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16. Abstract <p>Three computer programs have been developed or adapted to assist in the design of multi-beam prestressed concrete box girder bridges. Programs DBOXSS and DBOXDS treat girders with straight and draped strands, respectively. Each program has a "design" option which selects concrete release strength and strand pattern for a specified cross section and 28-day concrete strength to minimize the total number of strands used. The programs also contain an "optimization" option which determines release and 28-day concrete strengths and strand pattern that minimize the total cost of the girder. An analysis program AMBB has also been developed to compute lateral load distribution factors for the members of a multi-beam bridge. Specifications governing the designs produced are from the American Association of State Highway and Transportation Officials, 1973 Bridge Specification and 1974 and 1975 Interim Specification.</p>			
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AUTOMATED DESIGN OF PRESTRESSED
CONCRETE BOX GIRDERS

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PREFACE

This report consists of six Chapters and four Appendices. For those interested in the underlying mathematical formulations, Chapters II and III develop the design of straight and draped strand beams as mathematical programming problems. For those interested in how to use the straight and draped strand design computer programs, complete input instructions and output interpretation are presented in Chapters IV and V. Each of these two Chapters is self-contained and can be understood without referring to other sections of the report. Chapter VI is concerned with the analysis of multi-beam bridges, and deals primarily with instructions on the use of a computer program. This Chapter is also independent of others in the report. The Appendices deal with program documentation. Should the user wish to modify the programs, he will find subroutine descriptions, variable definitions and flow charts in Appendices B, C and D. Appendix E contains a listing of each program as it existed at the time of this report.

The equations required for problem formulation are extensive, since they are developed in their entirety. For clarity, highlighters (solid arrows) have been attached to those equations in the text which are the end result of manipulating preceding equations or which are especially significant.

Recently the Texas Highway Department (THD) became a part of the Texas Department of Highways and Public Transportation (TDHPT). References in the text to THD pertain to this latter organization.

ABSTRACT

Three computer programs have been developed or adapted to assist in the design of multi-beam prestressed concrete box girder bridges. Programs DBOXSS and DBOXDS treat girders with straight and draped strands, respectively. Each program has a "design" option which selects concrete release strength and strand pattern for a specified cross section and 28-day concrete strength to minimize the total number of strands used. The programs also contain an "optimization" option which determines release and 28-day concrete strengths and strand pattern that minimize the total cost of the girder. An analysis program AMBB has also been developed to compute lateral load distribution factors for the members of a multi-beam bridge. Specifications governing the designs produced are from the American Association of State Highway and Transportation Officials, 1973 Bridge Specification and 1974 and 1975 Interim Specification.

DISCLAIMER

The contents of this report reflect the views of the authors who are responsible for the facts and the accuracy of the data presented herein. The contents do not necessarily reflect the official views or policies of the Federal Highway Administration. This report does not constitute a standard, specification or regulation.

SUMMARY

This report presents formulations for the automated design and analysis of multi-beam prestressed concrete box girder bridges and documentation of the computer programs implementing these formulations. Simple span box girders of specified cross sectional dimensions are considered. Computer programs DBOXSS and DBOXDS treat girders containing straight and draped strands, respectively. The design variables determined by the programs include the number of strands in each strand row, concrete release and 28-day strengths and stirrup spacing. For straight strand designs, the extent and location of bond breakage is determined and for draped strand designs, the end eccentricity of the strands are computed. Each program has a "design" option which, for a specified 28-day strength, determines the strand pattern and release strength which minimizes the total number of strands used. An "optimization" option is also available with each program which determines release and 28-day concrete strengths and strand patterns that minimizes the total cost of the girder, based on the costs of concrete and strands supplied by the user.

Specifications governing design are those of the American Association of State Highway and Transportation Officials, 1973 Bridge Specifications and 1974 and 1975 Interim Specifications. Design restrictions include limits on release and service load stresses, upper and lower bounds on camber at release, ultimate and cracking moment capacities and maximum and minimum concrete strengths.

An existing computer program for the rigorous analysis of multi-beam bridges has been modified to compute lateral distribution factors for maximum moment for individual beams in a multi-beam bridge. Standard

AASHTO truck and lane loadings as well as arbitrary multi-axle vehicles can be treated by the program AMBB.

All programs have standard, simplified input forms and concise output formats. The computer core requirements for the programs in source form are 170,000 bytes for DBOXDS, 264,000 for DBOXSS and 294,000 for AMBB.

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RECOMMENDATION FOR IMPLEMENTATION

These computer programs are available to assist the bridge designer in carrying out the routine calculations associated with his job. In addition, their optimization options automatically produce the optimum design under a rather restricted set of conditions. Because of the rapidity with which proposed designs can be processed, these programs will permit designers to explore a wider range of possible solutions to a design problem. The programs should be equally useful for routine designs utilizing standardized cross sections as well as to explore new concepts for possible future standardization.

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I. INTRODUCTION

The simple span multi-beam prestressed concrete box girder bridge is a special use structure which may be the most economical selection when traffic disruptions, limited clearance, or other unusual conditions exist. This type of bridge construction consists of a number of box girders (not necessarily of identical cross sectional dimensions) laid side by side across the bridge bents. Lateral continuity between the girders is established by placing a concrete key (Figure 1) and transverse post tensioning at one or more points along the span. The bridge is usually completed with the addition of an asphalt wearing surface.

The design of this type of bridge requires the selection of cross sectional dimensions of the box girder(s) to be used, the release and 28 day strengths of the concrete, the number and placement details of the prestressing strands, the spacing of stirrups and designation of other conventional reinforcing details. A number of "standard" box shapes have been established by various states, including Texas, and fabricators in those states usually have considerable capital invested in steel forms and other hardware peculiar to the standard beams used in highway construction there. Thus, it is generally necessary for the designer to utilize standard box girder dimensions (with the possible exceptions of box width or void size) in order to gain maximum economy. Therefore, the question of what cross sectional dimensions to use for a particular design has not been addressed in the research efforts reported herein. Likewise, the design of conventional reinforcing details (with the exception of stirrups)

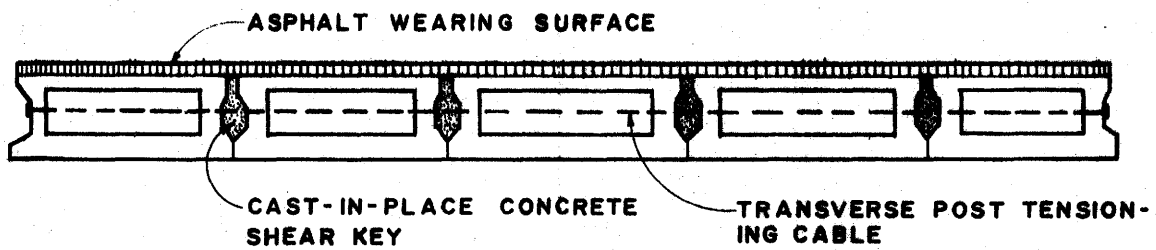


FIGURE 1. CROSS SECTION OF TYPICAL MULTI-BEAM PRESTRESSED CONCRETE BOX GIRDER BRIDGE

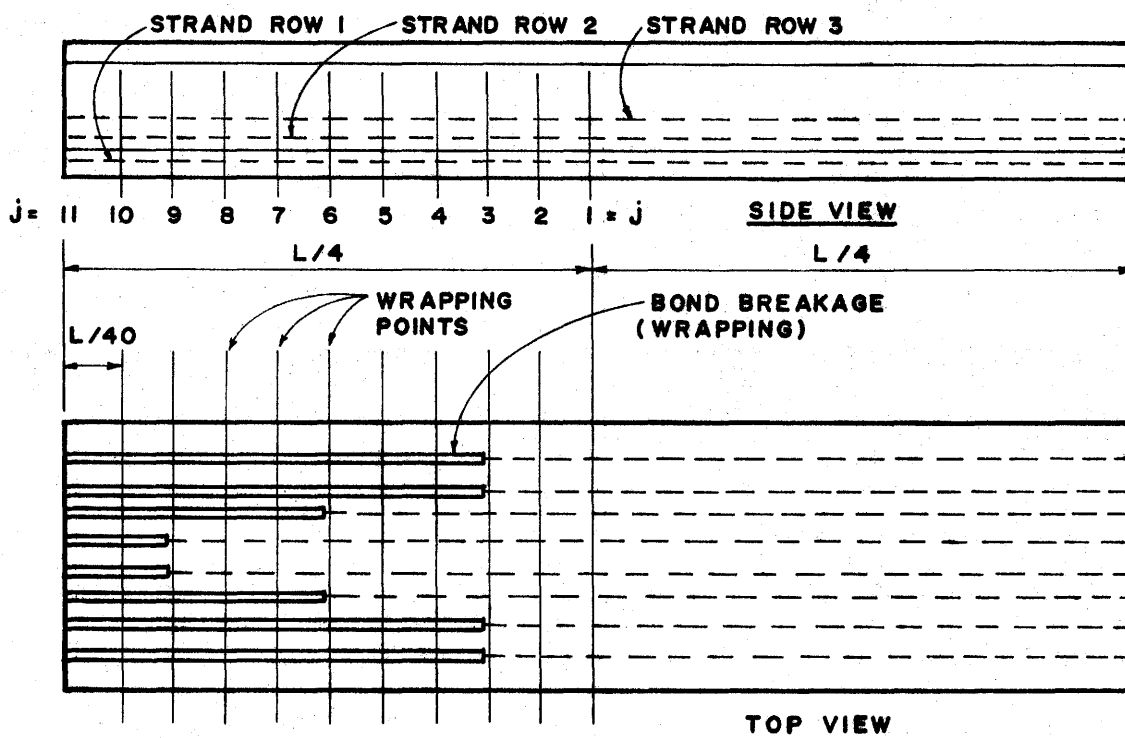


FIGURE 2. NOTATION FOR DESCRIPTION OF BOND BREAKAGE ON PRESTRESSED STRANDS

has been omitted from consideration due to the difficulty of insuring that all necessary reinforcing can be fitted into a particular box design and fabrication still be feasible. While the design items listed above that were omitted from consideration are by no means unimportant, those items which remain lend themselves to an automated (computer designed) approach.

Two distinct types of prestressing cable arrangements are commonly used in box girders. The first incorporates straight cables, which may have bond breakage near the ends of the beam to control stresses and camber. The second utilizes draped cables, in combination with straight cables, to attain the same controls. The latter arrangement rarely uses bond breakage unless end splitting problems are encountered during fabrication. When bond breakage is used to alleviate end splitting, it generally only extends over a few feet adjacent to the ends of the beam. Two automated design computer programs were developed in this study to treat the two types of cable layouts. The first has been given the name DBOXSS (Design of Box girders with Straight Strands) and the other DBOXDS (Design of Box girders with Draped Strands). Each of these programs has two options available to the designer. The "optimization" option automatically selects the minimum cost design, based on the costs of concrete and prestressing strands used. If the designer wishes to exert more control over the design or does not have at hand the unit cost information required by the optimization option, he may select the "design" option which computes a design based on the minimum number of strands that can be used. The underlying mathematical formulations used in both the optimization and design options are taken from the theories of linear and integer programming. Chapters II and III of this report present a brief description of the mathematical structure

of these optimization formulations and develops the design of straight and draped strand box beams in these formats. The input to the programs has been simplified through the use of standard input forms. A description of program input and interpretation of output is contained in Chapters IV and V for DBOXSS and DBOXDS, respectively. Also presented there are several example problems to assist the designer in understanding the use of the programs.

The design programs produce a design for a single box girder from a complete bridge. The fraction of the total load carried by the bridge which is assigned to the box girder under consideration is determined automatically by the current (1974 Interim) AASHTO Specification provision covering lateral load distribution in multi-beam bridges. This provision is empirical and the limits of its applicability can be examined in the research reports on which the provision is based. Situations frequently arise where the use of this means of determining lateral load distribution is questionable. To assist the designer in such cases, this study has adapted a third computer program AMBB (Analysis of Multi-Beam Bridges) which carries out a rigorous analysis of a multi-beam bridge and determines the fraction of total bridge live loads carried by each beam. The designer may thus choose to exercise this program first to obtain the lateral load distribution factors for beams in a proposed bridge and input them to the appropriate design program. This analysis program can also compute forces acting on the joints between beams which may be of assistance in designing concrete keys and transverse post tensioning. The program has a simplified standard input form which is described in Chapter VI, together with interpretation of program output and several example problems.

II. DESIGN OF BOX GIRDERS WITH STRAIGHT STRANDS

The design of prestressed concrete girders with straight strands can be cast as a linear, constrained optimization problem in which the design variables are concrete strengths and prestressing strand layout, and the constraints are restrictions on structural behavior outlined below.

Once the design problem has been cast in this format, standard computational procedures are available for its solution (1)*. The general form of the linear, constrained optimization problem (Linear Programming problem or LP problem) is:

$$\text{minimize } c_1x_1 + c_2x_2 + \dots + c_nx_n \quad (1)$$

$$\text{subject to: } a_{11}x_1 + a_{12}x_2 + \dots + a_{1n}x_n \leq b_1 \quad (2)$$

$$\begin{array}{ccccccc} \vdots & \vdots & & \vdots & \vdots \\ a_{m1}x_1 + a_{m2}x_2 + \dots + a_{mn}x_n & \leq & b_m \end{array}$$

$$x_1, x_2, \dots, x_n \geq 0.$$

where x_1, \dots, x_n are the variables, Eq. (1) the objective function and Eq. (2) the constraint set. This chapter is devoted to formulating the beam design problem in the mathematical form given above.

2.1 DESIGN CONSIDERATIONS

The arrangement of prestressing strands in a beam have a direct effect on the stresses at release and under service loads and on camber of the beam. The position of strands in the beam and the extent of bond breakage (also referred to as "wrapping") can be described by a doubly subscripted variable NS_{ij} for the general case shown in Figure 2. Here,

*Numerals in parenthesis refer to entries in the Reference section of this report.

wrapping is assumed to occur in lengths which are integer multiples of $L/40$, up to a maximum length of $L/4$. Wrapping is assumed to terminate just to the left of one of the 11 wrapping points. N_{ij} is defined as the number of bonded strands present in strand row i , at wrapping point j . In Figure 2, if the row shown in plain view were number 2, then

$$\begin{aligned} NS_{2,1} &= NS_{2,2} = NS_{2,3} = 8 \\ NS_{2,4} &= NS_{2,5} = NS_{2,6} = 4 \\ NS_{2,7} &= NS_{2,8} = NS_{2,9} = 2 \\ NS_{2,10} &= NS_{2,11} = 0. \end{aligned}$$

The wrapping of strands reduces the prestress induced stresses toward the end of the beam, where load induced stresses to offset them are small. The total stress at the top and bottom of the beam (taking tension stress as positive) at release at a wrapping point can be written as

$$\sigma_j^{(T)} = (1 - \xi) F_o \sum_{i=1}^{NR} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,j} - \frac{M_j}{Z_t} \quad (3)$$

$$\sigma_j^{(B)} = (1 - \xi) F_o \sum_{i=1}^{NR} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,j} + \frac{M_j}{Z_b} \quad (4)$$

where

NR = number of rows which can contain strands,

F_o = force in a single prestress strand prior to release,

ξ = fraction of initial prestress force lost immediately after release,

A = cross sectional area of the box girder,

d_i = distance from the c.g. axis of the beam to strand row i (positive if row i above c.g. of beam),

Z_t & Z_b = section moduli of beam (both positive quantities),

M_j = bending moment at point j due to beam weight.

Stresses at any location between quarter points can be obtained by setting subscript $j = 1$ and replacing M_j with the moment at that location. Equations (3) and (4) ignore the effect of strand development length. The cross sectional properties A and Z include the transformed area of conventional compression reinforcing in the top of the beam, if present. The stresses existing under service load conditions can be computed from

$$\sigma_j^{(T)} = (1 - \eta) F_{o \sum_{i=1}^{NR}} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,j} - \frac{M_j}{Z_t} - \frac{\bar{M}_j}{\bar{Z}_t} \quad (5)$$

$$\sigma_j^{(B)} = (1 - \eta) F_{o \sum_{i=1}^{NR}} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,j} + \frac{M_j}{Z_b} + \frac{\bar{M}_j}{\bar{Z}_b} \quad (6)$$

where

η = fraction of initial prestress force lost under service load conditions,

\bar{Z}_t & \bar{Z}_b = section moduli of beam plus shear key,

M_j = moment at point j due to beam and shear key weight

\bar{M}_j = total live and dead load moment acting on the composite section (i.e., with shear key) at point j .

Stresses at any section between quarter points ($j = 1$) can be obtained from Eqs. (3) thru (6) by setting subscript $j = 1$ and substituting the moments acting at that section for M_j and \bar{M}_j .

Camber control is an important consideration in the design of prestressed concrete box girders. If the beam camber is upward upon release, there is a tendency for the camber to increase with time due to creep and shrinkage effects in the concrete and because of the absence of significant additional dead load such as a deck slab. A downward camber on release may tend to become more downward with time. Although long term camber is the quantity the

designer seeks to control, its accurate computation is difficult. Generally accepted analytical means for its computation (2, 3) require a knowledge of the creep and shrinkage properties of the concrete, which in turn depend on the materials and mix design used as well as curing conditions. In the absence of accurate creep and shrinkage data, many designers rely on cambers computed at release as a guide to insuring satisfactory long term behavior. The release camber can be computed from previously defined quantities (see Figure 3) by

$$\Delta = \Delta_{DL} - \frac{1}{E_{ci} I} \left\{ (1 - \epsilon) F_o \sum_{j=1}^{11} h_j \delta_j y_j \right\} \quad (7)$$

where

$$h_j = \sum_{i=1}^{NR} d_i \cdot NS_{i,j} \quad (8)$$

$$y_j = \begin{cases} 3L/8 & ; j=1 \\ (11-j) \frac{L}{40} + \frac{L}{80} & ; j=2, \dots, 11 \end{cases} \quad (9)$$

$$\delta_j = \begin{cases} L/4 & ; j=1 \\ L/40 & ; j=2, \dots, 11 \end{cases}$$

and

Δ = midspan camber (positive upward),

Δ_{DL} = midspan deflection due to beam weight (positive upward),

E_{ci} = modulus of elasticity of concrete at release.

In addition to satisfactory behavior under release and service load conditions, a box girder must have adequate ultimate moment capacity. The current AASHTO Specification (4) requires that the computed ultimate moment capacity of a section M_u be not less than M_{ur} , where

$$M_{ur} = 1.30 \left\{ M_{DL} + \frac{5}{3} (I \cdot M_{LL}) \right\} \quad (10)$$

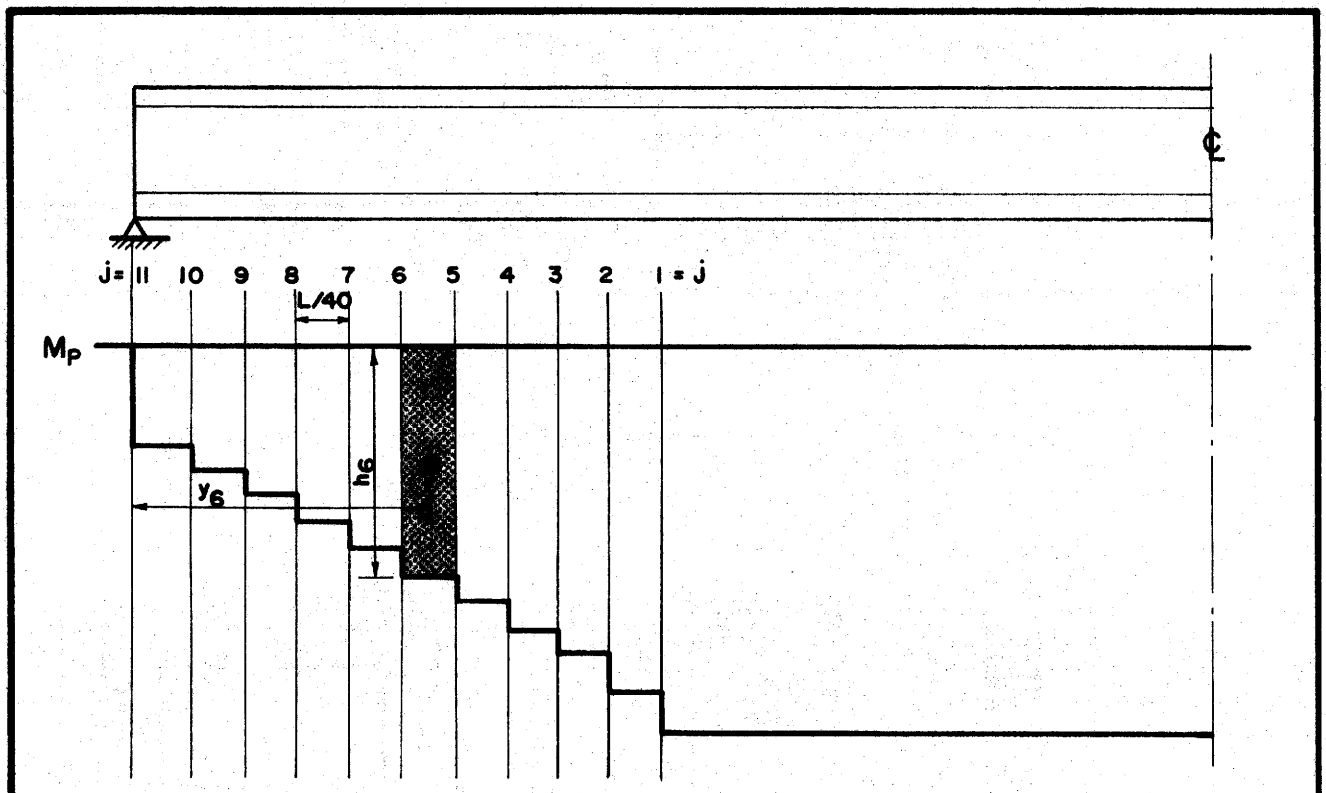


FIGURE 3. PRESTRESS INDUCED MOMENT DIAGRAM FOR CALCULATION OF RELEASE CAMBER IN BEAMS WITH STRAIGHT STRANDS

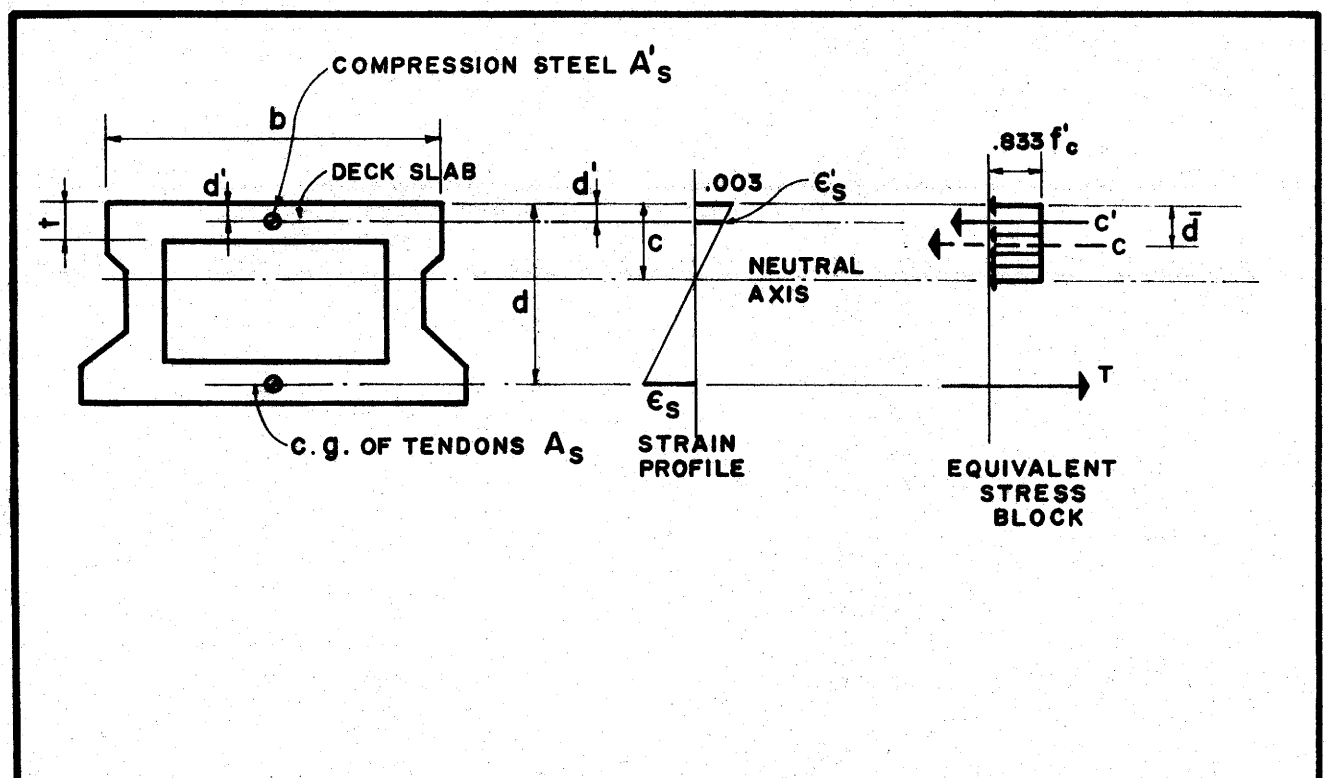


FIGURE 4. STRAIN PROFILE AND RESULTANT FORCES AT ULTIMATE MOMENT

M_{LL} = maximum live load moment,
 M_{DL} = dead load moment,
 $I = \frac{50}{(L+125)}$, the impact factor,
 L = span length (ft).

The method of computing ultimate moment capacity at midspan depends upon the location of the neutral axis in the cross section. If the neutral axis falls within the deck slab (Figure 4), then M_u is given by (4),

$$M_u = A_s^* f_{su}^* d \left\{ 1.0 - 0.6 \frac{p^* f_{su}^*}{f_c'} \right\} \quad (11)$$

where

$$f_{su}^* = f_s' \left\{ 1.0 - 0.5 \frac{p^* f_s'}{f_c'} \right\} \quad (12)$$

and

f_{su}^* = average stress in the prestressing strands at ultimate,

A_s^* = total area of prestressing steel,

f_s' = ultimate strength of strand,

f_c' = compressive strength of concrete,

$p^* = A_s^*/bd$,

b, d = (See Figure 4).

This computation neglects the contribution of the compression steel to moment capacity and is justified on the basis of the proximity of the compression reinforcing to the neutral axis when the latter lies in the deck slab. When the neutral axis lies below the slab, as indicated in Figure 4, then a trial and error approach is required to determine M_u . For a prescribed location of the neutral axis (the dimension c), the total compressive force in the concrete C is computed from

$$C = .833f'_c A_c \quad (13)$$

where A_c is the area between the neutral axis and the top of the section and \bar{d} is the location of the c.g. of this area. Equilibrium of horizontal forces requires that

$$C' + C = T \quad (14)$$

The force C' is the force in the compression steel, given by

$$C' = \begin{cases} \epsilon'_s E_s A'_s & ; \epsilon'_s \leq \text{yield strain} \\ f_y A'_s & ; \epsilon'_s > \text{yield strain} \end{cases} \quad (15)$$

The force in the tendons at ultimate is

$$T = f_{su}^* A_s \quad (16)$$

where f_{su}^* is the average tendon stress, determined from the stress-strain characteristics of the tendon material, which can be approximated by (5)

$$\epsilon_{su} = \begin{cases} f_{su}^*/E_s & ; f_{su}^* \leq f_{pl} \\ a_1[a_2 - a_3/f_{su}^*(f'_s - f_{su}^*)] & ; f_{su}^* > f_{pl} \end{cases} \quad (17)$$

where

ϵ_{su} = average strain in the prestress strands,

f_{pl} = proportional limit stress of strand material,

E_s = modulus of strand material,

$a_1 = f_{pl}/E_s$

$a_2 = 1. + (f'_s - f_{pl})/(f'_s - 2f_{pl})$

$a_3 = f_{pl}(f'_s - f_{pl})^2/(f'_s - 2f_{pl})$

If the total tensile force T exceeds the total compression $(C + C')$, then the neutral axis depth c is too small. If $(C + C')$ exceeds T , then the correct c value is less than that assumed. Once the proper c has been obtained, the ultimate moment capacity can be computed from

$$M_u = C'(d - d') + C(d - d') \quad (18)$$

The average compressive stress over the concrete compression zone $(.833f'_c)$, and the stress-strain relationship for the tendon (Eq. (17)) were derived on the condition that Eqs. (11) and (18) give the same moment capacity when $A'_s = 0$ and the neutral axis is located in the deck slab (5).

It will later prove useful to have a relationship between the strand pattern in a box girder and its ultimate moment capacity. Let

$$\rho = - \sum_{i=1}^{NR} d_i \cdot NS_{i,1} \quad (19)$$

define a positive parameter (the d_i are normally negative quantities) which is a measure of the total available strand force eccentricity. For a specified concrete strength f'_c , the ultimate moment capacity of a section M_u can be plotted against the parameter ρ , as shown schematically in Figure 5. For a specified required ultimate moment capacity M_{ur} , a minimum value of the strand force eccentricity $\bar{\rho}$ exists for each concrete strength. The plot of $\bar{\rho}$ vs. f'_c , shown in Figure 6, provides a convenient means of insuring that the final strand pattern and 28 day concrete strength selected will yield an adequate ultimate moment capacity.

Current prestress concrete design practice recognizes the importance of adequate warning of impending failure in an overloaded structural member. A natural means of achieving this end is to insure that signifi-

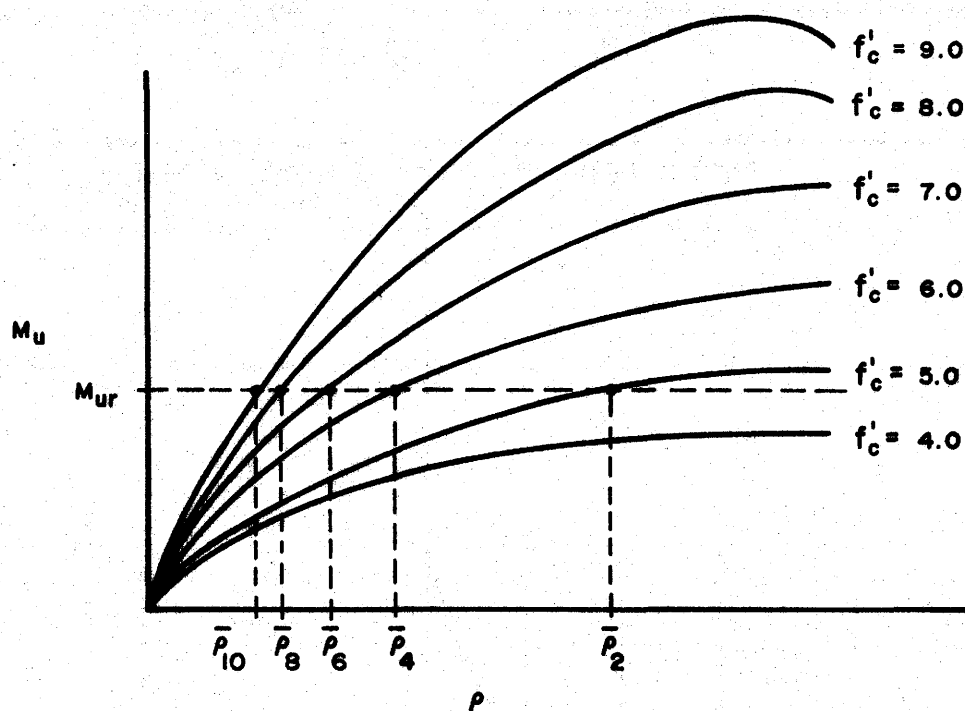


FIGURE 5. ULTIMATE MOMENT CAPACITY VS. TOTAL STRAND FORCE ECCENTRICITY

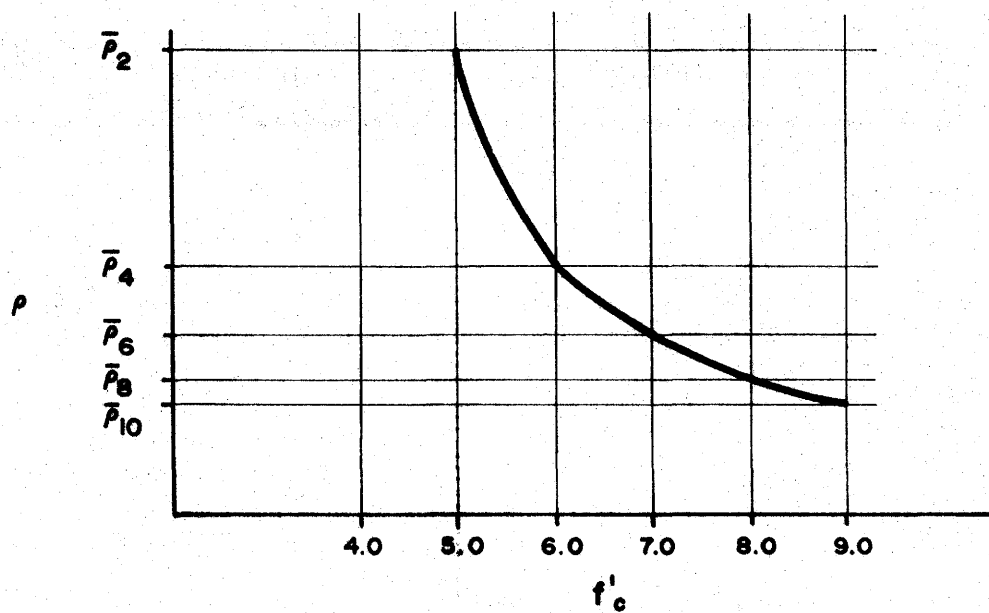


FIGURE 6. REQUIRED TOTAL STRAND FORCE ECCENTRICITY VS. 28 DAY CONCRETE STRENGTH

cant flexural cracking of the section occurs prior to failure. Thus, the AASHTO Specification (4) requires that

$$M_u \geq 1.2 \cdot M_{cr} \quad (20)$$

where M_{cr} is the moment required to produce a tensile stress at the bottom of the section equal to the modulus of rupture strength of the concrete.

The net prestress force in a strand at release and under service load conditions is dependent on the loss factors ξ and η (Eqs. 3 thru 6). The AASHTO Specification (6) provides a method of long term prestress loss calculation which includes all factors currently thought to have a significant effect. The loss may be written as

$$\eta = [SH + ES + CR_c + CR_s]/f_e \quad (21)$$

where

f_e = stress in strand immediately after initial tensioning (ksi),

SH = loss due to concrete shrinkage (ksi),

ES = loss due to elastic shortening (ksi),

CR_c = loss due to creep of concrete (ksi),

CR_s = loss due to relaxation of prestressing strand (ksi).

The four components of prestress loss are computed from

$$SH = 17.0 - 0.15RH \quad (22)$$

$$ES = E_s f_{cir}/E_{ci} \quad (23)$$

$$CR_c = 12f_{cir} - 7 f_{cds} \quad (24)$$

$$CR_s = 20. - .4ES - .2(SH + CR_c) \quad (25)$$

where

RH = average annual relative humidity in percent,

E_s = modulus of elasticity of prestress strand,

E_{ci} = modulus of elasticity of concrete at time of strand release,

f_{cir} = concrete stress at c.g. of strands due to prestress force immediately after release and beam weight. The stress is computed at the point of maximum moment.

f_{cds} = concrete stress at c.g. of strands due to all dead loads except those present at release (i.e., beam weight).

The prestress loss immediately after release can be estimated from

$$\xi = [ES + .5CR_s]/f_e \quad (26)$$

The fraction of the total live load on a multi-beam bridge that is carried by a single box girder must be determined prior to design. In the absence of a rigorous analysis, the AASHTO Specification (8) suggests the following empirical estimate:

$$S/D = \text{fraction of } \underline{\text{axle}} \text{ load carried by the girder} \quad (27)$$

where

$$S = 0.5(12N_L + 9)/N_g \quad (28)$$

$$D = \begin{cases} 5 + N_L/10 + (3-2N_L/7)(1-C/3)^2 & ; C \leq 3 \\ 5 + N_L/10 & ; C > 3 \end{cases} \quad (29)$$

and

N_L = total number of traffic lanes,

N_g = number of longitudinal beams,

$C = K(W/L)$, a stiffness parameter,

$K = 1$. for box sections,

W = overall bridge width,

L = span length.

A rigorous analysis for multi-beam bridges has been developed by Ghose and Powell (9) and programmed by Ghose. The method is based on Fourier series expansion representations of applied loads and individual beam

responses and compatibility of displacements at the juncture of adjacent beams. The computer program has been obtained from the authors and modified so that lateral distribution factors for each box girder in a bridge are automatically computed for AASHTO truck and lane loads and for an arbitrary axle train configuration. The designer may use this program to determine lateral load distribution factors in lieu of those computed from Eq. (27). The details of the program's use are contained in Chapter VI.

Stirrup requirements are computed from current AASHTO Specification provisions (4). The stirrup spacing s_j at the j th tenth point is given by

$$s_j = 2A_v f_{sy} J d_j / (V_U^{(j)} - V_C^{(j)}) \quad (30)$$

where

$$V_C^{(j)} = 0.06 f'_c b_j d_j \leq 180 b' d_j \quad (31)$$

$$V_U^{(j)} = \frac{1.30}{\phi} \left\{ V_{DL}^{(j)} + \frac{5}{3} (I \cdot V_{LL}^{(j)}) \right\} \quad (32)$$

and

s_j = stirrup spacing at i th tenth point,

A_v = area of stirrup,

f_{sy} = yield strength of stirrups,

b' = total width of beam web,

d_j = distance from c.g. of strands to top of section at i th tenth point,

J = fraction of d_j which gives the distance from the center of compression to the c.g. of strands; taken as 0.9,

$V_{DL}^{(j)}$ = total dead load shear at j th tenth point,

$V_{LL}^{(j)}$ = total live load shear at j th tenth point,

ϕ = strength factor, taken as 0.9,

$I = \frac{50}{L + 125}$, the impact factor,

L = span length (ft).

The "best" prestressed concrete beam (whose structural behavior is satisfactory) is the one with the lowest bid price. Bid price is influenced by some factors over which the designer has control and by others which he can not control. The latter category includes differences in pricing procedures among fabricators and little correlation between the geographical location of a bridge and the fabricator who produces the beams for it. However, despite the uncontrollable nature of some factors, it is believed that the cost model developed below provides a means of ranking beams according to expected bid price.

The final cost of a beam is assumed to consist of the cost of concrete, cost of strand and cost of strand wrapping. The cost of concrete is primarily a function of release strength. Higher release strengths require some additional materials (cement, admixtures, etc.) but the principle cause for increased cost is the additional curing time needed. This trend is evident from the results of a survey of producers of highway beams in the state of Texas. Four responses to the questionnaire shown in Appendix A were received. The questionnaire asked the fabricator to list the in-place cost of concrete with release strengths ranging from 4.0 to 8.0 ksi, assuming that the cost of 4.0 ksi release strength concrete is \$1.00/cu.yd. This method of cost presentation was used in an attempt to circumvent fabricators' natural reluctance to divulging actual cost information. Concrete cost is plotted against release strength in Figure 7 for the four responses received.

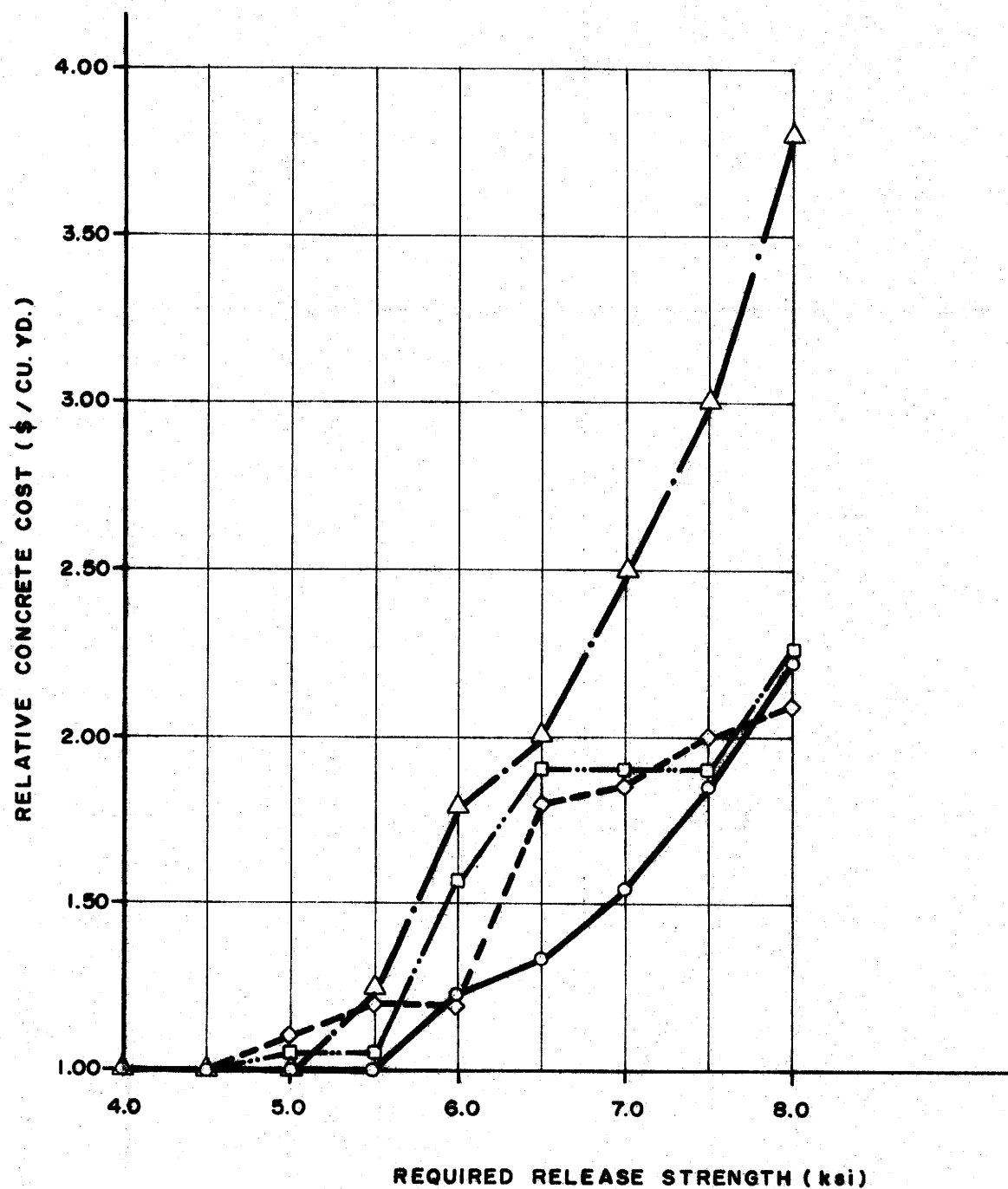


FIGURE 7. RELATIVE CONCRETE COST VS. RELEASE STRENGTH
FOR FOUR TEXAS FABRICATORS

Costs associated with prestressing strands consists of the cost of materials and cost of placing strands and wrapping them. The reported in-place cost of 1/2 in. diameter grade 270K 7 wire strand ranged from \$0.20 to \$0.25 per foot. Unfortunately, no cost figures on strand wrapping were sought in the questionnaire. Additional consideration is given to determining cost figures for design in Chapter IV.

Although the release strength f'_{ci} and 28-day strength f'_c of beam concrete frequently are treated as independent parameters in design, fabrication practices indicate a strong correlation between the two quantities. Most fabricators have a relatively small number of mix designs which are used to cover the usual range of required strengths. For a specific mix design, the release strength may vary considerably, depending on the method and length of time of curing, but the 28-day strength attained is largely independent of these factors. Thus, if one specifies a release strength of 6.0 ksi, and a 28-day strength of 6.5 ksi, he may actually get an f'_c of 7.0 ksi, depending on the fabricator involved. Thus, the design does not take full advantage of the concrete strength available under service load conditions. For a particular fabricator, one can generally construct a plot of f'_{ci} vs. f'_c whose general form will follow that shown in Figure 8.

2.2 STRAIGHT STRAND DESIGN FORMULATION - OPTIMIZATION OPTION

In this section the problem of determining the concrete release and 28-day strengths and strand pattern layout which minimizes the total cost of a box girder is formulated as a linear programming problem whose mathematical structure was given in Eqs. (1) and (2). The notation used here differs slightly from Eqs. (1) and (2) in that the design variables x_1, \dots, x_n

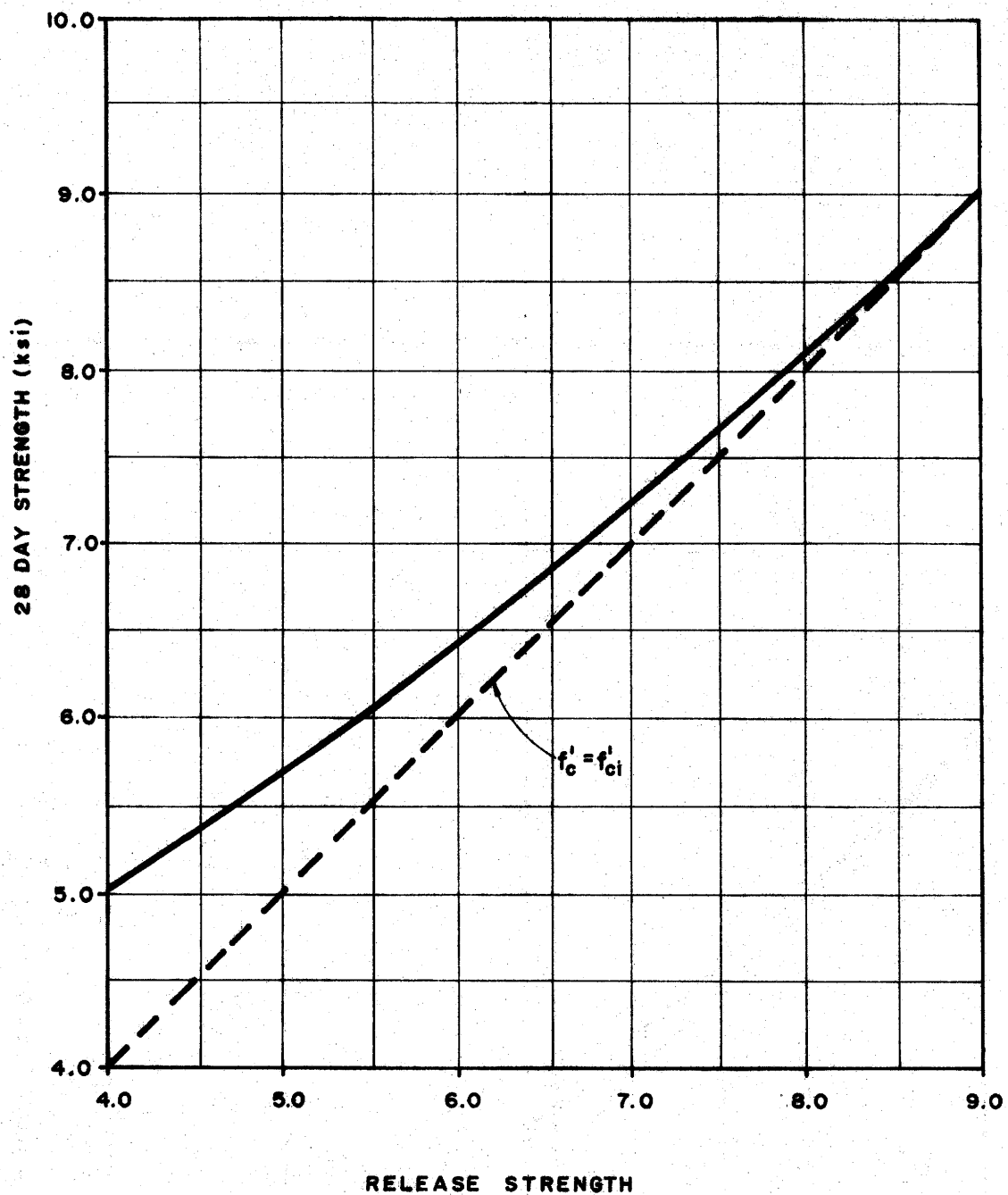


FIGURE 8. HYPOTHETICAL 28 - DAY RELEASE STRENGTH RELATIONSHIP

are represented by symbols defined in the preceding section in order to more clearly preserve the physical significance of the equations.

Before proceeding to the formulation of the objective function and constraints, additional notation must be introduced. Let the concrete release strength f'_{ci} be given by:

$$f'_{ci} = 4.0 + \sum_{i=1}^{10} f_i \quad (33)$$

where f_1, \dots, f_{10} are design variables whose values satisfy the inequalities:

$$0 \leq f_i \leq 0.5 \quad i=1, \dots, 10 \quad (34)$$

$$f_{i+1} \leq f_i \quad i=1, \dots, 9 \quad (35)$$

Note that by this definition, any release strength between 4.0 ksi ($f_1=f_2=\dots=f_{10}=0$) and 9.0 ksi ($f_1=f_2=\dots=f_{10}=0.5$) is admissible. For example, a release strength of 5.35 ksi would result if $f_1=f_2=0.5$, $f_3=0.35$, $f_4=f_5=\dots=f_{10}=0$. A minimum release strength of 4.0 ksi was selected to conform with current AASHTO standards, while an upper limit of 9.0 ksi was selected because it is at the extreme upper limits of concrete strength which fabricators in the state of Texas are able to produce. The expression for f'_{ci} given in Eq. (33) was derived on the basis of a need to maintain linearity in objective function and constraint equations which follow.

2.2.1 Objective Function

The total cost of the box girder is assumed to be the sum of concrete cost C_c , strand cost C_s and strand wrapping cost C_w . As developed in Section 2.1, the cost of concrete is assumed to be a function of release strength. Let c_0, c_1, \dots, c_{10} denote the cost of concrete with $f'_{ci} = 4.0, 4.5, \dots, 9.0$ ksi. Then the cost of one cubic yard of concrete can be written

as

$$c_0 + 2 \sum_{i=1}^{10} (c_i - c_{i-1}) f_i \quad (36)$$

where f_i are defined in Eqs. (33) thru (35). Note that Eq. (36) assumes a piecewise linear variation in concrete cost, as shown in Figure 9. The total cost of the concrete is then given by

$$C_c = \frac{A \cdot L}{3888} \left\{ c_0 + 2 \sum_{i=1}^{10} (c_i - c_{i-1}) f_i \right\} \quad (37)$$

where A is the area of the section in square inches and L is the length of the beam in feet. Note that Eq. (37) neglects additional concrete used in forming interior diaphragms and end closures.

The total number of strands used in the beam is given by $\sum_{i=1}^{NR} NS_{i,1}$. If c_s is the cost per foot of strand, the total strand cost is given by

$$C_s = c_s L \sum_{i=1}^{NR} NS_{i,1} \quad (38)$$

Taking c_w as the cost per foot of strand wrapping, the total cost of wrapping strands may be written as

$$C_w = c_w \left\{ .5L \sum_{i=1}^{NR} NS_{i,1} - \frac{2L}{40} \sum_{j=2}^{11} \sum_{i=1}^{NR} NS_{i,j} \right\} \quad (39)$$

Thus, the total cost of the beam (objective function) becomes

$$\begin{aligned} \text{Minimize } & (c_s L + .5c_w L) \sum_{i=1}^{NR} NS_{i,1} - \frac{2c_w L}{40} \sum_{j=2}^{11} \sum_{i=1}^{NR} NS_{i,j} \\ & + \frac{A \cdot L}{1944} \sum_{i=1}^{10} (c_i - c_{i-1}) f_i + \frac{A \cdot L \cdot c_0}{3888} \end{aligned} \quad (40)$$

Equation (40) is linear in the design variables $NS_{i,j}$ and f_i as required by Eq. (1).

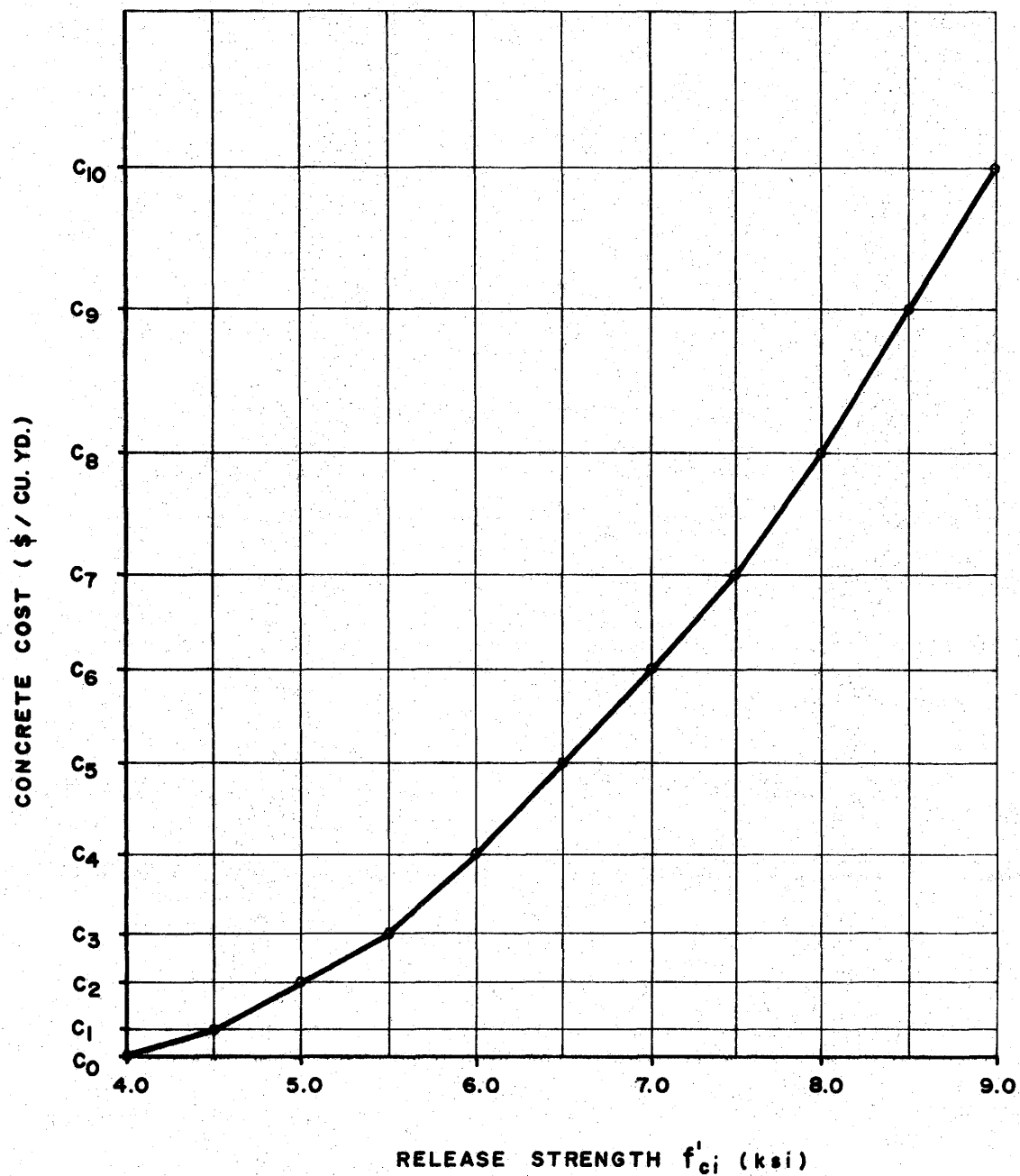


FIGURE 9. LINEARIZED REPRESENTATION OF CONCRETE COST VS. RELEASE STRENGTH

2.2.2 Constraints on Release Stresses [Constraints 1 thru 12]

When the strands are released, stresses are produced in the beam by prestress force and the weight of the beam. These stresses are generally tensile at points along the top of the beam and compressive at the bottom. In order to prevent damage to the beam, stresses must be held within certain limits which are a function of the concrete release strength. Generally the stress limits (in ksi) are of the form

$$\sigma_{tj} = .031623 S_{tj} \sqrt{f'_{ci}} \quad (f'_{ci} \text{ in ksi}) \quad (41)$$

$$\sigma_{cj} = S_{cj} f'_{ci} \quad (42)$$

where S_{tj} and S_{cj} are constants, which in general may vary from point to point along the beam (hence the subscript j). The AASHTO Bridge Specifications (4) currently stipulate that $S_{tj} = 7.5$ and $S_{cj} = 0.6$. The square root in Eq. (41) causes difficulties since it applies to f'_{ci} which is a design variable and thus introduces a nonlinearity into the formulation. With little error, Eq. (41) can be written as

$$\sigma_{tj} = S_{tj} \left\{ .007454 f'_{ci} + .03355 \right\} \quad (f'_{ci} \text{ and } \sigma_{tj} \text{ in ksi}) \quad (43)$$

Equation (43) is obtained from Eq. (41) by replacing $\sqrt{f'_{ci}}$ with a first order Taylor series expansion about the point $f'_{ci} = 4.5$ ksi. The error in this expression is 4.2% at $f'_{ci} = 8.0$ ksi, and decreases as f'_{ci} approaches 4.5 ksi.

Release stresses are checked top and bottom at the end of the beam ($j = 11$ in Figure 2), $L/20$ ($j = 9$), $2L/20$ ($j = 7$), $3L/20$ ($j = 5$), $4L/20$ ($j = 3$) and $L/4$ ($j = 1$). For points on the top of the beam, tensile stress is limited to σ_{tj} by

$$\sigma_j^{(T)} - \sigma_{tj} \leq 0 \quad j=1, 3, \dots, 11 \quad (44)$$

Substituting Eqs. (3) and (43) into (44) gives

$$\begin{aligned} (1 - \xi) F_{\sum_{i=1}^{NR}} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,j} - .007454 S_{tj} f'_{ci} \\ \leq \frac{M_j}{Z_t} + .03355 S_{tj} \quad ; j=1, 3, \dots, 11 \end{aligned} \quad (45)$$

Replacing f'_{ci} in Eq. (45) with Eq. (33) results in the following linear inequality constraint in the design variables:

$$\begin{aligned} (1 - \xi) F_{\sum_{i=1}^{NR}} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,j} - .007454 S_{tj} \sum_{i=1}^{10} f_i \\ \leq \frac{M_j}{Z_t} + .06337 S_{tj} \quad ; j=1, 3, \dots, 11 \end{aligned} \quad (46)$$

Letting j in Eq. (46) range over 1, 3, 5, ..., 11 produces 6 constraints which limit release stresses in the top of the beam to the tensile allowable. For points on the bottom of the beam, compression stress is limited to σ_{cj} by

$$-\sigma_j^{(B)} - \sigma_{cj} \leq 0 \quad j=1, 3, \dots, 11 \quad (47)$$

Substituting Eqs. (4), (33) and (42) into (47) yields

$$\begin{aligned} -(1 - \xi) F_{\sum_{i=1}^{NR}} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,j} - S_{cj} \sum_{i=1}^{10} f_i \\ \leq \frac{M_j}{Z_b} + 4.0 S_{cj} \quad ; j = 1, 3, \dots, 11 \end{aligned} \quad (48)$$

Six constraints limiting compression stress in the bottom of the beam result from taking $j=1, 3, \dots, 11$.

2.2.3 Constraints on Service Load Stresses [Constraints 13 thru 20]

Let

$$\bar{\sigma}_{tj} = .031623 \bar{S}_{tj} \sqrt{f'_c} \quad (f'_c \text{ in ksi}) \quad (49)$$

$$\bar{\sigma}_{cj} = \bar{S}_{cj} f'_c \quad (50)$$

denote the allowable tension and compression stresses under service load conditions. Stress checks are made on compression in the top and tension in the bottom of the beam at midspan, 2/10 and 1/10 points, and for tension in the top and compression in the bottom at the end of the beam. Using the Taylor series expansion to eliminate the radical, Eq. (49) becomes

$$\bar{\sigma}_{tj} = \bar{S}_{tj} \left\{ .007454 f'_c + .03355 \right\} \quad (f'_c \text{ and } \bar{\sigma}_{tj} \text{ in ksi}) \quad (51)$$

The 28 day strength f'_c depends on release strength in a manner depicted in Figure 8. A piecewise continuous linear relationship between f'_c and f'_{ci} is given by

$$f'_c = g_0 + 2 \sum_{i=1}^{10} (g_i - g_{i-1}) f_i \quad (52)$$

Where g_0, g_1, \dots, g_{10} are the 28 day strengths which correspond to release strengths of 4.0, 4.5, ..., 9.0 ksi (see Figure 10). Substitution of Eq. (52) into Eq. (51) gives

$$\bar{\sigma}_{tj} = \bar{S}_{tj} \left\{ .007454 \left[g_0 + 2 \sum_{i=1}^{10} (g_i - g_{i-1}) f_i \right] + .03355 \right\} \quad (53)$$

Tension stresses in the bottom of the beam are limited by

$$\bar{\sigma}_j^{(B)} - \bar{\sigma}_{tj} \leq 0 \quad (j=0, 3, 7) \quad (54)$$

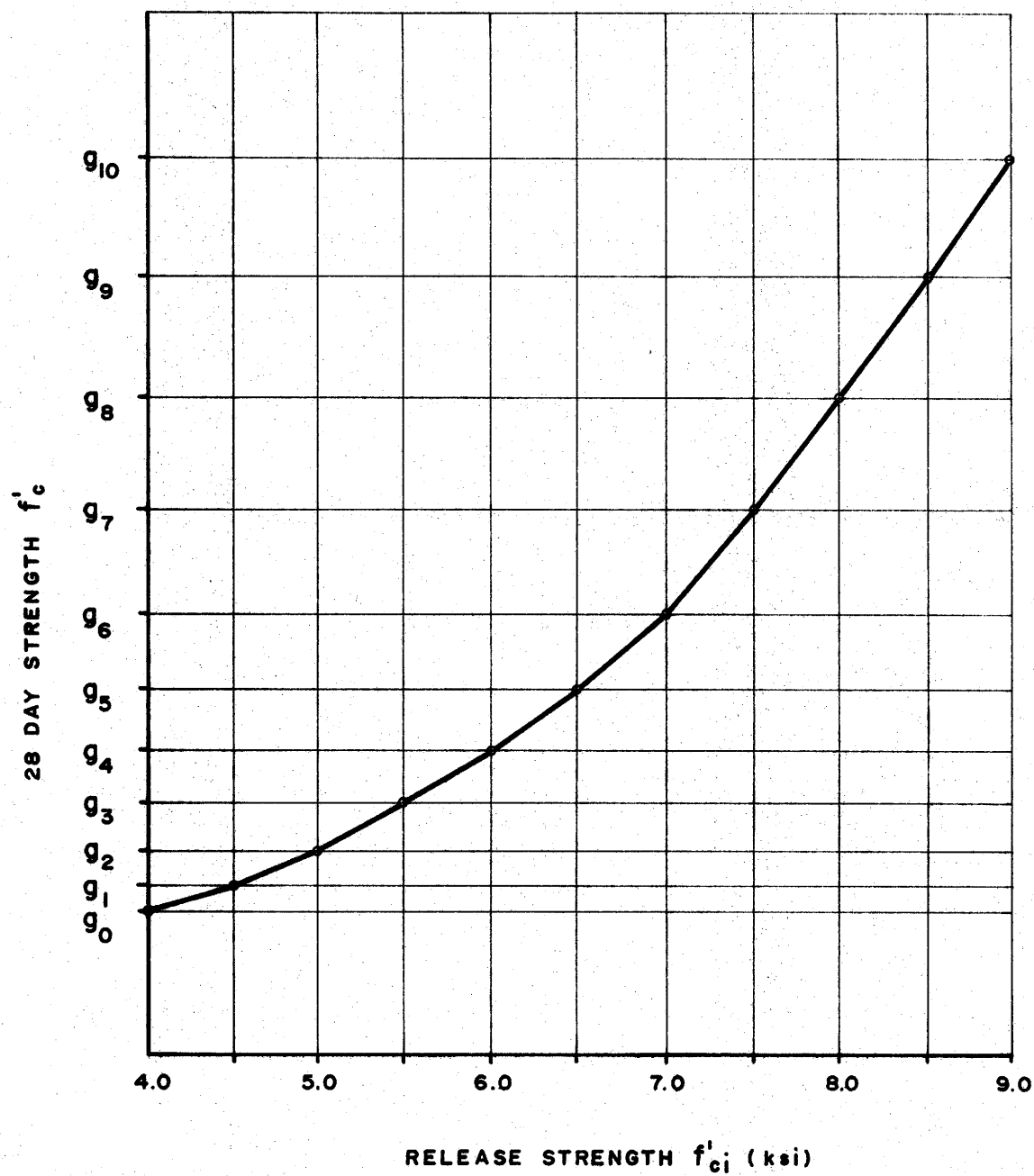


FIGURE 10. LINEARIZED REPRESENTATION OF 28 DAY VS. RELEASE STRENGTH

The subscript value $j = 0$ denotes midspan of the beam. Substitution of Eqs. (5) and (53) into (54) gives

$$(1 - \eta)F_{o_{i=1}}^{NR} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,j} - .01491 \bar{S}_{tj_{i=1}}^{10} (g_i - g_{i-1}) f_i \\ \leq - \frac{M_j}{Z_b} - \frac{\bar{M}_j}{\bar{Z}_b} + .007454 \bar{S}_{tj_0} + .03355 \bar{S}_{tj} \quad (j=1, 3, 7) \quad (55)$$

A discrepancy in notation exists between Eqs. (54) and (55). In the former, the subscript j takes values of 0, 3, 7 while in the latter, $j = 1, 3, 7$. The use of $j = 1$ indicates that prestress force induced stress is computed with bonded strands at the quarter point (i.e., $NS_{i,1}$), which is valid since the number of bonded strands there is the same as that at midspan. The moments M_j and \bar{M}_j however, should be replaced with those occurring at midspan when $j = 1$.

The tensile stress in the top of the beam at the end is limited to $\bar{\sigma}_{t11}$ by

$$\bar{\sigma}_{11}^{(T)} - \bar{\sigma}_{t11} \leq 0 \quad (56)$$

Noting that the load induced stresses are zero at the end, Eq. (56) becomes

$$(1 - \eta)F_{o_{i=1}}^{NR} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,11} - .01491 \bar{S}_{t11_{i=1}}^{10} (g_i - g_{i-1}) f_i \\ \leq .007454 \bar{S}_{t11_0} + .03355 \bar{S}_{t11} \quad (57)$$

The compression stress in the top of the beam under service load is limited by

$$\bar{\sigma}_j^{(T)} - \bar{\sigma}_{cj} \leq 0 \quad (j=0, 3, 7) \quad (58)$$

The allowable compression stress $\bar{\sigma}_{cj}$ is

$$\bar{\sigma}_{cj} = \bar{S}_{cj} f'_c \quad (59)$$

which, after the substitution of Eq. (52) becomes

$$\bar{\sigma}_{cj} = \bar{S}_{cj} \left\{ g_0 + 2 \sum_{i=1}^{10} (g_i - g_{i-1}) f_i \right\} \quad (60)$$

Replacing $\bar{\sigma}_j^{(T)}$ with Eq. (5) and $\bar{\sigma}_{cj}$ with Eq. (60) yields

$$\begin{aligned} & -(1 - \eta) F_o \sum_{i=1}^{NR} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,j} - 2 \bar{S}_{cj} \sum_{i=1}^{10} (g_i - g_{i-1}) f_i \\ & \leq \frac{M_j}{Z_t} - \frac{\bar{M}_j}{Z_t} + \bar{S}_{cj} g_0 \quad (j=1, 3, 7) \end{aligned} \quad (61)$$

As in Eq. (55), when $j=1$, M_j and \bar{M}_j are taken as the moments at midspan.

The compression stress at the bottom of the beam at its end is limited to σ_c by

$$-\sigma_{11}^{(B)} - \sigma_{c11} \leq 0 \quad (62)$$

Substituting Eq. (6) for $\bar{\sigma}_{11}$ and Eq. (60) for $\bar{\sigma}_{c11}$ and noting that load induced stresses are zero at the end of the beam, gives

$$\begin{aligned} & -(1 - \eta) F_o \sum_{i=1}^{NR} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,11} - 2 \bar{S}_{c11} \sum_{i=1}^{10} (g_i - g_{i-1}) f_i \\ & \leq \bar{S}_{c11} g_0 \end{aligned} \quad (63)$$

2.2.4 Constraints to Insure Proper Strand Wrapping [Constraints 21 thru (20 + 10·NR)]

Bond breakage is initiated at the end of the beam and proceeds toward the quarter point (see Figure 2). The variables $NS_{i,j}$ give the number of bonded strands present in the i th strand row at point j . If

wrapping begins at the end of the beam and terminates just to the left of one of the wrapping points (denoted by j), then $NS_{i,j}$ must be greater than or equal to $NS_{i,j+1}$ for all rows and wrapping points. This requirement is imposed through the $(10 \cdot NR)$ inequality constraints.

$$\begin{aligned} NS_{i,2} - NS_{i,1} &\leq 0 \\ NS_{i,3} - NS_{i,2} &\leq 0 \\ \vdots \\ NS_{i,11} - NS_{i,10} &\leq 0 \end{aligned} \quad (i = 1, 2, \dots, NR) \quad (64)$$

2.2.5 Constraints Limiting the Number of Strands in Each Row [Constraints $(21 + 10 \cdot NR)$ thru $(20 + 11 \cdot NR)$]

The number of strands that may be placed in a row is limited by the dimensions of the box cross section and the necessity of maintaining adequate clearance between strands and between the strands and the edges of the section. If NM_i denotes the maximum number of strands that can be placed in row i , then

$$NS_{i,1} \leq NM_i \quad (i = 1, 2, \dots, NR) \quad (65)$$

2.2.6 Constraints to Insure Proper Release Strength Representation [Constraints $(21 + 11 \cdot NR)$ thru $(39 + 11 \cdot NR)$]

The concrete release strength representation used in Eq. (33) is valid only if the constraints given in Eqs. (34) and (35) are satisfied. Thus, to obtain a proper problem formulation, the constraint set must include

$$f_i \leq 0.5 \quad (i = 1, \dots, 10) \quad (66)$$

$$f_{i+1} - f_i \leq 0 \quad (i = 1, \dots, 9) \quad (67)$$

2.2.7 Bounds on Initial Beam Camber [Constraints (40 + 11·NR) and (41 + 11·NR)]

Let Δ^+ and Δ^- denote the maximum and minimum initial midspan deflections admissible in a particular design, with positive deflections taken as upward. If, for example, a designer wished to insure that a beam did not have an upward camber of more than 3.25 in. nor less 0.75 in. (the lower bound perhaps being imposed to insure that the long term camber under the additional weight of wearing surface and shear key was not downward), Δ^+ would be +3.25 in. and Δ^- would be 0.75 in. The initial camber Δ is the sum of the deflections due to prestress and weight of the beam (Δ_{DL}). The initial deflection Δ_{DL} due to beam weight is

$$\Delta_{DL} = -22.5 \frac{wL^4}{E_{ci}I} \quad (\text{in.}) \quad (68)$$

where

w = beam weight (kips/ft),

L = span length (ft),

I = moment of inertia of beam section (in^4),

E_{ci} = modulus of elasticity of beam concrete at release (ksi).

Substituting Eq. (68) into Eq. (7) yields the following expression for initial camber

$$\Delta = \frac{1}{E_{ci}I} \left\{ 22.5wL^4 - (1 - \xi)F_o \sum_{j=1}^{11} \left[\sum_{i=1}^{NR} d_i \cdot NS_{i,j} \right] \delta_j y_j \right\} \quad (69)$$

where y_j and δ_j are given by Eq. (9). The upper bound on camber is enforced by

$$\Delta \leq \Delta^+ \quad (70)$$

which upon substitution of Eq. (69) and rearrangement becomes

$$-(1 - \xi)F_o \sum_{j=1}^{11} \left[\sum_{i=1}^{NR} d_i \cdot NS_{i,j} \right] \delta_j y_j \leq E_{ci}I\Delta^+ + 22.5wL^4 \quad (71)$$

The form of Eq. (71) is not yet acceptable because it involves the modulus of elasticity of the concrete at release, which depends on the release strength f'_{ci} . The modulus of elasticity frequently is assumed to vary with the square root of cylinder strength; i.e.,

$$E_{ci} = .031623K\sqrt{f'_{ci}} \quad (E_{ci} \text{ and } f'_{ci} \text{ in ksi}) \quad (72)$$

Replacing the radical with a Taylor series expansion and substituting Eq. (33) for f'_{ci} gives

$$E_{ci} = K \left\{ .007454 \sum_{i=1}^{10} f_i + .06337 \right\} \quad (73)$$

The constant K depends on the unit weight of the concrete (10) and can be taken as 57,000 for normal weight concrete. Substitution of Eq. (73) into (71) gives the final form of the constraint

$$\begin{aligned} & -(1 - \xi)F_o \sum_{j=1}^{11} \left\{ \sum_{i=1}^{NR} d_i \cdot NS_{i,j} \right\} \delta_j y_j - .007454 \Delta^+ K \sum_{i=1}^{10} f_i \\ & \leq .06337 K I \Delta^+ + 22.5 w L^4 \end{aligned} \quad (74)$$

In a similar fashion, the lower bound constraint is given by

$$\begin{aligned} & (1 - \xi)F_o \sum_{j=1}^{11} \left\{ \sum_{i=1}^{NR} d_i \cdot NS_{i,j} \right\} \delta_j y_j + .007454 \Delta^- K \sum_{i=1}^{10} f_i \\ & \leq .06337 K I \Delta^- - 22.5 w L^4 \end{aligned} \quad (75)$$

2.2.8 Constraints to Insure Adequate Ultimate Moment Capacity [Constraint (42 + 11·NR)]

The computed ultimate moment capacity of the beam M_u must be greater than or equal to M_{ur} , the required ultimate moment capacity defined in Eq. (10). This requirement can be written as

$$-M_u \leq -M_{ur} \quad (76)$$

The ultimate moment capacity M_u is not linearly related to the strand pattern at midspan (described by the design variables $NS_{i,1}$) nor the concrete strength (described by the design variables f_i). Thus, an indirect method, one which is linear in the design variables, must be used. The device for accomplishing this was developed in Section 2.1 (Eq. 19 and Figures 5 and 6). Figure 6 shows the relation between the parameter $\bar{\rho}$, which is a measure of total strand force eccentricity, and concrete strength f'_c . Those beams having a midspan strand pattern and concrete strength which yield a moment capacity in excess of M_{ur} are represented by points that lie above and to the right of the curve shown in Figure 6. Let $\bar{\rho}_0, \bar{\rho}_1, \dots, \bar{\rho}_{10}$ be the minimum total strand force eccentricities necessary for $M_u = M_{ur}$, for 28 day concrete strengths corresponding to release strengths of 4.0, 4.5, ..., 9.0 ksi. The curve shown in Figure 6 can be approximated with the following relation

$$\bar{\rho} = \bar{\rho}_0 + 2 \sum_{i=1}^{10} (\bar{\rho}_i - \bar{\rho}_{i-1}) f_i \quad (77)$$

Note that Eq. (77) defines a piecewise linear approximation to the curve in Figure 6. That is, for f_i values which give a release strength that is an integer multiple of 0.5 ksi, the value of $\bar{\rho}$ computed from Eq. (77) lies on the curve. In order to insure that adequate total strand force eccentricity is present, we write

$$-\rho \leq -\bar{\rho} \quad (78)$$

and substituting Eqs. (19) and (77) into (78) we have

$$\sum_{i=1}^{NR} d_i \cdot NS_{i,1} + 2 \sum_{i=1}^{10} (\bar{\rho}_i - \bar{\rho}_{i-1}) f_i \leq -\bar{\rho}_0 \quad (80)$$

Moving the design variables to the left of the inequality yields the final

form

$$\sum_{i=1}^{NR} d_i \cdot NS_{i,1} + 2 \sum_{i=1}^{10} (\bar{\rho}_i - \bar{\rho}_{i-1}) f_i \leq -\bar{\rho}_0 \quad (80)$$

The computational procedure for constructing the $\bar{\rho}_i$ in Eq. (80) is straightforward. Beginning with a release strength of 4.0, the corresponding 28 day strength g_i is used to compute moment capacity. Strands are added to the section, 2 at a time, beginning with first (bottom) row and the ultimate moment capacity M_u is computed using Eqs. (11) or (18). If $M_u < M_{ur}$ additional strands are added, progressively filling the first, then the second row, etc., until $M_u \geq M_{ur}$. The corresponding value of ρ computed with Eq. (19) is then taken as $\bar{\rho}_i$.

2.2.9 Constraint to Insure $M_u \geq 1.2M_{cr}$ [Constraint (43 + 11·NR)]

The cracking moment capacity is defined as that moment which produces a tensile stress of $7.5\sqrt{f'_c}$ (f'_c in psi) at the bottom of the beam (4). Using the strand pattern at midspan, the cracking moment is given by

$$M_{cr} = \bar{Z}_b \left\{ -(1 - \eta) F_o \sum_{i=1}^{NR} - \left[\frac{1}{A} - \frac{d_i}{Z_b} \right] NS_{i,1} + .2371 \sqrt{f'_c} - \frac{M_o}{Z_b} \right\} \quad (81)$$

where M_o = midspan moment due to beam weight and f'_c is in ksi. Once again an indirect approach to formulation of this constraint must be used to avoid introducing nonlinear terms. Figure 11 shows schematically the relationship between the ultimate moment capacity of the section M_u , 1.2 times the cracking moment M_{cr} and the total strand force eccentricity (defined in Eq. (19)). For small values of ρ , the cracking moment capacity exceeds the ultimate moment capacity. As strands are added, M_u rises more sharply than does $1.2M_{cr}$, and at the point ρ' , exceeds $1.2M_{cr}$. Thus, for the

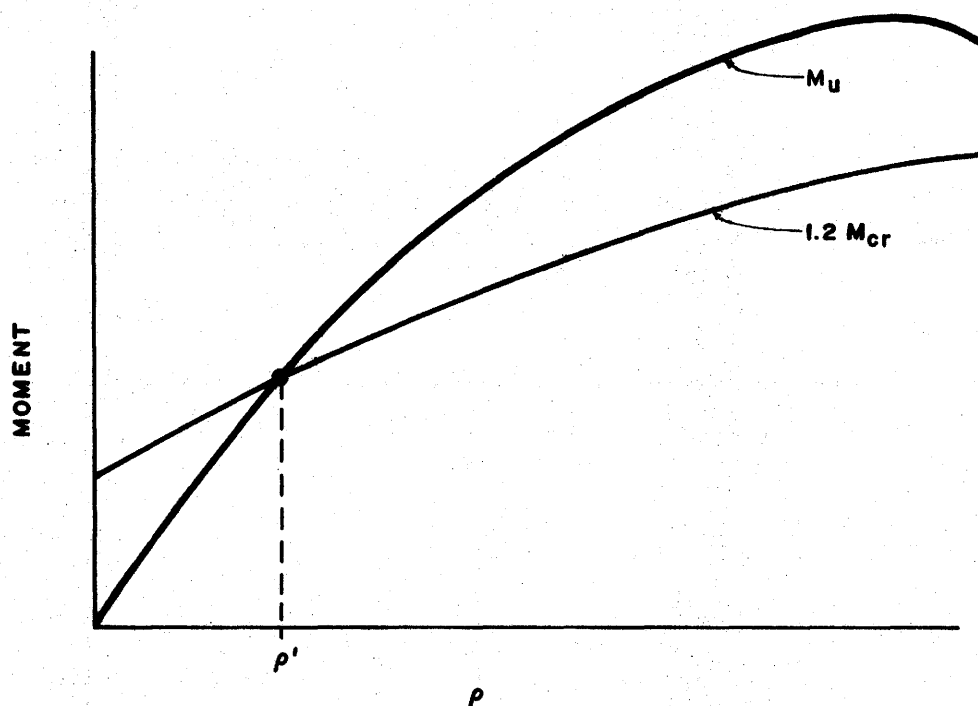


FIGURE 11. CRACKING AND ULTIMATE MOMENTS PLOTTED AGAINST TOTAL STRAND FORCE ECCENTRICITY

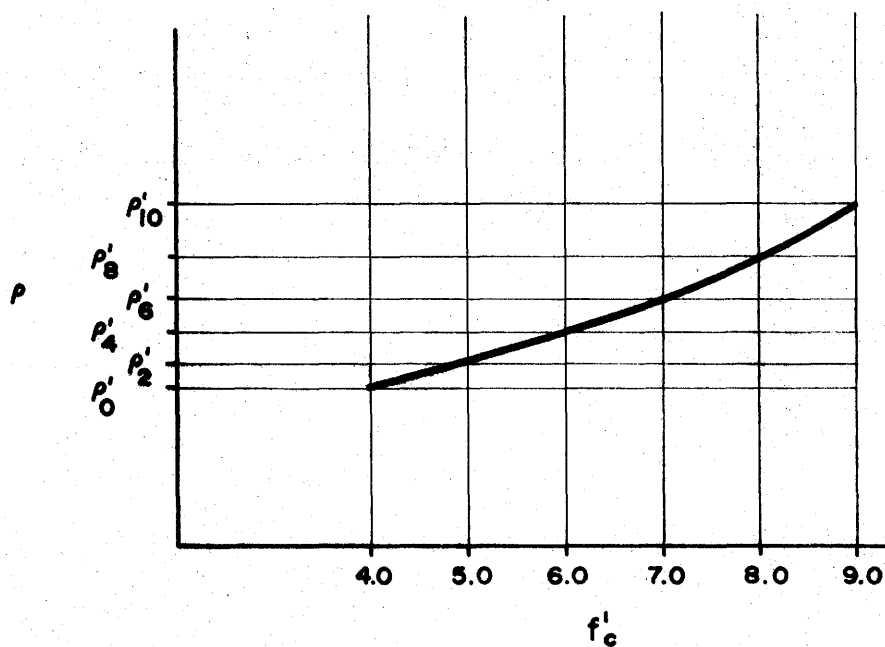


FIGURE 12. MINIMUM REQUIRED STRAND FORCE ECCENTRICITY VS. RELEASE STRENGTH

particular f'_c under consideration, the requirement that

$$-M_u \leq -1.2M_{cr} \quad (82)$$

can be stated as

$$-\rho \leq -\rho' \quad (83)$$

As f'_c increases, the point of intersection of the M_u and $1.2M_{cr}$ curves generally moves to the right. Figure 12 shows the variation of ρ' with concrete strength. Let $\rho'_0, \rho'_1, \dots, \rho'_{10}$ denote the minimum total strand force eccentricity necessary for $M_u \geq 1.2M_{cr}$, for 28-day strengths corresponding to release strengths f'_{ci} of 4.0, 4.5, ..., 9.0 ksi. The curve of Figure 12 is approximated with straight line segments by

$$\rho' = \rho'_0 + 2 \sum_{i=1}^{10} (\rho'_i - \rho'_{i-1}) f_i \quad (84)$$

Substituting Eqs. (19) and (84) into (83) yields

$$\sum_{i=1}^{NR} d_i \cdot NS_{i,1} + 2 \sum_{i=1}^{10} (\rho'_i - \rho'_{i-1}) f_i \leq -\rho'_0 \quad (85)$$

The computation of ρ'_i can be carried out in a manner analogous to that for $\bar{\rho}_i$ in the ultimate moment constraint.

2.2.10 Lower and Upper Bounds on Concrete Strength [Constraints (44 + 11·NR) and (45 + 11·NR)]

If during the computation of the $\bar{\rho}_i$ for the ultimate moment capacity constraint, it is found that M_{ur} can not be attained for a particular 28-day strength g_i , the release strength corresponding to g_{i+1} must be taken as the minimum permissible release strength. That is,

$$-4.0 - \sum_{i=1}^{10} f_i \leq (f'_{ci})_{\min} \quad (86)$$

If M_{ur} can be obtained for all prescribed 28 day strengths, $(f'_{ci})_{min}$ is taken as 4.0 ksi.

Release strengths in this formulation are assumed to range up to 9.0 ksi. Should this be greater than the actual release strength that can be obtained, the release strength variables must be bounded from above by

$$4.0 + \sum_{i=1}^{10} f_i \leq (f'_{ci})_{max} \quad (87)$$

where $(f'_{ci})_{max}$ is maximum attainable value of f'_{ci} .

2.3 STRAIGHT STRAND DESIGN FORMULATION - DESIGN OPTION

Should the designer wish to specify the 28 day strength f'_c to be used and obtain the strand pattern and minimum release strength, he may specify the "design" option. For this problem formulation, the design variables are limited to those which define the strand pattern $(NS_{i,j})$ and the release strength f'_{ci} .

2.3.1 Objective Function

The objective in this case is to minimize the total number of strands, while wrapping strands only where necessary to keep the release strength to a minimum or to control camber. Mathematically, this is equivalent to

$$\begin{aligned} \text{Minimize } & (c_s L + .5c_w L) \sum_{i=1}^{NR} NS_{i,1} - \frac{2Lc_w}{40} \sum_{j=2}^{11} \sum_{i=1}^{NR} NS_{i,j} \\ & + c_c \cdot \frac{A \cdot L}{1944} \cdot f'_{ci} \end{aligned} \quad (88)$$

By taking c_s , the cost per foot for strand, very large (say, \$100.00), we are assured of obtaining the minimum number of strands. Assigning a

cost of $c_c \cdot A \cdot L / 1944$ to concrete insures that f'_{ci} will be as small as possible. We select c_c such that the total concrete cost will be a small fraction of the strand cost (say, equal to the cost of one strand). Finally, a small cost for wrapping (say $c_w = \$0.01$) insures that strands will be wrapped only where necessary, but will always be used if it results in a lower release strength f'_{ci} .

2.3.2 Constraints on Release Stresses [Constraints 1 thru 12]

Release stress constraints in this case differ from those developed in Section 2.2.2 in the concrete strength variable. Noting that the release strength is given by f'_{ci} , Eq. (45) limits the tension stress at each section ($j=1, 3, \dots, 11$) to the allowable tension stress by

$$\begin{aligned} (1 - \epsilon) F_{o \sum_{i=1}^{NR}} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,j} - .007454 S_{tj} f'_{ci} \\ \leq \frac{M_j}{Z_t} + .03355 S_{tj} \quad (j=1, 3, \dots, 11) \end{aligned} \quad (89)$$

Compression stress at points on the bottom of the beam are limited to the allowable compression stress through modification of Eq. (48) to obtain

$$\begin{aligned} -(1 - \epsilon) F_{o \sum_{i=1}^{NR}} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,j} - S_{cj} f'_{ci} \\ \leq \frac{M_j}{Z_b} \quad (j=1, 3, \dots, 11) \end{aligned} \quad (90)$$

2.3.3 Constraints on Service Load Stresses [Constraints 13 thru 20]

The expression for 28-day strength used in Section 2.2.3 is replaced with the specified strength f'_{28} . Tension stresses at the bottom for sections at midspan, $j = 3$, and $j = 7$ are limited by

$$\begin{aligned} (1 - n) F_{o \sum_{i=1}^{NR}} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,j} \leq -\frac{M_j}{Z_b} - \frac{\bar{M}_j}{Z_b} + \bar{S}_{tj} \sqrt{f'_{28}} \\ (j=1, 3, 7) \end{aligned} \quad (91)$$

As before, M_j and \bar{M}_j are taken as midspan moments when $j = 1$.

The tensile stress in the top of the beam at its end is limited by

$$(1 - n)F_o \sum_{i=1}^{NR} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,11} \leq \bar{S}_{t11} \sqrt{f'_{28}} \quad (92)$$

while the compression stress at the bottom is limited by

$$-(1 - n)F_o \sum_{i=1}^{NR} - \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_{i,11} \leq \bar{S}_{c11} f'_{28} \quad (93)$$

Compression stresses in the top of the beam at midspan, $j = 3$, and $j = 7$ are limited by

$$-(1 - n)F_o \sum_{i=1}^{NR} - \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_{i,j} \leq \bar{S}_{cj} f'_{28} \quad (94)$$

2.3.4 Constraints to Insure Proper Strand Wrapping [Constraints 21 thru (20 + 10·NR)]

The constraint set to insure proper strand wrapping for this formulation is identical to that used in the optimization option and is given by Eq. (64).

2.3.5 Constraints Limiting the Number of Strands in Each Row [Constraints (21 + 10·NR) thru (20 + 11·NR)]

This constraint set is also identical to that defined previously, and is given by Eq. (65).

2.3.6 Bounds on Release Camber [Constraints (21 + 11·NR) and (22 + 11·NR)]

The revised form of these constraints is obtained by substituting the expression for modulus of elasticity at release

$$E_{ci} = K \{ .007454f'_{ci} + .03355 \} (f'_{ci} \text{ \& } E_{ci} \text{ in ksi}) \quad (95)$$

into Eq. (71) to obtain

$$-(1 - \xi)F_o \sum_{j=1}^{11} \left[\sum_{i=1}^{NR} d_i \cdot NS_{i,j} \right] \delta_j y_j - .007454KI\Delta^+ f'_{ci} \leq .03355KI\Delta^+ + 22.5wL^4 \quad (96)$$

and

$$(1 - \xi)F_o \sum_{j=1}^{11} \left[\sum_{i=1}^{NR} d_i \cdot NS_{i,j} \right] \delta_j y_j + .007454KI\Delta^- f'_{ci} \leq .03355KI\Delta^- - 22.5wL^4 \quad (97)$$

2.3.7 Ultimate Moment Capacity [Constraint (23 + 11·NR)]

Letting $\bar{\rho}$ denote the minimum total strand force eccentricity which provides M_u greater than or equal to M_{ur} for the specified 28 day strength f'_{28} , and altering Eq. (79), we have

$$\sum_{i=1}^{NR} d_i \cdot NS_{i,11} \leq -\bar{\rho} \quad (98)$$

2.3.8 Cracking Moment Capacity [Constraint (24 + 11·NR)]

Taking ρ' as the minimum total strand force eccentricity which insures that for the concrete strength f'_{28} , the ultimate moment capacity is equal to or greater than $1.2 M_{cr}$, gives

$$\sum_{i=1}^{NR} d_i \cdot NS_{i,11} \leq -\rho' \quad (99)$$

2.3.9 Lower and Upper Bounds on Concrete Strength [Constraints (25 + 11·NR) and (26 + 11·NR)]

The release strength f'_{ci} can not be less than the THD standard minimum of 4.0 ksi,

$$-f'_{ci} \leq -4.0 \quad (100)$$

nor exceed the specified 28 day strength,

$$f'_{ci} \leq f'_{28}$$

(101) 

2.4 DETERMINATION OF FINAL DESIGN - STRAIGHT STRANDS

The linear programming formulation used requires that prestress loss at release ξ and long term loss η be specified. These quantities are actually dependent, in part, on the design variables defining concrete strength and strand pattern. This shortcoming in the formulation can be treated through an iterative process. Initial values for ξ and η are assumed, and the linear program solved to obtain concrete strengths and strand pattern. Corrected values of ξ and η are then computed by the procedure described in Section 2.1. If the computed losses are less than those assumed, the design is adequate. If not, the new computed losses are incorporated in the constraint equations and the linear program resolved. Experience with the computer programs indicate that this process generally converges in 3 to 5 iterations, starting with $\xi=0.05$ and $\eta=0.10$.

The variables in a standard linear programming formulation are assumed to be continuous. Thus, one may obtain a non-integer number of strands in a solution. The final design is obtained by rounding the strand variables to the nearest integer number. In straight strand box beam designs there are normally a sufficient number of strands present in a strand row so that rounding does not significantly affect the final solution (for example, 12 strands in row 2 as opposed to 11.78 strands obtained from the L.P. solution).

The linear programming formulation permits the early detection of unrealistic design requirements that may inadvertently be imposed by the designer. These can occur in a variety of ways, such as too stringent

camber restrictions or insufficient number of strands permitted to sustain the loads which are to be carried. Such a condition is automatically detected by the Simplex algorithm (1) during an attempt to solve the linear program.

After a solution has been obtained for an L.P. problem, the behavior restrictions (allowable stresses, camber limits, ultimate moment capacity, etc.) which control the design can be easily determined. Those inequality constraints which are "tight" at the final solution (i.e., those which are satisfied as equalities) control, while those which are satisfied by some margin have no effect on the final design. This information can be used by the designer should he choose to alter some of initially specified properties of the design (such as section dimensions or maximum number of strands permitted in a row) in order to obtain a more efficient beam.

2.5 VARIABLE CORRESPONDENCE FOR STRAIGHT STRAND DESIGN

The linear programming problem format given in Eqs. (1) and (2) utilizes design variables x_1, \dots, x_n , while the notation used in the objective function and constraint relationships retained $NS_{i,j}$ for the variables representing the number of bonded strands at each wrapping point and f_1, \dots, f_{10} to denote variables associated with release strength (f'_{ci} for the design option). Correspondence between these two sets of notation are as follows:

2.5.1 Optimization Option

The $NS_{i,j}$ correspond to

$$x_1 = NS_{1,1}$$

$$x_2 = NS_{2,1}$$

$$\begin{aligned}
& \cdot \\
& \cdot \\
& \cdot \\
& x_{\text{NRAV}} = NS_{\text{NRAV},1} \\
& x_{\text{NRAV}+2} = NS_{2,2} \\
& \cdot \\
& \cdot \\
& \cdot \\
& x_{2 \cdot \text{NRAV}} = NS_{\text{NRAV},2} \\
& \cdot \\
& \cdot \\
& \cdot \\
& x_{10 \cdot \text{NRAV}} = NS_{1,11} \\
& x_{10 \cdot \text{NRAV}+1} = NS_{2,11} \\
& \cdot \\
& \cdot \\
& \cdot \\
& x_{11 \cdot \text{NRAV}} = NS_{\text{NRAV},11}
\end{aligned} \tag{102}$$

and the f_i correspond to

$$\begin{aligned}
& x_{11 \cdot \text{NRAV}+1} = f_1 \\
& \cdot \\
& \cdot \\
& \cdot \\
& \cdot \\
& x_{11 \cdot \text{NRAV}+11} = f_{10}
\end{aligned} \tag{103}$$

The total number of variables n is equal to $(11 \cdot \text{NRAV} + 11)$.

2.5.2 Design Option

The correspondence between x_k and $NS_{i,j}$ is the same as that given in Eq. (102). The release strength is $x_{11 \cdot \text{NRAV}+1}$ and the total number of variables n is $(11 \cdot \text{NRAV} + 1)$.

III. DESIGN OF BOX GIRDERS WITH DRAPED STRANDS

In this section, the problem of determining the concrete release and 28-day strengths and strand pattern layout which minimizes the total cost of a box girder which may contain draped strands and has no strand bond breakage is formulated as an integer programming problem. The integer programming problem has the same mathematical structure as the linear program described by Eqs. (1) and (2) with the exception of the design variables x_1, \dots, x_n which are required to take only integer values. While the integer programming formulation more closely reflects the true nature of the design problem, its solution requires considerably more computational effort than does the linear program. It is used here in lieu of the linear programming approach because of variable rounding difficulties inherent in the draped strand formulation. Draped strand design practice requires that a fixed number of strands be draped in a row (as many as six strands, depending on web width). If any drapable strands in a row are to be draped, all must be draped. Thus, if a linear programming formulation was used, and the final solution indicated that 2.9 strands were to be draped, this value would have to be truncated to zero or raised to 6 (assuming 6 drapable strands per row). This obviously would lead to considerable differences between the L.P. optimum design and that obtained from it by rounding.

3.1 DESIGN CONSIDERATIONS

Let NS_i denote the total number of strands present in strand row i (row number 1 is the bottom most strand row) and I_i be a binary variable (either 0 or 1) indicating the presence of draped strands in row i

($I_i = 1$) or their absence ($I_i = 0$). For the case shown in Figure 13,

$$\begin{array}{ll} NS_1 = 10 & I_1 = 0 \\ NS_2 = 10 & I_2 = 0 \\ NS_3 = 6 & I_3 = 1 \\ NS_4 = 4 & I_4 = 1 \\ NS_5 = 4 & I_5 = 1 \end{array} \quad (104)$$

Let NR equal the number of rows which may contain strands, NRAV equal the number of the top-most strand row in the section (NRAV = 10 in Figure 13), NB be the row number of the first row containing draped strands (NB = 3 in Figure 13), and EN be the product of the number of rows of draped strands and the number of rows by which the strands are raised at the end of the beam. In Figure 13, the number of rows of draped strands is 3 and the number of rows by which they are raised is 4, giving EN=12. Define NW as the number of drapable strands per row, e as the distance between the straight and draped strands in a row at the end of the beam, e_j as this distance at point j along the beam, and αL as the distance from the end of the beam to the holddown point (Figure 14).

The stress in the top of the beam at point j due to prestress in row i is given by

$$\begin{aligned} \sigma_{i,j}^{(T)} = & -(1 - \xi)F_0 \frac{1}{A} NS_i + (1 - \xi)F_0 (-d_i - e_j)NW \frac{1}{Z_t} I_i \\ & + (1 - \xi)F_0 (-d_i) \frac{1}{Z_t} (NS_i - NW \cdot I_i) \end{aligned} \quad (105)$$

Collecting common terms and factoring yields

$$\sigma_{i,j}^{(T)} = -(1 - \xi)F_0 \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i - (1 - \xi)F_0 \frac{NW}{Z_t} e_j I_i \quad (106)$$

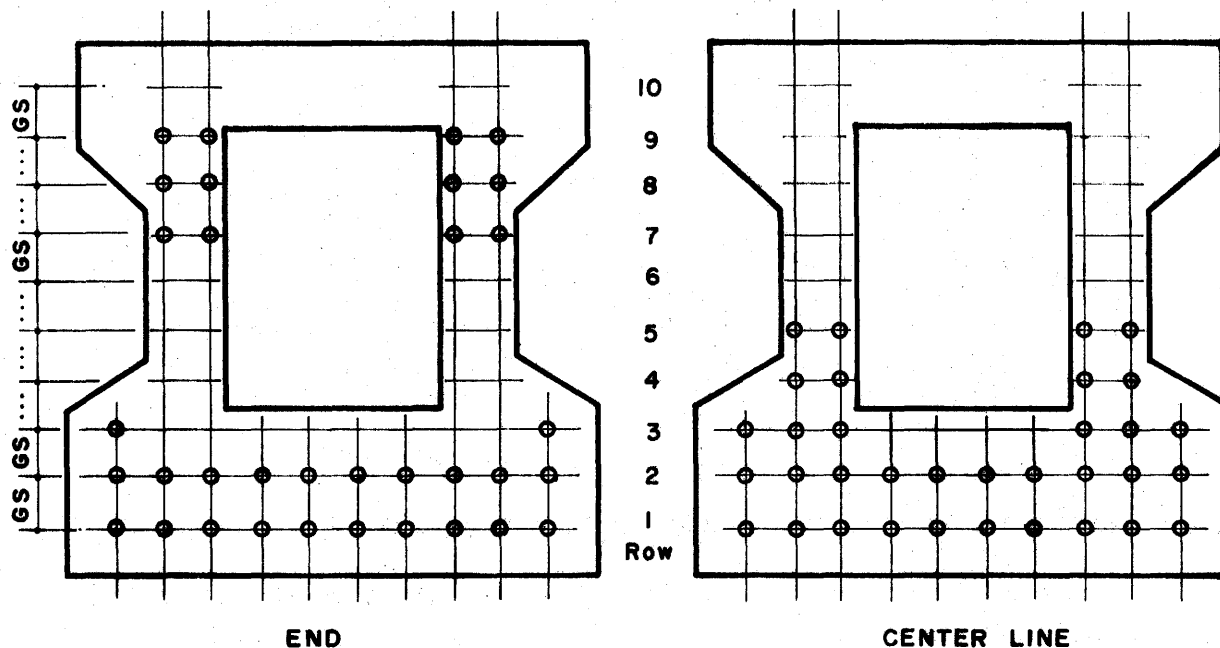


FIGURE 13. CENTER LINE AND END SECTIONS OF DRAPED STRAND BEAM

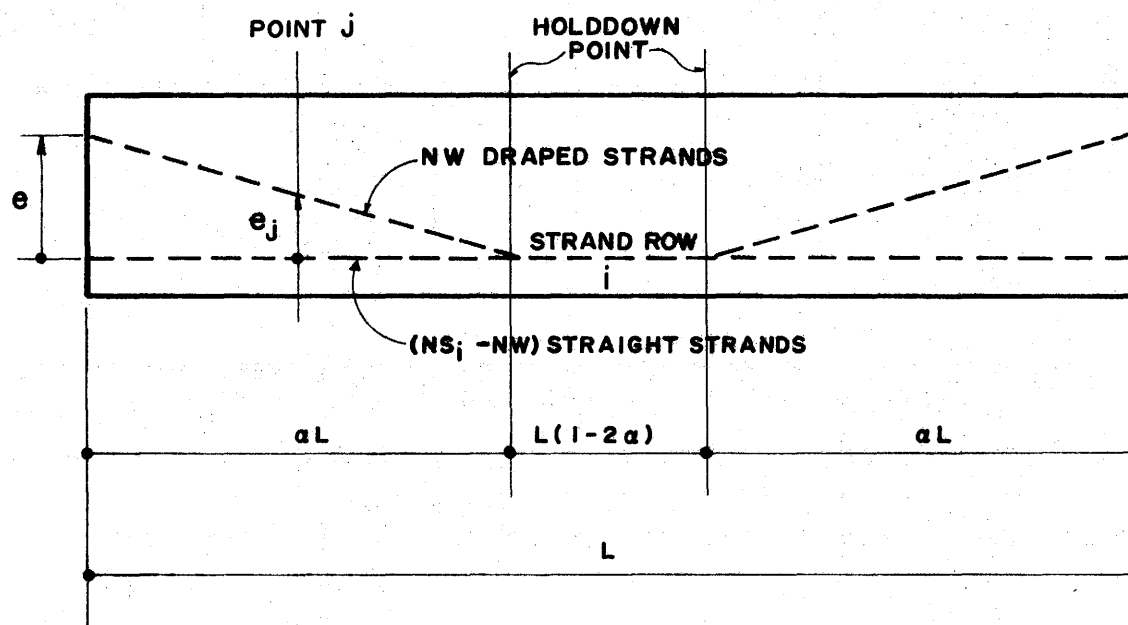


FIGURE 14. NOTATION DESCRIBING A STRAND ROW

Noting that $e_j = \tau_j e$, where τ_j is a factor dependent on the location of point j , Eq. (106) becomes

$$\sigma_{i,j}^{(T)} = -(1 - \epsilon)F_0 \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i - (1 - \epsilon)F_0 \frac{NW}{Z_t} \tau_j e I_i \quad (107)$$

The total stress in the top of the beam at point j due to all rows of strands is obtained by summing the effects of each row, i.e.

$$\sigma_j^{(T)} = \sum_{i=1}^{NR} \sigma_{i,j}^{(T)} \quad (108)$$

Substituting Eq. (107) into (108) and collecting terms gives

$$\sigma_j^{(T)} = -(1 - \epsilon)F_0 \sum_{i=1}^{NR} \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i - (1 - \epsilon)F_0 \frac{NW}{Z_t} \tau_j e \sum_{i=1}^{NR} I_i \quad (109)$$

The strand end eccentricity e must be an integer multiple of the strand row spacing GS . The spacing between all rows must be the same in order that a given design can be fabricated with equal ease by hardware that depresses strands at the holddown points or that lifts strands at the ends of the beam. Thus, e may be written as

$$e = N \cdot GS \quad (110)$$

where N is the number of rows by which the draped strands are raised at the end of the beam. Substituting Eq. (110) into Eq. (109) gives

$$\sigma_j^{(T)} = - (1 - \xi) F_o \sum_{i=1}^{NR} \left[\frac{1}{A} + \frac{d_i}{Z_t} \right] NS_i - (1 - \xi) F_o \frac{NW}{Z_t} - \tau_j GS \cdot N \sum_{i=1}^{NR} I_i \quad (111)$$

It follows from the definition of I_i that $\sum I_i$ is equal to the number of rows with draped strands, and that $N \sum I_i$ is equal to EN which was defined previously. Thus Eq. (111) becomes

$$\sigma_j^{(T)} = - (1 - \xi) F_o \sum_{i=1}^{NR} \left[\frac{1}{A} + \frac{d_i}{Z_t} \right] NS_i - (1 - \xi) F_o \frac{NW}{Z_t} - \tau_j GS \cdot EN \quad (112)$$

At this point a comment is in order concerning the seemingly bizarre set of variables (NS_i , I_i , NB and EN) used to arrive at an expression for prestress induced stress. The integer programming format used to formulate the beam design problem requires that all constraint expressions (including those developed later to limit beam stresses) be linear in the design variables. If the expression for stress given in Eq. (109) were used, it would contain the product of design variables (e and the I_i). The introduction of transformed variables to produce a linear expression is not without its complications. As will be shown later, a rather complex set of additional constraints must be introduced to insure that the minimum cost design obtained in the transformed design space corresponds in a unique way to an obtainable design.

Using previously defined terminology and a procedure analogous to that just explained, the stress at point j in a beam produced by all sources under release conditions can be written as

$$\sigma_j^{(T)} = -(1 - \xi) F_{o \sum_{i=1}^{NR}} \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i - (1 - \xi) F_o \frac{NW}{Z_t} \tau_j GS \cdot EN - \frac{M_j}{Z_t} \quad (113)$$

$$\sigma_j^{(B)} = -(1 - \xi) F_{o \sum_{i=1}^{NR}} \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_i + (1 - \xi) F_o \frac{NW}{Z_b} \tau_j GS \cdot EN + \frac{M_j}{Z_b} \quad (114)$$

and for service load conditions

$$\bar{\sigma}_j^{(T)} = -(1 - \eta) F_{o \sum_{i=1}^{NR}} \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i - (1 - \eta) F_o \frac{NW}{Z_t} \tau_j GS \cdot EN - \frac{M_j}{Z_t} - \frac{\bar{M}_j}{Z_t} \quad (115)$$

$$\bar{\sigma}_j^{(B)} = -(1 - \eta) F_{o \sum_{i=1}^{NR}} \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_i + (1 - \eta) F_o \frac{NW}{Z_b} \tau_j GS \cdot EN + \frac{M_j}{Z_b} + \frac{\bar{M}_j}{Z_b} \quad (116)$$

Camber at midspan after strand release can be written as

$$\Delta = \Delta_{DL} - \frac{1}{E_{ci} I} \left\{ \frac{L^2}{8} M_s + \frac{(\alpha L)^2}{6} M_E + \left[\frac{L^2}{8} - \frac{(\alpha L)^2}{6} \right] M_D \right\} \quad (117)$$

where M_s , M_E and M_D are shown in Figure 15 and given by

$$M_s = (1 - \xi) F_{o \sum_{i=1}^{NR}} d_i (NS_i - NW \cdot I_i) \quad (118)$$

$$M_D = (1 - \xi) F_o NW \sum_{i=1}^{NR} d_i I_i \quad (119)$$

$$M_E = M_D + (1 - \xi) F_o NW \cdot GS \cdot EN \quad (120)$$

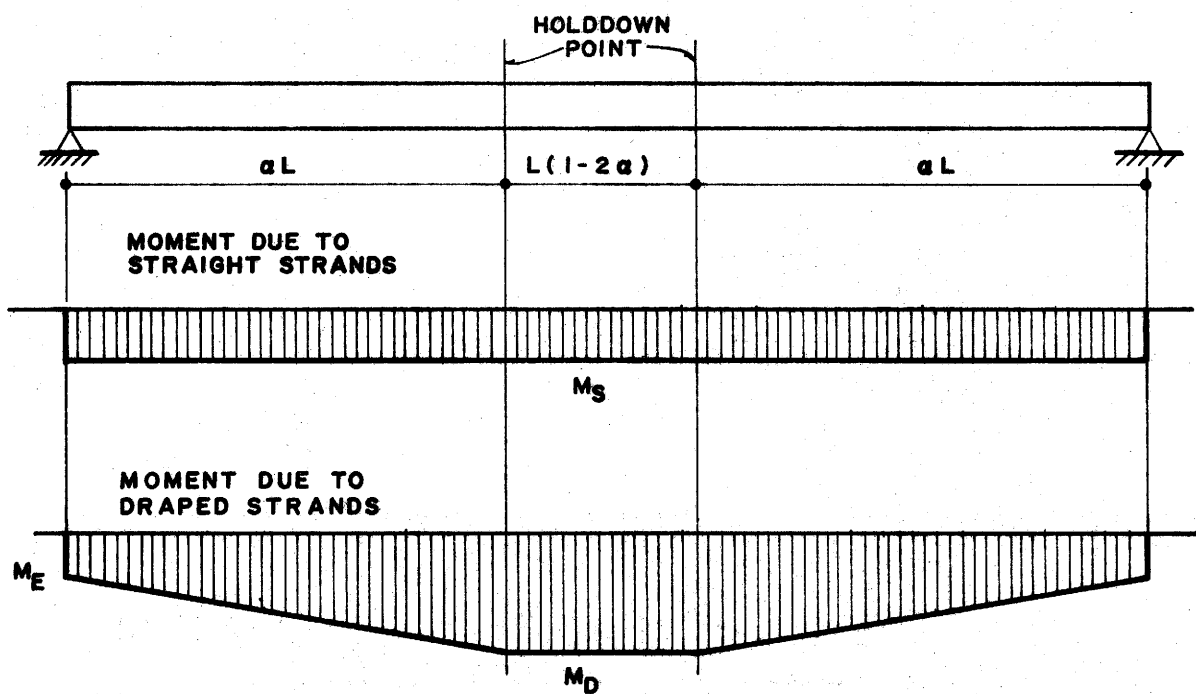


FIGURE 15. PRESTRESS INDUCED MOMENT DIAGRAMS FOR CALCULATION OF RELEASE CAMBER IN BEAMS WITH DRAPED STRANDS

The computation of ultimate moment capacity, cracking moment and prestress loss is identical to that described in Section 2.1. The calculation of stirrup spacing differs from that previously developed only in that the shear force to be resisted by the stirrups $V_U^{(i)}$ is reduced by the amount of the vertical component of force exerted by draped strands at points between the end of the beam and holddown point.

3.2 DRAPED STRAND FORMULATION - OPTIMIZATION OPTION

In this section, the problem of determining the concrete release strength, the mid-span strand pattern and end eccentricity of draped strands which minimizes the total cost of a box girder is formulated as an integer program. The variables NS_i , NB and EN are integer, assuming any integer values which satisfy the constraints defined below. The variables I_i and the variables K_i (introduced below) are binary variables, which can assume only the values 0 or 1. The release strength of beam concrete can be written as

$$f'_{ci} = 4.0 + 0.5 \sum_{i=1}^{10} K_i \quad (121)$$

where K_i are binary variables (taking values of either 0 or 1) satisfying the inequalities

$$-K_i + K_{i+1} \leq 0 \quad i=1, \dots, 9 \quad (122)$$

This form renders discrete values of release strength in 0.5 ksi increments. The 28 day strength corresponding to a specific release strength can be

expressed as

$$f'_c = g_0 + \sum_{i=1}^{10} (g_i - g_{i-1})K_i \quad (123)$$

which is a discontinuous step approximation to the function relating release strength to 28-day strength (Figure 16). The cost per cubic yard of concrete as a function of release strength is given by the discontinuous step approximation

$$c_0 + \sum_{i=1}^{10} (c_i - c_{i-1})K_i \quad (124)$$

3.2.1 Objective Function

The total cost of the beam is assumed to consist of the cost of concrete and cost of strands. This can be written as

$$\text{Minimize } c_s L \sum_{i=1}^{NR} NS_i + \frac{A \cdot L}{1944} \sum_{i=1}^{10} (c_i - c_{i-1})K_i + \frac{A \cdot L}{1944} c_0 \quad (125)$$

3.2.2 Constraints on Release Stresses [Constraints 1 thru 8]

Release stresses are checked at the top and bottom of the beam, at the holddown ($j=8$), $5L/40$ ($j=5$), $L/10$ ($j=6$) and at the end ($j=7$) (see Figure 17). The point $5L/40$ is used so that different allowable stresses could be imposed, at the designers' option, for points between the end of the beam and the first tenth point and for those points between the tenth point and the holddown. The points chosen are the end points of each of these two intervals. At release, if the allowable stresses are satisfied at each end of the interval, they will be satisfied at all points within the interval because the offsetting stresses produced by beam weight increase parabolically while those due to prestress increase linearly with position along the interval.

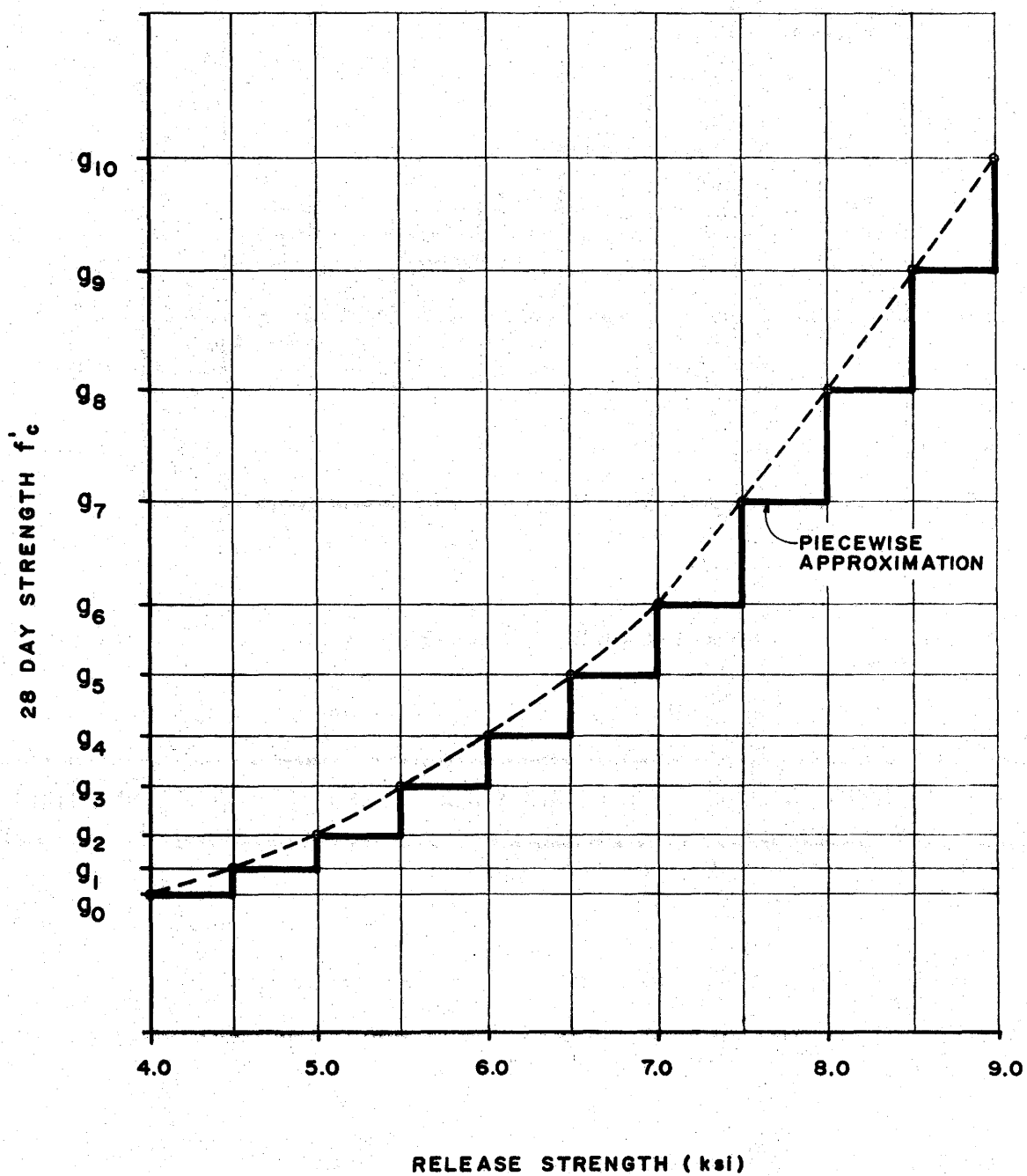


FIGURE 16. APPROXIMATION OF 28 DAY STRENGTH VS. RELEASE STRENGTH

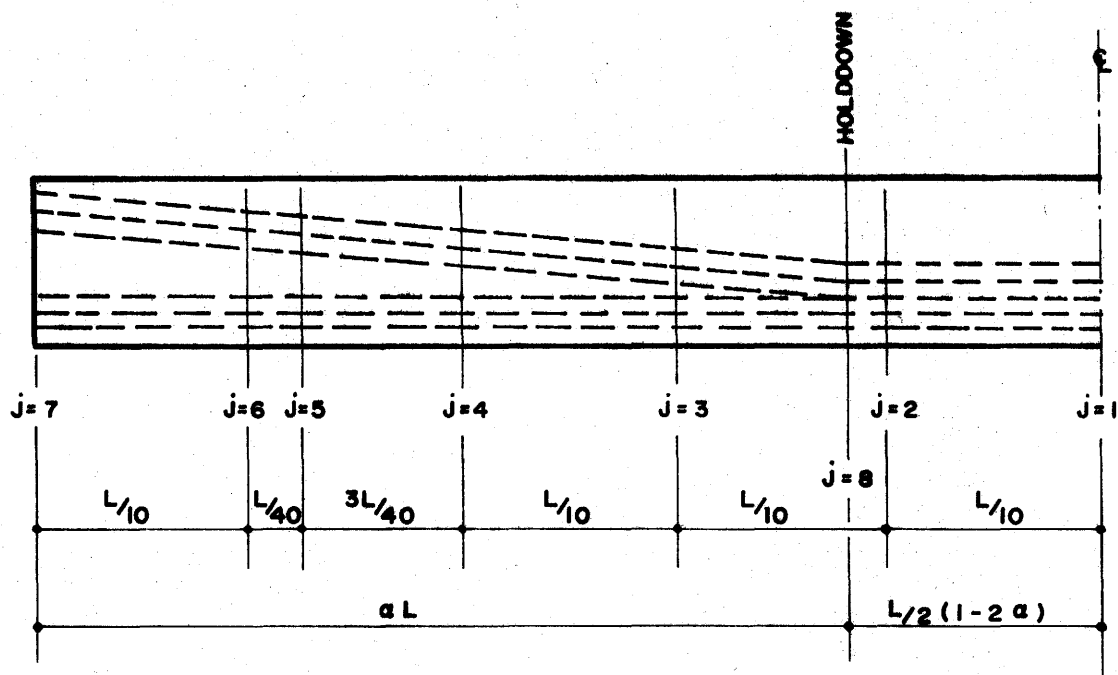


FIGURE 17. STATIONS FOR STRESS CHECKS IN DRAPED STRAND DESIGN

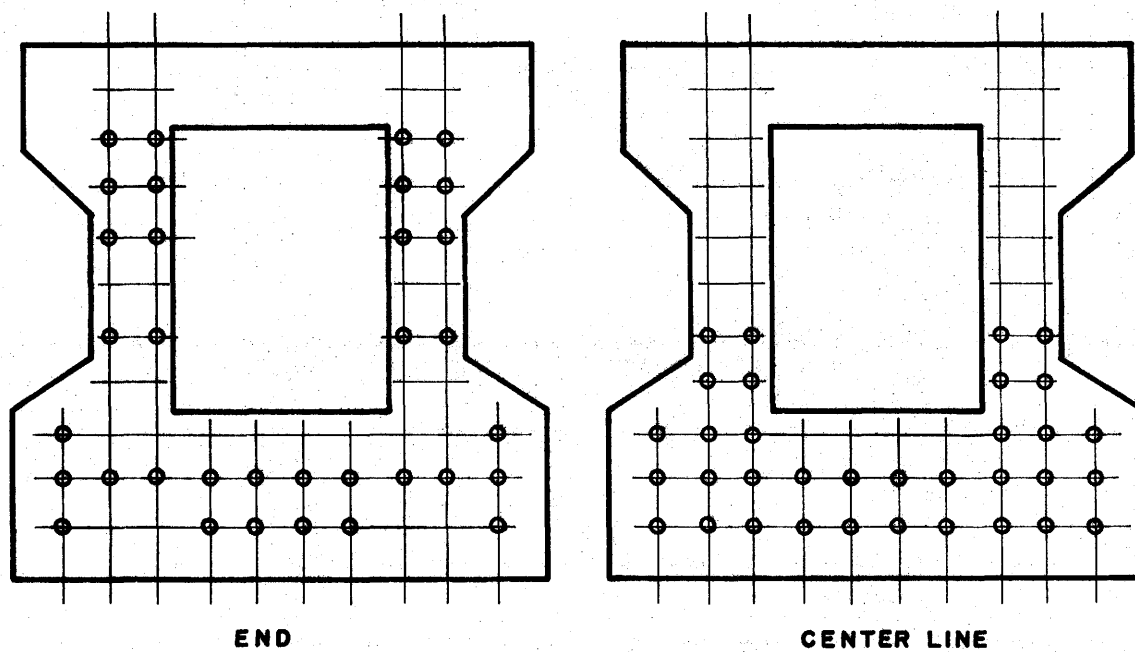


FIGURE 18. CENTERLINE AND END SECTIONS OF BEAM WITH IMPROPER DRAPING OF STRANDS

To insure that the allowable tensile stress in the top of the beam is not exceeded, it is necessary that

$$-(1 - \xi)F_o \sum_{i=1}^{NR} \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i - (1 - \xi)F_o \frac{NW}{Z_t} \tau_j GS \cdot EN$$

$$-.003727S_{tj} \sum_{i=1}^{10} K_i \leq \frac{M_j}{Z_t} + .06387S_{tj}; \quad j=5, 6, 7, 8 \quad (126)$$

Referring to Figure 16, τ_j is given by

$$\tau_j = \begin{matrix} 0. & ; j=1 & \frac{\langle \alpha - .125 \rangle}{\alpha} & ; j=5 \\ \frac{\langle \alpha - .40 \rangle}{\alpha} & ; j=2 & \frac{\langle \alpha - .10 \rangle}{\alpha} & ; j=6 \\ \frac{\langle \alpha - .30 \rangle}{\alpha} & ; j=3 & 1. & ; j=7 \\ \frac{\langle \alpha - .20 \rangle}{\alpha} & ; j=4 & 0. & ; j=8 \end{matrix} \quad (127)$$

The bracketed quantity in Eq. (127) has the following interpretation:

$\langle x \rangle = x$ if $x > 0.$, $\langle x \rangle = 0.$ if $x \leq 0.$

To insure that the allowable compressive stress in the bottom of the beam is not exceeded, it is necessary that

$$(1 - \xi)F_o \sum_{i=1}^{NR} \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_i - (1 - \xi)F_o \frac{NW}{Z_b} \tau_j GS \cdot EN$$

$$-0.5S_{cj} \sum_{i=1}^{10} K_i \leq 4.0S_{cj} + \frac{M_j}{Z_b}; \quad j = 5, 6, 7, 8 \quad (128)$$

3.2.3 Constraints on Service Load Stresses [Constraints 9 thru 22]

Service load stresses are checked top and bottom at midspan, $4L/10$, $3L/10$, $2L/10$, $5L/40$, $L/10$ and the end. To insure that the allowable tension stress in the bottom of the beam at all points except the end do not ex-

ceed the allowable tensile stress, it is necessary that

$$\begin{aligned}
 & -(1 - \eta)F_0 \sum_{i=1}^{NR} \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_i + (1 - \eta)F_0 \frac{NW}{Z_b} \tau_j GS \cdot EN - .007454 \bar{S}_{tj} \cdot \\
 & \sum_{i=1}^{10} (g_i - g_{i-2}) K_i \leq - \frac{M_j}{Z_b} - \frac{\bar{M}_j}{Z_b} + .007454 \bar{S}_{tj} g_0 + .03355 \bar{S}_{tj} ; \\
 & j=1, \dots, 6
 \end{aligned} \tag{129}$$

At the end of the beam ($j=7$), tension stress in the top is limited by

$$\begin{aligned}
 & -(1 - \eta)F_0 \sum_{i=1}^{NR} \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i - (1 - \eta)F_0 \frac{NW}{Z_t} \tau_7 GS \cdot EN \\
 & - .007454 \bar{S}_{t7} \sum_{i=1}^{10} (g_i - g_{i-1}) K_i \leq .007454 \bar{S}_{t7} g_0 + .03355 \bar{S}_{t7} \tag{130}
 \end{aligned}$$

The compression stress in the top of the beam at all points except the end is limited to the allowable by

$$\begin{aligned}
 & (1 - \eta)F_0 \sum_{i=1}^{NR} \left\{ \frac{1}{A} + \frac{d_i}{Z_t} \right\} NS_i + (1 - \eta)F_0 \frac{NW}{Z_t} \tau_j GS \cdot EN \\
 & - \bar{S}_{cj} \sum_{i=1}^{10} (g_i - g_{i-1}) K_i \leq - \frac{M_j}{Z_t} - \frac{\bar{M}_j}{Z_t} + \bar{S}_{cj} g_0 ; \\
 & j=1, \dots, 6
 \end{aligned} \tag{131}$$

and at the end of the beam, the compression stress in the bottom is limited by

$$\begin{aligned}
 & (1 - \eta)F_0 \sum_{i=1}^{NR} \left\{ \frac{1}{A} - \frac{d_i}{Z_b} \right\} NS_i - (1 - \eta)F_0 \frac{NW}{Z_b} \tau_j GS \cdot EN \\
 & - \bar{S}_{c7} \sum_{i=1}^{10} (g_i - g_{i-1}) K_i \leq \bar{S}_{c7} g_0
 \end{aligned} \tag{132}$$

3.2.4 Sufficient Number of Strands in Row for Draping [Constraints 23 thru (22 + NR)]

Two variables, NS_i and I_i are associated with each strand row. If strands are to be draped in row i ($I_i = 1$), then there must be at least NW strands present in that row ($NS_i \geq NW$). This is assured by the constraint set

$$-NS_i + NW \cdot I_i \leq 0 \quad i=1, \dots, NR \quad (133)$$

3.2.5 Contiguous Draped Strands [Constraints (23 + NR) thru (21 + 2·NR)]

Fabrication practices require that strands which are draped must be in adjacent rows. For example, this condition would not be acceptable: row 1 with draped strands ($I_1=1$), row 2 with no draped strands ($I_2=0$) and rows 3, 4 and 5 with draped strands ($I_3=I_4=I_5=1$). This situation is depicted in Figure 18, where it can be seen that the draped strands in row 1 would have to cross over those in the second row as the first row strands were raised. This condition is precluded by the constraint set

$$i(I_{i+1} - I_i) - NB \leq 0 \quad i=1, \dots, (NR-1) \quad (134)$$

where NB is the row number of the first row containing draped strands.

3.2.6 Upper Bound on EN [Constraints (22 + 2·NR) thru (21 + 3·NR)]

The maximum number of rows by which draped strands can be raised at the end of the beam depends on how many of the NR possible rows that may contain strands actually have them. Referring to Figure 13, which has 5 rows filled, it can be seen that at most, the top most 3 rows that are draped can be raised by 5 rows. If there were six rows containing strands

(with the 3 top most draped), then the maximum lift at the end would be 4 rows. In the first case, the maximum value permitted for EN would be 15. For the latter case, EN could not exceed 12. Thus, the upper bound on EN depends on the total number of rows containing strands, which is given by

$$NB-1 + \sum_{i=1}^{NR} I_i \quad (135)$$

and the total number of strand rows available, NRAV. Upper bounds on EN are imposed through the following set of constraints

$$EN + [J + 1]NB - [NRAV - 2J - 1] \sum_{i=1}^{NR} I_i \leq J^2 + 2J + 1;$$

$$J=0, 1, \dots, (NR-1) \quad (136)$$

3.2.7 Upper and Lower Bounds on NB [Constraints (22 + 3·NR) and (23 + 3·NR)]

The number of the first row containing draped strands, NB, must lie between 1 and NR, i.e.,

$$NB \leq NR \quad (137)$$

$$-NB \leq -1 \quad (138)$$

3.2.8 Constraints to Insure that if $NB \geq i + 1$, then $I_i = 0$

[Constraints (24 + 3·NR) thru (22 + 4·NR)]

From the definition of NB and I_i it follows that for those rows below row NB, $I_i=0$ ($i=1, \dots, NB-1$). The constraint set defined in Section 3.2.5 insures that draped strand rows are contiguous, that is, $I_i=0$ for rows i

below the draped strand rows and above the top most draped strand. However, as yet, nothing ties the first draped strand row (the first row i for which $I_i \neq 0$) to NB. For example, at this point, nothing would prevent the occurrence of a situation where $NB=3$ and $I_1=I_2=I_3=\dots=1$. The following constraint set insures that this arrangement does not occur.

$$NB + NR \cdot I_i \leq NR + i \quad i=1, 2, \dots, (NR-1) \quad (139)$$

3.2.9 Constraints to Insure that if $NB = i$, then $I_i = 1$ [Constraints (23 + 4·NR) thru (22 + 5·NR)]

The constraint sets in Sections 3.2.5 and 3.2.8 do not preclude the occurrence of a situation where $NB=i$ and $I_i \neq 1$. For example, $NB=3$ and $I_1=I_2=I_3=\dots=0$. This condition is prevented by

$$-NB - (i + 1) \sum_{j=1}^{i-1} I_j - I_i \leq -(i + 1); \quad i=1, 2, \dots, NR \quad (140)$$

3.2.10 Maximum Number of Strands per Row [Constraints (23 + 5·NR) thru (22 + 6·NR)]

Letting the maximum number of strands that can be placed in row i (straight strands plus draped strands) be denoted by NM_i , we have

$$NS_i \leq NM_i \quad ; \quad i=1, 2, \dots, NR \quad (141)$$

3.2.11 Constraints to Insure Proper Release Strength Representation [Constraints (23 + 6·NR) thru (31 + 6·NR)]

The binary variable representation of f'_{ci} defined in Eq. (121) is valid only if the inequalities appearing in Eq. (122) are satisfied. Thus, it is required that

$$-K_i + K_{i+1} \leq 0 \quad i=1, \dots, 9 \quad (142)$$

3.2.12 Bounds on Release Camber [Constraints (32 + 6·NR) and (33 + 6·NR)]

Letting Δ^+ and Δ^- denote the maximum and minimum midspan camber allowed, and using Eq. (117) for the camber which occurs, we have

$$E_{ci} I_{\Delta DL} - \left\{ \frac{L^2}{8} M_s + \frac{(\alpha L)^2}{6} M_E + \left[\frac{L^2}{8} - \frac{(\alpha L)^2}{6} \right] M_D \right\} \leq E_{ci} I_{\Delta^+} \quad (143)$$

for the upper bound constraint on camber. Substitution of Eqs. (118) thru (120) for M_s , M_E and M_D and noting that the modulus of the concrete E_{ci} can be written approximately as

$$E_{ci} = K \left\{ .003727 \sum_{i=1}^{10} K_i + .06337 \right\} \quad (144)$$

where K is defined in Eq. (72), we have

$$\begin{aligned} & -\frac{L^2}{8} (1 - \xi) F_0 \sum_{i=1}^{NR} d_i N S_i \\ & - \frac{(\alpha L)^2}{6} (1 - \xi) F_0 N W \cdot G S \cdot E N - .003727 I \cdot K \Delta^+ \sum_{i=1}^{10} K_i \\ & \leq .06337 K \cdot I_{\Delta^+} + 22.5 W L^4 \end{aligned} \quad (145)$$

In a similar manner, the lower bound Δ^- is imposed through

$$\begin{aligned} & \frac{L^2}{8} (1 - \xi) F_0 \sum_{i=1}^{NR} d_i NS_i \\ & + \frac{(\alpha L)^2}{6} (1 - \xi) F_0 NW \cdot GS \cdot EN + .003727 \cdot K \Delta^- \sum_{i=1}^{10} K_i \\ & \leq .06337K \cdot I \Delta^- - 22.5L^4 \end{aligned} \quad (146)$$

3.2.13 Adequate Ultimate Moment Capacity [Constraint (34 + 6·NR)]

The discussion in Section 2.2.8 applies in this case, with Eq. (80) being modified to give

$$\sum_{i=1}^{NR} d_i NS_i + \sum_{i=1}^{10} (\bar{\rho}_i - \bar{\rho}_{i-1}) K_i \leq -\bar{\rho}_0 \quad (147)$$

3.2.14 Ultimate Moment Capacity $M_u \geq 1.2M_{cr}$ [Constraint (35 + 6·NR)]

The change of variables used above also applies for this constraint, giving

$$\sum_{i=1}^{NR} d_i NS_i + \sum_{i=1}^{10} (\rho_i - \rho_{i-1}) K_i \leq \rho_0 \quad (148)$$

3.2.15 Lower and Upper Bounds on Concrete Strength [Constraints (36 + 6·NR) and (37 + 6·NR)]

The lower limit $(f'_{ci})_{\min}$ and upper limit $(f'_{ci})_{\max}$ on concrete release strength are enforced by

$$-0.5 \sum_{i=1}^{10} K_i \leq 4.0 - (f'_{ci})_{\min} \quad (149)$$

accomplished by first solving the integer formulation of Section 3.2 or 3.3 as a standard linear programming problem (i.e., assuming all variables are continuous). A benefit derived from this approach is the possible detection of unrealistic design requirements as discussed in Section 2.4. There is no guarantee, however, that such a condition will be detected in the L.P. problem since it is theoretically possible to obtain a solution to a continuous problem where no feasible solution in integers exists.

3.5 VARIABLE CORRESPONDENCE FOR DRAPED STRAND DESIGN

The correspondence between the variable notation, x_1, \dots, x_n used in Eqs. (1) and (2) and that incorporated in this chapter is as follows:

3.5.1 Optimization Option

$$\begin{aligned}
 x_1 &= NS_1 \\
 &\cdot \\
 &\cdot \\
 &\cdot \\
 x_{NR} &= NS_{NR} \\
 &= NB \\
 x_{NR+1} &= EN \\
 x_{NR+2} &= EN \\
 x_{NR+3} &= I_1 \\
 &\cdot \\
 &\cdot \\
 &\cdot \\
 x_{2 \cdot NR+2} &= I_{NR} \\
 x_{2 \cdot NR+3} &= K_1 \\
 &\cdot \\
 &\cdot \\
 &\cdot \\
 &\cdot \\
 &\cdot \\
 x_{2 \cdot NR+12} &= K_{10}
 \end{aligned} \tag{162}$$

3.5.2 Design Option

$$x_1 = f'_{ci}$$

$$x_2 = NS_1$$

•

•

•

•

$$x_{NR+1} = NS_{NR}$$

$$x_{NR+2} = NB$$

(163)

$$x_{NR} + 3 = EN$$

$$x_{NR} + 4 = I_1$$

•

•

$$x_{2 \cdot NR+3} = I_{NR}$$

IV. PROGRAM DOCUMENTATION - DBOXSS

The computer program DBOXSS implements the box girder design formulation developed in Chapter II. Described below are the standard input form and its use, interpretation of program output and several example problems.

4.1 PROGRAM INPUT

Figure 19 shows the input form to be used with the program.

4.1.1 Title Cards

The first three input cards are title cards providing a means of job reference. The information preprinted on the form in various columns need not be punched on the data cards - it will be printed out automatically during output. The information on these cards is optional. The first two cards should only be input once per computer run. The third title card is the first card in a data pack when multiple problem runs are made, as explained below.

4.1.2 Load and Options Card



















The type of standard AASHTO loading (H-15, H-20, HS-15 or HS-20) is entered in columns 5 - 6 and 8 - 9. The live load distribution factor entered in columns 13 - 16 is the fraction of an axle load to be carried by the beam. This distribution factor is applied to the axle train loading (if used) as well as AASHTO truck and lane loadings. If columns 13 - 16 are left blank, the program automatically computes lateral distribution using Eqs. (24) thru (26), (the AASHTO distribution factor). If a vehicle other than an AASHTO truck is to be used for design, enter a "1" in column 20 and complete the axle train data cards. If both axle train and AASHTO

Sheet ____ of ____

BOX BEAM DESIGN PROGRAM (STRAIGHT STRANDS)

AXLE TRAIN

Axle Loads (kips)

$i = 2$ $i = 3$ $i = 4$ $i = 5$ $i = 6$ $i = 7$ $i = 8$ $i = 9$ $i = 10$ $i = 11$ $i = 12$ $i = 13$ $i = 14$ $i = 15$ $i = 16$ $i = 17$ $i = 18$

Distance From Axle 1 to Axle i

CONCENTRATED LOADS ON SINGLE BEAM

Load (kips)
Distance Fine Left Support (ft)

BEAM DIMENSIONS

A (in)		B (in)		C (in)		D (in)		E (in)		F (in)		G (in)		H (in)		M (in)		T (in)		W (in)		X (in)		Y (in)	
4	7	9	12	14	17	19	22	24	27	29	32	34	37	39	42	44	47	49	52	54	57	59	62	64	67

Span Length (ft)	Bridge Width (ft)	Number Of Traffic Lanes	Number Of Beams	Area of Compression Reinforcing (in ²)	Distance to c.g. of Compression Reinforcing (in)	Maximum Initial Camber (in)	Minimum Initial Camber (in)	Distance From Bottom of Beam to Strand Row
4 7	11 14	18 19	23 26	29 32	36 39	43 47	51 55	59 60

Enter 1 to Read Nonstandard Grid Spacing Card
Enter 1 to Read Misc. Properties Card

MAXIMUM NUMBER OF STRANDS PERMITTED IN ROW i

4 5 7 8 10 11 13 14 16 17 19 20 22 23 25 26 28 29 31 32 34 35 37 38 40 41 43 44 46 47 49 50 52 53 55 56 58 59 61 62 64 65 67 68 70 71 73 74 76 77 79 80

FIGURE 19. INPUT FORM FOR DBOXSS

FIGURE 19. (CONTINUED)

loadings are specified, the larger of the axle train, AASHTO truck and AASHTO lane moments are used at each design point (Figure 2). A uniform dead load carried by a single box girder (i.e. with no lateral distribution of load) is entered in columns 24 - 27. This provision allows the user to include the weight of such things as a wearing surface in the design. Concentrated dead load forces applied to a single girder are indicated by placing a "1" in column 31 and completing the concentrated loads on a single beam data cards. Column 48 dictates which program option is to be used. If a "1" is entered, the program determines the minimum cost design, based on cost information input from part 2 of the form. This is the "optimization option". If column 48 is left blank, the program exercises the "design option", in which the strand pattern and release strength are selected which minimize the number of strands used, assuming the beam concrete has the 28 day strength entered in columns 35 - 37.

4.1.3 Axle Train Cards (Use Only if "1" Entered in Column 20 of Load and Options Card)

A moving load pattern of up to 18 axles may be used for design. The first card contains the total load on each axle. To facilitate input, the user should sketch the axle train configuration, labeling either the right-most or left-most axle as axle 1 and numbering the remaining axles in sequential order. The weight of each axle is then placed in the appropriate columns of the first data card. The spacing of axles is input on the second data card, where axle spacing is defined as the distance from axle 1 to the axle under consideration. As an example, an AASHTO HS-20 truck (with rear axles separated 14 feet) whose

light axle was designated as axle 1, would require 8.0 in columns 4 - 6 on the first card and 32.0 in columns 8 - 10 and 12 - 14. The second card would contain 14. in columns 8 - 10 and 28. in columns 12 - 14. The program automatically scales axle train axle loads by the lateral distribution factor, but no impact factor is applied.

4.1.4 Concentrated Loads on Single Beam Cards (Use Only if Column 31 of Load and Options Card Contains "1")

Up to 10 concentrated forces acting on a single beam (no lateral distribution assumed) may be input. The first card contains the magnitude of the load, while the second card contains the distance of each load from the left support. This program provision is intended for small loads only. Service load stress checks are based on the assumption that the maximum moment due to all dead plus live load occurs either at the 1/10, 2/10 or 5/10 point. If large concentrated forces are entered, this assumption may be in error, resulting in an overstressed design.

4.1.5 Beam Dimensions Card

The dimensions of the beam cross section which are to be input are shown on the figure at the upper right corner of part 1 of the input form. The fillets (dimensions X and Y) are assumed to slope at 45 degrees. Most any cross sectional shape can be accommodated with the dimensions shown. An ordinary rectangular voided section can be obtained, for example, by inputting dimensions such that $A=B=(2 \cdot W+M)$ and $G=H=E=C=0$ (or left blank).

4.1.6 Bridge and Beam Properties Card

The information input on this card is used to compute the lateral distribution factor (if columns 13 - 16 of the load and options card is left blank) and other quantities used to formulate the constraint set.

The span length is entered in columns 4 - 7, bridge width in columns 11 - 14, number of traffic lanes in columns 18 - 19 and number of longitudinal beams in columns 23 - 26. The number of longitudinal beams is input as a decimal number to accommodate unusual conditions (such as a mixture of two or more different box cross sections in the same bridge). Compression steel is sometimes used in box sections to help control long term camber. The area of this steel, which is input in columns 29 - 32, is considered in the computation of section properties, using a transformed steel area of $(n-1)$ for properties with shear key and $2(n-1)$ without. The distance from the top of the beam to c.g. of compression steel is entered in columns 36 - 39. If left blank, the program assumes $T/2$. Maximum and minimum acceptable release cambers are input in columns 43 - 47 and 51 - 55 (upward camber is positive, downward camber negative). These apply to the midspan camber at release produced by prestress and beam weight. A typical application of the lower bound camber would be to insure that a beam did not deflect downward under full dead load (say, shear key plus wearing surface). If an estimate of the final release strength is made, then a modulus of elasticity can be computed, and the midspan downward deflection under shear key and wearing surface weights determined. This value is entered (as a positive number in this case) under minimum initial camber. This will insure that the final design has enough upward initial camber to offset the downward deflection caused by the addition of shear key and wearing surface. If columns 43 - 47 or 51 - 55 are left blank, then the constraint is ignored during design. The distance from the bottom of the beam to the centerline of the first (bottom-most) strand row is input in columns 59 - 60. If column 64 is left blank, the spacing between all rows of strands is assumed

to be 2.0 inches. If this is not the case, enter "1" in column 64 and complete the non-standard grid spacing card. If column 68 is left blank, the program assumes normal weight concrete (150 lbs/ft^3), 50% relative humidity and $.153 \text{ in.}^2$ grade 270 strands. The allowable stress coefficients are taken as 0.6 for compression and 7.5 for tension at release and 0.4 and 6.0 under service loads.

4.1.7 Maximum Number of Strands per Row

The maximum number of strands, as well as the number of rows available for strands is determined from this input data. Strand rows are numbered consecutively, taking the bottom-most row as row 1. The computation time required to obtain a final design increases rapidly as the number of available strand rows increases. Thus, one should include only those rows which will likely be used. As written, the program is limited to 10 rows of strands.

4.1.8 Nonstandard Grid Spacing Card (Use Only if "1" Entered in Column 64 of Bridge and Beam Properties Card)

The spacings entered on the data card are the distance from the row under consideration to the row above. Thus, the center to center spacing between rows 1 and 2 would be placed in columns 4 - 5, between rows 2 and 3 in columns 7 - 8, etc. If a uniform spacing (different from 2.0 in.) is to be used, only columns 4 - 5 need be completed. The program will automatically assume this uniform spacing throughout if it encounters no other entries beyond columns 4 - 5.

4.1.9 Miscellaneous Properties Card (Use Only if "1" Entered in Column 68 of Bridge and Beam Properties Card)

If properties other than the standard values listed in Section 4.1.6 are to be used, they must be entered on this card. Only those properties which differ from standard values need be entered. If the program encounters

blanks on the card where a property is to be read, it automatically assumes the standard value. The unit weight of concrete, if different from .150 k/ft³, is entered in columns 4 - 6. Relative humidity is entered in columns 11 - 12, strand area in columns 17 - 19 and ultimate strength of strands in columns 24 - 26. The coefficients used to specify allowable stresses are entered in columns 31 - 55. If the allowable compressive release stress differs from $0.6 f'_{ci}$ or the allowable tensile stress from $7.5 \sqrt{f'_{ci}}$, then the coefficients (those that replace 0.6 and 7.5) should be entered in columns 31 - 32 and 34 - 35 for the end 1/10 of the beam ($j = 8, 9, 10$ & 11 in Figure 2) and in columns 37 - 38 and 40 - 41 for the remainder of the beam. If the allowable compressive long term stress differs from $0.4 f'_c$ or the allowable long term tensile stress from $6.0 \sqrt{f'_c}$, then the new coefficients should be entered in columns 45 - 46 and 48 - 49 for the end 1/10 of the beam and in columns 51 - 52 and 54 - 55 for the remainder of the beam.

The final beam camber at midspan after all prestress losses and creep and shrinkage effects have occurred is computed using the method developed in reference (2). Cambers are computed and displayed using four different sets of creep and shrinkage coefficients typical of concretes in highway beams produced in four localities in Texas. Should the designer have information on the creep and shrinkage properties of the concrete he expects to be used in a particular design, he may enter the appropriate coefficients in columns 58 - 75. The program will then compute and display the expected midspan camber for these conditions.

4.1.10 Concrete Cost Coefficients Cards (Use Only if "1" Entered in Column 48 of Load & Options Card)

The cost of concrete in dollars per cubic yard can be input for concrete release strengths up to 9.0 ksi. If an estimate of the cost per cubic yard for 4.0 ksi release strength can be made, Figure 7 can be used as a guide to establishing the cost of higher strengths in the absence of actual cost data. Should release strengths beyond some value (say, 7.0 ksi) not be feasible, then the values (7.5, 8.0, 8.5 and 9.0) beyond that point should be left blank.

4.1.11 Strand Cost Card (Use Only if "1" Entered in Column 48 of Load & Option Card)

The cost per foot for strand is entered in columns 14 - 16 and the cost of strand wrapping in columns 63 - 65.

4.1.12 28 Day Concrete Strength Cards (Use Only if "1" Entered in Column 48 of Load & Options Card)

The relationship between release strength and 28 day strength is defined by the data input from these cards. It isn't possible to construct a general relationship between release and 28 day strengths because of the many factors that influence it. There are similarities in the operations of the major producers of highway beams in the state of Texas (12) which permit a reasonable estimate of the relationship.

Fabricators generally use a 24 hour steam curing production sequence. Beams are cast in the afternoon, allowed to gain their initial set (minimum of 3 hours after casting before steam curing is begun (13)) and then steam cured overnight. A total steam curing period of 18 hours at 140° to 150° F.

is typical of most operations. Hanson (14) collected data on concretes made with type III Portland cement and subjected to 15 hours of steam curing at 150° F. commencing 3 hours after casting. Concrete strengths were generally around 4 ksi at 18 hours (release) and 5 ksi at 28 days. For the data reported, the 18 hour strengths averaged 74% of the 28 day strengths. This percentage is probably valid over the usual range of release strengths utilized in THD prestressed designs, which is approximately 6 ksi. If this percentage is applied to release strengths between 4 and 6 ksi, the first straight line segment shown in Figure 20 is produced. Higher release strengths generally demand longer periods of steam curing and result in smaller percentage gains in release strength over 28 day strength. At the extreme limit of 9 ksi release strength, the fabricator would no doubt be forced to keep the beams under special cure for the majority of a 28 day period before release strength was reached. Under these conditions, the ratio of release strength to 28 day strength should be approximately 1. If a linear variation in strength gain over release strength is assumed between 6. and 9. ksi release strengths, the second straight line segment shown in Figure 20 is obtained.

4.1.13 Multiple Problem Runs

The user may process more than one design in a single computer run. The first problem must contain the three title cards described in Section 4.1.1. Each additional problem which is run should have the third title card as the first card in the data set.

4.2 SAMPLE PROBLEMS

Described below are several example problems demonstrating the use of

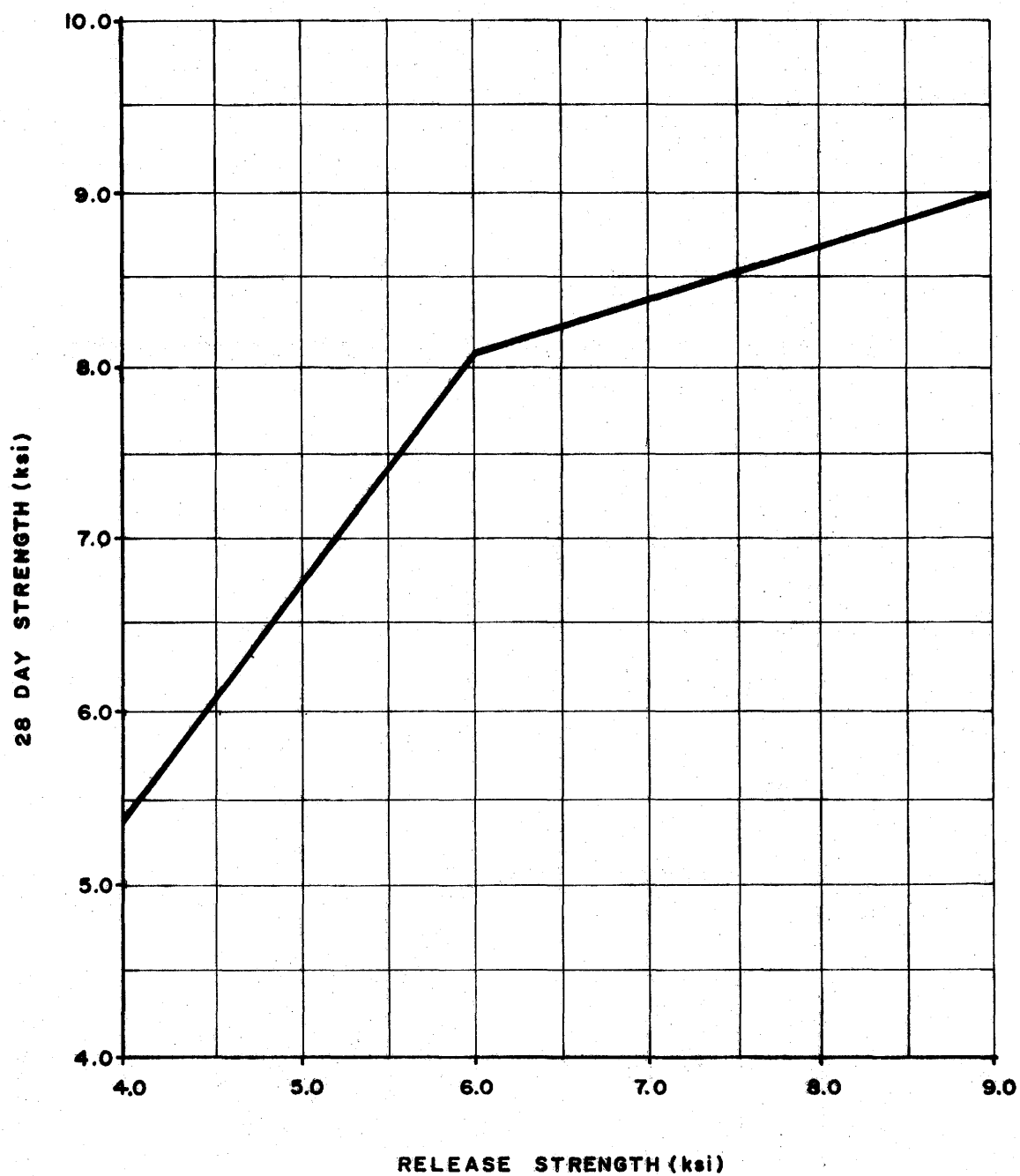


FIGURE 20. PROPOSED RELATIONSHIP BETWEEN 28 DAY STRENGTH
AND RELEASE STRENGTH

the standard input form.

4.2.1 Example Problem 1

A multibeam box girder bridge is to carry two lanes of traffic with HS-20 loading and span 35.0 ft. The girder cross section is to be formed using the THD type "B" side form and is to have an overall width of 4.0 ft. A 27 3/4 in. wide by 23 1/2 in. deep void is to be used, and the girder has one interior diaphragm weighing 440 lbs. Seven boxes will be used, with a 1.2 in. space between adjacent boxes, giving an overall bridge width of 28.6 ft. The bridge will be surfaced with a 2 in. asphaltic concrete topping. The design will incorporate a 5.0 ksi 28 day strength and 1/2 in. diameter grade 270k strands. The dimensions of the section are such that 5 rows of strands can be placed. The first row of strands is 2.5 in. above the bottom of the section and can accommodate a maximum of 22 strands. The remaining 4 rows are spaced uniformly at 2.0 in. increments and each can hold 6 strands. However, no more than one row of 22 strands should be required. Six number 9, grade 60 conventional reinforcing bars are to be placed in the deck slab, 2.5 in. down from the surface.

The shear key weighs approximately 180 lbs/ft. (1/2 key on either side of the beam) and the 2 in. wearing surface 90 lbs./ft. Assuming the release strength for the design will be approximately 4.0 ksi, the modulus of elasticity of the concrete will be 3.83 million psi. The moment of inertia of the cross section without shear key is approximately 122,000 in.⁴. The total downward deflection of a single girder, under the weight of shear key and topping based on the assumed modulus is computed to be 0.02 in. This value is specified as the minimum initial camber permitted. For this relatively

short span, excessive camber should not be a factor and is therefore to be ignored in design.

The bridge is to be situated in a coastal environment, so no tension will be allowed in the bottom of the beam, under service loads, where it is exposed to the air. Tension may occur in the top of the beam near the end, due to prestress, but since the top will be protected by a wearing surface, the usual tension allowable of $6.0 \sqrt{f'_c}$ will be imposed for the end 1/10 of the beam.

The design option is used for this problem. The completed input form is shown in Figure 21, and the program output in Figure 24.

4.2.2 Example Problem 2

A 44 ft. wide by 50 ft. long bridge is to be constructed from 16 box girder units. The box cross section is that proposed as a standard by the Federal Highway Administration (15) and is shown in Figure 22. The vertical positioning of strand rows is that suggested in Reference (15); the first row 2 3/4 in. above the bottom of the beam and the second row 4 in. above the first. A lateral spacing of 2 in. center-to-center is used between strands in a row and thus permits a maximum of 17 strands in row 1 and 4 strands in row 2. Grade A-416, 7/16 in. diameter strands (ultimate strength of 250 ksi and cross sectional area of .1089 in.²) are to be used. The section contains 5 No. 4 bars for compression reinforcing as shown. The completed structure will carry an asphaltic concrete wearing surface whose average thickness is 5 in. over the two beams at the center of the bridge. The resulting dead load is approximately 160 lbs./ft. per beam. The beam must have an upward camber at release, but can not exceed .75 in. to insure a relatively level riding surface under full dead load. Allowable release and service load stresses in the concrete are those specified in the AASHTO Bridge Specification

FIGURE 21. COMPLETED INPUT FORM FOR EXAMPLE PROBLEM NO. 1

Part 2 of 2

NONSTANDARD GRID SPACING CARD (Enter This Card Only if Previously Specified)

Distance From Row 1 To Row (i+1)

4 5	7 8	10 11	13 14	16 17	19 21	22 23	25 26	29 29	31 32	34 35	37 38	40 41	43 44	46 47	49 50	52 53	55 56	58 59	61 62	64 65	67 68	70 71	73 74	75 76
1=1	1=2	1=3	1=4	1=5	1=6	1=7	1=8	1=9	1=10	1=11	1=12	1=13	1=14	1=15	1=16	1=17	1=18	1=19	1=20	1=21	1=22	1=23	1=24	1=25

MISCELLANEOUS PROPERTIES CARD (Enter This Card Only if Previously Specified)

Unit Weight Concrete (K/Cu.Ft.)	Relative Humidity (%)	Strand Area (in ²)	Strand Ultimate Strength (ksi)	Allowable Stress Coefficients (ksi)				Creep & Shrinkage Coefficients (μ -in. & Days)							
				Release		Long Term		Creep 1		Creep 2		SHRK 1		SHRK 2	
4 6	11 12	17 19	24 26	C	T	C	T	C	T	C	T	58 60	63 65	68 70	73 75
				31 32	34 35	37 38	40 41	45 46	48 49	51 52	54 55				
				End 1/10	Remainder	End 1/10	Remainder								

CONCRETE COST COEFFICIENTS (Include Only for Optimization Option)

Release Strength (Ksi) / \$ Cost YD						
4.0Ksi / \$	4.5Ksi / \$	5.0Ksi / \$	5.5Ksi / \$	6.0Ksi / \$	6.5Ksi / \$	
7.0Ksi / \$	7.5Ksi / \$	8.0Ksi / \$	8.5Ksi / \$	9.0Ksi / \$		
11 14	24 27	37 40	50 53	63 66	76 79	

STRAND COST \$ / Linear Foot (Optimization Only)

14	16
----	----

STRAND WRAPPING COST \$ / Linear Foot (Optimization Only)

63	65
----	----

28 DAY CONCRETE STRENGTHS (Include Only for Optimization)

Release Strength / 28 Day Strength					
4.0Ksi /	4.5Ksi /	5.0Ksi /	5.5Ksi /	6.0Ksi /	6.5Ksi /
7.0Ksi /	7.5Ksi /	8.0Ksi /	8.5Ksi /	9.0Ksi /	
11 13	24 26	37 39	50 52	63 65	76 78

FIGURE 21 (CONTINUED)

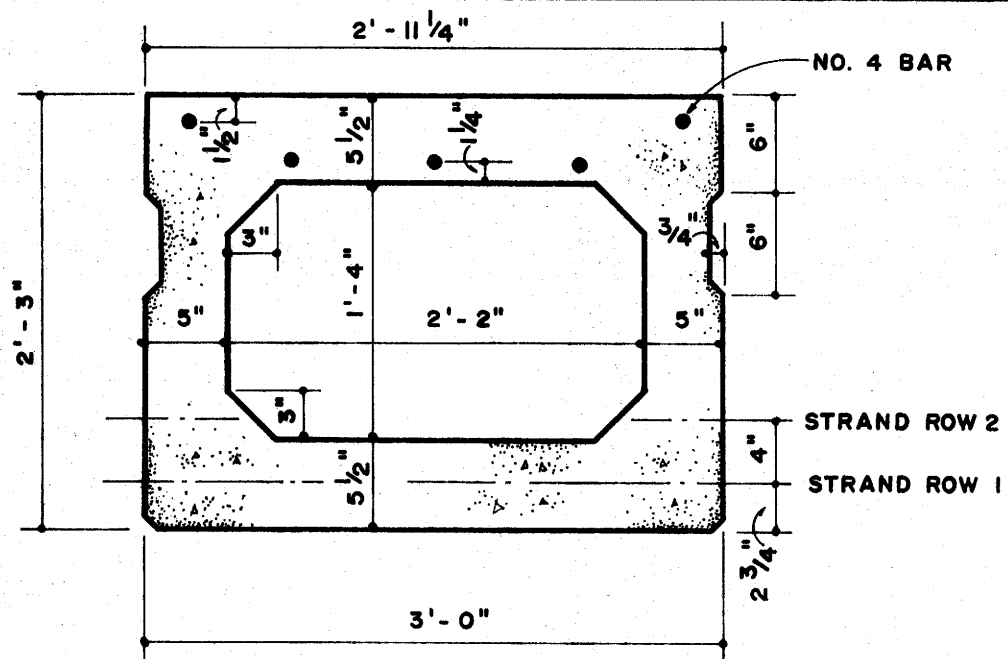


Figure 22. BEAM CROSS SECTION FOR EXAMPLE NO. 2

(4) for inland regions.

The in-place cost of beam concrete is assumed to be \$150/cu.yd. for release strengths up to 5.5 ksi, \$200 for $f'_{ci} = 6.0$ ksi and \$230 for $f'_{ci} = 6.5$ ksi. Release strengths greater than 6.5 ksi are assumed to be unavailable. The 28 day concrete strength corresponding to each release strength is assumed to be that given in Figure 20.

The optimization option is used for this problem with the completed input form shown in Figure 23. Program output for this problem is shown in Figure 25.

4.3 INTERPRETATION OF PROGRAM OUTPUT

The program first reads input data and performs basic checks for input errors. At present, the program checks include; proper AASHTO loading designation and omitted span length and omitted number of longitudinal beams (if no live load distribution factor is input). The input data is printed out before design calculations are begun so that the user can locate input errors which might cause the program to terminate abnormally before producing any other output. The output format is essentially the same for design and optimization options (see first sheet, Figures 24 and 25).

The second sheet of output summarizes the design results, as seen in Figures 24 and 25. The first items listed are the release and 28-day concrete strengths. For the design option, the 28-day strength is that specified on input, while the release strength is computed by the program. In the optimization option, release strength is computed and 28 day strength is obtained from 28-day vs. release strength input data. The modulus of elasticity listed is computed using the ACI equation (10) and the unit

Part 1 of 2

Sheet 1 of 2

AXLE TRAIN

Axle Loads (kips)

CONCENTRATED LOADS ON SINGLE BEAM

4 8 10 14 16 20 22 26 28 32 34 38 40 44 46 50 52 56 58 62

Load (kips)
Distance From Left Support (ft)

BEAM DIMENSIONS

A (in)	B (in)	C (in)	D (in)	E (in)	F (in)	G (in)	H (in)	M (in)	T (in)	W (in)	X (in)	Y (in)
35	25	15	27	75	55	75	6	26	55	5	3	3
4 7	9 12	14 17	19 22	24 27	29 32	34 37	39 42	44 47	49 52	54 57	59 62	64

Span Length (ft)	Bridge Width (ft)	Number Of Traffic Lanes	Number Of Beams	Area of Compression Reinforcing (in²)	Distance to c.g. of Compression Reinforcing (in)	Maximum Initial Camber (in)	Minimum Initial Camber (in)	Distance From Bottom of Beam to Strand Row
<div style="border: 1px solid black; padding: 2px;">50</div>	<div style="border: 1px solid black; padding: 2px;">44</div>	<div style="border: 1px solid black; padding: 2px;">4</div>	<div style="border: 1px solid black; padding: 2px;">16</div>	<div style="border: 1px solid black; padding: 2px;">10</div>	<div style="border: 1px solid black; padding: 2px;">31.5</div>	<div style="border: 1px solid black; padding: 2px;">.75</div>	<div style="border: 1px solid black; padding: 2px;">0.</div>	Enter 1 to Read Nonstandard Grid Spacing Card Enter 1 to Read Misc. Properties Card
4 7	11 14	18 19	23 26	29 32	36 39	43 47	51 55	<div style="margin-right: 20px;"><div style="border-top: 1px solid black; width: 20px;"></div>2.8 59.60</div> <div style="margin-right: 20px;"><div style="border-top: 1px solid black; width: 20px;"></div>1 64</div> <div><div style="border-top: 1px solid black; width: 20px;"></div>1 68</div>

MAXIMUM NUMBER OF STRANDS PERMITTED IN ROW :

[illegible]














FIGURE 23. COMPLETED INPUT FORM FOR EXAMPLE PROBLEM NO. 2

Distance From Row i To Row $(i+1)$

4 5 7 8 10 11 13 14 16 17 19 21 22 23 25 26 28 29 31 32 34 35 37 38 40 41 43 44 46 47 49 50 52 53 55 56 58 59 61 62 64 65 67 68 70 71 73 74 75 76

$i=1$ $i=2$ $i=3$ $i=4$ $i=5$ $i=6$ $i=7$ $i=8$ $i=9$ $i=10$ $i=11$ $i=12$ $i=13$ $i=14$ $i=15$ $i=16$ $i=17$ $i=18$ $i=19$ $i=20$ $i=21$ $i=22$ $i=23$ $i=24$ $i=25$

MISCELLANEOUS PROPERTIES CARD (Enter This Card Only if Previously Specified)

Unit Weight Concrete (k / Cu.Ft.)	Relative Humidity (%)	Strand Area (in²)	Strand Ultimate Strength (ksi)	Allowable Stress Coefficients (ksi)								Creep & Shrinkage Coefficients (μ-in. & Days)							
				Release				(Long Term)											
				C	T	C	T	C	T	C	T	CREEP 1	CREEP 2	SHRK 1	SHRK 2				
																			
4 6	11 12	17 19	24 26	31 32	34 35	37 38	40 41	45 46	48 49	51 52	54 55	58 60	63 65	68 70	73 75				
				End 1/10 Remainder				End 1/10 Remainder											

CONCRETE COST COEFFICIENTS (Include Only for Optimization Option)

Release Strength (Ksi) / \$ Cost YD

4.0Ksi / g
7.0Ksi / g

1	5	0	
		0	

4.5Ksi / σ
7.5Ksi / σ

1	5	0	.
		0	.

24

5.0 Ksi / g
8.0 Ksi / g

1	5	0	
		0	

37 40

5.5Ksi / g
8.5Ksi / g

1	5	0	
		0	
50			53

6.0 Ksi /
9.0 Ksi /

2	0	0
		0
63		6

6.5 KSI / 45

230	
76	79

STRAND COST \$ **0.25** / Lineal Foot (Optimization Only)

STRAND WRAPPING COST \$

0	5
63	65

 / Linear Foot (Optimization Only)

28 DAY CONCRETE STRENGTHS (Include Only for Optimization)

Release Strength / 28 Day Strength

4.0 Ksi
7.0 Ksi

	54
0.	

11 13

4.5 Ksi /
7.5 Ksi /

	6.1	
	9	
24		26

5.0Ksi /
8.0Ksi /

	6	7
	9	

37 39

5.5 Ksi /
8.0 Ksi /

	7.4
	2
50	52

6.0 Ksi
9.0 Ksi

81	
63	8

6.3 KSI 7

76	78
----	----

FIGURE 23. (CONTINUED)

87

FIGURE 24. OUTPUT FOR EXAMPLE PROBLEM NO. 1

```

*****
*                                     LOADING CONDITIONS
*                                     *****
*
* AASHTO LL = HS-20      L.L. DISTRIBUTION FACTOR = 0.363 (TRUCKS)
*
* .....CONCENTRATED FORCES ON SINGLE BEAM.....
* *LCAD (KIPS)      0.4      0.0      0.0      0.0      0.0      0.0      0.0      0.0      0.0
* *DISTANCE FROM
* *LEFT SUPPORT (FT) 17.5      0.0      0.0      0.0      0.0      0.0      0.0      0.0      0.0
*
* *UNIFORM LOAD ON SINGLE BEAM = 0.00 (K/FT)
*****

```

FIGURE 24. (CONTINUED)

THE COMMAND IS TO SELECT STRANDS

```

*****
*                                     DESIGN PROPERTIES
*                                     *****
*
* RELEASE STRENGTH = 4.00 (KSI)  CONCRETE MODULUS (RELEASE) = 3024.2 (KSI)  INITIAL PRESTRESS LOSS = 5.57 PERCENT
* 28-DAY STRENGTH = 5.00 (KSI)  TOTAL PRESTRESS LOSS = 16.73 PERCENT
*****

```

```

*****
*                                     DESIGN RESULTS
*                                     *****

```

```

* .....STRAND LAYOUT.....
*
*          ROW      STRANDS      WRAPPED STRANDS IN EACH ROW
*          NUMBER    PER ROW
*          1          7
*
*          THERE ARE NO WRAPPED STRANDS IN THIS ROW

```

```

* .....COMPUTED DEFLECTION.....

```

CONDITION	SHORT TERM		LONG TERM	
	MODULUS	DEFLECTION		
BWWT	RELEASE	0.07 INCHES	-0.08 INCHES (BASED UPON DALLAS CONCRETE PROPERTIES)	
BWWT + KEY	5 MILLION	0.03 INCHES	-0.09 INCHES (BASED UPON ODESSA CONCRETE PROPERTIES)	
BWWT + KEY + DEAD LOAD	5 MILLION	0.02 INCHES	-0.07 INCHES (BASED UPON SAN ANTONIO CONCRETE PROPERTIES)	
			-0.09 INCHES (BASED UPON LUFKIN CONCRETE PROPERTIES)	

```

* .....STIRRUP SPACING - AASHTO 1973 - STIRRUP AREA = 0.11 IN2.....

```

SECTION	0/10	1/10	2/10	1/4	3/10	4/10	5/10
SPACING (IN.)	13.20	13.20	13.20	13.20	13.20	13.20	13.20

FIGURE 24. (CONTINUED)

***** CRITICAL DESIGN FACTORS *****									
***** RELEASE STRESSES *****					***** SERVICE LOAD STRESSES *****				
(SYMBOL X DENOTES STRESS AT ALLOWABLE)					(SYMBOL X DENOTES STRESS AT ALLOWABLE)				
SECTION	STRESS TOP (KSI)	STRESS BOTTOM (KSI)		SECTION	STRESS TOP (KSI)	STRESS BOTTOM (KSI)			
0/20	-0.1430E 00	0.5951E 00		0/10	-0.1462E 00	0.4877E 00			
1/20	-0.1023E 00	0.5536E 00		1/10	0.6773E-01	0.2702E 00			
2/20	-0.6580E-01	0.5164E 00		2/10	0.2236E 00	0.1117E 00			
3/20	-0.3363E-01	0.4836E 00		3/10	0.3275E 00	0.6056E-02			
4/20	-0.5754E-02	0.4552E 00		4/10	0.3774E 00	-0.4463E-01	X		
5/20	-0.1784E-01	0.4312E 00		5/10	0.3833E 00	-0.5071E-01	X		
***** LIST OF DESIGN CONSTRAINTS *****									
(SYMBOL X INDICATES CONSTRAINT CONTROLS FINAL DESIGN)									
MINIMUM CONCRETE STRENGTH			X	ULTIMATE MOMENT		MINIMUM INITIAL CAMBER			
MAXIMUM CONCRETE STRENGTH				CRACKING MOMENT		MAXIMUM INITIAL CAMBER			
***** MOMENT AND SHEAR SUMMARY *****									
SECTION	BEAM DT. PLUS SHEAR KEY MOMENTS (KIP-FT)	OTHER D.L. MOMENTS (KIP-FT)		L.L. MOMENTS (KIP-FT)	TOTAL MOMENTS (KIP-FT)		ULTIMATE SHEAR (KIPS)		
0/10	0.0	0.0		0.0	0.0		0.67154E 02		
1/10	0.5746E 02	0.57312E 01		0.75405E 02	0.13860E 03		0.74153E 02		
2/10	0.10217E 03	0.10360E 02		0.12700E 03	0.23652E 03		0.61152E 02		
1/4	0.11973E 03	0.12261E 02		0.14552E 03	0.27751E 03		0.54651E 02		
3/10	0.13409E 02	0.13886E 02		0.15675E 03	0.30673E 03		0.48151E 02		
4/10	0.15325E 02	0.16310E 02		0.16933E 03	0.33889E 03		0.35150E 02		
5/10	0.15963E 02	0.17631E 02		0.16536E 03	0.34263E 03		0.29604E 02		

ULTIMATE MOMENT REQUIRED = 0.59733E 03 KIP-FT									
ULTIMATE MOMENT CAPACITY = 0.72482E 03 KIP-FT									
CRACKING MOMENT CAPACITY = 0.53942E 03 KIP-FT									

FIGURE 24. (CONTINUED)

concrete weight and release strength indicated. The initial and total prestress losses shown are computed as described in Section 2.1. The strand pattern and strand wrapping requirements are listed under the heading of Design Results. The number of strands required in each strand row is printed, together with number and wrapping lengths for strands in that row. The wrapping distance is measured inward, from the end of the beam. Neither of the examples in Figures 24 and 25 contain wrapped strands. Deflections are displayed for short term (no creep and shrinkage effects) and long term (all creep and shrinkage effects have occurred) conditions. The long term deflections are computed using the method of Sinno and Furr (2). Positive deflections are upward. The stirrup spacing required at each tenth point of the beam which is output is based on No. 4, grade 60 stirrups. For the optimization option, the final items of design results are cost totals. The total beam cost and cost per foot figures shown are of course based only on concrete, strand and strand wrapping costs and therefore are incomplete.

It may happen that the user inadvertently may seek a design which is impossible to obtain within the imposed restrictions. When the program determines that no feasible design exists, it prints the message

```
*****  
*SORRY, THIS BEAM WILL NOT WORK.*  
*****
```

When this abnormal termination occurs, the user should inspect the input data on the first sheet of output for errors. This is the most frequent cause of abnormal termination. Other possible causes include; deflection

DISTRICT 21 HIGHLAND COUNTY HIGHWAY NO. US 281
 CONTROL NO. 220-02 IPE 670 SUBMITTED BY HLJ
 DESCRIPTION EXAMPLE PROBLEM NO. 2

```

*****
*                                BEAM DIMENSIONS AND PROPERTIES                                *
*****
*.....(DIMENSIONS IN INCHES).....*                                *..SECTION PROPERTIES (WITHOUT SHEAR KEY)..**
* A   B   C   D   E   F   G   H   I   J   K   L   M   N   O   P   Q   R   S   T   U   V   W   X   Y   Z   *
* 35.25 36.00 15.00 27.00 0.75 5.50 0.75 6.00 26.00 5.50 5.00 3.00 3.00 51510. 560.7 13.35 13.61 *
*
* COMPRESSION MAXIMUM MINIMUM STRAND STRAND CONCRETE *..SECTION PROPERTIES (WITH SHEAR KEY)..**
* REINFORCING INITIAL INITIAL STRAND ULTIMATE RELATIVE UNIT *..SECTION PROPERTIES (WITH SHEAR KEY)..**
* AREA CAMBER CAMBER AREA STRENGTH HUMIDITY WEIGHT *..SECTION PROPERTIES (WITH SHEAR KEY)..**
* (IN**2) (IN) (IN) (IN**2) (KSI) (%) (K/FT**3) *..SECTION PROPERTIES (WITH SHEAR KEY)..**
* 1.00 0.750 0.0 0.109 250. 50. 0.150 *..SECTION PROPERTIES (WITH SHEAR KEY)..**
*
*.....STRAND INFORMATION.....*
*
* RCH NUMBER 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 *
* MAXIMUM NO. OF STRANDS 17 4 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 *
* SPACING (ROW 1=1 TO 1) 2.0 4.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 *
*
*.....ALLOWABLE STRESS COEFFICIENTS.....*
*
* RELEASE END 1/10 REPAIR SERVICE END 1/10 REPAIR
* C 0.60 0.60 C 0.40 0.40
* T 7.50 7.50 T 6.00 6.00
*
*.....CREEP AND SHRINKAGE COEFFICIENTS.....*
*
* CREEP1 = 0. CREEP2 = 0. SHRINK1 = 0. SHRINK2 = 0.
*
*.....CONCRETE COST COEFFICIENTS($/YD**3).....*
* 4.0KSI/$150.0 4.5KSI/$150.0 5.0KSI/$150.0 5.5KSI/$150.0 6.0KSI/$200.0 6.5KSI/$230.0 7.0KSI/$230.0 7.5KSI/$230.0
* 8.0KSI/$230.0 8.5KSI/$230.0 9.0KSI/$230.0
*
*.....28 DAY CONCRETE STRENGTH COEFFICIENTS (RELEASE STRENGTH/28 DAY STRENGTH).....*
* 4.0KSI/ 5.4KSI 4.5KSI/ 6.1KSI 5.0KSI/ 6.7KSI 5.5KSI/ 7.4KSI 6.0KSI/ 8.1KSI 6.5KSI/ 8.2KSI 7.0KSI/ 8.2KSI 7.5KSI/ 8.2KSI
* 8.0KSI/ 8.2KSI 8.5KSI/ 8.2KSI 9.0KSI/ 8.2KSI
*
* STRAND COST = $0.25/FT STRAND WRAPPING COST = $0.25/FT MAXIMUM RELEASE STRENGTH ALLOWED = 6.5 KSI
*
*.....BRIDGE PROPERTIES.....*
*
* SPAN LENGTH = 50.0(FT) BRIDGE WIDTH = 44.0(FT) NUMBER TRAFFIC LANES = 4 NUMBER BEAMS = 16.00
*

```

FIGURE 25. OUTPUT FOR EXAMPLE PROBLEM NO. 2

***** CRITICAL DESIGN FACTORS *****									
***** RELEASE STRESSES *****					***** SERVICE LOAD STRESSES *****				
(SYMBOL X DENOTES STRESS AT ALLOWABLE)					(SYMBOL X DENOTES STRESS AT ALLOWABLE)				
SECTION	STRESS TOP (KSI)	STRESS BOTTOM (KSI)		SECTION	STRESS TOP (KSI)	STRESS BOTTOM (KSI)			
0/20	-0.2900E 00	0.1316E 01		0/10	-0.2537E 00	0.1117E 01			
1/20	-0.1220E 00	0.1206E 01		1/10	0.2995E 00	0.5613E 00			
2/20	-0.8542E-01	0.1108E 01		2/10	0.7127E 00	0.1460E 00			
3/20	-0.1844E-03	0.1022E 01		3/10	0.9860E 00	-0.1269E 00			
4/20	0.7268E-01	0.5464E 00		4/10	0.1145E 01	-0.2888E 00			
5/20	0.1362E 00	0.8825E 00		5/10	0.1177E 01	-0.3208E 00			
***** LIST OF DESIGN CONSTRAINTS *****									
(SYMBOL X INDICATES CONSTRAINT CONTROLS FINAL DESIGN)									
MINIMUM CONCRETE STRENGTH			X	ULTIMATE MOMENT		X	MINIMUM INITIAL CAMBER		
MAXIMUM CONCRETE STRENGTH				CRACKING MOMENT			MAXIMUM INITIAL CAMBER		
***** MOMENT AND SHEAR SUMMARY *****									
SECTION	BEAM WT. PLUS SHEAR KEY MOMENTS (KIP-FT)	OTHER D.L. MOMENTS (KIP-FT)		L.L. MOMENTS (KIP-FT)	TOTAL MOMENTS (KIP-FT)		ULTIMATE SHEAR (KIPS)		
0/10	C.C	C.C		0.0	0.0		C.74795E 02		
1/10	0.65676E 02	0.18000E 02		0.52947E 02	0.17662E 03		0.63850E 02		
2/10	0.11676E 03	0.32000E 02		0.15983E 03	0.30859E 03		C.52905E 02		
1/4	0.13683E 03	C.37500E 02		0.18351E 03	0.35783E 03		0.47432E 02		
3/10	0.15324E 03	0.42030E 02		0.20066E 03	0.39591E 03		0.41820E 02		
4/10	0.17514E 03	0.48000E 02		0.22354E 03	0.44667E 03		C.3C178E 02		
5/10	C.1E242E 03	0.50000E 02		0.22441E 03	0.45684E 03		C.1E537E 02		

ULTIMATE MOMENT REQUIRED = 0.78838E 03 KIP-FT									
ULTIMATE MOMENT CAPACITY = 0.79234E 03 KIP-FT									
CRACKING MOMENT CAPACITY = 0.34570E 03 KIP-FT									

FIGURE 25. (CONTINUED)

constraints are too "tight", i.e., upper bound too small or lower bound too large, inadequate number of strand rows provided, or inadequate concrete strength specified (design option).

The final sheet of output (see the third sheet in Figures 24 and 25) provides information on factors controlling the final design. Release and service load stresses are displayed first, with compression stress positive. An "x" beside a stress indicates that it is at the allowable (see Figure 25). Generally, the stresses shown may in some cases slightly exceed the allowables. This is the result of permitting a slight variation in computed prestress loss on successive iterations (see Section 2.4) and of rounding the final number of strands in each row to an integer value. This is demonstrated in Figure 24, sheet 3, where the service load stress at the bottom of the beam at midspan is slight tension where zero tension is permitted. Stress computations are based on the strand pattern, strand wrapping and prestress losses indicated on the second output sheet. For release stresses, beam weight and prestress forces are considered. Service load stresses are based on live load moment, dead load moment (beam weight, shear key, uniform and concentrated loads), and prestress force.

Active design constraints are denoted with an "x" under the List of Design Constraints section of the third output sheet. For example, the final design of the first example problem (Figure 24) is controlled by ultimate moment considerations. As shown at the bottom of the Moment and Shear Summary table, the required ultimate moment capacity is 788 k-ft., while that supplied is 815 k-ft. These two numbers are not identical because of rounding of the number of strands to an integer value.

Moments and shears used in design are listed in the Moment and Shear Summary. Live load moments include lateral distribution and impact factors. Ultimate shears are total live load plus dead load, with load factors defined in Section 2.1.

V. PROGRAM DOCUMENTATION - DBOXDS

The computer program DBOXDS implements the box girder design formulation developed in Chapter III. Described below are the standard input form and its use, interpretation of program output and several example problems.

5.1 PROGRAM INPUT

Figure 26 shows the input form to be used with the program.

5.1.1 Title Cards

The first three input cards are title cards providing a means of job reference. The information preprinted on the form in various columns need not be punched on the data cards - it will be printed out automatically during output. The information on these cards is optional. The first two cards should only be input once per computer run. The third title card is the first card in a data pack when multiple problem runs are made, as explained below.

5.1.2 Load and Options Card

The type of standard AASHTO loading (H-15, H-20, HS-15 or HS-20 is entered in columns 5 - 6 and 8 - 9. The live load distribution factor entered in columns 13 - 16 is the fraction of an axle load to be carried by the beam. This distribution factor is applied to the axle train loading (if used) as well as AASHTO truck and lane loadings. If columns 13 - 16 are left blank, the program automatically computes lateral distribution using Eqs. (24) thru (26), (the AASHTO distribution factor). If a vehicle other than an AASHTO truck is to be used for design, enter a "1" in column 20 and complete the axle train data cards. If both axle train and AASHTO loadings

BOX BEAM DESIGN PROGRAM (DRAPED STRANDS)

Sheet _____ of _____

AXLE TRAIN

Axle Loads (kips)

Axle 1	Axle 2	Axle 3	Axle 4	Axle 5	Axle 6	Axle 7	Axle 8	Axle 9	Axle 10	Axle 11	Axle 12	Axle 13	Axle 14	Axle 15	Axle 16	Axle 17	Axle 18
4	8	12	16	20	24	28	32	36	40	44	48	52	56	60	64	68	72
	10	14	18	22	26	30	34	38	42	46	50	54	58	62	66	70	74
	i = 2	i = 3	i = 4	i = 5	i = 6	i = 7	i = 8	i = 9	i = 10	i = 11	i = 12	i = 13	i = 14	i = 15	i = 16	i = 17	i = 18
	Distance from Axle 1 to Axle i																

Load (kips)
 Distance From Left Support (ft)

4 8 10 14 16 20 22 26 28 32 34 38 40 44 46 50 52 56 58 62

A (in) B (in) C (in) D (in) E (in) F (in) G (in) H (in) M (in) T (in) W (in) X (in) Y (in)

A 7 B 12 C 14 17 D 19 22 E 24 27 F 29 32 G 34 37 H 39 42 M 44 47 T 49 52 W 54 57 X 59 62 Y 64 67

Span Length (ft)	Bridge Width (ft)	Number Of Traffic Lanes	Number Of Beams	Area of Compression Reinforcing (in ²)	Distance to c.g. of Compression Reinforcing (in)	Maximum Initial Camber (in)	Minimum Initial Camber (in)	Distance From Bottom Of Beam To Strand Row 1 (in)	Number Of Web Strands	Distance From Centerline Of Beam To Harping Point (ft)	Enter 1 To Read Misc. Properties Card
4 7	11 14	18 19	23 26	29 32	36 39	43 47	51 55	60 61	66 67	72 74	79

4 5 7 8 10 11 13 14 16 17 19 20 22 23 25 26 28 29 31 32 34 35 37 38 40 41 43 44 46 47 49 50 52 53 55 56 58 59 61 62 64 65 67 68 70 71 73 74 76 77 79 80

FIGURE 26. INPUT FORM FOR DBOXDS

Sheet ____ of ____

MISCELLANEOUS PROPERTIES CARD (Enter This Card Only if Previously Specified)

CONCRETE COST COEFFICIENTS (Include Only for Optimization Option)

98

28 DAY CONCRETE STRENGTHS (Include Only for Optimization)

Release Strength / 28 Day Strength

Release Strength	28 Day Strength
4.0 Ksi /	4.5 Ksi /
7.0 Ksi /	7.5 Ksi /
11 13	24 26

Release Strength	28 Day Strength
5.0 Ksi /	8.0 Ksi /
37 39	

Release Strength	28 Day Strength
5.5 Ksi /	8.0 Ksi /
50 52	

Release Strength	28 Day Strength
6.0 Ksi /	9.0 Ksi /
63 65	

Release Strength	28 Day Strength
6.5 Ksi /	76 78

FIGURE 26. (CONTINUED)

are specified, the larger of the axle train, AASHTO truck and AASHTO lane moments are used at each design point (Figure 2). A uniform dead load carried by a single box girder (i.e., with no lateral distribution of load) is entered in columns 24 - 27. This provision allows the user to include the weight of such things as a wearing surface in the design. Concentrated dead load forces applied to a single girder are indicated by placing a "1" in column 31 and completing the concentrated loads on single beam data cards. Column 48 dictates which program option is to be used. If a "1" is entered, the program determines the minimum cost design, based on cost information input from part 2 of the form. This is the "optimization option". If column 48 is left blank, the program exercises the "design option", in which the strand pattern and release strength are selected which minimize the number of strands used, assuming the beam concrete has the 28-day strength entered in columns 35 - 37.

5.1.3 Axle Train Cards (Use Only if "1" Entered in Column 20 of Load and Options Card)

A moving load pattern of up to 18 axles may be used for design. The first card contains the total load on each axle. To facilitate input, the user should sketch the axle train configuration, labeling either the right-most or left-most axle as axle 1 and numbering the remaining axles in sequential order. The weight of each axle is then placed in the appropriate columns of the first data card. The spacing of axles is input on the second data card, where axle spacing is defined as the distance from axle 1 to the axle under consideration. As an example, an AASHTO HS-20 truck (with rear axles separated 14 feet) whose

light axle was designated as axle 1, would require 8.0 in columns 4 - 6 on the first card and 32.0 in columns 8 - 10 and 12 - 14. The second card would contain 14. in columns 8 - 10 and 28. in columns 12 - 14. The program automatically scales axle train axle loads by the lateral distribution factor, but no impact factor is applied.

5.1.4 Concentrated Loads on Single Beam Cards (Use Only if Column 31 of Load and Option Card Contains "1")

Up to 10 concentrated forces acting on a single beam (no lateral distribution assumed) may be input. The first card contains the magnitude of the load, while the second card contains the distance of each load from the left support. This program provision is intended for small loads only. Service load stress checks assume that maximum moment due to total dead load plus live load occurs at a tenth point. If large concentrated forces not at tenth points are entered, this assumption may be in error, resulting in an overstressed design.

5.1.5 Beam Dimensions Card

The dimensions of the beam cross section which are to be input are shown on the figure at the upper right corner of part 1 of the input form. The fillets (dimensions X and Y) are assumed to slope at 45 degrees. Most any cross sectional shape can be accommodated with the dimensions shown. An ordinary rectangular voided section can be obtained, for example, by inputting dimensions such that $A=B=(2 \cdot W+M)$ and $C=H=E=C=0$ (or left blank).

5.1.6 Bridge and Beam Properties Card

The information on this card is used to compute the lateral distribution factor (if columns 13 - 16 of the load and options card is

left blank) and other quantities used to formulate the constraint set. The span length is entered in columns 4 - 7, bridge width in columns 11 - 14, number of traffic lanes in columns 18 - 19 and number of longitudinal beams in columns 23 - 26. The number of longitudinal beams is input as a decimal number to accommodate unusual conditions (such as a mixture of two or more different box cross sections in the same bridge). Compression steel is sometimes used in box sections to help control long term camber. The area of this steel, which is input in columns 29 - 32, is considered in the computation of section properties, using a transformed steel area of $(n-1)$ for properties with shear key and $2(n-1)$ without. The distance from the top of the beam to c.g. of compression steel is entered in columns 36 - 39. If left blank, the program assumes $T/2$. Maximum and minimum acceptable release cambers are input in columns 43 - 47 and 51 - 55 (upward camber is positive, downward camber negative). These apply to the midspan camber at release produced by prestress and beam weight. A typical application of the lower bound camber would be to insure that a beam did not deflect downward under full dead load (say, shear key plus wearing surface). If an estimate of the final release strength is made, then a modulus of elasticity can be computed, and the midspan downward deflection under shear key and wearing surface weights determined. This value is entered (as a positive number in this case) under minimum initial camber. This will insure that the final design has enough upward initial camber to offset the downward deflection caused by the addition of shear key and wearing surface. If columns 43 - 47 or 51 - 55 are left blank, then the constraint is ignored during design. The distance from the bottom of the beam to the centerline of the first (bottom-most) strand row is input in columns 60 - 61. The number of web strands, or the number of strands that may be draped in

any given row, is input in columns 66 - 67. The distance from the centerline of the beam to the harping point, should there be a need for draped strands, is input in columns 72 - 74. Should column 79 be left blank, the program assumes normal weight concrete (150 lbs/ft^3), 50% relative humidity, 0.153 in.^2 grade 270 strands, and a standard grid spacing of 2.0 inches. The allowable stress coefficients are taken as 0.6 for compression and 7.5 for tension at release and 0.4 and 6.0 under service loads. In order to change any or all of these properties, enter "1" in column 79 and complete the miscellaneous properties card.

5.1.7 Maximum Number of Strands and Top-Most Grid Row Card

The maximum number of strands, the number of rows available for strands, and the top-most grid row (if less than or equal to 26) are determined from this input data. Strand rows are numbered consecutively, taking the bottom-most row as row 1. The maximum number of strands to be allowed in each row should be input in the columns corresponding to that row on the input form. The computation time required to obtain a final design increases rapidly as the number of available strand rows increases. Thus, one should include only those rows which will likely be used. The top-most grid row is the upper-most row to which strands may be draped at the ends of the beam. This is input by entering "TP" in the two columns corresponding to the row number of the top-most grid row. If the top-most row number is greater than 26, enter 1 in column 79 of the bridge and beam properties card and enter the row number in columns 79-80 of the miscellaneous properties card. At present the program is limited to 10 rows of strands.

5.1.8 Miscellaneous Properties Card (Use Only if "1" Entered in Column 79 of Bridge and Beam Properties Card)

If properties other than the standard values listed in Section 5.1.6 are to be used, they must be entered on this card. Only those properties which differ from standard values need to be entered. If the program encounters blanks on the card where a property is to be read, it automatically assumes the standard value. The unit weight of concrete, if different from .150 k/ft³, is entered in columns 4 - 6. Relative humidity is entered in columns 10 - 11, strand area in columns 14 - 16, ultimate strength of strands in columns 19 - 21 and grid spacing in columns 25 - 26. The coefficients used to specify allowable stresses are entered in columns 31 - 55. If the allowable compressive release stress differs from $0.6 f'_{ci}$ or the allowable tensile stress from $7.5 \sqrt{f'_{ci}}$, then the coefficients (those that replace 0.6 and 7.5) should be entered in columns 31 - 32 and 34 - 35 for the end 1/10 of the beam ($j = 6 \text{ \& } 7$ in Figure 17) and in columns 37 - 38 and 40 - 41 for the remainder of the beam. If the allowable compressive long term stress differs from $0.4 f'_c$ or the allowable long term tensile stress from $6.0 \sqrt{f'_c}$, then the new coefficients should be entered in columns 45 - 46 and 48 - 49 for the end 1/10 of the beam and in columns 51 - 52 and 54 - 55 for the remainder of the beam.

The final beam camber at midspan after all prestress losses and creep and shrinkage effects have occurred is computed using the method developed in Reference (2). Cambers are computed and displayed using four different sets of creep and shrinkage coefficients typical of concretes in highway beams produced in four localities in Texas. Should the designer have information on the creep and shrinkage properties of the concrete he expects to be used in a particular design, he may enter the appropriate coefficients in columns 58 - 75. The program will then compute and display the expected midspan camber for these conditions.

The number of the top-most grid row, if greater than 26, must be entered in columns 79 - 80.

5.1.9 Concrete Cost Coefficients Cards (Use Only if "1" Entered in Column 48 of Load & Options Card)

The cost of concrete in dollars per cubic yard can be input for concrete release strengths up to 9.0 ksi. If an estimate of the cost per cubic yard for 4.0 ksi release strength can be made, Figure 7 can be used as a guide to establishing the cost of higher strengths in the absence of actual cost data. Should release strengths beyond some value (say, 7.0 ksi) not be feasible, then the values (7.5, 8.0, 8.5 and 9.0) beyond that point should be left blank.

5.1.10 Strand Cost Card (Use Only if "1" Entered in Column 48 of Load & Option Card)

The cost per foot for strand is entered in columns 14 - 16.

5.1.11 28-Day Concrete Strength Cards (Use Only if "1" Entered in Column 48 of Load & Options Card)

The relationship between release strength and 28-day strength is defined by the data input from these cards. It isn't possible to construct a general relationship between release and 28-day strengths because of the many factors that influence it. There are similarities in the operations of the major producers of highway beams in the state of Texas (12) which permit a reasonable estimate of the relationship.

Fabricators generally use a 24 hour steam curing production sequence. Beams are cast in the afternoon, allowed to gain their initial set (minimum of 3 hours after casting before steam curing is begun (13)) and then steam cured overnight. A total steam curing period of 18 hours at 140° to 150° F.

is typical of most operations. Hanson (14) collected data on concretes made with type III Portland cement and subjected to 15 hours of steam curing at 150° F. commencing 3 hours after casting. Concrete strengths were generally around 4 ksi at 18 hours (release) and 5 ksi at 28 days. For the data reported, the 18 hour strengths averaged 74% of the 28-day strengths. This percentage is probably valid over the usual range of release strengths utilized in THD prestressed designs, which is approximately 6 ksi. If this percentage is applied to release strengths between 4 and 6 ksi, the first straight line segment shown in Figure 20 is produced. Higher release strengths generally demand longer periods of steam curing and result in smaller percentage gains in release strength over 28-day strength. At the extreme limit of 9 ksi release strength, the fabricator would no doubt be forced to keep the beams under special cure for the majority of a 28-day period before release strength was reached. Under these conditions, the ratio of release strength to 28-day strength should be approximately 1. If a linear variation in strength gain over release strength is assumed between 6. and 9. ksi release strengths, the second straight line segment shown in Figure 20 is obtained.

5.1.12 Multiple Problem Runs

The user may process more than one design in a single computer run. The first problem must contain the three title cards described in Section 5.1.1. Each additional problem which is run should have the third title card as the first card in the data set.

5.2 SAMPLE PROBLEMS

Described below are several example problems demonstrating the use of

the standard input form.

5.2.1 Example Problem 3

A multibeam box girder bridge is to carry two lanes of traffic with HS-20 loading and span 80.0 ft. The girder cross section is to be formed using a standard FHWA design for an 80.0 ft. span and an overall beam width of 3.0 ft. Ten boxes will be used to support an overall bridge width of 30.0 ft. and a uniform dead load of 160 lbs./ft. The design will incorporate two strand rows, using 1/2 in. diameter grade 270k strands. A maximum of ten strands will be allowed in the first strand row, which is to be placed 2.5 in. above the bottom of the section. The second row is located the standard 2.0 in. above the first, and may also contain 10 strands. Since the section is 42. in. deep, with a standard grid spacing of 2.0 in., the top-most grid row available for draped strands will be taken as row number 18. Each 5 in. web width will accommodate 2 strands in each row, hence the number of web strands, or the number of strands available for draping, is taken as 4. Five number 4, grade 60 conventional reinforcing bars are to be placed in the deck slab, 2.5 in. down from the surface.

The beam dimensions C,E,G, and H have been specified as 0.0 in. on the input form, which effectively modifies the beam section shown on the input form to accommodate this specific beam cross section. For this design, there is essentially no shear key.

The midspan camber of the beam is assumed to be a non-controlling factor in the design, and is therefore ignored in the computations.

In order to prevent tension cracks in the concrete at the ends of the beam, there will be no tension stress allowed at the ends of the beam at release, however all other allowable stresses will be taken as the standard

values specified in section 5.1.6.

The in-place cost of beam concrete is assumed to be \$150/cu.yd. for release strengths up to 5.5 ksi, \$200 for $f'_{ci} = 6.0$ ksi and \$230 for $f'_{ci} = 6.5$ ksi. Release strengths greater than 6.5 ksi are assumed to be unavailable. The 28 day concrete strength corresponding to each release strength is assumed to be that given in Figure 20.

5.2.2 Example Problem 4

A 48 ft. wide multibeam bridge consisting of 16 box girder units is to be constructed to accommodate 2 lanes of traffic and span 80 ft. The design loads consist of an HS-20 truck and a uniform load of 160 lbs./ft. The box cross section to be used is a 3 ft. wide bituminous surface beam proposed as a standard by the Commonwealth of Pennsylvania Department of Transportation. The vertical positioning of strand rows is such that the first row is placed 1.5 in. above the bottom of the section, with subsequent rows spaced evenly at 2.0 in. A lateral spacing of 2.0 in. center-to-center will permit a maximum of 15 strands per row. Grade A-416, 7/16 in. diameter strands (ultimate strength of 270k and cross sectional area of 0.117 in.²) are to be used. The 5 in. web width and 42 in. depth of the section provide for 4 web strands and a top-most grid row of 18, should there be a need for draping. Compression reinforcing consists of 4 No. 4 grade A615 bars placed 2.5 in. from the top of the section. Downward camber of the beam at release will not be permitted, however the amount of upward camber is not critical. Allowable release and service load stresses in the concrete are those specified in the AASHTO Bridge Specification (4) for inland regions.

Due to the large number of boxes in the bridge and the standard allowable stress coefficients permitted, the design of each section should not be overly critical. Thus, there should not be a need for more than 2 strand

BOX BEAM DESIGN PROGRAM (DRAPED STRANDS)

Sheet _____ of _____

The diagram shows a cross-section of a beam with an octagonal hole. The overall width is labeled A and the overall height is labeled D . The hole is defined by points $G, H, T, X, W, M, Y, C, E, F$ connected by a dashed line. The points are located at the corners and midpoints of the hole's boundary.

AXLE TRAIN

CONCENTRATED LOADS ON SINGLE BEAM

BEAM DIMENSIONS

Span Length (ft)	Bridge Width (ft)	Number Of Traffic Lanes	Number Of Beams	Area of Compression Reinforcing (in ²)	Distance to C.G. of Compression Reinforcing (in)	Maximum Initial Camber (in)	Minimum Initial Camber (in)	Distance From Bottom Of Beam To Strand Row 1 (in.)	Number Of Web Strands	Distance From Centerline Of Beam To Harping Point (ft)	Enter 1 To Read Misc. Properties Card
80 4 7	30 11 14	2 18 19	10 23 26	1 29 32	25 36 39	 43 47	 51 55	25 60 61	4 66 67	5 72 74	1 79

MAXIMUM NUMBER OF STRANDS PERMITTED IN ROW i (Enter "TP" in TOP-MOST GRID ROW, if ≤ 26)

[illegible]

FIGURE 27. COMPLETED INPUT FORM FOR EXAMPLE PROBLEM 3.

BOX BEAM DESIGN PROGRAM (DRAPED STRANDS)

Part 2 of 2

Sheet ___ of ___

MISCELLANEOUS PROPERTIES CARD (Enter This Card Only if Previously Specified)

Unit Weight Concrete (k/Cu Ft)	Relative Humidity (%)	Strand Area (in ²)	Strand Ultimate Strength (ksi)	Grid Spacing (in)	Allowable Stress Coefficients (ksi)				Creep & Shrinkage Coefficients (μ-in. & Days)				Top-Most Grid Row (if > 26)				
					Release (Long Term)												
					C	T	C	T	C	T	C	T					
4 6	10 11	14 16	19 21	25 26	31 32	34 35	37 38	40 41	45 46	48 49	51 52	54 55		58 60	62 65	68 70	73 75
					End 1/10				End 1/10								

CONCRETE COST COEFFICIENTS (Include Only for Optimization Option)

Release Strength (Ksi) / \$ Cost YD									
4.0 Ksi / \$	1 5 0	4.5 Ksi / \$	1 5 0	5.0 Ksi / \$	1 5 0	5.5 Ksi / \$	1 5 0	6.0 Ksi / \$	2 0 0
7.0 Ksi / \$	2 3 0	7.5 Ksi / \$	2 3 0	8.0 Ksi / \$	2 3 0	8.5 Ksi / \$	2 3 0	9.0 Ksi / \$	2 3 0
	11 14		24 27		37 40		50 53		63 66
									76 79

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STRAND COST \$ / Linear Foot (Optimization Only)

28 DAY CONCRETE STRENGTHS (Include Only for Optimization)

Release Strength / 28 Day Strength									
4.0 Ksi /	5 4	4.5 Ksi /	6 1	5.0 Ksi /	6 7	5.5 Ksi /	7 4	6.0 Ksi /	8 1
7.0 Ksi /	8 2	7.5 Ksi /	8 2	8.0 Ksi /	8 2	8.5 Ksi /	8 2	9.0 Ksi /	8 2
	11 13		24 26		37 39		50 52		63 65
									76 78

FIGURE 27. (CONTINUED)

rows and probably no reason to use draped strands.

The design option is to be used for this problem with the completed input form shown in Figure 28. Program output for this problem is shown in Figure 30.

5.3 INTERPRETATION OF PROGRAM INPUT

The program first reads input data and performs basic checks for input errors. At present, the program checks include; design option specified but no 28 day concrete strength given, proper AASHTO loading designation and omitted span length and omitted number of longitudinal beams (if no live load distribution factor is input), top-most grid row not specified, and an unrecognizable AASHTO truck loading. The input data is printed out before design calculations are begun so that the user can locate input errors which might cause the program to terminate abnormally before producing any other input. The output format is essentially the same for design and optimization options (see first sheet, Figures 29 and 30).

The second sheet of output summarizes the design results, as seen in Figures 29 and 30. The strength and modulus of the concrete, and the prestress losses in the strands are listed under the heading of Design Properties. The first items listed are the release and 28-day concrete strengths. For the design option, the 28-day strength is that specified on input, while the release strength is computed and 28-day strength is obtained from 28-day vs. release strength input data. The modulus of elasticity listed is computed using the ACI equation (10) and the unit concrete weight and release strength indicated. The initial and total prestress losses shown are computed as described in Section 2.1. The strand pattern, deflections, and stirrup spacing requirements are listed under the heading of Design Results. Under

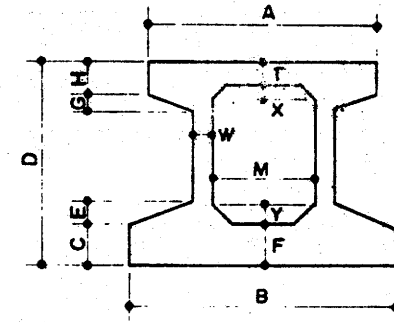
TEXAS HIGHWAY DEPARTMENT
BRIDGE DIVISION

Part 1 of 2

BOX BEAM DESIGN PROGRAM
(DRAPED STRANDS)

Sheet ____ of ____

DISTRICT 02 TARRANT COUNTY HIGHWAY NO. SH-121
CONTROL NO. 31-152 IPE 108 SUBMITTED BY HLJ
DESCRIPTION EXAMPLE PROBLEM NO. 4



Enter 1 For Axle Train Enter 1 if Concentrated Forces Applied To Single Beam Enter 1 For Optimization Option

A.A.S.H.T.O. L.L. HS-20
Live Load Distribution Factor 13 16
Uniform Load on Single Beam (k/ft) 0.16
28 Day Strength (ksi) 5
Omit for Optimization Option 48

AXLE TRAIN

Axle Loads (kips)
Axle 1 4 6
Axle 2 8 10
Axle 3 12 14
Axle 4 16 18
Axle 5 20 22
Axle 6 24 26
Axle 7 28 30
Axle 8 32 34
Axle 9 36 38
Axle 10 40 42
Axle 11 44 46
Axle 12 48 50
Axle 13 52 54
Axle 14 56 58
Axle 15 60 62
Axle 16 64 66
Axle 17 68 70
Axle 18 72 74

i = 2 i = 3 i = 4 i = 5 i = 6 i = 7 i = 8 i = 9 i = 10 i = 11 i = 12 i = 13 i = 14 i = 15 i = 16 i = 17 i = 18

Distance From Axle 1 to Axle i

CONCENTRATED LOADS ON SINGLE BEAM

Load (kips)
Distance From Left Support (ft)
4 8 10 14 16 20 22 26 28 32 34 38 40 44 46 50 52 56 58 62

BEAM DIMENSIONS

A (in) 36 7
B (in) 36 12
C (in) 36 13
D (in) 42 19
E (in) 0.13 24
F (in) 5 27
G (in) 0.13 29
H (in) 5.88 32
M (in) 26 34
T (in) 5.5 37
W (in) 5 39
X (in) 3 42
Y (in) 3 44

Span Length (ft) 80 7
Bridge Width (ft) 48 11
Number Of Traffic Lanes 2 18
Number Of Beams 16 23
Area of Compression Reinforcing (in²) 0.80 29
Distance to c.g. of Compression Reinforcing (in) 25 36
Maximum Initial Camber (in) 4 43
Minimum Initial Camber (in) 0.0 51
Distance From Bottom Of Beam To Strand Row 1 (in) 15 60
Number Of Web Strands 4 66
Distance From Centerline Of Beam To Harping Point (ft) 133 72
Enter 1 To Read Misc. Properties Card 1 79

MAXIMUM NUMBER OF STRANDS PERMITTED IN ROW i (Enter "TP" in TOP-MOST GRID ROW, If ≤ 26)

15 15 10 11 13 14 16 17 19 20 22 23 25 26 28 29 31 32 34 35 37 38 40 41 43 44 46 47 49 50 52 53 55 56 58 59 61 62 64 65 67 68 70 71 73 74 76 77 79 80
i = 1 i = 2 i = 3 i = 4 i = 5 i = 6 i = 7 i = 8 i = 9 i = 10 i = 11 i = 12 i = 13 i = 14 i = 15 i = 16 i = 17 i = 18 i = 19 i = 20 i = 21 i = 22 i = 23 i = 24 i = 25 i = 26

FIGURE 28. COMPLETED INPUT FORM FOR EXAMPLE PROBLEM 4.

BOX BEAM DESIGN PROGRAM (DRAPED STRANDS)

Sheet ___ of ___

Part 2 of 2

MISCELLANEOUS PROPERTIES CARD (Enter This Card Only if Previously Specified)

Unit Weight Concrete (k/Cu Ft)	Relative Humidity (%)	Strand Area (in ²)	Strand Ultimate Strength (ksi)	Grid Spacing (in)	Allowable Stress Coefficients (ksi)				Creep & Shrinkage Coefficients (μ - in. & Days)				Top-Most Grid Row (if > 26)				
					Release		(Long Term)		CREEP 1		CREEP 2			SHRK 1		SHRK 2	
					C	T	C	T	C	T							
4 6	10 11	14 16	19 21	25 26	31 32	34 35	37 38	40 41	45 46	48 49	51 52	54 55	58 60	63 65	68 70	73 75	79 80
					End 1/10		Remainder		End 1/10		Remainder						

CONCRETE COST COEFFICIENTS (Include Only for Optimization Option)

Release Strength (Ksi) / Cost YD

4.0Ksi / \$
7.0Ksi / \$

4.5Ksi / \$
7.5Ksi / \$

5.0Ksi / \$
8.0Ksi / \$

5.5Ksi / \$
8.5Ksi / \$

6.0Ksi / \$
9.0Ksi / \$

6.5Ksi / \$

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STRAND COST \$

--	--	--

 / Lineal Foot (Optimization Only)

28 DAY CONCRETE STRENGTHS (Include Only for Optimization)

Release Strength / 28 Day Strength

4.0Ksi /
7.0Ksi /

4.5Ksi /
7.5Ksi /

5.0Ksi /
8.0Ksi /

5.5Ksi /
8.0Ksi /

6.0Ksi /
9.0Ksi /

6.5Ksi /

FIGURE 28. (CONTINUED)

DISTRICT C2 TARRANT COUNTY HIGHWAY NO. SH-121
 CONTROL NO. 31-152 IPE 108 SUBMITTED BY HLJ
 DESCRIPTION EXAMPLE PRECEM NO. 3

```

*****
*                                BEAM DIMENSIONS AND PROPERTIES                                *
*****
*..... DIMENSIONS IN INCHES .....*
*
*      A      B      C      D      E      F      G      H      M      T      W      X      Y
*    36.00  36.00   0.0  42.00   0.0   5.50   0.0   0.0  26.00   5.50   5.00   3.00   3.00
*
*.....(WITHOUT SHEAR KEY)..... SECTION PROPERTIES .....(WITH SHEAR KEY).....*
*
*      I(IN**4)      A(IN**2)      YT(IN)      YE(IN)      I(IN**4)      A(IN**2)      YT(IN)      YB(IN)
*    164644.        715.4        20.72       21.28       162985.        710.4        21.00       21.00
*
*  DISTANCE TO CG OF COMPRESSION REINFORCING (IN) 2.50
*  COMPRESSION REINFORCING AREA (IN**2) 1.00
*  MAXIMUM INITIAL CAMBER (IN)
*  MINIMUM INITIAL CAMBER (IN)
*  STRAND AREA (IN**2) 0.153
*  STRAND ULTIMATE STRENGTH (KSI) 270.
*  RELATIVE HUMIDITY (%) 50.
*  CONCRETE UNIT WEIGHT (K/FT**3) 0.150
*  DISTANCE FROM CENTERLINE OF BEAM TO HARPING POINT (FT) 5.00
*
*..... STRAND INFORMATION .....*
*
*  DISTANCE FROM BOTTOM OF BEAM TO STRAND ROW 1 = 2.50 IN.
*  GRID SPACING = 2.00 IN.
*  NUMBER OF WEB STRANDS = 4
*  TOP = NCST GRID ROW = 18
*
*  ROW NUMBER      1      2      3      4      5      6      7      8      9     10     11     12     13     14     15     16     17     18     19     20     21     22     23     24     25     26
*  MAX. NO. OF STRANDS 10    10    0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0      0
*
*..... ALLOWABLE STRESS COEFFICIENTS ..... CREEP AND SHRINKAGE COEFFICIENTS .....*
*
*      RELEASE          SERVICE          CREEP1 = 0.      SHRINK1 = 0.
*      END 1/10 REMAINDER  END 1/10 REMAINDER  CREEP2 = 0.      SHRINK2 = 0.
*  COMPRESSION      0.60      0.60      0.40      0.40
*  TENSION          0.0      7.50      6.00      6.00
*
*.....CONCRETE COST COEFFICIENTS($/YD**3).....*
*  4.0KSI/$150.0  4.5KSI/$150.0  5.0KSI/$150.0  5.5KSI/$150.0  6.0KSI/$200.0  6.5KSI/$230.0  7.0KSI/$230.0  7.5KSI/$230.0
*  8.0KSI/$230.0  8.5KSI/$230.0  9.0KSI/$230.0
*
*.....28 DAY CONCRETE STRENGTH COEFFICIENTS (RELEASE STRENGTH/28 DAY STRENGTH).....*
*  4.0KSI/ 5.4KSI  4.5KSI/ 6.1KSI  5.0KSI/ 6.7KSI  5.5KSI/ 7.4KSI  6.0KSI/ 8.1KSI  6.5KSI/ 8.2KSI  7.0KSI/ 8.2KSI  7.5KSI/ 8.2KSI
*  8.0KSI/ 8.2KSI  8.5KSI/ 8.2KSI  9.0KSI/ 8.2KSI
*
*  STRAND COST = 10.25/FT
*  MAXIMUM RELEASE STRENGTH ALLOWED = 9.0 KSI
*
*..... BRIDGE PROPERTIES .....*
*
*  SPAN LENGTH = 50.0(FT)  BRIDGE WIDTH = 70.0(FT)  NUMBER TRAFFIC LANES = 2  NUMBER BEAMS = 10.0
  
```

FIGURE 29. OUTPUT FOR EXAMPLE NO. 3

THE COMMAND IS TO OPTIMIZE

```

*****
*                               DESIGN PROPERTIES                               *
*****
*   RELEASE STRENGTH = 5.50 (KSI)   CONCRETE MODULUS(RELEASE) = 4496.1 (KSI)   INITIAL PRESTRESS LOSS = 5.77 PERCENT *
*   28-DAY STRENGTH = 7.40 (KSI)                                     TOTAL PRESTRESS LOSS = 17.31 PERCENT *
*****

```

```

*****
*                               DESIGN RESULTS                               *
*****
*                               STRAND LAYOUT                               *
*****
*   LOCATION                                DISTANCE FROM                DISTANCE FROM
*   END OF BEAM                             BOTTOM OF BEAM TO C.G.         BOTTOM OF BEAM TO C.G.
*   CENTERLINE                             OF DRAPED STRANDS           OF STRAIGHT STRANDS
*
*   15.50                                   2.50
*   3.50                                   2.50
*
*   TOTAL NUMBER OF STRANDS                = 14
*   NUMBER OF DRAPED STRANDS              = 8
*   NUMBER OF STRANDS IN ROW 2            = 4
*   NUMBER OF STRANDS IN ROW 1            = 10

```

AT THE END OF THE BEAM, BEGINNING WITH ROW 1, RAISE 4 STRANDS IN EACH ROW 12.0 INCHES ABOVE STRAIGHT STRANDS IN THAT ROW

```

*****
*                               COMPUTED DEFLECTION                               *
*****
*   SHORT TERM DEFLECTION (IN.)                                LONG TERM DEFLECTION (IN.)
*
*   BEAM WEIGHT (RELEASE) = 0.05                                BASED ON CALLAS CONCRETE PROPERTIES = -2.27
*   FIBT + SHEAR KEY (E=5000.) = -0.06                          BASED ON ODESSA CONCRETE PROPERTIES = -2.80
*   EMWT + KEY + DEAD LOADS (E=5000.) = -0.24                   BASED ON SAN ANTONIO CONCRETE PROPERTIES = -1.74
*                                                                BASED ON LUFKIN CONCRETE PROPERTIES = -2.51

```

```

*****
*   STIRRUP SPACING - AASHTO 1973 - STIRRUP AREA = 0.11 IN2 *****
*
*   SECTION * C/10 * 1/10 * 5/40 * 2/10 * 1/4 * 3/10 * 4/10 * 5/10 * FIBT *
*   SPACING (IN.) * 13.20 * 13.20 * 13.20 * 13.20 * 13.20 * 13.20 * 13.20 * 13.20 * 13.20 *

```

```

*****
*   CCST AND MATERIAL REQUIREMENTS OF BEAM *****
*
*   ITEM          AMOUNT          CCST          PERCENTAGE OF TOTAL CCST          TOTAL CCST OF BEAM = 2488.14
*   CONCRETE*      14.72 YD**3      $ 2206.14          89.75 %
*   STRANDS        1120.00 FT        $ 280.00          11.25 %
*   CCST PER FOOT          $ 31.10
*
*   *CCST NOT INCLUDE END SECTION

```

FIGURE 29. (CONTINUED)

CRITICAL DESIGN FACTORS									
RELEASE STRESSES.....					SERVICE LOAD STRESSES.....				
(SYMBOL X DENOTES STRESS AT ALLOWABLE)					(SYMBOL X DENOTES STRESS AT ALLOWABLE)				
SECTION	STRESS TOP (KSI)		STRESS BOTTOM (KSI)		SECTION	STRESS TOP (KSI)		STRESS BOTTOM (KSI)	
0/10	-0.1084E-01	X	0.1092E 01		0/10	-0.9514E-02		0.5580E 00	
1/10	0.2377E 00		0.6366E 00		1/10	0.5220E 00		0.4197E 00	
5/40	0.2886E 00		0.7843E 00		5/40	0.6330E 00		0.3072E 00	
2/10	0.4144E 00		0.6552E 00		2/10	0.9140E 00		0.2277E-01	
1/4	0.4757E 00		0.5923E 00		1/4	0.1058E 01		-0.1227E 00	
3/10	0.5191E 00		0.5478E 00		3/10	0.1167E 01		-0.2328E 00	
4/10	0.5518E 00		0.5141E 00		4/10	0.1290E 01		-0.3570E 00	
5/10	0.5596E 00		0.5061E 00		5/10	0.1320E 01		-0.3872E 00	
MDPT	0.5456E 00		0.5206E 00		MDPT	0.1302E 01		-0.3651E 00	
.....LIST OF DESIGN CONSTRAINTS.....									
(SYMBOL X INDICATES CONSTRAINT CONTROLS FINAL DESIGN)									
MINIMUM CONCRETE STRENGTH		ULTIMATE MOMENT		MINIMUM INITIAL CAPAC		MAXIMUM INITIAL CAPAC			
BASED ON CONCRETE STRENGTH		CRACKING MOMENT							
.....MOMENT AND SHEAR SUMMARY.....									
DISTANCE FROM END OF BEAM (FT)	EM. WT. + SHEAR KEYS (KIP-FT)	OTHER C.L. MOMENTS (KIP-FT)	L.L. MOMENTS (KIP-FT)	TOTAL MOMENTS (KIP-FT)	ULTIMATE SHEAR (KIPS)				
0.0	0.0	C.C	0.0	0.0	0.52849E 02				
8.00	0.21313E 03	0.46080E 02	0.12118E 03	0.39039E 03	0.77974E 02				
10.00	0.25901E 03	0.56000E 02	0.15874E 03	0.47376E 03	0.74255E 02				
16.00	0.37850E 03	0.61520E 02	0.22687E 03	0.68969E 03	0.62555E 02				
20.00	0.44402E 03	0.96000E 02	0.26515E 03	0.80518E 03	0.55241E 02				
24.00	0.45731E 03	0.10752E 03	0.29307E 03	0.89789E 03	0.47523E 02				
32.00	0.56835E 03	0.12288E 03	0.33028E 03	0.10215E 04	0.32088E 02				
40.00	0.59203E 03	0.12800E 03	0.33726E 03	0.10573E 04	0.16657E 02				
MDPT	0.58278E 03	0.12600E 03	0.33682E 03	0.10456E 04	0.26200E 02				
ULTIMATE MOMENT REQUIRED = 0.16668E 04 KIP-FT									
ULTIMATE MOMENT CAPACITY = 0.17645E 04 KIP-FT									
CRACKING MOMENT CAPACITY = 0.63233E 03 KIP-FT									

FIGURE 29. (CONTINUED)

DISTRICT 02 TARRANT COUNTY HIGHWAY NO. SH-121
 CONTROL NO. 21-152 IPE 108 SUBMITTED BY HLJ
 DESCRIPTION EXAMPLE PROBLEM NO. 4

```

*****
***** EEA DIMENSIONS AND PROPERTIES *****
*****
***** DIMENSIONS IN INCHES *****
*****
      A      B      C      C      E      F      G      H      M      T      W      X      Y
      36.00  36.00  36.13  42.00  0.13  5.00  0.13  5.88  26.00  5.50  5.00  3.00  3.00
***** (W/CUT SHEAR KEY) ***** SECTION PROPERTIES ***** (WITH SHEAR KEY) *****
*****
      I(IN**4)      A(IN**2)      YT(IN)      YB(IN)      I(IN**4)      A(IN**2)      YT(IN)      YB(IN)
      160739.      700.4      20.48      21.52      159443.      696.4      20.71      21.29
*****
      DISTANCE TO CG CF      COMPRESSION      MAXIMUM      MINIMUM      STRAND      STRAND      RELATIVE      CONCRETE      DISTANCE FROM
      REINFORCING      REINFORCING      INITIAL      INITIAL      AREA      ULTIMATE      HUMIDITY      UNIT      CENTERLINE CF
      (IN)      (IN**2)      CAMBER      CAMBER      (IN**2)      STRENGTH      (%)      WEIGHT      BEAM TO
      2.50      0.00      (IN)      (IN)      (IN**2)      (KSI)      (K/FT**3)      HAFFING POINT
      2.50      0.00      (IN)      (IN)      (IN**2)      (KSI)      (K/FT**3)      (FT)
*****
***** STRAND INFORMATION *****
*****
      DISTANCE FROM ECTEN      GRID SPACING = 2.00 IN.      NUMBER OF WEB STRANDS = 4
      OF BEAM TO STRAND ROW 1 = 1.50 IN.
*****
      ROW NUMBER      1  2  3  4  5  6  7  8  9  10  11  12  13  14  15  16  17  18  19  20  21  22  23  24  25  26
      MAX. NO. OF STRANDS  15  15  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0  0
*****
***** ALLOWABLE STRESS COEFFICIENTS ***** CREEP AND SHRINKAGE COEFFICIENTS *****
*****
      RELEASE      SERVICE
      END 1/10 REMAINDER      END 1/10 REMAINDER      CREEP1 = 0.      SHRK1 = 0.
      COMPRESSION      0.60      0.60      0.40      0.40      CREEP2 = 0.      SHRK2 = 0.
      TENSION      7.50      7.50      6.00      6.00
*****
***** BRIDGE PROPERTIES *****
*****
      SPAN LENGTH = 37.0 (FT)      BRIDGE WIDTH = 42.0 (FT)      NUMBER TRAFFIC LANES = 2      NUMBER BEAMS = 16.00
*****
  
```

FIGURE 30. OUTPUT FOR EXAMPLE NO. 4

***** CRITICAL DESIGN FACTORS *****									
***** RELEASE STRESSES *****				***** SERVICE LOAD STRESSES *****					
(SYMBOL X DENOTES STRESS AT ALLOWABLE)				(SYMBOL X DENOTES STRESS AT ALLOWABLE)					
SECTION	STRESS TOP (KSI)	STRESS BOTTOM (KSI)		SECTION	STRESS TOP (KSI)	STRESS BOTTOM (KSI)			
0/10	-0.3504E 00	0.1285E 01		C/10	-0.3108E 00	0.1140E 01			
1/10	-0.2564E 01	0.9475E 00		1/10	0.2150E 00	0.5515E 00			
5/40	0.3542E 01	0.6753E 00		5/40	0.3275E 00	0.4746E 00			
2/10	0.2199E 00	0.6857E 00		2/10	0.6195E 00	0.1703E 00			
1/4	0.3175E 00	0.5627E 00		1/4	0.7763E 00	0.6754E 02			
3/10	0.3581E 00	0.4564E 00		3/10	0.9028E 00	-0.1251E 00			
4/10	0.5055E 00	0.3861E 00		4/10	0.1071E 01	-0.3005E 00			
5/10	0.5406E 00	0.3466E 00		5/10	0.1122E 01	-0.3535E 00			
MDPT	0.4421E 00	0.4521E 00		MDPT	0.9711E 00	-0.1963E 00			
***** LIST OF DESIGN CONSTRAINTS *****									
(SYMBOL X INDICATES CONSTRAINT CONTROLS FINAL DESIGN)									
MINIMUM CONCRETE STRENGTH		X	ULTIMATE MOMENT		MINIMUM INITIAL CAMBER				
MAXIMUM CONCRETE STRENGTH			CRACKING MOMENT		MAXIMUM INITIAL CAMBER				
***** MOMENT AND SHEAR SUMMARY *****									
DISTANCE FROM END OF BEAM (FT)	PM. MT. + SHEAR KEY MOMENTS (KIP-FT)		OTHER D.L. MOMENTS (KIP-FT)		L.L. MOMENTS (KIP-FT)		TOTAL MOMENTS (KIP-FT)		ULTIMATE SHEAR (KIPS)
0.0	0.0		0.0		0.0		0.0		0.77837E 02
8.00	0.20693E 03		0.46000E 02		0.85692E 02		0.34070E 03		0.64683E 02
10.00	0.25391E 03		0.56000E 02		0.10370E 03		0.41361E 03		0.61354E 02
16.00	0.37143E 03		0.81920E 02		0.14950E 03		0.60286E 03		0.51437E 02
20.00	0.43527E 03		0.56000E 02		0.17321E 03		0.70448E 03		0.44678E 02
24.00	0.46751E 03		0.10752E 03		0.15144E 03		0.78646E 03		0.37518E 02
32.00	0.55715E 03		0.12298E 03		0.21575E 03		0.89578E 03		0.24358E 02
40.00	0.56036E 03		0.12800E 03		0.22031E 03		0.92867E 03		0.10878E 02
MDPT	0.51620E 03		0.11385E 02		0.20066E 03		0.83073E 03		0.33355E 02

ULTIMATE MOMENT REQUIRED =				0.13582E 04 KIP-FT					
ULTIMATE MOMENT CAPACITY =				0.14889E 04 KIP-FT					
CRACKING MOMENT CAPACITY =				0.45858E 03 KIP-FT					

FIGURE 30. (CONTINUED)

Strand Layout, the total number of strands, number of draped strands, and the number of strands in each row are printed, together with the distance from the bottom of the beam section to the centroid of the straight and the draped strands, at the ends of the beam and at the centerline. Using the bottom-most strand row as row number 1, the number of the first row containing draped strands, the number of strands to be draped, and the distance the strands are raised at the end of the beam are also shown. Deflections are displayed for short term (no creep and shrinkage effects) and long term (all creep and shrinkage effects have occurred) conditions. The long term deflections are computed using the method of Sinno and Furr (2). Positive deflections are upward. The stirrup spacing required at each tenth point, $5L/40$, and the holddown location of the beam is based on No. 4, grade 60 stirrups. For the optimization option, the final items of design results are cost totals. The total beam cost and cost per foot figures are based only on concrete and strand cost and therefore are incomplete.

It may happen that the user inadvertently may seek a design which is impossible to obtain within the imposed restrictions. When the program determines that no feasible design exists, it prints the message

```
*****  
*SORRY, THIS BEAM WILL NOT WORK*  
*****
```

When this abnormal termination occurs, the user should inspect the input data on the first sheet of output for errors. This is the most frequent cause of abnormal termination. Other possible causes include; deflection constraints are too "tight", i.e., upper bound too small or lower bound too

large, inadequate number of strand rows provided, or inadequate concrete strength specified (design option).

The final sheet of output (see the third sheet in Figures 29 and 30) provides information on factors controlling the final design. Release and service load stresses are displayed first, with compression stress positive. An "x" beside a stress indicates that it is at the allowable. In the third example problem (see Figure 29), although there are no stresses at the allowable, the stress in the top of the beam at the end is very nearly 0.0, and due to the fact that no tension was allowed at this point, it is obvious that this stress was a critical factor in the design. Generally, the stresses shown may in some cases slightly exceed the allowables. This is the result of permitting a 3 percent variation in computed prestress loss on successive iterations (see Section 2.4). Stress computations are based on the strand pattern and prestress losses indicated on the second output sheet. For release stresses, beam weight and prestress forces are considered. Service load stresses are based on live load moment, dead load moment (beam weight, shear key, uniform and concentrated loads), and prestress force.

Active design constraints are denoted with an "x" under the List of Design Constraints section of the third output sheet. For example, the final design of the fourth example problem (Figure 30) is controlled by minimum concrete strength considerations, indicating that the design was not critical.

Moments and shears used in design are listed in the Moment and Shear Summary. Live load moments include lateral distribution and impact factors. Ultimate shears are total live load plus dead load, with load factors defined in Section 2.1.

VI. ANALYSIS OF MULTI-BEAM BRIDGES

When a load is placed on one of the beams of a multibeam bridge, the loaded beam deflects and due to the presence of shear keys, adjoining beams deflect with it. This action is transmitted to other beams in a similar way, deforming the entire cross-section of the structure and thus distributing the applied load to all beams in varying amounts. At each longitudinal joint between beams there are forces in three directions and one moment which tend to keep the beams together, as shown in Figure 31. These joint forces are; vertical force v_i , longitudinal force a_i , transverse force h_i and transverse moment m_i . These forces, of course, act in an opposite sense on the beams adjacent to the joint. The joint forces on each edge of a beam and the applied loads produce the forces V_i , A_i , H_i and moments M_{yi} , M_{zi} , M_{ti} shown in Figure 31 on the beam. The four components of joint forces and six components of beam forces vary with position, x , along the beam. They have units of force or moment per unit length and are positive in the direction shown.

The method of analysis employed uses Fourier series to represent the loads applied to the structure and forces and deformations produced by them. The loading may consist of a number of point or patch loads acting vertically anywhere on the structure. These are approximated by Fourier series representations and the response of the structure obtained for each harmonic in the series. The total response is obtained from the superposition of harmonics.

The details of the method are treated adequately in References (9)

and (16). The purpose of this Chapter is to familiarize the reader with the use of a computer program AMBB developed by Ghose (9) implementing the method. Input to the program in its original form is explained in the next section. The program has been modified by the authors to compute lateral distribution factors for axle train and standard AASHTO loadings. The use of the program in this mode is described in Section 6.2.

6.1 ROUTINE ANALYSES USING PROGRAM AMBB

In its original form, the computer program computes displacement and the joint and beam forces (Figure 31) at points along the span specified by the user. The data cards necessary to utilize this portion of the program are described below.

6.1.1 Program Input

1. Title Card (15A4) - Columns 1-60. Title to be printed with output.
2. Control Card (3F10.0,4I5) - Use consistent length and force units.

Col. 1-10 - Span of bridge

11-20 - Young's modulus

21-30 - Poisson's ratio

31-35 - Number of beams (max. 20)

36-40 - Number of different beam types (max 10)

41-45 - Number of different joint types (max 10)

46-50 - Number of harmonic terms of the Fourier series representation. 10-20 for uniformly distributed loads.
20-50 for concentrated forces.

3. Beam Cards - Two cards for each different beam type.

First Card (4F10.0)

Col. 1-10 - Moment of inertia about z (vertical) axis.

Col. 11-20 - Moment of inertia about y (transverse) axis.

Col. 21-30 - Area of cross section.

Col. 31-40 - St. Venant torsion constant of the beam.

Second Card (4F10.0)

Col. 1-10 - Vertical distance between center of gravity of the beam section and left shear hinge, measured positive downwards.

Col. 11-20 - Horizontal distance between center of gravity of beam section and left shear hinge, measured positive to the right.

Col. 21-30 - Vertical distance between center of gravity of the beam section and right shear hinge, measured positive downwards.

Col. 31-40 - Horizontal distance between center of gravity of beam section and right shear hinge, measured positive to the right.

4. Beam-type Identification Card (20I2) For each beam, starting from the left, enter the beam type number in the above format.

5. Hinge-flexibility Card (4F10.0) One card for each different hinge type.

Col. 1-10 - Flexibility of hinge in the longitudinal (x) direction.

Col. 11-20 - Flexibility in the transverse (y) direction .

Col. 21-30 - Flexibility in the vertical (z) direction .

Col. 31-40 - Flexibility of hinge for transverse rotation.

For a rigid connection, enter zero flexibility. For zero restraint, enter a large number (say 10^8).

6. Hinge-type Identification Card (20I2) For each hinge, starting from the left, enter the hinge type number in the above format.
7. Load Control Card (2I5)
 - Col. 1-5 - Number of load cases to be analyzed (no maximum).
 - Col. 6-10 - Number of load cards (maximum 40).
8. Load Cards (2I5, 4F10.0) One card for each beam load.
 - Col. 1-5 - Load case number.
 - Col. 6-10 - Number of loaded beam.
 - Col. 11-20 - Magnitude of load.
 - Col. 21-30 - x-coordinate of the load centroid.
 - Col. 31-50 - Eccentricity of the centroid of the load from the centroidal axis of the beam, measured positive in positive direction of y-axis (i.e. positive to the right).

A uniformly distributed load in the y direction should be reduced to equivalent line loads acting on the separate beams.
9. Results Card I (2I5, 9F5.0) - One card per load case.
 - Col. 1-5 - Load case number.
 - Col. 6-10 - Number of output positions along span (max 9).
 - Col. 11-55 - X-coordinates of positions, fields of 5 columns each.
10. Results Card II (I5, 8F5.0) One card for each beam type.
 - Col. 1-5 - Number of output positions on the cross section for computation of longitudinal fiber stresses (max 4).
 - Col. 6-45 - One pair of y and z coordinates respectively, for each output position, in fields of 5 columns for each coordinate.

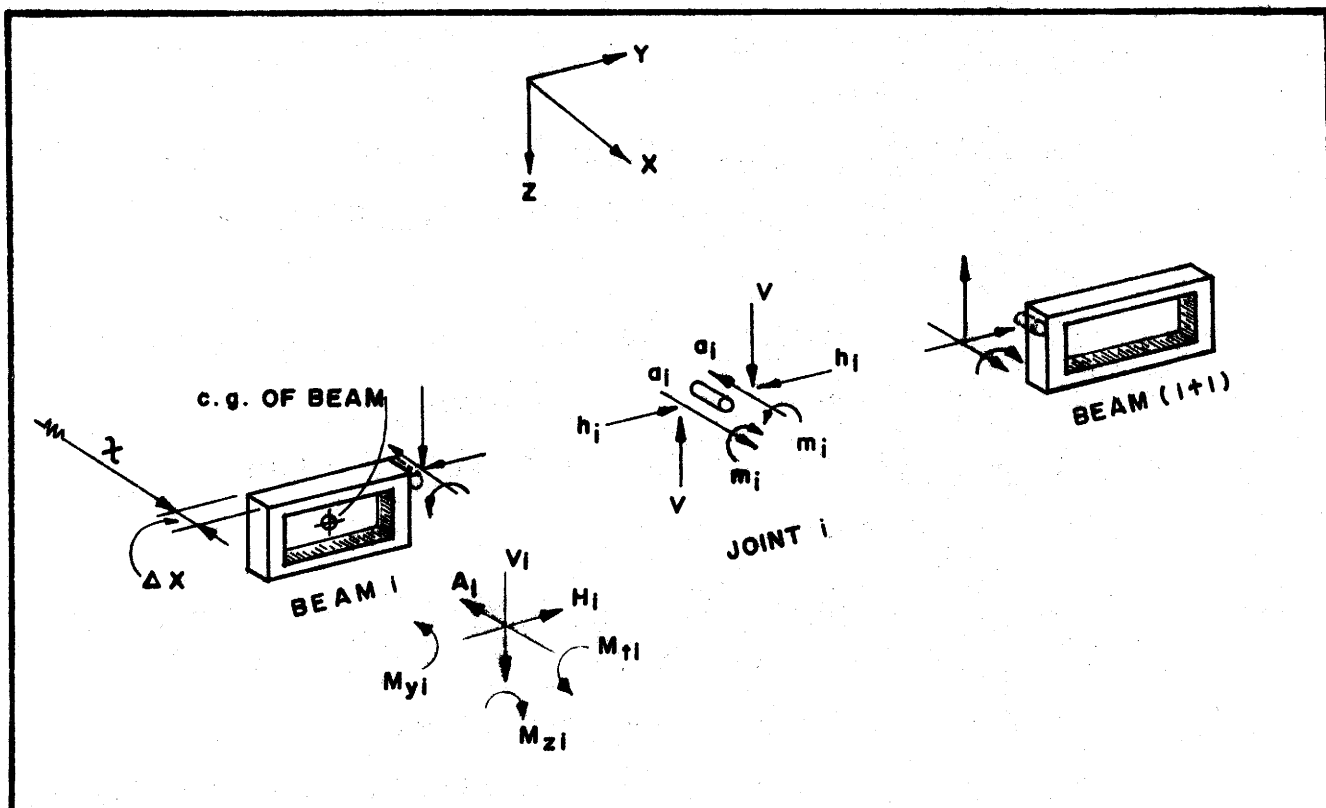


FIGURE 31. JOINT AND BEAM FORCES IN MULTI-BEAM BRIDGE

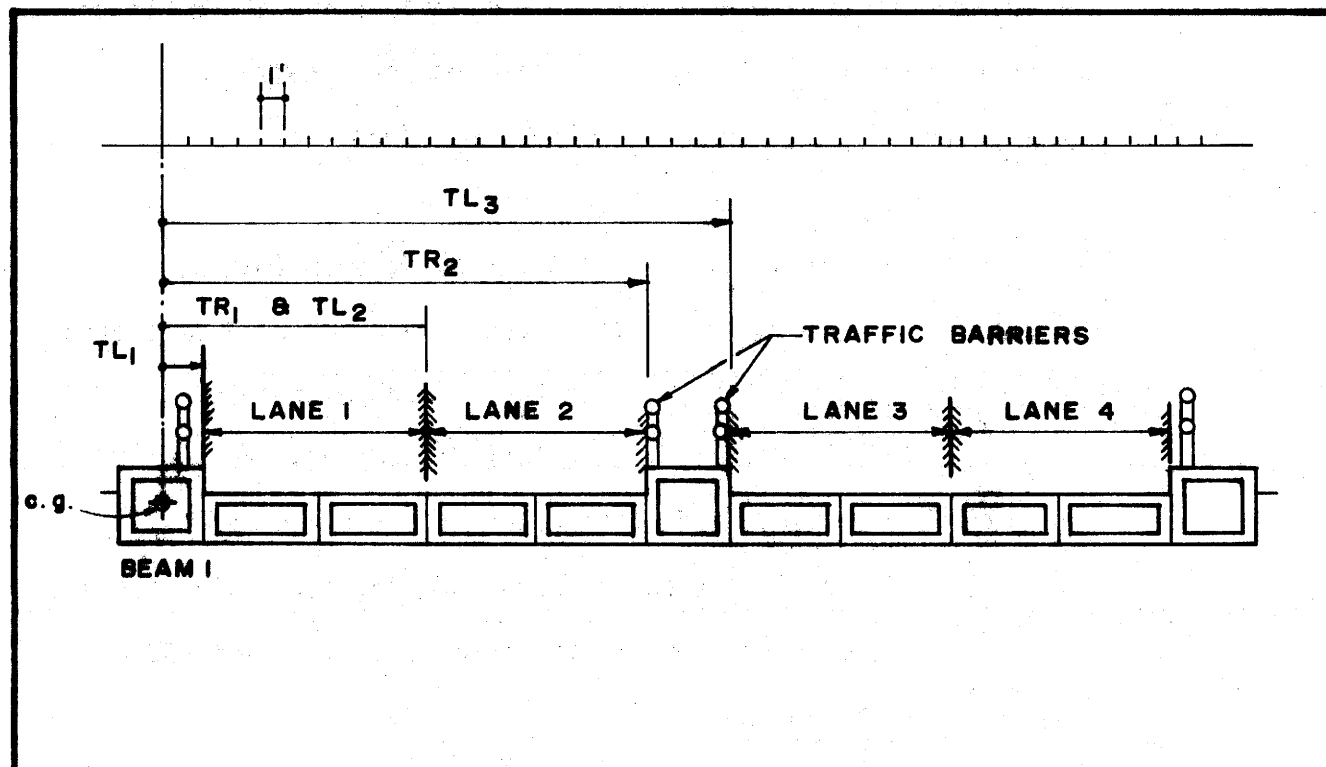


FIGURE 32. MULTIPLE LANE BRIDGE

6.1.2 Example Problem 5

The bridge of example 4.2.1 is to be modified to carry a concrete median barrier (CMB). This is accomplished by adding one 4.0 ft. box to the cross section for a total of eight boxes. The CMB is to be placed over the shear key between beams 4 and 5. The base width of the CMB is 2 ft.-3 in. and it weighs 485 lbs./ft. The forces produced by a vehicle impact on the CMB are not considered. The analysis is to determine what fraction of the barrier's weight is carried by the center beams. The moment of inertia of the box section is approximately 129,000 in.⁴ about the horizontal centroidal axis and 191,000 in.⁴ about the vertical axis. The St. Venant torsional constant J is computed from (16)

$$J = 2t_v t_h b^2 h^2 (1 - t_v/b)^2 (1 - t_h/h)^2 / \{bt_v + ht_h - (t_v + t_h)^2\} \quad (164)$$

where dimensions t_v , t_h , b and h pertain to a rectangular box inscribed within the box section under consideration and

t_v = thickness of vertical walls,

t_h = average thickness of top and bottom slab,

b = width of section,

h = depth of section.

For this section J is approximately 177,000 in.⁴. The input data are shown on the coding sheet in Figure 33. The loading is idealized as two uniformly distributed line loads of 242.5 lbs./ft., carried by beams 4 and 5 (the two center-most beams). The line of action of the two line loads is taken as 6 3/4 in. on either side off the middle of the shear key. The hinge joint between all beams is assumed to be at the c.g. of the shear key. The joint is assumed to transmit all shear and transverse forces but no transverse

CONTINUATION

DATA PROCESSING CENTER

TEXAS A & M UNIVERSITY

PROBLEM: ROUTINE INPUT FOR AMBB

PROGRAMMER: HLJ

FORTRAN

STATEMENT

FORTRAN										STATEMENT									
1	5	6	7	10	20	30	40	50	60	1	2	3	4	5	6	7	8	9	10
EXAMPLE OF ROUTINE ANALYSIS WITH AMBB										← Title Card									
420. 5000. .166 8 1 1 15										← Control Card									
191000. 129000. 859. 177000.										← Beam Card 1									
5.5 -24. 5.5 24.										← Beam Card 2									
1 1 1 1 1 1 1										← Beam Type I.D. Card									
0. 0. 0. 10000000.										← Hinge Flexibility Card									
1 1 1 1 1 1										← Hinge Type I.D. Card									
1 2										← Load Control Card									
1 4 8.488 210. 420. 17.25										← Load Card 1									
1 5 8.488 210. 420. -17.25										← Load Card 2									
1 1 210.										← Results Card 1									
2 0. -16.8 0. 17.2										← Results Card 2									

FIGURE 33. INPUT FOR EXAMPLE PROBLEM NO. 5

FIGURE 33. INPUT FOR EXAMPLE PROBLEM NO. 5

moment. The units chosen for input data are inches and kips.

The output for this example is shown in Figure 34. The fifth sheet of output lists the bending moment about the Y-axis for each beam at midspan. The moments carried by both beams 4 and 5 is 169.4 k-in. The moment produced by a uniform load of 485 lbs./ft. acting on a single 35 ft. long beam is 891.2 k-in. Thus, to include the effects of the CMB in design, a load of $.485 \text{ k/ft} \times (169.4/891.2) = .092 \text{ k/ft}$ would be input as uniform load on a single beam.

6.2 DETERMINATION OF LIVE LOAD LATERAL DISTRIBUTION FACTORS USING PROGRAM AMBB

Situations frequently occur where a rational approach (rather than an empirical expression) is needed for calculation of the lateral distribution of wheel loads on a multi-beam bridge. The analysis program, in its original form, could be used to accomplish this although it would be impractical because of the voluminous input required. The method by which it could be done manually (and which has been added to the program so that it is done automatically) is described below.

Current design practice stipulates that vehicles be confined to lanes on a bridge. Only one vehicle is permitted (laterally) within a lane for calculation of design moments. Lateral distribution factor is defined here as the ratio of the largest midspan moment produced in a beam by a vehicle or vehicles on the bridge to the maximum midspan moment of one vehicle carried entirely by that beam. Thus, lateral distribution is in terms of fractions of a vehicle (truck). It is assumed that the longitudinal position of a vehicle


```
*****ANALYSIS OF MULTI-BEAM BRIDGE*****  
EXAMPLE OF ROUTINE ANALYSIS WITH AMBB  
BRIDGE SPAN = 420.000  
YOUNGS MODULUS OF ELASTICITY = 5000.  
POISSONS RATIO = 0.166  
NUMBER OF BEAMS = 8  
NUMBER OF BEAM-TYPES = 1  
NUMBER OF JOINT-TYPES = 1  
NUMBER OF HARMONICS = 15
```

FIGURE 34. OUTPUT FOR EXAMPLE PROBLEM NO. 5

*****BEAM PROPERTIES*****

TYPE	I-ZZ	I-YY	AREA	TORS J	ZHL	YHL	ZHR	YHR
1	191000.0	129000.0	859.0	177000.0	5.50	-24.00	5.50	24.00

BEAM NO	TYPE
1	1
2	1
3	1
4	1
5	1
6	1
7	1
8	1

*****HINGE FLEXIBILITIES*****

TYPE	LCNG.	HORIZ.	VERT.	ROT.
1	0.0	0.0	0.0	1.00000E 07

JOINT NO	TYPE
1	1
2	1
3	1
4	1
5	1
6	1
7	1

*****LOADING CONDITIONS*****

LOAD CASE	BEAM NO	LOAD	X COORD	LENGTH	ECC.
1	4	8.488	210.000	420.000	17.250
1	5	8.488	210.000	420.000	-17.250

*****JOINT X-COORDS FOR RESULTS*****

LOAD CASE	NO OF POSITIONS	X1	X2	X3	X4	X5	X6	X7	X8	X9
1	1	210.0								

*****COORDINATES FOR OUTPUT OF AXIAL STRESS*****

BEAM TYPE	NO OF POSITIONS	Y1	Z1	Y2	Z2	Y3	Z3	Y4	Z4
1	2	0.0	-16.8	0.0	17.2				

FIGURE 34. (CONTINUED)

*****BEAM CENTER-LINE DISPLACEMENTS LOAD CASE NO 1*****

LOCATIONS ON BEAM 210.0

BEAM NO

1	1.950E-03
2	2.399E-03
3	3.395E-03
4	4.951E-03
5	4.951E-03
6	3.395E-03
7	2.399E-03
8	1.950E-03

FIGURE 34. (CONTINUED)

*****BENDING MOMENTS ABOUT Y-AXIS LOAD CASE NO 1*****

LOCATIONS ON BEAM 210.0

BEAM NO

1	7.027E 01
2	8.602E 01
3	1.199E 02
4	1.694E 02
5	1.694E 02
6	1.199E 02
7	8.602E 01
8	7.027E 01

FIGURE 34. (CONTINUED)

*****BENDING MOMENTS ABOUT Z-AXIS LOAD CASE NO 1*****

LOCATIONS ON BEAM 210.0

BEAM NO

1	-1.288E 00
2	-4.056E 00
3	-7.426E 00
4	-9.392E 00
5	9.392E 00
6	7.426E 00
7	4.056E 00
8	1.288E 00

FIGURE 34. (CONTINUED)

*****VERTICAL SHEARS LOAD CASE NO 1*****

LOCATIONS ON BEAM 210.0

BEAM NO	
1	6.713E-07
2	7.556E-07
3	1.170E-06
4	6.837E-07
5	6.837E-07
6	1.170E-06
7	7.556E-07
8	6.713E-07

FIGURE 34. (CONTINUED)

*****AXIAL FORCES LOAD CASE NO 1*****

LOCATIONS ON BEAM 210.0

BEAM NO

1	-9.946E-06
2	2.007E-06
3	1.019E-05
4	-4.816E-07
5	-1.702E-06
6	6.749E-06
7	-2.575E-06
8	-4.244E-06

FIGURE 34. (CONTINUED)

*****TORSIONAL MOMENTS LOAD CASE NO 1*****

LOCATIONS ON BEAM 210.0

BEAM NO

1	1.596E-05
2	5.035E-05
3	9.580E-05
4	1.206E-04
5	-1.206E-04
6	-9.580E-05
7	-5.035E-05
8	-1.596E-05

FIGURE 34. (CONTINUED)

*****FORCES ALONG LONGITUDINAL JOINT LOAD CASE NO 1*****

LONGITUDINAL SHEAR ON JOINT
LOCATIONS ON BEAM 210.0

JOINT NO	
1	-4.455E-14
2	-1.308E-13
3	-1.522E-13
4	9.872E-15
5	1.153E-13
6	1.230E-13
7	1.011E-13

TRANSVERSE FORCE ON JOINT
LOCATIONS ON BEAM 210.0

JOINT NO	
1	-1.478E-04
2	-1.619E-04
3	-6.485E-04
4	4.917E-05
5	-6.485E-04
6	-1.619E-04
7	-1.478E-04

VERTICAL SHEAR ON JOINT
LOCATIONS ON BEAM 210.0

JOINT NO	
1	-3.912E-03
2	-8.301E-03
3	-1.417E-02
4	-2.328E-09
5	1.417E-02
6	8.301E-03
7	3.912E-03

TRANSVERSE MOMENT ON JOINT
LOCATIONS ON BEAM 210.0

JOINT NO	
1	-9.844E-13

FIGURE 34. (CONTINUED)

-1.857E-12
-1.034E-12
7.516E-12
-1.034E-12
-1.257E-12
-5.254E-13

2 3 4 5 6 7

FIGURE 34. (CONTINUED)

*****AXIAL STRESS LOAD CASE NO 1*****

BEAM NO 1

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-9.151E-03
0.0	17.2		9.369E-03

BEAM NO 2

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-1.120E-02
0.0	17.2		1.147E-02

BEAM NO 3

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-1.561E-02
0.0	17.2		1.598E-02

BEAM NO 4

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-2.206E-02
0.0	17.2		2.259E-02

BEAM NO 5

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-2.206E-02
0.0	17.2		2.259E-02

BEAM NO 6

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-1.561E-02
0.0	17.2		1.598E-02

BEAM NO 7

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-1.120E-02
0.0	17.2		1.147E-02

BEAM NO 8

LOCATIONS	CN	BEAM	210.0
0.0	-16.8		-9.151E-03
0.0	17.2		9.369E-03

FIGURE 34. (CONTINUED)

on a bridge when it produces maximum midspan moment in a beam is independent of its transverse location on the bridge.

The first step is to determine the lateral position of a vehicle in each lane which produces maximum moment in each beam. This is an influence line problem. For convenience, the transverse expanse of the bridge is divided into one foot segments (Figure 32), using the c.g. location of beam 1 as a reference point. One line of wheels from the vehicle, positioned longitudinally for maximum moment, is moved transversely across the bridge and the moment produced in each beam for each station is stored. For a specific beam, the location of the vehicle in lane 1 that produces maximum moment can be found by moving the two wheel lines of the vehicle from the left to the right edge of the lane (observing required side clearances; e.g., 2 ft. for AASHTO trucks) and adding the two ordinates of the influence line to obtain the total moment. This process is repeated for all beams and all lanes and the results for each stored. The final step is to sum the effects, for a particular beam, of vehicles in each lane and applying a frequency of occurrence factor if appropriate. AASHTO, for example, allows the moment produced by three lanes loaded simultaneously to be reduced by 10 percent for design purposes.

6.2.1 Program Input

A standard input form shown in Figure 35 has been developed for use with the program in this mode. The input quantities are explained below.

1. Title Cards - Three title cards as indicated. Column 62 on the first title card must contain "1" in order that the input for this program mode can be distinguished from the conventional analysis mode.
2. Control Card - Span length, modulus, number of beams and traffic lanes input as indicated. AASHTO loading is designated in columns 38-42. If an axle train is to be used, complete columns 48-67. The

1
62
Input Trigger

FIGURE 35. INPUT FORM FOR AMBB

Hinge Type Number	Longitudinal Shear	Vertical Shear	Transverse Force	Transverse Moment
4	14	22	30	38

Hinge Type Identification Number For Hinge i

$i=1$ $i=2$ $i=3$ $i=4$ $i=5$ $i=6$ $i=7$ $i=8$ $i=9$ $i=10$ $i=11$ $i=12$ $i=13$ $i=14$ $i=15$ $i=16$ $i=17$ $i=18$ $i=19$

Distance (ft) From c.g. Axis of Beam 1 to

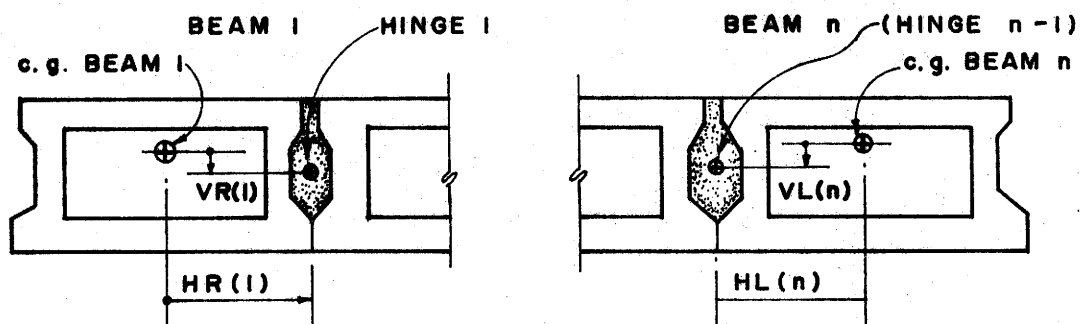
Left Edge	Right Edge	Left Edge	Right Edge	Left Edge	Right Edge	Left Edge	Right Edge	Left Edge	Right Edge
Lane 1	Lane 1	Lane 2	Lane 2	Lane 3	Lane 3	Lane 4	Lane 4	Lane 5	Lane 5
5 8	13 16	21 24	29 32	37 40	45 48	53 56	61 64	69 72	77 80

Axle Train

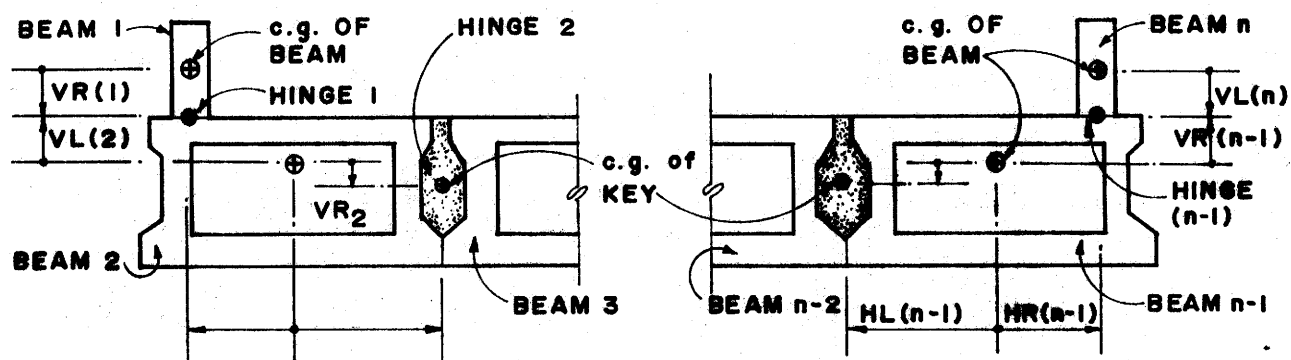
FIGURE 35. (CONTINUED)

axle train side clearance is the minimum distance permitted between a wheel line and an edge of a lane. The number of axle trains present on the bridge is the maximum number of lanes that will be loaded in attempting to produce maximum moment in a beam.

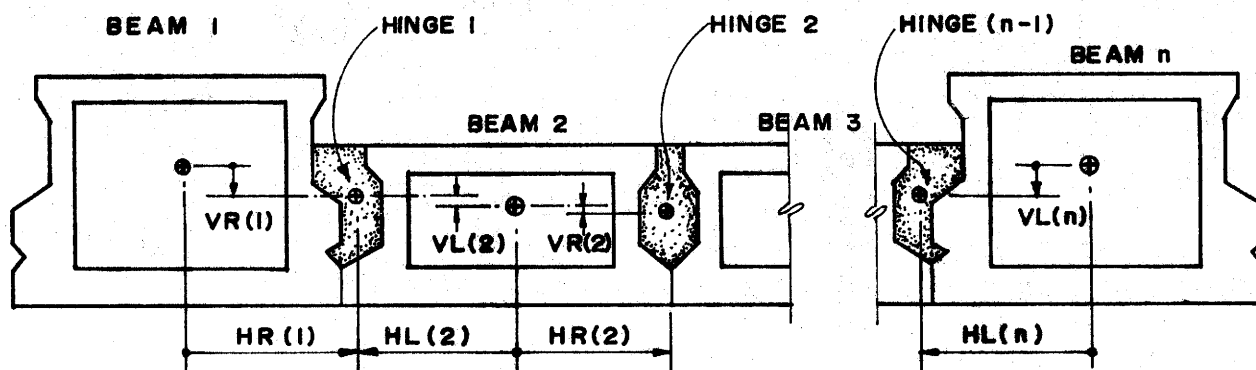
3. Beam Card - Moment of inertia about a horizontal axis refers to the y-axis shown in Figure 31, which in most cases will be horizontal. The vertical axis refers to the Z-axis in Figure 31. The torsional stiffness in columns 45-52 can be computed from Eq. (164) for most sections. HL and VL are the horizontal and vertical distances from the centroid of the beam cross section to the left hinge. The hinge can conveniently be taken at the centroid of the shear key. HL is positive if the hinge is to the left of the centroid of the beam (the negative y-direction in Figure 31) and VL is positive if the hinge lies below the beam centroid. HR and VR define the position of the right hinge, with VR positive if the right hinge is below the beam centroid and HR positive if the hinge is to the right of the beam centroid. Several typical situations are shown in Figure 36 with the various dimensions labeled.
4. Beam Type Identification Card - For each beam, starting from the left, enter the beam type number.
5. Hinge Card - A hinge is assumed to be either completely flexible (no force transmission) or completely rigid (full force transmission) in each of its 4 possible modes of displacement. If the hinge transmits longitudinal shear force (a_j in Figure 31), "Y" is entered in column 14. If a_j must be zero, then column 14 should contain "N" or left blank. The remaining components of joint force



CASE I



CASE II



CASE III

FIGURE 36. TYPICAL BEAM AND HINGE ARRANGEMENTS

are vertical shear (v_i in Figure 31), transverse force (h_i) and transverse moment (m_i).

6. Hinge Type Identification Card - For each hinge, beginning with the hinge between beams 1 and 2, enter the hinge type number.
7. Lane Location Card - The limits of traffic lanes are measured with respect to the centroid of the left-most beam (beam number 1) in the bridge. Distance is positive to the right of the centroid. Lane limits are shown as TR_i and TL_i in Figure 32.
8. Axle Train Cards - Enter only if column 48 of the control card contains "1". Either the leading or trailing axle may be designated as axle 1, with the remaining axles numbered in sequential order. The first card contains the axle loads. The second card contains axle spacings (See Section 4.1.3)

6.2.2 Example Problem 6

Lateral distribution factors are to be computed for the structure shown in Figure 32. The shallow boxes are 6 ft. wide by 2 ft.-3 in. deep with 5 1/2 in. thick horizontal walls and 5 in. vertical walls. The exterior and center beam are 3 ft. wide and 3 ft.-3 in. deep with the same wall thicknesses. A work sheet has been provided (Figure 37) to assist in data input. The completed work sheet for this example is seen in Figure 38. The structure is sketched to some convenient scale and the layout for beams, joints and traffic lanes are indicated. With the completed work sheet, it is a simple matter to complete the input form as shown in Figure 39. For this problem, four 12 ft. traffic lanes are utilized, with HS-20 loading.

The results of this analysis are shown in Figure 40. The first two sheets list input data. The location of loads within each lane which produce maximum moment for a beam are displayed at the top of the third sheet. For

TEXAS HIGHWAY DEPARTMENT
BRIDGE DIVISION

DATA INPUT WORK SHEET

Sketch of Bridge Cross Section	
Beam Number	
Beam Type Number	
Hinge Number	
Hinge Transmits: (Y or N)	
Longitudinal Shear	
Vertical Shear	
Transverse Force	
Transverse Moment	
Hinge Type Number	

FIGURE 37. DATA INPUT WORK SHEET FOR AMBB

MULTIBEAM BRIDGE
ANALYSIS PROGRAM

TEXAS HIGHWAY DEPARTMENT
BRIDGE DIVISION
DATA INPUT WORK SHEET

Sketch of Bridge Cross Section

Beam Number	1	2	3	4	5	6	7	8	9	10	11
Beam Type Number	1	2	2	2	2	1	2	2	2	2	1
Hinge Number	1	2	3	4	5	6	7	8	9	10	
Hinge Transmits: (Y or N)											
Longitudinal Shear	Y	Y	Y	Y	Y	Y	Y	Y	Y	Y	Y
Vertical Shear	Y	Y	Y	Y	Y	Y	Y	Y	Y	Y	Y
Transverse Force	Y	Y	Y	Y	Y	Y	Y	Y	Y	Y	Y
Transverse Moment	N	N	N	N	N	N	N	N	N	N	N
Hinge Type Number	1	1	1	1	1	1	1	1	1	1	1

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FIGURE 38. COMPLETED WORK SHEET FOR EXAMPLE PROBLEM NO. 6

MULTIBEAM BRIDGE
ANALYSIS PROGRAM

TEXAS HIGHWAY DEPARTMENT
BRIDGE DIVISION

Sheet 1 of 2

DISTRICT **03** ARCHER COUNTY HIGHWAY NO. **SH 25**

CONTROL NO. IPE SUBMITTED BY **HLJ**

DESCRIPTION **EXAMPLE PROBLEM NO. 6**

GENERAL INFORMATION

1
62
Input Trigger

Span (ft.) **80** Modulus of Elasticity (Ksi) **5000** Number of Beams **11** Number of Traffic Lanes **4** AASHTO Loading **HS-20** Enter 1 for Axle Train **1** Transverse Axle Train Wheel Spacing (ft.) **62** Axle Train Side Clearance (ft.) **61** Number of Axle Trains on Bridge **67**

Beam Type Number **1** Inertia About Horizontal Axis (in.⁴) **130400** Inertia About Vertical Axis (in.⁴) **110600** Area (in.²) **676** Torsional Stiffness (in.⁴) **208700** HL (in) **18** VL (in) **-25** HR (in) **18** VR (in) **-25**

Beam Type Identification Number For Beam i

i=1	i=2	i=3	i=4	i=5	i=6	i=7	i=8	i=9	i=10	i=11	i=12	i=13	i=14	i=15	i=16	i=17	i=18	i=19	i=20
1	2	2	2	2	1	2	2	2	2	1									

FIGURE 39. INPUT FOR EXAMPLE PROBLEM NO. 6

Hinge Type Number	Longitudinal Shear	Vertical Shear	Transverse Force	Transverse Moment
1	Y	Y	Y	N
4	14	22	30	38

i=1 i=2 i=3 i=4 i=5 i=6 i=7 i=8 i=9 i=10 i=11 i=12 i=13 i=14 i=15 i=16 i=17 i=18 i=19

 5 6 8 9 11 12 14 15 17 18 20 21 23 24 26 27 29 30 32 33 35 36 38 39 41 42 44 45 47 48 50 51 53 54 56 57 59 60

Left Edge Lane 1	Right Edge Lane 1	Left Edge Lane 2	Right Edge Lane 2	Left Edge Lane 3	Right Edge Lane 3	Left Edge Lane 4	Right Edge Lane 4	Left Edge Lane 5	Right Edge Lane 5
5 8	13 16	21 24	29 32	37 40	45 48	53 56	61 64	69 72	77 80

Diagram illustrating the 18-axle test vehicle configuration. The vehicle is shown with 18 axles, labeled Axle 1 through Axle 18. Each axle is represented by a 2x2 grid of wheels. The diagram shows the relative positions of the axles and the wheels. A legend on the right indicates 'Axle Load (1 Dist. From Axle 1 To Axle i (ft))'.

FIGURE 39. (CONTINUED)

VALUES ASSIGNED TO INPUT DATA

DISTRICT C3 ARCHER COUNTY HIGHWAY NO. SH 25
 CONTROL NO. IPE SUBMITTED BY: FLJ
 DESCRIPTION: EXAMPLE PROBLEM NO. 6

..... BRIDGE PROPERTIES

SPAN (FT.)	MODULUS OF ELASTICITY (KSI)	NUMBER OF BEAMS	NUMBER OF TRAFFIC LANES	AASHTO LOADING	TRANSVERSE AXLE TRAIN WHEEL SPACING (FT.)	AXLE TRAIN SIDE CLEARANCE (FT.)	NUMBER OF AXLE TRAINS ON BRIDGE
80.0	5000.0	11	4	HS-20	0	0	0

..... BEAM DIMENSIONS AND PROPERTIES

BEAM TYPE NUMBER	INERTIA ABOUT HORIZONTAL AXIS (IN. ⁴)	INERTIA ABOUT VERTICAL AXIS (IN. ⁴)	AREA (IN. ²)	TORSIONAL STIFFNESS (IN. ⁴)	HL (IN.)	VL (IN.)	HR (IN.)	VR (IN.)
1	130400.0	110600.0	676.0	208700.0	-18.00	-2.50	16.00	-2.50
2	96900.0	522000.0	952.0	286600.0	-36.00	-8.50	36.00	-8.50

..... BEAM TYPE IDENTIFICATION NUMBER FOR BEAM 1

I=1	I=2	I=3	I=4	I=5	I=6	I=7	I=8	I=9	I=10	I=11	I=12	I=13	I=14	I=15	I=16	I=17	I=18	I=19	I=20
1	2	2	2	2	1	2	2	2	2	2	1								

..... HINGE FORCE TRANSMISSION

HINGE TYPE NUMBER	LONGITUDINAL SHEAR	VERTICAL SHEAR	TRANSVERSE FORCE	TRANSVERSE MOMENT
1	Y	Y	Y	N

..... HINGE TYPE IDENTIFICATION NUMBER FOR HINGE 1

I=1	I=2	I=3	I=4	I=5	I=6	I=7	I=8	I=9	I=10	I=11	I=12	I=13	I=14	I=15	I=16	I=17	I=18	I=19

FIGURE 40. OUTPUT FOR EXAMPLE PROBLEM NO. 6

FIGURE 40. OUTPUT FOR EXAMPLE PROBLEM NO. 6

 * LATERAL POSITION OF LOADS *
 * FOR MAXIMUM MOMENT AT MIDSPAN *

***** AASHTO TRUCK *****
 (POSITION OF TRUCK WHEELS IN LOADED LANES WITH RESPECT TO C.G. OF BEAM NO.1 = IN FT.)

BEAM NO.LANE 1.....	LANE 2.....	LANE 3.....	LANE 4.....	LANE 5.....	
	TO LEFT WHEEL	TO RIGHT WHEEL	TO LEFT WHEEL	TO RIGHT WHEEL	TO LEFT WHEEL	TO RIGHT WHEEL	TO LEFT WHEEL	TO RIGHT WHEEL	TO LEFT WHEEL	TO RIGHT WHEEL
1	3.5	5.5	15.5	21.5	30.5	36.5	42.5	48.5		
2	3.5	9.5	15.5	21.5	30.5	36.5	42.5	48.5		
3	5.5	11.5	15.5	21.5	30.5	36.5	42.5	48.5		
4	5.5	11.5	15.5	21.5	30.5	36.5	42.5	48.5		
5	5.5	11.5	17.5	23.5	30.5	36.5	42.5	48.5		
6	5.5	11.5	17.5	23.5	30.5	36.5	42.5	48.5		
7	5.5	11.5	17.5	23.5	30.5	36.5	42.5	48.5		
8	5.5	11.5	17.5	23.5	32.5	38.5	42.5	48.5		
9	5.5	11.5	17.5	23.5	32.5	38.5	42.5	48.5		
10	5.5	11.5	17.5	23.5	32.5	38.5	44.5	50.5		
11	5.5	11.5	17.5	23.5	32.5	38.5	44.5	50.5		

***** AASHTO LANE *****
 (POSITION OF LANE LOADING WITHIN LOADED LANE WITH RESPECT TO C.G. OF BEAM NO.1 = IN FT.)

BEAM NO.LANE 1.....	LANE 2.....	LANE 3.....	LANE 4.....	LANE 5.....	
	TO LEFT EDGE	TO RIGHT EDGE	TO LEFT EDGE	TO RIGHT EDGE	TO LEFT EDGE	TO RIGHT EDGE	TO LEFT EDGE	TO RIGHT EDGE	TO LEFT EDGE	TO RIGHT EDGE
1	1.5	11.5	13.5	23.5	28.5	38.5	40.5	50.5		
2	1.5	11.5	13.5	23.5	28.5	38.5	40.5	50.5		
3	3.5	13.5	13.5	23.5	28.5	38.5	40.5	50.5		
4	3.5	13.5	13.5	23.5	28.5	38.5	40.5	50.5		
5	3.5	13.5	15.5	25.5	28.5	38.5	40.5	50.5		
6	3.5	13.5	15.5	25.5	28.5	38.5	40.5	50.5		
7	3.5	13.5	15.5	25.5	28.5	38.5	40.5	50.5		
8	3.5	13.5	15.5	25.5	30.5	40.5	40.5	50.5		

FIGURE 40. (CONTINUED)

9	3.5	13.5	15.5	25.5	30.5	40.5	40.5	50.5
10	3.5	13.5	15.5	25.5	30.5	40.5	42.5	52.5
11	3.5	13.5	15.5	25.5	30.5	40.5	42.5	52.5

 * MAXIMUM MIDSPAN MOMENTS AND *
 * LATERAL DISTRIBUTION FACTORS *

BEAM NO.	MOMENT FROM AASHTO TRUCK (KIP-FT.)	FRACTION OF FULL AASHTO TRUCK APPLIED	MOMENT FROM AASHTO LANE (KIP-FT.)	FRACTION OF FULL AASHTO LANE APPLIED	MOMENT FROM AXLE TRAIN (KIP-FT.)	FRACTION OF FULL AXLE TRAIN APPLIED
1	486.5	0.339	366.2	0.338		
2	382.2	0.265	287.5	0.265		
3	400.1	0.277	299.9	0.276		
4	398.1	0.276	298.3	0.275		
5	377.1	0.261	283.7	0.262		
6	498.6	0.346	373.2	0.344		
7	377.1	0.261	283.7	0.262		
8	398.1	0.276	298.3	0.275		
9	400.1	0.277	299.9	0.276		
10	382.2	0.265	287.5	0.265		
11	486.5	0.339	366.2	0.338		

FIGURE 40. (CONTINUED)

example, if only lane 2 were loaded, the truck wheels should be located 2 ft. from the left edge of the lane to produce maximum moment in beams 1 through 4 and 2 ft. from the right edge of the lane for the remaining beams. Comparable information on the location of the 10 ft. wide lane loading is also shown. The last entry on the last sheet displays maximum moment and lateral distribution factor for each beam. The maximum moments include the AASHTO impact factor for axle train as well as AASHTO loadings.

REFERENCES

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APPENDIX A

BEAM COST SURVEY DATA

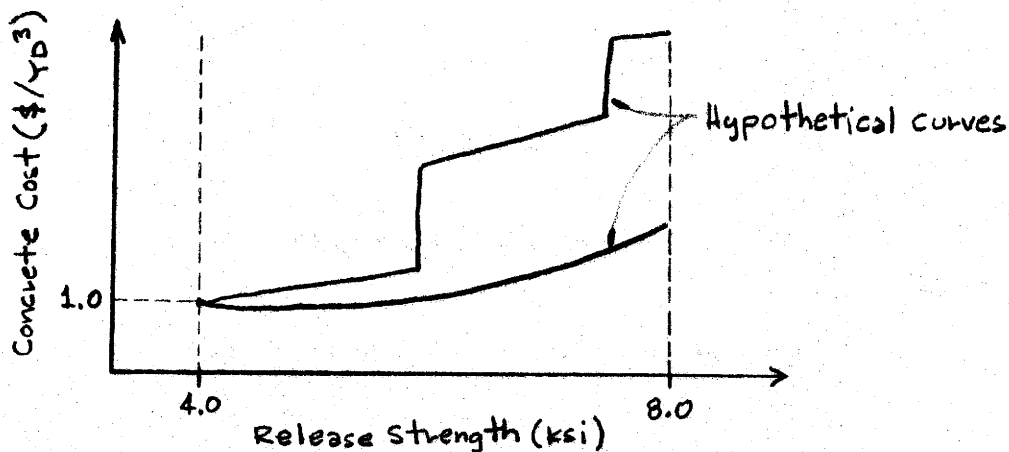
MATERIALS COST INFORMATION SHEET

Company Name _____

Location _____

Date _____

1. The cost of one cubic yard of concrete in a completed highway beam is effected by the required release strength, especially when high release strengths are specified. Our objective is to produce a graph similar to that shown below, which would be representative of concrete cost changes with change in release strength.



We are asking that you provide us with costs of various release strength concretes so we can construct such a graph. Since we are interested only in relative costs, we have arbitrarily set the cost of one cubic yard of 4000 psi. release strength concrete at \$1.00. Using this as a base, would you please complete table below, up through the highest release strength your company would be willing to produce.

QUESTIONNAIRE SENT TO 7 FABRICATORS IN TEXAS

Release Strength (psi)	Concrete Cost (\$)
4000	<u>\$1.00</u>
4500	<u>\$</u>
5000	<u>\$</u>
5500	<u>\$</u>
6000	<u>\$</u>
6500	<u>\$</u>
7000	<u>\$</u>
7500	<u>\$</u>
8000	<u>\$</u>

2. Indicate those items listed below that were considered in arriving at the costs computed for question 1.

- Yes___ No___ (a) Cost of materials (cement, aggregate, admixtures, etc.)
- Yes___ No___ (b) Cost of energy used in curing(e.g., natural gas, oil, electricity, etc.)
- Yes___ No___ (c) Cost of labor in placing concrete
- Yes___ No___ (d) Amortization of permanent equipment (such as forms, curing facilities, etc.) through periodic depreciation.
- Yes___ No___ (e) Increased cost due to decreased production associated with leaving beams in forms to obtain high release strength.
- Yes___ No___ (f) Increased overhead due to reduced production.
- (g) Other

3. What would be the in-place cost per foot of strand for 1/2 inch diameter, grade 270, 7 wire strand?

Release
Strength (psi)

Concrete
Cost (\$)

4000	<u>\$1.00</u>
4500	<u>\$1.00</u>
5000	<u>\$1.00</u>
5500	<u>\$1.00</u>
6000	<u>\$1.22</u>
6500	<u>\$1.34</u>
7000	<u>\$1.54</u>
7500	<u>\$1.85</u>
8000	<u>\$2.22</u>

2. Indicate those items listed below that were considered in arriving at the costs computed for question 1.

- Yes x No (a) Cost of materials (cement, aggregate, admixtures, etc.)
- Yes x No (b) Cost of energy used in curing (e.g., natural gas, oil, electricity, etc.)
- Yes No x (c) Cost of labor in placing concrete
- Yes No x (d) Amortization of permanent equipment (such as forms, curing facilities, etc.) through periodic depreciation.
- Yes x No (e) Increased cost due to decreased production associated with leaving beams in forms to obtain high release strength.
- Yes x No (f) Increased overhead due to reduced production.
- (g) Other

3. What would be the in-place cost per foot of strand for 1/2 inch diameter, grade 270, 7 wire strand?

\$.20 per ft.

RESPONSE FROM COMPANY 1

Release
Strength (psi)

Concrete
Cost (\$)

(In place, finished
product)

4000

\$1.00

4500

\$ 1.00

5000

\$ 1.00

5500

\$ 1.25

6000

\$ 1.80

6500

\$ 2.00

7000

\$ 2.50

7500

\$ 3.00

Probably would not - 8000

\$ 3.80 - Would be difficult to
obtain in a reasonable
production cycle.

2. Indicate those items listed below that were considered in arriving at the costs computed for question 1.

- Yes X No (a) Cost of materials (cement, aggregate, admixtures, etc.)
- Yes X No (b) Cost of energy used in curing (e.g., natural gas, oil, electricity, etc.)
- Yes No X (c) Cost of labor in placing concrete
- Yes No X (d) Amortization of permanent equipment (such as forms, curing facilities, etc.) through periodic depreciation.
- Yes X No (e) Increased cost due to decreased production associated with leaving beams in forms to obtain high release strength.
- Yes X No (f) Increased overhead due to reduced production.
- (g) Other - Due to long production time - would be unable to bid other work.

3. What would be the in-place cost per foot of strand for 1/2 inch diameter, grade 270, 7 wire strand?

25¢/ lin. ft.

RESPONSE FROM COMPANY 2

Release Strength (psi)	Concrete Cost (\$)
4000	<u>\$1.00</u>
4500	<u>\$1.00</u>
5000	<u>\$1.10</u>
5500	<u>\$1.20</u>
6000	<u>\$1.20</u>
6500	<u>\$1.80</u>
7000	<u>\$1.85</u>
7500	<u>\$2.00</u>
8000	<u>\$2.10</u>

2. Indicate those items listed below that were considered in arriving at the costs computed for question 1.

- Yes XX No (a) Cost of materials (cement, aggregate, admixtures, etc.)
- Yes XX No (b) Cost of energy used in curing (e.g., natural gas, oil, electricity, etc.)
- Yes No XX (c) Cost of labor in placing concrete
- Yes No XX (d) Amortization of permanent equipment (such as forms, curing facilities, etc.) through periodic depreciation.
- Yes XX No (e) Increased cost due to decreased production associated with leaving beams in forms to obtain high release strength.
- Yes XX No (f) Increased overhead due to reduced production.
- (g) Other

3. What would be the in-place cost per foot of strand for 1/2 inch diameter, grade 270, 7 wire strand? \$.25

RESPONSE FROM COMPANY 3

Release
Strength (psi)

Concrete
Cost (\$)

4000	<u>\$1.00</u>
4500	<u>\$1.00</u>
5000	<u>\$1.05</u>
5500	<u>\$1.05</u>
6000	<u>\$1.57</u>
6500	<u>\$1.90</u>
7000	<u>\$1.90</u>
7500	<u>\$1.90</u>
8000	<u>\$2.25</u>

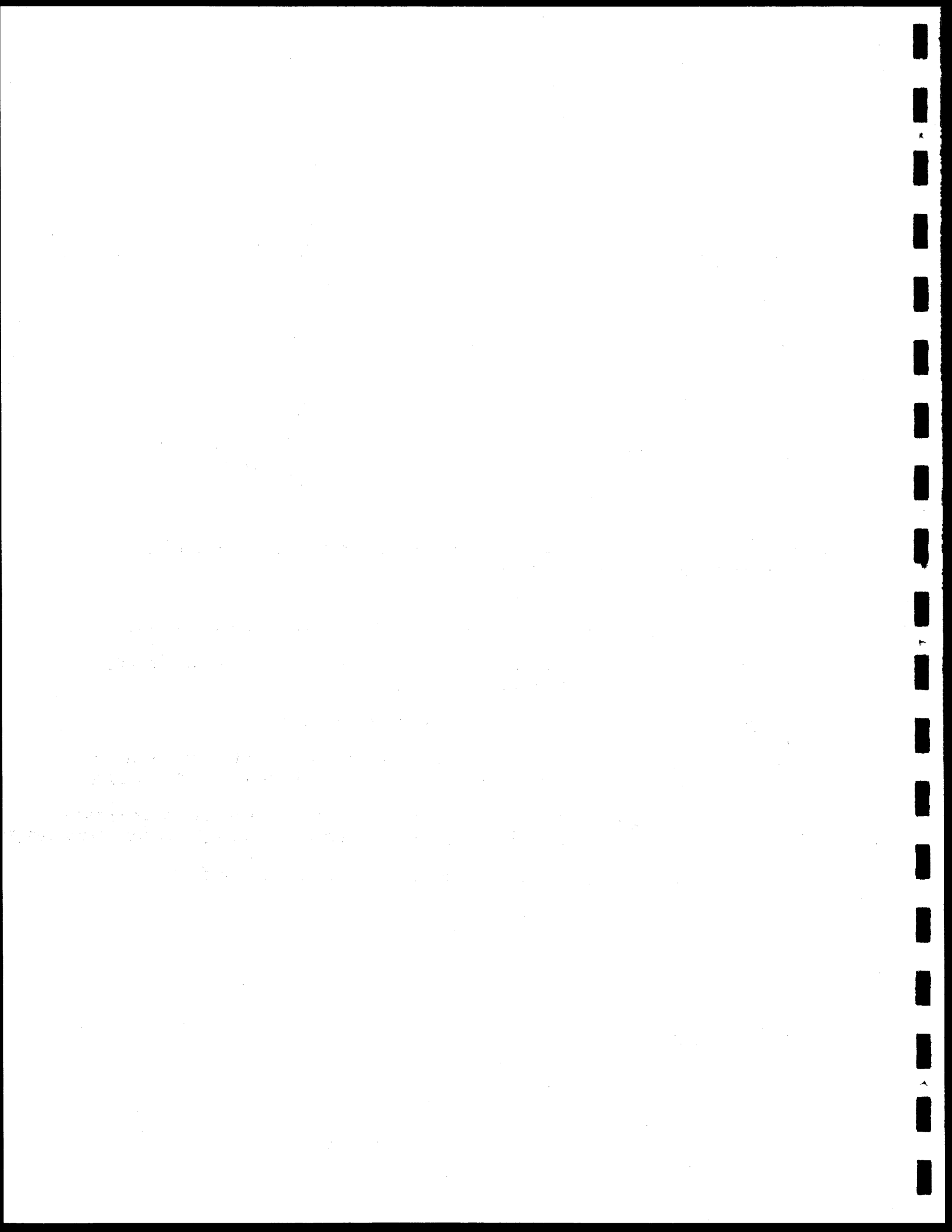
2. Indicate those items listed below that were considered in arriving at the costs computed for question 1.

- Yes ☒ No ☐ (a) Cost of materials (cement, aggregate, admixtures, etc.)
- Yes ☒ No ☐ (b) Cost of energy used in curing (e.g., natural gas, oil, electricity, etc.)
- Yes ☒ No ☐ (c) Cost of labor in placing concrete
- Yes ☐ No ☒ (d) Amortization of permanent equipment (such as forms, curing facilities, etc.) through periodic depreciation.
- Yes ☒ No ☐ (e) Increased cost due to decreased production associated with leaving beams in forms to obtain high release strength.
- Yes ☐ No ☒ (f) Increased overhead due to reduced production.
- (g) Other

3. What would be the in-place cost per foot of strand for 1/2 inch diameter, grade 270, 7 wire strand?

0.22 / LF.

RESPONSE FROM COMPANY 4



APPENDIX B

DEFINITION OF VARIABLES APPEARING IN LABELLED COMMON BLOCKS OF DBOXSS AND DBOXDS

Labelled common blocks used in both DBOXSS and DBOXDS are essentially identical. In a few cases, variable names are unique to one program or the other and in this situation both names are listed under the appropriate common block name.

COMMON/BLK1/

ADIM, BDIM, CDIM, DDIM, EDIM, FDIM, GDIM, HDIM, TDIM, XDIM, YDIM -

- the cross sectional dimensions A, B, ..., Y on input form (in) .

WHDIM - the dimension M on input form (in) .

WDDIM - the dimension W on input form (in) .

ACONC - cross sectional area of beam (in^2) .

BINERT - moment of inertia of beam about c.g. axis of beam (in^4) .

DTOP - distance from c.g. of beam cross section to bottom of beam (in) .

ZT - section modulus for computing stress in top of beam (in^3) .

ZB - section modulus for computing stress in bottom of beam (in^3) .

ACONCK - area of beam cross section plus shear key (in^2) .

BINERK - moment of inertia of beam cross section and shear key about c.g.
axis of composite beam and shear key section (in^4) .

DTOPK - distance from c.g. of beam and shear key composite section to
bottom of beam (in) .

ZTK - section modulus for computing stress at top of composite beam and key
section (in^3) .

ZBK - section modulus for computing stress at bottom of composite beam and
key section (in^3) .

ZL - span length (ft) .

F28 - 28 day concrete strength (ksi) .

JOPT - program option code: if = 0, design option is performed; if = 1, optimization option performed.

ASSCLR - fraction of initial prestress lost immediately after strand release.

ASSPLS - fraction of initial prestress lost after all losses have occurred.

APRIME - area of compression steel reinforcing (in^2).

CBOT - distance from bottom of beam to center of first strand row (in).

COSTWP - cost of strand (\$/FT) midspan.

DEFMIN - minimum permissible midspan camber upon release (in).

ALDEF - maximum permissible midspan camber upon release (in).

NRAV - the number of rows available for strand placement in DBOXSS and the maximum number of strand rows that can be fitted in the beam cross section for DBOXDS.

NWHEEL - number of axles in axle train vehicle.

DISTF - lateral distribution factor applied to axle loads.

FPCMAX - maximum permissible concrete release strength (ksi).

FPCMIN - minimum permissible concrete release strength (ksi).

ELASC - constant which when multiplied times the square root of concrete strength (ksi) gives the modulus of elasticity in ksi (ksi^2).

ULTMRQ - required ultimate moment capacity (k-ft).

DCR - distance from top of beam to c.g. of compression reinforcing steel (in).

W - beam weight (k/ft).

WB - shear key weight (k/ft).

FO - initial strand force (kips).

NR - the number of rows available for placement of strands.

HDPT - distance from center line of beam to holddown point (ft).

- NW - number of strands to be draped in strand rows .
- ALPHA - fraction which when multiplied times the span length gives the distance from the end of the beam to the holddown point .

COMMON/BLK2/

- PAXLE(I) - weight of Ith axle in axle train .
- NWHL(I) - distance from axle 1 to axle I in axle train (ft) .
- STRMAX(I) - maximum number of strands permitted in strand row I .
- FCNC(I) - magnitude of Ith concentrated force applied to a single beam (kips) .
- DCNC(I) - distance from the left end of the beam to the Ith concentrated force .
- G(I) - cost of concrete with release strength of $3.5 + 0.5I (\$/yd^3)$.
- F(I) - 28-day strength of concrete with a release strength of $3.5 + 0.5I$ (ksi) .
- D(I) - distance from c.g. of beam cross section to strand row I (in) .
Positive if strand row is above c.g. of beam .
- GRIDS(I) - distance from strand row I to row (I + 1) (in) .
- BMMA(I) - bending moment at points along beam due to live load. The locations are: I=1, end of beam; I=2, L/10; I=3, 2L/10; I=4, L/4; I=5, 3L/10; I=6, 4L/10; I=7, L/2 (k-ft) .
- BVMA(I) - live load shears at points along the beam (kips) .
- DLMO(I) - dead load moment at points along the beam due to uniform load and concentrated forces (kip-ft) .
- DLSH(I) - ultimate dead load shear at points along the beam due to uniform load and concentrated forces (kip-ft) .
- CBRMA(I) - array used to store long term camber computed for 4 sets of concrete properties and a set which is input .

- PRLMAX(I) - array containing estimates of prestress loss computed for 4 sets of concrete properties and input properties in SUBROUTINE CAMBER .
- ZLOS(I) - contains final prestress loss fractions 0.1, 0.2, 0.3, 0.4 .
- PECRK(I,J) - contains the total prestress force eccentricity at which the ultimate moment capacity first exceeds 1.2 times the cracking moment capacity for the 28 day concrete strength corresponding to a release strength of $3.5 + 0.5I$ and a prestress loss of $0.1J$ (k-in) .
- ZWRAP(I,J) - for strand row I, contains the number of wrapped strands in $J=1$ and the total length in feet ($J=2$) and inches ($J=3$) of wrapping .
- NSDIF(I) - number of different wrapping lengths in strand row I .

COMMON/DEFINE/

- UWC - unit weight of beam and shear key concrete (k/ft^3) .
- HUM - average relative humidity used in computing prestress loss due to shrinkage (%) .
- AS - area of a strand (in^2) .
- FPS - ultimate strength of the strand (ksi) .
- CTR1 - factor which when multiplied times the square root of concrete strength (in psi) gives the allowable tension stress in the concrete in psi for the end of the beam and the 1/10 point at release .
- CTR2 - same as CTR1, but for points between the 1/10 and 5/10 points on the beam .

CTS1 - same as CTR1, but for service load conditions .
 CTS2 - same as CTR2, but for service load conditions .
 CBR1 - factor which when multiplied by concrete strength gives the allowable
 compression stress in the concrete at the end of the beam and
 the 1/10 point at release .
 CBR2 - same as CBR1, but for points between the 1/10 and 5/10 points .
 CBS1 - same as CBR1, but for service load conditions .
 CBS2 - same as CBR2, but for service load conditions .
 CREEP1 - constant appearing in the numerator of the hyperbolic function
 defining the unit creep function for the concrete .
 CREEP2 - constant appearing in the denominator of the unit creep function .
 SHRK1 - constant appearing in numerator of the hyperbolic function defining
 the shrinkage properties of the concrete .
 SHRK2 - constant appearing in the denominator of the shrinkage function .
 RATNOD - ratio of the modulus of elasticity of compression steel to that
 of concrete. Used in computing transformed section properties .
 FPL - proportional limit stress for strand material (ksi) .
 FSY - yield strength of compression reinforcing and stirrups (ksi) .
 ASTIRP - total area of stirrup reinforcing (in^2) .
 GSP - spacing between strand row for DBOXDS .

COMMON/YZ/

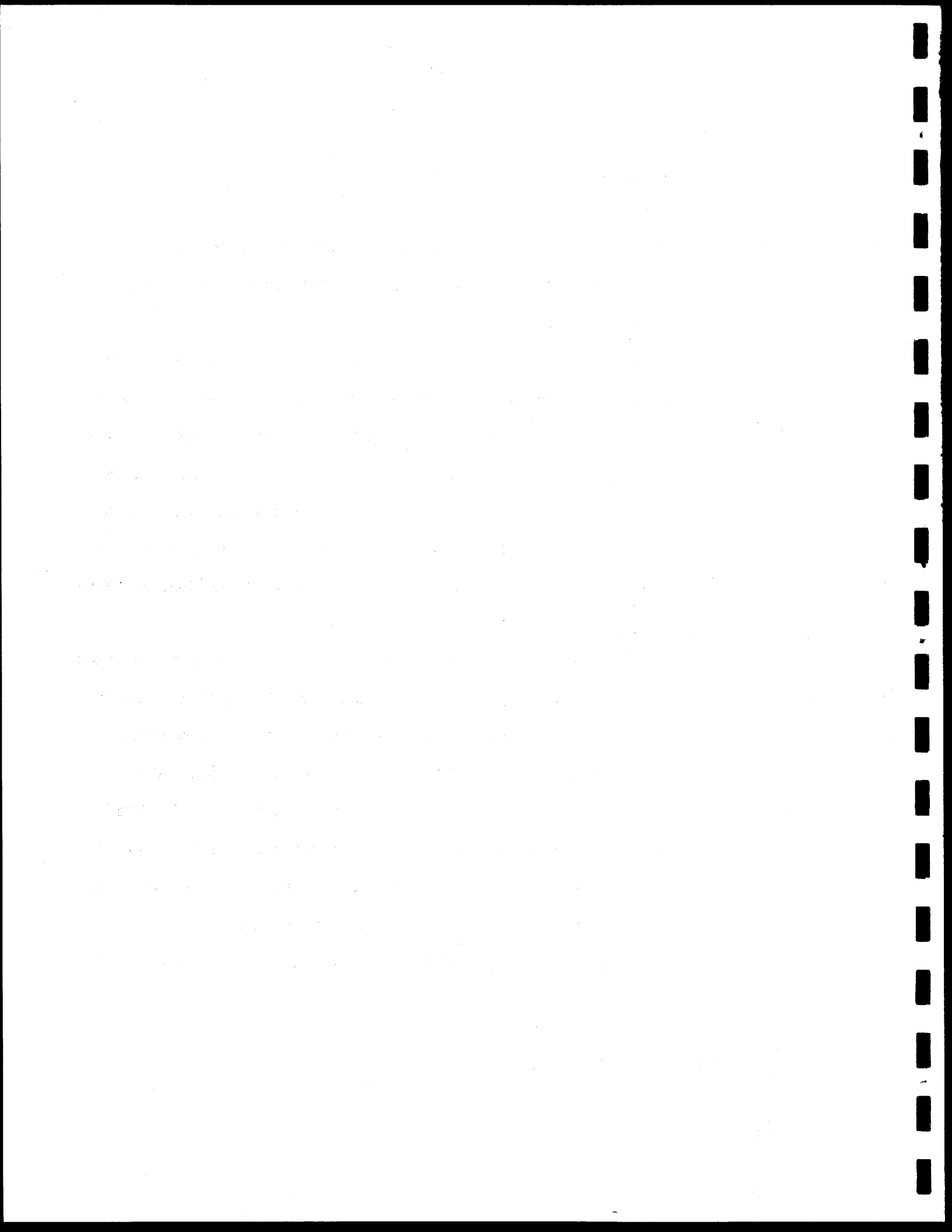
Y1, Y2, Y3, Y4, Z1, Z2, Z3, Z4 - characteristic widths and depths used in
 computing area and c.g. of compression zone for ultimate
 moment computations (see SUBROUTINE ULTMP) .

COMMON/DUMP/

- TITLE(I,J) - used to store input title cards .
- YJ(I) - scratch storage.
- FROW(I) - used to store number of strands placed in each row during generation of coefficients in ultimate and cracking moment constraints in SUBROUTINE EQGEN .
- PEF(I,J) - used in EQGEN to form ultimate and cracking moment constraints
Row I corresponds to the placement of 2·I strands in the beam (except when one or more row has an odd maximum number of strands).
Column 1 contains the total number of strands placed. Column 2 contains the sum of the products of the number of strands in each row times the distance of that row from the c.g. of beam. Column 3 contains the distance from the c.g. of PEF(I,1) strands to the c.g. of the beam.
- ZNE(I) - contains the product of the number of strands in each row times their distance from the c.g. of beam when the ultimate moment capacity first exceeds that required, for a 28 day concrete strength corresponding to a release strength of $3.5 + 0.5I$.
- KKODE(I) - used in EQGEN to form cracking moment constraint. If KKODE(I) = 1, then for current total strand force eccentricity, the ultimate moment capacity just exceeds 1.2 times the cracking moment capacity for a final prestress loss of ZLOS(I).
- ZMCR(I) - contains the cracking moment capacity when KKODE(I) set equal to 1.

COMMON/D314/

These variables are used in SUBROUTINES LPCODE and INTPRG and are defined in the subroutine descriptions.



APPENDIX C

DESCRIPTION OF SUBROUTINES USED IN DBOXSS AND DBOXDS

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The primary differences between programs DBOXSS and DBOXDS occur in the MAIN programs. The logic and storage requirements of DBOXDS are more involved than those in DBOXSS because it incorporates an integer programming solution (subroutine INTPRG) to obtain final designs, working from an approximate L.P. solution generated in subroutine LPCODE. During the project, two separate programs evolved as a natural result of our efforts to minimize programming complexities by dealing with each problem separately. The majority of the computer core storage requirements (with programs in object form) arise from the large arrays used in LPCODE and INTPRG. By dividing the programs, it was possible to overlay arrays in a straightforward way (see the shifts in variable names appearing in COMMON/D314/) and thus reduce the total storage needed by DBOXDS. The reader will find an almost complete correspondence between variable names used in the two programs, since DBOXDS was constructed from a reproduced version of DBOXSS. The flowcharts of logic for the MAIN programs are presented together, with branches indicated for each program. Variable names common to both programs as well as those used in only one or the other of the programs appear together in the definition of variables. The subroutines are for the most part self-contained and differences in them occur in the way in which calling parameters are formed. Each subroutine is described in a separate section, listing its function, the definition of variables which it uses and a macro level flowcharts of logic when necessary for understanding its operation.

MAIN PROGRAM

Function

The main program reads and checks input data, computes the quantities required by subroutines, iterates on prestress loss until an acceptable design is obtained and outputs the final design.

Variable Definitions

- WIDTH - overall bridge width, used to compute AASHTO lateral distribution factor (ft).
- JTNTL - number of traffic lanes, used in computation of AASHTO lateral distribution factor.
- TNLB - number of longitudinal beams, used in computing AASHTO lateral distribution factor.
- ZAXLE(I) - contains axles loads of designated AASHTO truck loading (kips).
- ZNWHL(I) - contains axle spacings of designated AASHTO truck (ft).
- NAXLE - number of axles in designated AASHTO truck.
- ULOAD - lane load for designated AASHTO loading (k/ft).
- CSLOAD - concentrated force used in computation of live load shear from designated AASHTO lane loading.
- CMLOAD - concentrated force used in computation of live load moment from designated AASHTO lane loading.
- ZIMP - live load impact factor.
- STRESS(I,J) - contains stresses in beam due to all sources, for final design. J=1, release stress top; J=2, release stress bottom; J=3, service stress top; J=4, service stress bottom. I runs from 1 to 6 and for release stresses correspond to the

following points; end, $L/20$, $L/10$, $3L/20$, $2L/10$ and $1/4$ point.

For service service stresses, I corresponds to end, $L/10$,

$2L/10$, $3L/10$, $4L/10$ and midspan. Tension stresses are negative and compression stresses are positive.

KSYM(I,J) - array containing the symbol "x" to be printed with those stresses (either release or service) which are at their allowable value ($I=1, \dots, 6$; $J=1, \dots, 4$). Column 5 contains "x" for behavior constraints which are binding on the final design. The constraints are: minimum concrete strength ($I=2$), ultimate moment capacity ($I=4$), minimum initial camber ($I=5$), maximum concrete strength ($I=1$), cracking moment capacity ($I=3$), maximum initial camber ($I=6$).

NSTRMX(I) - the maximum number of strands permitted in strand row I , stored in integer form.

NWRAP(I) - contains strand wrapping information used in output.

STRSP(I) - stirrups spacings (in). $I=1$, end; $I=2$, $L/10$; $I=3$, $2L/10$; $I=4$, quarter point; $I=5$, $3L/10$; $I=6$, $4L/10$; $I=7$, midspan.

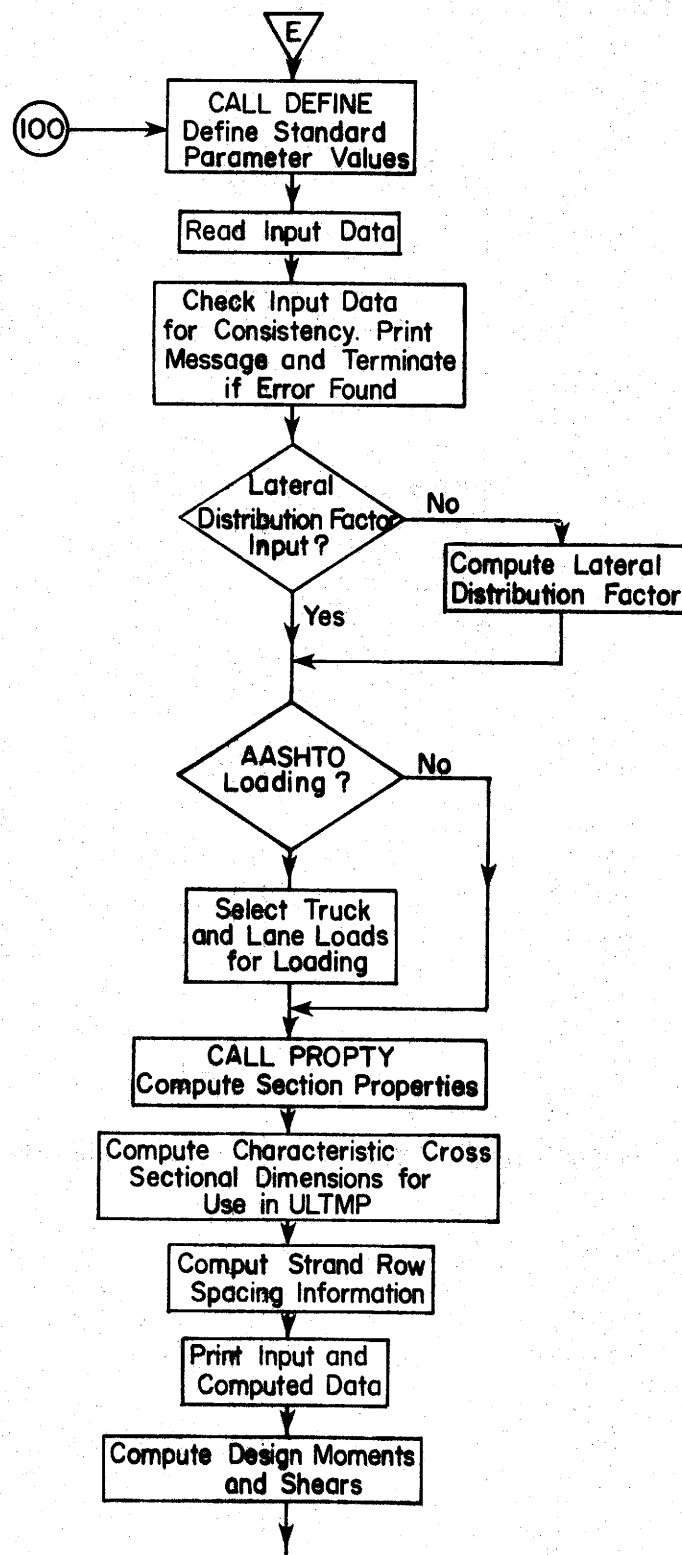


FIGURE C1. Flow Chart for Main Program

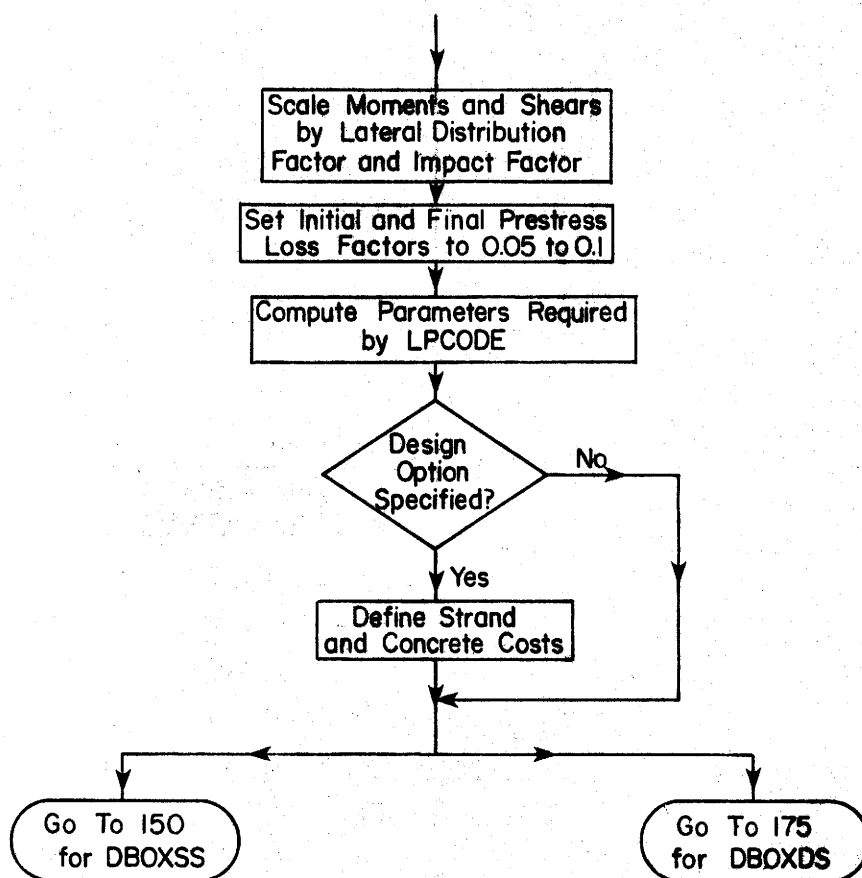


FIGURE C1. (continued)

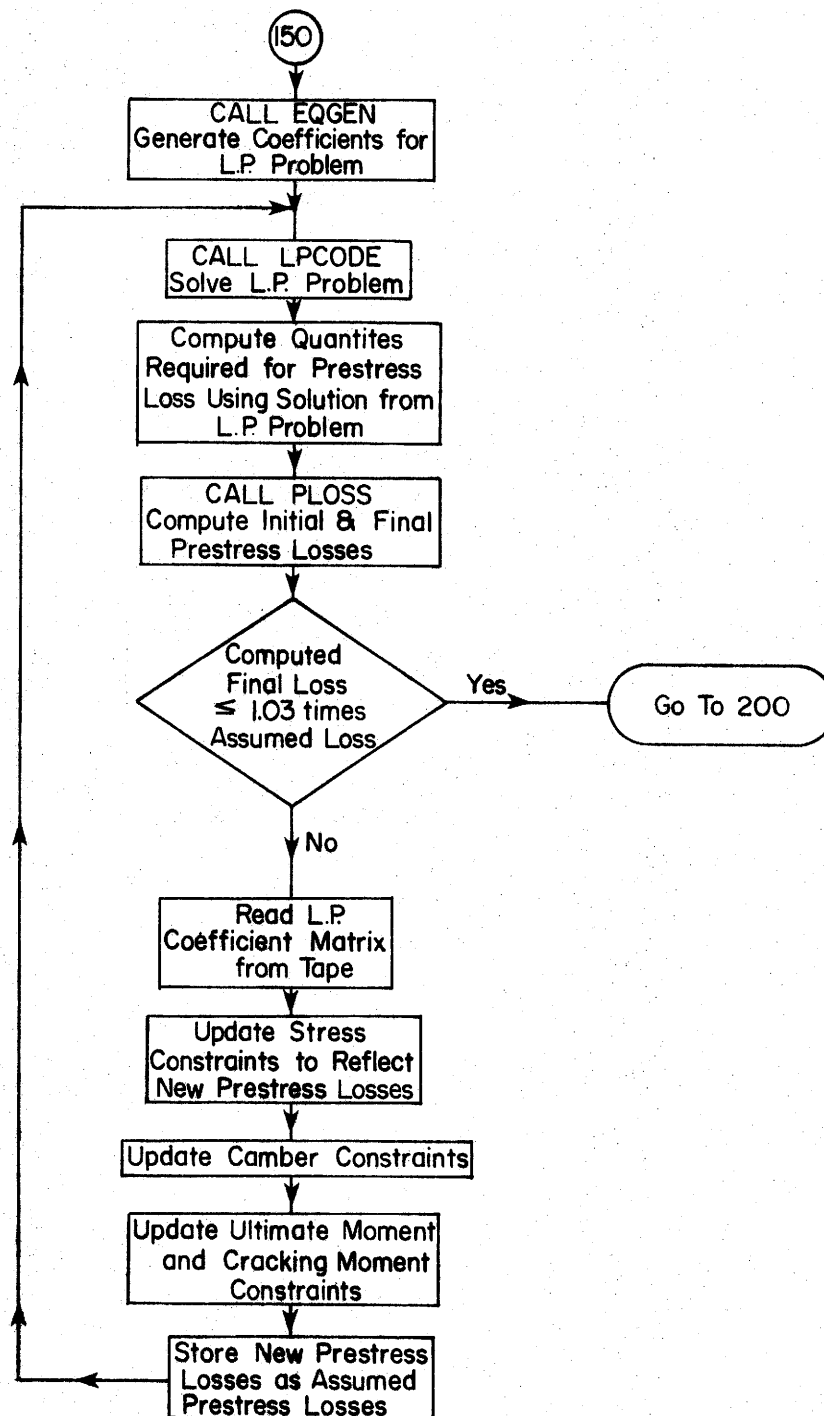


FIGURE C1. (continued)

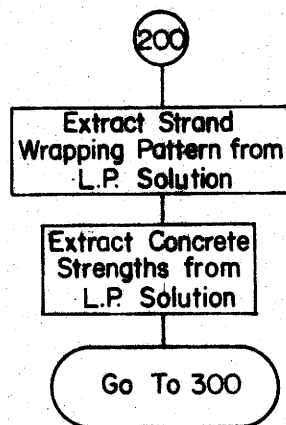


FIGURE C1. (continued)

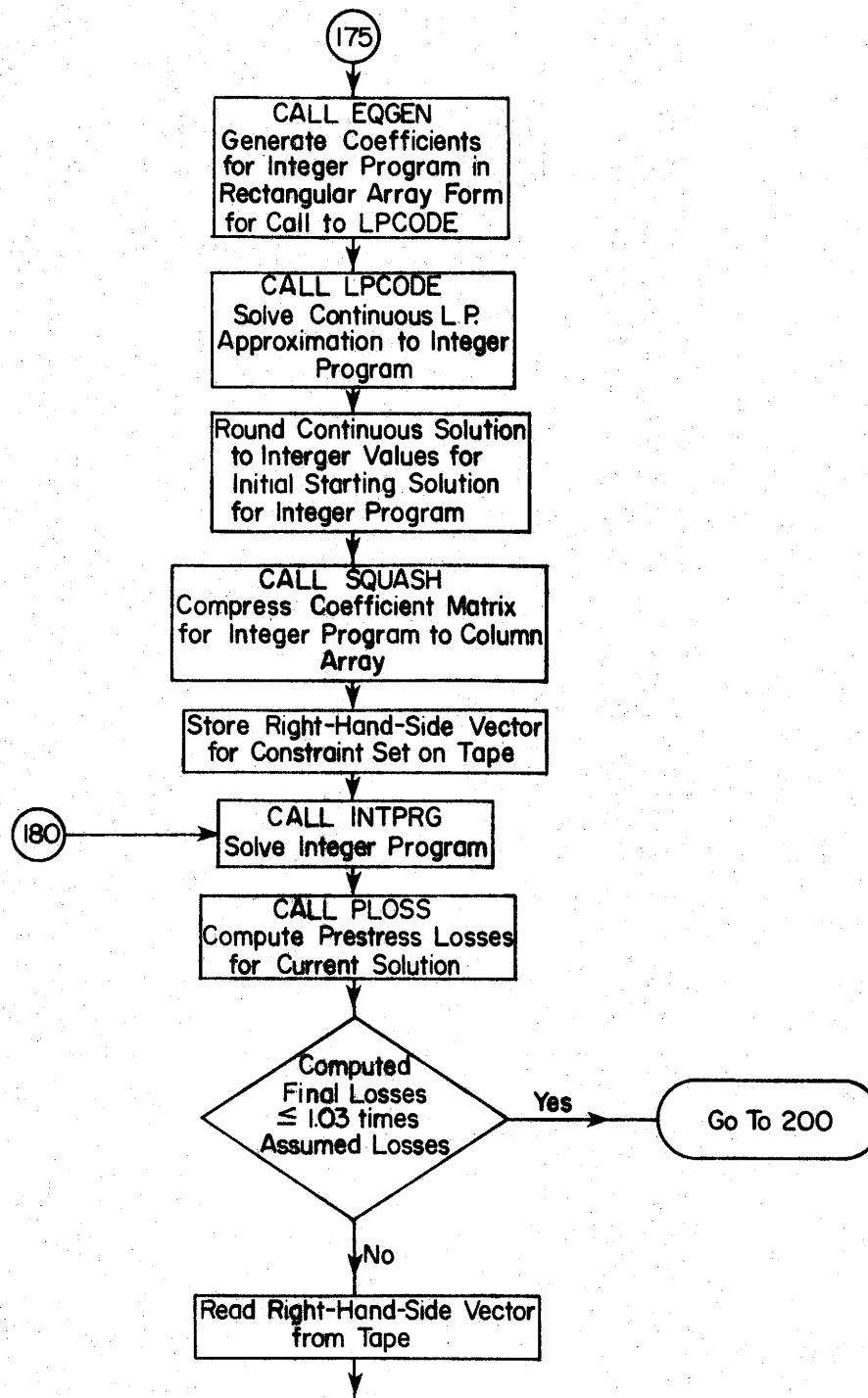


FIGURE C1. (continued)

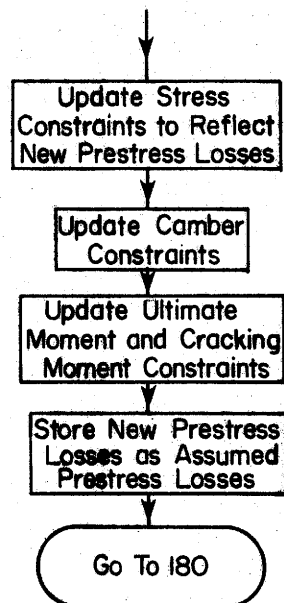


FIGURE C1. (continued)

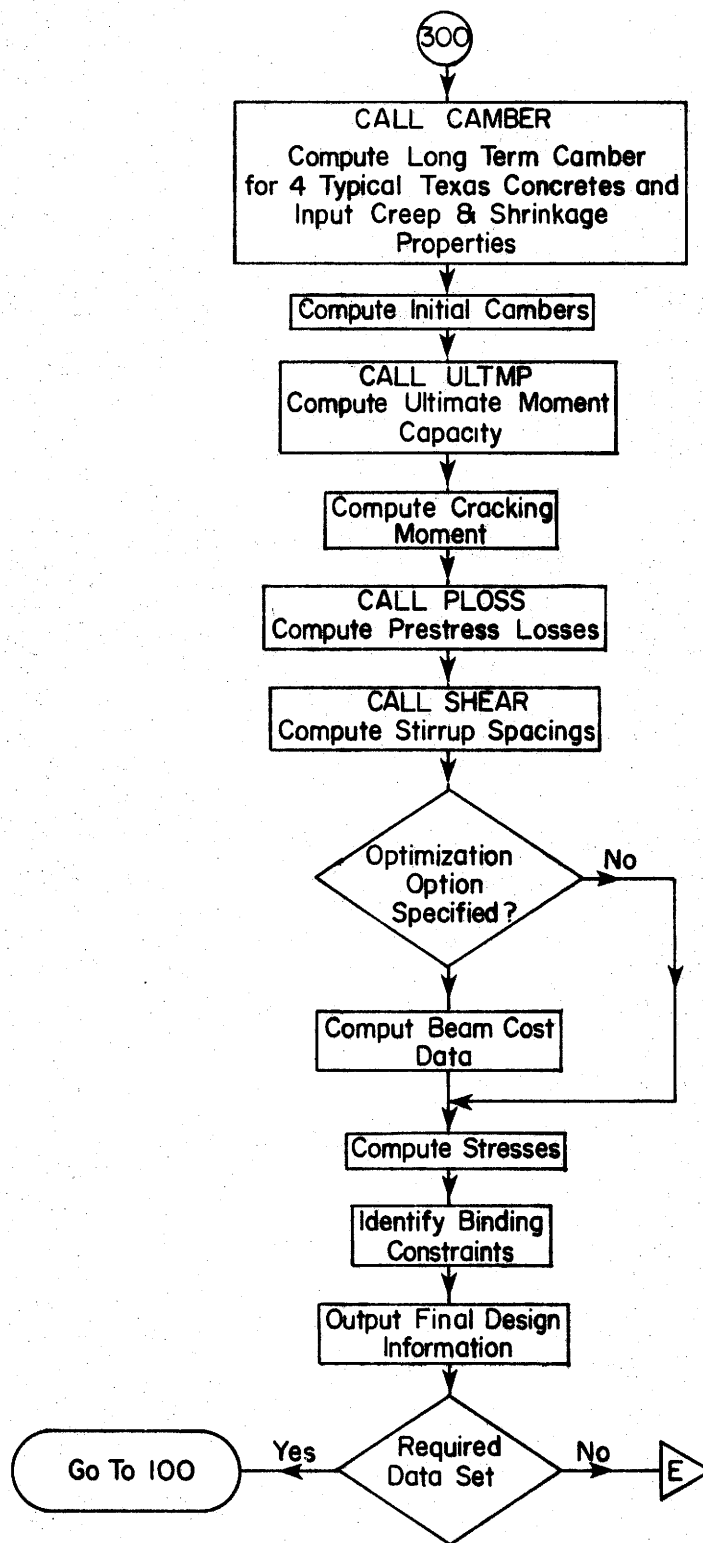


FIGURE C1. (continued)

SUBROUTINE DEFINE

Subroutine Function

This subroutine defines standard parameter values used in the program. The subroutine is called before input data are read, for each data set processed. Thus, only non-standard values of input parameters need be entered on the program input forms. The parameters initialized are: unit weight of concrete, average relative humidity, ultimate strength of strand, proportional limit of strand, allowable stress coefficients, unit creep and shrinkage constants for the concrete, ratio of modulus of elasticity of compression reinforcing to that of the concrete, yield strength of stirrup and compression reinforcing and area of stirrups.

Definition of Variables

The variables used in this subroutine appear in COMMON/DEFINE/ and are described in Appendix B.

SUBROUTINE EQGEN

Subroutine Function

This subroutine name appears in both DBOXSS and DBOXDS. Its function in both programs is to generate the coefficients defining the objective function and constraint set for the programming problem. The coefficients used are contained in the equations of Chapter II for DBOXSS and Chapter III for DBOXDS.

Additional Considerations - DBOXDS

In addition to the basic function defined above, two other actions are contained in the version in DBOXDS. The coefficient matrix is the same, regardless of whether the problem is to be solved as a continuous linear programming (L.P.) problem or an integer program. DBOXDS uses an L.P. solution as a starting point for the solution to the integer program. To insure that the starting point is realistic, it is necessary to place an upper bound of 1.0 on binary variables (which take values of either 0 or 1) which appear in the integer formulation. This is done at the end of EQGEN by generating an additional set of upper bound constraints which are activated only when the coefficient matrix is passed to subroutine LPCODE. The coefficient matrix is destroyed during the solution of the linear program in LPCODE. Thus, to preserve the matrix for later iterations on prestress loss, it is necessary to store it on scratch tape (unit 3). This is done as the last step in the subroutine.

Variable Definitions

The coefficient matrix (which includes the objective function and right-hand-side vector as well as the constraints) is placed in the

variable name ARRAY(I,J). Other variable names which appear in this subroutine are defined either in the labeled commons (Appendix B) or in subroutine LPCODE and INTPRG.

SUBROUTINE LPCODE (NFRCE, NEQS, INDX, KODE)

Subroutine Function

This subroutine solves the linear program defined by Eqs. using the simplex method. It calls subroutine PIVOT to perform tableaux transformations. The principle variables used are those contained in COMMON/D314/ which are defined below.

Additional Considerations - DBOXSS

The coefficient matrix is destroyed during the solution process. To preserve it for later use in iterations on prestress loss (and thus save the computational effort of recomputing it each time) the matrix is stored on scratch tape (unit 4). This is performed immediately after entry to the subroutine.

Variable Definitions

NFRCE - number of unrestricted variables (those which may assume negative values). This parameter must be zero for the version of LPCODE incorporated in the program.

NEQS - number of equality constraints. Must be zero in this program.

INDX - parameter indicating whether the primal (INDX=0) or dual (INDX=1) problem is to be solved. Must be zero in this program.

KODE - code indicating whether this is the first call to LPCODE (KODE=0). On the first call, slack variables are added to the coefficient matrix and it is stored on tape unit 4.

N - the number of constraints plus 1.

M - the number of variables.

A(I,J) - the coefficient matrix.

B(I),XD(I) - arrays used as flags during computation.

X(I) - on return, contains the problem solution.

OBJ - contains the objective function value.

SUBROUTINE INTPRG

Subroutine Function

This subroutine uses a heuristic algorithm to solve the integer programming problem. It calls subroutine PHASE1, PHASE2 and PHASE3. It is used only in DBOXDS. To improve computational efficiency, it works with a compressed version of the coefficient matrix which is obtained by deleting zero entries and "stacking" the columns of this matrix in the singly subscripted variable Y(I). In this form, several additional arrays (ROW(I) and COL(I)) are required to keep track of indexing. The compression of the coefficient matrix into a column vector is carried out in subroutine SQUASH.

Definition of Variables

- N1 - number of continuous variables in the problem.
- N2 - number of integer variables in the problem.
- N3 - number of binary (0 or 1) variables in the problem.
- TR - a tolerance on constraints. When the difference between the left and right side of an inequality is less than TR in absolute value, the constraint is assumed active.
- TV - a tolerance on variables. When any variable takes a value less than TV, it is assigned the value zero.
- NR - number of constraints in the problem.
- DXMAX - the maximum amount by which any variable is incremented in testing for a potential solution point.
- IT(I) - array containing the number of iterations used in each of the four phases of the algorithm.
- X(I) - contains the values of the variables.

- Y(I) - contains the nonzero terms of the coefficient matrix in "stacked" form.
- ROW(I) - contains the row in the coefficient matrix from which the Ith element of Y(I) was taken.
- COL(I) - contains the element number of Y(J) where the first nonzero element from row I of the coefficient matrix is stored.
- BB(I) - scratch storage used to accumulate the value of each constraint.
- C(I) - contains the coefficients of the objective function.
- B(I) - contains the right-hand-side vector for the constraints.
- XX(I) - scratch storage space.

SUBROUTINE PROPTY

Subroutine Function

This subroutine calculates the location of the centroid of the cross section, the moment of inertia with respect to the centroidal axis of bending, and the section moduli for the top and bottom of the cross section. Each of the computed quantities previously mentioned is calculated for the cross section with and without the shear key. If the dimension A is greater than or equal to B, the area of the shear key is taken as zero.

Definition of Variables

A, B, C, C1, C2, D, E, F, G, H, T, WD, WH, XDIM, YDIM - See Figure C2

AREA - area of cross section without shear key and with compression steel replaced with an equivalent area of concrete obtained by multiplying the area of steel by $(\text{RATNOD}-1)(\text{in.}^2)$.

AREAK - area of cross section with shear key and with compression steel transformed using $(2 \cdot \text{RATNOD}-1)(\text{in.}^2)$.

YB - distance from c.g. of beam to bottom of beam (in.).

YBK - distance from c.g. of beam to bottom of beam, with shear key (in.).

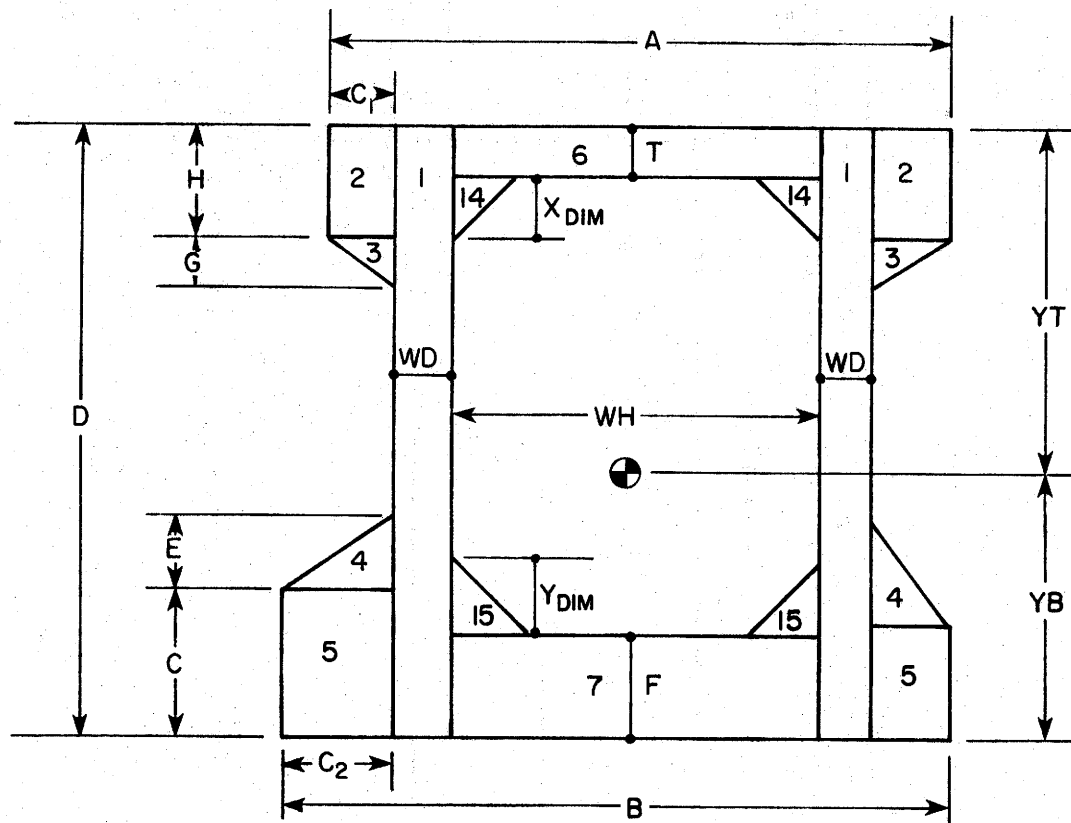
Y1 thru Y15 - distance from bottom of the cross section to the c.g. of the areas 1 thru 15 shown in Figure C2 (in.).

I1 thru I15 - moment of inertia of areas 1 thru 15 shown in Figure C2 (in.^4).

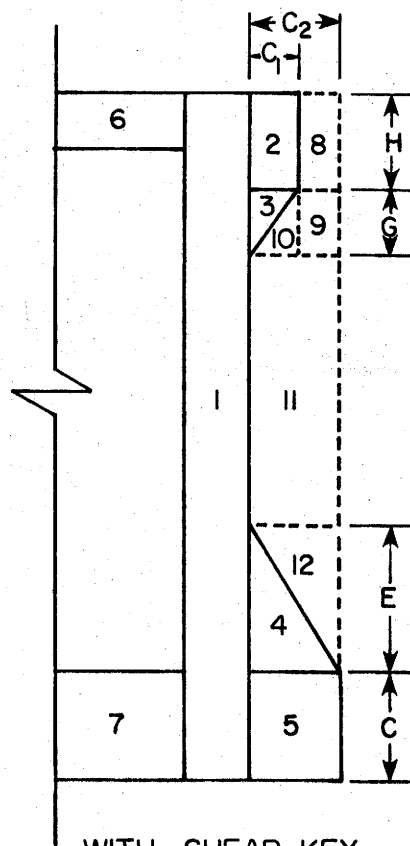
JVKEY = 1 - shear key omitted.

JVKEY = 2 - shear key included.

ZT - section modulus at top excluding the shear key (in.³).
ZB - section modulus at bottom excluding the shear key (in.³).
ZTK - same as ZT except including shear key (in.³).
ZBK - same as ZB except including shear key (in.³).
APRIME - area of compression steel in the top flange (in.²).
RATNOD - modular ratio.



WITHOUT SHEAR KEY



WITH SHEAR KEY

FIGURE C2. Component Areas Used In Computing Cross Sectional Properties

SUBROUTINE MOMSHR (DL, NWHL, NWHEEL, XSEC, PAXLE, MAXMOM, MAXSHR)

Subroutine Function

This subroutine determines the maximum bending moment and shear force at a point on a simply supported beam due to a series of moving concentrated forces.

Definition of Variables

- DL - length of span (ft).
- NWHL(I) - distance from concentrated force 1 to concentrated force I (ft).
- NWHEEL - number of concentrated forces in moving force pattern.
- XSEC* - distance from left end of beam to point where maximum effects to be computed as load pattern moves from right to left (ft).
- PAXLE(I) - weight of Ith concentrated force (kips).
- MAXMOM - on return, contains maximum bending moment at point of interest (kip-ft).
- MAXSHR - on return, contains maximum shear force at point of interest (kips).
- NS - shift number, which is the number of the force located over the point of interest on the beam.
- NST - number of intervals between concentrated forces.
- IPL(I) - array indicating which concentrated force is located over the section of interest for shift number I.
- IPR(I)* - array indicating which concentrated force is located nearest the right end of the beam and still on the beam for shift I, as load pattern moves from right to left.

- D2* - distance from any concentrated force to the right end of the beam for load pattern moving right to left (ft).
- REACT(I)* - left reaction force for shift I, as load pattern moves from right to left (kips).
- DM* - distance from point of interest on beam to any concentrated force to the left of the point and still on the beam when load pattern moves from right to left (ft).
- SHEAR(I) - shear force at point of interest on beam for Ith shift (kips).
- MOMENT(I) - bending moment at point of interest on beam for Ith shift (kip-ft).
- II - a trigger used to indicate direction of movement of load pattern. When II = 1, load pattern moves right to left and when II = 2, movement is left to right.

* to obtain correct interpretation of the variable when the load pattern moves from left to right, substitute the word right for left and left-to-right for right-to-left.

SUBROUTINE LOCATE (DL, XSEC, NST, NWHL)

Subroutine Function

This subroutine calculates the number of the concentrated force in the moving load pattern located at the point of interest on the beam and the number of the concentrated force on the beam nearest the end of the beam from which the load pattern is moving.

Definition of Variables

The variables used in this subroutine are defined in the description of subroutine MOMSHR.

SUBROUTINE ULTMP (ASTAR, FPCBM, FPS, ASPRM, FPL, D, DPTH, FSY, DCR
Y1, Y2, Y3, Y4, Z1, Z2, Z3, Z4, ZMUL)

Subroutine Function

This subroutine computes the ultimate moment capacity of the section. Two cases are considered: the neutral axis in the slab and the neutral axis below the slab. The methodology used to compute moment capacity was developed in Chapter II. This subroutine calls function subprogram BRACK.

Definition of Variables

ASTAR - total area of prestressing strands (in^2).

FPCBM - 28 day concrete strength (ksi).

FPS - ultimate strength of strand (ksi).

ASPRM - total area of compression steel reinforcing present in slab (in^2).

FPL - proportional limit stress for strand material (ksi).

D - distance from top of section to c.g. of strands (in).

DPTH - depth of section (in).

FSY - yield strength of compression reinforcing steel (ksi).

DCR - distance from top of section to c.g. of compression steel (in).

Y1,Y2,Y3,Y4,

Z1,Z2,Z3,Z4 - dimensions used to compute area and c.g. of concrete compression zone (see Figure C3).

ZMUL - on return, contains the ultimate moment capacity (k-ft).

CLONG - fraction of prestress force lost due to elastic and inelastic effects.

ESINI - average strain in the strands after all losses (in/in).

BEFF - width of top of section (in).

THK - thickness of top slab (in).

CC - total compression force over concrete compression zone (kips).

T - total tension force in strands (kips).

X - distance from top of beam to neutral axis (in).

ES - average strain in strands (in/in).

ESP - average strain in compression reinforcing (in/in).

CS - total compression force in compression reinforcing.

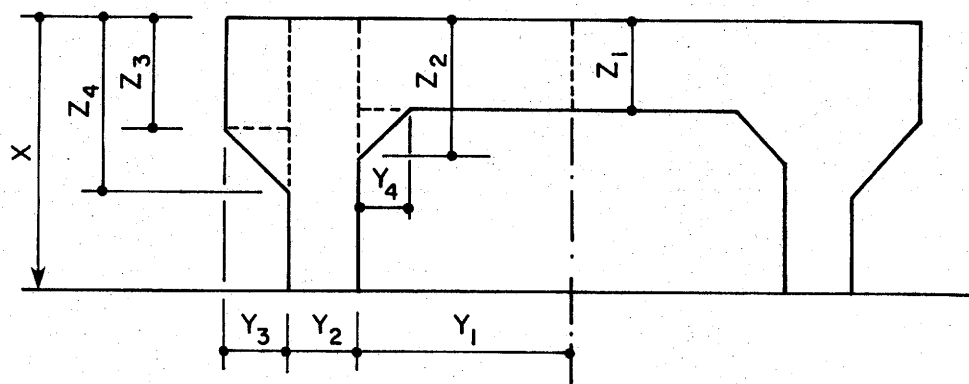


FIGURE C3. Dimensions of Cross Section Used
in Computing Concrete Compression Zone

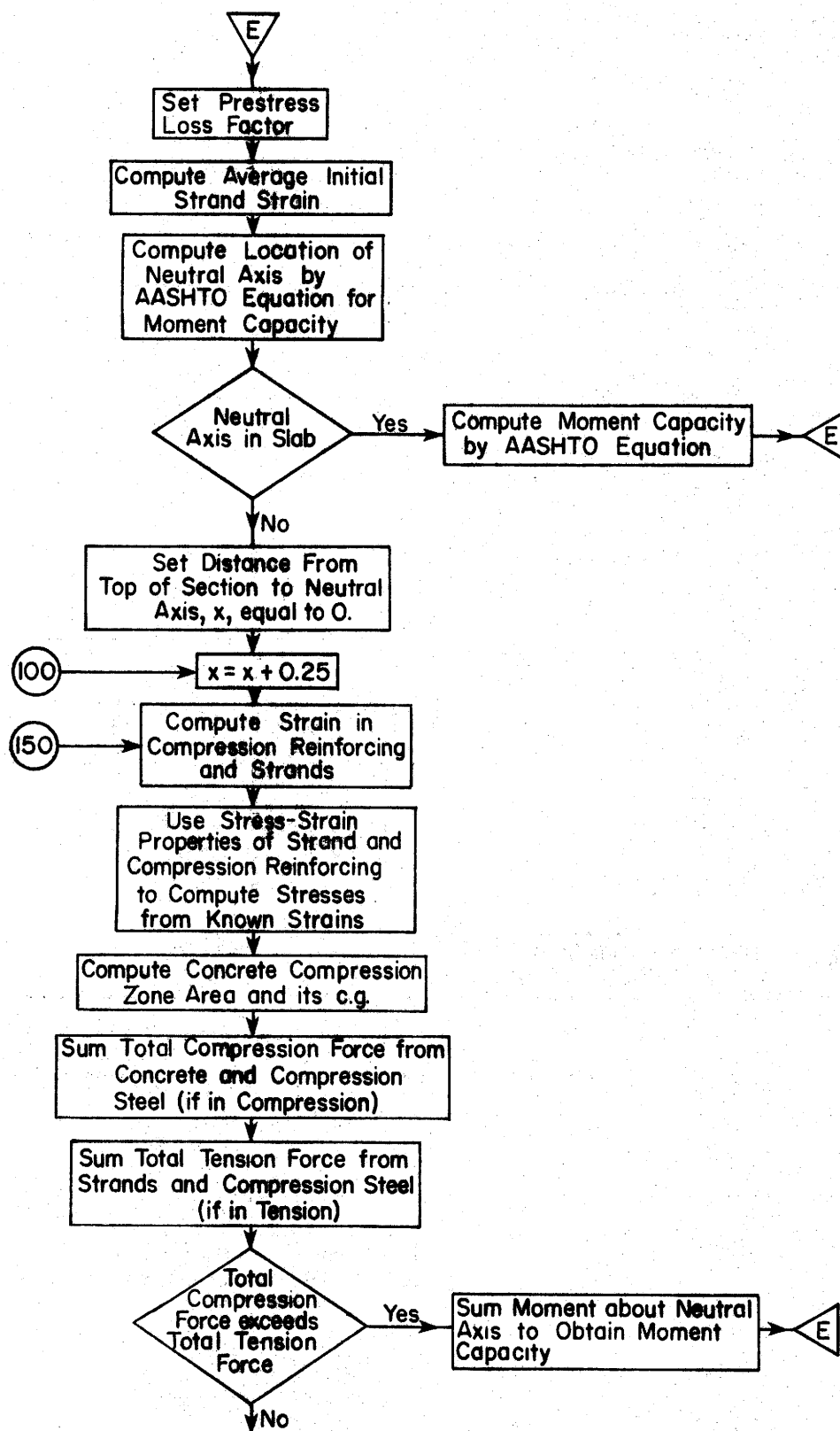


FIGURE C4. Flow Chart for Subroutine ULTMP

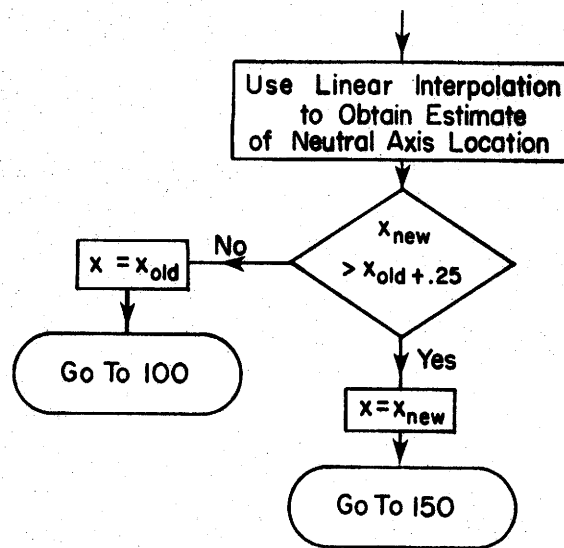


FIGURE C4. (continued)

SUBROUTINE PLOSS (FPCR, ZMBW, ZMC, ZMNC, FSU, AS, AB, ZI, ZIC,
YB, YBC, EC, HUM, SPAN, ZLOSS, ZINLOS, UWC)

Subroutine Function

This subroutine computes the fraction of initial strand stress lost immediately after release and when all creep, shrinkage and strand relaxation losses have occurred. The losses are computed using the expressions presented in Chapter II.

Definitions of Variables

- FPCR - concrete strength at release (ksi).
- ZMBW - moment at midspan due to beam weight (k-ft).
- ZMC - other dead load moment at midspan acting on composite section (beam plus shear key) (k-ft).
- ZMNC - other dead load moment at midspan acting on non-composite section (beam without shear key) (k-ft).
- FSU - ultimate strength of strand (ksi).
- AS - total area of strands (in.²).
- AB - cross sectional area of beam (in.²).
- ZI - moment of inertia of beam cross section (in.⁴).
- ZIC - moment of inertia of composite section (in.⁴).
- YB - distance from c.g. of beam to bottom of beam (in.).
- YBC - distance from c.g. of composite section to bottom of beam (in.).
- EC - distance from bottom of beam to c.g. of strands (in.).
- HUM - average relative humidity present during life of beam (%).
- SPAN - span length (ft).
- ZINLOS - fraction of initial strand stress (.7 FSU) lost at release.

ZLOSS - fraction of initial strand stress lost under service load conditions.

UWC - unit weight of beam concrete (k/ft^3).

SH - prestress loss due to shrinkage of concrete (ksi).

ES - prestress loss due to elastic shortening (ksi).

CRC - prestress loss due to creep of concrete (ksi).

CRS - prestress loss due to strand relaxation (ksi).

ECI - modulus of elasticity of concrete at release (ksi).

Computed from ACI equation.

SUBROUTINE CAMBER (ES, EC, ASTRN, STRNS, UWB, AREA, SPANL, ECCL,
IB, FO, ENDECC, PRLMAX, CBRMAX, HDPT)

Subroutine Function

This subroutine computes midspan camber under dead load due to elastic and inelastic (creep and shrinkage) behavior of the concrete. The method is developed in reference (2). Upward camber is positive.

Definitions of Variables

ACR - unit creep at time infinity (in./in./psi).
ASH - shrinkage at time infinity (in./in.).
AST - total area of prestressing strands (in.²).
BCR - time at which one-half ACR is reached (days).
BSH - time at which one-half ASH is reached (days).
ES - modulus of elasticity of steel (10⁶ psi).
EC - modulus of elasticity of concrete (10⁶ psi).
ASTRN - area of a single prestressing strand (in.²).
STRNS - total number of prestressing strands.
UWB - unit weight of beam concrete (lbs./ft³).
AREA - cross-sectional area of beam (in.²).
SPANL - span length (ft).
ECCL - eccentricity of the strand pattern at the center
line of the beam measured from the cg (in.).
IB - beam moment of inertia (in.⁴).
FO - total initial prestressing force (lbs).
ENDECC - eccentricity of the strand pattern at the ends of
the beam measured from the cg (in.).

PRLMAX - total prestress loss at time infinity (%).

CBRMAX - total camber at time infinity (in.).

HDPT - distance from the centerline of the beam to the
hold-down point (ft).

CNST(i,j) - creep and shrinkage coefficients based upon Dallas,
Odessa, San Antonio, and Lufkin concrete properties.

SUBROUTINE SHEAR (B, DEPTH, D, FPC, FSY, AREA, VU, SPACE)

Subroutine Function

This subroutine computes the stirrup spacing requirements at selected sections of the beam according to The American Association of State Highway Officials Specifications, 1973.

Definitions of Variables

- B - width of a web of the beam cross-section (in.).
- DEPTH - depth of the beam (in.).
- D - distance from extreme compressive fiber to the centroid of the prestressing force (in.).
- FPC - compressive strength of concrete at 28 days (ksi).
- FSY - yield strength of non-prestressed conventional reinforcement in compression (ksi).
- AREA - area of web reinforcement (in.²).
- VU - shear due to ultimate load and effect of prestressing (kips).
- SPACE - longitudinal spacing of the web reinforcement (in.).
- AV - total area of web reinforcement (in.²).
- RJ - ratio of distance between centroid of compression and centroid of tension to the depth D.
- VC - shear carried by the concrete (kips).

APPENDIX D

DESCRIPTION OF MODIFICATIONS TO
GHOSE'S MULTIBEAM BRIDGE ANALYSIS PROGRAM

Chapter VI described briefly the computer program by Ghose (9) for the analysis of multibeam bridges and the modified version, AMBB, developed for the calculation of lateral distribution factors for axle train and standard AASHTO loadings. This Appendix describes the modifications to the program written by Ghose.

The modifications consist of changes in program input and output and three added subroutines, INPUT, OUTPUT, and INFLN. The appropriate input routine is called by the main program, based on the entry in column 62 of the first card in the data deck (see Figure 35). For input to the original program, subroutine INPTT is called. If a 1 is found in column 62, control is transferred to subroutine INPUT for reading of the data on the form in Figure 35. Output is handled in the same way, with subroutine OUTPTT being used when a blank is encountered in column 62 of the first card and subroutine OUTPUT utilized for output when the program is to be used to compute lateral distribution factors. The longitudinal position of vehicle axles (either axle train or AASHTO truck) is first computed. Next, the influence lines for midspan moment in each beam is computed by moving a single line of wheels transversely across the bridge in one foot moves. Finally, the maximum moment in each beam and the corresponding position of axle train, truck or lane loading is computed from the influence lines. Loadings for producing the influence lines are assembled in subroutine INPUT, and maximum moments and vehicle positions are computed in subroutine INFLN.

The variables listed below are used in the modifications to the original program.

ZPAN - span length (ft.).
 E - modulus of elasticity (ksi).
 NBEAMS - number of beams in bridge.
 NTRFL - number of traffic lanes.
 KAXT - trigger: if 1, axle train loading input; if not equal to 1,
 no axle train considered.
 NAXTSP - lateral spacing of wheel lines in axle train (ft. - an integer
 number).
 NAXCL - side clearance of axle train vehicle. The distance between
 the wheel line and outside of the vehicle. The wheel line
 may be no closer than NAXCL feet from the edge of a traffic
 lane.
 NAXT - number of axle trains that can simultaneously be located
 transversely on the bridge ($NAXT \leq NTRFL$).
 YMI(I) - moment of inertia about y-axis (see Figure 31) of beam type I
 (in^4).
 ZMI(I) - moment of inertia about z-axis of beam type I (in^4).
 BMA(I) - cross sectional area of beam type I (in^2).
 BMJ(I) - torsional stiffness (polar moment of inertia) of beam type I
 (see Eq. 164), (in^4).
 YH(I,1) - distance, parallel to y-axis, between c.g. of beam type I and
 left hinge (HL on input form), (in).
 YH(I,2) - distance, parallel to y-axis, between c.g. of beam type I
 and right hinge (HR on input form), (in).

ZH(I,1) - distance, parallel to z-axis, between c.g. of beam type I and left hinge (VL on input form), (in).
 ZH(I,2) - distance, parallel to z-axis, between c.g. of beam type I and right hinge (VR on input form), (in).
 NTYPES - number of different beam types.
 NTY(I) - contains beam type number for beam I.
 HINGTP(I,J) - for hinge type I: contains "Y" in J = 1 if hinge transmits longitudinal shear force or "N" if it does not; contains "Y" in J = 2 if hinge transmits vertical shear (shear in z-direction, Figure 31), "N" if not; contains "Y" in J = 3 if hinge transmits transverse force (y-direction Figure 31), "N" if not; contains "Y" in J = 4 if hinge transmits transverse moment (about x-axis Figure 31), "N" if not.
 JTYPES - number of different hinge types.
 JTY(I) - contains hinge type number for hinge I.
 TLN(I,J) - contains distance between c.g. of beam 1 and left edge of traffic lane I in J = 1. Distance between c.g. of beam 1 and right edge of traffic lane I in J = 2. Distances are positive to the right of beam 1 c.g. and negative to left (ft).
 NWHEEL - number of axles in axle train.
 PWHEEL(I) - weight of axle I in axle train (kips).
 ZNWHL(I) - distance between axle 1 and axle I (ft).

- KASAST(I,J) - for traffic lane I, J = 1 contains load case number where AASHTO truck is as close to left edge of traffic lane I as side clearance (2 ft.) permits. For J = 2, contains load case number for truck as close as possible to right edge of traffic lane I.
- KASASL(I,J) - for traffic lane I, J = 1 contains load case number where 10 ft. wide lane load is positioned at left edge of lane. J = 2 contains load case number where lane load is at right edge of traffic lane I.
- KASAXT(I,J) - same as KASAST(I,J) but for axle train whose side clearance is NAXCL.
- FULMAT - maximum moment at midspan due to single AASHTO truck (k-in).
- FULMAL - maximum midspan moment due to full AASHTO lane load (k-in).
- FULMAX - maximum moment at midspan due to full axle train (k-in).
- ZIMP - impact factor.
- ZMAST(I,J) - maximum midspan moment, beam I produced by AASHTO truck positioned in lane J (k-in).
- ZMASL(I,J) - maximum midspan moment, beam I produced by AASHTO lane load in lane J (k-in).
- ZMAXT(I,J) - maximum midspan moment, beam I, produced by axle train in Lane J (k-in).
- ZMMAST(I) - maximum moment at midspan of beam I due to AASHTO trucks in one or more lanes, using AASHTO lane reduction factors (k-in).

ZMMASL(I) - same as ZMMASL(I), but for lane loads applied instead of AASHTO trucks (k-in).

ZMMAXT(I) - maximum midspan moment for beam I by placing from one up to NAXT axle trains simultaneously in the various traffic lanes. AASHTO lane reduction factors are not used (k-in).

POSAT(I,J) - contains the position of left and right wheel lines of AASHTO truck which produces maximum moment in beam I. J = 1 contains distance from c.g. of beam 1 to left wheel line for truck in lane 1. J = 2 contains distance to right wheel line of truck in lane 1. J = 3 contains distance from c.g. of beam 1 to left wheel line for truck in lane 2. J = 4 contains distance to right wheel line, etc. (ft).

POSLN(I,J) - see POSAT(I,J), but for AASHTO lane loading instead of AASHTO truck (ft).

POSAX(I,J) - see POSAT(I,J), but for axle train vehicle (ft).

NLLAST(I,J) - coded array: if NLLAST(I,J) \neq 0, then lane J is loaded when maximum moment at midspan of beam I occurs under AASHTO truck loadings.

NLLALN(I,J) - same as NLLAST(I,J), but for AASHTO lane loading.

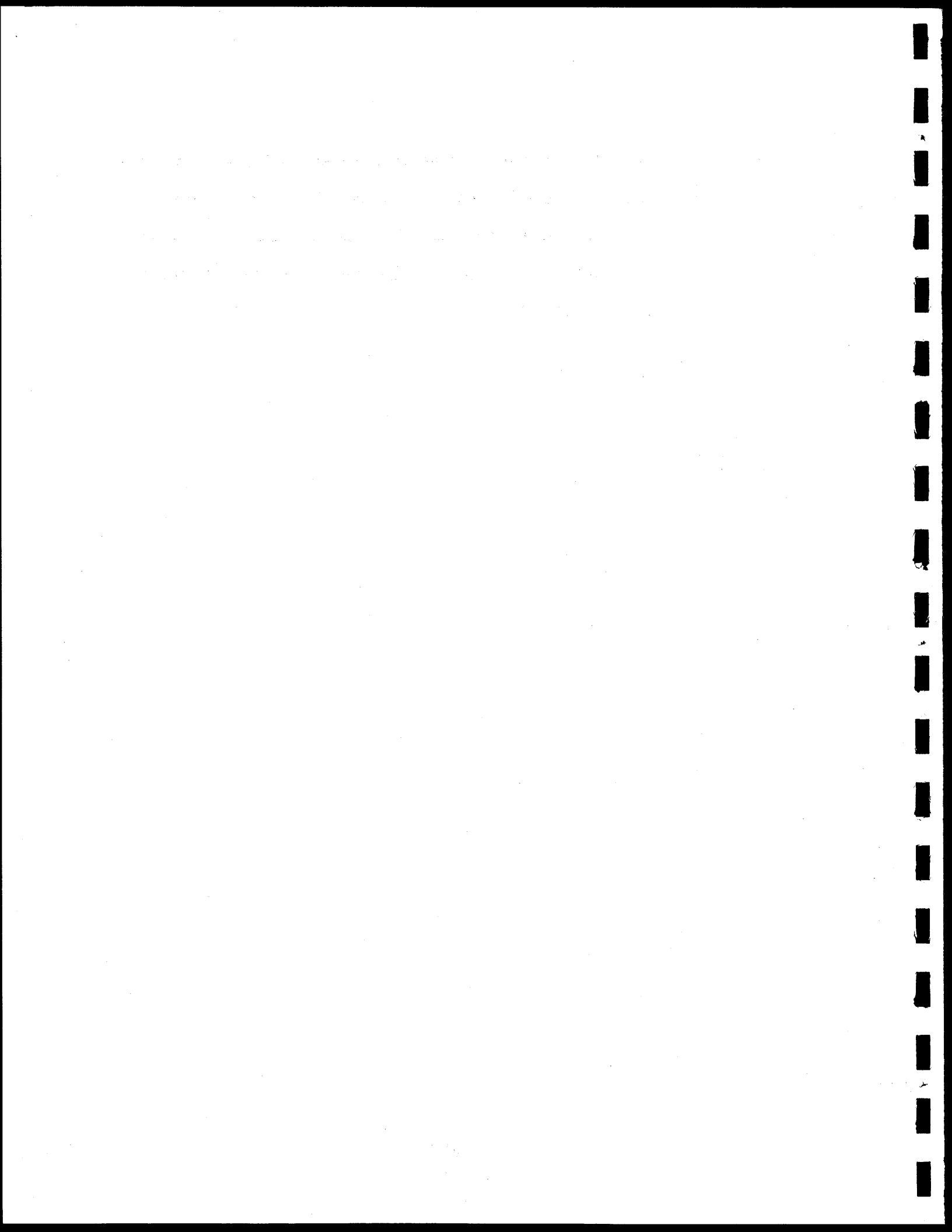
NLLAXT(I,J) - same as NLLAST(I,J), but for axle train vehicle.

DISTAT(I) - fraction of full AASHTO truck applied to single beam I which would produce same moment as that recorded in ZMMASL(I).

DISTAL(I) - same as DISTAT(I), but for AASHTO lane loading.

DISTAX(I) - same as DISTAT(I), but for axle train vehicle.

YMOUT(I,J,K) - moment about y-axis in beam J, under load case I, for Kth x-coordinate position along beam. This variable occurs in Ghose's original program. When used in added subroutines described here, only $K = 1$ (which corresponds to midspan) is used (k-in).



APPENDIX E
PROGRAM LISTINGS

DBOXSS.	215
DBOXDS.	268
AMBB.	335

	PAXLE(I)=0.	SS000410
	NWHL(I)=0.	SS000420
927	CONTINUE	SS000430
	IF(JLOAD.NE.1)GO TO 861	SS000440
	READ(5,502)(PAXLE(I),I=1,18)	SS000450
502	FORMAT(3X,18(F3.1,1X))	SS000460
	DO 503 N=1,18	SS000470
	IF(PAXLE(N).NE.0.)NWHEEL=N	SS000480
503	CONTINUE	SS000490
	READ(5,505)(NWHL(I),I=1,17)	SS000500
505	FORMAT(7X,17(F3.0,1X))	SS000510
C	CONCENTRATED FORCES APPLIED TO SINGLE BEAM	SS000520
861	DO 590 I=1,10	SS000530
	FCONC(I)=0.	SS000540
	DCONC(I)=0.	SS000550
590	CONTINUE	SS000560
	IF(JCONC.NE.1) GO TO 925	SS000570
	READ(5,591)(FCONC(I),I=1,10)	SS000580
	READ(5,591)(DCONC(I),I=1,10)	SS000590
591	FORMAT(3X,10(F5.2,1X))	SS000600
C	BEAM DIMENSIONS	SS000610
925	READ(5,907)ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,WHDIM,TDIM,WDDISS000620	
	*M,XDIM,YDIM	SS000630
907	FORMAT(3X,13(F4.2,1X))	SS000640
C	GENERAL INFORMATION	SS000650
	READ(5,908) ZL,WIDTH,JTNLT, TNLB,APRIME,DCR,ALDEF,DEFMIN,	SS000660
	*CBOT,JGRID,JPROP	SS000670
908	FORMAT(3X,F4.1,3X,F4.1,3X,I2,3X,F6.4 ,F4.2,3X,F4.2,3X,F5.3,3X,	SS000680
	1F5.3,3X,F2.1,3X,I1,3X,I1)	SS000690
	IF(DISTF.NE.0.0) GO TO 4102	SS000700
	IF(ZL.LT.10..OR. TNLB.EQ.0) WRITE(6,917) ZL, TNLB	SS000710
917	FORMAT(//////,40X,*CHECK YOUR DATA = BEAM LENGTH AND NUMBER OF BEASS000720	
	MS ARE/,/,60X,F10.2,5X,F5.2)	SS000730
	IF(ZL.LT.10..OR. TNLB.EQ.0) STOP	SS000740
4102	READ(99,9083) IAB,IAC,IAD,IAE	SS000750
9083	FORMAT(42X,4A4)	SS000760
	IF(IAB.EQ.BLANK.AND.IAC.EQ.BLANK) ALDEF=1000.	SS000770
	IF(IAD.EQ.BLANK.AND.IAE.EQ.BLANK) DEFMIN=-1000.	SS000780
C	MAXIMUM NUMBER OF STRANDS PER ROW	SS000790
	READ(5,509)(NSTRMX(I),I=1,26)	SS000800

909	FORMAT(3X,25(I2 ,1X),I2)	SS000810
	DO 910 J=1,26	SS000820
	STRMAX(J)=NSTRMX(J)	SS000830
	IF(NSTRMX(J).NE.0)NRAV=J	SS000840
910	CONTINUE	SS000850
	DO 878 J1=1,26	SS000860
878	GRIDS(J1)=0.	SS000870
	GRIDS(1)=CBOT	SS000880
	IF(JGRID.NE.1) GO TO 862	SS000890
C	NONSTANDARD GRID SPACING	SS000900
	READ(5,880) (GRIDS(I),I=2,26)	SS000910
880	FORMAT(3X,25(F2.1,1X))	SS000920
	IF(GRIDS(2).EQ.0.) WRITE(6,897)	SS000930
897	FORMAT(1X,130(1H*)/1X,2(1H*),' INCORRECT DATA INPUT - DISTANCE FROSS	SS000940
	*M STRAND ROW 1 TO ROW 2 IS 0 - CHECK INPUT CARD (NONSTANDARD GRID	SS000950
	SPACING), COLS. 4 & 5 ',3(1H)/1X,130(1H*))	SS000960
	IF(GRIDS(2).EQ.0.) STOP	SS000970
C	MISCELLANECUS PROPERTIES	SS000980
862	IF(JPROP.NE.1) GO TO 863	SS000990
	READ(5,882) CUW,UHM,SA,SPF, BR2, TR2, BR1, TR1, BS2, TS2, BS1, TS1	SS001000
	*,CREEP1,CREEP2,SHRK1,SHRK2	SS001010
882	FORMAT(3X,F3.3,4X,F2.0,4X,F3.3,4X,F3.0,4X,F2.2,1X,F2.1,1X,F2.2,1X,	SS001020
	*F2.1,3X,F2.2,1X,F2.1,1X,F2.2,1X,F2.1,4X,4(F3.0,2X))	SS001030
	IF(SPF.NE.0.) FPS=SPF	SS001040
	IF(SA.NE.0.) AS=SA	SS001050
	IF(UHM.NE.0.) HUM=UHM	SS001060
	IF(CUW.NE.0.) UWC=CUW	SS001070
	READ(99,9082) (IAA(J1),J1=1,16)	SS001080
9082	FORMAT(30X,2A1,1X,2A1,1X,2A1,1X,2A1,3X,2A1,1X,2A1,1X,2A1,1X,2A1)	SS001090
	IF(IAA(1).NE.BLANK.OR.IAA(2).NE.BLANK) CBR2=BR2	SS001100
	IF(IAA(3).NE.BLANK.OR.IAA(4).NE.BLANK) CTR2=TR2	SS001110
	IF(IAA(5).NE.BLANK.OR.IAA(6).NE.BLANK) CBR1=BR1	SS001120
	IF(IAA(7).NE.BLANK.OR.IAA(8).NE.BLANK) CTR1=TR1	SS001130
	IF(IAA(9).NE.BLANK.OR.IAA(10).NE.BLANK) CBS2=BS2	SS001140
	IF(IAA(11).NE.BLANK.OR.IAA(12).NE.BLANK) CTS2=TS2	SS001150
	IF(IAA(13).NE.BLANK.OR.IAA(14).NE.BLANK) CBS1=BS1	SS001160
	IF(IAA(15).NE.BLANK.OR.IAA(16).NE.BLANK) CTS1=TS1	SS001170
C	CONCRETE COST COEFFICIENTS	SS001180
863	IF(JOPT.NE.1) GO TO 1001	SS001190
912	READ(5,915)(G(I),I=1,6)	SS001200

	READ(5,915)(G(I),I=7,11)	SS001210
915	FORMAT(10X,5(F4.1,9X),F4.1)	SS001220
C	STRAND AND STRAND WRAPPING COST	SS001230
	READ(5,914)COSTFT,COSTWP	SS001240
914	FORMAT(13X,F3.2,46X,F3.2)	SS001250
913	FORMAT(10X,5(F3.1,10X),F3.1)	SS001260
C	28 DAY CONCRETE STRENGTHS	SS001270
	READ(5,913)(F(I),I=1,6)	SS001280
	READ(5,913)(F(I),I=7,11)	SS001290
	DO 916 J=1,11	SS001300
	IF(F(J).NE.0.0) FFCMAX=4.0+(J-1)*0.5	SS001310
	IF(F(J).EQ.0.0) F(J)=F(J-1)	SS001320
	IF(G(J).EQ.0.0) G(J)=G(J-1)	SS001330
916	CONTINUE	SS001340
1001	CONTINUE	SS001350
C		SS001360
C	LIVE LOAD DISTRIBUTION FACTOR	SS001370
C		SS001380
	IF(DISTF.NE.0.0) GO TO 933	SS001390
	CONSNT=1.0	SS001400
	C11 = CONSNT*WIDTH/ZL	SS001410
	D11=5.+JTNTL/10.+(3.-2.*JTNTL/7.)*((1.-C11/3.)*2)	SS001420
	IF(C11.GT.3) D11=5.+JTNTL/10.	SS001430
	S11=(12.*JTNTL+9.)/ TNLB	SS001440
	DISTF =(S11/D11)*0.5	SS001450
933	CONTINUE	SS001460
	IAASHO=1	SS001470
C		SS001480
C	AASHTO TRUCK LOADINGS	SS001490
C		SS001500
	IF(IA.EQ.BLANK.AND.IB.EQ.BLANK) IAASHO=0	SS001510
	IF(IA.EQ.BLANK.AND.IB.EQ.BLANK) GO TO 6000	SS001520
	IF(IA.EQ.BLANK.AND.IB.EQ.ONE)GO TO 1000	SS001530
	IF(IA.EQ.HHH.AND.IB.EQ.ONE)GO TO 1000	SS001540
	IF(IA.EQ.SSS.AND.IB.EQ.ONE)GO TO 2000	SS001550
	IF(IA.EQ.BLANK.AND.IB.EQ.TWO)GO TO 3000	SS001560
	IF(IA.EQ.HHH.AND.IB.EQ.TWO)GO TO 3000	SS001570
	IF(IA.EQ.SSS.AND.IB.EQ.TWO)GO TO 4000	SS001580
	WRITE(6,950)	SS001590
950	FORMAT(1X,130(1H*)/1X,30(1H*),*UNRECOGNIZABLE AASHTO TRUCK LOADINGS	SS001600

=CHECK INPUT CARD 4, COLS. 5 THRU 8.30(1H*)/1X.130(1H*))
STOP

C
C H=15 TRUCK
C

1000 ZAXLE(1)=6.
ZAXLE(2)=24.
ZNWHL(1)=14.
NAXLE=2
ULOAD=0.480
CSLOAD=19.5
CMLOAD=13.5
GO TO 6000

C
C HS=15 TRUCK
C

2000 ZAXLE(1)=6.
ZAXLE(2)=24.
ZAXLE(3)=24.
ZNWHL(1)=14.
ZNWHL(2)=28.
NAXLE=3
ULOAD=0.480
CSLOAD=19.5
CMLOAD=13.5
GO TO 6000

C
C H=20 TRUCK
C

3000 ZAXLE(1)=8.
ZAXLE(2)=32.
ZNWHL(1)=14.
NAXLE=2
ULOAD=0.640
CSLOAD=26.
CMLOAD=18.
GO TO 6000

C
C HS=20 TRUCK
C

SS001610
SS001620
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SS001900
SS001910
SS001920
SS001930
SS001940
SS001950
SS001960
SS001970
SS001980
SS001990
SS002000

4000	ZAXLE(1)=8.	SS002010
	ZAXLE(2)=32.	SS002020
	ZAXLE(3)=32.	SS002030
	ZNWL(1)=14.	SS002040
	ZNWL(2)=28.	SS002050
	NAXLE=3	SS002060
	ULOAD=0.640	SS002070
	CSLOAD=26.	SS002080
	CMLOAD=18.	SS002090
6000	CONTINUE	SS002100
	CALL PROPTY	SS002110
	Z1=TDIM	SS002120
	Z2=TDIM+XDIM	SS002130
	Z3=HDIM	SS002140
	Z4=HDIM+GDIM	SS002150
	Y1=WHDIM/2.	SS002160
	Y2=WDDIM	SS002170
	Y3=ADIM/2.-Y1-Y2	SS002180
	Y4=XDIM	SS002190
	D(1)=-DBOT+GRIDS(1)	SS002200
	IF(NRAV.EQ.1) GO TO 588	SS002210
	IF(JGRID.EQ.1) GO TO 582	SS002220
	DO 580 J1=2,NRAV	SS002230
580	GRIDS(J1)=2.0	SS002240
582	DO 584 J1=2,NRAV	SS002250
	IF(GRIDS(J1).EQ.0.) GRIDS(J1)=GRIDS(J1-1)	SS002260
584	D(J1)=D(J1-1)+GRIDS(J1)	SS002270
588	CONTINUE	SS002280
C*****		SS002290
C****	PRINT OUT INPUT QUANTITIES	SS002300
C*****		SS002310
	WRITE(6,9080) (TITLE(1,J1),J1=10,11),(TITLE(1,J1),J1=14,26),	SS002320
	1(TITLE(1,J1),J1=48,54),(TITLE(2,J2),J2=13,19),(TITLE(2,J2),J2=26,2	SS002330
	28),(TITLE(2,J2),J2=44,54),(TITLE(3,J3),J3=13,54)	SS002340
9080	FORMAT(1H1,37X,'DISTRICT ',2A1,1X,13A1,' COUNTY HIGHWAY NO. ',	SS002350
	17A1/38X,'CENTFCL NO. ',7A1,' IPE ',3A1,' SUBMITTED BY ',11A1/	SS002360
	238X,'DESCRIPTION ',42A1)	SS002370
600	FORMAT(1H1)	SS002380
	WRITE(6,601)	SS002390
601	FORMAT(/,1X,129(' '*'))	SS002400

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        WRITE(6,610)
610  FORMAT(' *',47X,'BEAM DIMENSIONS AND PROPERTIES',50X,'*')
        WRITE(6,602)
602  FORMAT(1X,129('**'))
        WRITE(6,611)
611  FORMAT(1X,'**',26('..'),*((DIMENSIONS IN INCHES)',27('..'),'**',5X,'*',
&.SECTION PROPERTIES (WITHOUT SHEAR KEY)...****')
        WRITE(6,613)
613  FORMAT(1X,'**',2X,'A',5X,'B',5X,'C',5X,'D',5X,'E',5X,'F',5X,'G',5X,
**H',5X,'M',5X,'T',5X,'W',5X,'X',5X,'Y',9X,'I(IN**4)',5X,'A(IN**2)',
*,4X,'YT(IN)',5X,'YB(IN) **')
        WRITE(6,650)ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,WHDIM,TDIM,WDD
&IM,XDIM,YDIM,BINERT,ACONC,DTOP,DBOT
650  FORMAT(1X,'**',F5.2,12F6.2,5X,F10.0,6X,F6.1,5X,F5.2,6X,F6.2,'*')
        WRITE(6,603)
603  FORMAT(1X,'**',127X,'**')
        WRITE(6,614) BINERK,ACONCK,DTOPK,DBOTK
614  FORMAT(1X,'* COMPRESSION  MAXIMUM  MINIMUM',13X,'STRAND',14X,'COSS
&NCRETE',9X,'*...SECTION PROPERTIES (WITH SHEAR KEY)...****',/,1X,'
&* REINFORCING  INITIAL  INITIAL  STRAND  ULTIMATE  RELATIVE',
&4X,
&'UNIT',13X,'I(IN**4)',5X,'A(IN**2)',4X,'YT(IN)',5X,'YB(IN) **',/,
&* AREA',7X,'CAMBER  CAMBER  AREA  STRENGTH  HUMIDITY
&WEIGHT',10X,F10.0,6X,F6.1,5X,F5.2,6X,F6.2,'*',/,1X,'* (IN**2)',
&6X,'(IN)',6X,'(IN)',5X,'(IN**2) (KSI)',6X,'(X) (K/FT**3)',5
&3X,'**')
        WRITE(6,651) APRIME,AS,FPS,HUM,UWC
651  FORMAT(' *',4X,F5.2,2X,20X,F10.3,5X,F4.0,7X,F3.0,7X,F5.3,55X,'*')
        IF(ALDEF.GT.999..AND.DEFMIN.GT.=999.) WRITE(6,7101) DEFMIN
        IF(ALDEF.LT.999..AND.DEFMIN.LT.=999.) WRITE(6,7102) ALDEF
        IF(ALDEF.LT.999..AND.DEFMIN.GT.=999.) WRITE(6,7103) ALDEF,DEFMIN
7101  FORMAT('+',22X,F10.3)
7102  FORMAT('+',12X,F10.3)
7103  FORMAT('+',12X,2F10.3)
        WRITE(6,603)
        WRITE(6,615)
615  FORMAT(' *',54('..'),'STRAND INFORMATION',55('..'),'**/' *',127X, '*
&',/, ' *ROW NUMBER',16X,'1  2  3  4  5  6  7  8  9  10  11
& 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26','**')
        WRITE(6,652) (NSTRMX(I),I=1,26)

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SS002410
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SS002500
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SS002780
SS002790
SS002800

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652 FORMAT(1X,'**',          'MAXIMUM NO. OF STRANDS ',26(1X,I3),'**') SS002810
    IF(NRAV.EQ.1) WRITE(6,894) CBCT SS002820
    IF(NRAV.EQ.1) GO TO 891 SS002830
    WRITE(6,894) (GRIDS(I),I=1,26) SS002840
894 FORMAT(1X,'**','SPACING (ROW I=1 TO I)',1X,26(F4.1),'**') SS002850
891 WRITE(6,603) SS002860
    WRITE(6,888) SS002870
888 FORMAT(1X,'**',49(' '), 'ALLOWABLE STRESS COEFFICIENTS',49(' '), '**') SS002880
    WRITE(6,603) SS002890
    WRITE(6,883) SS002900
883 FORMAT(1X,'**',15X,'RELEASE',8X,'END 1/10',2X,'REMAINDER',29X, SS002910
    '** SERVICE',8X,'END 1/10',2X,'REMAINDER',15X,'**') SS002920
    WRITE(6,896) CBR2,CBR1,CBS2,CBS1,CTR2,CTR1,CTS2,CTS1 SS002930
896 FORMAT(1X,'**',27X,'C',4X,F4.2,6X,F4.2,44X,'C',4X,F4.2,6X,F4.2, SS002940
    *18X,'**'/1X,'**',27X,'T',4X,F4.2,6X,F4.2,44X,'T',4X,F4.2,6X,F4.2, SS002950
    *18X,'**') SS002960
    WRITE(6,603) SS002970
    WRITE(6,895) CREEP1,CREEP2,SHRK1,SHRK2 SS002980
895 FORMAT(1X,'**',47(' '), 'CREEP AND SHRINKAGE COEFFICIENTS',48(' '), SS002990
    '**'/1X,'**',27X,100X,'**'/1X,'**',30X,'CREEP1 = ',F4.0,5X,'CREEP2 = ' SS003000
    ',F4.0,7X,'SHRK1 = ',F4.0,5X,'SHRK2 = ',F4.0,30X,'**') SS003010
    WRITE(6,603) SS003020
    IF(JOPT.NE.1) GO TO 695 SS003030
    WRITE(6,616) SS003040
616 FORMAT(' **',46(' '), 'CONCRETE COST COEFFICIENTS($/YD**3)',46(' '), SS003050
    &'**') SS003060
    WRITE(6,653)(G(I),I=1,8) SS003070
653 FORMAT(' *4.0KSI/$',F5.1,' 4.5KSI/$',F5.1,' 5.0KSI/$',F5.1,' 5. SS003080
    &5KSI/$',F5.1,' 6.0KSI/$',F5.1,' 6.5KSI/$',F5.1,' 7.0KSI/$',F5.1 SS003090
    &,' 7.5KSI/$',F5.1,9X,'**') SS003100
    WRITE(6,654)(G(I),I=9,11) SS003110
654 FORMAT(' *8.0KSI/$',F5.1,' 8.5KSI/$',F5.1,' 9.0KSI/$',F5.1,84X,' SS003120
    &'**') SS003130
    WRITE(6,603) SS003140
    WRITE(6,617) SS003150
617 FORMAT(' **',28(' '), '28 DAY CONCRETE STRENGTH COEFFICIENTS (RELEASSS003160
    &E STRENGTH/28 DAY STRENGTH)',27(' '), '**') SS003170
    WRITE(6,655)(F(I),I=1,8) SS003180
655 FORMAT(' *4.0KSI/°',F4.1,' KSI 4.5KSI/°',F4.1,' KSI 5.0KSI/°',F4.1,' SS003190
    &KSI 5.5KSI/°',F4.1,' KSI 6.0KSI/°',F4.1,' KSI 6.5KSI/°',F4.1,' KSI 7SS003200

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      6.0KSI/' ,F4.1,'KSI  7.5KSI/' ,F4.1,'KSI**')
      WRITE(6,656)(F(I),I=9,11)
656  FORMAT(' *8.0KSI/' ,F4.1,'KSI  8.5KSI/' ,F4.1,'KSI  9.0KSI/' ,F4.1,'
      6KSI' ,80X,'**')
      WRITE(6,603)
      WRITE(6,657)COSTFT, COSTWP,FPCMAX
657  FORMAT(' *STRAND COST = $' ,F4.2,'/FT' ,16X,'STRAND WRAPPING COST =
      &$' ,F4.2,'/FT' ,16X,'MAXIMUM RELEASE STRENGTH ALLOWED =' ,F4.1,' KSI*
      **')
695  WRITE(6,602)
      WRITE(6,9007)
9007 FORMAT(/1X,129('**'))
      WRITE(6,618)
618  FORMAT(' *' ,53X,'BRIDGE PROPERTIES' ,57X,'**')
      WRITE(6,602)
      WRITE(6,603)
      WRITE(6,658)ZL ,WIDTH,JTNTL, TNLB
658  FORMAT(' *SPAN LENGTH = ' ,F5.1,'(FT)' ,5X,'BRIDGE WIDTH = ' ,F5.1,'(
      &FT)' ,5X,'NUMBER TRAFFIC LANES = ' ,12,5X,'NUMBER BEAMS = ' ,F5.2,20X
      *,**')
      WRITE(6,602)
      WRITE(6,600)
      WRITE(6,602)
      WRITE(6,619)
619  FORMAT(' *' ,52X,'LOADING CONDITIONS' ,57X,'**')
      WRITE(6,602)
      WRITE(6,603)
      IF(IA.EQ.BLANK.AND.IB.EQ.BLANK) GO TO 696
      IF(IB.EQ.TWO) WRITE(6,659)IA,DISTF
659  FORMAT(' * AASHTO LL = H' ,A1,'-20' ,7X, 'L.L. DISTRIBUTION FACTOR
      &= ' ,F5.3,'(TRUCKS)' ,62X,'**')
      IF(IB.EQ.ONE) WRITE(6,660)IA,DISTF
660  FORMAT(' * AASHTO LL = H' ,A1,'-15' ,7X, 'L.L. DISTRIBUTION FACTOR
      &= ' ,F5.3,'(TRUCKS)' ,62X,'**')
      WRITE(6,603)
696  IF(JLOAD.NE.1) GO TO 697
      WRITE(6,620)(I,I=1,18)
620  FORMAT(' *' ,57('.'),'AXLE TRAIN' ,60('.'),'*' ,/, ' *AXLE NUMBER
      &18I6,5X,'**')
      WRITE(6,661)(PAXLE(I),I=1,18)

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SS003210
SS003220
SS003230
SS003240
SS003250
SS003260
SS003270
SS003280
SS003290
SS003300
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SS003470
SS003480
SS003490
SS003500
SS003510
SS003520
SS003530
SS003540
SS003550
SS003560
SS003570
SS003580
SS003590
SS003600

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661 FORMAT(' *AXLE LOAD(KIPS)',18F6.1,4X,'**')
WRITE(6,662)(NWHL(I),I=1,17)
662 FORMAT(' *DISTANCE TO AXLE(FT) ',17F4.0,4X,'**')
WRITE(6,603)
697 IF(JCONC.NE.1) GO TO 698
WRITE(6,621)
621 FORMAT(' *',45(' '), 'CONCENTRATED FORCES ON SINGLE BEAM',48(' '),
&*)
WRITE(6,664)(FCONC(I),I=1,10)
664 FORMAT(' *LOAD(KIPS) ',10F9.1,23X,'**')
WRITE(6,665)(DCCNC(I),I=1,10)
665 FORMAT(' *DISTANCE FROM',114X,'*',/, ' *LEFT SUPPORT(FT) ',F5.1,9F
&9.1,23X,'**')
WRITE(6,603)
698 IF(ULSB.EQ.0.0) GO TO 699
WRITE(6,666)ULSB
666 FORMAT(' *UNIFORM LOAD ON SINGLE BEAM = ',F5.2,'(K/FT)',86X,'**')
699 CONTINUE
WRITE(6,602)
C*****SS003800
C**** COMPUTE DESIGN MOMENTS AND SHEARS SS003810
C*****SS003820
DO 20 J1=1,7
DLMDM(J1)=0.
DLSHR(J1)=0.
IF(J1.EQ.1) ZX=0.
IF(J1.EQ.2) ZX=ZL/10.
IF(J1.EQ.3) ZX=2.*ZL/10.
IF(J1.EQ.4) ZX=ZL/4.
IF(J1.EQ.5) ZX=3.*ZL/10.
IF(J1.EQ.6) ZX=4.*ZL/10.
IF(J1.EQ.7) ZX=ZL/2.
ZMDL=(ACONCK*UWC/144.)*ZX/2.*(ZL-ZX)
IF(J1.EQ.7) ZMBW=ZMDL
ZSDL=(ACONCK*UWC/144.)*(ZL/2.-ZX)
IF(JCONC.NE.1) GO TO 14
SUMM=0.
SUMV=0.
DO 12 J2=1,10
IF(DCONC(J2).EQ.0.) GO TO 14

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R=FCNC(J2)*(ZL=DCNC(J2))/ZL	SS004010
SUMM=SUMM+R*ZX	SS004020
SUMV=R	SS004030
IF(DCNC(J2).GT.ZX+.1) GO TO 10	SS004040
SUMM=SUMM-FCNC(J2)*(ZX=DCNC(J2))	SS004050
SUMV=SUMV-FCNC(J2)	SS004060
10 DLMOM(J1)=DLMOM(J1)+SUMM	SS004070
DL SHR(J1)=DL SHR(J1)+SUMV	SS004080
12 CONTINUE	SS004090
14 DLMOM(J1)=DLMOM(J1)+ULSB*ZX/2.*(ZL=ZX)	SS004100
DL SHR(J1)=DL SHR(J1)+ULSB*(ZL/2.=ZX)	SS004110
ZMOML=0.	SS004120
ZMOMT=0.	SS004130
ZSHRL=0.	SS004140
ZSHRT=0.	SS004150
ZMOMAX=0.	SS004160
ZSHRAX=0.	SS004170
ZIMP=1.+50./(125.+ZL)	SS004180
IF(ZIMP.GT.1.30) ZIMP=1.30	SS004190
IF(IAASH0.EQ.0) GO TO 16	SS004200
ZMOML=DISTF*ZIMP*ULOAD*ZX/2.*(ZL=ZX)	SS004210
ZSHRL=DISTF*ZIMP*ULOAD*(ZL/2.=ZX)	SS004220
R=CMLOAD*(ZL=ZX)/ZL	SS004230
ZMOML=ZMOML+R*ZX*DISTF*ZIMP	SS004240
ZSHRL=ZSHRL+R*DISTF*ZIMP	SS004250
CALL MOMSHR(ZL,ZNWHL,NAXLE,ZX,ZAXLE,ZMOMT,ZSHRT)	SS004260
ZMOMT=ZMOMT*DISTF*ZIMP	SS004270
ZSHRT=ZSHRT*DISTF*ZIMP	SS004280
16 IF(JLOAD.EQ.0) GO TO 18	SS004290
CALL MOMSHR(ZL,NWHL,NWHEEL,ZX,PAXLE,ZMOMAX,ZSHRAX)	SS004300
ZMOMAX=ZMOMAX*DISTF	SS004310
ZSHRAX=ZSHRAX*DISTF	SS004320
18 BMMA X(J1)=AMAX1(ZMOML,ZMOMT,ZMOMAX)	SS004330
BVMA X(J1)=AMAX1(ZSHRL,ZSHRT,ZSHRAX)	SS004340
20 CONTINUE	SS004350
ZMAX=0.0	SS004360
DO 4103 J1=1,7	SS004370
IF(BMMA X(J1).GT.ZMAX) ZMAX =BMMA X(J1)	SS004380
4103 CONTINUE	SS004390
ZMNC=ABS(ACONCK=ACONC)*UWC*ZL**2/(144.*8.)	SS004400

ZMC=DL MOM (7)	SS004410
ULTRQ=1.3*(ZMCL+DL MOM (7)+(5./3.)*ZMAX)	SS004420
C*****	SS004430
C**** GENERATE OBJECTIVE FUNCTION AND INITIAL CONSTRAINT SET	SS004440
C*****	SS004450
ASSCLR=0.05	SS004460
ASSPLS=0.1	SS004470
ZSSCLR=ASSCLR	SS004480
ZSSPLS=ASSPLS	SS004490
ELASC=1.04355*(1000.*UWC)**1.5	SS004500
NFRCE=0	SS004510
NEQS=0	SS004520
INDX=0	SS004530
IF(JOPT.EQ.0) GO TO 110	SS004540
N=1+45+11*NRAV	SS004550
M=11*NRAV+10	SS004560
GO TO 112	SS004570
110 N=1+26+11*NRAV	SS004580
M=11*NRAV+1	SS004590
112 KK=M+N-1	SS004600
K=N+M-1	SS004610
IF(JOPT.EQ.1) GO TO 108	SS004620
C	SS004630
C DEFINE COST COEFFICIENTS FOR DESIGN OPTION	SS004640
C	SS004650
COSTFT=100.	SS004660
G(1)=1944.*COSTFT/(4.0*ACONC)	SS004670
COSTWP=0.1	SS004680
108 CALL EGEN	SS004690
KODE=0	SS004700
C*****	SS004710
C**** ITERATE ON PRESTRESS LOSS	SS004720
C*****	SS004730
700 CONTINUE	SS004740
CALL LPCODE(NFRCE,NEQS,INDX,KODE,KBOMB)	SS004750
C	SS004760
C COMPUTE NEW PRESTRESS LOSSES	SS004770
C	SS004780
AMOM=0.	SS004790
ASUM=0.	SS004800

DO 706 J1=1,NRAV	SS004810
ASUM=ASUM+X(J1)*AS	SS004820
706 AMOM=AMOM-X(J1)*AS*D(J1)	SS004830
DCG=DBOT-AMOM/ASUM	SS004840
IF(JOPT.NE.0) GO TO 708	SS004850
FPCR=X(11*NRAV+1)	SS004860
GO TO 712	SS004870
708 FPCR=4.0	SS004880
KNT=11*NRAV	SS004890
DO 710 J1=1,10	SS004900
710 FPCR=FPCR+X(KNT+J1)	SS004910
712 CALL PLCSS(FPCR,ZMBW,ZMC,ZMNC,FPS,ASUM,ACONC,EINERT,BINERK,DBOT,	SS004920
*DBOTK,DCG,HUM,ZL,ZLCSS,ZINLOS,UWC)	SS004930
C	SS004940
C TERMINATE ITERATIONS IF COMPUTED LONG TERM LOSS DOES	SS004950
C NOT EXCEED ASSUMED LOSS BY MORE THAN 3 PERCENT	SS004960
C	SS004970
IF(KBOMB.NE.0) GO TO 740	SS004971
IF(ZLOSS.LE.1.03*ASSPLS) GO TO 740	SS004980
C	SS004990
C UPDATE CONSTRAINTS TO REFLECT NEW PRESTRESS LOSSES	SS005000
C	SS005010
READ(4) ARRAY	SS005020
REWIND 4	SS005030
JST=11*NRAV	SS005040
C RELEASE AND SERVICE STRESS CONSTRAINTS	SS005050
DO 714 J1=2,13	SS005060
DO 714 J2=1,JST	SS005070
714 ARRAY(J1,J2)=ARRAY(J1,J2)*((1.-ZINLOS)/(1.-ZSSCLR)	SS005080
DO 716 J1=14,21	SS005090
DO 716 J2=1,JST	SS005100
716 ARRAY(J1,J2)=ARRAY(J1,J2)*((1.-ZLOSS)/(1.-ZSSPLS)	SS005110
C BOUNDS ON INITIAL CAMBER	SS005120
JR=22+11*NRAV	SS005130
IF(JOPT.NE.0) JR=41+11*NRAV	SS005140
DO 718 J1=1,11	SS005150
DO 718 J2=1,NRAV	SS005160
ARRAY(JR,(J1-1)*NRAV+J2)=ARRAY(JR,(J1-1)*NRAV+J2)*((1.-ZINLOS)/	SS005170
*(1.-ZSSCLR)	SS005180
718 ARRAY(JR+1,(J1-1)*NRAV+J2)=ARRAY(JR+1,(J1-1)*NRAV+J2)*((1.-ZINLOS)/	SS005190

	*(1.=ZSSCLR)	SS005200
C	CRACKING MOMENT CAPACITY	SS005210
	JR=25+11*NRAV	SS005220
	IF(JOPT.NE.0) JR=44+11*NRAV	SS005230
	DO 719 J1=1,NRAV	SS005240
719	ARRAY(JR,J1) = ARRAY(JR,J1)*(1.=ZLOSS)/(1.=ZSSPLS)	SS005250
	IF(ZLOSS.LE.0.2) I1=1	SS005260
	IF(ZLOSS.LE.0.2) I2=2	SS005270
	IF(0.2.LT.ZLOSS.AND.ZLOSS.LE.0.3) I1=2	SS005280
	IF(0.2.LT.ZLOSS.AND.ZLOSS.LE.0.3) I2=3	SS005290
	IF(ZLOSS.GT.0.3) I1=3	SS005300
	IF(ZLOSS.GT.0.3) I2=4	SS005310
	S1=I1/10.	SS005320
	S2=I2/10.	SS005330
	IF(JOPT.NE.0) GO TO 720	SS005340
	ARRAY(25+11*NRAV,K+1) = ((PECRK(1,I2)-PECRK(1,I1))*ZLOSS/0.1 -	SS005350
	*(PECRK(1,I2)*S1-PECRK(1,I1)*S2)/0.1)	SS005360
	GO TO 726	SS005370
720	DO 722 J1=1,11	SS005380
722	ZNE(J1) = (PECRK(J1,I2)-PECRK(J1,I1))*ZLOSS/0.1 - (PECRK(J1,I2)*S1 -	SS005390
	*PECRK(1,I1)*S2)/0.1	SS005400
	DO 724 J1=1,10	SS005410
724	ARRAY(44+11*NRAV,11*NRAV+J1) = 2.*(ZNE(J1+1)-ZNE(J1))	SS005420
	ARRAY(44+11*NRAV,K+1) = -ZNE(1)	SS005430
726	CONTINUE	SS005440
	JC=11*NRAV+10	SS005450
	JR=44+11*NRAV	SS005460
	IF(JOPT.EQ.0) JC=11*NRAV+1	SS005470
	IF(JOPT.EQ.0) JR=25+11*NRAV	SS005480
	KODE=1	SS005490
	ASSPLS=ZLOSS	SS005500
	ASSCLR=ZINLOS	SS005510
	GO TO 700	SS005520
740	CONTINUE	SS005530
C		SS005540
C	UNSCRAMBLE L.P. NOTATION FOR BOND BREAKAGE	SS005550
C		SS005560
	DO 402 J1=1,NRAV	SS005570
	DO 402 J2=1,11	SS005580
	IDX=J1+(J2-1)*NRAV	SS005590

S1 = X(IDX)	SS005600
S2 = AINT(S1)	SS005610
IF(S1-S2.GE.0.5) X(IDX) = S2+1	SS005620
IF(S1-S2.LT.0.5) X(IDX) = S2	SS005630
402 CONTINUE	SS005640
DEL=ZL/40.	SS005650
JT = 0	SS005660
DO 410 J1=1,NRAV	SS005670
JS = 0	SS005680
DO 408 J2=1,10	SS005690
IDX=J1+(J2-1)*NRAV	SS005700
IF(ABS(X(IDX)-X(IDX+NRAV)).LE.0.001) GO TO 408	SS005710
JS = JS + 1	SS005720
ZWRAP(JS,1)=X(IDX)-X(IDX+NRAV)	SS005730
NWRAP(JS,1)=ZWRAP(JS,1)	SS005740
ZWRAP(JS,2) = (11-J2)*DEL	SS005750
408 CONTINUE	SS005760
NSDIF(J1) = JS	SS005770
410 JT = JT + JS	SS005780
IF(JT.EQ.0) GO TO 422	SS005790
DO 414 J1=1,JT	SS005800
SS = ZWRAP(J1,2)	SS005810
S1 = AINT(SS)	SS005820
DEL = SS-S1	SS005830
DO 412 J2=1,13	SS005840
S2=(J2-1)/12.	SS005850
IF(DEL.GT.S2) GO TO 412	SS005860
ZWRAP(J1,2)=ABS(S1)	SS005870
ZWRAP(J1,3)=J2-1	SS005880
IF(ZWRAP(J1,3).LT.12.) GO TO 414	SS005890
ZWRAP(J1,2)=ZWRAP(J1,2)+1.	SS005900
ZWRAP(J1,3)=0.	SS005910
GO TO 414	SS005920
412 CONTINUE	SS005930
414 CONTINUE	SS005940
422 CONTINUE	SS005950
C 28=DAY CONCRETE STRENGTH	SS005960
IF(JOPT.EQ.0) GO TO 648	SS005970
F28=F(1)	SS005980
DO 642 J3=2,11	SS005990

642	F28=F28+2.*(F(J3)-F(J3-1))*X(11*NRAV+J3-1)	SS006000
C	LONG TERM CAMBER	SS006010
648	SUM=0.	SS006020
	SUM10=0.0	SS006030
	SUM11=0.0	SS006040
	DO 432 J=1,NRAV	SS006050
	SUM10=SUM10+AS*X(J)	SS006060
	SUM11=SUM11-AS*D(J)*X(J)	SS006070
432	SUM=SUM+X(J)	SS006080
	STRNS=SUM	SS006090
	EC=ELASC*SQRT(F28)/1.E+03	SS006100
	ES=29.	SS006110
	ECCL=SUM11/SUM10	SS006120
	ENDECC=ECCL	SS006130
	FP=FO*1000.	SS006140
	UWB=UWC*1000.	SS006150
	CALL CAMBER(ES,EC,AS,STRNS,UWB,ACONC,ZL,ECCL,BINERT,FP,ESS	SS006160
	*NDECC,PRLMAX,CBRMAX)	SS006170
C	DEFLECTION CALCULATIONS	SS006180
	ECR=ELASC*SQRT(FPCR)	SS006190
	SUM6=0.0	SS006200
	DO 9057 J1=1,11	SS006210
	SUM7=0.0	SS006220
	DO 9058 J2=1,NRAV	SS006230
9058	SUM7=SUM7+D(J2)*X((J1-1)*NRAV+J2)	SS006240
	YJC=((11-J1)*ZL/40.+ZL/80.)*12.	SS006250
	DJC=ZL/40.*12.	SS006260
	IF(J1.EQ.1) YJC=3.*ZL/8.*12.	SS006270
	IF(J1.EQ.1) DJC=ZL/4.*12.	SS006280
9057	SUM6=SUM6+SUM7*DJC*YJC	SS006290
	DEFBWK=(-22.5*WB*ZL**4-(1.-ZLOSS)*FO*SUM6)/5000./BINERT	SS006300
	DEFCF=0.	SS006310
	IF(JCONC.NE.1) GO TO 2445	SS006320
	DO 2444 JN=1,10	SS006330
	ZBX1=ZL-DCCNC(JN)	SS006340
	ZBX2=DCCNC(JN)	SS006350
	ZBX=AMIN1(ZBX1,ZBX2)*12.	SS006360
2444	DEFCF=DEFCF+FCONC(JN)*ZBX*(3.*ZL**2*144.-4.*ZEX**2)/48.	SS006370
2445	CONTINUE	SS006380
	DEFBWU=(-22.5*(ULSB+WB)*ZL**4-(1.-ZLOSS)*FO*SUM6-DEFCF)/5000./	SS006390


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*BINERK
C      ULTIMATE MOMENT AND CRACKING MOMENT CAPACITY
      DD=DTOP+ECCL
      CALL ULTMP(SUM10,F28,FPS,APRIME,FPL,DD,DDIM,FSY,DCR,
*Y1,Y2,Y3,Y4,Z1,Z2,Z3,Z4,ZMUL)
      SUM8=0.0
      DO 9056 J2=1,NRAV
9056  SUM8=SUM8+(1./ACONC=D(J2)/ZB)*X(J2)
      ZCRACK=((1.-ZLOSS)*ZBK*FC*SUM8+7.5*ZBK*0.031623*SQRT(F28)-W *ZL**2
1/8.*12.*ZBK/ZB)/12.
C*****
C**** PRINT OUT RESULTS
C*****
      WRITE(6,600)
      IF(KBOMB.NE.0) WRITE(6,70)
70  FORMAT(50X,32('*'),/,50X,'*SORRY, THIS BEAM WILL NOT WORK*',/,50X,
*32('*'))
      IF(JOPT.EQ.0) WRITE(6,622)
622  FORMAT( / , ' ',47X,'THE COMMAND IS TO SELECT STRANDS',48X,' ')
      IF(JOPT.EQ.1) WRITE(6,623)
623  FORMAT( / , ' ',50X,'THE COMMAND IS TO OPTIMIZE',51X,' ')
      WRITE(6,601)
      WRITE(6,624)
624  FORMAT(' ',54X,'DESIGN PROPERTIES',56X,'*')
      WRITE(6,602)
      WRITE(6,603)
      ZLS=ZLOSS*100.
      ZINL=ZINLOS*100.
      WRITE(6,625) FPCR,ECR,ZINL,F28,ZLS
625  FORMAT(' ',4X,'RELEASE STRENGTH = ',F5.2,' (KSI)',4X,'CONCRETE MSS
10DULUS(RELEASE) = ',F7.1,' (KSI)',4X,'INITIAL PRESTRESS LOSS = ',
2F5.2,' PERCENT',4X,'**/' ' ',4X,'28-DAY STRENGTH = ',F5.2,' (KSI)
3',50X,'TOTAL PRESTRESS LOSS = ',F5.2,' PERCENT',4X,'*')
      WRITE(6,602)
      WRITE(6,601)
      WRITE(6,628)
628  FORMAT(' ',56X,'DESIGN RESULTS',57X,'*')
      WRITE(6,602)
      WRITE(6,603)
      WRITE(6,629)

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SS006400
SS006410
SS006420
SS006430
SS006440
SS006450
SS006460
SS006470
SS006480
SS006490
SS006500
SS006510
SS006520
SS006521
SS006522
SS006523
SS006524
SS006530
SS006540
SS006550
SS006560
SS006570
SS006580
SS006590
SS006600
SS006610
SS006620
SS006630
SS006640
SS006650
SS006660
SS006670
SS006680
SS006690
SS006700
SS006710
SS006720
SS006730
SS006740
SS006750

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629  FORMAT(' *',56(' '), 'STRAND LAYOUT',58(' '), '**')
      WRITE(6,603)
      WRITE(6,630)
630  FORMAT(' *',31X,'ROW',7X,'STRANDS' WRAPPED STRANDS IN EACH ROW'
1,47X,'**',/, ' *',30X,'NUMBER' PER ROW',79X,'**')
      JCNT = 1
      DO 631 K = 1,NRAV
      INTX=X(K)
      JSTP = NSDIF(K) + JCNT-1
      IF(NSDIF(K).EQ.0) WRITE(6,640) K,INTX
640  FORMAT(' *',32X,I2,9X,I3,1X,6X,'THERE ARE NO WRAPPED STRANDS IN TH
1IS ROW',34X,'**')
      IF(NSDIF(K).EQ.0) GO TO 631
      IF(NSDIF(K).GT.1) WRITE(6,632) K,INTX,(NWRAP(J3,1),ZWRAP(J3,2),
1ZWRAP(J3,3),J3=JCNT,JSTP)
      IF(NSDIF(K).LE.1) WRITE(6,9073)K,INTX,(NWRAP(J3,1),ZWRAP(J3,2),
1ZWRAP(J3,3),J3=JCNT,JSTP)
632  FORMAT(' *',32X,I2,9X,I3,1X,6X,I3 , ' STRANDS WRAPPED FOR ',F4.1,'
1FT = ',F4.1,' INCHES',29X,'**',/,( ' *',53X,I3 , ' STRANDS WRAPPED F
2OR ',F4.1,' FT = ',F4.1,' INCHES',29X,'**'))
9073  FORMAT(' *',32X,I2,9X,I3,1X,6X,I3 , ' STRANDS WRAPPED FOR ',F4.1,'
1FT = ',F4.1,' INCHES',29X,'**')
631  JCNT = JCNT + NSDIF(K)
      WRITE(6,603)
C    PUT STIRRUP SPACING & CAMBER PRINT OUT ROUTINES HERE
      WRITE(6,603)
      CBRI=ALDEF=X(KK-5)*1.E+04/(ELASC*SQRT(FPCR)*BINERT)
      WRITE(6,634)CBRMAX(1),CBRI,CBRMAX(2),DEFBWK,CBRMAX(3),DEFBWU,
1CBRMAX(4)
      IF(CREEP1.NE.0.) WRITE(6,9049) CBRMAX(5)
634  FORMAT(1X,'**',53(' '), 'COMPUTED DEFLECTION',55(' '), '**'/1X,'**',
1127X,'**'/1X,'**',22X,'SHORT TERM',25X,' *',29X,'LONG TERM',30X,
2 '**'/1X,'**',57X,' *',68X,'**'/1X,'**',8X,'CONDITION',9X,'**',3X,
4 'MODULUS',3X,'**',3X,'DEFLECTION',3X,' *',68X,'**'/1X,'**',26X,'**',13
5X,'**',16X,' *',3X, F5.2,' INCHES (BASED UPON DALLAS CONCRETE PROPE
6RTIES)',13X,'**'/ ' *', 2X,'BMWT',20X,'**',3X,'RELEASE',3X,'**',2X,
7F5.2,' INCHES',2X,' *',3X,F5.2,' INCHES (BASED UPON',
3 ' ODESSA CONCRETE PROPERTIES)',13X,'**'/
81X,'**',2X,'BMWT + KEY',14X,'**', ' 5 MILLION',2X,'**',2X,F5.2,
9' INCHES',2X,' *',3X,F5.2,' INCHES (BASED UPON SAN ANTONIO CONCRET

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1E PROPERTIES)*. 8X, '**/1X, '**.2X, BMWT + KEY + DEAD LOAD*.2X, SS007160
 2** ,.5 MILLION*.2X, '**.2X, F5.2, ' INCHES*.2X, ' *.3X, F5.2, ' INCHES SS007170
 3S (BASED UPON LUFKIN CONCRETE PROPERTIES)*.13X, '**) SS007180
 9049 FORMAT(1X, '**.26X, '**.13X, '**.16X, ' *.3X, F5.2, ' INCHES (BASED UPOSS007190
 IN GIVEN CONCRETE PROPERTIES)*.13X, '**) SS007200
 C SS007210
 C STIRRUP SPACING OUTPUT SS007220
 C SS007230
 DO 9078 J1=1,7 SS007240
 ZX=(J1-1)*ZL/10. SS007250
 IF(J1.GT.3) ZX=(J1-2)*ZL/10. SS007260
 IF(J1.EQ.4) ZX=ZL/4. SS007270
 VU(J1)=1.444*(WB*(ZL/2.-ZX)+DL SHR(J1)+5./3.*BVMAX(J1)) SS007280
 9078 CONTINUE SS007290
 WEB=2.*WDDIM SS007300
 DO 9075 J1=1,7 SS007310
 J2=0 SS007320
 IF(J1.EQ.1) J2=10*NRAV SS007330
 IF(J1.EQ.2) J2=6*NRAV SS007340
 IF(J1.EQ.3) J2=2*NRAV SS007350
 SUM1=0.0 SS007360
 SUM2=0.0 SS007370
 DO 9076 J3=1, NRAV SS007380
 SUM1=SUM1+AS*X(J2+J3) SS007390
 9076 SUM2=SUM2+AS*D(J3)*X(J2+J3) SS007400
 IF(SUM1.LT.0.001) DISTCG=DTOP+ECCL SS007410
 IF(SUM1.LT.0.001) GO TO 9091 SS007420
 DISTCG=DTOP+SUM2/SUM1 SS007430
 9091 CALL SHEAR(WEB, DDIM, DISTCG, F28, FSY, ASTIRP, VU(J1), SPACE) SS007440
 STRSP(J1)=SPACE SS007450
 9075 CONTINUE SS007460
 WRITE(6,603) SS007470
 WRITE(6,9077) ASTIRP, (STRSP(J4), J4=1,7) SS007480
 9077 FORMAT(1X, '**.36('.''), 'STIRRUP SPACING - AASHTO 1973 - STIRRUP ARESS007490
 5A = ', F4.2, ' IN2*.36('.''), '**/ SS007500
 11X, '**.127X, '**/1X, '**.5X, 'SECTION*.10X, '**.5X, '0/10*.5X, '**.5X, SS007510
 2*1/10*.5X, '**.5X, '2/10*.5X, '**.5X, '1/4*.6X, '**.5X, '3/10*.5X, '**. SS007520
 35X, '4/10*.5X, '**.5X, '5/10*.5X, '**/1X, '**.22X, '**.7(14X, '**)/1X, '** SS007530
 4.5X, 'SPACING (IN.)*.4X, '**.7(4X, F5.2, 5X, '**)) SS007540
 C SS007550

C	COST DATA PRINTOUT	SS007560
C		SS007570
	IF(JOPT.EQ.0) WRITE(6,603)	SS007580
	IF(JOPT.EQ.0) GO TO 9020	SS007590
	CONCV=ACONC/144.*ZL/27.	SS007600
	STRFT=STRNS*ZL	SS007610
	WRPFT=0.0	SS007620
	J2=0	SS007630
	J3=1	SS007640
	DO 433 J1=1,NRAV	SS007650
	IF(NSDIF(J1).EQ.0) GO TO 433	SS007660
	J2=J2+NSDIF(J1)	SS007670
	J4=J3	SS007680
	DO 436 I=J4,J2	SS007690
	WRPFT=WRPFT+ZWRAP(I,1)*(ZWRAP(I,2)+(ZWRAP(I,3)/12.))	SS007700
	J3=J3+NSDIF(J1)	SS007710
436	CONTINUE	SS007720
433	CONTINUE	SS007730
	JC1=11*NRAV+1	SS007740
	DO 434 J1=JC1,M	SS007750
	COSTC=G(J1-11*NRAV)+X(J1)*2.*(G(J1-11*NRAV+1)-G(J1-11*NRAV))	SS007760
	IF(X(J1).NE.0.5) GO TO 435	SS007770
434	CONTINUE	SS007780
435	CSTCON=COSTC*CONCV	SS007790
	CSTSTR=STRFT*COSTFT	SS007800
	CSTWRP=WRPFT*CSTWP*2.	SS007810
	CSTTOT=CSTCCN+CSTSTR+CSTWRP	SS007820
	CPRCST=CSTCON/CSTTOT*100.	SS007830
	SPRCST=CSTSTR/CSTTOT*100.	SS007840
	WPRCST=CSTWRP/CSTTOT*100.	SS007850
	CSTPFT=CSTTOT/ZL	SS007860
	WRITE(6,603)	SS007870
	WRITE(6,670)	SS007880
670	FORMAT(1X,'*',44(' '),*COST AND MATERIAL REQUIREMENTS OF BEAM*,4	SS007890
	*5(' '),**)	SS007900
	WRITE(6,603)	SS007910
	WRITE(6,671)	SS007920
671	FORMAT(1X,'*',8X,'ITEM',15X,'AMCUNT',12X,'COST',8X,'PERCENTAGE OF	SS007930
	1TOTAL COST',6X,'*',39X,'**)	SS007940
	WRITE(6,672) CSTTOT	SS007950

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672 FORMAT(1X,'**',87X,'**',7X,'TOTAL COST OF BEAM $ ',F8.2,2X,'**')      SS007960
    WRITE(6,673) CCNCV,CSTCON,CPRCST                                         SS007970
673 FORMAT(1X,'**',6X,'CONCRETE**',9X,F7.2,' YD**3',5X,'$',F8.2,14X,F5.2,SS007980
    1,' $',15X,'**',39X,'**')                                               SS007990
    WRITE(6,674) STRFT,CSTSTR,SPRCST,CSTPFT                                   SS008000
674 FORMAT(1X,'**',7X,'STRANDS',10X,F7.2,' FT',8X,'$',F8.2,14X,F5.2,' $SS008010
    1',15X,'**',7X,'COST PER FOOT $ ',F8.2,2X,'**')                       SS008020
    WRITE(6,675) WRPFT,CSTWRP,WPRCST                                         SS008030
675 FORMAT(1X,'**',3X,'WRAPPED STRANDS',6X,F7.2,' FT',8X,'$',F8.2,14X,F5.2,SS008040
    15.2,' $',15X,'**',39X,'**')                                           SS008050
    WRITE(6,676)                                                             SS008060
676 FORMAT(1X,'**',127X,'**'/1X,'**',5X,'*DOES NOT INCLUDE END SECTION',9SS008070
    *3X,'**')                                                                SS008080
9020 WRITE(6,602)                                                            SS008090
C                                                                              SS008100
C    CRITICAL DESIGN FACTORS OUTPUT                                         SS008110
C                                                                              SS008120
    WRITE(6,600)                                                            SS008130
    DO 800 J1=1,6                                                            SS008140
    DO 800 J2=1,5                                                            SS008150
800 KSYM(J1,J2)=EX(1)                                                       SS008160
    WRITE(6,602)                                                            SS008170
    WRITE(6,810)                                                            SS008180
810 FORMAT(1X,'**',52X,'CRITICAL DESIGN FACTORS',52X,'**')               SS008190
    WRITE(6,602)                                                            SS008200
    WRITE(6,603)                                                            SS008210
    WRITE(6,811)                                                            SS008220
811 FORMAT(' **',23(' '), 'RELEASE STRESSES',24(' '), '**',21(' '), 'SERVICSS008230
    1E LOAD STRESSES',21(' '), '**'/ ' **',12X,'(SYMBOL X DENOTES STRESS ATSS008240
    2 ALLOWABLE)',13X,'**',13X,'(SYMECL X DENOTES STRESS AT ALLOWABLE)',SS008250
    312X,'**'/ ' **',63X,'**',63X,'**')                                     SS008260
    WRITE(6,812)                                                            SS008270
812 FORMAT(' **',2X,'SECTION **',7X,'STRESS TOP',8X,'**',6X,'STRESS BOTTS008280
    10M',6X,'**',2X,'SECTION **',7X,'STRESS TOP',8X,'**',6X,'STRESS BOTTS008290
    2M',6X,'**'/ ' **',11X,'**',9X,'(KSI)',11X,'**',10X,'(KSI)',10X,'**',11X,SS008300
    3 '**',9X,'(KSI)',11X,'**',10X,'(KSI)',10X,'**'/ ' **',11X,'**',25X,'**',25SS008310
    4X,'**',11X,'**',25X,'**',25X,'**')                                     SS008320
    DO 801 J1=1,6                                                            SS008330
    XSX1=(J1-1)/20.*ZL                                                       SS008340
    XSX2=(J1-1)/10.*ZL                                                       SS008350

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ZMJ=W*XSX1/2.*(ZL-XSX1)	SS008360
ZMJB=WB*XSX2/2.*(ZL-XSX2)	SS008370
ZMJBK=DLMCM(J1)+BMMAX(J1)	SS008380
IF(J1.GT.3) ZMJBK=DLMCM(J1+1)+BMMAX(J1+1)	SS008390
J3=(10-(J1-1)*4)*NRAV	SS008400
IF(J1.GT.3) J3=0	SS008410
SUM1=0.0	SS008420
SUM2=0.0	SS008430
SUM3=0.0	SS008440
SUM4=0.0	SS008450
DO 804 J2=1,NRAV	SS008460
SUM1=SUM1+(-(1./ACONC+D(J2)/ZT)*X((12-2*J1)*NRAV+J2))	SS008470
SUM2=SUM2+(-(1./ACONC=D(J2)/ZB)*X((12-2*J1)*NRAV+J2))	SS008480
SUM3=SUM3+(-(1./ACONCK+D(J2)/ZTK)*X(J3+J2))	SS008490
804 SUM4=SUM4+(-(1./ACONCK=D(J2)/ZBK)*X(J3+J2))	SS008500
STRESS(J1,1)=((1.-ZINLOS)*F0*SUM1-ZMJ*12./ZT)*(-1.)	SS008510
STRESS(J1,2)=((1.-ZINLOS)*F0*SUM2+ZMJ*12./ZB)*(-1.)	SS008520
STRESS(J1,3)=((1.-ZLOSS)*F0*SUM3-ZMJB*12./ZT-ZMJBK*12./ZTK)*(-1.)	SS008530
STRESS(J1,4)=((1.-ZLOSS)*F0*SUM4+ZMJB*12./ZB+ZMJBK*12./ZBK)*(-1.)	SS008540
STR=CTR1	SS008550
SCR=CBR1	SS008560
STS=CTS1	SS008570
SCS=CBS1	SS008580
IF(XSX1.LE.ZL/10.+0.1) STR=CTR2	SS008590
IF(XSX1.LE.ZL/10.+0.1) SCR=CBR2	SS008600
IF(XSX2.LE.ZL/10.+0.1) STS=CTS2	SS008610
IF(XSX2.LE.ZL/10.+0.1) SCS=CBS2	SS008620
C	SS008621
C IF STRESS WITHIN 1 PERCENT OF ALLOWABLE, CALL IT CRITICAL.	SS008622
C	SS008623
IF(STRESS(J1,1).LE.-.00099*STR*SQRT(FPCR*1000.)) KSYM(J1,1)=EX(2)	SS008630
IF(STRESS(J1,2).GE..99*SCR*FPCR) KSYM(J1,2)=EX(2)	SS008640
IF(STRESS(J1,3).GE..99*SCS*F28) KSYM(J1,3)=EX(2)	SS008650
IF(STRESS(J1,4).LE.-.00099*STS*SQRT(F28*1000.)) KSYM(J1,4)=EX(2)	SS008660
IF(J1.NE.1) GO TO 9053	SS008670
IF(STRESS(1,3).LE.-.00099*STS*SQRT(F28*1000.)) KSYM(1,3)=EX(2)	SS008680
IF(STRESS(1,4).GE..99*SCS*F28) KSYM(1,4)=EX(2)	SS008690
9053 CONTINUE	SS008700
J4=J1-1	SS008710
WRITE(6,813) J4,STRESS(J1,1),KSYM(J1,1),STRESS(J1,2),KSYM(J1,2),	SS008720

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      *J4,STRESS(J1,3),KSYM(J1,3),STRESS(J1,4),KSYM(J1,4)          SS008730
      IF(X(KK=J1+1).EQ.0.) KSYM(J1,5)=EX(2)                        SS008740
801  CONTINUE                                                         SS008750
813  FORMAT(' **',4X,I1,'/20',3X,'**',6X,E11.4,3X,A1,4X,'**',6X,E11.4,3X,A SS008760
      11,4X,'**',4X,I1,'/10',3X,'**',6X,E11.4,3X,A1,4X,'**',6X,E11.4,3X,A1, SS008770
      24X,'**')                                                       SS008780
      WRITE(6,814)                                                     SS008790
814  FORMAT(' **',11X,'**',25X,'**',25X,'**',11X,'**',25X,'**',25X,'**'/ ' **', SS008800
      111(' '),**',25(' '),**',25(' '),**',11(' '),**',25(' '),**',25(' ') SS008810
      2),'**')                                                         SS008820
      WRITE(6,603)                                                     SS008830
      WRITE(6,822)                                                     SS008840
822  FORMAT(1X,'**',50(' '),*LIST OF DESIGN CCONSTRAINTS*,51(' '),**'/1X, SS008850
      1**',37X,'(SYMBOL X INDICATES CONSTRAINT CONTROLS FINAL DESIGN)', SS008860
      237X,'**')                                                       SS008870
      WRITE(6,603)                                                     SS008880
      IF(CBRI.GE.ALDEF-0.05) KSYM(6,5)=EX(2)                        SS008881
      IF(CBRI.LE.DEFMIN+0.05) KSYM(5,5)=EX(2)                      SS008882
      WRITE(6,805) KSYM(2,5),KSYM(4,5),KSYM(5,5),KSYM(1,5),KSYM(3,5), SS008890
      *KSYM(6,5)                                                       SS008900
805  FORMAT(1X,'**',16X,*MINIMUM CONCRETE STRENGTH*,3X,A1,11X,*ULTIMATE SS008910
      1MOMENT*,3X,A1,11X,*MINIMUM INITIAL CAMBER*,3X,A1,15X,'**'/1X,'**', SS008920
      216X,*MAXIMUM CONCRETE STRENGTH*,3X,A1,11X,*CRACKING MOMENT*,3X,A1, SS008930
      3 11X,*MAXIMUM INITIAL CAMBER*,3X,A1,15X,'**')                SS008940
      WRITE(6,603)                                                     SS008950
      WRITE(6,602)                                                     SS008960
      SS008970
C    DESIGN SHEAR AND MOMENTS AT TENTH POINTS                      SS008980
C    SS008990
C    WRITE(6,9045)                                                     SS009000
9045  FORMAT(/)                                                       SS009010
      WRITE(6,602)                                                     SS009020
      WRITE(6,9041)                                                     SS009030
9041  FORMAT(1X,'**',51X,*MOMENT AND SHEAR SUMMARY*,52X,'**')        SS009040
      WRITE(6,602)                                                     SS009050
      WRITE(6,603)                                                     SS009060
      WRITE(6,9042)                                                     SS009070
9042  FORMAT(1X,'**',5X,*SECTION*,5X,'**',6X,*BEAM WT.*,7X,'**',21X,'**',21X SS009080
      1,'**',21X,'**',21X,'**'/1X,'**',17X,'**',3X,*PLUS SHEAR KEY*,4X,'**',6X, SS009090
      2*OTHER D.L.*,5X,'**',9X,*L.L.*,8X,'**',8X,*TOTAL*,8X,'**',7X,*ULTIMAT SS009100

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3E*,6X, '**/' *,17X, '**,4(7X, 'MOMENTS',7X, '**),8X, 'SHEAR',8X, '**/' SS009110
4** , 17X, '**,4(7X, '(KIP=FT)',6X, '**),8X, '(KIPS)',7X, '**/1X, '**,17X, SS009120
5** ,5(21X, '**)) SS009130
DO 9040 J1=1,7 SS009140
ZX=(J1-1)*ZL/10. SS009150
IF(J1.GT.3) ZX=(J1-2)*ZL/10. SS009160
IF(J1.EQ.4) ZX=ZL/4. SS009170
J2=J1-1 SS009180
IF(J1.GT.4) J2=J1-2 SS009190
ZMJB=WB*ZX/2.*(ZL-ZX) SS009200
ZMJT=ZMJB+DLMOM(J1)+BMMA(X(J1)) SS009210
VU(J1)=1.444*(WB*(ZL/2.-ZX)+DL SHR(J1)+5./3.*BVMAX(J1)) SS009220
IF(J1.EQ.4) WRITE(6,9043) ZMJB,DLMOM(J1),BMMA(X(J1),ZMJT,VU(J1) SS009230
IF(J1.NE.4)WRITE(6,9047) J2,ZMJB,DLMOM(J1),BMMA(X(J1),ZMJT,VU(J1) SS009240
9040 CONTINUE SS009250
9047 FORMAT(1X, '**,6X, I1, '/10',7X, '**,5(4X,E12.5,5X, '**)) SS009260
9043 FORMAT(1X, '**,6X, '1/4',8X, '**,5(4X,E12.5,5X, '**)) SS009270
WRITE(6,603) SS009280
WRITE(6,9044) ULTMREQ,ZMUL,ZCRACK SS009290
9044 FORMAT(1X, '**,42X, 'ULTIMATE MOMENT REQUIRED = ',E12.5, ' KIP=FT',39SS009300
1X, '**/1X, '**,42X, 'ULTIMATE MOMENT CAPACITY = ',E12.5, ' KIP=FT',39XSS009310
2, '**/1X, '**,42X, 'CRACKING MOMENT CAPACITY = ',E12.5, ' KIP=FT',39X,SS009320
3**)) SS009330
WRITE(6,603) SS009340
WRITE(6,602) SS009350
WRITE(6,641) SS009360
641 FORMAT(1H1) SS009370
NCOUNT=NCOUNT+1 SS009380
GO TO 3007 SS009390
2500 CONTINUE SS009400
STOP SS009410
END SS009420

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SUBROUTINE EQGEN
COMMON/BLK1/ ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,TDIM,
1WDDIM,WHDIM,XDIM,YDIM,ACONC,BINERT,DTOP,DBOT,ZT,ZB,ACONCK,
2BINERK,DTOPK,DBCTK,ZTK,ZBK,ZL,F2B,JOPT,ASSCLR,ASSPLS,APRIME,
3CBOT,COSTWP,COSTFT,DEFMIN,ALDEF,NRAV,NWHEEL,DISTF,FPCMAX,
4FPCMIN,ELASC,ULTMRQ,CTOP,W,WB,F0,DCR
COMMON/BLK2/ PAXLE(18),NWHL(18),STRMAX(26),FCONC(10),DCONC(10),
1G(11),F(11),D(10),GRIDS(26),BMMAX(7),EVMAX(7),CBRMAX(5),PRLMAX(5),
2DLNOM(7),DLSHR(7),PECRK(11,4),ZLOS(4),ZWRAP(10,3),NSDIF(4)
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2,
1CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP
COMMON/YZ/Z1,Z2,Z3,Z4,Y1,Y2,Y3,Y4
COMMON/DUMP/ TITLE(3,54),YJ(11),FROW(26),
1PEF(50,3),ZNE(11),KKODE(4),ZMCR(4)
COMMON/D314/ N,M,ARRAY(156,276),B(276),X(276),XD(276),OBJ,KP1,K
JRR=28+11*NRAV
JCC=1+11*NRAV
IF(JOPT.NE.0) JRR=46+11*NRAV
IF(JOPT.NE.0) JCC=10+11*NRAV
DO 2 J1=1,JRR
ARRAY(J1,K+1)=0.0
DO 2 J2=1,JCC
2 ARRAY(J1,J2)=0.
C*****SS009660
C OBJECTIVE FUNCTION SS009670
C*****SS009680
DO 4 J1=1,NRAV SS009690
DO 5 J2=1,11 SS009700
5 ARRAY(1,(J1-1)*11+J2)=2.*COSTWP*ZL/40. SS009710
4 ARRAY(1,J1)=- (COSTFT*ZL+0.5*COSTWP*ZL) SS009720
ARRAY(1,NRAV*11+1)=-ACONC*ZL*G(1)/1944. SS009730
IF(JOPT.EQ.0) GO TO 8 SS009740
DO 6 J1=1,10 SS009750
6 ARRAY(1,11*NRAV+J1)=-ACONC*ZL*(G(J1+1)-G(J1))/1944. SS009760
8 CONTINUE SS009770
C*****SS009780
C RELEASE STRESSES = CONSTRAINTS 1 THRU 12 SS009790
C*****SS009800
F0 = 0.7*AS*FPS SS009810
W=UWC*ACONC/144. SS009820

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DO 16 J1=1,11,2
ZX=(11-J1)*ZL/40.
ZMJ=0.5*W*(ZL*ZX=ZX**2)*12.
JR=1+J1
ST=CTR1
SC=CBR1
IF(ZX.LE.ZL/10.+.1) ST=CTR2
IF(ZX.LE.ZL/10.+.1) SC=CBR2
DO 10 J2=1,NRAV
ARRAY(JR,(J1=1)*NRAV+J2)=-{(1.-ASSCLR)*FO*(1./ACONC+D(J2)/ZT)}
10 ARRAY(JR+1,(J1=1)*NRAV+J2)={(1.-ASSCLR)*FO*(1./ACONC-D(J2)/ZB)}
IF(JOPT.EQ.0) GO TO 14
DO 12 J2=1,10
ARRAY(JR,NRAV*11+J2)=-.0074535*ST
12 ARRAY(JR+1,NRAV*11+J2)=-SC
ARRAY(JR,K+1)=ZMJ/ZT+.063366*ST
ARRAY(JR+1,K+1)=ZMJ/ZB+4.0*SC
GO TO 16
14 ARRAY(JR,NRAV*11+1)=-.0074535*ST
ARRAY(JR+1,NRAV*11+1)=-SC
ARRAY(JR,K+1)=ZMJ/ZT+.033552*ST
ARRAY(JR+1,K+1)=ZMJ/ZB
16 CONTINUE
C *****
C SERVICE LOAD STRESSES - CONSTRAINTS 13 THRU 20
C *****
C CONSTRAINTS 13 THRU 18
WB=UWC*ACONCK/144.
DO 24 J1=1,7,2
IF(J1.EQ.5) GO TO 24
IF(J1.EQ.1) ZMJB=BMMA(7)*12.+DLMOM(7)*12.
IF(J1.EQ.3) ZMJB=BMMA(3)*12.+DLMOM(3)*12.
IF(J1.EQ.7) ZMJB=BMMA(2)*12.+DLMOM(2)*12.
ZX=(11-J1)*ZL/40.
IF(J1.EQ.1) ZX=ZL/2.
ZMJ=0.5*WB*(ZL*ZX=ZX**2)*12.
JR=13+J1
IF(J1.EQ.7) JR=18
ST=CTS1
SC=CBS1

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SS009830
SS009840
SS009850
SS009860
SS009870
SS009880
SS009890
SS009900
SS009910
SS009920
SS009930
SS009940
SS009950
SS009960
SS009970
SS009980
SS009990
SS010000
SS010010
SS010020
SS010030
SS010040
SS010050
SS010060
SS010070
SS010080
SS010090
SS010100
SS010110
SS010120
SS010130
SS010140
SS010150
SS010160
SS010170
SS010180
SS010190
SS010200
SS010210
SS010220

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IF(ZX.LE.ZL/10.+.1) ST=CTS2	SS010230
IF(ZX.LE.ZL/10.+.1) SC=CBS2	SS010240
DO 18 J2=1,NRAV	SS010250
ARRAY(JR,(J1=1)*NRAV+J2)=(1.-ASSPLS)*FO*(1./ACONC+D(J2)/ZT)	SS010260
18 ARRAY(JR+1,(J1=1)*NRAV+J2)=- (1.-ASSPLS)*FO*(1./ACONC-D(J2)/ZB)	SS010270
IF(JOPT.EQ.0) GO TO 22	SS010280
DO 20 J2=1,10	SS010290
ARRAY(JR,NRAV*11+J2)=-2.*SC*(F(J2+1)-F(J2))	SS010300
20 ARRAY(JR+1,NRAV*11+J2)=-.014907*ST*(F(J2+1)-F(J2))	SS010310
ARRAY(JR,K+1)=-ZMJ/ZT-ZMJB/ZTK+SC*F(1)	SS010320
ARRAY(JR+1,K+1)=-ZMJ/ZB-ZMJB/ZBK+.0074535*ST*F(1)+.033552*ST	SS010330
GO TO 24	SS010340
22 ARRAY(JR,K+1)=-ZMJ/ZT-ZMJB/ZTK+SC*F28	SS010350
ARRAY(JR+1,K+1)=-ZMJ/ZB-ZMJB/ZBK+.031623*SQRT(F28)*ST	SS010360
24 CONTINUE	SS010370
C CONSTRAINTS 19 AND 20	SS010380
DO 26 J2=1,NRAV	SS010390
ARRAY(20,NRAV*10+J2)=- (1.-ASSPLS)*(1./ACONC+D(J2)/ZT)*FO	SS010400
26 ARRAY(21,NRAV*10+J2)=(1.-ASSPLS)*(1./ACONC-D(J2)/ZB)*FO	SS010410
IF(JOPT.EQ.0) GO TO 30	SS010420
DO 28 J2=1,10	SS010430
ARRAY(20,NRAV*11+J2)=-.014907*CTS2*(F(J2+1)-F(J2))	SS010440
28 ARRAY(21,NRAV*11+J2)=-2.*CBS2*(F(J2+1)-F(J2))	SS010450
ARRAY(20,K+1)=.0074535*CTS2*F(1)+.033552*CTS2	SS010460
ARRAY(21,K+1)=CBS2*F(1)	SS010470
GO TO 32	SS010480
30 ARRAY(20,K+1)=CTS2*.031623*CTS2*SQRT(F28)	SS010490
ARRAY(21,K+1)=CBS2*F28	SS010500
C*****	SS010510
C STRAND WRAPPING CONSTRAINTS = CONSTRAINTS 21 THRU (20+10*NRAV)	SS010520
C*****	SS010530
32 DO 34 J1=1,NRAV	SS010540
DO 34 J2=1,10	SS010550
ARRAY(21+(J1=1)*10+J2,NRAV*J2+J1)=1.	SS010560
34 ARRAY(21+(J1=1)*10+J2,NRAV*(J2=1)+J1)=-1.	SS010570
C*****	SS010580
C MAXIMUM NUMBER OF STRANDS PER ROW = CONSTRAINTS (21+10*NRAV)	SS010590
C THRU (20+11*NRAV)	SS010600
C*****	SS010610
DO 36 J1=1,NRAV	SS010620

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      ARRAY(21+10*NRAV+J1,J1)=1.                                SS010630
      ARRAY(21+10*NRAV+J1,K+1)=STRMAX(J1)                      SS010640
36  CONTINUE                                                    SS010650
C*****SS010660
C      JOPT=1, PROPER RELEASE STRENGTH REPRESENTATION = CONSTRAINTS SS010670
C      (21+11*NRAV) THRU (39+11*NRAV)                          SS010680
C*****SS010690
      IF(JOPT.NE.1) GO TO 44                                     SS010700
      DO 38 J1=1,10                                             SS010710
      ARRAY(21+11*NRAV+J1,11*NRAV+J1)=1.                      SS010720
38  ARRAY(21+11*NRAV+J1,K+1)=0.5                               SS010730
      DO 40 J1=1,9                                              SS010740
      ARRAY(31+11*NRAV+J1,11*NRAV+J1)=-1.                     SS010750
40  ARRAY(31+11*NRAV+J1,11*NRAV+J1+1)=1.                     SS010760
C*****SS010770
C      BOUNDS ON INITIAL CAMBER                                SS010780
C      JOPT=0,CONSTRAINTS (21+11*NRAV) THRU (22+11*NRAV)      SS010790
C      JOPT=1,CONSTRAINTS (40+11*NRAV) THRU (41+11*NRAV)      SS010800
C*****SS010810
44  YJ(1)=3.*(ZL*12.)/8.                                       SS010820
      DO 46 J1=2,11                                             SS010830
46  YJ(J1)=(11-J1)*(ZL*12.)/40.+(ZL*12.)/80.                SS010840
      IF(JOPT.EQ.0) JR=22+11*NRAV                               SS010850
      IF(JOPT.NE.0) JR=41+11*NRAV                               SS010860
      DO 48 J1=1,11                                             SS010870
      DELTAJ=ZL*12./40.                                         SS010880
      IF(J1.EQ.1) DELTAJ=ZL*12./4.                             SS010890
      DO 48 J2=1,NRAV                                           SS010900
      ARRAY(JR,(J1-1)*NRAV+J2)=- (1.-ASSCLR)*FO*YJ(J1)*D(J2)*DELTAJ SS010910
48  ARRAY(JR+1,(J1-1)*NRAV+J2)=ARRAY(JR,(J1-1)*NRAV+J2)      SS010920
      IF(JOPT.EQ.0) GO TO 52                                     SS010930
      DO 50 J1=1,10                                             SS010940
      ARRAY(JR,11*NRAV+J1)=-.235252 *BINERT*ELASC*ALDEF       SS010950
50  ARRAY(JR+1,11*NRAV+J1)=.235252 *BINERT*DEFMIN*ELASC       SS010960
      ARRAY(JR,K+1)=2.0 *ELASC*ALDEF*BINERT+22.5*W*ZL**4      SS010970
      ARRAY(JR+1,K+1)=-2.0 *ELASC*DEFMIN*BINERT-22.5*W*ZL**4  SS010980
      GO TO 60                                                  SS010990
52  ARRAY(JR,11*NRAV+1)=-.235252 *BINERT*ELASC*ALDEF         SS011000
      ARRAY(JR+1,11*NRAV+1)=.235252 *BINERT*ELASC*DEFMIN      SS011010
      ARRAY(JR,K+1)=1.058991*BINERT*ELASC*ALDEF+22.5*W*ZL**4  SS011020

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	ARRAY(JR+1,K+1)=-1.058991*BINERT*ELASC*DEFMIN=22.5*W*ZL**4	SS011030
C		SS011040
60	ARRAY(JR,K+1)=ARRAY(JR,K+1)/1.E+04	SS011050
	ARRAY(JR+1,K+1)=ARRAY(JR+1,K+1)/1.E+04	SS011060
	DO 56 J1=1,M	SS011070
	ARRAY(JR,J1)=ARRAY(JR,J1)/1.E+04	SS011080
56	ARRAY(JR+1,J1)=ARRAY(JR+1,J1)/1.E+04	SS011090
		SS011100
C	*****	SS011110
C	ULTIMATE MOMENT AND CRACKING MOMENT CONSTRAINTS	SS011120
C	JOPT=0, CONSTRAINTS (23+11*NRAV) AND (24+11*NRAV)	SS011130
C	JOPT=1, CONSTRAINTS (42+11*NRAV) AND (43+11*NRAV)	SS011140
C	*****	SS011150
C		SS011160
C	SET UP NO. STRANDS AND STRAND ECCENTRICITY ARRAY	SS011170
C		SS011180
	DO 62 J1=1,NRAV	SS011190
62	FROW(J1)=0.	SS011200
	SUM=0.	SS011210
	KNT=0	SS011220
	PF=0.	SS011230
	DO 64 J1=1,NRAV	SS011240
67	IF(FROW(J1).EQ.STRMAX(J1)) GO TO 64	SS011250
	IADD=2	SS011260
	IF(STRMAX(J1)-FROW(J1).LE.1) IADD=1	SS011270
	FROW(J1)=FROW(J1)+IADD	SS011280
	PF=PF+IADD*(=D(J1))	SS011290
	SUM=SUM+IADD	SS011300
	KNT=KNT+1	SS011310
	PEF(KNT,1)=SUM	SS011320
	PEF(KNT,2)=PF	SS011330
	PEF(KNT,3)=PF/SUM	SS011340
	GO TO 67	SS011350
64	CONTINUE	SS011360
C	SET UP FOR CALLS TO ULTMP	SS011370
	ZMDL=W*ZL**2/8.	SS011380
	DO 63 J1=1,4	SS011390
63	ZLOS(J1)=0.1*J1	SS011400
	DO 65 J1=1,11	SS011410
	ZNE(J1)=0.	SS011420

DO 65 J2=1,4	SS011430
65 PECRK(J1,J2)=0.	SS011440
JSTOP=11	SS011450
IF(JOPT.NE.0) GO TO 66	SS011460
F(1)=F28	SS011470
JSTOP=1	SS011480
C GENERATE TOTAL FORCE ECCENTRICITIES FOR ULTIMATE MOMENT AND CRACK	SS011490
C MOMENT CONSTRAINTS	SS011500
66 FPCMIN=4.0	SS011510
DO 84 J1=1,JSTOP	SS011520
IF(J1.EQ.1) GO TO 69	SS011530
IF(F(J1).NE.F(J1-1)) GO TO 69	SS011540
ZNE(J1)=ZNE(J1-1)	SS011550
DO 68 J2=1,4	SS011560
68 PECRK(J1,J2)=PECRK(J1-1,J2)	SS011570
GO TO 84	SS011580
69 FPCBM=F(J1)	SS011590
DO 70 J2=1,4	SS011600
70 KKODE(J2)=0	SS011610
KODEMU=0	SS011620
ZMOLD=0.	SS011630
DO 82 J2=1,KNT	SS011640
ASTL=AS*PEF(J2,1)	SS011650
DD=DTOP+PEF(J2,3)	SS011660
CALL ULTMP(ASTL,FPCBM,FPS,APRIME,FPL,CD,DDIM,FSY,DCR,	SS011670
*Y1,Y2,Y3,Y4,Z1,Z2,Z3,Z4,ZMUL)	SS011680
DO 72 J3=1,4	SS011690
ZMCR(J3)=(1.-ZLOS(J3))*ZBK*FO*(PEF(J2,1)/ACONC+PEF(J2,2)/ZB)	SS011700
1+7.5*ZBK*.031623*SQR(T(FPCBM)-ZBK*ZMDL*12./ZB	SS011710
72 ZMCR(J3)=ZMCR(J3)*1.2/12.	SS011720
DO 73 J3=1,4	SS011730
IF(KKODE(J3).EQ.1) GO TO 73	SS011740
IF(ZMUL.LT.ZMCR(J3)) GO TO 73	SS011750
PECRK(J1,J3)=PEF(J2,2)	SS011760
KKODE(J3)=1	SS011770
73 CONTINUE	SS011780
74 IF(KODEMU.EQ.1) GO TO 78	SS011790
IF(ZMUL.GE.ULTMRG) GO TO 76	SS011800
E1=ZMOLD	SS011810
IF(E1.LT.ZMUL) ZMOLD=ZMUL	SS011820

IF(E1.LT.ZMUL) GO TO 82	SS011830
FPCMIN=4.0+(J1-1)*0.5	SS011840
GO TO 84	SS011850
76 ZNE(J1)=PEF(J2,2)	SS011860
KODEMU=1	SS011870
78 DO 80 J3=1,4	SS011880
IF(KKODE(J3).EQ.0) GO TO 82	SS011890
80 CONTINUE	SS011900
IF(KODEMU.EQ.1) GO TO 84	SS011910
82 CONTINUE	SS011920
84 CONTINUE	SS011930
C	SS011940
C FORM ULTIMATE MOMENT AND CRACKING MOMENT CONSTRAINTS	SS011950
C	SS011960
IF(JOPT.EQ.0) JR=1+23+11*NRAV	SS011970
IF(JOPT.NE.0) JR=1+42+11*NRAV	SS011980
DO 86 J1=1,NRAV	SS011990
ARRAY(JR,J1)=D(J1)	SS012000
86 ARRAY(JR+1,J1)=D(J1)	SS012010
IF(JOPT.EQ.0) GO TO 90	SS012020
DO 88 J1=1,10	SS012030
ARRAY(JR,11*NRAV+J1)=2.*(ZNE(J1+1)-ZNE(J1))	SS012040
88 ARRAY(JR+1,11*NRAV+J1)=2.*(PECRK(J1+1,1)-PECRK(J1,1))	SS012050
ARRAY(JR,K+1)=-ZNE(1)	SS012060
ARRAY(JR+1,K+1)=-PECRK(1,1)	SS012070
GO TO 92	SS012080
90 ARRAY(JR,K+1)=-ZNE(1)	SS012090
ARRAY(JR+1,K+1)=-PECRK(1,1)	SS012100
C*****	SS012110
C MINIMUM AND MAXIMUM CONCRETE STRENGTH CONSTRAINTS	SS012120
C JOPT=0, CONSTRAINTS (25+11*NRAV) AND (26+11*NRAV)	SS012130
C JOPT=1, CONSTRAINTS (44+11*NRAV) AND (45+11*NRAV)	SS012140
C*****	SS012150
IF(JOPT.NE.0) GO TO 92	SS012160
ARRAY(26+11*NRAV,11*NRAV+1)=1.	SS012170
ARRAY(26+11*NRAV,K+1)=-4.0	SS012180
ARRAY(27+11*NRAV,11*NRAV+1)=1.0	SS012190
ARRAY(27+11*NRAV,K+1)=F28	SS012200
GO TO 96	SS012210
92 DO 94 J1=1,10	SS012220

ARRAY(45+11*NRAV,11*NRAV+J1)=-1.
94 ARRAY(46+11*NRAV,11*NRAV+J1)=1.
ARRAY(45+11*NRAV,K+1)=4.0=FPCMIN
ARRAY(46+11*NRAV,K+1)=FPCMAX=4.0
96 CONTINUE
RETURN
END

SS012230
SS012240
SS012250
SS012260
SS012270
SS012280
SS012290

SUBROUTINE DEFIN
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2,
CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP
UWC=.150
HUM=50.
AS=0.153
FPS=270.
FPL=0.63*FPS
CTR1=7.5
CTR2=7.5
CBR1=0.6
CBR2=0.6
CTS1=6.0
CTS2=6.0
CBS1=0.4
CBS2=0.4
CREEP1=0.
CREEP2=0.
SHRK1=0.
SHRK2=0.
RATNOD=6.0
FSY=60.
ASTIRP=0.11
RETURN
END

SS012300
SS012310
SS012320
SS012330
SS012340
SS012350
SS012360
SS012370
SS012380
SS012390
SS012400
SS012410
SS012420
SS012430
SS012440
SS012450
SS012460
SS012470
SS012480
SS012490
SS012500
SS012510
SS012520
SS012530
SS012540

	SUBROUTINE LPCODE (NFRCE,NEQS,INDX,KODE,KBOMB)	SS012550
	COMMON/D314/ N,M,A (156,276),B(276),X(276),XD(276),OBJ,KP1,K	SS012560
C	LINEAR PROGRAMMING ALGORITHM	SS012570
C		SS012580
C		SS012590
C	SET UP MATRIX	SS012600
C		SS012610
	KBOMB=0	SS012620
	KP1=N+M	SS012630
	K=N+M-1	SS012640
	KK=K-1	SS012650
	IF(KODE.NE.0) GO TO 200	SS012660
	DO 1 I=1,N	SS012670
	DO 1 J=M,KK	SS012680
	1 A(I,J+1)=0.0	SS012690
	DO 2 I=2,N	SS012700
	IPM=I+M-1	SS012710
	2 A(I,IPM)=1.0	SS012720
	WRITE(4) A	SS012730
	REWIND 4	SS012740
C		SS012750
C	FLAG BASIS	SS012760
C		SS012770
	200 DO 5 I=1,K	SS012780
	XD(I)=0.0	SS012790
	5 X(I)=0.0	SS012800
	DO 6 I=1,N	SS012810
	IPM=I+M	SS012820
	6 X(IPM)=1.0	SS012830
	DO 7 I=1,N	SS012840
	7 B(I)=0.0	SS012850
	10 CONTINUE	SS012860
	C*****	SS012870
	C**** FEASIBILITY SECTION	SS012880
	C*****	SS012890
	INEG=2	SS012900
	11 DO 14 I=2,N	SS012910
	IF (B(I)) 12,12,14	SS012920
	12 CONTINUE	SS012930
	IF (A(I,KP1)-A(INEG,KP1)) 13,14,14	SS012940

13	INEG=I	SS012950
14	CONTINUE	SS012960
	IF (A(INEG,KP1)) 15,23,23	SS012970
15	IF (B(INEG)) 16,16,23	SS012980
16	JSM=1	SS012990
	DO 19 J=2,K	SS013000
	IF (XD(J)) 17,17,19	SS013010
17	CONTINUE	SS013020
	IF (A(INEG,J)=A(INEG,JSM)) 18,19,19	SS013030
18	JSM=J	SS013040
19	CONTINUE	SS013050
	IF (XD(JSM)) 20,20,23	SS013060
20	IF (A(INEG,JSM)) 22,21,21	SS013070
		SS013080
C		SS013090
C	NO FEASIBLE SOLUTION	SS013100
C		SS013110
21	KBOMB=51	SS013120
	GO TO 38	SS013130
22	CALL PIVOT (INEG,JSM)	SS013140
	GO TO 10	SS013150
C	*****	SS013160
C	OPTIMALITY SECTION	SS013170
C	*****	SS013180
23	JBGST=1	SS013190
C		SS013200
C	SELECT INCOMING VECTOR	SS013210
C		SS013220
	DO 26 J=1,K	SS013230
	IF (XD(J)) 24,24,26	SS013240
24	CONTINUE	SS013250
	IF (A(1,J)=A(1,JBGST)) 26,26,25	SS013260
25	JBGST=J	SS013270
26	CONTINUE	SS013280
	IF (A(1,JBGST)) 38,38,27	SS013290
C		SS013300
C	CHECK FOR UNBOUNDED SOLUTION	SS013310
C		SS013320
27	DO 29 I=2,N	SS013330
	ISPOT=I	SS013340
	IF (B(I)) 28,28,29	

28	CONTINUE	SS013350
	IF (A(I,JBGST)) 29,29,30	SS013360
29	CONTINUE	SS013370
30	CONTINUE	SS013380
	IF (A(ISPOT,JBGST)) 31,31,32	SS013390
31	KBOMB=50	SS013400
	GO TO 38	SS013410
C		SS013420
C	SELECT OUTGOING VECTOR	SS013430
C		SS013440
32	KK=ISPOT	SS013450
	DO 36 I=KK,N	SS013460
	IF (B(I)) 33,33,36	SS013470
33	CONTINUE	SS013480
	IF(A(I,JBGST)) 36,36,34	SS013490
34	IF (A(I,K+1)/A(I,JBGST)-A(ISPOT,K+1)/A(ISPOT,JBGST)) 35,36,36	SS013500
35	ISPOT=I	SS013510
36	CONTINUE	SS013520
	IF (B(ISPOT)) 37,37,31	SS013530
37	CALL PIVCT (ISPOT,JBGST)	SS013540
	GO TO 23	SS013550
C*****		SS013560
C****	OUTPUT SECTION	SS013570
C*****		SS013580
38	OBJ=-A(1,KP1)	SS013590
	IF (INDX=1) 40,39,40	SS013600
39	OBJ=-OBJ	SS013610
40	DO 45 I=1,K	SS013620
	IF (X(I)) 44,44,41	SS013630
41	DO 42 J=2,N	SS013640
	IF (A(J,I)) 42,42,43	SS013650
42	CONTINUE	SS013660
43	X(I)=A(J,KP1)	SS013670
	GO TO 45	SS013680
44	X(I)=0.	SS013690
45	CONTINUE	SS013700
	DO 49 J=1,K	SS013710
	IF (J=N+1) 46,46,47	SS013720
46	JJ=J+M	SS013730
	GO TO 48	SS013740

47 JJ=J=N+1
48 XD(J)=-A(1,JJ)
49 CONTINUE
IF(KBOMB.EQ.50) WRITE(6,50)
50 FORMAT (/1CHOUNBOUNDED)
RETURN
END

SS013750
SS013760
SS013770
SS013780
SS013800
SS013830
SS013850

SUBROUTINE PIVOT (I,J)	SS013860
COMMGN/D314/ N,M,A (156,276),B(276),X(276),XD(276),OBJ,KP1,K	SS013870
DO 2 JJ=1,K	SS013880
IF (X(JJ)) 2,2,1	SS013890
1 IF (A(I,JJ)) 2,2,3	SS013900
2 CONTINUE	SS013910
3 X(JJ)=0.0	SS013920
NM1=N-1	SS013930
R=A(I,J)	SS013940
DO 4 L=1,KP1	SS013950
4 A(I,L)=A(I,L)/R	SS013960
DO 5 L=2,I	SS013970
F=A(L-1,J)	SS013980
DO 5 M=1,KP1	SS013990
5 A(L-1,M)=A(L-1,M)-A(I,M)*F	SS014000
IF (I=N) 6,8,8	SS014010
6 DO 7 L=I,NM1	SS014020
F=A(L+1,J)	SS014030
DO 7 M=1,KP1	SS014040
7 A(L+1,M)=A(L+1,M)-A(I,M)*F	SS014050
8 CONTINUE	SS014060
X(J)=1.0	SS014070
M=KP1-N	SS014080
RETURN	SS014090
END	SS014100

SUBROUTINE PROPTY	SS014110
REAL*4 I1,I2,I3,I4,I5,I6,I7,I8,I9,I10,I11,I12 ,I13,I14,I15	SS014120
COMMON/BLK1/ ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,TDIM,	SS014130
1WDDIM,WHDIM,XDIM,YDIM,ACCNC,BINERT,DTCP,DBOT,ZT,ZB,ACONCK,	SS014140
2BINERK,DTOPK,DBOTK,ZTK,ZBK,ZL,F28,JOPT,ASSCLR,ASSPLS,APRIME,	SS014150
3CBOT,COSTWP,COSTFT,DEFMIN,ALDEF,NRAV,NWHEEL,DISTF,FPCMAX,	SS014160
4FPCMIN,ELASC,ULTMRQ,CTOP,W,WB,FG,DCR	SS014170
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2,	SS014180
1CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP	SS014190
EQUIVALENCE (AREA,ACONC),(YB,DBOT),(YT,DTOP),(YBK,DBOTK),	SS014200
*(YTK,DTOPK),(AREAK,ACONCK),(A,ADIM),(B,BDIM),(C,CDIM),(D,DDIM),	SS014210
*(E,EDIM),(F,FDIM),(G,GDIM),(T,TDIM),(WC,WDCIM),(WH,WHDIM),(H,HDIM)	SS014220
CTOP=DCR	SS014230
C1 = (A=(WH+2.*WD))/2.	SS014240
C2 = (B=(WH +2.*WD))/2.	SS014250
A1 = WD*D	SS014260
A2 = C1*H	SS014270
A3 = C1*G/2.	SS014280
A4 = E*C2/2.	SS014290
A5 = C*C2	SS014300
A6 = WH*T	SS014310
A7 = WH*F	SS014320
A8 = (C2-C1)*H	SS014330
A9 = (C2-C1)*G	SS014340
A10 = A3	SS014350
A11 = C2*(D=H=G=E=C)	SS014360
A12 = A4	SS014370
IF(A.LT.B) GO TO 80	SS014380
A8=0.	SS014390
A9=0.	SS014400
A10=0.	SS014410
A11=0.	SS014420
A12=0.	SS014430
80 CONTINUE	SS014440
A14=(XDIM**2)/2.	SS014450
A15=(YDIM**2)/2.	SS014460
AREA = A1*2. +A2*2. + A3*2. + A4*2. + A5*2. + A6 + A7	SS014470
+2.(RATNOD=1.)*APRIME=.5625	SS014480
AREA1= A1*2. +A2*2. + A3*2. + A4*2. + A5*2. + A6 + A7	SS014490
*+(RATNOD=1.)*APRIME=.5625	SS014500

AREAK= AREA1+ 2.0*A8 + 2.0*A9+2.0*A10+2.0*A11+2.0*A12	SS014510
Y1 = D/2.	SS014520
Y2 = D=H/2.	SS014530
Y3 = D=(H+G/3.)	SS014540
Y4 = C + E/3.	SS014550
Y5 = C/2.	SS014560
Y6 = D=T/2.	SS014570
Y7 = F/2.	SS014580
Y8 = D=(H/2.0)	SS014590
Y9 = D=(H+ G/2.0)	SS014600
Y10= D=(H+ 2.0*G/3.0)	SS014610
Y11 = (D=H=G+E+C)/2.0	SS014620
Y12 = C + 2.0*E/3.0	SS014630
Y14=D=T=XDIM/3.	SS014640
Y15=F+YDIM/3.	SS014650
YB = (Y1*A1*2. + Y2*A2*2. + Y3*A3*2. + Y4*A4*2. + Y5*A5*2. + Y6*A6	SS014660
E + Y7*A7 + (D=CTOP)*2.*(RATNOD=1.)*APRIME)/AREA	SS014670
YT = D=YB	SS014680
YB1= (Y1*A1*2. + Y2*A2*2. + Y3*A3*2. + Y4*A4*2. + Y5*A5*2. + Y6*A6	SS014690
E + Y7*A7 + (D=CTOP)*(RATNOD=1.)*APRIME)/AREA	SS014700
YBK=(YB1*AREA1+Y8*A8*2.+Y9*A9*2.+Y10*A10*2. +Y11*A11*2. +Y12*A12*	SS014710
22.0)/AREAK	SS014720
YTK=D=YBK	SS014730
DO 10 J1=1,2	SS014740
JVKEY=J1-1	SS014750
DY=YB	SS014760
IF(JVKEY.EQ.1) DY = YBK	SS014770
I1 = WD*(D**3)/12. + A1*((Y1=DY)**2)	SS014780
I2 = C1*(H**3)/12. + A2*((Y2=DY)**2)	SS014790
I3 = C1*(G**3)/36. + A3*((Y3=DY)**2)	SS014800
I4 = C2*(E**3)/36. + A4*((Y4=DY)**2)	SS014810
I5 = C2*(C**3)/12. + A5*((Y5=DY)**2)	SS014820
I6 = WH*(T**3)/12. + A6*((Y6=DY)**2)	SS014830
I7 = WH*(F**3)/12. + A7*((Y7=DY)**2)	SS014840
I13=2.*(RATNOD=1.)*APRIME*((D=CTOP=DY)**2)	SS014850
I131= (RATNOD=1.)*APRIME*((D=CTOP=DY)**2)	SS014860
I14=(XDIM**4)/36.+A14*((Y14=DY)**2)	SS014870
I15=(YDIM**4)/36.+A15*((Y15=DY)**2)	SS014880
XINERT = I1*2. + I2*2. + I3*2. + I4*2. + I5*2. + I6 + I7 + I13+I14	SS014890
**2.+I15*2.	SS014900

IF(JVKEY.EQ.0) GO TO 5	SS014910
IF(A.GE.B) XINERK=XINERT+I131-I13	SS014920
IF(A.GE.B) GO TO 5	SS014930
I8 = (C2-C1)*(H**3)/12. + A8*((Y8-DY)**2)	SS014940
I9 = (C2-C1)*(G**3)/12. + A9*((Y9-DY)**2)	SS014950
I10 = C1*(G**3)/36. + A10*((Y10-DY)**2)	SS014960
I11 = C2*((D-H-G-E-C)**3)/12. + A11*((Y11-DY)**2)	SS014970
I12 = C2*(E**3)/36. + A12*((Y12-DY)**2)	SS014980
IF(JVKEY.EQ.1) XINERK =	SS014990
&XINERT+I131-I13 + I8*2.0 + I9*2.0 + I10*2.0 + I11*2.0 + I12*2.0	SS015000
5 CONTINUE	SS015010
IF(JVKEY.EQ.1) GO TO 8	SS015020
ZT=XINERT/YT	SS015030
ZB=XINERT/YB	SS015040
BINERT=XINERT	SS015050
BINERT=BINERT-0.75**4/18.=2.*(.28125*(YB=.50)**2)	SS015060
GO TO 10	SS015070
8 ZTK=XINERK/YTK	SS015080
ZBK=XINERK/YBK	SS015090
BINERK=XINERK	SS015100
BINERK=BINERK-0.75**4/18.=2.*(.28125*(YBK=0.50)**2)	SS015110
10 CONTINUE	SS015120
RETURN	SS015130
END	SS015140

256

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SUBROUTINE MOMSHR(DL,NWHL,NWHEEL,XSEC,PAXLE,MAXMOM,MAXSHR)
REAL*4 MAXMOM, MAXSHR,NWHL,MOMENT
COMMON/DUMP/ MOMENT(12),SHEAR(12),IPL(20),IPR(20),REACT(20)
DIMENSION NWHL(18),PAXLE(18)
NST=NWHEEL-1
DO 11 II = 1,2
  IF(II.EQ.2) XSEC = DL - XSEC
  XSECR= DL-XSEC
DO 3 NS = 1,NST
  IL = NS
  CALL LOCATE(DL,XSEC,NST,NS,NWHL)
  N1 = IPL(IL)
  N2 = IPR(IL)
  IF(N1.EQ.0.AND.N2.EQ.0) PROD = PAXLE(IL+1)*XSECR
  IF(N1.EQ.0.AND.N2.EQ.0) GO TO 33
  IF(N1.EQ.0) N1 = IL+1
  IF(N2.EQ.0) N2 = IL+1
C  OBTAIN THE LEFT REACTION FOR ANY SHIFT
  PROD = 0.
  DO 4 I = N1,N2
    IF(I.EQ.1) D2 = DL-(XSEC-NWHL(IL))
    IF(I.EQ.1) GO TO 36
    IF(I.EQ.(IL+1).AND.IPL(IL).EQ.0) D2 = XSECR
    IF(I.EQ.(IL+1).AND.IPL(IL).EQ.0) GO TO 36
    IF(I.LE.IL) D2 = DL-(XSEC-(NWHL(IL)-NWHL(I-1)))
    IF(I.LE.IL) GO TO 36
    IF(I.GT.IL) D2 = XSECR-(NWHL(I-1)-NWHL(IL))
  36 CONTINUE
  DELT = PAXLE(I)*D2
  4 PROD = PROD+DELT
  33 CONTINUE
  REACT(IL) = PROD/DL
  SUMV = 0.
  SUMM = 0.
  IF(IPL(IL).EQ.0) SHEAR(IL) = REACT(IL)
  IF(IPL(IL).EQ.0) MOMENT(IL) = REACT(IL) *XSEC
  IF(IPL(IL).EQ.0) GO TO 3
  DO 5 I = N1,IL
    IF(I.EQ.1) DM = NWHL(IL)
    IF(I.EQ.1) GO TO 34

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SS015150
SS015160
SS015170
SS015180
SS015190
SS015200
SS015210
SS015220
SS015230
SS015240
SS015250
SS015260
SS015270
SS015280
SS015290
SS015300
SS015310
SS015320
SS015330
SS015340
SS015350
SS015360
SS015370
SS015380
SS015390
SS015400
SS015410
SS015420
SS015430
SS015440
SS015450
SS015460
SS015470
SS015480
SS015490
SS015500
SS015510
SS015520
SS015530
SS015540

DM = NWPL(IL)-NWHL(I-1)	SS015550
34 DELTM = PAXLE(I)*DM	SS015560
DELT V = PAXLE(I)	SS015570
SUMM = SUMM+DELT M	SS015580
SUMV = SUMV+DELT V	SS015590
5 CONTINUE	SS015600
SHEAR(IL) = REACT(IL)-SUMV	SS015610
MOMENT(IL) = REACT(IL)*XSEC-SUMM	SS015620
3 CONTINUE	SS015630
NA = 0	SS015640
IF(II.EQ.1) MAXMOM = MOMENT(1)	SS015650
IF(II.EQ.1) MAXSHR = SHEAR(1)	SS015660
NSTA = NST + 1	SS015670
IF(NSTA.EQ.0) GO TO 16	SS015680
DO 13 LL = 1,NSTA	SS015690
NA = NA + 1	SS015700
NB = LL + 1	SS015710
AAA = MOMENT(NA)	SS015720
BBB = SHEAR(NA)	SS015730
IF(II.EQ.2) AAA = MAXMOM	SS015740
IF(II.EQ.2) BBB = MAXSHR	SS015750
IF(MOMENT(NB).GT.AAA) MAXMOM = MOMENT(NB)	SS015760
IF(ABS(SHEAR(NB)).GT.BBB) MAXSHR = ABS(SHEAR(NB))	SS015770
IF(MOMENT(NB).GT.AAA) GO TO 15	SS015780
NA = NA + 1	SS015790
GO TO 13	SS015800
15 NA = NB + 1	SS015810
13 CONTINUE	SS015820
16 CONTINUE	SS015830
11 CONTINUE	SS015840
XSEC=DL-XSEC	SS015850
RETURN	SS015860
END	SS015870

SUBROUTINE LOCATE(DL,XSEC,NST,NS,NWHL)	SS015880
REAL*4 NWHL,MOMENT	SS015890
COMMON/DUMP/ MOMENT(12),SHEAR(12),IPL(20),IPR(20),REACT(20)	SS015900
DIMENSION NWHL(18)	SS015910
XSECR = DL-XSEC	SS015920
DTERM = 0.	SS015930
DO 1 I = 1,NST	SS015940
DLE = NWHL(NS)-DTERM	SS015950
IF(DLE.LE.XSEC) IPL(NS) = 1	SS015960
IF(DLE.LE.XSEC) GO TO 2	SS015970
IF(I.EQ.NS) IPL(NS) = 0	SS015980
IF(I.EQ.NS) GO TO 2	SS015990
DTERM = NWHL(I)	SS016000
1 CONTINUE	SS016010
2 CONTINUE	SS016020
DO 4 IC= 1,NST	SS016030
NSC = NS+IC	SS016040
IF((NS+1).EQ.(NST+1)) IPR(NS) = 0	SS016050
IF(NSC.GT.NST) GO TO 5	SS016060
DELTR = NWHL(NS+IC)-NWHL(NS)	SS016070
IF(DELTR.GT.XSECR.AND.IC.EQ.1) IPR(NS) = 0	SS016080
IF(DELTR.GT.XSECR) GO TO 5	SS016090
IPR(NS) = NS+IC+1	SS016100
4 CONTINUE	SS016110
5 CONTINUE	SS016120
RETURN	SS016130
END	SS016140

SUBROUTINE ULTMF(ASTAR,FPCBM,FPS,ASPRM,FPL,D,DPTH,FSY, DCR,	SS016150
*Y1,Y2,Y3,Y4,Z1,Z2,Z3,Z4,ZMUL)	SS016160
CLONG=0.2	SS016170
ESINI=0.7*FPS*(1.-CLONG)/28.E+03	SS016180
CON1=(FPL/28000.)*(1.+(FPS=FPL)/(FPS=2.*FPL))	SS016190
CON2=-(FPL/28000.)*FPL*(FPS=FPL)**2/(FPS=2.*FPL)	SS016200
BEFF=2.*(Y1+Y2+Y3)	SS016210
THK=Z1	SS016220
Z4MZ3=Z4-Z3	SS016230
IF(ABS(Z4-Z3).LE.1.E-06) Z4MZ3=1.E-06	SS016240
Z2MZ1=Z2-Z1	SS016250
IF(ABS(Z2-Z1).LE.1.E-06) Z2MZ1=1.E-06	SS016260
C*****	SS016270
C**** POSITIVE MOMENT CAPACITY = N.A. IN SLAB	SS016280
C*****	SS016290
C	SS016300
C CHECK TO SEE IF N.A. IN SLAB	SS016310
C	SS016320
PSTAR=ASTAR/(BEFF*D)	SS016330
FSUSTR=FPS*(1.-0.5*PSTAR*FPS/FPCBM)	SS016340
T=ASTAR*FSUSTR	SS016350
CC=.833*FPCBM*BEFF*THK	SS016360
IF(CC.LT.T) GO TO 10	SS016370
C	SS016380
C N.A. IN SLAB	SS016390
C	SS016400
ZMUL =ASTAR*FSUSTR*D*(1.-0.6*PSTAR*FSUSTR/FPCBM)/12	SS016410
RI=PSTAR*FSUSTR/FPCBM	SS016420
IF(RI.GT.0.3)ZMUL =0.25*FPCBM *BEFF*D**2/12.	SS016430
RETURN	SS016440
C*****	SS016450
C**** POSITIVE MOMENT CAPACITY = N.A. BELOW SLAB	SS016460
C*****	SS016470
10 CONTINUE	SS016480
C	SS016490
C BEGIN ITERATION TO LOCATE N.A.	SS016500
C	SS016510
JCNT=0	SS016520
X=0.	SS016530
12 X=X+0.25	SS016540

```

13 JCNT=JCNT+1
   IF(X.GT.DPTH) ZMUL=0.
   IF(X.GT.DPTH) RETURN
C
C   COMPUTE STRAND STRAIN AND FORCE IN DECK STEEL
C
   ES=.003*(D-X)/X+ESINI
   ESP=.003*(X-DCR)/X
   CS=29.E+03*ABS(ESP)
   IF(CS.GT.FSY) CS=FSY
   IF(ESP.LE.0) CS=-CS
   CS=CS*ASPRM
C
C   COMPUTE RESULTANT COMPRESSIVE FORCE ON CONCRETE AND ITS LOCATION
C
   KODE=1
   GO TO 1000
14 DBAR=D-YC
   CC=C*.833*FPCBM
   CTOT=CS+CC
   GO TO 2000
C
C   COMPUTE STRAND STRESS AND STRAND FORCE
C
16 T=ASTAR*FS
   SUMFOR=T-CTOT
   IF(SUMFOR.LT.0.) GO TO 18
   IF(JCNT.EQ.2) GO TO 17
   SAVEF1=SUMFOR
   SAVEX1=X
   GO TO 12
17 SAVEF2=SUMFOR
   SAVEX2=X
   X=SAVEX1+(SAVEX2-SAVEX1)*SAVEF1/(SAVEF1-SAVEF2)
   IF(X=SAVEX1.LT..25) X=SAVEX1+.25
   JCNT=0
   GO TO 13
18 ZMUL=(CC*DBAR+CS*(D-DCR))/12.
   GO TO 28
C*****

```

SS016550
SS016560
SS016570
SS016580
SS016590
SS016600
SS016610
SS016620
SS016630
SS016640
SS016650
SS016660
SS016670
SS016680
SS016690
SS016700
SS016710
SS016720
SS016730
SS016740
SS016750
SS016760
SS016770
SS016780
SS016790
SS016800
SS016810
SS016820
SS016830
SS016840
SS016850
SS016860
SS016870
SS016880
SS016890
SS016900
SS016910
SS016920
SS016930
SS016940

```

C*****THIS SECTION COMPUTES CONCRETE COMPRESSION AREA AND ITS C.G.      SS016950
C*****SS016960
1000 C =(Y1*BRACK(0.,X,Z1)+Y2*X+Y3*BRACK(0.,X,Z3)+BRACK(Z1,X,Z2)*Y4      SS016970
*   =0.5*Y4*BRACK(Z1,X,Z2)**2/Z2MZ1+BRACK(Z3,X,Z4)*Y3=0.5*Y3*      SS016980
*   BRACK(Z3,X,Z4)**2/Z4MZ3)*2.      SS016990
YC=(0.5*Y1*BRACK(0.,X,Z1)**2+0.5*Y2*X**2+0.5*Y3*BRACK(0.,X,Z3)**2      SS017000
*   +Y4*Z1*BRACK(Z1,X,Z2)+0.5*Y4*BRACK(Z1,X,Z2)**2=0.5*Z1*Y4*      SS017010
*   BRACK(Z1,X,Z2)**2/Z2MZ1=.33333*Y4*BRACK(Z1,X,Z2)**3/Z2MZ1      SS017020
*   +Y3*Z3*BRACK(Z3,X,Z4)+0.5*Y3*BRACK(Z3,X,Z4)**2      SS017030
*   =0.5*Z3*Y3*BRACK(Z3,X,Z4)**2/Z4MZ3=.33333*Y3      SS017040
*   *BRACK(Z3,X,Z4)**3/Z4MZ3)*2./C      SS017050
GO TO 14      SS017060
C*****SS017070
C***** THIS SECTION COMPUTES STRAND STRESS      SS017080
C*****SS017090
2000 FS=ES*28000      SS017100
IF(FS.GT.FPL) GO TO 2002      SS017110
2002 FS=.5*FPS+.5*SQRT(FPS**2=4.*CON2/(ES=CON1))      SS017120
GO TO 16      SS017130
28 RETURN      SS017140
END      SS017150

```

FUNCTION BRACK(ZL,X,ZU)
IF(X.LE.ZL) BRACK=0.
IF(ZL.LT.X.AND.X.LE.ZU) BRACK=X-ZL
IF(X.GT.ZU) BRACK=ZU-ZL
RETURN
END

SS017160
SS017170
SS017180
SS017190
SS017200
SS017210

SUBROUTINE PLOSS(FPCR,ZMBW,ZMC,ZMNC,FSU,AS,AB,ZI,ZIC,YB,YBC,EC,
 *HUM,SPAN,ZLOSS,ZINLOS,UWC)

THIS SUBROUTINE COMPUTES PRESTRESS LOSS BY 1975 AASHTO
 INTERIM SPEC.

FPCR = CONCRETE RELEASE STRENGTH (KSI)

ZMBW=D.L. MOMENT DUE TO BEAM WEIGHT AT MIDSPAN(K-FT)

ZMC = TOTAL D.L. MOMENT (EXCEPT BEAM WEIGHT) AT MIDSPAN
 ACTING ON COMPOSITE SECTION(K-FT)

ZMNC = TOTAL D.L. MOMENT (EXCEPT BEAM WEIGHT) AT MIDSPAN
 ACTING ON NONCOMPOSITE SECTION (K-FT)

FSU = ULTIMATE STRENGTH OF STRAND (KSI)

AS = TOTAL STRAND AREA (IN**2)

AB = CROSS SECTIONAL AREA OF BEAM (IN**2)

ZI = M. OF I. OF NONCOMPOSITE BEAM (IN**4)

ZIC = M. OF I. OF COMPOSITE BEAM (IN**4)

YB = DISTANCE FROM C.G. OF BEAM TO BOTTOM FIBER (IN)

YBC = DISTANCE FROM C.G. OF COMPOSITE BEAM TO BOTTOM FIBER (IN)

EC = DISTANCE FROM BOTTOM OF BEAM TO C.G. OF STRANDS (IN)

HUM = RELATIVE HUMIDITY (PERCENT)

SPAN = SPAN LENGTH (FT)

ZINLOS=FRACTION OF INITIAL STRESS(.7*FSU) LOST (RELEASE)

ZLOSS = FRACTION OF INITIAL STRESS (.7*FSU) LOST (SERVICE)

(COMPRESSION STRESS IS POSITIVE)

SHRINKAGE LOSS

SH=(17000.-150*HUM)/1000.

ELASTIC SHORTENING

A 10 PERCENT LOSS IN STRAND FORCE DUE TO RELAXATION AND ELASTIC
 SHORTENING PRIOR TO RELEASE IS ASSUMED AT TIME OF RELEASE

FEFF=0.9*0.7*FSU*AS

FCIR=FEFF/AB+FEFF*(YB-EC)*ABS(YB-EC)/ZI-12.*ZMBW*(YB-EC)/ZI

ECI=(UWC*1000.)*1.5*33.*SQRT(1000.*FPCR)

SS017220
 SS017230
 SS017240
 SS017250
 SS017260
 SS017270
 SS017280
 SS017290
 SS017300
 SS017310
 SS017320
 SS017330
 SS017340
 SS017350
 SS017360
 SS017370
 SS017380
 SS017390
 SS017400
 SS017410
 SS017420
 SS017430
 SS017440
 SS017450
 SS017460
 SS017470
 SS017480
 SS017490
 SS017500
 SS017510
 SS017520
 SS017530
 SS017540
 SS017550
 SS017560
 SS017570
 SS017580
 SS017590
 SS017600
 SS017610

C	ES=(28E+06*FCIR/ECI)	SS017620
C	CREEP LOSS	SS017630
C		SS017640
	FCDS=12.*ZMNC*(YB=EC)/ZI+12.*ZMC*(YBC=EC)/ZIC	SS017650
	CRC=12.*FCIR=7.*FCDS	SS017660
C		SS017670
C	STRAND RELAXATION LOSS	SS017680
C		SS017690
	CRS=20.*0.4*ES=0.2*(SH+CRC)	SS017700
C		SS017710
C	TOTAL LOSS	SS017720
C		SS017730
	DELTFS=SH+ES+CRC+CRS	SS017740
	DELFSI=ES+0.5*CRS	SS017750
C		SS017760
C	LOSS FACTOR	SS017770
C		SS017780
	ZLOSS=DELTFS/(.7*FSU)	SS017790
	ZINLOS=DELFSI/(.7*FSU)	SS017800
	RETURN	SS017810
	END	SS017820
		SS017830

```

SUBROUTINE CAMBER(ES,EC,ASTRN,STRNS,UWB,AREA,SPANL,ECCL,IB,FO,ENDESS017840
*CC,PRLMAX,CBRMAX) SS017850
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2, SS017860
ICBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP SS017870
DIMENSION CNST(4,5),PRLMAX(5),CBRMAX(5) SS017880
DATA CNST/315.,20.,440.,60.,525.,10.,675.,40.,380.,25.,400.,50.,29 SS017890
*0.,25.,460.,50.,0.,0.,0.,0./ SS017900
REAL IE SS017910
C SS017920
C SS017930
C SS017940
C CAMBER AND STRESS LOSS CALCULATIONS SS017950
C MIDSPAN CAMBER AND STRESS LOSS DUE TO INITIAL PRESTRESS AND BEAM SS017960
C SS017970
HDPT=5.0 SS017970
IF(CREEP1.EQ.0.) J1=4 SS017980
IF(CREEP1.EQ.0.) GO TO 2 SS017990
CNST(1,5)=SHRK1 SS018000
CNST(2,5)=SHRK2 SS018010
CNST(3,5)=CREEP1 SS018020
CNST(4,5)=CREEP2 SS018030
J1=5 SS018040
2 DO 1 N=1,J1 SS018050
ASH=0.000001*CNST(1,N) SS018060
BSH=CNST(2,N) SS018070
ACRR=0.000001*CNST(3,N) SS018080
BCR=CNST(4,N) SS018090
ACR = ACRR*0.001 SS018100
RN = ES/EC SS018110
AST = ASTRN*STRNS SS018120
W = UWB*AREA/144. SS018130
DLM = (W*SPANL*SPANL/8.)*12. SS018140
TEMP = 1.+(RN*AST/AREA)+(RN*AST*ECCL*ECCL/IB) SS018150
FR = FO/TEMP +(DLM*ECCL*RN*AST/(IB*TEMP)) SS018160
PLI = ((FO=FR)/FO)*100. SS018170
CONST = (1./AREA)+(ECCL*ECCL/IB) SS018180
FCSO = FR*CONST=(DLM*ECCL/IB) SS018190
STRN1 = ACR*FCSO+ASH SS018200
STRN2 = STRN1-STRN1*(RN*AST*CONST) SS018210
DFCS = STRN2*ES*AST*CONST * 10.0 ** 6 SS018220
STRN4 = ACR*(FCSO-DFCS/2.)+ASH SS018230

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STRN5 = STRN4=STRN4*RN*AST*CONST	SS018240
DFCS1 = STRN5*ES*AST*CCNST * 10.0 ** 6	SS018250
STRN6 = ACR*(FCSD=DFCS1/2.)+ASH	SS018260
STRN7 = STRN6=STRN6*RN*AST*CONST	SS018270
PLINF = (STRN7*ES*AST*10.0**6/FC)*100.	SS018280
PLMAX = PLINF+PLI	SS018290
PRLMAX(N)=PLMAX	SS018300
CCNST = 1./(EC*IB*10.**6)	SS018310
HSPAN = SPANL/2.	SS018320
CI1 = CCNST*(FR*ENDECC *HSPAN*0.5*HSPAN*144.)	SS018330
CI2 = CCNST*(FR*(ECCL=ENDECC)*(HSPAN=HDPT)*0.5*0.67*(HSPAN	SS018340
1=HDPT)*144.0)	SS018350
CI3 = CCNST*(FR*(ECCL=ENDECC)*HDPT*(HSPAN=HDPT/2.)*144.)	SS018360
CI4 = CCNST*{(5./384.)*(W*SPANL*SPANL*SPANL*SPANL*12.*12.*12.)}	SS018370
CI = CI1 +CI2 +CI3 =CI4	SS018380
STRAIN=FCSD/(EC*10.**6)	SS018390
CMAx = CI*{(ACR*(FCSD=(DFCS/2.))+STRAIN)/STRAIN)*(1.-(PLINF/100.))	SS018400
CBRMAX(N)=CMAx	SS018410
1 CONTINUE	SS018420
RETURN	SS018430
END	SS018440

SUBROUTINE SHEAR(B,DEPTH,D,FPC,FSY,AREA,VU,SPACE)
AV=2.*AREA
S1=(AV*FSY)/(0.100*B)
SMAX=0.75*DEPTH
IF(S1.LT.SMAX) SMAX=S1
RJ=0.90
VCMAX=0.180*B*RJ*D
VC=0.06*FPC*B*RJ*D
IF(VC.GT.VCMAX) VC=VCMAX
SPACE=(2.*AV*FSY*RJ*D)/(VU-VC)
IF(SPACE.LT.0.0.OR.SPACE.GT.SMAX) SPACE=SMAX
RETURN
END

SS018450
SS018460
SS018470
SS018480
SS018490
SS018500
SS018510
SS018520
SS018530
SS018540
SS018550
SS018560
SS018570

```

INTEGER*2 ROW,COL
COMMON/BLK1/ADIM,EDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,TDIM,WDDIM,
1WHDIM,XDIM,YDIM,ACONC,BINERT,DTOP,DBOT,ZT,ZB,ACONCK,BINERK,DTOPK,
2DBOTK,ZTK,ZBK,ZL,F28,JOPT,ASSCLR,ASSPLS,APRIME,CBOT,COSTFT,
3DEFMIN,ALDEF,NRAV,NWHEEL,DISTF,FPCMAX,FPCMIN,ELASC,ULTMRQ,CTOP,
4W,WB,FO,DCR,NR,HDPT,NW,ALPHA
COMMON/BLK2/PAXLE(18),NWHL(18),STRMAX(26),FCONC(10),DCONC(10),
1G(11),F(11),D(10),BMMAX(9),BVMAX(9),CBRMAX(5),PRLMAX(5),DLMOM(9),
2DLSHR(9),PECRK(11,4),ZLOS(4)
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2,
1CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP,GSP
COMMON/YZ/Z1,Z2,Z3,Z4,Y1,Y2,Y3,Y4
COMMON/D314/N,M,OBJ,KP1,K,NCON,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,
1SUM,IGNOR,IT(4),JCONT,X(150),Y(3136),ROW(3136),COL(150),BB(150),
2C(150),B(150),XX(150)
COMMON/DUMP/
TITLE(3,54),YJ(11),FROW(26),
1PEF(50,3),ZNE(11),KKODE(4),ZMCR(4)
DIMENSION STRESS(9,4),NSTRMX(26),ZNWHL(18),ZAXLE(18),VU(9),
1EX(2),KSYM(9,5),STRSP(9),IAA(16)
INTEGER ONE,TWO,BLANK,HHH,SSS,EX
DATA NTP/'TP' //
DATA EX/' ','X' //
DATA ONE,TWO,BLANK,HHH,SSS/'1' ,'2' ,' ' ,'H' ,'S' //
CALL REREAD
NCOUNT=1
C*****
C**** INPUT ROUTINE
C*****
3007 FPC1=4.0
IF(NCOUNT.NE.1) GO TO 573
READ(5,500)(TITLE(1,J),J=1,54)
READ(5,500)(TITLE(2,J),J=1,54)
573 READ(5,500,END=2500) (TITLE(3,J1),J1=1,54)
500 FORMAT(80A1)
CALL DEFIN
READ(5,501)ID,IA,IB,DISTF ,JLOAD, ULSB,JCONC,F28,JOPT
501 FORMAT(A1,4X,A1,1X,A1,4X,F4.3,3X,I1,3X, F4.2,3X,I1,3X,F3.1,
*10X,I1)
IF(JOPT.EQ.0.AND.F28.EQ.0.) WRITE(6,525)
IF(JOPT.EQ.0.AND.F28.EQ.0.) STOP

```

525	FORMAT(//////.35X,**DESIGN OPTION SPECIFIED BUT NO 28 DAY CONCRETE	DS000410
	1STRENGTH GIVEN**)	DS000420
C	AXLE TRAIN	DS000430
	DO 927 I=1,18	DS000440
	PAXLE(I)=0.	DS000450
	NWHL(I)=0.	DS000460
927	CONTINUE	DS000470
	IF(JLOAD.NE.1)GO TO 861	DS000480
	READ(5,502)(PAXLE(I),I=1,18)	DS000490
502	FORMAT(3X,18(F3.1,1X))	DS000500
	DO 503 N=1,18	DS000510
	IF(PAXLE(N).NE.0.)NWHEEL=N	DS000520
503	CONTINUE	DS000530
	READ(5,505)(NWHL(I),I=1,17)	DS000540
505	FORMAT(7X,17(F3.0,1X))	DS000550
C	CONCENTRATED FORCES APPLIED TO SINGLE BEAM	DS000560
861	DO 590 I=1,10	DS000570
	FCONC(I)=0.	DS000580
	DCONC(I)=0.	DS000590
590	CONTINUE	DS000600
	IF(JCONC.NE.1) GO TO 925	DS000610
	READ(5,591)(FCONC(I),I=1,10)	DS000620
	READ(5,591)(DCCNC(I),I=1,10)	DS000630
591	FORMAT(3X,10(F5.2,1X))	DS000640
C	BEAM DIMENSIONS	DS000650
925	READ(5,907)ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,WHDIM,TDIM,WDDI	DS000660
	*M,XDIM,YDIM	DS000670
907	FORMAT(3X,13(F4.2,1X))	DS000680
C	GENERAL INFORMATION	DS000690
	READ(5,908) ZL,WIDTH,JTNLT, TNLB,APRIME,DCR,ALDEF,DEFMIN,	DS000700
	1CBOT,NW,HDPT,JPROP	DS000710
908	FORMAT(3X,F4.1,3X,F4.1,3X,I2,3X,F6.4 ,F4.2,3X,F4.2,3X,F5.3,3X,	DS000720
	1F5.3,4X,F2.1,4X,I2,4X,F3.1,4X,I1)	DS000730
	IF(ZL.LT.10..OR. TNLB.EQ.0) WRITE(6,917) ZL, TNLB	DS000740
917	FORMAT(//////.40X,**CHECK YOUR DATA = BEAM LENGTH AND NUMBER OF BEADS	DS000750
	MS ARE/,.60X,F10.2,5X,F5.2)	DS000760
	IF(ZL.LT.10..OR. TNLB.EQ.0) STOP	DS000770
	READ(99,9083) IAB,IAC,IAD,IAE	DS000780
9083	FORMAT(42X,4A4)	DS000790
	IF(IAB.EQ.BLANK.AND.IAC.EQ.BLANK) ALDEF=1000.	DS000800

	IF(IAD.EQ.BLANK.AND.IAE.EQ.BLANK) DEFMIN==1000.	DS000810
C	MAXIMUM NUMBER OF STRANDS PER ROW	DS000820
	NRAV=0	DS000830
	READ(5,9094) (NSTRMX(J1),J1=1,26)	DS000840
9094	FORMAT(3X,25(A2,1X),A2)	DS000850
	DO 9100 J1=1,26	DS000860
	IF(NSTRMX(J1).NE.NTP) J2=J1	DS000870
	IF(NSTRMX(J1).EQ.NTP) NRAV=J1	DS000880
	IF(NSTRMX(J1).EQ.NTP) GO TO 9101	DS000890
9100	CONTINUE	DS000900
9101	DO 9404 J1=1,26	DS000910
9404	NSTRMX(J1)=0	DS000920
	READ(99,909) (NSTRMX(J1),J1=1,J2)	DS000930
909	FORMAT(3X,25(I2,1X),I2)	DS000940
	DO 910 J=1,26	DS000950
	STRMAX(J)=NSTRMX(J)	DS000960
	IF(NSTRMX(J).NE.0) NR=J	DS000970
910	CONTINUE	DS000980
C	MISCELLANEOUS PROPERTIES	DS000990
862	IF(JPROP.NE.1) GO TO 863	DS001000
	READ(5,882) CUW,UHM,SA,SPF,PSG,BR2,TR2,BR1,TR1,BS2,TS2,BS1,TS1,	DS001010
	1CREEP1,CREEP2,SHRK1,SHRK2,NVRA	DS001020
882	FORMAT(3X,F3.3,3X,F2.0,2X,F3.3,2X,F3.0,3X,F2.1,4X,F2.2,1X,F2.1,1X,	DS001030
	1F2.2,1X,F2.1,3X,F2.2,1X,F2.1,1X,F2.2,1X,F2.1,4X,4(F3.0,1X),1X,I2)	DS001040
	IF(SPF.NE.0.) FPS=SPF	DS001050
	IF(SA.NE.0.) AS=SA	DS001060
	IF(UHM.NE.0.) HUM=UHM	DS001070
	IF(CUW.NE.0.) UWC=CUW	DS001080
	IF(PSG.NE.0.0) GSP=PSG	DS001090
	IF(NVRA.NE.0) NRAV=NVRA	DS001100
	IF(NRAV.EQ.0) WRITE(6,9181)	DS001110
	IF(NRAV.EQ.0) STOP	DS001120
9181	FORMAT(1X,130(1H*)/1X,60(*'),* INCORRECT DATA INPUT - TOP-MOST GRDS	DS001130
	11D ROW WAS NOT SPECIFIED - CHECK *,60(*'),/,1X,60(*'),*MAXIMUM NDS	DS001140
	2UMBER OF STRANDS CARD AND/OR MISCELLANEOUS PROPERTIES CARD *,60('DS	DS001150
	3*'))	DS001160
	READ(99,9082) (IAA(J1),J1=1,16)	DS001170
9082	FORMAT(30X,2A1,1X,2A1,1X,2A1,1X,2A1,3X,2A1,1X,2A1,1X,2A1,1X,2A1)	DS001180
	IF(IAA(1).NE.BLANK.CR.IAA(2).NE.BLANK) CER2=BR2	DS001190
	IF(IAA(3).NE.BLANK.OR.IAA(4).NE.BLANK) CTR2=TR2	DS001200

IF(IAA(5).NE.BLANK.CR.IAA(6).NE.BLANK)	CBR1=BR1	DS001210
IF(IAA(7).NE.BLANK.OR.IAA(8).NE.BLANK)	CTR1=TR1	DS001220
IF(IAA(9).NE.BLANK.OR.IAA(10).NE.BLANK)	CBS2=BS2	DS001230
IF(IAA(11).NE.BLANK.OR.IAA(12).NE.BLANK)	CTS2=TS2	DS001240
IF(IAA(13).NE.BLANK.OR.IAA(14).NE.BLANK)	CBS1=BS1	DS001250
IF(IAA(15).NE.BLANK.OR.IAA(16).NE.BLANK)	CTS1=TS1	DS001260
C	CONCRETE COST COEFFICIENTS	DS001270
863	IF(JOPT.NE.1) GO TO 1001	DS001280
912	READ(5,915)(G(I),I=1,6)	DS001290
	READ(5,915)(G(I),I=7,11)	DS001300
915	FORMAT(10X,5(F4.1,9X),F4.1)	DS001310
C	STRAND COST	DS001320
	READ(5,914) COSTFT	DS001330
914	FORMAT(13X,F3.2,46X,F3.2)	DS001340
913	FORMAT(10X,5(F3.1,10X),F3.1)	DS001350
C	28 DAY CONCRETE STRENGTHS	DS001360
	READ(5,913)(F(I),I=1,6)	DS001370
	READ(5,913)(F(I),I=7,11)	DS001380
	DO 916 J=1,11	DS001390
	IF(F(J).NE.0.0) FPCMAX=4.0+(J-1)*0.5	DS001400
	IF(F(J).EQ.0.0) F(J)=F(J-1)	DS001410
	IF(G(J).EQ.0.0) G(J)=G(J-1)	DS001420
916	CONTINUE	DS001430
1001	CONTINUE	DS001440
C		DS001450
C	LIVE LOAD DISTRIBUTION FACTOR	DS001460
C		DS001470
	IF(DISTF.NE.0.) GO TO 933	DS001480
	CONSNT=1.0	DS001490
	C11 = CONSNT*WIDTH/ZL	DS001500
	D11=5.+JTNTL/10.+(3.-2.*JTNTL/7.)*((1.-C11/3.)**2)	DS001510
	IF(C11.GT.3) D11=5.+JTNTL/10.	DS001520
	S11=(12.*JTNTL+9.)/ TNLB	DS001530
	DISTF =(S11/D11)*0.5	DS001540
933	CONTINUE	DS001550
	IAASHO=1	DS001560
C		DS001570
C	AASHTO TRUCK LOADINGS	DS001580
C		DS001590
	IF(IA.EQ.BLANK.AND.IB.EQ.BLANK) IAASHO=0	DS001600

IF(IA.EQ.BLANK.AND.IB.EQ.BLANK) GO TO 6000	DS001610
IF(IA.EQ.BLANK.AND.IB.EQ.ONE)GO TO 1000	DS001620
IF(IA.EQ.HHH.AND.IB.EQ.ONE)GO TO 1000	DS001630
IF(IA.EQ.SSS.AND.IB.EQ.ONE)GO TO 2000	DS001640
IF(IA.EQ.BLANK.AND.IB.EQ.TWO)GO TO 3000	DS001650
IF(IA.EQ.HHH.AND.IB.EQ.TWO)GO TO 3000	DS001660
IF(IA.EQ.SSS.AND.IB.EQ.TWO)GO TO 4000	DS001670
WRITE(6,950)	DS001680
950 FORMAT(1X,130(1H*)/1X,30(1H*),*UNRECOGNIZABLE AASHTO TRUCK LOADING	DS001690
=CHECK INPUT CARD 4, COLS. 5 THRU 8,30(1H*)/1X,130(1H*))	DS001700
STOP	DS001710
C	DS001720
C H=15 TRUCK	DS001730
C	DS001740
1000 ZAXLE(1)=6.	DS001750
ZAXLE(2)=24.	DS001760
ZNWHL(1)=14.	DS001770
NAXLE=2	DS001780
ULOAD=0.480	DS001790
CSLOAD=19.5	DS001800
CMLOAD=13.5	DS001810
GO TO 6000	DS001820
C	DS001830
C HS=15 TRUCK	DS001840
C	DS001850
2000 ZAXLE(1)=6.	DS001860
ZAXLE(2)=24.	DS001870
ZAXLE(3)=24.	DS001880
ZNWHL(1)=14.	DS001890
ZNWHL(2)=28.	DS001900
NAXLE=3	DS001910
ULOAD=0.480	DS001920
CSLOAD=19.5	DS001930
CMLOAD=13.5	DS001940
GO TO 6000	DS001950
C	DS001960
C H=20 TRUCK	DS001970
C	DS001980
3000 ZAXLE(1)=8.	DS001990
ZAXLE(2)=32.	DS002000

ZNWHL(1)=14.	DS002010
NAXLE=2	DS002020
ULOAD=0.640	DS002030
CSLOAD=26.	DS002040
CMLOAD=18.	DS002050
GO TO 6000	DS002060
C	DS002070
C HS=20 TRUCK	DS002080
C	DS002090
4000 ZAXLE(1)=8.	DS002100
ZAXLE(2)=32.	DS002110
ZAXLE(3)=32.	DS002120
ZNWHL(1)=14.	DS002130
ZNWHL(2)=28.	DS002140
NAXLE=3	DS002150
ULOAD=0.640	DS002160
CSLOAD=26.	DS002170
CMLOAD=18.	DS002180
6000 CONTINUE	DS002190
CALL PROPTY	DS002200
Z1=TDIM	DS002210
Z2=TDIM+XDIM	DS002220
Z3=HDIM	DS002230
Z4=HDIM+GDIM	DS002240
Y1=WHDIM/2.	DS002250
Y2=WDDIM	DS002260
Y3=ADIM/2.-Y1-Y2	DS002270
Y4=XDIM	DS002280
D(1)=DBOT+CBOT	DS002290
DO 9102 J1=2,NR	DS002300
9102 D(J1)=D(J1-1)+GSP	DS002310
ALPHA=0.5=HDPT/ZL	DS002320
C*****	DS002330
C**** PRINT OUT INPUT QUANTITIES	DS002340
C*****	DS002350
WRITE(6,9080) (TITLE(1,J1),J1=10,11),(TITLE(1,J1),J1=14,26).	DS002360
1(TITLE(1,J1),J1=48,54),(TITLE(2,J2),J2=13,19),(TITLE(2,J2),J2=26,28),	DS002370
(TITLE(2,J2),J2=44,54),(TITLE(3,J3),J3=13,54)	DS002380
9080 FORMAT(1H1,37X,'DISTRICT ',2A1,1X,13A1,' COUNTY HIGHWAY NO. ',	DS002390
17A1/38X,'CONTRCL NO. ',7A1,' IPE ',3A1,' SUBMITTED BY ',11A1/	DS002400

238X,'DESCRIPTION ',42A1)	DS002410
600 FORMAT(1H1)	DS002420
WRITE(6,601)	DS002430
601 FORMAT(/,1X,129('**'))	DS002440
WRITE(6,610)	DS002450
610 FORMAT(' ',47X,'BEAM DIMENSIONS AND PROPERTIES',50X,'**')	DS002460
WRITE(6,602)	DS002470
602 FORMAT(1X,129('**'))	DS002480
WRITE(6,603)	DS002490
603 FORMAT(1X,'**',127X,'**')	DS002500
WRITE(6,611) ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,WHDIM,TDIM,	DS002510
1WDDIM,XDIM,YDIM	DS002520
611 FORMAT(1X,'**',8X,'**',44(' '),', DIMENSIONS IN INCHES ',43(' '),	DS002530
1'**,8X,'**'/1X,'**',127X,'**'/1X,'**',21X,'A',6X,'B',6X,'C',6X,'D',	DS002540
26X,'E',6X,'F',6X,'G',6X,'H',6X,'M',6X,'T',6X,'W',6X,'X',6X,'Y',	DS002550
321X,'**'/1X,'**',19X,13(F5.2,2X),17X,'**')	DS002560
WRITE(6,603)	DS002570
WRITE(6,613) BINERT,ACONC,DTOP,DBOT,BINERK,ACONCK,DTOPK,DBOTK	DS002580
613 FORMAT(1X,'**',8X,'**',14(' '),', (WITOUT SHEAR KEY)',12(' '),', SECTION	DS002590
IN PROPERTIES ',12(' '),', (WITH SHEAR KEY)',17(' '),',**',8X,'**'/1X,	DS002600
2'**,127X,'**'/1X,'**',14X,2('I(IN**4)',5X,'A(IN**2)',5X,'YT(IN)',5X,	DS002610
3'VB(IN)',13X),1X,'**'/1X,'**',12X,2(F10.0,6X,F6.1,6X,F5.2,6X,F5.2,	DS002620
413X),1X,'**')	DS002630
WRITE(6,603)	DS002640
WRITE(6,650) DCR,APRIME,AS,FPS,HUM,UWC,HDPT	DS002650
650 FORMAT(1X,'**',4X,'DISTANCE TO',95X,'DISTANCE FROM',4X,'**'/1X,'**',	DS002660
17X,'CG OF',8X,'COMPRESSION',5X,'MAXIMUM',5X,'MINIMUM',17X,'STRAND'	DS002670
2,19X,'CONCRETE',5X,'CENTERLINE OF',4X,'**'/1X,'**',4X,'COMPRESSION',	DS002680
35X,'REINFORCING',2(5X,'INITIAL'),5X,'STRAND',5X,'ULTIMATE',5X,	DS002690
4'RELATIVE',7X,'UNIT',10X,'BEAM TO',7X,'**'/1X,'**',4X,'REINFORCING',	DS002700
58X,'AREA',9X,2('CAMBER',6X),', AREA',6X,'STRENGTH',5X,'HUMIDITY',	DS002710
66X,'WEIGHT',6X,'HARPING POINT',4X,'**'/1X,'**',8X,'(IN)',10X,'(IN**2	DS002720
7)',2(8X,'(IN)'),7X,'(IN**2)',5X,'(KSI)',9X,'(X)',8X,'(K/FT**3)',	DS002730
88X,'(FT)',9X,'**'/1X,'**',7X,F5.2,10X,F5.2,6X,9X ,3X,9X ,7X,F5.3,	DS002740
97X,F4.0,9X,F3.0,9X,F5.3,10X,F5.2,9X,'**')	DS002750
IF(ALDEF.GT.999..AND.DEFMIN.GT.=999.) WRITE(6,9401) DEFMIN	DS002760
IF(ALDEF.LT.999..AND.DEFMIN.LT.=999.) WRITE(6,9501) ALDEF	DS002770
IF(ALDEF.LT.999..AND.DEFMIN.GT.=999.) WRITE(6,9601) ALDEF,DEFMIN	DS002780
9401 FORMAT('+',46X,F9.3)	DS002790
9501 FORMAT('+',34X,F9.3)	DS002800

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9601 FORMAT('+',34X,F9.3,3X,F9.3) DS002810
      WRITE(6,603) DS002820
      WRITE(6,614) NW,CBOT,GSP,NRAV DS002830
614 FORMAT(1X,'*',54(' '),* STRAND INFORMATION *,53(' '),**/1X,'*', DS002840
      1127X,'**/1X,'*',10X,'DISTANCE FROM BOTTOM',56X,'NUMBER OF WEB STRADS DS002850
      2NDS = ',13,14X,'**/1X,'*',10X,'OF BEAM TO STRAND ROW 1 = ',F5.2, DS002860
      3' IN.',8X,'GRID SPACING = ',F5.2,' IN.',9X,'TOP = MOST GRID ROW', DS002870
      43X,'= ',13,14X,'**/1X,'*',127X,'**') DS002880
      WRITE(6,651) (J1,J1=1,26),(NSTRMX(J1),J1=1,26) DS002890
651 FORMAT(1X,'*',2X,'ROW NUMBER',9X,26I4,2X,'**/1X,'*',2X,'MAX. NO. DS002900
      1F STRANDS',26I4,2X,'**/1X,'*',127X,'**') DS002910
      WRITE(6,615) CREEP1,SHRK1,CBR2,CBR1,CBS2,CBS1,CREEP2,SHRK2, DS002920
      1CTR2,CTR1,CTS2,CTS1 DS002930
615 FORMAT(1X,'*',17(' '),* ALLOWABLE STRESS COEFFICIENTS *,32(' '), DS002940
      1' CREEP AND SHRINKAGE COEFFICIENTS ',13(' '),**/1X,'*',127X,'**/ DS002950
      21X,'*',27X,'RELEASE',19X,'SERVICE',67X,'**/1X,'*',21X,2('END 1/10' DS002960
      2.3X,'REMAINDER',6X),8X,'CREEP1 = ',F4.0,8X,'SHRK1 = ',F4.0,13X, DS002970
      4'**/1X,'*',4X,'COMPRESSION',8X,2(F4.2,7X,F4.2,11X),6X,'CREEP2 = ', DS002980
      5F4.0,8X,'SHRK2 = ',F4.0,13X,'**/1X,'*',4X,'TENSION',12X,2(F4.2, DS002990
      67X,F4.2,11X),52X,'**/1X,'*',127X,'**') DS003000
      IF(JOPT.NE.1) GO TO 695 DS003010
      WRITE(6,616) DS003020
616 FORMAT(' ',46(' '),* CONCRETE COST COEFFICIENTS($/YD**3)*,46(' '), DS003030
      6'**) DS003040
      WRITE(6,653)(G(I),I=1,8) DS003050
653 FORMAT(' *4.0KSI/$',F5.1,' 4.5KSI/$',F5.1,' 5.0KSI/$',F5.1,' 5. DS003060
      65KSI/$',F5.1,' 6.0KSI/$',F5.1,' 6.5KSI/$',F5.1,' 7.0KSI/$',F5.1 DS003070
      6,' 7.5KSI/$',F5.1,9X,'**') DS003080
      WRITE(6,654)(G(I),I=9,11) DS003090
654 FORMAT(' *8.0KSI/$',F5.1,' 8.5KSI/$',F5.1,' 9.0KSI/$',F5.1,84X,' DS003100
      6'**) DS003110
      WRITE(6,603) DS003120
      WRITE(6,617) DS003130
617 FORMAT(' ',28(' '),* 28 DAY CONCRETE STRENGTH COEFFICIENTS (RELEAS DS003140
      6E STRENGTH/28 DAY STRENGTH)*,27(' '),**') DS003150
      WRITE(6,655)(F(I),I=1,8) DS003160
655 FORMAT(' *4.0KSI/',F4.1,'KSI 4.5KSI/',F4.1,'KSI 5.0KSI/',F4.1,' DS003170
      6KSI 5.5KSI/',F4.1,'KSI 6.0KSI/',F4.1,'KSI 6.5KSI/',F4.1,'KSI 7 DS003180
      6.0KSI/',F4.1,'KSI 7.5KSI/',F4.1,'KSI*') DS003190
      WRITE(6,656)(F(I),I=9,11) DS003200

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656 FORMAT(' *8.0KSI/',F4.1,'KSI 8.5KSI/',F4.1,'KSI 9.0KSI/',F4.1,'DS003210
      8KSI',80X,'**') DS003220
      WRITE(6,603) DS003230
      WRITE(6,657) COSTFT,FPCMAX DS003240
657 FORMAT(' **',21X,'STRAND COST = $',F4.2,'/FT',21X,'MAXIMUM RELEASE DS003250
      1STRENGTH ALLCWD = ',F4.1,' KSI',21X,'**') DS003260
695 WRITE(6,602) DS003270
      WRITE(6,9007) DS003280
9007 FORMAT(/1X,129('**')) DS003290
      WRITE(6,618) DS003300
618 FORMAT(' **',53X,'BRIDGE PROPERTIES',57X,'**') DS003310
      WRITE(6,602) DS003320
      WRITE(6,603) DS003330
      WRITE(6,658)ZL ,WIDTH,JTNTL, TNLB DS003340
658 FORMAT(' *SPAN LENGTH = ',F5.1,'(FT)',5X,'BRIDGE WIDTH = ',F5.1,'(DS003350
      8FT)',5X,'NUMBER TRAFFIC LANES = ',I2,5X,'NUMBER BEAMS = ',F5.2,20XDS003360
      *,**') DS003370
      WRITE(6,602) DS003380
      WRITE(6,600) DS003390
      WRITE(6,602) DS003400
      WRITE(6,619) DS003410
619 FORMAT(' **',52X,'LOADING CONDITIONS',57X,'**') DS003420
      WRITE(6,602) DS003430
      WRITE(6,603) DS003440
      IF(IA.EQ.BLANK.AND.IB.EQ.BLANK) GO TO 696 DS003450
      IF(IB.EQ.TWO) WRITE(6,659)IA,DISTF DS003460
659 FORMAT(' * AASHTO LL = H',A1,'-20',7X, 'L.L. DISTRIBUTION FACTOR DS003470
      8= ',F5.3,'(TRUCKS)',62X,'**') DS003480
      IF(IB.EQ.ONE) WRITE(6,660)IA,DISTF DS003490
660 FORMAT(' * AASHTO LL = H',A1,'-15',7X, 'L.L. DISTRIBUTION FACTOR DS003500
      8= ',F5.3,'(TRUCKS)',62X,'**') DS003510
      WRITE(6,603) DS003520
696 IF(JLOAD.NE.1) GO TO 697 DS003530
      WRITE(6,620)(I,I=1,18) DS003540
620 FORMAT(' **',57(' '), 'AXLE TRAIN',60(' '), '**',/, ' *AXLE NUMBER ',DS003550
      818I6,5X,'**') DS003560
      WRITE(6,661)(PAXLE(I),I=1,18) DS003570
661 FORMAT(' *AXLE LOAD(KIPS)',18F6.1,4X,'**') DS003580
      WRITE(6,662)(NWHL(I),I=1,17) DS003590
662 FORMAT(' *DISTANCE TO AXLE(FT) ',17F4.0,4X,'**') DS003600

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        WRITE(6,603)
697  IF(JCONC.NE.1) GO TO 698
        WRITE(6,621)
621  FORMAT(' **',45(' '), 'CONCENTRATED FORCES ON SINGLE BEAM',48(' '),
        &' ')
        WRITE(6,664)(FCCNC(I),I=1,10)
664  FORMAT(' *LOAD(KIPS)      ',10F9.1,23X,'**')
        WRITE(6,665)(DCCNC(I),I=1,10)
665  FORMAT(' *DISTANCE FROM',114X,'**',/, ' *LEFT SUPPORT(FT)  ',F5.1,9F
        &9.1,23X,'**')
        WRITE(6,603)
698  IF(ULSB.EQ.0.0) GO TO 699
        WRITE(6,666)ULSB
666  FORMAT(' *UNIFORM LOAD ON SINGLE BEAM = ',F5.2,'(K/FT)',86X,'**')
699  CONTINUE
        WRITE(6,602)
C*****
C**** COMPUTE DESIGN MOMENTS AND SHEARS
C*****
        DO 8104 J1=1,9
        DLMOM(J1)=0.
        DLSHR(J1)=0.
        ZX=(J1-1)*ZL/10.
        IF(J1.EQ.3) ZX=5.*ZL/40.
        IF(J1.EQ.4) ZX=2.*ZL/10.
        IF(J1.EQ.5) ZX=ZL/4.
        IF(J1.GE.6) ZX=(J1-3)*ZL/10.
        IF(J1.EQ.9) ZX=ALPHA*ZL
        ZMDL=(ACONCK*UWC/144.)*ZX/2.*(ZL-ZX)
        IF(J1.EQ.8) ZMBW=ZMDL
        ZSDL=(ACONCK*UWC/144.)*(ZL/2.-ZX)
        IF(JCONC.NE.1) GO TO 14
        SUMM=0.
        SUMV=0.
        DO 12 J2=1,10
        IF(DCONC(J2).EQ.0.) GO TO 14
        R=FCONC(J2)*(ZL-DCONC(J2))/ZL
        SUMM=SUMM+R*ZX
        SUMV=R
        IF(DCONC(J2).GT.ZX+.1) GO TO 10

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DS003610
DS003620
DS003630
DS003640
DS003650
DS003660
DS003670
DS003680
DS003690
DS003700
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DS003900
DS003910
DS003920
DS003930
DS003940
DS003950
DS003960
DS003970
DS003980
DS003990
DS004000

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SUMM=SUMM-FCONC(J2)*(ZX-DCONC(J2))
SUMV=SUMV-FCONC(J2)
10 DLMOM(J1)=DLMOM(J1)+SUMM
   DLSHR(J1)=DLSHR(J1)+SUMV
12 CONTINUE
14 DLMOM(J1)=DLMOM(J1)+ULSB*ZX/2.*(ZL-ZX)
   DLSHR(J1)=DLSHR(J1)+ULSB*(ZL/2.-ZX)
   ZMOML=0.
   ZMOMT=0.
   ZSHRL=0.
   ZSHRT=0.
   ZMOMAX=0.
   ZSHRAX=0.
   ZIMP=1.+50./(125.+ZL)
   IF(ZIMP.GT.1.30) ZIMP=1.30
   IF(IAASHO.EQ.0) GO TO 16
   ZMOML=DISTF*ZIMP*ULOAD*ZX/2.*(ZL-ZX)
   ZSHRL=DISTF*ZIMP*ULOAD*(ZL/2.-ZX)
   R=CMLOAD*(ZL-ZX)/ZL
   ZMOML=ZMOML+R*ZX*DISTF*ZIMP
   ZSHRL=ZSHRL+R*DISTF*ZIMP
   CALL MOMSHR(ZL,ZNWL,NAXLE,ZX,ZAXLE,ZMOMT,ZSHRT)
   ZMOMT=ZMOMT*DISTF*ZIMP
   ZSHRT=ZSHRT*DISTF*ZIMP
16 IF(JLOAD.EQ.0) GO TO 18
   CALL MOMSHR(ZL,NWL,NWFEEL,ZX,PAXLE,ZMOMAX,ZSHRAX)
   ZMOMAX=ZMOMAX*DISTF
   ZSHRAX=ZSHRAX*DISTF
18 BMMAX(J1)=AMAX1(ZMOML,ZMOMT,ZMOMAX)
   BVMAX(J1)=AMAX1(ZSHRL,ZSHRT,ZSHRAX)
8104 CONTINUE
   ZMAX=0.0
   DO 5106 J1=1,9
   IF(BMMAX(J1).GT.ZMAX) ZMAX=BMMAX(J1)
5106 CONTINUE
   ZMNC=ABS(ACONCK-ACONC)*UWC*ZL**2/(144.*8.)
   ZMC=DLMCM(8)
   ULTMRO=1.3*(ZMBW+DLMCM(8)+(5./3.)*ZMAX)
C*****
C**** GENERATE OBJECTIVE FUNCTION AND INITIAL CONSTRAINT SET

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DS004010
DS004020
DS004030
DS004040
DS004050
DS004060
DS004070
DS004080
DS004090
DS004100
DS004110
DS004120
DS004130
DS004140
DS004150
DS004160
DS004170
DS004180
DS004190
DS004200
DS004210
DS004220
DS004230
DS004240
DS004250
DS004260
DS004270
DS004280
DS004290
DS004300
DS004310
DS004320
DS004330
DS004340
DS004350
DS004360
DS004370
DS004380
DS004390
DS004400


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C *****DS004410
  ASSCLR=0.05DS004420
  ASSPLS=0.1DS004430
  ELASC=1.04355*(1000.*UWC)**1.5DS004440
  IF(JOPT.EQ.0) GO TO 110DS004450
  N=48+7*NRDS004460
  M=12+2*NRDS004470
  GO TO 112DS004480
110 N=29+7*NRDS004490
  M=3+2*NRDS004500
112 KK=M+N-1DS004510
  K=N+M-1DS004520
  IF(JOPT.EQ.1) GO TO 108DS004530
CDS004540
C  DEFINE COST COEFFICIENTS FOR DESIGN OPTIONDS004550
CDS004560
  COSTFT=100.DS004570
  G(1)=1944.*COSTFT/(4.0*ACONC)DS004580
108 CALL EQGENDS004590
C *****DS004600
C  CALL LPCODE FOR INITIAL SOLUTIONDS004610
C *****DS004620
  KODE=0DS004630
  NFRCE=0DS004640
  NEQS=0DS004650
  INDX=0DS004660
  CALL LPCODE(NFRCE,NEQS,INDX,KODE)DS004670
CDS004680
C  ROUND LP SOLUTION FOR INTRODUCTION TO INTEGER ROUTINEDS004690
CDS004700
  IF(JOPT.EQ.0) JR=M-1DS004710
  IF(JOPT.NE.0) JR=MDS004720
  DO 402 J1=1,JRDS004730
  S1=X(1-JOPT+J1)DS004740
  S2=AIN(T(S1)DS004750
  IF(S1-S2.GE.0.5) X(1-JOPT+J1)=S2+1DS004760
  IF(S1-S2.LT.0.5) X(1-JOPT+J1)=S2DS004770
402 CONTINUEDS004780
C *****DS004790
C  SET UP FOR CALL TO INTEGER ROUTINEDS004800

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C*****DS004810
    DXMAX=50.                                DS004820
    IS=0                                      DS004830
    IF(JOPT.EQ.0) GO TO 113                   DS004840
    NCON=37+6*NR                             DS004850
    N1=0                                      DS004860
    N2=NR+2                                  DS004870
    N3=NR+10                                 DS004880
    GO TO 114                                DS004890
113 NCON=28+6*NR                             DS004900
    N1=1                                      DS004910
    N2=NR+2                                  DS004920
    N3=NR                                     DS004930
114 NRA=NCON+1                               DS004940
    NCA=N1+N2+N3+1                           DS004950
    CALL SQUASH                               DS004960
    WRITE(3) BB                               DS004970
    REWIND 3                                  DS004980
C*****DS004990
C    ITERATE ON PRESTRESS LOSS                DS005000
C*****DS005010
    ZSSCLR=ASSCLR                             DS005020
    ZSSPLS=ASSPLS                             DS005030
    700 CONTINUE                             DS005040
    CALL INTPRG(KBCMB)                        DS005050
    DO 192 J1=1,NC                           DS005060
192  X(J1)=XX(J1)                             DS005070
    DO 5130 J1=1,NR                           DS005080
    IF(NSTRMX(J1)/2*2.NE.NSTRMX(J1)) GO TO 5130 DS005090
    X(1-JOPT+J1)=X(1-JOPT+J1)*2.             DS005100
5130 CONTINUE                                DS005110
C                                              DS005120
C    COMPUTE NEW PRESTRESS LOSSES            DS005130
C                                              DS005140
    IF(JOPT.NE.0) GO TO 708                   DS005150
    FPCR=X(1)                                 DS005160
    GO TO 712                                 DS005170
708 FPCR=4.0                                  DS005180
    DO 710 J1=1,10                            DS005190
710 FPCR=FPCR+0.5*X(2*NR+2+J1)              DS005200

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712	AMOM=0.0	DS005210
	ASUM=0.0	DS005220
	DO 706 J1=1,NR	DS005230
	ASUM=ASUM+AS*X(1-JOPT+J1)	DS005240
706	AMOM=AMOM-AS*X(1-JOPT+J1)*D(J1)	DS005250
	DCG=DBOT-AMOM/ASUM	DS005260
	CALL PLOSS(FPCR,ZMBW,ZMC,ZMNC,FPS,ASUM,ACCNC,BINERT,BINERK,DBOT,	DS005270
	1DBOTK,DCG,HUM,ZL,ZLCSS,ZINLOS,UWC)	DS005280
C		DS005290
C	TERMINATE ITERATIONS IF COMPUTED LONG TERM LOSS DOES	DS005300
C	NOT EXCEED ASSUMED LOSS BY MORE THAN 3 PERCENT	DS005310
C		DS005320
	IF(KBOMB.NE.0) GO TO 740	DS005321
	IF(ZLOSS.LE.1.03*ZSSPLS) GO TO 740	DS005330
C		DS005340
C	UPDATE CONSTRAINTS TO REFLECT NEW PRESTRESS LOSSES	DS005350
C		DS005360
	READ(3) BB	DS005370
	REWIND 3	DS005380
	FACT1=(1.-ZINLOS)/(1.-ZSSCLR)	DS005390
	FACT2=(1.-ZLOSS)/(1.-ZSSPLS)	DS005400
	J11=0	DS005410
	IF(JOPT.EQ.0) NR2=NR*2+3	DS005420
	IF(JOPT.NE.0) NR2=NR*2+2+10	DS005430
	DO 9300 J1=1,NR2	DS005440
	IF(J1.EQ.1-JOPT+NR+1) GO TO 9300	DS005450
	IA=COL(J1)	DS005460
	IB=COL(J1+1)-1	DS005470
	DO 9320 J2=IA,IB	DS005480
	IF(J1.GE.1-JOPT+2*NR+3) GO TO 9013	DS005490
	IF(J1.GE.1-JOPT+NR+3) GO TO 9310	DS005500
C		DS005510
C	RELEASE STRESSES	DS005520
C		DS005530
	IF(ROW(J2).LE.8) Y(J2)=Y(J2)*FACT1	DS005540
C		DS005550
C	SERVICE STRESSES	DS005560
C		DS005570
	IF(ROW(J2).GT.8.AND.ROW(J2).LE.22) Y(J2)=Y(J2)*FACT2	DS005580
C		DS005590

C CRACKING MOMENT CAPACITY

C

9013 JR=26+6*NR
IF(JOPT.EQ.1) JR=35+6*NR
IF(J11.NE.0) GO TO 724
J11=J11+1
IF(ZLOSS.LE.0.2) I1=1
IF(ZLOSS.LE.0.2) I2=2
IF(0.2.LT.ZLOSS.AND.ZLOSS.LE.0.3) I1=2
IF(0.2.LT.ZLOSS.AND.ZLOSS.LE.0.3) I2=3
IF(ZLOSS.GT.0.3) I1=3
IF(ZLOSS.GT.0.3) I2=4
S1=I1/10.
S2=I2/10.
IF(JOPT.NE.0) GO TO 720
BB(26+6*NR)=-((PECRK(1,I2)-PECRK(1,I1))*ZLOSS/0.1-(PECRK(1,I2)*S1
1-PECRK(1,I1)*S2)/0.1)
GO TO 726
720 DO 722 J4=1,I1
722 ZNE(J4)=(PECRK(J4,I2)-PECRK(J4,I1))*ZLOSS/0.1-(PECRK(J4,I2)*S1
1-PECRK(1,I1)*S2)/0.1
BB(35+6*NR)=-ZNE(1)
724 IF(ROW(J2).NE.JR) GO TO 726
IF(J1.LT.1-JOPT+2*NR+3) Y(J2)=Y(J2)*FACT2
IF(J1.LT.1-JOPT+2*NR+3) GO TO 726
IF(JOPT.EQ.0) GO TO 726
Y(J2)=2.*(ZNE(J1-2*NR+2+1)-ZNE(J1-2*NR+2))
726 CONTINUE
IF(J1.GE.1-JOPT+2*NR+3) GO TO 9320

C

C

CAMBER CONSTRAINTS

C

9310 JR=32+6*NR
IF(JOPT.EQ.0) JR=23+6*NR
IF(ROW(J2).EQ.JR.OR.ROW(J2).EQ.JR+1) Y(J2)=Y(J2)*FACT1
9320 CONTINUE
9300 CONTINUE
ZSSPLS=ZLOSS
ZSSCLR=ZINLOS
DO 5140 J1=1, NR

DS005600
DS005610
DS005620
DS005630
DS005640
DS005650
DS005660
DS005670
DS005680
DS005690
DS005700
DS005710
DS005720
DS005730
DS005740
DS005750
DS005760
DS005770
DS005780
DS005790
DS005800
DS005810
DS005820
DS005830
DS005840
DS005850
DS005860
DS005870
DS005880
DS005890
DS005900
DS005910
DS005920
DS005930
DS005940
DS005950
DS005960
DS005970
DS005980
DS005990

IF(NSTRMX(J1)/2*2.NE.NSTRMX(J1)) GO TO 5140	DS006000
X(1-JOPT+J1)=0.5*X(1-JOPT+J1)	DS006010
5140 CONTINUE	DS006020
GO TO 700	DS006030
740 CONTINUE	DS006040
C*****	DS006050
C COMPUTE DESIGN RESULTS	DS006060
C*****	DS006070
C	DS006080
C ECCENTRICITIES	DS006090
C	DS006100
SUM1=0.0	DS006110
SUM2=0.0	DS006120
ISUMI=0	DS006130
XMOM1=0.0	DS006140
XMOM2=0.0	DS006150
DO 9156 J1=1,NR	DS006160
IF(X(NR+3-JOPT+J1).EQ.1.) ISUMI=ISUMI+1	DS006170
SUM1=SUM1+X(1-JOPT+J1)	DS006180
SUM2=SUM2+NW*X(1-JOPT+NR+2+J1)	DS006190
XMOM1=XMOM1-X(1-JOPT+J1)*D(J1)	DS006200
9156 XMOM2=XMOM2-NW*X(1-JOPT+NR+2+J1)*D(J1)	DS006210
EN=X(NR+3-JOPT)	DS006220
ENI=EN/ISUMI+0.5	DS006230
DRAP1=AINTE(ENI)*GSP	DS006240
XMOM3=-NW*GSP*X(1-JOPT+NR+2)	DS006250
IF(XMOM3.EQ.0.0) XMOM2=0.0	DS006260
IF(XMOM3.EQ.0.0) SUM2=0.0	DS006270
ECCL=XMOM1/SUM1	DS006280
ENDECC=(XMOM1+XMOM3)/SUM1	DS006290
SS1=DSBOT=(XMOM1-XMOM2)/(SUM1-SUM2)	DS006300
IF(SUM2.NE.0.0) DS1=DSBOT=XMOM2/SUM2	DS006310
IF(SUM2.NE.0.0) DS2=DSBOT=(XMOM2+XMOM3)/SUM2	DS006320
IF(SUM2.EQ.0.0) DS1=SS1	DS006330
IF(SUM2.EQ.0.0) DS2=SS1	DS006340
NSUM1=SUM1	DS006350
NSUM2=SUM2	DS006360
C	DS006370
C 28-DAY CONCRETE STRENGTH	DS006380
C	DS006390

IF(JOPT.EQ.0) GO TO 648	DS006400
F28=F(1)	DS006410
DO 642 J3=2,11	DS006420
642 F28=F28+(F(J3)-F(J3-1))*X(2*NR+2+J3-1)	DS006430
648 CONTINUE	DS006440
C	DS006450
C DEFLECTION CALCULATIONS	DS006460
C	DS006470
ECR=ELASC*SQRT(FPCR)	DS006480
CBRI=(-22.5*W*ZL**4+(1.-ZINLOS)*FO*ZL**2*(XMM1/8.+ALPHA**2/6.*	DS006490
1XMM3)*144.)/ECR/BINERT	DS006500
DEFBWK=(-22.5*WB*ZL**4+(1.-ZLOSS)*FO*ZL**2*(XMM1/8.+ALPHA**2/6.*	DS006510
1XMM3)*144.)/5000./BINERT	DS006520
DEFCF=0.	DS006530
IF(JCONC.NE.1) GO TO 2445	DS006540
DO 2444 JN=1,10	DS006550
ZBX1=ZL-DCCNC(JN)	DS006560
ZBX2=DCCNC(JN)	DS006570
ZBX=AMIN1(ZBX1,ZBX2)*12.	DS006580
2444 DEFCF=DEFCF+FCONC(JN)*ZBX*(3.*ZL**2*144.-4.*ZBX**2)/48.	DS006590
2445 CONTINUE	DS006600
DEFBWU=(-22.5*(WB+ULSB)*ZL**4+(1.-ZLOSS)*FO*ZL**2*(XMM1/8.+	DS006610
1ALPHA**2/6.*XMM3)*144.-DEFCF)/5000./BINERT	DS006620
C LONG TERM CAMBER	DS006630
EC=ELASC*SQRT(F28)/1.E+03	DS006640
ES=29.	DS006650
FP=FO*1000.	DS006660
UWB=UWC*1000.	DS006670
CALL CAMBER(ES,EC,AS,SUM1,UWB,ACONC,ZL,ECCL,BINERT,FP,ENDECC,	DS006680
1PRLMAX,CBRMAX,HDPT)	DS006690
C	DS006700
C ULTIMATE AND CRACKING MOMENT CALCULATIONS	DS006710
C	DS006720
DD=DTOP+ECCL	DS006730
AREA=SUM1*AS	DS006740
CALL ULTMP(AREA,F28,FPS,APRIME,FPL,DD,DDIM,FSY,DCR,	DS006750
1Y1,Y2,Y3,Y4,Z1,Z2,Z3,Z4,ZMUL)	DS006760
SUM8=0.C	DS006770
DO 9155 J1=1,NR	DS006780
9155 SUM8=SUM8+(1./ACONC-D(J1)/ZB)*X(1-JOPT+J1)	DS006790

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      ZCRACK=((1.-ZLOSS)*ZBK*F0*SUM8+7.5*ZBK*0.031623*SQRT(F28)
1=W*ZL**2/8.*12.*ZBK/ZB)/12.
C
C      STIRRUP SPACING
C
      WEB=2.*WDDIM
      IF(SUM2.EQ.0.0) PFORCE=0.0
      IF(SUM2.NE.0.0) PFORCE=-XMOM3/SUM2/(ALPHA*ZL*12.)*F0*(1.-ZLOSS)
      DO 8103 J1=1,9
      ZX=(J1-1)*ZL/10.
      IF(J1.EQ.3) ZX=5.*ZL/40.
      IF(J1.EQ.4) ZX=2.*ZL/10.
      IF(J1.EQ.5) ZX=ZL/4.
      IF(J1.GE.6) ZX=(J1-3)*ZL/10.
      IF(J1.EQ.9) ZX=ALPHA*ZL
      VU(J1)=1.444*(WB*(ZL/2.-ZX)+DLSHR(J1)+5./3.*BVMAX(J1))-PFORCE
      TAU=(ALPHA-ZX/ZL)/ALPHA
      IF(TAU.LT.0.0) TAU=0.0
      DISTCG=DTOP+(XMOM1+TAU*XMOM3)/SUM1
      CALL SHEAR(WEB,DDIM,DISTCG,F28,FSY,ASTIRP,VU(J1),SPACE)
      STRSP(J1)=SPACE
8103 CONTINUE
C
C      COST AND MATERIAL REQUIREMENTS OF BEAM
C
      IF(JOPT.EQ.0) GO TO 9183
      CONCV=ACONC/144.*ZL/27.
      STRFT=SUM1*ZL
      COSTC=G(1)
      DO 434 J1=1,10
      COSTC=COSTC+(G(J1+1)-G(J1))*X(1-JOPT+2*NR+2+J1)
434 CONTINUE
435 CSTCON=COSTC*CONCV
      CSTSTR=STRFT*CSTFT
      CSTTOT=CSTCON+CSTSTR
      CPRCST=CSTCON/CSTTOT*100.
      SPRCST=CSTSTR/CSTTOT*100.
      CSTPFT=CSTTOT/ZL
9183 CONTINUE
C*****CS007190

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DS006800
 DS006810
 DS006820
 DS006830
 DS006840
 DS006850
 DS006860
 DS006870
 DS006880
 DS006890
 DS006900
 DS006910
 DS006920
 DS006930
 DS006940
 DS006950
 DS006960
 DS006970
 DS006980
 DS006990
 DS007000
 DS007010
 DS007020
 DS007030
 DS007040
 DS007050
 DS007060
 DS007070
 DS007080
 DS007090
 DS007100
 DS007110
 DS007120
 DS007130
 DS007140
 DS007150
 DS007160
 DS007170
 DS007180
 DS007190

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C**** PRINT OUT RESULTS
C*****
WRITE(6,600)
IF(KBOMB.NE.0) WRITE(6,70)
70 FORMAT(50X,32('**'),/,50X,'*SORRY, THIS BEAM WILL NOT WORK*',/,50X,
*32('**'))
IF(JOPT.EQ.0) WRITE(6,622)
622 FORMAT( / , ' ',47X,'THE COMMAND IS TO SELECT STRANDS',48X,' ')
IF(JOPT.EQ.1) WRITE(6,623)
623 FORMAT( / , ' ',50X,'THE COMMAND IS TO OPTIMIZE',51X,' ')
WRITE(6,601)
WRITE(6,624)
624 FORMAT(' **',54X,'DESIGN PROPERTIES',56X,'**')
WRITE(6,602)
WRITE(6,603)
ZLS=ZLOSS*100.
ZINL=ZINLOS*100.
WRITE(6,625) FPCR,ECR,ZINL,F28,ZLS
625 FORMAT(' **',4X,'RELEASE STRENGTH = ',F5.2,' (KSI)',4X,'CONCRETE M
10DULUS(RELEASE) = ',F7.1,' (KSI)',4X,'INITIAL PRESTRESS LOSS = ',
2F5.2,' PERCENT',4X,'**/' ' ',4X,'28=DAY STRENGTH = ',F5.2,' (KSI)
3',50X,'TOTAL PRESTRESS LOSS = ',F5.2,' PERCENT',4X,'**')
WRITE(6,602)
WRITE(6,601)
WRITE(6,628)
628 FORMAT(' **',56X,'DESIGN RESULTS',57X,'**')
WRITE(6,602)
WRITE(6,603)
WRITE(6,629)
629 FORMAT(' **',56(' '), 'STRAND LAYOUT',58(' '), '**')
WRITE(6,603)
WRITE(6,9157) DS2,SS1,DS1,SS1,NSUM1,NSUM2
9157 FORMAT(1X,'**',51X,2('DISTANCE FROM',12X),26X,'**'/1X,'**',34X,
1'LOCATION',5X,2('BOTTOM OF BEAM TO C.G.',3X),30X,'**'/1X,'**',49X,
2'OF DRAPED STRANDS',7X,'OF STRAIGHT STRANDS',35X,'**'/1X,'**',127X,
3'**'/1X,'**',33X,'END OF BEAM',10X,2(F6.2,20X),21X,'**'/
31X,'**',33X,'CENTERLINE ',10X,2(F6.2,20X),21X,'**'/1X,'**',127X,
4'**'/1X,'**',45X,'TOTAL NUMBER OF STRANDS',7X,'= ',13,47X,'**'/
51X,'**',45X,'NUMBER OF DRAPED STRANDS',6X,'= ',13,47X,'**')
DO 9159 J1=1,NR

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DS007200

DS007210

DS007211

DS007212

DS007213

DS007214

DS007220

DS007230

DS007240

DS007250

DS007260

DS007270

DS007280

DS007290

DS007300

DS007310

DS007320

DS007330

DS007340

DS007350

DS007360

DS007370

DS007380

DS007390

DS007400

DS007410

DS007420

DS007430

DS007440

DS007450

DS007460

DS007470

DS007480

DS007490

DS007500

DS007510

DS007520

DS007530

DS007540

DS007550


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J2=NR+1-J1
INTX=X(1-JOPT+NR+1-J1)
9159 WRITE(6,9160) J2,INTX
9160 FORMAT(1X,'*',45X,'NUMBER OF STRANDS IN ROW',I3,' =',I4,47X,'*')
NB=X(1-JOPT+NR+1)
WRITE(6,603)
WRITE(6,9161) NB,NW,DRAP1
9161 FORMAT(1X,'*',2X,'AT THE END OF THE BEAM, BEGINNING WITH ROW',
1I3,'', RAISE',I3,' STRANDS IN EACH ROW',F6.1,' INCHES ABOVE STRAIGDS
2HT STRANDS IN THAT ROW',2X,'*')
WRITE(6,603)
WRITE(6,9227) CBR1,CBRMAX(1),DEFBWK,CBRMAX(2),DEFBWU,CBRMAX(3),
1CBRMAX(4)
9227 FORMAT(1X,'*',53('.'),'COMPUTED DEFLECTION',55('.'),'*'/1X,'*',
1127X,'*'/1X,'*',19X,'SHORT TERM DEFLECTION (IN.)',31X,'LONG TERM
2EFLECTION (IN.)',24X,'*'/1X,'*',127X,'*'/1X,'*',12X,
2'BEAM WEIGHT (RELEASE)',14X,
3'= ',F6.2,11X,'BASED ON DALLAS CONCRETE PROPERTIES',7X,'= ',F6.2,
411X,'*'/1X,'*',12X,'BMWT + SHEAR KEY (E=5000.)',9X,'= ',F6.2,11X,
5'BASED ON CDESSA CONCRETE PROPERTIES',7X,'= ',F6.2,11X,'*'/1X,'*',
612X,'BMWT + KEY + DEAD LOADS (E=5000.) = ',F6.2,11X,'BASED ON SANDS
7 ANTONIO CONCRETE PROPERTIES = ',F6.2,11X,'*'/1X,'*',66X,'BASED O
8N LUFKIN CONCRETE PROPERTIES',7X,'= ',F6.2,11X,'*')
IF(CREEP1.NE.0.0) WRITE(6,9049) CBRMAX(5)
9049 FORMAT(1X,'*',66X,'BASED ON GIVEN CONCRETE PROPERTIES',8X,'= ',
1F6.2,11X,'*')
C STIRRUP SPACING OUTPUT
WRITE(6,603)
WRITE(6,9077) ASTIRP,(STRSP(J4),J4=1,5)
9077 FORMAT(' *',35('.'),' STIRRUP SPACING - AASHTO 1973 - STIRRUP AREAS
1 = ',F4.2,' IN2 ',35('.'),'*'/1X,'*',127X,'*'/
11X,'*',3X,'SECTION',9X,'*',4X,'0/10',3X,'*',4X,'1/10',3X,'*',4X,
2'5/40',3X,'*',4X,'2/10',3X,'*',4X,'1/4',4X,'*',4X,'3/10',3X,'*',
34X,'4/10',3X,'*',4X,'5/10',3X,'*',4X,'HDPT',3X,'*'/1X,'*',3X,
4'SPACING (IN.)',3X,'*',9(2X,F6.2,3X,'*'))
C COST DATA PRINTCUT
IF(JOPT.EQ.0) WRITE(6,603)
IF(JOPT.EQ.0) GO TO 9020
WRITE(6,603)
WRITE(6,670)

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DS007560
DS007570
DS007580
DS007590
DS007600
DS007610
DS007620
DS007630
DS007640
DS007650
DS007660
DS007670
DS007680
DS007690
DS007700
DS007710
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DS007860
DS007870
DS007880
DS007890
DS007900
DS007910
DS007920
DS007930
DS007940
DS007950

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670 FORMAT( 1X,'**',44(' '),*COST AND MATERIAL REQUIREMENTS OF BEAM*,4DS007960
      *5(' '),***)
      WRITE(6,603)
      WRITE(6,671)
671 FORMAT(1X,'**',8X,'ITEM',15X,'AMOUNT',12X,'COST',8X,'PERCENTAGE CF DS008000
      1TOTAL COST',6X,' ',39X,'**')
      WRITE(6,672) CSTTOT
672 FORMAT(1X,'**',87X,'**',7X,'TOTAL COST OF BEAM $ ',F8.2,2X,'**')
      WRITE(6,673) CONCV,CSTCON,CPRCST
673 FORMAT(1X,'**',6X,'CONCRETE*',9X,F7.2,' YD**3',5X,'$',F8.2,14X,F5.2DS008050
      1,' X',15X,'**',39X,'**')
      WRITE(6,674) STRFT,CSTSTR,SPRCST,CSTPFT
674 FORMAT(1X,'**',7X,'STRANDS',10X,F7.2,' FT',8X,'$',F8.2,14X,F5.2,' XDS008080
      1',15X,'**',7X,'CCST PER FOOT $ ',F8.2,2X,'**')
      WRITE(6,676)
676 FORMAT(1X,'**',127X,'**'/1X,'**',5X,'*DOES NOT INCLUDE END SECTION',9DS008110
      *3X,'**')
9020 WRITE(6,602)
C*****
C COMPUTE AND PRINTOUT CRITICAL DESIGN FACTORS
C*****
      WRITE(6,600)
      DO 800 J1=1,9
      DO 800 J2=1,5
800 KSYM(J1,J2)=EX(1)
      WRITE(6,602)
      WRITE(6,810)
810 FORMAT(1X,'**',52X,'CRITICAL DESIGN FACTORS',52X,'**')
      WRITE(6,602)
      WRITE(6,603)
      WRITE(6,811)
811 FORMAT(' *',23(' '),*RELEASE STRESSES',24(' '),**',21(' '),*SERVICDS008270
      1E LOAD STRESSES',21(' '),**'/ ' **',12X,'(SYMBOL X DENOTES STRESS ATDS008280
      2 ALLOWABLE)',13X,'**',13X,'(SYMBOL X DENOTES STRESS AT ALLOWABLE)',DS008290
      312X,'**'/ ' **',63X,'**',63X,'**')
      WRITE(6,812)
812 FORMAT(' *',2X,'SECTION *',7X,'STRESS TOP',8X,'**',6X,'STRESS BOTTD0S008320
      10M',6X,'**',2X,'SECTION *',7X,'STRESS TOP',8X,'**',6X,'STRESS BOTTD0S008330
      2M',6X,'**'/ ' **',11X,'**',9X,'(KSI)',11X,'**',10X,'(KSI)',10X,'**',11X,DS008340
      3*' **',9X,'(KSI)',11X,'**',10X,'(KSI)',10X,'**'/ ' **',11X,'**',25X,'**',25DS008350

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4X,'*',11X,'*',25X,'*',25X,'*')
DO 8101 J1=1,9
ZX=(J1-1)*ZL/10.
IF(J1.EQ.3) ZX=5.*ZL/40.
IF(J1.EQ.4) ZX=2.*ZL/10.
IF(J1.EQ.5) ZX=ZL/4.
IF(J1.GE.6) ZX=(J1-3)*ZL/10.
IF(J1.EQ.9) ZX=ALPHA*ZL
ZMJ=W*ZX/2.*(ZL-ZX)*12.
ZMJB=WB*ZX/2.*(ZL-ZX)*12.
ZMJBK=DLMCM(J1)*12.+BMMAX(J1)*12.
TAU=(ALPHA-ZX/ZL)/ALPHA
IF(TAU.LT.0.0) TAU=0.0
SUM10=0.0
SUM20=0.0
DO 804 J3=1,NR
SUM10=SUM10+(1./ACONC+D(J3)/ZT)*X(J3+1-JOPT)
804 SUM20=SUM20+(1./ACONC-D(J3)/ZB)*X(J3+1-JOPT)
TERM1=SUM10+NW*TAU/ZT*GSP*X(3-JOPT+NR)
TERM2=SUM20-NW*TAU/ZB*GSP*X(3-JOPT+NR)
STRESS(J1,1)=-((1.-ZINLOS)*FO*TERM1-ZMJ/ZT)*(-1.)
STRESS(J1,2)=-((1.-ZINLOS)*FO*TERM2+ZMJ/ZB)*(-1.)
STRESS(J1,3)=-((1.-ZLOSS)*FO*TERM1-ZMJ/ZT-ZMJBK/ZTK)*(-1.)
STRESS(J1,4)=-((1.-ZLCSS)*FO*TERM2+ZMJ/ZB+ZMJBK/ZBK)*(-1.)
STR=CTR1
SCR=CBR1
STS=CTS1
SCS=CBS1
IF(ZX.LE.ZL/10.+0.1) STR=CTR2
IF(ZX.LE.ZL/10.+0.1) SCR=CBR2
IF(ZX.LE.ZL/10.+0.1) STS=CTS2
IF(ZX.LE.ZL/10.+0.1) SCS=CBS2
C
C IF STRESS WITHIN 1 PERCENT OF ALLOWABLE, CALL IT CRITICAL.
C
IF(STRESS(J1,1).LE.-.00099*STR*SQRT(FPCR*1000.)) KSYM(J1,1)=EX(2)
IF(STRESS(J1,2).GE..99*SCR*FPCR) KSYM(J1,2)=EX(2)
IF(STRESS(J1,3).GE..99*SCS*F28) KSYM(J1,3)=EX(2)
IF(STRESS(J1,4).LE.-.00099*STS*SQRT(F28*1000.)) KSYM(J1,4)=EX(2)
IF(J1.NE.1) GO TO 9053

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DS008360
DS008370
DS008380
DS008390
DS008400
DS008410
DS008420
DS008430
DS008440
DS008450
DS008460
DS008470
DS008480
DS008490
DS008500
DS008510
DS008520
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DS008570
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DS008590
DS008600
DS008610
DS008620
DS008630
DS008640
DS008650
DS008660
DS008670
DS008680
DS008690
DS008700
DS008710
DS008720
DS008730
DS008740
DS008750

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      IF(STRESS(1,3).LE.-.00099*STS*SQRT(F28*1000.)) KSYM(1,3)=EX(2)      DS008760
      IF(STRESS(1,4).GE..99*SCS*F28) KSYM(1,4)=EX(2)                      DS008770
9053 CONTINUE                                                                DS008780
      IF(BB(NCON-J1+1).LE.0.0) KSYM(J1,5)=EX(2)                          DS008790
      J5=(ZX*10.+0.001)/ZL                                                DS008800
      IF(J1.EQ.3) WRITE(6,9110) STRESS(J1,1),KSYM(J1,1),STRESS(J1,2),    DS008810
      1KSYM(J1,2),STRESS(J1,3),KSYM(J1,3),STRESS(J1,4),KSYM(J1,4)        DS008820
      IF(J1.EQ.5) WRITE(6,9113) STRESS(J1,1),KSYM(J1,1),STRESS(J1,2),    DS008830
      1KSYM(J1,2),STRESS(J1,3),KSYM(J1,3),STRESS(J1,4),KSYM(J1,4)        DS008840
      IF(J1.EQ.9) WRITE(6,9111) STRESS(J1,1),KSYM(J1,1),STRESS(J1,2),    DS008850
      1KSYM(J1,2),STRESS(J1,3),KSYM(J1,3),STRESS(J1,4),KSYM(J1,4)        DS008860
      IF(J1.NE.3.AND.J1.NE.5.AND.J1.NE.9) WRITE(6,9112) J5,STRESS(J1,1), DS008870
      1KSYM(J1,1),STRESS(J1,2),KSYM(J1,2),J5,STRESS(J1,3),KSYM(J1,3),    DS008880
      2STRESS(J1,4),KSYM(J1,4)                                             DS008890
8101 CONTINUE                                                                DS008900
9110 FORMAT(1X,'*',4X,'5/40',3X,'*',2(6X,E11.4,3X,A1,4X,'*'),4X,        DS008910
      1'5/40',3X,'*',2(6X,E11.4,3X,A1,4X,'*'))                          DS008920
9111 FORMAT(1X,'*',4X,'HDPT',3X,'*',2(6X,E11.4,3X,A1,4X,'*'),4X,        DS008930
      1'HDPT',3X,'*',2(6X,E11.4,3X,A1,4X,'*'))                          DS008940
9113 FORMAT(1X,'*',4X,'1/4 ',3X,'*',2(6X,E11.4,3X,A1,4X,'*'),4X,        DS008950
      1'1/4 ',3X,'*',2(6X,E11.4,3X,A1,4X,'*'))                          DS008960
9112 FORMAT(1X,'*',4X,'11','/10',3X,'*',2(6X,E11.4,3X,A1,4X,'*'),4X,    DS008970
      111,'/10',3X,'*',2(6X,E11.4,3X,A1,4X,'*'))                       DS008980
      WRITE(6,814)                                                         DS008990
814  FORMAT(' ',11X,'*',25X,'*',25X,'*',11X,'*',25X,'*',25X,'*'/ ' ', DS009000
      111(' '),**',25(' '),**',25(' '),**',11(' '),**',25(' '),**',25(' 'DS009010
      2),**')                                                             DS009020
      WRITE(6,603)                                                         DS009030
      WRITE(6,822)                                                         DS009040
822  FORMAT(1X,'*',50(' '),*LIST OF DESIGN CONSTRAINTS',51(' '),**'/1X, DS009050
      1**',37X,'(SYMBOL X INDICATES CONSTRAINT CONTROLS FINAL DESIGN)', DS009060
      237X,'**')                                                         DS009070
      WRITE(6,603)                                                         DS009080
      IF(CBRI.GE.ALDEF=0.05) KSYM(6,5)=EX(2)                             DS009081
      IF(CBRI.LE.DEFMIN+0.05) KSYM(5,5)=EX(2)                           DS009082
      WRITE(6,805) KSYM(2,5),KSYM(4,5),KSYM(5,5),KSYM(1,5),KSYM(3,5),    DS009090
      *KSYM(6,5)                                                         DS009100
805  FORMAT(1X,'*',16X,'MINIMUM CONCRETE STRENGTH',3X,A1,11X,'ULTIMATE DS009110
      1MOMENT',3X,A1,11X,'MINIMUM INITIAL CAMBER',3X,A1,15X,'**'/1X,'**', DS009120
      216X,'MAXIMUM CONCRETE STRENGTH',3X,A1,11X,'CRACKING MOMENT',3X,A1,DS009130

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3 11X,'MAXIMUM INITIAL CAMBER',3X,A1,15X,'*') DS009140
WRITE(6,603) DS009150
WRITE(6,602) DS009160
C***** DS009170
C RECALCULATE AND PRINTOUT DESIGN SHEARS AND MOMENTS DS009180
C***** DS009190
WRITE(6,9045) DS009200
9045 FORMAT(/) DS009210
WRITE(6,602) DS009220
WRITE(6,9041) DS009230
9041 FORMAT(1X,'*',51X,'MOMENT AND SHEAR SUMMARY',52X,'*') DS009240
WRITE(6,602) DS009250
WRITE(6,603) DS009260
WRITE(6,9042) DS009270
9042 FORMAT(1X,'*',2X,'DISTANCE FROM',2X,'*', ' BM. WT. + SHEAR KEY ', DS009280
1' ',6X,'OTHER D.L.',5X,'*',9X,'L.L.',8X,'*',8X,'TOTAL',8X,'*',7X, DS009290
2'ULTIMATE',6X,'*/1X,'*',3X,'END OF BEAM',3X,'*',4(7X,'MOMENTS', DS009300
37X,'*'),8X,'SHEAR',8X,'*/1X,'*',6X,'(FT)',7X,'*',4(7X,'(KIP-FT)', DS009310
46X,'*'),8X,'(KIPS)',7X,'*/1X,'*',17X,'*',5(21X,'*')) DS009320
DO 8102 J1=1,9 DS009330
ZX=(J1-1)*ZL/10. DS009340
IF(J1.EQ.3) ZX=5.*ZL/40. DS009350
IF(J1.EQ.4) ZX=2.*ZL/10. DS009360
IF(J1.EQ.5) ZX=ZL/4. DS009370
IF(J1.GE.6) ZX=(J1-3)*ZL/10. DS009380
IF(J1.EQ.9) ZX=ALPHA*ZL DS009390
ZMJB=WB*ZX/2.*(ZL-ZX) DS009400
ZMJT=ZMJB+DLMOM(J1)+BM MAX(J1) DS009410
VU(J1)=1.444*(WB*(ZL/2.-ZX)+DL SHR(J1)+5./3.*BV MAX(J1)) DS009420
IF(J1.NE.9) WRITE(6,9043) ZX,ZMJB,DLMOM(J1),BM MAX(J1),ZMJT,VU(J1) DS009430
IF(J1.EQ.9) WRITE(6,9047) ZMJB,DLMOM(J1),BM MAX(J1),ZMJT,VU(J1) DS009440
8102 CONTINUE DS009450
9043 FORMAT(1X,'*',6X,F5.2,6X,'*',5(4X,E12.5,5X,'*')) DS009460
9047 FORMAT(1X,'*',6X,'HDPT',7X,'*',5(4X,E12.5,5X,'*')) DS009470
WRITE(6,603) DS009480
WRITE(6,9044) ULTMRC,ZMUL,ZCRACK DS009490
9044 FORMAT(1X,'*',42X,'ULTIMATE MOMENT REQUIRED = ',E12.5,' KIP-FT',39 DS009500
1X,'*/1X,'*',42X,'ULTIMATE MOMENT CAPACITY = ',E12.5,' KIP-FT',39X DS009510
2,'*/1X,'*',42X,'CRACKING MOMENT CAPACITY = ',E12.5,' KIP-FT',39X, DS009520
3'*) DS009530

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WRITE(6,603)
WRITE(6,602)
WRITE(6,641)
641 FORMAT(1H1)
NCOUNT=NCOUNT+1
GO TO 3007
2500 CONTINUE
STOP
END

DS009540
DS009550
DS009560
DS009570
DS009580
DS009590
DS009600
DS009610
DS009620

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SUBROUTINE EQGEN
COMMON/BLK1/ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,TDIM,WDDIM,
1WHDIM,XDIM,YDIM,ACONC,BINERT,DTOP,DBOT,ZT,ZB,ACONCK,BINERK,DTOPK,
2DBOTK,ZTK,ZBK,ZL,F28,JOPT,ASSCLR,ASSPLS,APRIME,CBOT,COSTFT,
3DEFMIN,ALDEF,NRAV,NWHEEL,DISTF,FPCMAX,FPCMIN,ELASC,ULTMRQ,CTOP,
4W,WB,FO,DCR,NR,HDPT,NW,ALPHA
COMMON/BLK2/PAXLE(18),NWHL(18),STRMAX(26),FCONC(10),DCONC(10),
1G(11),F(11),D(10),BMMAX(9),BVMAX(9),CBRMAX(5),PRLMAX(5),DLMOM(9),
2DLSHR(9),PECRK(11,4),ZLOS(4)
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2,
1CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP,GSP
COMMON/YZ/Z1,Z2,Z3,Z4,Y1,Y2,Y3,Y4
COMMON/D314/N,M,OBJ,KP1,K,NCON,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,
1SUM,IGNOR,IT(4),JCONT,X(150),ARRAY(118,150),B(150),XD(150)
COMMON/DUMP/ TITLE(3,54),YJ(11),FROW(26),
1PEF(50,3),ZNE(11),KKODE(4),ZMCR(4)
FO=0.7*AS*FPS
W=UWC*ACONC/144.
WB=UWC*ACONCK/144.
JRR=48+7*NR
JCC=12+2*NR
IF(JOPT.NE.1) JRR=29+7*NR
IF(JOPT.NE.1) JCC=3+2*NR
DO 2 J1=1,JRR
ARRAY(J1,K+1)=0.0
DO 2 J2=1,JCC
2 ARRAY(J1,J2)=0.0
C*****
C OBJECTIVE FUNCTION
C*****
IF(JOPT.EQ.1) GO TO 3
ARRAY(1,1)=G(1)*ACONC*ZL/1944.
ARRAY(1,1+NR+2)=0.1*G(1)*ACONC*ZL/1944.
DO 5 J1=1,NR
5 ARRAY(1,J1+1)=CCSTFT*ZL
GO TO 8
3 DO 4 J1=1,NR
4 ARRAY(1,J1)=COSTFT*ZL
C
C PUT IN TOKEN COST FOR DRAPING.

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DS009630
DS009640
DS009650
DS009660
DS009670
DS009680
DS009690
DS009700
DS009710
DS009720
DS009730
DS009740
DS009750
DS009760
DS009770
DS009780
DS009790
DS009800
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DS009920
DS009930
DS009940
DS009950
DS009960
DS009970
DS009980
DS009990
DS010000
DS010010
DS010020

C		DS010030
	ARRAY(1,NR+2)=(CCSTFT*ZL/(NR*NRAV))*0.1	DS010040
	DO 6 J1=1,10	DS010050
6	ARRAY(1,2*NR+2+J1)=ACONC*ZL/1944.*(G(J1+1)-G(J1))	DS010060
8	CONTINUE	DS010070
	DO 1 J1=1,M	DS010080
1	ARRAY(1,J1)=-ARRAY(1,J1)	DS010090
C	*****	DS010100
C	RELEASE STRESSES = CONSTRAINTS 1 THRU 8	DS010110
C	*****	DS010120
	DO 10 J1=2,9,2	DS010130
	IF(J1.EQ.2) ZX=ALPHA*ZL	DS010140
	IF(J1.EQ.4) ZX=5.*ZL/40.	DS010150
	IF(J1.EQ.6) ZX=ZL/10.	DS010160
	IF(J1.EQ.8) ZX=0.0	DS010170
	IF(J1.EQ.2) TAU=0.	DS010180
	IF(J1.EQ.4) TAU=(ALPHA=0.125)/ALPHA	DS010190
	IF(J1.EQ.6) TAU=(ALPHA=0.10)/ALPHA	DS010200
	IF(J1.EQ.8) TAU=1.0	DS010210
	IF(TAU.LT.0.0) TAU=0.0	DS010220
	ST=CTR1	DS010230
	SC=CBR1	DS010240
	IF(ZX.LE.ZL/10.+0.1) ST=CTR2	DS010250
	IF(ZX.LE.ZL/10.+0.1) SC=CBR2	DS010260
	DO 12 J2=1,NR	DS010270
	ARRAY(J1,1-JOPT+J2)=-((1.-ASSCLR)*FO*(1./ACONC+D(J2)/ZT)	DS010280
12	ARRAY(J1+1,1-JOPT+J2)= ((1.-ASSCLR)*FO*(1./ACONC-D(J2)/ZB)	DS010290
	ARRAY(J1,1-JOPT+NR+2)=-((1.-ASSCLR)*FO*NW/ZT*TAU*GSP	DS010300
	ARRAY(J1+1,1-JOPT+NR+2)=-((1.-ASSCLR)*FO*NW/ZB*TAU*GSP	DS010310
	IF(JOPT.EQ.0) GC TO 9	DS010320
	DO 14 J2=1,10	DS010330
	ARRAY(J1,2*NR+2+J2)=-0.003727*ST	DS010340
14	ARRAY(J1+1,2*NR+2+J2)=-0.50*SC	DS010350
	ARRAY(J1,K+1)=0.50*W*ZX*(ZL-ZX)*12./ZT+0.063366*ST	DS010360
	ARRAY(J1+1,K+1)=0.50*W*ZX*(ZL-ZX)*12./ZB+4.0*SC	DS010370
	GO TO 10	DS010380
9	ARRAY(J1,1)=-0.0074535*ST	DS010390
	ARRAY(J1+1,1)=-SC	DS010400
	ARRAY(J1,K+1)=0.50*W*ZX*(ZL-ZX)*12./ZT+0.033552*ST	DS010410
	ARRAY(J1+1,K+1)=0.50*W*ZX*(ZL-ZX)*12./ZB	DS010420


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10 CONTINUE
C*****DS010430
C SERVICE LOAD STRESSES = CONSTRAINTS 9 THRU 22DS010440
C*****DS010450
C POINTS 1,2,3,4, AND 5 TENTHS ; AND 5L/40DS010460
DO 18 J1=1,6DS010470
JR=8+2*J1DS010480
TAU=(ALPHA-J1/10.)/ALPHA DS010490
IF(J1.EQ.6) TAU=(ALPHA=0.125)/ALPHA DS010500
IF(TAU.LT.0.0) TAU=0.0 DS010510
ZX=J1/10.*ZL DS010520
IF(J1.EQ.6) ZX=5.*ZL/40. DS010530
ZMJ=0.50*WE*ZX*(ZL-ZX)*12. DS010540
ZMJB=BMMAX(J1+2)*12.+DLMOM(J1+2)*12. DS010550
IF(J1.EQ.1) ZMJB=BMMAX(2)*12.+DLMOM(2)*12. DS010560
IF(J1.EQ.2) ZMJB=BMMAX(3)*12.+DLMOM(3)*12. DS010570
ST=CTS1 DS010580
SC=CBS1 DS010590
IF(ZX.LE.ZL/10.+0.1) ST=CTS2 DS010600
IF(ZX.LE.ZL/10.+0.1) SC=CBS2 DS010610
DO 20 J2=1,NR DS010620
ARRAY(JR,1-JOPT+J2)=(1.-ASSPLS)*FO*(1./ACONC+D(J2)/ZT) DS010630
20 ARRAY(JR+1,1-JOPT+J2)=-(1.-ASSPLS)*FO*(1./ACONC+D(J2)/ZB) DS010640
ARRAY(JR,1-JOPT+NR+2)=(1.-ASSPLS)*FO*NW/ZT*TAU*GSP DS010650
ARRAY(JR+1,1-JOPT+NR+2)=-(1.-ASSPLS)*FO*NW/ZB*TAU*GSP DS010660
IF(JOPT.EQ.0) GO TO 21 DS010670
DO 22 J2=1,10 DS010680
ARRAY(JR,2*NR+2+J2)=-SC*(F(J2+1)-F(J2)) DS010690
22 ARRAY(JR+1,2*NR+2+J2)=-0.0074535*ST*(F(J2+1)-F(J2)) DS010700
ARRAY(JR,K+1)=-ZMJ/ZT-ZMJB/ZTK+SC*F(1) DS010710
ARRAY(JR+1,K+1)=-ZMJ/ZB-ZMJB/ZBK+0.0074535*ST*F(1)+0.033552*ST DS010720
GO TO 18 DS010730
21 ARRAY(JR,K+1)=-ZMJ/ZT-ZMJB/ZTK+SC*F28 DS010740
ARRAY(JR+1,K+1)=-ZMJ/ZB-ZMJB/ZBK+0.031623*ST*SQRT(F28) DS010750
18 CONTINUE DS010760
C ENDS OF THE BEAM DS010770
DO 24 J2=1,NR DS010780
ARRAY(22,1-JOPT+J2)=-(1.-ASSPLS)*FO*(1./ACONC+D(J2)/ZT) DS010790
24 ARRAY(23,1-JOPT+J2)= (1.-ASSPLS)*FO*(1./ACONC+D(J2)/ZE) DS010800
ARRAY(22,1-JOPT+NR+2)=-(1.-ASSPLS)*FO*NW/ZT*GSP DS010810
DS010820

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        ARRAY(23,1-JOPT+NR+2)=-(1.-ASSPLS)*F0*NW/ZB*GSP                      DS010830
        IF(JOPT.EQ.0) GO TO 25                                                DS010840
        DO 23 J2=1,10                                                         DS010850
        ARRAY(22,2*NR+2)=0.0074535*CTS2*(F(J2+1)-F(J2))                     DS010860
23      ARRAY(23,2*NR+2)=CBS2*(F(J2+1)-F(J2))                               DS010870
        ARRAY(22,K+1)=0.0074535*CTS2*F(1)+0.033552*CTS2                     DS010880
        ARRAY(23,K+1)=CBS2*F(1)                                              DS010890
        GO TO 27                                                              DS010900
25      ARRAY(22,K+1)=0.031623*CTS2*SQRT(F28)                                DS010910
        ARRAY(23,K+1)=CBS2*F28                                              DS010920
27      CONTINUE                                                            DS010930
C *****                                                                    DS010940
C      SUFFICIENT NUMBER OF STRANDS IN ROW FOR DRAPING                      DS010950
C      CONSTRAINTS 23 THRU 22 + NR                                          DS010960
C *****                                                                    DS010970
        DO 26 J1 = 1, NR                                                       DS010980
        ARRAY(23+J1,1-JOPT+J1)=1.                                           DS010990
26      ARRAY(23+J1,1-JCPT+2+NR+J1)=NW                                       DS011000
C *****                                                                    DS011010
C      CONTIGUOUS DRAPED STRANDS                                           DS011020
C      CONSTRAINTS 23 + NR THRU 21 + 2*NR                                   DS011030
C *****                                                                    DS011040
        DO 28 J1 = 2, NR                                                       DS011050
        ARRAY(23+NR+J1-1,1-JCPT+NR+2+J1)=J1-1.                             DS011060
        ARRAY(23+NR+J1-1,1-JOPT+NR+2+J1-1)=-(J1-1.)                       DS011070
28      ARRAY(23+NR+J1-1,1-JOPT+NR+1)=1.                                     DS011080
C *****                                                                    DS011090
C      UPPER BOUND ON EN                                                    DS011100
C      CONSTRAINTS 22 + 2*NR THRU 21 + 3*NR                                DS011110
C *****                                                                    DS011120
        DO 30 J1 = 1, NR                                                       DS011130
        ARRAY(22+2*NR+J1,1-JOPT+NR+2) = 1.                                  DS011140
        ARRAY(22+2*NR+J1,1-JOPT+NR+1) = J1                                  DS011150
        ARRAY(22+2*NR+J1,K+1)=(J1-1)**2+2.*(J1-1)+1.                       DS011160
        DO 30 J2 = 1, NR                                                       DS011170
30      ARRAY(22+2*NR+J1,1-JOPT+NR+2+J2)=-(NRAV-2.*(J1-1.))-1.             DS011180
C *****                                                                    DS011190
C      UPPER AND LOWER BOUNDS ON NB                                         DS011200
C      CONSTRAINTS 22+3*NR AND 23+3*NR                                       DS011210
C *****                                                                    DS011220

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        ARRAY(23+3*NR, 1-JOPT+NR+1) = 1.                      DS011230
        ARRAY(23+3*NR,K+1) = NR                                DS011240
        ARRAY(24+3*NR,1-JCPT+NR+1)=-1.                        DS011250
        ARRAY(24 + 3*NR, K+1) = -1.                            DS011260
C*****DS011270
C      CONSTRAINTS TO INSURE THAT IF (NB.GE.I+1) THEN II = 0  DS011280
C      CONSTRAINTS 24 + 3*NR THRU 22 + 4*NR                    DS011290
C*****DS011300
        DO 32 J1 = 2, NR                                         DS011310
            ARRAY(24 + 3*NR + J1-1,1-JOPT+NR+1) = 1.           DS011320
            ARRAY(24 + 3*NR+J1-1,1-JOPT+NR+2+J1-1) = NR         DS011330
        32 ARRAY(24+3*NR+J1-1,K+1)=NR+J1-1.                    CS011340
C*****DS011350
C      CONSTRAINTS TO INSURE THAT IF (NB.EQ.I) THEN II = 1    DS011360
C      CONSTRAINTS 23 + 4 * NR THRU 22 + 5*NR                  DS011370
C*****DS011380
        DO 34 J1=1,NR                                           DS011390
            ARRAY(23 + 4*NR + J1,1-JOPT+NR+1) = -1.             DS011400
            ARRAY(23 + 4*NR+J1,1-JOPT+NR+2+J1) = -1.            DS011410
            ARRAY(23 + 4*NR+J1,K+1) = -(J1+1.)                  DS011420
            IF(J1.EQ.1) GO TO 34                                   DS011430
        DO 36 J2 = 2,J1                                           DS011440
        36 ARRAY(23+4*NR+J1,1-JOPT+NR+2+J2-1) = -(J1+1.)       DS011450
        34 CONTINUE                                              DS011460
C*****DS011470
C      MAXIMUM NUMBER OF STRANDS PER ROW                        DS011480
C      CCNSTRANTS 23 + 5*NR THRU 22 + 6*NR                      DS011490
C*****DS011500
        DO 38 J1 = 1,NR                                         DS011510
            ARRAY(23+5*NR+J1,1-JCPT+J1) = 1.                    DS011520
        38 ARRAY(23+5*NR+J1,K+1)=STRMAX(J1)                     DS011530
C*****DS011540
C      CONSTRAINTS TO INSURE PROPER RELEASE STRENGTH REPRESENTATION DS011550
C      JOPT = 1, CONSTRAINTS 23 + 6*NR THRU 31 + 6*NR          DS011560
C*****DS011570
        IF(JOPT.EQ.0) GO TO 39                                   DS011580
        DO 40 J1 = 1,9                                           DS011590
            ARRAY(23+6*NR+J1,2*NR+2+J1) = -1.                   DS011600
        40 ARRAY(23+6*NR+J1,2*NR+2+J1+1) = 1.                   DS011610
C*****DS011620

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```

C      BOUNDS ON RELEASE CAMBER                                DS011630
C      JOPT = 0, CONSTRAINTS 23 + 6*NR THRU 24 + 6*NR          DS011640
C      JOPT=1, CONSTRAINTS 32+6*NR THRU 33+6*NR                DS011650
C *****DS011660
  39 IF(JOPT.EQ.0) JR=24+6*NR                                DS011670
    IF(JOPT.NE.0) JR=33+6*NR                                DS011680
    DO 42 J1=1,NR                                            DS011690
      ARRAY(JR,1-JOPT+J1)=-(ZL*12.)**2/8.*(1.-ASSCLR)*FO*D(J1) DS011700
      ARRAY(JR+1,1-JOPT+J1)=(ZL*12.)**2/8.*(1.-ASSCLR)*FO*D(J1) DS011710
      ARRAY(JR,1-JOPT+NR+2)=-(ALPHA*ZL*12.)**2/6.*(1.-ASSCLR)*FO*NW*GSP DS011720
  42 ARRAY(JR+1,1-JOPT+NR+2)=(ALPHA*ZL*12.)**2/6.*(1.-ASSCLR)*FO*NW*GSP DS011730
    IF(JOPT.EQ.0) GO TO 41
    DO 44 J1=1,10                                           DS011740
      ARRAY(JR,2*NR+2+J1)=-0.117634*BINERT*ELASC*ALDEF      DS011750
  44 ARRAY(JR+1,2*NR+2+J1)=0.117634*BINERT*ELASC*DEFMIN      DS011760
      ARRAY(JR,K+1)=2.0 *ELASC*BINERT*ALDEF+22.5*W*ZL**4      DS011770
      ARRAY(JR+1,K+1)=-2.0 *ELASC*BINERT*DEFMIN-22.5*W*ZL**4 DS011780
      GO TO 43                                                DS011790
  41 ARRAY(JR,1)=-0.2352523*ELASC*BINERT*ALDEF              DS011800
      ARRAY(JR+1,1)=0.2352523*ELASC*BINERT*DEFMIN            DS011810
      ARRAY(JR,K+1)=1.058991*ELASC*BINERT*ALDEF+22.5*W*ZL**4 DS011820
      ARRAY(JR+1,K+1)=-1.058991*ELASC*BINERT*DEFMIN-22.5*W*ZL**4 DS011830
  43 ARRAY(JR,K+1)=ARRAY(JR,K+1)/1.E+04                     DS011840
      ARRAY(JR+1,K+1)=ARRAY(JR+1,K+1)/1.E+04                 DS011850
      DO 46 J1=1,M                                           DS011860
        ARRAY(JR,J1)=ARRAY(JR,J1)/1.E+04                     DS011870
  46 ARRAY(JR+1,J1)=ARRAY(JR+1,J1)/1.E+04                     DS011880
C *****DS011890
C      ADEQUATE ULTIMATE MOMENT CAPACITY                      DS011900
C      ULTIMATE MOMENT CAPACITY MU .GE. 1.2 * CRACKING MOMENT DS011910
C      JOPT=0, CONSTRAINTS 25+6*NR AND 26+6*NR                DS011920
C      JOPT=1, CONSTRAINTS 34+6*NR AND 35+6*NR                DS011930
C *****DS011940
C      SET UP NO. STRANDS AND STRAND ECCENTRICITY ARRAY        DS011950
C                                                                DS011960
C                                                                DS011970
C                                                                DS011980
      DO 62 J1=1,NR                                           DS011990
  62 FROW(J1)=0.                                              DS012000
      SUM=0.0                                                DS012010
      KNT=0                                                  DS012020

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```

      PF=0.
      DO 64 J1=1,NR
67    IF(FROW(J1).EQ.STRMAX(J1)) GO TO 64
      IADD=2
      IF(STRMAX(J1)=FROW(J1).LE.1) IADD=1
      FROW(J1)=FROW(J1)+IADD
      PF=PF+IADD*(-D(J1))
      SUM=SUM+IADD
      KNT=KNT+1
      PEF(KNT,1)=SUM
      PEF(KNT,2)=PF
      PEF(KNT,3)=PF/SUM
      GO TO 67
64    CONTINUE
C    SET UP FOR CALLS TO ULTMP
      ZMDL=W*ZL**2/8.
      DO 63 J1=1,4
63    ZLOS(J1)=0.1*J1
      DO 65 J1=1,11
      ZNE(J1)=0.0
      DO 65 J2=1,4
65    PECRK(J1,J2)=0.
      JSTOP=11
      IF(JOPT.NE.0) GO TO 66
      F(1)=F28
      JSTOP=1
C    GENERATE TOTAL FORCE ECCENTRICITIES FOR ULTIMATE MOMENT AND
C    CRACKING MOMENT CONSTRAINTS
66    FPCMIN=4.0
      DO 84 J1=1,JSTOP
      IF(J1.EQ.1) GO TO 69
      IF(F(J1).NE.F(J1-1)) GO TO 69
      ZNE(J1)=ZNE(J1-1)
      DO 68 J2=1,4
68    PECRK(J1,J2)=PECRK(J1-1,J2)
      GO TO 84
69    FPCBM=F(J1)
      DO 70 J2=1,4
70    KCODE(J2)=0
      KODEMU=0

```

```

DS012030
DS012040
DS012050
DS012060
DS012070
DS012080
DS012090
DS012100
DS012110
DS012120
DS012130
DS012140
DS012150
DS012160
DS012170
DS012180
DS012190
DS012200
DS012210
DS012220
DS012230
DS012240
DS012250
DS012260
DS012270
DS012280
DS012290
DS012300
DS012310
DS012320
DS012330
DS012340
DS012350
DS012360
DS012370
DS012380
DS012390
DS012400
DS012410
DS012420

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```

      ZMOLD=0.
      DO 82 J2=1,KNT
      ASTL=AS*PEF(J2,1)
      DD=DTOP+PEF(J2,3)
      CALL ULTMP(ASTL,FPCBM,FPS,APRIME,FPL,DD,DDIM,FSY,DCR,
      *Y1,Y2,Y3,Y4,Z1,Z2,Z3,Z4,ZMUL)
      DO 72 J3=1,4
      ZMCR(J3)=(1.-ZLOS(J3))*ZBK*FO*(PEF(J2,1)/ACONC+PEF(J2,2)/ZB)
      1+7.5*ZBK*.031623*SQRT(FPCBM)=ZBK*ZMDL*12./ZB
72  ZMCR(J3)=ZMCR(J3)*1.2/12.
      DO 73 J3=1,4
      IF(KKODE(J3).EQ.1) GO TO 73
      IF(ZMUL.LT.ZMCR(J3)) GO TO 73
      PECRK(J1,J3)=PEF(J2,2)
      KKODE(J3)=1
73  CONTINUE
74  IF(KODEMU.EQ.1) GO TO 78
      IF(ZMUL.GE.ULTMRQ) GO TO 76
      E1=ZMOLD
      IF(E1.LT.ZMUL) ZMOLD=ZMUL
      IF(E1.LT.ZMUL) GO TO 82
      FPCMIN=4.0+(J1-1)*0.5
      GO TO 84
76  ZNE(J1)=PEF(J2,2)
      KODEMU=1
78  DO 80 J3=1,4
      IF(KKODE(J3).EQ.0) GO TO 82
80  CONTINUE
      IF(KODEMU.EQ.1) GO TO 84
82  CONTINUE
84  CONTINUE
C
C      FORM ULTIMATE MOMENT AND CRACKING MOMENT CONSTRAINTS
C
      IF(JOPT.EQ.0) JR=26+6*NR
      IF(JOPT.NE.0) JR=35+6*NR
      DO 100 J1=1,NR
      ARRAY(JR,1-JOPT+J1)=D(J1)
100  ARRAY(JR+1,1-JOPT+J1)=D(J1)
      IF(JOPT.EQ.0) GO TO 101

```

```

DS012430
DS012440
DS012450
DS012460
DS012470
DS012480
DS012490
DS012500
DS012510
DS012520
DS012530
DS012540
DS012550
DS012560
DS012570
DS012580
DS012590
DS012600
DS012610
DS012620
DS012630
DS012640
DS012650
DS012660
DS012670
DS012680
DS012690
DS012700
DS012710
DS012720
DS012730
DS012740
DS012750
DS012760
DS012770
DS012780
DS012790
DS012800
DS012810
DS012820

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DO 102 J1=1,10 DS012830
  ARRAY(JR,2*NR+2+J1)=ZNE(J1+1)-ZNE(J1) DS012840
102 ARRAY(JR+1,2*NR+2+J1)=PECRK(J1+1,1)-PECRK(J1,1) DS012850
101 ARRAY(JR,K+1)=-ZNE(1) DS012860
  ARRAY(JR+1,K+1)=-PECRK(1,1) DS012870
C***** DS012880
C LOWER AND UPPER BOUNDS ON CONCRETE STRENGTH DS012890
C JOPT=0, CONSTRAINTS 27+6*NR AND 28+6*NR DS012900
C JOPT=1, CONSTRAINTS 36+6*NR AND 37+6*NR DS012910
C***** DS012920
  IF(JOPT.EQ.0) GO TO 103 DS012930
  DO 104 J1=1,10 DS012940
    ARRAY(37+6*NR,2*NR+2+J1)=-0.5 DS012950
104 ARRAY(38+6*NR,2*NR+2+J1)=0.5 DS012960
    ARRAY(37+6*NR,K+1)=4.0-FPCMIN DS012970
    ARRAY(38+6*NR,K+1)=FPCMAX=4.0 DS012980
    GO TO 105 DS012990
103 ARRAY(28+6*NR,1)=-1. DS013000
    ARRAY(29+6*NR,1)=1. DS013010
    ARRAY(28+6*NR,K+1)=-4.0 DS013020
    ARRAY(29+6*NR,K+1)=F28 DS013030
105 CONTINUE DS013040
C***** DS013050
C IF NSMAX IS ODD, NSI=NO. OF STRANDS DS013060
C IF NSMAX IS EVEN, NSI=1/2 NO. OF STRANDS DS013070
C***** DS013080
  IF(JOPT.EQ.0) J4=29+6*NR DS013090
  IF(JOPT.EQ.1) J4=38+6*NR DS013100
  DO 120 J1=1,NR DS013110
    NSMAX=STRMAX(J1) DS013120
    IF(NSMAX/2*2.NE.NSMAX) GO TO 120 DS013130
    DO 121 J2=1,J4 DS013140
121 ARRAY(J2,1=JOPT+J1)=2.*ARRAY(J2,1=JOPT+J1) DS013150
120 CONTINUE DS013160
C***** DS013170
C WRITE ARRAY ON UNIT (3) DS013180
C***** DS013190
  DO 111 J2=1,M DS013200
111 WRITE(3) (ARRAY(J3,J2),J3=2,J4),ARRAY(1,J2) DS013210
    WRITE(3) (ARRAY(J3,K+1),J3=2,J4),ARRAY(1,K+1) DS013220

```

REWIND 3	DS013230
C*****	DS013240
C CONSTRAINTS TO APPROXIMATE BINARY VARIABLES IN LP SOLUTION	DS013250
C JOPT=0, CONSTRAINTS 29+6*NR THRU 28+7*NR	DS013260
C JOPT=1, CONSTRAINTS 38+6*NR THRU 47+7*NR	DS013270
C*****	DS013280
JR=39+6*NR	DS013290
IF(JOPT.EQ.0) JR=30+6*NR	DS013300
DO 106 J1=1, NR	DS013310
ARRAY(JR+J1-1, 1-JOPT+NR+2+J1)=1.	DS013320
106 ARRAY(JR+J1-1, K+1)=1.	DS013330
IF(JOPT.EQ.0) GO TO 107	DS013340
DO 108 J1=1, 10	DS013350
ARRAY(JR+NR+J1-1, 2*NR+2+J1)=1.	DS013360
108 ARRAY(JR+NR+J1-1, K+1)=1.	DS013370
107 CONTINUE	DS013380
RETURN	DS013390
END	DS013400


```

SUBROUTINE ULTMP(ASTAR,FPCBM,FPS,ASPRM,FPL,D,DPTH,FSY,DCR,
*Y1,Y2,Y3,Y4,Z1,Z2,Z3,Z4,ZMUL)
CLONG=0.2
ESINI=0.7*FPS*(1.-CLONG)/28.E+03
CON1=(FPL/28000.)*(1.+(FPS=FPL)/(FPS=2.*FPL))
CON2=-(FPL/28000.)*FPL*(FPS=FPL)**2/(FPS=2.*FPL)
BEFF=2.*(Y1+Y2+Y3)
THK=Z1
Z4MZ3=Z4-Z3
IF(ABS(Z4-Z3).LE.1.E-06) Z4MZ3=1.E-06
Z2MZ1=Z2-Z1
IF(ABS(Z2-Z1).LE.1.E-06) Z2MZ1=1.E-06
C*****DS013530
C**** POSITIVE MOMENT CAPACITY = N.A. IN SLAB DS013540
C*****DS013550
C
CHECK TO SEE IF N.A. IN SLAB
C
PSTAR=ASTAR/(BEFF*D)
FSUSTR=FPS*(1.-0.5*PSTAR*FPS/FPCBM)
T=ASTAR*FSUSTR
CC=.833*FPCBM*BEFF*THK
IF(CC.LT.T) GO TO 10
C
C N.A. IN SLAB
C
ZMUL =ASTAR*FSUSTR*D*(1.-0.6*PSTAR*FSUSTR/FPCBM )/12
RI=PSTAR*FSUSTR/FPCBM
IF(RI.GT.0.3) ZMUL =0.25*FPCBM *BEFF*D**2/12.
RETURN
C*****DS013710
C**** POSITIVE MOMENT CAPACITY = N.A. BELOW SLAB DS013720
C*****DS013730
10 CONTINUE
C
BEGIN ITERATION TO LOCATE N.A.
C
JCNT=0
X=0.
12 X=X+0.25

```

DS013410
DS013420
DS013430
DS013440
DS013450
DS013460
DS013470
DS013480
DS013490
DS013500
DS013510
DS013520
DS013530
DS013540
DS013550
DS013560
DS013570
DS013580
DS013590
DS013600
DS013610
DS013620
DS013630
DS013640
DS013650
DS013660
DS013670
DS013680
DS013690
DS013700
DS013710
DS013720
DS013730
DS013740
DS013750
DS013760
DS013770
DS013780
DS013790
DS013800

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13 JCNT=JCNT+1
IF(X.GT.DPTH) ZMUL=0.
IF(X.GT.DPTH) RETURN
C
C   COMPUTE STRAND STRAIN AND FORCE IN DECK STEEL
C
ES=.003*(D-X)/X+ESINI
ESP=.003*(X-DCR)/X
CS=29.E+03*ABS(ESP)
IF(CS.GT.FSY) CS=FSY
IF(ESP.LE.0) CS=-CS
CS=CS*ASPRM
C
C   COMPUTE RESULTANT COMPRESSIVE FORCE ON CONCRETE AND ITS LOCATION
C
KODE=1
GO TO 1000
14 DBAR=D-YC
CC=C*.833*FPCBM
CTOT=CS+CC
GO TO 2000
C
C   COMPUTE STRAND STRESS AND STRAND FORCE
C
16 T=ASTAR*FS
SUMFOR=T-CTOT
IF(SUMFOR.LT.0.) GO TO 18
IF(JCNT.EQ.2) GO TO 17
SAVEF1=SUMFOR
SAVEX1=X
GO TO 12
17 SAVEF2=SUMFOR
SAVEX2=X
X=SAVEX1+(SAVEX2-SAVEX1)*SAVEF1/(SAVEF1-SAVEF2)
IF(X=SAVEX1.LT..25) X=SAVEX1+.25
JCNT=0
GO TO 13
18 ZMUL=(CC*DBAR+CS*(D-DCR))/12.
GO TO 28
C*****
DS013810
DS013820
DS013830
DS013840
DS013850
DS013860
DS013870
DS013880
DS013890
DS013900
DS013910
DS013920
DS013930
DS013940
DS013950
DS013960
DS013970
DS013980
DS013990
DS014000
DS014010
DS014020
DS014030
DS014040
DS014050
DS014060
DS014070
DS014080
DS014090
DS014100
DS014110
DS014120
DS014130
DS014140
DS014150
DS014160
DS014170
DS014180
DS014190
DS014200

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```

C*****THIS SECTION COMPUTES CONCRETE COMPRESSION AREA AND ITS C.G.      DS014210
C*****DS014220
1000 C =(Y1*BRACK(0.,X,Z1)+Y2*X+Y3*BRACK(0.,X,Z3)+EFACK(Z1,X,Z2)*Y4      DS014230
*   =0.5*Y4*BRACK(Z1,X,Z2)**2/Z2MZ1+BRACK(Z3,X,Z4)*Y3=0.5*Y3*      DS014240
*   BRACK(Z3,X,Z4)**2/Z4MZ3)*2.      DS014250
YC=(0.5*Y1*BRACK(0.,X,Z1)**2+0.5*Y2*X**2+0.5*Y3*BRACK(0.,X,Z3)**2      DS014260
*   +Y4*Z1*BRACK(Z1,X,Z2)+0.5*Y4*BRACK(Z1,X,Z2)**2=0.5*Z1*Y4*      DS014270
*   BRACK(Z1,X,Z2)**2/Z2MZ1=.33333*Y4*BRACK(Z1,X,Z2)**3/Z2MZ1      DS014280
*   +Y3*Z3*BRACK(Z3,X,Z4)+0.5*Y3*BRACK(Z3,X,Z4)**2      DS014290
*   =0.5*Z3*Y3*BRACK(Z3,X,Z4)**2/Z4MZ3=.33333*Y3      DS014300
*   *BRACK(Z3,X,Z4)**3/Z4MZ3)*2./C      DS014310
GO TO 14      DS014320
C*****DS014330
C***** THIS SECTION COMPUTES STRAND STRESS      DS014340
C*****DS014350
2000 FS=ES*28000      DS014360
IF(FS.GT.FPL) GO TO 2002      DS014370
2002 FS=.5*FPS+.5*SQRT(FPS**2=4.*CON2/(ES=CON1))      DS014380
GO TO 16      DS014390
28 RETURN      DS014400
END      DS014410

```

FUNCTION BRACK(ZL,X,ZU)
IF(X.LE.ZL) BRACK=0.
IF(ZL.LT.X.AND.X.LE.ZU) BRACK=X-ZL
IF(X.GT.ZU) BRACK=ZU-ZL
RETURN
END

DS014420
DS014430
DS014440
DS014450
DS014460
DS014470

SUBROUTINE DEFIN
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2,
1 CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP,GSP
UWC=.150
HUM=50.
AS=0.153
FPS=270.
FPL=0.63*FPS
CTR1=7.5
CTR2=7.5
CBR1=0.6
CBR2=0.6
CTS1=6.0
CTS2=6.0
CBS1=0.4
CBS2=0.4
CREEP1=0.
CREEP2=0.
SHRK1=0.
SHRK2=0.
RATNOD=6.0
FSY=60.
GSP=2.00
ASTIRP=0.11
RETURN
END

DS014480
DS014490
DS014500
DS014510
DS014520
DS014530
DS014540
DS014550
DS014560
DS014570
DS014580
DS014590
DS014600
DS014610
DS014620
DS014630
DS014640
DS014650
DS014660
DS014670
DS014680
DS014690
DS014700
DS014710
DS014720
DS014730

SUBROUTINE PROPTY	DS014740
REAL*4 I1,I2,I3,I4,I5,I6,I7,I8,I9,I10,I11,I12 ,I13,I14,I15	DS014750
COMMON/BLK1/ADIM,BDIM,CDIM,DDIM,EDIM,FDIM,GDIM,HDIM,TDIM,WDDIM,	DS014760
1 WHDIM,XDIM,YDIM,ACONC,BINERT,DTOP,DBOT,ZT,ZB,ACONCK,BINERK,DTOPK,	DS014770
2 DBOTK,ZTK,ZBK,ZL,F28,JOPT,ASSCLR,ASSPLS,APRIME,CBOT,COSTFT,	DS014780
3 DEFMIN,ALDEF,NRAV,NWHEEL,DISTF,FPCMAX,FPCMIN,ELASC,ULTMRQ,CTOP,	DS014790
4 W,WB,FO,DCR,NR,HDPT,NW,ALPHA	DS014800
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CBR1,CBR2,CTS1,CTS2,	DS014810
1 CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP,GSP	DS014820
EQUIVALENCE (AREA,ACONC),(YB,DBCT),(YT,DTOP),(YBK,DBOTK),	DS014830
*(YTK,DTOPK),(AREAK,ACONCK),(A,ADIM),(B,BDIM),(C,CDIM),(D,DDIM),	DS014840
* (E,EDIM),(F,FDIM),(G,GDIM),(T,TDIM),(WD,WDDIM),(WH,WHDIM),(H,HDIM)	DS014850
CTOP=DCR	DS014860
C1 = (A=(WH+2.*WD))/2.	DS014870
C2 = (B=(WH +2.*WD))/2.	DS014880
A1 = WD*D	DS014890
A2 = C1*H	DS014900
A3 = C1*G/2.	DS014910
A4 = E*C2/2.	DS014920
A5 = C*C2	DS014930
A6 = WH*T	DS014940
A7 = WH*F	DS014950
A8 = (C2-C1)*H	DS014960
A9 = (C2-C1)*G	DS014970
A10 = A3	DS014980
A11 = C2*(D-H-G-E-C)	DS014990
A12 = A4	DS015000
IF(A.LT.B) GO TO 80	DS015010
A8=0.	DS015020
A9=0.	DS015030
A10=0.	DS015040
A11=0.	DS015050
A12=0.	DS015060
80 CONTINUE	DS015070
A14=(XDIM**2)/2.	DS015080
A15=(YDIM**2)/2.	DS015090
AREA = A1*2. +A2*2. + A3*2. + A4*2. + A5*2. + A6 + A7	DS015100
++2.*(RATNOD-1.)*APRIME=.5625	DS015110
AREA1= A1*2. +A2*2. + A3*2. + A4*2. + A5*2. + A6 + A7	DS015120
++(RATNOD-1.)*APRIME=.5625	DS015130

AREAK= AREA1+ 2.0*A8 + 2.0*A9+2.0*A10+2.0*A11+2.0*A12	DS015140
Y1 = D/2.	DS015150
Y2 = D=H/2.	DS015160
Y3 = D=(H+G/3.)	DS015170
Y4 = C + E/3.	DS015180
Y5 = C/2.	DS015190
Y6 = D=T/2.	DS015200
Y7 = F/2.	DS015210
Y8 = D=(H/2.0)	DS015220
Y9 = D=(H+ G/2.0)	DS015230
Y10= D=(H+ 2.0*G/3.0)	DS015240
Y11 = (C=H=G+E+C)/2.0	DS015250
Y12 = C + 2.0*E/3.0	DS015260
Y14=D=T=XDIM/3.	DS015270
Y15=F+YDIM/3.	DS015280
YB = (Y1*A1*2. + Y2*A2*2. + Y3*A3*2. + Y4*A4*2. + Y5*A5*2. + Y6*A6	DS015290
E + Y7*A7 + (D=CTOP)*2.*(RATNOD=1.)*APRIME)/AREA	DS015300
YT = D=YB	DS015310
YB1= (Y1*A1*2. + Y2*A2*2. + Y3*A3*2. + Y4*A4*2. + Y5*A5*2. + Y6*A6	DS015320
E + Y7*A7 + (D=CTOP)*(RATNOD=1.)*APRIME)/AREA	DS015330
YBK=(YB1*AREA1+Y8*A8*2.+Y9*A9*2.+Y10*A10*2. +Y11*A11*2. +Y12*A12*	DS015340
22.0)/AREAK	DS015350
YTK=D=YBK	DS015360
DO 10 J1=1,2	DS015370
JVKEY=J1=1	DS015380
DY=YB	DS015390
IF(JVKEY.EQ.1) DY = YBK	DS015400
I1 = WD*(D**3)/12. + A1*((Y1=DY)**2)	DS015410
I2 = C1*(H**3)/12. + A2*((Y2=DY)**2)	DS015420
I3 = C1*(G**3)/36. + A3*((Y3=DY)**2)	DS015430
I4 = C2*(E**3)/36. + A4*((Y4=DY)**2)	DS015440
I5 = C2*(C**3)/12. + A5*((Y5=DY)**2)	DS015450
I6 = WH*(T**3)/12. + A6*((Y6=DY)**2)	DS015460
I7 = WH*(F**3)/12. + A7*((Y7=DY)**2)	DS015470
I13=2.*(RATNOD=1.)*APRIME*((D=CTOP=DY)**2)	DS015480
I131= (RATNOD=1.)*APRIME*((D=CTOP=DY)**2)	DS015490
I14=(XDIM**4)/36.+A14*((Y14=DY)**2)	DS015500
I15=(YDIM**4)/36.+A15*((Y15=DY)**2)	DS015510
XINERT = I1*2. + I2*2. + I3*2. + I4*2. + I5*2. + I6 + I7 + I13+I14	DS015520
**2.+I15*2.	DS015530

IF(JVKEY.EQ.0) GO TO 5	DS015540
IF(A.GE.B) XINERK=XINERT+I131-I13	DS015550
IF(A.GE.B) GO TO 5	DS015560
I8 = (C2-C1)*(H**3)/12. + A8*((Y8-DY)**2)	DS015570
I9 = (C2-C1)*(G**3)/12. + A9*((Y9-DY)**2)	DS015580
I10 = C1*(G**3)/36. + A10*((Y10-DY)**2)	DS015590
I11 = C2*((D-H-G-E-C)**3)/12. + A11*((Y11-DY)**2)	DS015600
I12 = C2*(E**3)/36. + A12*((Y12-DY)**2)	DS015610
IF(JVKEY.EQ.1) XINERK =	DS015620
6XINERT+I131-I13 + I8*2.0 + I9*2.0 + I10*2.0 + I11*2.0 + I12*2.0	DS015630
5 CONTINUE	DS015640
IF(JVKEY.EQ.1) GO TO 8	DS015650
ZT=XINERT/YT	DS015660
ZB=XINERT/YB	DS015670
BINERT=XINERT	DS015680
BINERT=BINERT-0.75**4/18.-2.*(.28125*(YB=.50)**2)	DS015690
GO TO 10	DS015700
8 ZTK=XINERK/YTK	DS015710
ZBK=XINERK/YBK	DS015720
BINERK=XINERK	DS015730
BINERK=BINERK-0.75**4/18.-2.*(.28125*(YBK=0.50)**2)	DS015740
10 CONTINUE	DS015750
RETURN	DS015760
END	DS015770


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SUBROUTINE MOMSHR(DL,NWHL,NWHEEL,XSEC,PAXLE,MAXMOM,MAXSHR)
REAL*4 MAXMOM, MAXSHR,NWHL,MOMENT
COMMON/DUMP/ MOMENT(12),SHEAR(12),IPL(20),IPR(20),REACT(20)
DIMENSION NWHL(18),PAXLE(18)
NST=NWHEEL-1
DO 11 II = 1,2
IF(II.EQ.2) XSEC = DL - XSEC
XSECR= DL-XSEC
DO 3 NS = 1,NST
IL = NS
CALL LOCATE(DL,XSEC,NST,NS,NWHL)
N1 = IPL(IL)
N2 = IPR(IL)
IF(N1.EQ.0.AND.N2.EQ.0) PROD = PAXLE(IL+1)*XSECR
IF(N1.EQ.0.AND.N2.EQ.0) GO TO 33
IF(N1.EQ.0) N1 = IL+1
IF(N2.EQ.0) N2 = IL+1
C OBTAIN THE LEFT REACTION FOR ANY SHIFT
PROD = 0.
DO 4 I = N1,N2
IF(I .EQ.1) D2 = DL-(XSEC-NWHL(IL))
IF(I .EQ.1) GO TO 36
IF(I .EQ.(IL+1).AND.IPL(IL).EQ.0) D2 = XSECR
IF(I .EQ.(IL+1).AND.IPL(IL).EQ.0) GO TO 36
IF(I.LE.IL) D2 = DL-(XSEC-(NWHL(IL)-NWHL(I-1)))
IF(I.LE.IL) GO TO 36
IF(I.GT.IL) D2 = XSECR-(NWHL(I-1)-NWHL(IL))
36 CONTINUE
DELT = PAXLE(I)*D2
4 PROD = PROD+DELT
33 CONTINUE
REACT(IL) = PROD/DL
SUMV = 0.
SUMM = 0.
IF(IPL(IL).EQ.0) SHEAR(IL) = REACT(IL)
IF(IPL(IL).EQ.0) MOMENT(IL) = REACT(IL) *XSEC
IF(IPL(IL).EQ.0) GO TO 3
DO 5 I = N1,IL
IF(I .EQ.1) DM = NWHL(IL)
IF(I .EQ.1) GO TO 34

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DS015780
DS015790
DS015800
DS015810
DS015820
DS015830
DS015840
DS015850
DS015860
DS015870
DS015880
DS015890
DS015900
DS015910
DS015920
DS015930
DS015940
DS015950
DS015960
DS015970
DS015980
DS015990
DS016000
DS016010
DS016020
DS016030
DS016040
DS016050
DS016060
DS016070
DS016080
DS016090
DS016100
DS016110
DS016120
DS016130
DS016140
DS016150
DS016160
DS016170

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      DM = NWHL(IL)-NWHL(I-1)
34  DELTM = PAXLE(I)*DM
      DELTV = PAXLE(I)
      SUMM = SUMM+DELTM
      SUMV = SUMV+DELTV
5   CONTINUE
      SHEAR(IL) = REACT(IL)-SUMV
      MOMENT(IL) = REACT(IL)*XSEC-SUMM
3   CONTINUE
      NA = 0
      IF(II.EQ.1) MAXMOM = MOMENT(1)
      IF(II.EQ.1) MAXSHR = SHEAR(1)
      NSTA = NST - 1
      IF(NSTA.EQ.0) GO TO 16
      DO 13 LL = 1,NSTA
      NA = NA + 1
      NB = LL + 1
      AAA = MOMENT(NA)
      BBB = SHEAR(NA)
      IF(II.EQ.2) AAA = MAXMOM
      IF(II.EQ.2) BBB = MAXSHR
      IF(MOMENT(NB).GT.AAA) MAXMOM = MOMENT(NB)
      IF(ABS(SHEAR(NB)).GT.BBB) MAXSHR = ABS(SHEAR(NB))
      IF(MOMENT(NB).GT.AAA) GO TO 15
      NA = NA - 1
      GO TO 13
15  NA = NB - 1
13  CONTINUE
16  CONTINUE
11  CONTINUE
      XSEC=DL-XSEC
      RETURN
      END

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DS016180
DS016190
DS016200
DS016210
DS016220
DS016230
DS016240
DS016250
DS016260
DS016270
DS016280
DS016290
DS016300
DS016310
DS016320
DS016330
DS016340
DS016350
DS016360
DS016370
DS016380
DS016390
DS016400
DS016410
DS016420
DS016430
DS016440
DS016450
DS016460
DS016470
DS016480
DS016490
DS016500

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SUBROUTINE LOCATE(DL,XSEC,NST,NS,NWHL)	DS016510
REAL*4 NWHL,MOMENT	DS016520
COMMON/DUMP/ MOMENT(12),SHEAR(12),IPL(20),IPR(20),REACT(20)	DS016530
DIMENSION NWHL(18)	DS016540
XSECR = DL-XSEC	DS016550
DTERM = 0.	DS016560
DO 1 I = 1,NST	DS016570
DLE = NWHL(NS)-DTERM	DS016580
IF(DLE.LE.XSEC) IPL(NS) = I	DS016590
IF(DLE.LE.XSEC) GO TO 2	DS016600
IF(I.EQ.NS) IPL(NS) = 0	DS016610
IF(I.EQ.NS) GO TO 2	DS016620
DTERM = NWHL(I)	DS016630
1 CONTINUE	DS016640
2 CONTINUE	DS016650
DO 4 IC= 1,NST	DS016660
NSC = NS+IC	DS016670
IF((NS+1).EQ.(NST+1)) IPR(NS) = 0	DS016680
IF(NSC.GT.NST) GO TO 5	DS016690
DELTR = NWHL(NS+IC)-NWHL(NS)	DS016700
IF(DELTR.GT.XSECR.AND.IC.EQ.1) IPR(NS) = 0	DS016710
IF(DELTR.GT.XSECR) GO TO 5	DS016720
IPR(NS) = NS+IC+1	DS016730
4 CONTINUE	DS016740
5 CONTINUE	DS016750
RETURN	DS016760
END	DS016770

	SUBROUTINE LPCCDE (NFFCE,NEGS,INDX,KODE)	DS016780
C	LINEAR PROGRAMMING ALGORITHM	DS016790
C		DS016800
	COMMON/D314/N,M,CBJ,KP1,K,NR,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,	DS016810
	1SUM,IGNOR,IT(4),JCONT,X(150),A(118,150),B(150),XD(150)	DS016820
C		DS016830
C	SET UP MATRIX	DS016840
C		DS016850
	KBOMB=0	DS016860
	KP1=N+M	DS016870
	K=N+M-1	DS016880
	KK=K-1	DS016890
	IF(KODE.NE.0) GO TO 200	DS016900
	DO 1 I=1,N	DS016910
	DO 1 J=M,KK	DS016920
	1 A(I,J+1)=0.0	DS016930
	DO 2 I=2,N	DS016940
	IPM=I+M-1	DS016950
	2 A(I,IPM)=1.0	DS016960
C		DS016970
C	FLAG BASIS	DS016980
C		DS016990
	200 DO 5 I=1,K	DS017000
	XD(I)=0.0	DS017010
	5 X(I)=0.0	DS017020
	DO 6 I=1,N	DS017030
	IPM=I+M	DS017040
	6 X(IPM)=1.0	DS017050
	DO 7 I=1,N	DS017060
	7 B(I)=0.0	DS017070
	10 CONTINUE	DS017080
	C*****	DS017090
	C**** FEASIBILITY SECTION	DS017100
	C*****	DS017110
	INEG=2	DS017120
	11 DO 14 I=2,N	DS017130
	IF (B(I)) 12,12,14	DS017140
	12 CONTINUE	DS017150
	IF (A(I,KP1)-A(INEG,KP1)) 13,14,14	DS017160
	13 INEG=I	DS017170

14	CONTINUE	DS017180
	IF (A(INEG,KP1)) 15,23,23	DS017190
15	IF (B(INEG)) 16,16,23	DS017200
16	JSM=1	DS017210
	DO 19 J=2,K	DS017220
	IF (XD(J)) 17,17,19	DS017230
17	CONTINUE	DS017240
	IF (A(INEG,J)-A(INEG,JSM)) 18,19,19	DS017250
18	JSM=J	DS017260
19	CONTINUE	DS017270
	IF (XD(JSM)) 20,20,23	DS017280
20	IF (A(INEG,JSM)) 22,21,21	DS017290
C		DS017300
C	NO FEASIBLE SOLUTION	DS017310
C		DS017320
21	KBOMB=51	DS017330
	GO TO 38	DS017340
22	CALL PIVOT (INEG,JSM)	DS017350
	GO TO 10	DS017360
C	*****	DS017370
C	OPTIMALITY SECTION	DS017380
C	*****	DS017390
23	JBGST=1	DS017400
C		DS017410
C	SELECT INCOMING VECTOR	DS017420
C		DS017430
	DO 26 J=1,K	DS017440
	IF (XD(J)) 24,24,26	DS017450
24	CONTINUE	DS017460
	IF (A(1,J)-A(1,JBGST)) 26,26,25	DS017470
25	JBGST=J	DS017480
26	CONTINUE	DS017490
	IF (A(1,JBGST)) 38,38,27	DS017500
C		DS017510
C	CHECK FOR UNBOUNDED SOLUTION	DS017520
C		DS017530
27	DO 29 I=2,N	DS017540
	ISPOT=I	DS017550
	IF (B(I)) 28,28,29	DS017560
28	CONTINUE	DS017570

IF (A(I,JBGST)) 29,29,30	DS017580
29 CONTINUE	DS017590
30 CONTINUE	DS017600
IF (A(ISPOT,JBGST)) 31,31,32	DS017610
31 KBOMB=50	DS017620
GO TO 38	DS017630
C	DS017640
C SELECT OUTGOING VECTOR	DS017650
C	DS017660
32 KK=ISPOT	DS017670
DO 36 I=KK,N	DS017680
IF (B(I)) 33,33,36	DS017690
33 CONTINUE	DS017700
IF(A(I,JBGST)) 36,36,34	DS017710
34 IF (A(I,K+1)/A(I,JBGST)=A(ISPOT,K+1)/A(ISPOT,JBGST)) 35,36,36	DS017720
35 ISPOT=I	DS017730
36 CONTINUE	DS017740
IF (B(ISPOT)) 37,37,31	DS017750
37 CALL PIVOT (ISPOT,JBGST)	DS017760
GO TO 23	DS017770
C*****	DS017780
C**** OUTPUT SECTION	DS017790
C*****	DS017800
38 OBJ=-A(1,KF1)	DS017810
IF (INDX=1) 40,39,40	DS017820
39 OBJ=-OBJ	DS017830
40 DO 45 I=1,K	DS017840
IF (X(I)) 44,44,41	DS017850
41 DO 42 J=2,N	DS017860
IF (A(J,I)) 42,42,43	DS017870
42 CONTINUE	DS017880
43 X(I)=A(J,KF1)	DS017890
GO TO 45	DS017900
44 X(I)=0.	DS017910
45 CONTINUE	DS017920
DO 49 J=1,K	DS017930
IF (J=N+1) 46,46,47	DS017940
46 JJ=J+M	DS017950
GO TO 48	DS017960
47 JJ=J=N+1	DS017970

48 XD(J)=-A(1,JJ)
49 CONTINUE
IF(KBOMB.EQ.50) WRITE(6,50)
50 FORMAT (/10HOUNBCUNDED)
RETURN
END

DS017980
DS017990
DS018000
DS018010
DS018050
DS018070

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SUBROUTINE PIVOT (I,J)
COMMON/D314/N,M,OBJ,KP1,K,NR,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,
1SUM,IGNOR,IT(4),JCONT,X(150),A(118,150),B(150),XD(150)
DO 2 JJ=1,K
IF (X(JJ)) 2,2,1
1 IF (A(I,JJ)) 2,2,3
2 CONTINUE
3 X(JJ)=0.0
NM1=N-1
R=A(I,J)
DO 4 L=1,KP1
4 A(I,L)=A(I,L)/R
DO 5 L=2,I
F=A(L-1,J)
DO 5 M=1,KP1
5 A(L-1,M)=A(L-1,M)-A(I,M)*F
IF (I=N) 6,8,8
6 DO 7 L=I,NM1
F=A(L+1,J)
DO 7 M=1,KP1
7 A(L+1,M)=A(L+1,M)-A(I,M)*F
8 CONTINUE
X(J)=1.0
M=KP1-N
RETURN
END

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DS018080
DS018090
DS018100
DS018110
DS018120
DS018130
DS018140
DS018150
DS018160
DS018170
DS018180
DS018190
DS018200
DS018210
DS018220
DS018230
DS018240
DS018250
DS018260
DS018270
DS018280
DS018290
DS018300
DS018310
DS018320
DS018330

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SUBROUTINE CAMBER(ES,EC,ASTRN,STRNS,UWB,AREA,SPANL,ECCL,IB,FO,ENDEDS018340
1 CC,PRLMAX,CBRMAX,HDPT) DS018350
COMMON/DEFINE/ UWC,HUM,AS,FPS,CTR1,CTR2,CER1,CER2,CTS1,CTS2, DS018360
1 CBS1,CBS2,CREEP1,CREEP2,SHRK1,SHRK2,RATNOD,FPL,FSY,ASTIRP,GSP DS018370
DIMENSION CNST(4,5),PRLMAX(5),CBRMAX(5) DS018380
DATA CNST/315.,20.,440.,60.,525.,10.,675.,40.,380.,25.,400.,50.,29 DS018390
*0.,25.,460.,50.,0.,0.,0.,0./ DS018400
REAL IB DS018410
DS018420
DS018430
DS018440
CAMBER AND STRESS LOSS CALCULATIONS DS018450
MIDSPAN CAMBER AND STRESS LOSS DUE TO INITIAL PRESTRESS AND BEAM DS018460
DS018470
IF(CREEP1.EQ.0.) J1=4 DS018480
IF(CREEP1.EQ.0.) GO TO 2 DS018490
CNST(1,5)=SHRK1 DS018500
CNST(2,5)=SHRK2 DS018510
CNST(3,5)=CREEP1 DS018520
CNST(4,5)=CREEP2 DS018530
J1=5 DS018540
2 DO 1 N=1,J1 DS018550
ASH=0.000001*CNST(1,N) DS018560
BSH=CNST(2,N) DS018570
ACRR=0.000001*CNST(3,N) DS018580
BCR=CNST(4,N) DS018590
ACR = ACRR*0.001 DS018600
RN = ES/EC DS018610
AST = ASTRN*STRNS DS018620
W = UWB*AREA/144. DS018630
DLM = (W*SPANL*SPANL/8.)*12. DS018640
TEMP = 1.+(RN*AST/AREA)+(RN*AST*ECCL*ECCL/IB) DS018650
FR = FO/TEMP +(DLM*ECCL*RN*AST/(IB*TEMP)) DS018660
PLI = ((FO=FR)/FO)*100. DS018670
CONST = (1./AREA)+(ECCL*ECCL/IB) DS018680
FCSO = FR*CONST-(DLM*ECCL/IB) DS018690
STRN1 = ACR*FCSO+ASH DS018700
STRN2 = STRN1-STRN1*(RN*AST*CONST) DS018710
DFCS = STRN2*ES*AST*CONST * 10.0 ** 6 DS018720
STRN4 = ACR*(FCSO-DFCS/2.)+ASH DS018730
STRN5 = STRN4-STRN4*RN*AST*CONST

```

C
C
C
C
C

DFCS1 = STRN5*ES*AST*CONST * 10.0 ** 6	DS018740
STRN6 = ACR*(FCS0-DFCS1/2.)+ASH	DS018750
STRN7 = STRN6-STRN6*RN*AST*CONST	DS018760
PLINF = (STRN7*ES*AST*10.0**6/FC)*100.	DS018770
PLMAX = PLINF+PLI	DS018780
PRLMAX(N)=PLMAX	DS018790
CCONST = 1./(EC*IB*10.**6)	DS018800
HSPAN = SPANL/2.	DS018810
CI1 = CCONST*(FR*ENDECC *HSPAN*0.5*HSPAN*144.)	DS018820
CI2 = CCONST*(FR*(ECCL-ENDECC)*(HSPAN=HDPT)*0.5*0.67*(HSPAN	DS018830
1=HDPT)*144.0)	DS018840
CI3 = CCONST*(FR*(ECCL-ENDECC)*HDPT*(HSPAN=HDPT/2.)*144.)	DS018850
CI4 = CCONST*((5./384.)*(W*SPANL*SPANL*SPANL*SPANL*12.*12.*12.))	DS018860
CI = CI1 +CI2 +CI3 -CI4	DS018870
STRAIN=FCS0/(EC*10.**6)	DS018880
CMAX = CI*((ACR*(FCS0-(DFCS/2.))+STRAIN)/STRAIN)*(1.-(PLINF/100.))	DS018890
CBRMAX(N)=CMAX	DS018900
1 CONTINUE	DS018910
RETURN	DS018920
END	DS018930

```
SUBROUTINE SHEAR(B,DEPTH,D,FPC,FSY,AREA,VU,SPACE)
AV=2.*AREA
S1=(AV*FSY)/(0.100*B)
SMAX=0.75*DEPTH
IF(S1.LT.SMAX) SMAX=S1
RJ=0.90
VCMAX=0.180*B*RJ*D
VC=0.06*FPC*B*RJ*D
IF(VC.GT.VCMAX) VC=VCMAX
SPACE=(2.*AV*FSY*RJ*D)/(ABS(VU)-VC)
IF(SPACE.LT.0.0.OR.SPACE.GT.SMAX) SPACE=SMAX
RETURN
END
```

```
DS018940
DS018950
DS018960
DS018970
DS018980
DS018990
DS019000
DS019010
DS019020
DS019030
DS019040
DS019050
DS019060
```

	SUBROUTINE INTPRG(KBOMB)	DS019070
C	HEURISTIC MIXED INTEGER LINEAR PROGRAMMING	DS019080
C	MAX WITH LE CONSTRAINTS	DS019090
C	PACKED DATA	DS019100
	INTEGER*2 ROW,COL	DS019110
	COMMON/D314/N,M,OBJ,KP1,K,NR,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,	DS019120
	1SUM,IGNOR,IT(4),JCONT,X(150),Y(3136),ROW(3136),COL(150),BB(150),	DS019130
	2C(150),B(150),XX(150)	DS019140
C	TR = ROW TOLERANCE	DS019150
	TR=0.1	DS019160
C	TV = VARIABLE TOLERANCE	DS019170
	TV=1.E-2	DS019180
C	TITLE = TITLE INFORMATION	DS019190
C	NR = NUMBER OF CONSTRAINTS. N1 = NUMBER OF CONTINUOUS VARIABLES	DS019200
C	N2 = NUMBER OF INTEGER VARIABLES, N3 = NUMBER OF BINARY VARIABLES	DS019210
C	IS = 0#INITIAL SOLUTION TO BE PROVIDED	DS019220
C	IS = 1 OR 2#NO INITIAL SOLUTION TO BE PROVIDED	DS019230
C	IS = 0 OR 1#PHASE 4 USED IF NO FEASIBLE SOLUTION IS	DS019240
C	LOCATED. DXMAX = MAXIMUM INCREMENT OF ANY X%J<	DS019250
C	INPUT ORDER	DS019260
C	CONT VAR, INTEGER, 0=1	DS019270
C	OBJ FUNC IS ROW NR + 1	DS019280
C	RHS IS IN NCA=N1 + N2 + N3 + 1	DS019290
C	BLANK CARD ENDS DATA	DS019300
	NRA=NR+1	DS019310
	NTWO=N2+N1	DS019320
	NC=NTWO+N3	DS019330
	NCA=NC+1	DS019340
C	CONTINUOUS VARIABLES MUST BE ENTERED FIRST	DS019350
	MCOL=0	DS019360
	KKK=0	DS019370
C	COLUMN ENTRIES ALL TOGETHER	DS019380
	IC=0	DS019390
	IGNOR=0	DS019400
	JJ=0	DS019410
	OPT=-1.E30	DS019420
	DO 3 I=1,4	DS019430
3	IT(I)=0	DS019440
	DO 4 J=1,NC	DS019450
4	XX(J) = 0.	DS019460

C IF(IS.GT.0) GO TO 10
 INITIAL SOLUTION (REQUIRED IF IS = 0)
 DO 7 J=1,NC
 IF(X(J).EQ.0.) GO TO 7
 XX(J)=X(J)
 C(NCA)=C(NCA)+X(J)*C(J)
 IA=COL(J)
 IB=COL(J+1)-1
 DO 6 I=IA,IB
 BB(ROW(I))=BB(ROW(I))-X(J)*Y(I)
 6 CONTINUE
 7 CONTINUE
 10 SUM=0.
 DO 15 I=1,NR
 IF(ABS(BB(I)).LT.TR) BB(I)=0.
 B(I)=BB(I)
 15 IF(B(I).LT.0.) SUM=SUM-B(I)
 SMIN=SUM
 TOPT=C(NCA)
 20 IF(SUM.EQ.0.) GO TO 24
 22 CALL PHASE1
 IF(SUM.GT.0.) IF(IC) 40,33,40
 24 CALL PHASE2
 IF(C(NCA).LE.OPT) IF(IC) 46,36,46
 DO 30 J=1,NC
 30 XX(J)=X(J)
 DO 32 I=1,NR
 32 B(I)=BB(I)
 IF(OPT.EQ.(-1.E30)) JJ=0
 OPT=C(NCA)
 GO TO 36
 33 IF(SUM.GE.SMIN) GO TO 36
 DO 34 J=1,NC
 34 XX(J)=X(J)
 DO 35 I=1,NR
 35 B(I)=BB(I)
 SMIN=SUM
 TOPT=C(NCA)
 36 CALL PHASE3
 IF(IGNOR.EQ.0) GO TO 40

DS019470
 DS019480
 DS019490
 DS019500
 DS019510
 DS019520
 DS019530
 DS019540
 DS019550
 DS019560
 DS019570
 DS019580
 DS019590
 DS019600
 DS019610
 DS019620
 DS019630
 DS019640
 DS019650
 DS019660
 DS019670
 DS019680
 DS019690
 DS019700
 DS019710
 DS019720
 DS019730
 DS019740
 DS019750
 DS019760
 DS019770
 DS019780
 DS019790
 DS019800
 DS019810
 DS019820
 DS019830
 DS019840
 DS019850
 DS019860

```

      IC=1
      GO TO 22
40  IF(OPT.GT.(-1.E30)) GO TO 46
      IF(SUM.GE.SMIN) GO TO 48
      DO 42 J=1,NC
42  XX(J)=X(J)
      DO 44 I=1,NR
44  B(I)=BB(I)
      SMIN=SUM
      TOPT=C(NCA)
      GO TO 54
46  IF(IS.LT.2) GO TO 56
      C(NCA)=OPT
      SUM=0.
      GO TO 50
48  C(NCA)=TOPT
      SUM=SMIN
50  DO 52 J=1,NC
52  X(J)=XX(J)
      DO 53 I=1,NR
53  BB(I)=B(I)
C   PHASE 4 = PERTURE THE CURRENT SOLUTION
54  IF(JJ.EQ.NC) GO TO 56
      JJ=JJ+1
      IF(X(JJ).LT.TV) GO TO 54
      XS=X(JJ)
      X(JJ)=AMAX1(0.,XS-3.*DXMAX)
      DX=X(JJ)-XS
      IA=COL(JJ)
      IB=COL(JJ+1)-1
      DO 55 I=IA,IB
      BB(ROW(I))=BB(ROW(I))-DX*Y(I)
55  IF(ABS(BB(ROW(I))).LT.TR) BB(ROW(I))=0.
      SUM=0.
      DO 155 I=1,NR
155 IF(BB(I).LT.0.) SUM=SUM-BB(I)
      C(NCA)=C(NCA)+DX*C(JJ)
      IC=0
      IGNOR=JJ
      IT(4)=IT(4)+1

```

```

DS019870
DS019880
DS019890
DS019900
DS019910
DS019920
DS019930
DS019940
DS019950
DS019960
DS019970
DS019980
DS019990
DS020000
DS020010
DS020020
DS020030
DS020040
DS020050
DS020060
DS020070
DS020080
DS020090
DS020100
DS020110
DS020120
DS020130
DS020140
DS020150
DS020160
DS020170
DS020180
DS020190
DS020200
DS020210
DS020220
DS020230
DS020240
DS020250
DS020260

```

```

      GO TO 20
56  IF(OPT.EQ.(-1.E30)) GO TO 60
C   OUTPUT FINAL FEASIBLE SOLUTION
      KBOMB=0
      RETURN
C   OUTPUT FINAL INFEASIBLE SOLUTION
60  DO 57 I=1,NC
57  XX(I)=X(I)
      DO 58 I=1,NR
58  B(I)=BB(I)
      KBOMB=1
      END

```

```

DS020270
DS020280
DS020290
DS020291
DS020300
DS020310
DS020311
DS020312
DS020313
DS020314
DS020315
DS020360

```

	SUBROUTINE PHASE1	DS020370
C	ATTEMPT TO REDUCE THE SUM OF INFEASIBILITIES	DS020380
	INTEGER*2 ROW,COL	DS020390
	COMMON/D314/N,M,OBJ,KP1,K,NR,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,	DS020400
	1SUM,IGNOR,IT(4),JCONT,X(150),Y(3136),ROW(3136),COL(150),BB(150),	DS020410
	2C(150),B(150),XX(150)	DS020420
	NTWO=N2+N1	DS020430
10	Q=-1.E38	DS020440
	P=-1.E38	DS020450
	DO 140 J=1,NC	DS020460
	IF(J.NE.IGNOR) GO TO 15	DS020470
	IGNOR=0	DS020480
	GO TO 140	DS020490
15	R=-1.E38	DS020500
	IF(X(J).EQ.0.) GO TO 18	DS020510
	S=1.E38	DS020520
	IF (J.GT.NTWO) R = 0.	DS020530
	GO TO 20	DS020540
18	S = 0.0	DS020550
20	IA=COL(J)	DS020560
	IB=COL(J+1)-1	DS020570
	DO 60 I=IA,IB	DS020580
	IF(BB(ROW(I)).GE.0.) GO TO 60	DS020590
	V=BB(ROW(I))/Y(I)	DS020600
	IF(ABS(V).LT.TV) GO TO 60	DS020610
	IF(V.LT.0.) GO TO 30	DS020620
	IF (R.EQ.0.0) GO TO 60	DS020630
	IF(J.GT.N1) V=AIN(T(V+.999)	DS020640
	IF (J.GT.NTWO) V = 1.0	DS020650
	IF(V.GT.R) R=V	DS020660
	GO TO 60	DS020670
30	IF(S.EQ.0.) GO TO 60	DS020680
	IF(J.GT.N1) V=AIN(T(V-.999)	DS020690
	IF(V.GE.S) GO TO 60	DS020700
	IF(V.LT.(-X(J))) V=-X(J)	DS020710
	S=V	DS020720
60	CONTINUE	DS020730
	IF(R.EQ. -1.E38 .OR.R.EQ.0.0) GO TO 90	DS020740
	T=0.	DS020750
	K=IA	DS020760

DO 70 I=1,NR
 IF(I.EQ.ROW(K)) GO TO 64
 IF(BB(I).GE.0.) GO TO 70
 T=T-BB(I)
 GO TO 66
 64 F=BB(I)-R*Y(K)
 IF(ABS(F).LT.TR) F=0.
 K=K+1
 IF(K.GT.IB) K=IB
 IF(F.GE.0.) GO TO 70
 T=T-F
 66 IF(T.GE.SUM) GO TO 90
 70 CONTINUE
 IF(T.EQ.0.) GO TO 110
 W=R*C(J)
 IF(W.LE.Q) GO TO 90
 Q=W
 80 DX=AMIN1(R,DXMAX)
 KK=J
 90 IF(S.EQ.1.E38.OR.S.EQ.0.) GO TO 140
 T=0.
 K=IA
 DO 100 I=1,NR
 IF(I.EQ.ROW(K)) GO TO 94
 IF(BB(I).GE.0.) GO TO 100
 T=T-BB(I)
 GO TO 96
 94 F=BB(I)-S*Y(K)
 IF(ABS(F).LT.TR) F=0.
 K=K+1
 IF(K.GT.IB) K=IB
 IF(F.GE.0.) GO TO 100
 T=T-F
 96 IF(T.GE.SUM) GO TO 140
 100 CONTINUE
 IF(T.EQ.0.) GO TO 120
 W=S*C(J)
 IF(W.LE.Q) GO TO 140
 Q=W
 105 DX=AMAX1(S,-DXMAX)

DS020770
 DS020780
 DS020790
 DS020800
 DS020810
 DS020820
 DS020830
 DS020840
 DS020850
 DS020860
 DS020870
 DS020880
 DS020890
 DS020900
 DS020910
 DS020920
 DS020930
 DS020940
 DS020950
 DS020960
 DS020970
 DS020980
 DS020990
 DS021000
 DS021010
 DS021020
 DS021030
 DS021040
 DS021050
 DS021060
 DS021070
 DS021080
 DS021090
 DS021100
 DS021110
 DS021120
 DS021130
 DS021140
 DS021150
 DS021160

```

      KK=J
      GO TO 140
110  TEMP=R*C(J)
      IF(TEMP.LE.P) GO TO 90
      P=TEMP
      Q=1.E38
      GO TO 80
120  TEMP=S*C(J)
      IF(TEMP.LE.P) GO TO 140
      P=TEMP
      Q=1.E38
      GO TO 105
140  CONTINUE
      IF(Q.EQ.(-1.E38)) RETURN
      X(KK)=X(KK)+DX
      IA=COL(KK)
      IB=COL(KK+1)-1
      DO 150 I=IA,IB
      BB(ROW(I))=BB(ROW(I))-DX*Y(I)
150  IF(ABS(BB(ROW(I))).LT.TR) BB(ROW(I))=0.
      SUM=0.
      DO 155 I=1,NR
155  IF(BB(I).LT.0.) SUM=SUM-BB(I)
      C(NCA)=C(NCA)+DX*C(KK)
      IT(1)=IT(1)+1
      IF(SUM.EQ.0.) RETURN
      GO TO 10
      END

```

```

DS021170
DS021180
DS021190
DS021200
DS021210
DS021220
DS021230
DS021240
DS021250
DS021260
DS021270
DS021280
DS021290
DS021300
DS021310
DS021320
DS021330
DS021340
DS021350
DS021360
DS021370
DS021380
DS021390
DS021400
DS021410
DS021420
DS021430
DS021440

```

SUBROUTINE PHASE2
 C ATTEMPT TO IMPROVE THE VALUE OF THE OBJECTIVE FUNCTION
 INTEGER*2 ROW,COL
 COMMON/D314/N,M,OBJ,KP1,K,NR,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,
 1SUM,IGNOR,IT(4),JCONT,X(150),Y(3136),ROW(3136),COL(150),BB(150),
 2C(150),B(150),XX(150)
 NTWO=N2+N1
 10 ZZ=-1.
 DO 30 J=1,NC
 IF(J.NE.IGNCR) GO TO 12
 IGNCR=0
 GO TO 30
 12 IF(C(J).GE.0.) GO TO 18
 IF(X(J).EQ.0.) GO TO 30
 R=-1.E38
 IA=COL(J)
 IB=COL(J+1)-1
 DO 15 I=IA,IB
 IF(BB(ROW(I)).EQ.0.) IF(Y(I)) 30,30,13
 V=BB(ROW(I))/Y(I)
 IF(V.GT.0.) GO TO 15
 IF(V.GT.=TV) GO TO 30
 IF(J.GT.N1.AND.V.GT.=1.) GO TO 30
 IF(V.GE.(-X(J))) GO TO 14
 13 V=-X(J)
 14 IF(V.GT.R) R=V
 15 CONTINUE
 GO TO 21
 18 IF (J.GT.NTWO)IF(X(J)) 19,19,30
 19 R = 1.0E38
 IA=COL(J)
 IB=COL(J+1)-1
 DO 20 I=IA,IB
 IF(BB(ROW(I)).EQ.0.) IF(Y(I)) 20,30,30
 V=BB(ROW(I))/Y(I)
 IF(V.LT.0.) GO TO 20
 IF(V.LT.TV) GO TO 30
 IF(J.GT.N1.AND.V.LT.1.) GO TO 30
 IF (J.GT.NTWO) V = 1.0
 IF(V.LT.R) R=V

DS021450
 DS021460
 DS021470
 DS021480
 DS021490
 DS021500
 DS021510
 DS021520
 DS021530
 DS021540
 DS021550
 DS021560
 DS021570
 DS021580
 DS021590
 DS021600
 DS021610
 DS021620
 DS021630
 DS021640
 DS021650
 DS021660
 DS021670
 DS021680
 DS021690
 DS021700
 DS021710
 DS021720
 DS021730
 DS021740
 DS021750
 DS021760
 DS021770
 DS021780
 DS021790
 DS021800
 DS021810
 DS021820
 DS021830
 DS021840

```

20 CONTINUE
21 IF(ABS(R).EQ.1.E38) GO TO 30
   IF(C(J).EQ.0.) R=AMIN1(1.,R)
   Q=R*C(J)
   IF(Q.LE.ZZ) GO TO 30
   ZZ=Q
   KK=J
   IF(J.GT.N1) R=AIN1(R)
   IF(R.LT.0.) GO TO 25
   DX=AMIN1(R,DXMAX)
   GO TO 30
25 DX=AMAX1(R,-DXMAX)
30 CONTINUE
   IF(ZZ.EQ.(-1.)) RETURN
   X(KK)=X(KK)+DX
   IA=COL(KK)
   IB=COL(KK+1)-1
   DO 40 I=IA,IB
     BB(ROW(I))=BB(ROW(I))-DX*Y(I)
     IF(ABS(BB(ROW(I))).LT.TR) BB(ROW(I))=0.
40 CONTINUE
   C(NCA)=C(NCA)+DX*C(KK)
   IT(2)=IT(2)+1
   GO TO 10
END

```

```

DS021850
DS021860
DS021870
DS021880
DS021890
DS021900
DS021910
DS021920
DS021930
DS021940
DS021950
DS021960
DS021970
DS021980
DS021990
DS022000
DS022010
DS022020
DS022030
DS022040
DS022050
DS022060
DS022070
DS022080
DS022090

```

C	SUBROUTINE PHASE3	DS022100
	PERTURB THE CURRENT SOLUTION	DS022110
	INTEGER*2 ROW,COL	DS022120
	COMMON/D314/N,M,OBJ,KF1,K,NR,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,	DS022130
	ISUM,IGNOR,IT(4),JCONT,X(150),Y(3136),ROW(3136),COL(150),BB(150),	DS022140
	2C(150),E(150),XX(150)	DS022150
	NTWO=N2+N1	DS022160
	ZZ=-1.	DS022170
	JJ=0	DS022180
	DO 10 J=1,NC	DS022190
	IF(X(J).LT.TV.AND.C(J).LT.0.) GO TO 10	DS022200
	IF(X(J).GE.1..OR.C(J).GE.0.) GO TO 2	DS022210
	TEMP=C(J)*X(J)	DS022220
	IF(TEMP.LT.ZZ) GO TO 10	DS022230
	DX=-X(J)	DS022240
	GO TO 4	DS022250
	2 TEMP=ABS(C(J))	DS022260
	IF(TEMP.LT.ZZ) GO TO 10	DS022270
	DX=1.	DS022280
	IF(C(J).LT.0.) DX=-1.	DS022290
	IF (J.GT,NTWO) IF (X(J)*DX) 4,4,10	DS022300
	4 ZZ=TEMP	DS022310
	JJ=J	DS022320
	10 CONTINUE	DS022330
	IF(JJ.EQ.0)GO TO 35	DS022340
	X(JJ)=X(JJ)+DX	DS022350
	IA=COL(JJ)	DS022360
	IB=COL(JJ+1)-1	DS022370
	DO 20 I=IA,IB	DS022380
	BB(ROW(I))=BB(ROW(I))-DX*Y(I)	DS022390
	20 IF(ABS(BB(ROW(I))).LT.TR) BB(ROW(I))=0.	DS022400
	SUM=0.	DS022410
	DO 30 I=1,NR	DS022420
	30 IF(BB(I).LT.0.) SUM=SUM+BB(I)	DS022430
	C(NCA)=C(NCA)+ZZ	DS022440
	35 IGNOR=JJ	DS022450
	IT(3)=IT(3)+1	DS022460
	RETURN	DS022470
	END	DS022480

SUBROUTINE SQUASH	DS022490
INTEGER*2 ROW,COL	DS022500
COMMON/D314/N,M,OBJ,KP1,K,NR,NC,NRA,NCA,N1,N2,N3,DXMAX,IS,TV,TR,	DS022510
1SUM,IGNOR,IT(4),JCONT,X(150),Y(3136),ROW(3136),COL(150),BB(150),	DS022520
2C(150),B(150),XX(150)	DS022530
DIMENSION STRCOL(150)	DS022540
JCONT=0	DS022550
J1=0	DS022560
DO 2 J=1,NCA	DS022570
K1=0	DS022580
READ(3) (STRCOL(J2),J2=1,NRA)	DS022590
C(J)=STRCOL(NRA)	DS022600
DO 3 I=1,NR	DS022610
IF(J.EQ.NCA) GO TO 6	DS022620
IF(STRCOL(I).EQ.0.0) GO TO 3	DS022630
6 J1=J1+1	DS022640
K1=K1+1	DS022650
IF(J.EQ.NCA) GO TO 50	DS022660
Y(J1)=STRCOL(I)	DS022670
JCONT=JCONT+1	DS022680
ROW(J1)=I	DS022690
50 CONTINUE	DS022700
IF(K1.GT.1) GO TO 3	DS022710
COL(J)=J1	DS022720
3 CONTINUE	DS022730
2 CONTINUE	DS022740
DO 10 J3=1,NR	DS022750
10 BB(J3)=STRCOL(J3)	DS022760
REWIND 3	DS022770
RETURN	DS022780
END	DS022790

```

SUBROUTINE PLOSS(FPCR,ZMBW,ZMC,ZMNC,FSU,AS,AB,ZI,ZIC,YB,YBC,EC,
*HUM,SPAN,ZLOSS,ZINLOS,UWC)

```

THIS SUBROUTINE COMPUTES PRESTRESS LOSS BY 1975 AASHTO
INTERIM SPEC.

DS022800
DS022810
DS022820
DS022830
DS022840
DS022850
DS022860
DS022870
DS022880
DS022890
DS022900
DS022910
DS022920
DS022930
DS022940
DS022950
DS022960
DS022970
DS022980
DS022990
DS023000
DS023010
DS023020
DS023030
DS023040
DS023050
DS023060
DS023070
DS023080
DS023090
DS023100
DS023110
DS023120
DS023130
DS023140
DS023150
DS023160
DS023170
DS023180
DS023190

	ES=(28E+06*FCIR/ECI)	DS023200
C		DS023210
C	CREEP LOSS	DS023220
C		DS023230
	FCDS=12.*ZMNC*(YB-EC)/ZI+12.*ZMC*(YBC-EC)/ZIC	DS023240
	CRC=12.*FCIR=7.*FCDS	DS023250
C		DS023260
C	STRAND RELAXATION LOSS	DS023270
C		DS023280
	CRS=20.-0.4*ES=0.2*(SH+CRC)	DS023290
C		DS023300
C	TOTAL LOSS	DS023310
C		DS023320
	DELTFIS=SH+ES+CRC+CRS	DS023330
	DELFSI=ES+0.5*CRS	DS023340
C		DS023350
C	LOSS FACTOR	DS023360
C		DS023370
	ZLOSS=DELTFIS/(.7*FSU)	DS023380
	ZINLOS=DELFSI/(.7*FSU)	DS023390
	RETURN	DS023400
	END	DS023410


```

COMMON/BLK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),
*ZSA(20),YSA(20),BMJ(20),YH(10,2),ZH(10,2),NTY(30),
*LC( 500),NLB( 500),LM(4),P( 500),XL( 500),PL( 500),
*ECC( 500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),
*NPOS( 500),XPOS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),
*NOUT(9),YSSS(9,4),ZSSS(9,4),X(8)
COMMON/BLK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYPES,NEQ,
*ALPHA,NLCASE,NLCARD,NX
COMMON/BLK 3/ XZERO(164),YMAMPL(20),ZVAMPL(20),
*FAMPL(20),SFAMPL(20),TAMPL(20),DAMPL(20),YMOLT( 200,20,9),
*ZMOUT(9,20,9),FCUT(9,20,9),SFOUT(9,20,9),TGUT(9,20,9),
*DOUT(9,20,9),XCUT(9,76,9),XS(4),XA(4)

```

C
C
C

MULTI-BEAM BRIDGE ANALYSIS,

NHARMS=30

CALL REREAD

1 READ(5,211,END=1900) (HEAD(J1),J1=1,15),KODE
211 FORMAT(15A4,1X,11)

C
C
C

LOAD INPUT DATA

IF(KODE,EQ.0)CALL INPTT

IF(KODE,NE.0)CALL INPUT

C
C
C

ZERO OUTPUT STORAGE AREA

NX = 4 * NBEAMS = 4

DO 220 I = 1, NLCASE

NP = NPOS(I)

DO 219 K= 1, NP

DO 210 J = 1, NBEAMS

YMOUT(I,J,K) = 0.0

IF(KODE,NE.0) GO TO 210

ZMOUT(I,J,K) = 0.0

SFOUT(I,J,K) = 0.0

FOUT(I,J,K) = 0.0

TOUT(I,J,K) = 0.0

DOUT(I,J,K)=0.0

210 CONTINUE

AM000010
AM000020
AM000030
AM000040
AM000050
AM000060
AM000070
AM000080
AM000090
AM000100
AM000110
AM000120
AM000130
AM000140
AM000150
AM000160
AM000170
AM000180
AM000190
AM000200
AM000210
AM000220
AM000230
AM000240
AM000250
AM000260
AM000270
AM000280
AM000290
AM000300
AM000310
AM000320
AM000330
AM000340
AM000350
AM000360
AM000370
AM000380
AM000390
AM000400

```

      IF(KODE.NE.0) GO TO 219
      DO 217 J=1,NX
217  XOUT(I,J,K) = 0.
219  CONTINUE
220  CONTINUE

```

```

C
C   FORM STRUCTURE FLEXIBILITY MATRIX FOR EACH HARMONIC
C
      DO 900 NH = 1, NHARMS
      HARM = NH
      ALPHA = HARM*3.14159265/SPAN
      CALL FLEX

```

```

C
C   REDUCE FLEXIBILITY MATRIX
C
      CALL SYMSOL (SF, XZERO, NEQ, 8, 1)

```

```

C
C   SET UP XZERO FOR EACH CASE
C

```

```

      DO 800 K = 1, NLCASE
      DO 510 J = 1, NEQ
510  XZERO(J) = 0.0

```

```

C
C   ZERO DAMPL, TAMPL, SFAMPL, FAMPL, YMAMPL, ZMAMPL
C

```

```

      DO 515 NB = 1, NBEAMS
      DAMPL(NB) = 0.0
      TAMPL(NB) = 0.0
      SFAMPL(NB) = 0.0
      FAMPL(NB) = 0.0
      YMAMPL(NB) = 0.0
515  ZMAMPL(NB) = 0.0

```

```

C
C   FORM LOAD VECTOR FOR EACH LOAD CASE
C

```

```

      DO 520 N = 1, NLCARD
      KK = LC(N)
      IF(KK.NE.K) GO TO 520
      NB = NLB(N)
      NT = NTY(NB)

```

```

AM000410
AM000420
AM000430
AM000440
AM000450
AM000460
AM000470
AM000480
AM000490
AM000500
AM000510
AM000520
AM000530
AM000540
AM000550
AM000560
AM000570
AM000580
AM000590
AM000600
AM000610
AM000620
AM000630
AM000640
AM000650
AM000660
AM000670
AM000680
AM000690
AM000700
AM000710
AM000720
AM000730
AM000740
AM000750
AM000760
AM000770
AM000780
AM000790
AM000800

```

337

```

PP= P(N) / PL(N)
PN = 4.*PP*SIN(ALPHA*XL(N))*SIN(ALPHA*PL(N)/2.)/(3.14159265*HARM)
C
NR = 4 * NB = 3
G1 = PN/(ALPHA**3*YMI(NT))
G2 = ECC(N)*PN/(ALPHA**2*BMJ(NT))
G3 = G1/ALPHA
XZERO(NR) = XZERO(NR) + ZH(NT,1)*G1
NR = NR + 1
XZERO(NR) = XZERO(NR) - ZH(NT,1)*G2
NR = NR + 1
XZERO(NR) = XZERO(NR) - G3 + YH(NT,1)*G2
NR = NR + 1
XZERO(NR) = XZERO(NR) + G2
NR = NR + 1
XZERO(NR) = XZERO(NR) - ZH(NT,2)*G1
NR = NR + 1
XZERO(NR) = XZERO(NR) + ZH(NT,2)*G2
NR = NR + 1
XZERO(NR) = XZERO(NR) + G3 - YH(NT,2)*G2
NR = NR + 1
XZERO(NR) = XZERO(NR) - G2
C
C STORE AMPLITUDES OF DISPL, MOMENT, SHEAR, AND TORSION DUE TO LOADS
C
DAMPL(NB) = DAMPL(NB) + G3
YMAMPL(NB) = YMAMPL(NB) - PN/ALPHA**2
SFAMPL(NB) = SFAMPL(NB) + PN/ALPHA
TAMPL(NB) = TAMPL(NB) - ECC(N)*PN/ALPHA
520 CONTINUE
DO 530 N = 1, 4
530 XZERO(N) = 0.0
NF = NEQ = 3
DO 535 N = NF, NEQ
535 XZERO(N) = 0.0
C
C REDUCE R.F.S. VECTOR AND BACK-SUBSTITUTE TO SOLVE FOR X=AMPL
C
CALL SYMSOL ( SF, XZERO, NEQ, 8, 2)
C

```

AM000810
 AM000820
 AM000830
 AM000840
 AM000850
 AM000860
 AM000870
 AM000880
 AM000890
 AM000900
 AM000910
 AM000920
 AM000930
 AM000940
 AM000950
 AM000960
 AM000970
 AM000980
 AM000990
 AM001000
 AM001010
 AM001020
 AM001030
 AM001040
 AM001050
 AM001060
 AM001070
 AM001080
 AM001090
 AM001100
 AM001110
 AM001120
 AM001130
 AM001140
 AM001150
 AM001160
 AM001170
 AM001180
 AM001190
 AM001200

C	TRANSFORM REDUNDANTS TO BEAM CENTER=LINE	AM001210
C		AM001220
	DO 550 NB = 1, NBEAMS	AM001230
	NT = NTY(NB)	AM001240
	DO 540 I = 1, 8	AM001250
	NN = 4 * NB - 4 + I	AM001260
540	X(I) = XZERO(NN)	AM001270
	X(4) = X(4) + YH(NT,1)*X(3) - ZH(NT,1)*X(2) - X(8) - YH(NT,2)*X(7)	AM001280
1	+ ZH(NT,2)*X(6)	AM001290
	X(3) = X(3) - ALPHA*ZH(NT,1)*X(1) - X(7) + ALPHA*ZH(NT,2)*X(5)	AM001300
	X(2) = X(2) - ALPHA*YH(NT,1)*X(1) - X(6) + ALPHA*YH(NT,2)*X(5)	AM001310
	X(1) = X(1) - X(5)	AM001320
C		AM001330
C	STORE AMPL. CF DISPL., MOMENTS, SHEAR, AXIAL FORCE AND TORSIONAL	AM001340
C	MOMENT DUE TO REDUNDANTS	AM001350
C		AM001360
	DAMPL(NB) = DAMPL(NB) + X(3)/(YMI(NT)*ALPHA**4)	AM001370
	YMAMPL(NB) = YMAMPL(NB) - X(3)/ALPHA**2	AM001380
	ZMAMPL(NB) = ZMAMPL(NB) - X(2)/ALPHA**2	AM001390
	FAMPL(NB) = FAMPL(NB) + X(1)/ALPHA	AM001400
	SFAMPL(NB) = SFAMPL(NB) + X(3)/ALPHA	AM001410
550	TAMPL(NB) = TAMPL(NB) + X(4)/ALPHA	AM001420
C		AM001430
C	COMPUTE BM, SF, TCRS. MOM., AND DISPL. FOR REQUIRED POSITIONS	AM001440
C		AM001450
	NP = NPOS(K)	AM001460
	DO 700 M = 1, NP	AM001470
	ARG=ALPHA*SPAN/2.	AM001480
	IF(KODE.EQ.0) ARG=ALPHA*XPOS(K,M)	AM001490
	SS=SIN(ARG)	AM001500
	CC=COS(ARG)	AM001510
	DO 600 NB = 1, NBEAMS	AM001520
	YMOUT(K,NB,M) = YMOUT(K,NB,M) + YMAMPL(NB)*SS	AM001530
	IF(KODE.NE.0) GO TO 600	AM001540
	DMOUT(K,NB,M) = DMOUT(K,NB,M) + DAMPL(NB)*SS	AM001550
	ZMOUT(K,NB,M) = ZMOUT(K,NB,M) + ZMAMPL(NB)*SS	AM001560
	FMOUT(K,NB,M) = FMOUT(K,NB,M) + FAMPL(NB)*SS	AM001570
	SFMOUT(K,NB,M) = SFMOUT(K,NB,M) + SFAMPL(NB)*CC	AM001580
	TMOUT(K,NB,M) = TMOUT(K,NB,M) + TAMPL(NB)*CC	AM001590
600	CONTINUE	AM001600

C
C
C

STORE AMPLITUDE OF REDUNDANTS

IF(KODE.NE.0) GO TO 700
DO 630 NN = 1,NX,4
NM = NN + 4
630 XOUT(K,NN,M) = XOUT(K,NN,M) + XZERO(NM)*CC
DO 640 NN = 3,NX,4
NM = NN + 4
640 XOUT(K,NN,M) = XOUT(K,NN,M) + XZERO(NM)*SS
DO 650 NN = 2, NX, 2
NM = NN + 4
650 XOUT(K,NN,M) = XOUT(K,NN,M) + XZERO(NM)*SS
700 CONTINUE
800 CONTINUE
900 CONTINUE
IF(KODE.EQ.0)CALL OUTPTT
IF(KODE.NE.0)CALL OUTPUT
GO TO 1
1900 CONTINUE
STOP
END

AM001610
AM001620
AM001630
AM001640
AM001650
AM001660
AM001670
AM001680
AM001690
AM001700
AM001710
AM001720
AM001730
AM001740
AM001750
AM001760
AM001770
AM001780
AM001790
AM001800
AM001810
AM001820

```

SUBROUTINE INPTT
COMMON/BLK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),
*ZSA(20),YSA(20),BMJ(20),YH(10,2),ZH(10,2),NTY(30),
*LC( 500),NLB( 500),LM(4),P( 500),XL( 500),PL( 500),
*ECC( 500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),
*NPOS( 500),XPOS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),
*NDUT(9),YSSS(9,4),ZSSS(9,4),X(8)
COMMON/BLK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYPES,NEQ,
*ALPHA,NLCASE,NLCARD,NX
READ(99,1000) (HEAD(J1),J1=1,15)
1000 FORMAT(15A4)
READ 1010, SPAN,E,G,NBEAMS,NTYPES,JTYPES,NHARMS
1010 FORMAT (3F10.0,4I5)
WRITE(6,2000) (HEAD(J1),J1=1,15)
2000 FORMAT(1H1,9X,' *****ANALYSIS OF MULTI-BEAM BRIDGE*****',//,1X,15A
14)
PRINT 2010, SPAN,E,G,NBEAMS,NTYPES,JTYPES,NHARMS
2010 FORMAT (10X,11HBRIDGE SPAN,29X,1H=,F12.3//
1 10X,28HYOUNGS MODULUS OF ELASTICITY,12X,1H=,F12.0//
2 10X,14HPCISSONS RATIO,26X,1H=,F12.3//
3 10X,15HNUMBER OF BEAMS,25X,1H=,I6//
4 10X,20HNUMBER OF BEAM-TYPES,20X,1H=,I6//
5 10X,21HNUMBER OF JOINT-TYPES,19X,1H=,I6//
6 10X,19HNUMBER OF HARMONICS,21X,1H=,I6//)
C
G = E/(2.*(1. + G))
C
C READ IN BEAM PROPERTIES FOR EACH TYPE
C
PRINT 2020
2020 FORMAT (1H1,9X,26H *****BEAM PROPERTIES*****//)
1 1X,4HTYPE,7X,4HI=ZZ,7X,4HI=YY,7X,4HAREA,
2 5X,6HTORS J,8X,3HZHL,8X,3HYHL,8X,3HZHR,8X,3HYHR/)
DO 150 N = 1, NTYPES
READ 1030, ZMI(N),YMI(N),EMA(N),BMJ(N),ZH(N,1),YH(N,1),ZH(N,2),
1 YH(N,2)
1030 FORMAT (4F10.0/4F10.0)
PRINT 2030, N,ZMI(N),YMI(N),BMA(N),BMJ(N),ZH(N,1),
1 YH(N,1),ZH(N,2),YH(N,2)
2030 FORMAT (/15,4F11.1,4F11.2)

```

AM001830
AM001840
AM001850
AM001860
AM001870
AM001880
AM001890
AM001900
AM001910
AM001920
AM001930
AM001940
AM001950
AM001960
AM001970
AM001980
AM001990
AM002000
AM002010
AM002020
AM002030
AM002040
AM002050
AM002060
AM002070
AM002080
AM002090
AM002100
AM002110
AM002120
AM002130
AM002140
AM002150
AM002160
AM002170
AM002180
AM002190
AM002200
AM002210
AM002220

150 CONTINUE	AM002230
C	AM002240
C IDENTIFY BEAMS BY TYPE	AM002250
C	AM002260
PRINT 2040	AM002270
2040 FORMAT (///2X,7HBEAM NO,5X,4HTYPE/)	AM002280
READ 1050, (NTY(N), N = 1, NBEAMS)	AM002290
1050 FORMAT (20I2)	AM002300
PRINT 2050, (N, NTY(N), N = 1, NBEAMS)	AM002310
2050 FORMAT (2I5)	AM002320
C	AM002330
C READ IN HINGE FLEXIBILITIES	AM002340
C	AM002350
PRINT 2060	AM002360
2060 FORMAT (////10X,30H *****HINGE FLEXIBILITIES*****//	AM002370
1 1X,4HTYPE,15X,5HLONG.,14X,6HHCRIZ.,15X,5HVERT.,16X,4HROT./)	AM002380
DO 160 J = 1, JTYPES	AM002390
READ 1070, (HF(J,N), N = 1,4)	AM002400
1070 FORMAT (4F10.0)	AM002410
160 PRINT 2065, J, (HF(J,N), N = 1,4)	AM002420
2065 FORMAT (I5,1P4E20.5)	AM002430
C	AM002440
C IDENTIFY JOINTS BY TYPE	AM002450
C	AM002460
NJ = NBEAMS = 1	AM002470
2070 FORMAT (///9H JOINT NO,5X,4HTYPE/)	AM002480
PRINT 2070	AM002490
READ 1075, (JTY(J), J = 1, NJ)	AM002500
1075 FORMAT (20I2)	AM002510
PRINT 2050, (J,JTY(J), J = 1, NJ)	AM002520
C	AM002530
C READ IN LOAD DATA	AM002540
C	AM002550
READ 1080, NLCASE,NLCARD	AM002560
1080 FORMAT (2I5)	AM002570
PRINT 2080	AM002580
2080 FORMAT (1H1,9X,29H *****LOADING CONDITIONS*****//	AM002590
1 1X,9HLOAD CASE,1X,7HBEAM NO,9X,4HLOAD,6X,7HX COORD,7X,	AM002600
2 6HLENGTH,6X,4HECC.,//)	AM002610
DO 175 N = 1, NLCARD	AM002620

READ 1090, LC(N),NLB(N),P(N),XL(N),PL(N),ECC(N)	AM002630
1090 FORMAT (2I5,4F10.0)	AM002640
PRINT 2090, LC(N),NLB(N),P(N),XL(N),PL(N),ECC(N)	AM002650
175 ECC(N) = -ECC(N)	AM002660
2090 FORMAT (/2I9,4F13.3)	AM002670
C	AM002680
C READ IN DATA SPECIFYING X COORDS FOR RESULTS	AM002690
C	AM002700
PRINT 2100	AM002710
2100 FORMAT (/////10X,37+ *****JOINT X=COORDS FOR RESULTS*****//	AM002720
1 22X,5HNO OF/	AM002730
2 9X,9HLOAD CASE,2X,9HPOSITIONS,9X,2HX1,7X,2HX2,7X,2HX3,7X,	AM002740
3 2HX4,7X,2HX5,7X,2HX6,7X,2HX7,7X,2HX8,7X,2HX9)	AM002750
DO 200 J = 1, NLCASE	AM002760
READ 1110, L,NP, (XPOS(L,N), N = 1, NP)	AM002770
1110 FORMAT (2I5,9F5.0)	AM002780
NPOS(L) = NP	AM002790
200 PRINT 2110, L,NP, (XPCS(L,N), N = 1, NP)	AM002800
2110 FORMAT (/I18,I7,F15.1,8F9.1)	AM002810
C	AM002820
C READ IN COORDINATES SPECIFIED FOR STRESS OUTPUT	AM002830
C	AM002840
PRINT 2120	AM002850
2120 FORMAT (/////10X,49H *****COORDINATES FOR OUTPUT OF AXIAL STRESS**	AM002860
1****//22X,5HNO OF/	AM002870
2 9X,9HBEAM TYPE,2X,9HPOSITIONS,11X,2HY1,7X,2HZ1,12X,2HY2,	AM002880
3 7X,2HZ2,12X,2HY3,7X,2HZ3,12X,2HY4,7X,2HZ4//)	AM002890
DO 225 NT = 1, NTYPES	AM002900
READ 1130, NOUT(NT),(YSTR(NT,N),ZSTR(NT,N), N = 1, 4)	AM002910
1130 FORMAT (I5,8F5.0)	AM002920
NO = NOUT(NT)	AM002930
225 PRINT 2130, NT,NO,(YSTR(NT,N),ZSTR(NT,N), N = 1, NO)	AM002940
2130 FORMAT (I18,I7,8X,2F9.1,5X,2F9.1,5X,2F9.1,5X,2F9.1)	AM002950
DO 300 NT = 1, NTYPES	AM002960
DO 275 N = 1, 4	AM002970
YSSS(NT,N) = YSTR(NT,N)/ZMI(NT)	AM002980
275 ZSSS(NT,N) = ZSTR(NT,N)/YMI(NT)	AM002990
ZMI(NT) = E * ZMI(NT)	AM003000
YMI(NT) = E * YMI(NT)	AM003010
BMJ(NT) = BMJ(NT) * G	AM003020

BMA(NT) = E * BMA(NT)
300 CONTINUE
C
RETURN
C
END

AM003030
AM003040
AM003050
AM003060
AM003070
AM003080

	SUBROUTINE SYMSCL (A,B,NN,MM,KKK)	AM003090
	DIMENSION A(164,8),B(164)	AM003100
C		AM003110
C		AM003120
C	A = COEFFICIENT MATRIX.	AM003130
C	B = RIGHT HAND SIDE MATRIX.	AM003140
C	NN = NUMBER OF EQUATIONS (MAX 800 HERE).	AM003150
C	MM = HALF BAND WIDTH (MAX 20 HERE).	AM003160
C		AM003170
	GO TO (1000,2000),KKK	AM003180
C		AM003190
C	REDUCE COEFFICIENT MATRIX.	AM003200
C		AM003210
	1000 NL = NN - 4	AM003220
	DO 280 N = 5, NL	AM003230
	DO 260 L=2,MM	AM003240
	C=A(N,L)/A(N,1)	AM003250
	I = N+L-1	AM003260
	IF(NL=I) 260,240,240	AM003270
	240 J=0	AM003280
	DO 250 K=L,MM	AM003290
	J=J+1	AM003300
	250 A(I,J)=A(I,J)-C*A(N,K)	AM003310
	260 A(N,L)=C	AM003320
	280 CONTINUE	AM003330
	GO TO 500	AM003340
C		AM003350
C	REDUCE RIGHT HAND SIDE MATRIX.	AM003360
C		AM003370
	2000 DO 290 N = 5, NL	AM003380
	DO 285 L=2,MM	AM003390
	I=N+L-1	AM003400
	IF(NL=I) 290,285,285	AM003410
	285 B(I)=B(I)-A(N,L)*B(N)	AM003420
	290 B(N)=B(N)/A(N,1)	AM003430
C		AM003440
C	CARRY OUT BACK SUBSTITUTION.	AM003450
C		AM003460
	N=NL	AM003470
	300 N = N-1	AM003480

```
      IF(N=4) 350,500,350
350 DO 400 K=2,MM
      L = N+K-1
      IF(NL=L) 400,370,370
370 B(N) = B(N) - A(N,K) * B(L)
400 CONTINUE
      GO TO 300
C
500 RETURN
C
      END
```

```
AM003490
AM003500
AM003510
AM003520
AM003530
AM003540
AM003550
AM003560
AM003570
AM003580
AM003590
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SUBROUTINE INPUT
  INTEGER*2 TITLE,HH,SS,DIGIT1,DIGIT2,BK,A1,A2,KEY,HINGTP,YES,NO,
  *A3,A4
  COMMON/BLK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),
  *ZSA(20),YSA(20),BMJ(20),YH(10,2),ZH(10,2),NTY(30),
  *LC( 500),NLB( 500),LM(4),P( 500),XL( 500),PL( 500),
  *ECC( 500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),
  *NPOS( 500),XPOS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),
  *NOUT(9),YSSS(9,4),ZSSS(9,4),X(8)
  COMMON/BLK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYPES,NEG,
  *ALPHA,NLCASE,NLCARD,NX
  COMMON/BLK 3/ XZERO(164),YMAMPL(20),ZMAMPL(20),
  *FAMPL(20),SFAMPL(20),TAMPL(20),CAMPL(20),YMOUT( 200,20,9),
  *ZMOUT(9,20,9),FOUT(9,20,9),SFOUT(9,20,9),TCUT(9,20,9),
  *DOUT(9,20,9),XOUT(9,76,9),XS(4),XA(4)
  COMMON/BLK 4/ TITLE(3,54),HINGTP(19,4),PWHEEL(18),ZNWHL(18),
  *DIST(18),WHLWNM(18),TLN(5,2),KASAST(5,2),KASASL(5,2),
  *KASAXT(5,2),ZMAST(20,5),ZMASL(20,5),ZMAXT(20,5),ZMMAST(20),
  *ZMMASL(20),ZMMAXT(20),POSAT(20,10),POSLN(20,10),POSAX(20,10),
  *NLLAST(20,5),NLLALN(20,5),NLLAXT(20,5),DISTAT(20),
  *DISTAL(20),DISTAX(20)
  COMMON/BLK 5/ NWHLS,NWHEEL,A1,A2,KAXT,FULMAT,FULMAL,FULMAX,PMAT,PM
  *AXT,NTRFL,NAXCL,NAXTSP,NAXT,JTYPES,A3,A4,ZPAN
  DATA HH,SS,DIGIT1,DIGIT2,BK/'H','S','1','2',' ' /
  DATA YES,NO/'Y','N' /
  READ(99,104) (TITLE(1,J2),J2=1,54)
  DO 102 J1=2,3
102 READ(5,104)(TITLE(J1,J2),J2=1,54)
104 FORMAT(54A1)
  READ(5,106)ZPAN,E,NBEAMS,NTRFL,A1,A2,A3,A4,KAXT,NAXTSP,NAXCL,NAXT
106 FORMAT(4X,F4.1,4X,F6.1,5X,I2,5X,I2,5X,2A1,1X,2A1,5X,I1,5X,I2,5X,I1
  *,5X,I1)
  SPAN=ZPAN*12.
  NTYPES=0
108 READ(5,112) KEY
112 FORMAT(3X,A1)
  IF(KEY.EQ.BK) GO TO 116
  READ(99,114) IBMN,YMI( IBMN),ZMI( IBMN),BMA( IBMN),BMJ( IBMN),
  *YH( IBMN,1),ZH( IBMN,1),YH( IBMN,2),ZH( IBMN,2)
114 FORMAT(3X,I1,7X,F8.1,4X,F8.1,4X,F5.1,4X,F8.1,3X,4(F5.2,1X))

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AM003600
AM003610
AM003620
AM003630
AM003640
AM003650
AM003660
AM003670
AM003680
AM003690
AM003700
AM003710
AM003720
AM003730
AM003740
AM003750
AM003760
AM003770
AM003780
AM003790
AM003800
AM003810
AM003820
AM003830
AM003840
AM003850
AM003860
AM003870
AM003880
AM003890
AM003900
AM003910
AM003920
AM003930
AM003940
AM003950
AM003960
AM003970
AM003980
AM003990

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YH(IBMN,1)=-YH(IBMN,1)	AM004000
NTYPES=NTYPES+1	AM004010
GO TO 108	AM004020
116 READ(99,118) (NTY(J1),J1=1,NBEAMS)	AM004030
118 FORMAT(4X,20(I2,1X))	AM004040
JTYPES=0	AM004050
122 READ(5,112) KEY	AM004060
IF(KEY.EQ.BK) GO TO 126	AM004070
READ(99,124) IJTN,(HINGTP(IJTN,J1),J1=1,4)	AM004080
124 FORMAT(3X,11,9X,4(A1,7X))	AM004090
JTYPES=JTYPES+1	AM004100
GO TO 122	AM004110
126 DO 132 J1=1,JTYPES	AM004120
DO 128 J2=1,4	AM004130
128 HF(J1,J2)=1.E+08	AM004140
IF(HINGTP(J1,1).EQ.YES) HF(J1,1)=0.	AM004150
IF(HINGTP(J1,2).EQ.YES) HF(J1,3)=0.	AM004160
IF(HINGTP(J1,3).EQ.YES) HF(J1,2)=0.	AM004170
IF(HINGTP(J1,4).EQ.YES) HF(J1,4)=0.	AM004180
132 CONTINUE	AM004190
NJ=NBEAMS-1	AM004200
READ(99,118) (JTY(J1),J1=1,NJ)	AM004210
READ(5,134) (TLN(J1,1),TLN(J1,2),J1=1,NTRFL)	AM004220
134 FORMAT(4X,5(F4.1,4X,F4.1,4X))	AM004230
IF(KAXT.EQ.0) GO TO 150	AM004240
READ(5,136) (PWHEEL(J1),J1=1,18)	AM004250
READ(5,138) (ZNWHL(J1),J1=2,18)	AM004260
136 FORMAT(4X,18(F3.1,1X))	AM004270
138 FORMAT(8X,17(F3.0,1X))	AM004280
NWHEEL=0	AM004290
DO 140 J1=1,18	AM004300
IF(PWHEEL(J1).EQ.0.) GO TO 142	AM004310
140 NWHEEL=NWHEEL+1	AM004320
142 CONTINUE	AM004330
150 NLCARD=0	AM004340
NLCASE=0	AM004350
IF(A1.EQ.BK.AND.A2.EQ.BK) GO TO 514	AM004360
C *****	AM004370
C SET UP INFLUENCE COEFFICIENT LOAD CASES=AASHTO TRUCK	AM004380
C *****	AM004390

NW=2	AM004400
IF(A1.EQ.HH.AND.A2.EQ.SS) NW=3	AM004410
SCALE=0.5	AM004420
IF(A3.EQ.DIGIT1) SCALE=0.375	AM004430
WHLWNM(1)=SCALE*8.	AM004440
DO 156 J1=2,NW	AM004450
WHLWNM(J1)=SCALE*32.	AM004460
156 DIST(J1)=(J1-1)*14.	AM004470
CALL MAXMOM(NW,WHLWNM,DIST,ZPAN,FULMAT)	AM004480
FULMAT=FULMAT*2.*12.	AM004490
DO 506 J1=1,NTRFL	AM004500
KASAST(J1,1)=NLCASE+1	AM004510
YDIST=TLN(J1,1)*12.-12.	AM004520
502 YDIST=YDIST+12.	AM004530
IF(YDIST.GT.12.*TLN(J1,2)) GO TO 506	AM004540
NLCASE=NLCASE+1	AM004550
CALL LOCATE(YDIST,NBM,ECCEN,NBEAMS,YH,NTY)	AM004560
DO 504 J2=1,NW	AM004570
NLCARD=NLCARD+1	AM004580
LC(NLCARD)=NLCASE	AM004590
NLB(NLCARD)=NBM	AM004600
P(NLCARD)=WHLWNM(J2)	AM004610
XL(NLCARD)=DIST(J2)*12.	AM004620
PL(NLCARD)=12.	AM004630
ECC(NLCARD)=ECCEN	AM004640
504 CONTINUE	AM004650
GO TO 502	AM004660
506 KASAST(J1,2)=NLCASE	AM004670
C *****	AM004680
C SET UP INFLUNCE COEFFICIENT LOAD CASES=AASHTC LANE	AM004690
C *****	AM004700
WW=0.48	AM004710
CONF=13.5	AM004720
IF(A3.EQ.DIGIT2) WW=0.640	AM004730
IF(A3.EQ.DIGIT2) CONF=18.	AM004740
FULMAL=(WW/12.)*SPAN**2/8.+CONF*SPAN/4.	AM004750
DO 512 J1=1,NTRFL	AM004760
KASASL(J1,1)=NLCASE+1	AM004770
YDIST=TLN(J1,1)*12.-12.	AM004780
508 YDIST=YDIST+12.	AM004790

IF(YDIST.GT.12.*TLN(J1,2)) GO TO 512	AM004800
NLCASE=NLCASE+1	AM004810
CALL LOCATE(YDIST,NBM,ECCEN,NBEAMS,YH,NTY)	AM004820
NLCARD=NLCARD+1	AM004830
LC(NLCARD)=NLCASE	AM004840
NLB(NLCARD)=NBM	AM004850
P(NLCARD)=(WW/10.)*SPAN/12.	AM004860
XL(NLCARD)=SPAN/2.	AM004870
PL(NLCARD)=SPAN	AM004880
ECC(NLCARD)=ECCEN	AM004890
NLCARD=NLCARD+1	AM004900
LC(NLCARD)=NLCASE	AM004910
NLB(NLCARD)=NBM	AM004920
P(NLCARD)=CONF/10.	AM004930
XL(NLCARD)=SPAN/2.	AM004940
PL(NLCARD)=12.	AM004950
ECC(NLCARD)=ECCEN	AM004960
GO TO 508	AM004970
512 KASASL(J1,2)=NLCASE	AM004980
C *****	AM004990
C SET UP INFLUENCE COEFFICIENT LOAD CASES=AXLE TRAIN	AM005000
C *****	AM005010
514 IF(KAXT.EQ.0) GO TO 526	AM005020
NWHL=NWHEEL	AM005030
WHLWNM(1)=PWHEEL(1)/2.	AM005040
DO 530 J1=2,NW	AM005050
WHLWNM(J1)=PWHEEL(J1)/2.	AM005060
530 DIST(J1)=ZNWHL(J1)	AM005070
CALL MAXMM(NWHL,WHLWNM,DIST,ZPAN,FULMAX)	AM005080
FULMAX=FULMAX*2.*12.	AM005090
DO 524 J1=1,NTRFL	AM005100
KASAXT(J1,1)=NLCASE+1	AM005110
YDIST=TLN(J1,1)*12.-12.	AM005120
516 YDIST=YDIST+12.	AM005130
IF(YDIST.GT.12.*TLN(J1,2)) GO TO 524	AM005140
NLCASE=NLCASE+1	AM005150
CALL LOCATE(YDIST,NBM,ECCEN,NBEAMS,YH,NTY)	AM005160
DO 518 J2=1,NWHL	AM005170
NLCARD=NLCARD+1	AM005180
LC(NLCARD)=NLCASE	AM005190

NLB(NLCARD)=NBM	AM005200
P(NLCARD)=WHLWNM(J2)	AM005210
XL(NLCARD)=DIST(J2)*12.	AM005220
PL(NLCARD)=12.	AM005230
ECC(NLCARD)=ECCEN	AM005240
518 CONTINUE	AM005250
GO TO 516	AM005260
524 KASAXT(J1,2)=NLCASE	AM005270
526 CONTINUE	AM005280
C	AM005290
C PRINT OUT INPUT	AM005300
C	AM005310
WRITE(6,1100)	AM005320
1100 FORMAT('1')	AM005330
WRITE(6,1101)	AM005340
WRITE(6,1103)	AM005350
WRITE(6,1101)	AM005360
WRITE(6,1104) (TITLE(1,J1),J1=10,11),(TITLE(1,J1),J1=14,26),	AM005370
1(TITLE(1,J1),J1=48,54)	AM005380
WRITE(6,1105) (TITLE(2,J2),J2=13,19),(TITLE(2,J2),J2=26,28),	AM005390
1(TITLE(2,J2),J2=45,54)	AM005400
WRITE(6,1106) (TITLE(3,J3),J3=13,54)	AM005410
WRITE(6,1101)	AM005420
WRITE(6,1107)	AM005430
WRITE(6,1108) ZPAN,E,NBEAMS,NT RFL,A1,A2,A3,A4,NAXTSP,NAXCL,NAXT	AM005440
WRITE(6,1102)	AM005450
WRITE(6,1109)	AM005460
DO 1240 IBMN=1,NTYPES	AM005470
WRITE(6,1110) IBMN,YMI(IBM),ZMI(IBM),BMA(IBM),BMJ(IBM),	AM005480
1YH(IBM,1),ZH(IBM,1),YH(IBM,2),ZH(IBM,2)	AM005490
1240 CONTINUE	AM005500
WRITE(6,1102)	AM005510
WRITE(6,1111)	AM005520
WRITE(6,1112) (NTY(J1),J1=1,NBEAMS)	AM005530
WRITE(6,1102)	AM005540
WRITE(6,1113)	AM005550
DO 1235 IHING=1,JTYPES	AM005560
1235 WRITE(6,1114) IHING,(HINGTP(IHING,J1),J1=1,4)	AM005570
WRITE(6,1102)	AM005580
WRITE(6,1115)	AM005590


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WRITE(6,1116) (JTY(J1),J1=1,NJ)
WRITE(6,1102)
WRITE(6,1117)
WRITE(6,1118) (TLN(J1,1),TLN(J1,2),J1=1,NTRFL)
WRITE(6,1102)
IF(KAXT.EQ.0) GC TO 1255
WRITE(6,1119)
WRITE(6,1120) (PWHEEL(J1),J1=1,18)
WRITE(6,1121)
WRITE(6,1122) (ZNWHL(J1),J1=2,18)
WRITE(6,1102)
1255 WRITE(6,1123)
WRITE(6,1101)
1101 FORMAT(' ',1X,129('*'))
1102 FORMAT(' ',1X,129(' '))
1103 FORMAT(' ',/,51X,'VALUES ASSIGNED TO INPUT DATA',/)
1104 FORMAT(' ',/,30X,'DISTRICT ',2A1,4X,13A1,' CCUNTY HIGHWAY NO. ',
1,7A1,/)
1105 FORMAT(' ',30X,'CONTROL NO. ',7A1,5X,'IPE ',3A1,4X,'SUBMITTED BY:',
1,1X,10A1,/)
1106 FORMAT(' ',30X,'DESCRIPTION: ',42A1,/)
1107 FORMAT(' ',/,51X,'..... BRIDGE PROPERTIES .....',/,25X,'MODULUS',
116X,'NUMBER',22X,'TRANSVERSE',5X,'AXLE TRAIN',/,28X,'OF',7X,
1'NUMBER',7X,'OF',24X,'AXLE TRAIN',7X,'SIDE',10X,'NUMBER OF',/,12X,
1'SPAN',8X,'ELASTICITY',5X,'OF',7X,'TRAFFIC',5X,'AASHTO',9X,
1'WHEEL SPACING CLEARANCE',6X,'AXLE TRAINS',/,12X,'(FT.)',10X,
1'(KSI)',6X,'BEAMS',6X,'LANES',6X,'LOADING',12X,'(FT.)',9X,'(FT.)',
19X,'ON BRIDGE',/)
1108 FORMAT(' ',9X,F6.1,8X,F8.1,7X,I2,9X,I2,9X,2A1,'-',2A1,14X,I2,
113X,I1,15X,I1,/)
1109 FORMAT(' ',/,44X,5(' '),* BEAM DIMENSIONS AND PROPERTIES .....',/,
1,18X,'INERTIA ABOUT INERTIA ABOUT',/,11X,'BEAM',4X,'HORIZONTAL',
16X,'VERTICAL',20X,'TORSIONAL',/,11X,'TYPE',7X,'AXIS',11X,'AXIS',
110X,'AREA',8X,'STIFFNESS',10X,'HL',8X,'VL',8X,'HR',8X,'VR',/,10X,
1'NUMBER',4X,'(IN.**4)',7X,'(IN.**4)',6X,'(IN.**2)',6X,'(IN.**4)',
110X,4(' (IN.) '),/)
1110 FORMAT('0',11X,I1,5X,F10.1,5X,F10.1,5X,F7.1,6X,F10.1,5X,4F10.2)
1111 FORMAT(' ',/,38X,'..... BEAM TYPE IDENTIFICATION NUMBER FOR BEAM I
1 .....',/,14X,'I=1 I=2 I=3 I=4 I=5 I=6 I=7 I=8 I=9 I=10
1I=11 I=12 I=13 I=14 I=15 I=16 I=17 I=18 I=19 I=20',/)

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AM005600
AM005610
AM005620
AM005630
AM005640
AM005650
AM005660
AM005670
AM005680
AM005690
AM005700
AM005710
AM005720
AM005730
AM005740
AM005750
AM005760
AM005770
AM005780
AM005790
AM005800
AM005810
AM005820
AM005830
AM005840
AM005850
AM005860
AM005870
AM005880
AM005890
AM005900
AM005910
AM005920
AM005930
AM005940
AM005950
AM005960
AM005970
AM005980
AM005990

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1112 FORMAT(' ',10X,20I5,/)
1113 FORMAT(' ',/,46X,'..... HINGE FORCE TRANSMISSION .....',/,28X,
1'HINGE TYPE',5X,'LONGITUDINAL',6X,'VERTICAL',7X,2('TRANSVERSE',6X)
1,/,30X,'NUMBER',10X,2('SHEAR',11X),'FORCE',11X,'MOMENT',/)
1114 FORMAT('0',31X,I1,1X,4A16)
1115 FORMAT(' ',/,37X,'..... HINGE TYPE IDENTIFICATION NUMBER FOR HINGE
1 I .....',/,17X,'I=1 I=2 I=3 I=4 I=5 I=6 I=7 I=8 I=9 I=10
10 I=11 I=12 I=13 I=14 I=15 I=16 I=17 I=18 I=19',/)
1116 FORMAT('0',13X,19I5)
1117 FORMAT(' ',/,36X,'..... DISTANCE (FT.) FROM C.G. AXIS OF BEAM 1 TO
1: .....',/,12X,6(' '),1X,6(' '),1X,6(' '),1X,6(' '),1X,6(' '),1X,6(' '),
1,1X,6(' '),1X,6(' '),1X,6(' '),1X,6(' '),1X,6(' '),1X,6(' '),1X,6(' '),
1'LANE 5',6(' '),/,14X,5('LEFT',7X,'RIGHT',4X),/,14X,5('EDGE',8X,
1'EDGE',4X),/)
1118 FORMAT('0',13X,10(F5.1,5X))
1119 FORMAT(' ',/,54X,'..... AXLE TRAIN .....',/,15X,18('AXLE',2X),/,
117X,'1',5X,'2',5X,'3',5X,'4',5X,'5',5X,'6',5X,'7',5X,'8',5X,'9',
14X,'10',4X,'11',4X,'12',4X,'13',4X,'14',4X,'15',4X,'16',4X,'17',
14X,'18',/,5X,'AXLE',/,5X,'LOAD',/,4X,'(KIPS)')
1120 FORMAT('+',13X,18(F5.1,1X),/)
1121 FORMAT(' ',3X,'DIST. FROM',/,4X,'AXLE 1 TO',/,4X,'AXLE 1 (FT.)')
1122 FORMAT('+',19X,17(F5.0,1X),/)
1123 FORMAT(/)
DO 170 J1=1,NLCARD
170 ECC(J1)=-ECC(J1)
G=E/(2.*(1+.166))
DO 172 J1=1,NTYPES
ZMI(J1)=E*ZMI(J1)
YMI(J1)=E*YMI(J1)
BMJ(J1)=G*BMJ(J1)
172 BMA(J1)=E*BMA(J1)
DO 164 J1=1,NLCASE
164 NPOS(J1)=1
RETURN
END

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AM006000
AM006010
AM006020
AM006030
AM006040
AM006050
AM006060
AM006070
AM006080
AM006090
AM006100
AM006110
AM006120
AM006130
AM006140
AM006150
AM006160
AM006170
AM006180
AM006190
AM006200
AM006210
AM006220
AM006230
AM006240
AM006250
AM006260
AM006270
AM006280
AM006290
AM006300
AM006310
AM006320
AM006330
AM006340

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	SUBROUTINE OUTPTT	AM006350
	COMMON/BLK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),	AM006360
	*ZSA(20),YSA(20),BMJ(20),YH(10,2),ZH(10,2),NTY(30),	AM006370
	*LC(500),NLB(500),LM(4),P(500),XL(500),PL(500),	AM006380
	*ECC(500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),	AM006390
	*NPOS(500),XPOS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),	AM006400
	*NOUT(9),YSSS(9,4),ZSSS(9,4),X(8)	AM006410
	COMMON/BLK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYFES,NEQ,	AM006420
	*ALPHA,NLCASE,NLCARD,NX	AM006430
	COMMON/BLK 3/ XZERO(164),YMAMPL(20),ZMAMPL(20),	AM006440
	*FAMPL(20),SFAMPL(20),TAMPL(20),DAMPL(20),YMOUT(200,20,9),	AM006450
	*ZMOUT(9,20,9),FCUT(9,20,9),SFOUT(9,20,9),TOUT(9,20,9),	AM006460
	*DOUT(9,20,9),XCUT(9,76,9),XS(4),XA(4)	AM006470
C		AM006480
C	PRINT RESULTS	AM006490
C		AM006500
	DO 1000 K = 1, NLCASE	AM006510
	NP = NPOS(K)	AM006520
	PRINT 10, K	AM006530
	10 FORMAT (1H1,9X,51H *****BEAM CENTER=LINE DISPLACEMENTS LOAD CASE	AM006540
	1 NO,I2,5H*****//)	AM006550
	PRINT 20, (XPOS(K,J), J = 1, NP)	AM006560
	20 FORMAT (18H0LOCATIONS ON BEAM, 9F11.1)	AM006570
	PRINT 25	AM006580
	25 FORMAT (//6X,7HBEAM NO)	AM006590
	DO 910 J = 1, NBEAMS	AM006600
	910 PRINT 30, J, (DOUT(K,J,M), M = 1, NP)	AM006610
	30 FORMAT (/8X,I3,7X,1P9E11.3)	AM006620
	PRINT 40, K	AM006630
	40 FORMAT (1H1,9X,49H *****BENDING MOMENTS ABOUT Y=AXIS LOAD CASE	AM006640
	10,I2,5H*****//)	AM006650
	PRINT 20, (XPOS(K,J), J = 1, NP)	AM006660
	PRINT 25	AM006670
	DO 43 J1=1,NBEAMS	AM006680
	DO 43 J2=1,NP	AM006690
	43 YMOUT(K,J1,J2)=YMOUT(K,J1,J2)	AM006700
	DO 920 J = 1, NBEAMS	AM006710
	920 PRINT 30, J, (YMOUT(K,J,M), M = 1, NP)	AM006720
	PRINT 45, K	AM006730
	45 FORMAT (1H1,9X,49H *****BENDING MOMENTS ABOUT Z=AXIS LOAD CASE	AM006740

10, I2, 5H*****/))	
PRINT 20, (XPOS(K,J), J = 1, NP)	AM006750
PRINT 25	AM006760
DO 925 J = 1, NBEAMS	AM006770
925 PRINT 30, J, (ZMOUT(K,J,M), M = 1, NP)	AM006780
PRINT 50, K	AM006790
50 FORMAT (1H1,9X,36H *****VERTICAL SHEARS LOAD CASE NO,I2,	AM006800
1 5H*****/))	AM006810
PRINT 20, (XPOS(K,J), J = 1, NP)	AM006820
PRINT 25	AM006830
DO 930 J = 1, NBEAMS	AM006840
930 PRINT 30, J, (SFCUT(K,J,M), M = 1, NP)	AM006850
PRINT 55, K	AM006860
55 FORMAT (1H1,9X,33H *****AXIAL FORCES LOAD CASE NO,I2,5H*****/))	AM006870
PRINT 20, (XPOS(K,J), J = 1, NP)	AM006880
PRINT 25	AM006890
DO 935 J = 1, NBEAMS	AM006900
935 PRINT 30, J, (FCUT(K,J,M), M = 1, NP)	AM006910
PRINT 60, K	AM006920
60 FORMAT (1H1,9X,38H *****TORSIONAL MOMENTS LOAD CASE NO,I2,	AM006930
1 5H*****/))	AM006940
PRINT 20, (XPOS(K,J), J = 1, NP)	AM006950
PRINT 25	AM006960
DO 940 J = 1, NBEAMS	AM006970
940 PRINT 30, J, (TCUT(K,J,M), M = 1, NP)	AM006980
PRINT 70, K	AM006990
70 FORMAT (1H1,9X,52H *****FORCES ALONG LONGITUDINAL JOINT LOAD CASE NO,I2,5H*****/))	AM007000
PRINT 75	AM007010
75 FORMAT (//28H0LONGITUDINAL SHEAR ON JOINT)	AM007020
PRINT 20, (XPOS(K,J), J = 1, NP)	AM007030
MM = NX / 4	AM007040
PRINT 95	AM007050
DO 950 J = 1, MM	AM007060
JJ = 4*J - 3	AM007070
950 PRINT 100, J, (XCUT(K,JJ,M), M = 1, NP)	AM007080
PRINT 80	AM007090
80 FORMAT (//26H0TRANSVERSE FORCE ON JOINT)	AM007100
PRINT 20, (XPCS(K,J), J = 1, NP)	AM007110
PRINT 95	AM007120
	AM007130
	AM007140

DO 960 J = 1, MM	AM007150
JJ = 4*J = 2	AM007160
960 PRINT 100,J, (XOUT(K,JJ,M), M = 1, NP)	AM007170
85 FORMAT (//24H0VERTICAL SHEAR ON JCINT)	AM007180
PRINT 85	AM007190
PRINT 20, (XPOS(K,J), J = 1, NP)	AM007200
PRINT 95	AM007210
DO 970 J = 1, MM	AM007220
JJ = 4*J = 1	AM007230
970 PRINT 100,J, (XOUT(K,JJ,M), M = 1, NP)	AM007240
PRINT 90	AM007250
90 FORMAT (//27H0TRANSVERSE MOMENT ON JOINT)	AM007260
PRINT 20, (XPOS(K,J), J = 1, NP)	AM007270
PRINT 95	AM007280
DO 980 J = 1, MM	AM007290
JJ = 4*J	AM007300
980 PRINT 100,J, (XOUT(K,JJ,M), M = 1, NP)	AM007310
95 FORMAT (/9H0JOINT NO)	AM007320
100 FORMAT (I5,I3X,I9E11.3)	AM007330
1000 CONTINUE	AM007340
C	AM007350
C COMPUTE AXIAL STRESS AT FOUR SPECIFIED POSITIONS	AM007360
C	AM007370
DO 1200 NE = 1, NBEAMS	AM007380
NT = NTY(NB)	AM007390
DO 1200 N = 1, 4	AM007400
NS = 4*NB-4+N	AM007410
DO 1200 K = 1, NLCASE	AM007420
NP = NPOS(K)	AM007430
DO 1200 M = 1, NP	AM007440
1200 XOUT(K,NS,M) = YMCUT(K,NB,M)*ZSSS(NT,N) + ZMOUT(K,NB,M)*	AM007450
1 YSSS(NT,N) + FOUT(K,NB,M)/BMA(NT)*E	AM007460
C	AM007470
C PRINT STRESSES	AM007480
C	AM007490
DO 1300 K = 1, NLCASE	AM007500
NP = NPOS(K)	AM007510
PRINT 110, K	AM007520
110 FORMAT (I1,I9X,33H *****AXIAL STRESS LOAD CASE NO,I2,5H*****/)	AM007530
DO 1300 NE = 1, NBEAMS	AM007540

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      NT = NTY(NB)
      NO = NOUT(NT)
      PRINT 120, NB
120  FORMAT (//8H BEAM NO, I2)
      PRINT 130, (XPOS(K,J), J = 1, NP)
130  FORMAT (18H0LCCATIONS ON BEAM,4X,9F11.1/4X,1HY,9X,1HZ/)
      DO 1300 N = 1, NO
      NS = 4*N-4+1
1300 PRINT 150, YSTR(NT,N),ZSTR(NT,N),(XOUT(K,NS,M), M = 1, NP)
150  FORMAT (2X,F5.1,F10.1,5X,1P9E11.3)
      RETURN
      END

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AM007550
AM007560
AM007570
AM007580
AM007590
AM007600
AM007610
AM007620
AM007630
AM007640
AM007650
AM007660

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SUBROUTINE OUTPUT	AM007670
INTEGER*2 TITLE,A1,A2,HINGTP,A3,A4	AM007680
COMMON/BLK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),	AM007690
*ZSA(20),YSA(20),BMJ(20),YH(10,2),ZH(10,2),NTY(30),	AM007700
*LC(500),NLB(500),LM(4),P(500),XL(500),PL(500),	AM007710
*ECC(500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),	AM007720
*NPOS(500),XPOS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),	AM007730
*NQUT(9),YSSS(9,4),ZSSS(9,4),X(8)	AM007740
COMMON/BLK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYPES,NEQ,	AM007750
*ALPHA,NLCASE,NLCARD,NX	AM007760
COMMON/BLK 3/ XZERO(164),YMAMPL(20),ZMAMPL(20),	AM007770
*FAMPL(20),SFAMPL(20),TAMPL(20),DAMPL(20),YMOUT(200,20,9),	AM007780
*ZMOUT(9,20,9),FOUT(9,20,9),SFOUT(9,20,9),TCUT(9,20,9),	AM007790
*DOUT(9,20,9),XOUT(9,76,9),XS(4),XA(4)	AM007800
COMMON/BLK 4/ TITLE(3,54),HINGTP(19,4),PWHEEL(18),ZNWHL(18),	AM007810
*DIST(18),WHLWNM(18),TLN(5,2),KASAST(5,2),KASASL(5,2),	AM007820
*KASAXT(5,2),ZMAST(20,5),ZMASL(20,5),ZMAXT(20,5),ZMMAST(20),	AM007830
*ZMMASL(20),ZMMAXT(20),POSAT(20,10),POSLN(20,10),POSAX(20,10),	AM007840
*NLLAST(20,5),NLLALN(20,5),NLLAXT(20,5),DISTAT(20),	AM007850
*DISTAL(20),DISTAX(20)	AM007860
COMMON/BLK 5/ NWHLN,NWHEEL,A1,A2,KAXT,FULMAT,FULMAL,FULMAX,PMAT,PM	AM007870
*AXT,NTRFL,NAXCL,NAXTSP,NAXT,JTYPES,A3,A4,ZPAN	AM007880
BEGIN PRINT OUT	AM007890
	AM007900
CALL INFLN	AM007910
	AM007920
CONVERT MOMENTS TO KIP FEET FOR OUTPUT	AM007930
	AM007940
DO 180 J2=1,NBEAMS	AM007950
IF(A3.NE.BK) ZMMAST(J2)=ZMMAST(J2)/12.	AM007960
IF(A3.NE.BK) ZMMASL(J2)=ZMMASL(J2)/12.	AM007970
IF(KAXT.NE.0) ZMMAXT(J2)=ZMMAXT(J2)/12.	AM007980
180 CONTINUE	AM007990
J3=2*NTRFL	AM008000
WRITE(6,1002)	AM008010
WRITE(6,1003)	AM008020
WRITE(6,1004)	AM008030
WRITE(6,1005)	AM008040
WRITE(6,1006)	AM008050
DO 1207 J2=1,NBEAMS	AM008060

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1207 WRITE(6,1009) J2,(POSAT(J2,J4),J4=1,J3)
      WRITE(6,1010)
      WRITE(6,1003)
      WRITE(6,1004)
      WRITE(6,1005)
      WRITE(6,1007)
      DO 1208 J2=1,NBEAMS
1208 WRITE(6,1009) J2,(POSLN(J2,J4),J4=1,J3)
      IF(KAXT.EQ.0) GO TO 1260
      WRITE(6,1011)
      WRITE(6,1003)
      WRITE(6,1004)
      WRITE(6,1005)
      WRITE(6,1006)
      DO 1209 J2=1,NBEAMS
1209 WRITE(6,1009) J2,(PCSAX(J2,J4),J4=1,J3)
1260 WRITE(6,1012)
      WRITE(6,1013)
      WRITE(6,1012)
      WRITE(6,1014)
      WRITE(6,1012)
      WRITE(6,1015)
      WRITE(6,1005)
      WRITE(6,1008)
      IF(KAXT.EQ.0.OR.A3.EQ.BK) GO TO 1265
      DO 1245 J2=1,NBEAMS
      WRITE(6,1016) J2,ZMMAST(J2),DISTAT(J2),ZMMASL(J2),DISTAL(J2),
1 ZMMAST(J2),DISTAX(J2)
1245 CONTINUE
      GO TO 1275
1265 IF(KAXT.EQ.0) GO TO 1270
      DO 1280 J2=1,NBEAMS
      WRITE(6,1018) J2,ZMMAST(J2),DISTAX(J2)
1280 CONTINUE
      GO TO 1275
1270 IF(A3.EQ.BK) GO TO 1275
      DO 1285 J2=1,NBEAMS
      WRITE(6,1019) J2,ZMMAST(J2),DISTAT(J2),ZMMASL(J2),DISTAL(J2)
1285 CONTINUE
1275 WRITE(6,1012)

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AM008070
AM008080
AM008090
AM008100
AM008110
AM008120
AM008130
AM008140
AM008150
AM008160
AM008170
AM008180
AM008190
AM008200
AM008210
AM008220
AM008230
AM008240
AM008250
AM008260
AM008270
AM008280
AM008290
AM008300
AM008310
AM008320
AM008330
AM008340
AM008350
AM008360
AM008370
AM008380
AM008390
AM008400
AM008410
AM008420
AM008430
AM008440
AM008450
AM008460

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WRITE(6,1017)
1002 FORMAT('1', 129('**'),/, ' **',50X,'LATERAL POSITION OF LOADS',52X, AM008470
1' **',/,1X,'**',48X,'FOR MAXIMUM MOMENT AT MIDSPAN',50X,'**',/,1X, AM008480
1129('**'),///,1X,57(' '), ' AASHTO TRUCK ',58(' '),/,21X,'(POSITION AM008490
1OF TRUCK WHEELS IN LOADED LANES WITH RESPECT TO C.G. OF BEAM NO.1 AM008500
1= IN FT.)',/,) AM008510
1003 FORMAT(' ',12X,6(' '), 'LANE 1',6(' '),1X,6(' '), 'LANE 2',6(' '), AM008520
11X,6(' '), 'LANE 3',6(' '),1X,6(' '), 'LANE 4',6(' '),1X,6(' '), AM008530
1'LANE 5',6(' ')) AM008540
1004 FORMAT(' ',14X,'TO LEFT TO RIGHT TO LEFT TO RIGHT TO LEFT', AM008550
1' TO RIGHT TO LEFT TO RIGHT TO LEFT TO RIGHT') AM008560
1005 FORMAT(' ',4X,'BEAM NO. ') AM008570
1006 FORMAT('++',16X,5('WHEEL WHEEL '),/,) AM008580
1007 FORMAT('++',13X,10(' EDGE '),/,) AM008590
1008 FORMAT('++',15X,'(KIP=FT.) TRUCK APPLIED (KIP=FT.) LANE APPLAM008600
1IED (KIP=FT.) TRAIN APPLIED',/,) AM008610
1009 FORMAT('0',6X,I2,2X,F9.1,F11.1,F9.1,F11.1,F9.1,F11.1,F9.1,F11.1, AM008620
1F9.1,F11.1) AM008630
1010 FORMAT(' ',///,1X,57(' '), ' AASHTO LANE ',59(' '),/,20X, AM008640
1'(POSITION OF LANE LOADING WITHIN LOADED LANE WITH RESPECT TO C.G. AM008650
1 OF BEAM NO.1 = IN FT.)',/,) AM008660
1011 FORMAT(' ',///,1X,57(' '), ' AXLE TRAIN ',60(' '),/,20X, AM008670
1'(POSITION OF AXLE TRAIN WHEELS IN LOADED LANES WITH RESPECT TO C. AM008680
1G. OF BEAM NO.1 = IN FT.)',/,) AM008690
1012 FORMAT(' ',1X,130('**')) AM008700
1013 FORMAT(//) AM008710
1014 FORMAT(' ', ' **',51X,'MAXIMUM MIDSPAN MOMENTS AND',50X,'**',/,2X,'**' AM008720
1,51X,'LATERAL DISTRIBUTION FACTORS',49X,'**') AM008730
1015 FORMAT(' ',/,15X,'MOMENT FROM FRACTION OF ',2('MOMENT FROM AM008740
1FRACTION OF '),/,15X,'AASHTO TRUCK FULL AASHTO AASHTO LANE AM008750
1 FULL AASHTO AXLE TRAIN FULL AXLE') AM008760
1016 FORMAT('0',7X,I2,5X,F8.1,8X,F6.3,7X,F8.1,7X,F6.3,8X,F8.1,7X,F6.3) AM008770
1017 FORMAT('1') AM008780
1018 FORMAT('0',7X,I2,63X,F8.1,7X,F6.3) AM008790
1019 FORMAT('0',7X,I2,5X,F8.1,8X,F6.3,7X,F8.1,7X,F6.3) AM008800
RETURN AM008810
END AM008820
AM008830

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SUBROUTINE INFLN
  INTEGER*2 TITLE,A1,A2,HINGTP
  COMMON/BLK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),
  *ZSA(20),YSA(20),BMJ(20),YH(10,2),ZH(10,2),NTY(30),
  *LC( 500),NLB( 500),LM(4),P( 500),XL( 500),PL( 500),
  *ECC( 500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),
  *NPOS( 500),XPCS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),
  *NOUT(9),YSSS(9,4),ZSSS(9,4),X(8)
  COMMON/BLK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYPES,NEQ,
  *ALPHA,NLCASE,NLCARD,NX
  COMMON/BLK 3/ XZERO(164),YMAMPL(20),ZMAMPL(20),
  *FAMPL(20),SFAMPL(20),TAMPL(20),DAMPL(20),YMOUT( 200,20,9),
  *ZMOUT(9,20,9),FOUT(9,20,9),SFOUT(9,20,9),TCUT(9,20,9),
  *DOUT(9,20,9),XOUT(9,76,9),XS(4),XA(4)
  COMMON/BLK 4/ TITLE(3,54),HINGTP(19,4),PWHEEL(18),ZNWHL(18),
  *DIST(18),WHLWNM(18),TLN(5,2),KASAST(5,2),KASASL(5,2),
  *KASAXT(5,2),ZMAST(20,5),ZMASL(20,5),ZMAXT(20,5),ZMMAST(20),
  *ZMMASL(20),ZMMAXT(20),POSAT(20,10),POSLN(20,10),POSAX(20,10),
  *NLLAST(20,5),NLLALN(20,5),NLLAXT(20,5),DISTAT(20),
  *DISTAL(20),DISTAX(20)
  COMMON/BLK 5/ NWHLS,NWHEEL,A1,A2,KAXT,FULMAT,FULMAL,FULMAX,PMAT,PMAM
  *AXT,NTRFL,NAXCL,NAXTSP,NAXT,JTYPES,A3,A4,ZFAN
  DATA BK/' '/
  DIMENSION ZZ(5),IZ(5)
  ZIMP=50./((125.+SPAN/12.))
  IF(ZIMP.GT.0.3) ZIMP=0.3
  FULMAT=FULMAT*(1.+ZIMP)
  FULMAL=FULMAL*(1.+ZIMP)
  FULMAX=FULMAX*(1.+ZIMP)
  DO 200 J1=1,NLCASE
  DO 200 J2=1,NBEAMS
200 YMOUT(J1,J2,1)=YMOUT(J1,J2,1)*(1.+ZIMP)
C
C   ADJUST YMOUT TO CLOSE EQUILIBRIUM OF MOMENTS GAP, SHOULD IT EXIST
C
  IF(A3.EQ.BK) GO TO 254
  JSTRT=KASAST(1,1)
  JSTP=KASAST(NTRFL,2)
  DO 214 J1=JSTRT,JSTP
  SUM=0.

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AM008840
 AM008850
 AM008860
 AM008870
 AM008880
 AM008890
 AM008900
 AM008910
 AM008920
 AM008930
 AM008940
 AM008950
 AM008960
 AM008970
 AM008980
 AM008990
 AM009000
 AM009010
 AM009020
 AM009030
 AM009040
 AM009050
 AM009060
 AM009070
 AM009080
 AM009090
 AM009100
 AM009110
 AM009120
 AM009130
 AM009140
 AM009150
 AM009160
 AM009170
 AM009180
 AM009190
 AM009200
 AM009210
 AM009220
 AM009230

DO 204 J2=1,NBEAMS
 204 SUM=SUM+YMOUT(J1,J2,1)
 SUM=SUM/(0.5*FULMAT)
 DO 206 J2=1,NBEAMS
 206 YMOUT(J1,J2,1)=YMOUT(J1,J2,1)/SUM
 214 CONTINUE
 JSTRT=KASASL(1,1)
 JSTP=KASASL(NTRFL,2)
 DO 252 J1=JSTRT,JSTP
 SUM=0.
 DO 248 J2=1,NBEAMS
 248 SUM=SUM+YMOUT(J1,J2,1)
 SUM=SUM/(0.1*FULMAL)
 DO 253 J2=1,NBEAMS
 253 YMOUT(J1,J2,1)=YMOUT(J1,J2,1)/SUM
 252 CONTINUE
 254 IF(KAXT.EQ.0) GO TO 264
 JSTRT=KASAXT(1,1)
 JSTP=KASAXT(NTRFL,2)
 DO 261 J1=JSTRT,JSTP
 SUM=0.
 DO 256 J2=1,NBEAMS
 256 SUM=SUM+YMOUT(J1,J2,1)
 SUM=SUM/(0.5*FULMAX)
 DO 258 J2=1,NBEAMS
 258 YMOUT(J1,J2,1)=YMOUT(J1,J2,1)/SUM
 261 CONTINUE
 264 CONTINUE
 DO 202 J1=1,5
 DO 202 J2=1,NBEAMS
 NLLAST(J2,J1)=0
 NLLALN(J2,J1)=0
 NLLAXT(J2,J1)=0
 ZMAST(J2,J1)=0.
 ZMASL(J2,J1)=0.
 202 ZMAXT(J2,J1)=0.
 C
 C DETEMINE MAXIMUM MOMENTS DUE TO AASHTO TRUCK
 C
 IF(A1.EQ.BK.ANC.A2.EQ.BK) GO TO 220

AM009240
 AM009250
 AM009260
 AM009270
 AM009280
 AM009290
 AM009300
 AM009310
 AM009320
 AM009330
 AM009340
 AM009350
 AM009360
 AM009370
 AM009380
 AM009390
 AM009400
 AM009410
 AM009420
 AM009430
 AM009440
 AM009450
 AM009460
 AM009470
 AM009480
 AM009490
 AM009500
 AM009510
 AM009520
 AM009530
 AM009540
 AM009550
 AM009560
 AM009570
 AM009580
 AM009590
 AM009600
 AM009610
 AM009620
 AM009630

DO 208 LANE=1,NTRFL	AM009640
JST=KASAST(LANE,1)+2	AM009650
JSP=KASAST(LANE,2)-6=2	AM009660
IF(JSP.LT.JST) WRITE(6,249) LANE	AM009670
IF(JSP.LT.JST) STOP	AM009680
249 FORMAT(/,10X,'****LANE NO. ',I2,' WILL NOT HOLD AASHTO TRUCK****')	AM009690
*)	AM009700
DO 208 J1=1,NBEAMS	AM009710
DO 208 J2=JST,JSP	AM009720
SUM=YMOUT(J2,J1,1)+YMOUT(J2+6,J1,1)	AM009730
IF(SUM.GT.ZMAST(J1,LANE)) POSAT(J1,2*LANE-1)=TLN(LANE,1)+2+(J2-JST)	AM009740
*)	AM009750
IF(SUM.GT.ZMAST(J1,LANE)) POSAT(J1,2*LANE)=POSAT(J1,2*LANE-1)+6.	AM009760
IF(SUM.GT.ZMAST(J1,LANE)) ZMAST(J1,LANE)=SUM	AM009770
J6 = J2+6	AM009780
208 CONTINUE	AM009790
C	AM009800
C DETERMINE MAXIMUM MOMENTS DUE TO AASHTO LANE LOAD	AM009810
C	AM009820
DO 212 LANE=1,NTRFL	AM009830
JST=KASASL(LANE,1)	AM009840
JSP=KASASL(LANE,2)-10	AM009850
IF(JSP.LT.JST) WRITE(6,251)LANE	AM009860
IF(JSP.LT.JST) STOP	AM009870
251 FORMAT(/,10X,'****LANE NO. ',I2,' LESS THAN 10 FT WIDE****')	AM009880
DO 212 J1=1,NBEAMS	AM009890
DO 212 J2=JST,JSP	AM009900
SUM=(YMOUT(J2,J1,1)+YMOUT(J2+10,J1,1))*0.5	AM009910
DO 210 J3=1,9	AM009920
210 SUM=SUM+YMOUT(J2+J3,J1,1)	AM009930
IF(SUM.GT.ZMASL(J1,LANE)) POSLN(J1,2*LANE-1)=TLN(LANE,1)+(J2-JST)	AM009940
IF(SUM.GT.ZMASL(J1,LANE)) POSLN(J1,2*LANE)=POSLN(J1,2*LANE-1)+10.	AM009950
IF(SUM.GT.ZMASL(J1,LANE)) ZMASL(J1,LANE)=SUM	AM009960
212 CONTINUE	AM009970
IF(KAXT.EQ.0) GO TO 224	AM009980
C	AM009990
C DETERMINE MAXIMUM MOMENTS DUE TO AXLE TRAIN	AM010000
C	AM010010
220 DO 222 LANE=1,NTRFL	AM010020
JST=KASAXT(LANE,1)+NAXCL	AM010030

JSP=KASAXT(LANE,2)=NAXTSP=NAXCL	AM010040
IF(JSP.LT.JST) WRITE(6,255)LANE	AM010050
255 FORMAT(//,10X,'LANE NO. ',I2,' WILL NOT HOLD AXLE TRAIN VEHICLE***	AM010060
***)	AM010070
IF(JSP.LT.JST) STCP	AM010080
DO 222 J1=1,NBEAMS	AM010090
DO 222 J2=JST,JSP	AM010100
SUM=YMOUT(J2,J1,1)+YMOUT(J2+NAXTSP ,J1,1)	AM010110
IF(SUM.GT.ZMAXT(J1,LANE)) POSAX(J1,2*LANE-1)=TLN(LANE,1)+(J2-JST)	AM010120
* +NAXCL	AM010130
IF(SUM.GT.ZMAXT(J1,LANE)) POSAX(J1,2*LANE)=POSAX(J1,2*LANE-1)+NAXT	AM010140
*SP	AM010150
IF(SUM.GT.ZMAXT(J1,LANE)) ZMAXT(J1,LANE)=SUM	AM010160
J6=J2+NAXTSP	AM010170
222 CONTINUE	AM010180
C	AM010190
C COMPUTE TOTAL MOMENT ON BEAMS FROM AASHTO TRUCK	AM010200
C	AM010210
224 IF(A1.EQ.BK.AND.A2.EQ.BK) GO TO 242	AM010220
DO 232 JB=1,NBEAMS	AM010230
DO 226 J1=1,5	AM010240
226 ZZ(J1)=0.	AM010250
DO 228 J1=1,NTRFL	AM010260
228 ZZ(J1)=ZMAST(JB,J1)	AM010270
CALL SORT(ZZ,IZ)	AM010280
SUM1=ZMAST(JB,IZ(1))	AM010290
SUM2=SUM1+ZMAST(JB,IZ(2))	AM010300
SUM3=(SUM2+ZMAST(JB,IZ(3)))*0.9	AM010310
SUM4=(SUM2+ZMAST(JB,IZ(3))+ZMAST(JB,IZ(4)))*0.75	AM010320
SUM5=(SUM2+ZMAST(JB,IZ(3))+ZMAST(JB,IZ(4))+ZMAST(JB,IZ(5)))*0.75	AM010330
ZMAST(JB)=AMAX1(SUM1,SUM2,SUM3,SUM4,SUM5)	AM010340
ZZ(1)=SUM1	AM010350
ZZ(2)=SUM2	AM010360
ZZ(3)=SUM3	AM010370
ZZ(4)=SUM4	AM010380
ZZ(5)=SUM5	AM010390
DO 265 J1=1,5	AM010400
IF(ZZ(J1).NE.ZMAST(JB)) GO TO 265	AM010410
DO 263 J2=1,J1	AM010420
263 NLLAST(JB,IZ(J2))=1	AM010430

GO TO 232	AM010440
265 CONTINUE	AM010450
232 CONTINUE	AM010460
C	AM010470
C COMPUTE TOTAL MOMENT ON BEAMS FROM AASHTO LANE	AM010480
C	AM010490
DO 240 JB=1,NBEAMS	AM010500
DO 234 J1=1,5	AM010510
234 ZZ(J1)=0.	AM010520
DO 236 J1=1,NTRFL	AM010530
236 ZZ(J1)=ZMASL(JB,J1)	AM010540
CALL SORT(ZZ,IZ)	AM010550
SUM1=ZMASL(JB,IZ(1))	AM010560
SUM2=ZMASL(JB,IZ(2))+SUM1	AM010570
SUM3=(SUM2+ZMASL(JB,IZ(3)))*0.9	AM010580
SUM4=(SUM2+ZMASL(JB,IZ(3))+ZMASL(JB,IZ(4)))*0.75	AM010590
SUM5=(SUM2+ZMASL(JB,IZ(3))+ZMASL(JB,IZ(4))+ZMASL(JB,IZ(5)))*0.75	AM010600
ZMMASL(JB)=AMAX1(SUM1,SUM2,SUM3,SUM4,SUM5)	AM010610
ZZ(1)=SUM1	AM010620
ZZ(2)=SUM2	AM010630
ZZ(3)=SUM3	AM010640
ZZ(4)=SUM4	AM010650
ZZ(5)=SUM5	AM010660
DO 269 J1=1,5	AM010670
IF(ZZ(J1).NE.ZMMASL(JB)) GO TO 269	AM010680
DO 267 J2=1,J1	AM010690
267 NLLALN(JB,IZ(J2))=1	AM010700
GO TO 240	AM010710
269 CONTINUE	AM010720
240 CONTINUE	AM010730
C	AM010740
C COMPUTE TOTAL MOMENT ON BEAMS FROM AXLE TRAIN	AM010750
C	AM010760
242 IF(KAXT.EG.0) GO TO 260	AM010770
DO 250 JB=1,NBEAMS	AM010780
SUM2=0.	AM010790
SUM3=0.	AM010800
SUM4=0.	AM010810
SUM5=0.	AM010820
DO 244 J1=1,5	AM010830

244	ZZ(J1)=0.	AM010840
	DO 246 J1=1,NTRFL	AM010850
246	ZZ(J1)=ZMAXT(JB,J1)	AM010860
	CALL SORT(ZZ,IZ)	AM010870
	SUM1=ZMAXT(JB,IZ(1))	AM010880
	IF(NAXT.EQ.1) GO TO 289	AM010890
	SUM2=ZMAXT(JB,IZ(2))+SUM1	AM010900
	IF(NAXT.EQ.2) GO TO 289	AM010910
	SUM3=SUM2+ZMAXT(JB,IZ(3))	AM010920
	IF(NAXT.EQ.3) GO TO 289	AM010930
	SUM4=SUM3+ZMAXT(JB,IZ(4))	AM010940
	IF(NAXT.EQ.4) GO TO 289	AM010950
	SUM5=SUM4+ZMAXT(JB,IZ(5))	AM010960
289	ZMMAXT(JB)=AMAX1(SUM1,SUM2,SUM3,SUM4,SUM5)	AM010970
	ZZ(1)=SUM1	AM010980
	ZZ(2)=SUM2	AM010990
	ZZ(3)=SUM3	AM011000
	ZZ(4)=SUM4	AM011010
	ZZ(5)=SUM5	AM011020
	DO 277 J1=1,5	AM011030
	IF(ZZ(J1).NE.ZMMAXT(JB)) GO TO 277	AM011040
	DO 273 J2=1,J1	AM011050
273	NLLAXT(JB,IZ(J2))=1	AM011060
	GO TO 250	AM011070
277	CONTINUE	AM011080
250	CONTINUE	AM011090
C		AM011100
260	CONTINUE	AM011110
	DO 40 J1=1,NBEAMS	AM011120
40	DISTAT(J1)=ZMMAST(J1)/FULMAT	AM011130
	DO 42 J1=1,NBEAMS	AM011140
42	DISTAL(J1)=ZMMASL(J1)/FULMAL	AM011150
	IF(KAXT.EQ.0) GO TO 46	AM011160
	DO 44 J1=1,NBEAMS	AM011170
44	DISTAX(J1)=ZMMAXT(J1)/FULMAX	AM011180
46	CONTINUE	AM011190
	RETURN	AM011200
	END	AM011210

```

SUBROUTINE FLEX
COMMON/BLK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),
*ZSA(20),YSA(20),BMJ(20),YH(10,2),ZH(10,2),NTY(30),
*LC( 500),NLB( 500),LM(4),P( 500),XL( 500),PL( 500),
*ECC( 500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),
*NPOS( 500),XPOS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),
*NOOT(9),YSSS(9,4),ZSSS(9,4),X(8)
COMMON/BLK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYPES,NEQ,
*ALPHA,NLCASE,NLCARD,NX

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```

C
C   SUBROUTINE FLEX
C
C   ZERO STRUCTURE FLEXIBILITY MATRIX, SF.
C
C   NEQ = 4*(NBEAMS + 1)
C   DO 250 I = 1, NEQ
C   DO 250 J = 1, 8
250 SF(I,J) = 0.0
C
C   FORM FLEXIBILITY MATRICES FOR EACH BEAM TYPE
C
C   CALL BMFLEX (1, 1, 1, 1, F11)
C
C   CALL BMFLEX (1, 1, 2, 2, F12)
C
C   CALL BMFLEX (2, 2, 2, 2, F22)
C
C   STORE BEAM FLEXIBILITY INTO STRUCTURE FLEX. FOR ALL BEAMS
C
C   DO 400 N = 1, NBEAMS
C   NT = NTY(N)
C
C   STORE F11 INTO SF
C
C   NN = 4*(N - 1)
C   DO 320 I = 1, 4
320 LM(I) = NN + I
C   DO 340 I = 1, 4
C   II = LM(I)
C   DO 340 J = I, 4

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AM011220
AM011230
AM011240
AM011250
AM011260
AM011270
AM011280
AM011290
AM011300
AM011310
AM011320
AM011330
AM011340
AM011350
AM011360
AM011370
AM011380
AM011390
AM011400
AM011410
AM011420
AM011430
AM011440
AM011450
AM011460
AM011470
AM011480
AM011490
AM011500
AM011510
AM011520
AM011530
AM011540
AM011550
AM011560
AM011570
AM011580
AM011590
AM011600
AM011610

```



```

      JJ = LM(J) - II + 1
340 SF(II,JJ) = SF(II,JJ) + F11(NT,I,J)
C
C   STORE F12 INTO SF
C
      DO 360 I = 1, 4
      II = LM(I)
      DO 360 J = 1, 4
      JJ = LM(J) + 5 - II
360 SF(II,JJ) = SF(II,JJ) - F12(NT,I,J)
C
C   STORE F22 INTO SF
C
      DO 380 I = 1, 4
      II = LM(I) + 4
      DO 380 J = 1, 4
      JJ = LM(J) + 5 - II
380 SF(II,JJ) = SF(II,JJ) + F22(NT,I,J)
400 CONTINUE
C
C   ADD HINGE FLEXIBILITIES ALONG DIAGONAL OF SF
C
      NJ = NBEAMS - 1
      DO 450 J = 1, NJ
      JT = JTY(J)
      DO 450 N = 1, 4
      I = 4*J + N
      SF(I,1) = SF(I,1) + HF(JT,N)
450 CONTINUE
C
      RETURN
C
      END

```

```

AM011620
AM011630
AM011640
AM011650
AM011660
AM011670
AM011680
AM011690
AM011700
AM011710
AM011720
AM011730
AM011740
AM011750
AM011760
AM011770
AM011780
AM011790
AM011800
AM011810
AM011820
AM011830
AM011840
AM011850
AM011860
AM011870
AM011880
AM011890
AM011900
AM011910
AM011920
AM011930
AM011940

```

```

SUBROUTINE BMFLEX (II, JJ, KK, LL, A)
COMMON/ELK 1/ HEAD(15),ZMI(20),YMI(20),BMA(20),
*ZSA(20),YSA(20),RMJ(20),YH(10,2),ZH(10,2),NTY(30),
*LC( 500),NLB( 500),LM(4),P( 500),XL( 500),PL( 500),
*ECC( 500),F11(10,4,4),F12(10,4,4),F22(10,4,4),SF(164,8),
*NPOS( 500),XPOS(9,9),HF(19,4),JTY(30),YSTR(9,4),ZSTR(9,4),
*NOUT(9),YSSS(9,4),ZSSS(9,4),X(8)
COMMON/ELK 2/ NHARMS,SPAN,NBEAMS,E,G,NTYPES,NEQ,
*ALPHA,NLCASE,NLCARD,NX
DIMENSION A(10,4,4)

```

FORM BEAM FLEXIBILITY ABOUT HINGES

```

DO 10 K = 1, NTYPES
F1 = 1./ (ALPHA**2*BMA(K))
F2 = 1./ (ALPHA**4*ZMI(K))
F3 = 1./ (ALPHA**4*YMI(K))
F4 = 1./ (ALPHA**2*BMJ(K))

Y1 = YH(K,II)*ALPHA
Z1 = ZH(K,JJ)*ALPHA
Y2 = YH(K,KK)*ALPHA
Z2 = ZH(K,LL)*ALPHA

A(K,1,1) = F1 + Y1*Y2*F2 + Z1*Z2*F3
A(K,1,2) = -Y1*F2
A(K,1,3) = -Z1*F3
A(K,1,4) = 0.0
A(K,2,1) = -Y2*F2
A(K,2,2) = F2 + ZH(K,JJ)*ZH(K,LL)*F4
A(K,2,3) = -ZH(K,JJ)*YH(K,KK)*F4
A(K,2,4) = -ZH(K,JJ)*F4
A(K,3,1) = -Z2*F3
A(K,3,2) = -YH(K,II)*ZH(K,LL)*F4
A(K,3,3) = F3 + YH(K,II)*YH(K,KK)*F4
A(K,3,4) = YH(K,II)*F4
A(K,4,1) = 0.0
A(K,4,2) = -ZH(K,LL)*F4
A(K,4,3) = YH(K,KK)*F4

```

AM011950
 AM011960
 AM011970
 AM011980
 AM011990
 AM012000
 AM012010
 AM012020
 AM012030
 AM012040
 AM012050
 AM012060
 AM012070
 AM012080
 AM012090
 AM012100
 AM012110
 AM012120
 AM012130
 AM012140
 AM012150
 AM012160
 AM012170
 AM012180
 AM012190
 AM012200
 AM012210
 AM012220
 AM012230
 AM012240
 AM012250
 AM012260
 AM012270
 AM012280
 AM012290
 AM012300
 AM012310
 AM012320
 AM012330
 AM012340

A(K,4,4) = F4
10 CONTINUE
C
RETURN
C
END

AM012350
AM012360
AM012370
AM012380
AM012390
AM012400

```

SUBROUTINE LOCATE(Y,N,EC,NBEAMS,YH,NTY)
  DIMENSION YH(10,2),NTY(30)
  SUM=0.
  DO 20 J1=1,NBEAMS
    N=J1
    IF(SUM+YH(NTY(J1),1).LE.Y.AND.SUM+YH(NTY(J1),2).GE.Y) GO TO 22
20  SUM=SUM+YH(NTY(J1),2)-YH(NTY(J1+1),1)
22  EC=Y-SUM
    RETURN
  END
SUBROUTINE SORT(ZZ,IZ)
  DIMENSION ZZ(5),IZ(5)
  DO 100 J1=1,5
    S1=ZZ(1)
    S2=ZZ(2)
    S3=ZZ(3)
    S4=ZZ(4)
    S5=ZZ(5)
    ZMAX=AMAX1(S1,S2,S3,S4,S5)
    DO 20 J2=1,5
      IF(ZMAX.NE.ZZ(J2)) GO TO 20
      IZ(J1)=J2
      ZZ(J2)=-10.*(10-J1)
    GO TO 100
20  CONTINUE
100 CONTINUE
  RETURN
  END

```

```

AM012410
AM012420
AM012430
AM012440
AM012450
AM012460
AM012470
AM012480
AM012490
AM012500
AM012510
AM012520
AM012530
AM012540
AM012550
AM012560
AM012570
AM012580
AM012590
AM012600
AM012610
AM012620
AM012630
AM012640
AM012650
AM012660
AM012670
AM012680

```

```

SUBROUTINE MAXMCM(N,P,XSPC,SPAN,ZMAX)
DIMENSION P(18),XSPC(18),ZSPC(18),XSTORE(18)
XSPC(1)=0.
DO 10 J1=1,N
10 ZSPC(J1)=XSPC(J1)
ZMAX=0.
DO 30 J1=1,N
XSPC(1)=SPAN/2.+ZSPC(J1)
DO 12 J2=2,N
12 XSPC(J2)=XSPC(1)-ZSPC(J2)
ZM=0.
DO 20 J2=1,N
IF(0.LE.XSPC(J2).AND.XSPC(J2).LE.SPAN) GO TO 14
GO TO 20
14 ZM=ZM+(SPAN-XSPC(J2))*P(J2)/2.
IF(XSPC(J2).LT.SPAN/2.) ZM=ZM-P(J2)*(SPAN/2.-XSPC(J2))
20 CONTINUE
IF(ZM.LE.ZMAX) GO TO 30
ZMAX=ZM
DO 24 J2=1,N
24 XSTORE(J2)=XSPC(J2)
30 CONTINUE
JSTRT=1
JSTOP=N
DO 36 J1=1,N
IF(XSTORE(J1).GE.SPAN) JSTRT=JSTRT+1
IF(XSTORE(J1).LE.0.) JSTOP=JSTOP-1
36 CONTINUE
N=0
DO 38 J1=JSTRT,JSTOP
N=N+1
P(N)=P(J1)
38 XSPC(N)=XSTORE(J1)
RETURN
END

```

```

AM012690
AM012700
AM012710
AM012720
AM012730
AM012740
AM012750
AM012760
AM012770
AM012780
AM012790
AM012800
AM012810
AM012820
AM012830
AM012840
AM012850
AM012860
AM012870
AM012880
AM012890
AM012900
AM012910
AM012920
AM012930
AM012940
AM012950
AM012960
AM012970
AM012980
AM012990
AM013000
AM013010
AM013020
AM013030

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