

Analysis Guidelines and Examples for Fracture Critical Steel Twin Tub Girder Bridges

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Twin steel tub girder bridges are an aesthetically pleasing structural option offering long span solutions in tight-radii direct connectors. However, these bridges require a routine two-year inspection frequency, as well as a thorough hands-on inspection because of their fracture critical designation. The heightened inspection requirements for fracture critical bridges come at a significant cost to the Texas Department of Transportation (TxDOT). Recent research has shown that tangent, or nearly tangent, twin steel tub girder sections can redistribute load to the intact girder after fracture of one of the girder bottom flanges. Additional research is required to develop recommendations for practical analysis of typical twin steel tub span configurations with the degree of curvature common to twin steel tub direct connectors. A finite element method (FEM), specifically Abaqus, and SAP2000 can be employed for both rigorous FEM and grillage solutions along with push-down plastic analysis typically available to TxDOT and its consultant bridge designers. These analysis and modeling methods take into account the capacity of the fractured girder, especially at the support locations, and realistically model the load distribution between the intact girder and the fractured girder. The modeling and analysis methods are incorporated in a straightforward manner on a large scale to the inventory of the steel tub bridges. The requirements outlined in the Federal Highway Administration memorandum dated June 12, 2012, and entitled "Clarification of Requirements for Fracture Critical Members" were met in the employed analysis methods.					
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ANALYSIS GUIDELINES AND EXAMPLES FOR FRACTURE CRITICAL STEEL TWIN TUB GIRDER BRIDGES

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DISCLAIMER

This research was performed in cooperation with the Texas Department of Transportation (TxDOT) and the Federal Highway Administration (FHWA). The contents of this report reflect the views of the authors, who are responsible for the facts and the accuracy of the data presented herein. The contents do not necessarily reflect the official view or policies of FHWA or TxDOT. This report does not constitute a standard, specification, or regulation.

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1. FOREWORD

The Federal Highway Administration (FHWA 2012) (Lwin 2012) defines a fracture critical member (FCM) as "a steel member in tension, or with a tension element, whose failure would probably cause a portion of or the entire bridge to collapse." FCMs are composed of nonredundant members that make the structure extremely vulnerable to fatigue and/or fracture failures. An engineer may encounter the dilemma of choosing either a steel twin tub girder (STTG) or long-span steel (or concrete) multi-girder superstructure. Despite the advantages that tub girders offer for long-span curved bridges, there may be a reluctance to select this option since it is deemed fracture critical by the above FHWA requirements. The classification of the structure as FCM implies significant ongoing inspection costs due to the hands-on nature of inspection requirements for FCMs. However, it is contended that the current classification based on load path redundancy alone may be markedly conservative if it does not consider the inherent reserve capacity of STTG bridges, especially those continuous structures with multiple spans.

If these bridges are reclassified as nonfracture critical based on the reserve capacity in a damaged state under the design loads, substantial savings in inspection costs will accrue over the life of the structure. At the time of design, certain simplifications are made whereby the degree of redundancy and reserve capacity are not recognized. The quantification of the reserve capacity of each twin tub bridge structure can potentially lead to the reassignment from the default position of a fracture critical bridge to a reclassification as nonfracture critical.

The finite element method (FEM) makes use of the advanced nonlinear elasto-plastic analysis incorporating a detailed mesh refinement and precise element modeling. The results generated are consequently the most accurate and reliable. A second but less rigorous computational method implements a lower-bound analysis method using a nonlinear push-down grillage analysis. This ease in the computational effort comes at the cost of a comparatively less precise simulation of the bridge material properties and loading conditions.

Recent research reported in Texas Department of Transportation (TxDOT) project 0-6937 suggests that many of the bridges may be safely reclassified as nonfracture critical based on the computed overstrength factors. Since this assessment ideally needs to be an economical and rapid process, the applicability of using the simpler computational method such as grillage, aided by the plastic upper and lower-bound solutions serving as a check, is examined in that project.

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The feasibility of these simper solutions is established through the comparison of the results obtained from the three independent methods. This volume primarily focuses on the two simpler methods that serve as the preliminary steps in the identification of the reserve capacity of the structure to determine whether the bridge could be reclassified as nonfracture critical.

In spite of the relative simplicity of the nonlinear grillage methods, there remains considerable time and effort to code and run such an analysis. Moreover, once complete, the analyst may remain unsure of the dependability of the result.

To confirm the dependability of results, the simplest approach is to conduct a limit plastic collapse analysis of the bridge deck in a span-by-span fashion. This limit analysis for a fractured twin tub bridge needs to be rooted in the use of yield line theory. As such, analytical solutions using yield line theories may be posed as either lower or upper-bound solutions to the problem (Park and Gamble 2000).

2. ANALYSIS SCHEMA

2.1 SCOPE

This section presents an overview of the analysis procedures that should ideally be followed so that an STTG bridge that is presumed to be fracture critical (meaning, by definition of FCMs as per FHWA, it contains a "steel member in tension, or with a tension element, whose failure would probably cause a portion of or the entire bridge to collapse") may be reclassified as nonfracture critical. The analysis schema shown in Figure 2.1 depicts the overall procedure and decision points.

In principle, two methods of analysis that are markedly different in their approach are used to arrive at a positive declassification decision. If the first method—the yield line theory that provides a limit analysis solution—does not indicate a positive solution, there is normally little hope the following methods will succeed. Therefore, the analysis may be terminated. The process of the evaluation of the STTG bridges for their declassification is charted using an algorithmic representation, as shown in Figure 2.1. The procedure is as follows:

2.2 COLLECTION AND ORGANIZATION OF DATA:

The necessary details of the bridge are documented by using the design drawings. The reports for the member properties and additional information and other pertinent attributes are systematically recorded as per the requirements of each method. For example, the yield line analysis does not model the 3-D elements, such as diaphragms and stiffeners, that the FEM model does; therefore, the details of those elements need not be documented for the first stage. Thus, a selective documentation of the data is encouraged for efficient collection of useful information.

2.3 STAGE 1: YIELD LINE ANALYSIS

Yield line analysis is an explicit direct method of plastic analysis used to provide an upper-bound solution for the reserve capacity of the bridge in terms of the overstrength factor, Ω . The first stage can be perceived as a litmus test for the reserve capacity of the bridge span under consideration since if this method concludes an overstrength, $\Omega_{Yield \ Line} < 1$, then the chances of

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the remaining two methods resulting in a higher reserve capacity are very slim, and the bridge will be deemed fracture critical. If the yield line analysis predicts an overstrength, $\Omega_{Yield\ Line} > 1$, then proceed to the second stage of analysis, which is computational.



Figure 2.1. Flowchart for Analysis Procedure of STTG Bridges.

2.4 STAGE 2: GRILLAGE ANALYSIS

The nonlinear push-down analysis is a computational method that may be carried out using reputable commercial software such as SAP2000 (Wilson and Habibullah 1997). The concept of the method is the strip method of plastic analysis that incorporates the modeling of structural components as equivalent grillage members whose capacity is assessed through hinge failure in an elasto-plastic analysis. The method generally leads to lower-bound solutions. However, if strain-hardening is used in the constitutive relations, solutions where $\Omega_{Grillage} > \Omega_{Yield Line}$ are possible; the lower of the two should be adopted. If $1 > \Omega_{Grillage} > \Omega_{Yield Line}$, the evaluation shall be terminated by classifying the bridge as fracture critical. In the event of an overstrength, $\Omega_{Grillage} > 1$, the analysis shall be taken to the next stage.

2.5 STAGE 3: FINITE ELEMENT METHOD

It is possible that following the Stage 2 analysis, a result may lead to $\Omega_{Yield\ Line} > 1$ and $\Omega_{Grillage} < 1$. This result does not necessarily mean the bridge cannot be reclassified as nonfracture critical; rather, a more exacting computational analysis is needed as a tie-breaker. Therefore, an advanced modeling and load simulation analysis should then be conducted using general-purpose nonlinear FEM software such as Abaqus (Dassault Systèmes 2017) to capture the complete behavior of the elements and their material and structural properties. This method is the most precise of all the three methods employed but is considerably more costly in terms of time and effort. Because this method is considered the most definitive, then if the overstrength factor $\Omega_{FEM} > 1$, the structure may be reclassified as nonfracture critical. Conservatively, if $\Omega_{FEM} < 1$, the bridge shall continue to be classified as fracture critical.

2.6 BACKGROUND THEORY AND EXAMPLES

Sections 3 and 4 provide background to yield line and grillage approaches, respectively. In Sections 5–7 of these guidelines, overstrength factor results respectively are presented for the following 3 bridges selected from the suite of 15 typical steel twin tub girder (STTG) bridges from the Texas bridge inventory that are described in the TxDOT (0-6937) report of Fracture Critical Steel Twin Tub Girder Bridges:

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- Single-span bridges: Bridge 2 is used as an example that has no support fixity.
- Two-span continuous bridges: Both exterior spans of Bridge 5 are used as examples with one degree of fixity over the interior support.
- Three-span or greater continuous bridges: Both exterior spans of Bridge 10, with one fixity over the interior support, and one interior span of Bridge 10, with fixity over both the interior supports, are provided as examples.

3. PLASTIC ANALYSIS USING YIELD LINE THEORIES

As described in Park and Gamble (2000), the overstrength factor is calculated using virtual work analysis by equating the factored external work to the internal work when a maximum deflection of unity takes place:

$$\Omega EWD_U = IWD_N \tag{3.1}$$

in which IWD_N = internal work done based on nominal material properties; EWD_U = external work done by factored ultimate design load; and Ω =overstrength factor.

The external work done is calculated as follows:

$$EWD_U = W_{ET}L_x^* \frac{\delta}{2} = 0.5W_{ET}L_x^*$$
(3.2)

in which W_{ET} = the total load that effectively participates in the collapse mechanism; $\delta = 1$ = virtual displacement; and L_x^* = the span length under consideration measured on the centerline (CL) of the collapse mechanism, such that:

$$L_{\chi}^* = \left(1 + \frac{B}{4R_{\Phi}}\right)L_{\chi} \tag{3.3}$$

where $L_x = CL$ of the bridge (midway between the twin tubs); B = the width of the bridge; and R_{Φ} = the radius of curvature measured along the CL of the bridge deck for a straight bridge $L_x^* = L_x$.

For AASHTO HL93 truck and lane loads, this results in:

$$W_{ET} = w_u L_x^* (b + 0.5s) + W_x L_x^* + \left(336 - \frac{523}{\lambda L_x^*} - \frac{2091}{(1 - \lambda)L_x^*}\right) K_{lane}$$
(3.4)

where s = the width of the area of the slab along which the mechanism under consideration is applied; b = the transverse distance of the interior flange of the fractured girder from the outer edge of the bridge; $w_u =$ the area load consisting of self-weight of the reinforced concrete deck slab and the applied lane load (kip/ft² = ksf); W_x = the line load consisting of the self-weight of the fractured tub girder and the guardrail (kip/ft); and λ = the critical location factor for the hinge to occur, normally at the location of maximum moments. For simply supported spans and the interior spans of three- or more-span continuous bridges, $\lambda = 0.5$, whereas for two-span bridges or the end span for three- or more-span bridges, $\lambda = 0.4$. Thus, for simply supported spans and for interior spans ($\lambda = 0.5$):

$$W_T = w_u L_x^* (b + 0.5s) + W_x L_x^* + \left(336 - \frac{5226}{L_x^*}\right) K_{lane}$$
(3.5)

For the end spans of continuous bridges (2 spans or greater, where $\lambda = 0.4$).

$$W_T = w_u L_x^* (b + 0.5s) + W_x L_x^* + \left(336 - \frac{4793}{L_x^*}\right) K_{lane}$$
(3.6)

For a wider bridge, the second lane of trucks may participate (in part) in the collapse mechanism, as depicted in Figure 3.1. The axle loads are required to be increased proportionally to their deflection with respect to the truck position over the fractured girder. Thus, the lane load requires modification through the scalar K_{lane} . For one line of truck wheels participating:

$$K_{lane} = 1 + 0.5 \frac{y}{s}$$
(3.7)

in which y = distance measured from the intact (unfractured) girder to the line of wheels. If both lines of wheels are participating in the mechanism, then:

$$K_{lane} = 1 + \frac{y}{s}; \qquad \qquad K_{lane} \le 2 \qquad (3.8)$$

where y = distance to the CL of the truck.

The internal work done is calculated as follows:

$$IWD_{N} = (m_{y}' + m_{y}) \left(\frac{L_{x}}{2s}\right) k_{bound} + \left(\frac{m_{x}b}{\lambda(1-\lambda)L_{x}^{*}}\right) + \left(\frac{0.5M_{p1}^{-}}{(1-\lambda)L_{x}^{*}}\right) + \left(\frac{0.5M_{p2}^{-}}{(1-\lambda)L_{x}^{*}}\right)$$
(3.9)

where m'_y and m_y are the negative and positive moment capacities per unit width in the y direction, respectively, and m'_x , and m_x are the negative and positive moment capacities per unit width in the x direction, respectively (units k-in./in. = k-ft/ft, or kN m/m or N-mm/mm); M_{p1}^{-1} and M_{p2}^{-} are the plastic moment capacities of the composite deck and the intact girders at the ends of the span in consideration (0.5 is used since the outside girder alone takes part in the critical mechanism); and λ = the critical location factor for the hinge to occur, normally at the location of maximum moments, as defined above.



Figure 3.1. Layout of Typical Interior Span with Yield Line Mechanism.

Note that for simply supported spans, there is no end fixity; therefore, M_{p1}^- and M_{p2}^- are set to zero, whereas exterior spans of the two-span and of the three-span bridges have one fixity at the end support; therefore, one of the M_{p1}^- and M_{p2}^- are set to zero for that case, and interior spans have fixity at both the end, implying that both M_{p1}^- and M_{p2}^- are non-zero.

For simply supported spans and the interior spans of three-or-more-span continuous bridges ($\lambda = 0.5$):

$$IWD_{Simple Spans} = \left(m'_{y} + m_{y}\right) \left(\frac{L_{x}}{2s}\right) k_{bound} + \frac{4}{L_{x}^{*}}(m_{x}b)$$
(3.10)

$$IWD_{Int.Spans} = (m'_{y} + m_{y}) \left(\frac{L_{x}}{2s}\right) k_{bound} + \left(\frac{4m_{x}b}{L_{x}^{*}}\right) + \left(0.5M_{p1}^{-} + 0.5M_{p2}^{-}\right) \left(\frac{2}{L_{x}^{*}}\right)$$
(3.11)

For two-span bridges or the end-span for three- or more-span bridges ($\lambda = 0.4$):

$$IWD_{Ext.Spans} = (m'_{y} + m_{y}) \left(\frac{L_{x}}{2s}\right) k_{bound} + \left(\frac{25m_{x}b}{6L_{x}}\right) + \left(\frac{25M_{p1}}{12L_{x}^{*}}\right)$$
(3.12)

The term k_{bound} represents the modifier term representing the upper and lower-bound solutions, as follows:

$$k_{bound}^{upper} = \left[1 + 2\frac{\tan\alpha}{\tan\theta}\right] = 1 + \frac{4s}{L_x} \sqrt{\left(\frac{m'_x + m_x}{m'_y + m_y}\right)}$$
(3.13)

and:

$$k_{bound}^{lower} = \left[1 + 2\frac{\tan^2\alpha}{\tan^2\theta}\right] = 1 + \frac{8s^2}{L_x^2} \left(\frac{m'_x + m_x}{m'_y + m_y}\right)$$
(3.14)

where:

$$\tan \alpha = \frac{s}{L_x/2} \tag{3.15}$$

and:

$$\tan \theta = \frac{sN}{L_x} = \sqrt{\frac{m'_y + m_y}{m'_x + m_x}}$$
(3.16)

The overstrength factors are computed using Equation (3.1).

$$\Omega_{Yield\ Line} = \frac{IWD_N}{EWD_U} = \frac{\left(m'_y + m_y\right)\left(\frac{L_x}{2s}\right)k_{bound} + \left(\frac{m_x b}{\lambda(1-\lambda)L_x^*}\right) + \left(\frac{0.5M_{p1}^- + 0.5M_{p2}^-}{2L_x^*(1-\lambda)}\right)}{0.5W_{ET}}$$
(3.17)

For simply supported spans:

$$\Omega_{Simply Supported} = \frac{\left(m'_{y} + m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound} + \frac{4}{L_{x}^{*}}(m_{x}b)}{0.5W_{ET}}$$
(3.18)

For interior spans:

$$\Omega_{Interior} = \frac{\left(m'_{y} + m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound} + \left(\frac{4m_{x}b}{L_{x}^{*}}\right) + \left(0.5M_{p1}^{-} + 0.5M_{p2}^{-}\right)\left(\frac{2}{L_{x}^{*}}\right)}{0.5W_{ET}}$$
(3.19)

For exterior spans:

$$\Omega_{Exterior} = \frac{\left(m'_{y} + m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound} + \left(\frac{25m_{x}b}{6L_{x}}\right) + \left(\frac{25M_{p1}}{12L_{x}^{*}}\right)}{0.5W_{ET}}$$
(3.20)

The longitudinal and transverse moment (positive and negative) capacities of the deck slab and the positive capacities of the composite intact section are computed based on the standard U.S. code procedure using the specified compressive strength of concrete and specified or as-built (if known) yield strength of reinforcing steel in the deck and the guardrail and in the structural steel of the twin tub girders. The negative capacities of the composite intact section are computed using plastic analysis of sections via the equal area method, assuming that the concrete has cracked completely and does not contribute to tension. Since the fractured outside girder alone takes part in the postulated critical mechanism, the negative moment capacity of half the section is used for the computation of the overstrength factor of the exterior spans.

The tabulations in the examples in Section 5, 6 and 7 that follow are presented such that the input values to be used depend on bridge geometry, the material properties of the deck and

the guardrail, the reinforcement, and the structural steel. They are indicated by yellow highlighting of the corresponding row number, with the value itself in boldface. Similarly, the values that need to be solved to ensure equilibrium and the corresponding equilibrium checks are indicated by blue highlighting of the corresponding row number, with the value itself in boldface.

The other rows can be automated by feeding the formulae presented in the column named FORMULA/DEFINITION/EQUATION, which also mentions the conditions for which each formula is applicable. Since Bridge 2 does not have support fixity at all, the moment calculations for the positive and negative composite deck and the intact girders are irrelevant for this bridge, and are therefore not included in this section. The results are also presented in boldface.

4. COMPUTATIONAL IMPLEMENTATION OF GRILLAGE ANALYSIS

The computational analysis of the fracture critical twin tub girder bridges (TTGBs) may be implemented using commercial nonlinear structural analysis software. Programs such as SAP2000 may be useful to carry out the following steps.

Step 1: Define Cylindrical Coordinate System for the Grillage Grid

For the TTGB, the longitudinal grids need to be located at the two exterior edges of the bridge, the CL of the two exterior top flanges, and the two interior top flanges. The transverse grillage grids need to be located at the ends on 7 ft spacing increments in the middle of the bridge (for easier assignment of the HS-20 truck load), and at pier locations in the case of a multi-span bridge. An illustration of the grid system for a single-span bridge is shown in Figure 4.1. The transverse spacing increments will need to be converted to a radial spacing in the cylindrical coordinate system using Equation (4.1):

Radial Spacing (deg.) =
$$\left(\frac{Spacing Length}{Radius of bridge}\right) * \left(\frac{180}{\pi}\right)$$
 (4.1)



Figure 4.1. Grillage Grid System for a Single-Span Bridge.

Step 2: Nonlinear Material Properties of the Members

The fractured TTGB will be analyzed at ultimate loading conditions; therefore, the steel and concrete components of the bridge will be taken beyond their elastic capacity. The composite

girder and deck system is composed of concrete that will reach cracking and crushing strains and of rebar and steel plates that will reach strains beyond yielding. The material models to be used are represented in Figure 4.2. Nonlinear constitutive material behavior is defined in the advanced properties within the material definition.



Step 3: Define Section Properties for Longitudinal and Transverse Bridge Members

Using the section designer feature in SAP2000, a composite tub, deck, and railing section can be generated. The exterior longitudinal member in Figure 4.3 consists of the railing, the deck from the CL of the girder to the exterior edge (with corresponding reinforcing bar), one top flange, one web, and half of the bottom flange. The interior longitudinal member consists of the deck from the CL of the bridge to the CL of the girder, with corresponding longitudinal reinforcing bar, one top flange, one web, and one half of the bottom flange. The transverse members in Figure 4.4 consist of concrete deck and transverse reinforcing bar. However, it is critical to set the weight modifier to zero of the transverse section to not double-count the concrete deck weight. It should be noted that as the steel plate members change dimensions and the reinforcing pattern changes throughout the length of the bridge, and new sections will need to be created to represent the new dimensions.



a) Exterior Longitudinal Member
 b) Interior Longitudinal Member
 Figure 4.3. Representative Longitudinal Members.



a) End Transverse Member Figure 4.4. Representative Transverse Members.

The fractured girder can be modeled by simply copying the exterior and interior longitudinal sections and removing the bottom flange, web, and top flange steel plate components.

Step 4: Define Hinge Properties

Following the creation of the necessary longitudinal and transverse members, plastic hinges need to be created for each section. The grillage push-down analysis will generate plastic hinge formation under the ultimate loading condition. Within the section designer of SAP2000, there is a moment curvature response tool that allows the user to generate moment curvature data for

each of the members created in Step 3. The data from SAP2000 is then exported into an Excel spreadsheet. Then the angle on the Moment Curvature window can be changed to 180 to get the negative moment curvature, and once again the data is exported to the same Excel spreadsheet. The moment curvature response is then normalized against the maximum positive and negative moments and their corresponding curvatures and then plotted. The Hinge Definition window in SAP2000 will only allow four normalized positive and negative moment curvature data points per section hinge. Therefore, a best fit plot for each moment curvature response needs to be generated in Excel using only 9 points (4 positive, 4 negative, and 1 zero). The hinge length is assigned as half the member depth. A representative hinge property is depicted in Figure 4.5. For ease of convergence, non-negative slopes are recommended for the hinge properties.



Figure 4.5. Representative Hinge Property.

Step 5: Assign Longitudinal and Transverse Members to the Grid

Using the "quick draw" frame section tool in SAP2000, the various longitudinal and transverse frame sections can be assigned to the grillage grid that was established in Step 1 by merely

selecting the desired section from the drop-down menu and clicking on the appropriate grillage grid. Figure 4.6 shows a screenshot from SAP2000 after all members have been assigned to a simple-span bridge.



Figure 4.6. Screenshot of SAP2000 Post Frame Section Assignment.

Step 6: Assign Hinges to Frame Members

At this stage, the longitudinal and transverse members are already assigned to the grillage grid. To allow for plastic hinge formation, hinges need to be assigned at the nodal intersection of all members, as represented in Figure 4.7. Longitudinal hinges need be placed at both joints at the end of each member. Transverse hinges need to be assigned at a distance of half a top flange width away from each node.



Figure 4.7. Representative Hinge Assignments.

Step 7: Assign Boundary Conditions

The support conditions of the physical bridge are elastomeric bearing pads. These pads are represented by springs with a lateral stiffness of 6 kip/in. and a vertical stiffness of 3050 kip/in. in the grillage model, as represented in Figure 4.8. For the single-span bridges, springs will be assigned at each end of the longitudinal joints. For the two- and three-span bridges, springs are also assigned to the ends of the longitudinal joints and to the joints at the pier location.



Figure 4.8. Spring Boundary Conditions.

Step 8: Define Load Patterns and Load Cases

For single-span bridges, two additional load patterns need to be created: the truck load and the lane load. For the two- and three-span bridges, a truck load pattern needs to be defined for each span, and the same follows for the lane load. Once load patterns are established, load cases need to be created. Each load case represents a load combination of 1.25*DL + 1.75(LL + IM), where DL = dead load, LL = live load, and IM = impact load, or 1.25*DL + 1.75*LL + 2.33*HS-20, where DL = dead load, LL = lane load, and HS-20 = truck load. Each load case should be set to nonlinear behavior. The first load case should start from a zero initial condition. Each proceeding load case should start from the end of the previous load case. Each span should have its own set of load cases. In addition, each load case should be divided into 20 or more steps.

Step 9: Assign Frame Loads

Two lanes of truck loading should be assigned as a series of point loads, and the two lanes of lane loads should be distributed as lane loads to the longitudinal members (as depicted in Figure 4.9). An HS20 truck load consists of 32 kip middle and rear axles and 8 kip front axle loads spaced 14 ft apart longitudinally and 6 ft transversely. The first line of wheels will be placed 3 ft from the curved edge of the deck (2 ft away from the face of the curb plus 1 ft curb thickness), the second 9 ft from the edge of the deck, the third 15 ft from the edge of the deck, and the fourth 21 ft from the edge of the deck. The middle axle load of each truck should be placed at half the span length on the single-span bridges and interior middle spans of three-span bridges. The middle axle should be placed at approximately 0.4*L from the end of the end spans of two-span and three-span bridges. Two lanes of lane loading (0.64 kip/in. each) should be located—one 8 ft from the edge and the second lane 20 ft from the edge. However, since the longitudinal members are placed according to the girder placement, the lane loads have to be distributed according to the tributary area.



Figure 4.9. Grillage Truck and Lane Load.

Step 10: Assign Data Collection Points at Centerline of Girder

At the location of each of the center axles, the transverse members between the outer longitudinal member and the interior longitudinal member need to be divided into two pieces using the Divide Lines feature in SAP2000. This process allows for the collection of data at the CL of the member.

Step 11: Run Analysis for Dead Load Only

In order to get to dead load value of the data, the intact bridge should be run solely under the dead load case. The reactions should be recorded for all supports.

Step 12: Run Analysis for All Load Cases for Each Intact Span

For the intact bridge, run all HL93 load cases for the span being evaluated. Once the program has run, collect the displacement data for points 1 thru 4 (seen in Figure 4.10) and the CL points of the inside and outside girder at the location of the center axle. Be sure to obtain the results in the step-by-step format so that the load case and step for each displacement point can be collected as well. This process will be completed once for each span.



Figure 4.10. Location of Grillage Data Collection Points.

Step 13: Replace Hinges at Fracture

At the location of the center axle for the span being evaluated, replace the longitudinal hinges on the outside girder with their fractured counterparts. The hinge assignment is depicted in Figure 4.11.



Figure 4.11. Fractured Hinge Pattern.

Step 14: Run Analysis for All Load Cases for Each Fractured Span

For the fractured bridge, run all HL93 load cases for the span being evaluated. Once the program has run, collect the displacement data for points 1 thru 4 (seen in Figure 4.10) and the CL points of the inside and outside girder at the location of the center axle. Be sure to obtain the results in the step-by-step format so that the load case and step for each displacement point can be collected as well. This process will be completed once for each span, making certain to replace intact hinges in the preceding span before assigning fractured hinges in the following span.

Step 15: Post-Process the Data

For both the intact and fractured bridge, the following calculations need to be made:

• Omega (Ω):

$$\circ \quad \varOmega_i = \varOmega_{i-1} + \Big(\frac{1}{\# \, of \, Steps \, in \, Load \, case} \Big).$$

- Longitudinal Chord Rotation of Interior and Exterior Girder:
 - Chord Rot._{Single Span or Interior Span} = $-1 * \left(\frac{\delta_{CL}}{0.5 * L} \right)$.
 - Chord Rot._{Exterior Span} = $-1 * \left(\frac{\delta_{CL}}{0.4 * L} \right)$.
 - The above equations are in radians.
- Transverse Deck Rotation:
 - Relative rotation of deck at inside flange of inside girder:

•
$$\alpha_{2-3} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_2 - \delta_1}{w}\right)$$

•
$$\alpha_{3-2} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_4 - \delta_3}{w}\right).$$

- Where s = spacing between the interior top flanges (at the CL of the web) of the inside and outside girders and w = spacing between the top flanges (at the CL of the web) of the same girder.
- The above rotations are in radians.
- Applied Load:
 - ο Calculate unit applied load or applied load at 1 Ω:
 - Unit Applied Load_{single span} = 1.25 * Total Reaction from Dead Load Case + 2 * (2.33 * HS20 truck + 1.75 * Lane Load).
 - Unit Applied Load_{multi span} = 1.25 *Total Reaction from Dead Load Case $*\left(\frac{L_{span}}{L_{Total}}\right) + 2 *$ (2.33 * HS20 truck + 1.75 * Lane Load).
 - Applied Load = Unit Applied Load $* \Omega$.
- Intact Stiffness of Intact Bridge:

o Initial Stiffness<sub>Intact at
$$\Omega = 0.4 = \frac{0.4}{Abs(\delta_{OG-CL})}$$
.</sub>
- Instantaneous Stiffness for Fractured Bridge:
 - Instantaneous Stiffness_{OG-Frac.i} = $\frac{\Omega_i \Omega_{i-1}}{\delta_i \delta_{i-1}}$.

The criteria above can be organized into an Excel spreadsheet, as noted in Figure 4.12.

1	A	8	С	D	E	F	G	н	1.1	J	K	L	м	v	x	Y	Z	AA	AB	AC	AD
1	Intact				OG-CL	IG	CL	Point 4	Point 3	Point 2	Point 1			Applied		0	G	IC	3		
	Indexe	1	•	0-11-11-1	ch and from the	Dalla (la)	ch and (and)	Dalla II-1	Dalla (In)	0-1-1-1	D			Load	0.111	Dalla Hal	Chord	Dalla II-1	Chord	α 23	α 32
2	Load Case	Load Step	u u	Delta (in)	Chord (rad)	Delta (in)	Chord (rad)	Deita (in)	Deita (in)	Deita (in)	Delta (in)	α 23 (rad)	α 32 (rad)	(kip)	ti (cai)	Deita (in)	(deg)	Deita (in)	(deg)	(deg)	(deg)
3	LC1	0	0	0	0.00000	0	0.00000	0	0	0	0	0	0	0	0.00	0.00	0.00000	0.00	0.00000	0	0
-4	LC1	1	0.05	-0.1037	0.00015	-0.087926	0.00013	-0.10541	-0.10129	-0.0942	-0.07905	7.89E-05	4.94E-05	66	0.05	0.10	0.00861	0.09	0.00730	0.004522	0.002831
5	LC1	2	0.1	-0.2074	0.00030	-0.175851	0.00025	-0.21081	-0.20259	-0.18839	-0.15809	0.000158	9.88E-05	132	0.10	0.21	0.01722	0.18	0.01460	0.009043	0.005662

Figure 4.12. Spreadsheet of Grillage Data.

Failure criteria are as follows:

- The instantaneous stiffness for the fractured outside girder is less than 5 percent of the initial stiffness of intact outside girder.
- The chord angle (deflection at midspan divided by half span length) of the outside girder for simple spans or interior spans is greater than 2 degrees. The chord angle for exterior spans in multi-span bridges is greater than 3 degrees.
- The transverse deck rotation is greater than 5 degrees.

5. BRIDGE 2

5.1 **YIELD LINE ANALYSIS EXAMPLE OF BRIDGE 2**

This section documents the steps to compute the overstrength factor of Bridge 2 via the yield line analysis (grillage analysis is explained in the next sub-section). Figure 5.1 presents the dimensional details of the simply supported span of Bridge 2. Further details can be found in Appendix A.





This section also presents the stepwise procedure of the yield line analysis conducted to establish the upper-bound and lower-bound solution range for the overstrength factor that is in conjunction with the theory of plastic analysis in Section 3.

The moment capacities (longitudinal and transverse) of the deck slab are calculated using the standard US code-based procedure following the Whitney's stress block approach. The capacities are calculated for one ft wide cross-section of the bridge. The geometric parameters namely B = the total width (breadth) of the deck slab, t = thickness and b = the width of each cross-section are noted from the structural plans. The various material properties of concrete and steel are obtained through the bridge plans and the reports associated with the respective bridges. These properties include f_c' = the specified compressive strength of concrete, ε_{cu} = the maximum strain at the extreme concrete compression fiber (computed as per Section 22.2.2 of ACI-318 (2017) which states the "assumptions for concrete."), f_y = the yield strength of steel of the reinforcement bars, E = the Young's modulus of steel and ε_y = the yield strain that is the ratio of yield strength and Young's modulus of steel in the reinforcement. β_1 = the factor relating depth of equivalent rectangular compressive stress block to depth of neutral axis is computed using Table 22.2.2.4.3 of ACI-318 (2017). The formula applicable for the strength of concrete used in this study is as follows

$$\beta_1 = 0.85 - 0.05(f'_c - 4)$$
 where f'_c is in ksi

The details of the reinforcement such as $\#_{bar}$ = the number of bars per one ft wide section (for the longitudinal capacities) or s = the on-center spacing (for the transverse capacities), d_b = the diameters of the bars, A_s the corresponding areas of steel, cc = the clear cover (from the bridge drawings) and d and d' = the subsequently computed effective depths of the tensile and compressive zones respectively, are recorded. ε_{top} and ε_{bot} = the net tensile strain in the extreme tension reinforcement at the top and bottom of the section respectively, are determined from a linear distribution (ACI-318 2017).

The section of concrete is considered to be divided into compressive and tensile zones by a neutral axis that is located at a depth c from the top fiber. The compressive force due to concrete (with a negative sign) is found using the formula

$$C_c = 0.85 f'_c b \beta_1 c = 0.85 f'_c b$$
 where $a = \beta_1 c$

If the reinforcement steel is in compression, the compressive force due to compression steel is denoted by a negative sign, as follows

$$T_{top/bottom} = -(A_s f_s - 0.85 f_c' A_s)$$

The tensile force due to the steel reinforcement is given by

$$T_{top/bottom} = A_s f_s$$

To ensure that the computation of the forces considers whether or not the steel has yielded, the following formula is used for the strength of steel at the extreme fiber of reinforcement

$$f_{s} = \frac{E\varepsilon_{top/bottom}}{\left\{1 + \left|\frac{E\varepsilon_{top/bottom}}{f_{y}}\right|^{20}\right\}^{0.05}}$$

The depth of neutral axis, c (indicated by blue highlighting) is obtained by equating the tensile and compressive forces. The moment capacities are found by taking moments of forces about the neutral axis.

The computation of the overstrength factor is based on the procedure explained in the Section 3. The data necessary for the calculations such as the geometric details, the moment capacities, the volume of girder and area of cross-section of rail are listed as input values indicated by yellow highlighting. The parameters needed to solve the equations in Section 3 are computed using the formulae and allowance factors for dead load, live load and the weight of the stiffeners. The parameters and their corresponding formulae and values are tabulated in a sequential order and can be regenerated using any spreadsheet program.

		MOMENT CAPACITY OF DECK SLAB						
		·	(BRIDGE 2)	1	1		4	
		Descríptíon	Parameter	Data				
N N N	1	Total Width	B in	317				
DECT	2	Thickness	t in	8				
Ο Ο	3	Section width	b in	12				
LETE LAL	4	Characterístíc Compressíve Strength	f_c' ksi	4				
NCF	5		eta_1	0.85				
CO M	6	Strain	Е _{си}	0.003				
IAL IAL	F	Yield Strength	f _y ksi	60				
EBA	8	Young's Modulus	E ksi	29000				
RI MA	9	Strain	$\mathcal{E}_{\mathcal{Y}}$	0.002070				
		Moment Computations		Longíti	ıdínal	Tra	nsverse	
	10			m'_x	m_x	m_y'	m_y	
	11			Тор	Bottom	Тор	Bottom	
	12	Bar No.		5	5	5	5	
	13	Díameter of Bar	d_b in	0.625	0.625	0.625	0.625	
ALC	14	Area of Bar	A_{Φ} in ²	0.31	0.31	0.31	0.31	
DETA	15	Spacing	s in	_	—	5	5	
BAR	16	No. of Bars	# _{bar}	38	32	_	_	
CING	17	Area of Steel	$A_s in^2$	0.446	0.376	0.744	0.744	
IFOR	18	Clear Cover	cc in	2	1.25	2	1.25	
REIN	19	Effective depth (tension)	d in	5.0625	5.8125	5.6875	6.4375	
	20	Effectíve depth (comp.)	d' in	2.1875	2.9375	1.5625	2.3125	
	21	Depth of NA	<i>c</i> in	1.35	1.42	1.44	1.81	

	MOMENT CAPACITY OF DECK SLAB						2
			(BRIDG	E 2)	4		
		Moment Compute	atíons	Longít	udínal	Transverse	
	22			m'_x	m_x	m'_y	m_y
	23			тор	Bottom	Тор	Bottom
AIN	24		ε _{top}	0.008227	0.003201	0.008829	0.000837
STR	25		ε_{bot}	0.001851	0.009270	0.000250	0.007682
	26	$\beta_1 c$	<i>a</i> in	1.15	1.21	1.23	1.54
	27	Compression-Concrete	C _c kip	-46.91	-49.29	-50.03	-62.70
CES	28	Tension/	T _{top} kip	26.76	26.76	44.64	18.06
FOR	29	Compression-Steel	T _{bot} kip	20.16	22.53	5.39	44.64
	30	Equílíbríum Check	$\Sigma T + C$ kip	0.00	0.00	0.00	0.00
RESULTS	31	Moment	<i>M_n</i> k-in/in	12.71	14.98	19.30	23.41
		Remarks:		Тор	Тор	Тор	Top Not
				Yielded	Yielded	Yielded	Yielded
				Bottom	Bottom	Bottom	Bottom
			Yielded	Yielded	Yielded	Yielded	
				<u> </u>	1	<u> </u>	<u>.</u>
				Both steel	Both steel	Both steel	Both steel
				ín	ín	ín	ín
				Tension	Tension	Tension	Tension

		COMPUTATION OF OVERSTRENGTH FACTOR				
			SINGLE SPAN CASE (BRIDGE 2)		4	
		Parameter	Formula/Definition/Equation	1	ata	
	1	L_x	Span length (center líne)	115.00	ft	
	2	R _{CL}	Radius of center line	1910	ft	
rt	3	В	width	26.4	ft	
ROMET	4	L_x^*	Outer region length $L_{\chi}^{*}=\left(1+rac{B}{4R_{C_{L}}} ight)L_{\chi}$	115.40	ft	
Ū	5	S	Inter-Gírder Spacing	6.1	ft	
	6	b	Width of Girder + Edge	10	ft	
	F	t	Deck Thickness	8	in	
	8	m_x	Longitudinal Positive Moment per A	15	k-in/in	
	9	m'x	Longitudinal Negative Moment per ft	13	k-in/in	
	10	m_y	Transverse Posítive Moment per A	23	k-in/in	
	11	m'_y	Transverse Negative Moment per ft	19	k-in/in	
	12	tan θ	$\tan \theta = \sqrt{\frac{m'_y + m_y}{m'_x + m_x}}$	1.24	(Θ = 50.8°)	
K DONE	13	tan α	$\tan \alpha = \frac{2s}{L_x}$	0.11	$(\Theta = 6.0^{\circ})$	
IAL WOR	14	k_{bound}^{upper}	$\left[1+2\frac{\tan\alpha}{\tan\theta}\right] = 1+\frac{4s}{L_x}\sqrt{\frac{m'_x+m_x}{m'_y+m_y}}$	1.17		
INTERN	15	k ^{lower} k ^{bound}	$\left[1+2\frac{\tan^2\alpha}{\tan^2\theta}\right] = 1 + \frac{8s^2}{L_x^2}\sqrt{\frac{m'_x + m_x}{m'_y + m_y}}$	1.01		
	16	IWDupper	$(m'_{y} + m_{y}) \left(\frac{L_{x}}{2s}\right) k^{upper}_{bound} + \frac{4m_{x}b}{L_{x}}$	478	k-ft *	
	17	IWD _{lower}	$\left(m'_{y}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k^{lower}_{bound}+\frac{4m_{x}b}{L_{x}}$	415	k-ft *	

Note: *: A unit deflection ($\delta = 1$) is considered; therefore, the unit of internal work is in k-ft

		CON	COMPUTATION OF OVERSTRENGTH FACTOR					
			SINGLE SPAN CASE (BRIDGE 2)					
		Parameter	Formula/Definition/Equation	Da	ta			
	18	DL	Dead Load Factor	1.25				
	19	LL	Live Load Factor	1.75				
	20	SAF	Stiffener Allowance Factor	1.15				
	21	γc	Unit weight of reinforced concrete	0.15	kcf			
Ą	22	Wu	Area load due to reinforced concrete + lane load $DL\gamma_c \frac{t}{12} + LL \cdot \frac{0.64}{12}$	0.218	ksf			
E, E	23	γs	Unit weight of steel	0.49	kcf			
DON	24	V_g	Volume of Gírder	148	ft ³			
Б Х Х	25	Ar	Area of Rail Cross-Section SSTR	2.75	ft ²			
MO	26	Vr	Volume of Raíl = $L_x A_r$	317	ft ³			
NAL	27	Wx	$1.25 \left(1.15 V_g \gamma_s + V_r \gamma_c\right) / L_x$	1.42	k/ft			
SXTER	28	<i>y</i> (Lane 2)	(b + s - 15) for $(b + s) < 21(b + s - 18)$ for $(b + s) > 21$	1.25	ft			
Ψ	29	Klane	$1 + 0.5 \frac{y}{s}$ for $(b+s) < 21; 1 + \frac{y}{s}$ for $(b+s) > 21$	1.10				
	30	EWD _{HS-20}	$\left(168 - \frac{2613}{L_x}\right) K_{Lane}$	160	k-ft			
	31	Wet	$w_u L_x(b + 0.5s) + W_x L_x + 2EWD_{HS20}$	817	k-ft			
	32	EWD	0.5 Wet	409	k-ft *			
ILTS	33	Ω_{upper}	IWD _{upper} /EWD	1.17				
RESU	34	Ω_{lower}	IWD _{lower} /EWD	1.02				

Note: *: A unit deflection ($\delta = 1$) is considered; therefore, the unit of external work is in k-ft

5.2 GRILLAGE ANALYSIS EXAMPLE OF BRIDGE 2

The following steps are explained to conduct the computational implementation of grillage analysis of Bridge 2.

1. Gather bridge geometry and material information.

Steel Tub Properties (fy = 50 ksi)									
Terretien	Top Flange		V	Veb	Bottom Flange				
ft	Width in.	Thickness in.	Width in.	Thickness in.	Width in.	Thickness in.			
0–115	18	1.00	79	0.625	50	1.00			

Location	Parameter	Description/Value
	Location	Harris County, I-610
	Year Designed/Year Built	2002/2004
Bridge	Design Load	HS25
C	Length, ft	115
	Spans, ft	115
	Radius of Curvature, ft	1909.86
	Width, ft	26.417
Deelr	Thickness, in.	8
Deck	Haunch, in.	4
	Rail Type	SSTR
	# of Bar Longitudinal Top Row (#5)	40
	# of Bar Longitudinal Bottom Row (#5)	32
Rebar	Transverse Spacing Top Row (#5)	5
	Transverse Spacing Bottom Row (#5)	5
	Rebar Strength (ksi)	60
	CL of Bridge to CL of Girder (in.)	79.5
Girder	CL of Top Flange to CL of Top Flange (in.)	86

2.	Determine material	constitutive	behavior.

Concrete (4 ksi)			
Stress	Strain		
(ksi)	(1/in)		
-4	-3.79E-03		
-4	-3.56E-03		
-4	-2.69E-03		
-4	-1.78E-03		
-3.8205	-1.40E-03		
-2.8718	-8.69E-04		
-0.6403	-1.78E-04		
0	0		
0.378	1.06E-04		
0.378	1.16E-03		

Rebar (60 ksi)				
Stress	Strain			
(ksi)	(1/in)			
-87.9	-0.095			
-87.9	-0.0944			
-86.6	-0.0761			
-78	-0.0386			
-60.7	-9.80E-03			
-60.3	-2.08E-03			
0	0			
60.3	2.08E-03			
60.7	9.80E-03			
78	0.0386			
86.6	0.0761			
87.9	0.0944			
87.9	0.095			

Steel (50 ksi)			
Stress	Strain		
(ksi)	(1/in)		
-71.6	-0.1		
-71.6	-0.097		
-71.6	-0.095		
-71.6	-0.0946		
-70.3	-0.0764		
-62.5	-0.039		
-50	-0.0196		
-50	-1.72E-03		
0	0		
50	1.72E-03		
50	0.0196		
62.5	0.039		
70.3	0.0764		
71.6	0.0946		
71.6	0.095		
71.6	0.097		
71.6	0.1		

3. Create a coordinate system for half width of the span.



4. Create a cylindrical coordinate system for the curved bridge and ensure that the middle transverse divisions are 7 ft because it will aid in applying the truck load, whose axles are separated by 14 ft.

A. # of Segments =
$$\left(\frac{\text{Length }(ft)*12}{84 \text{ in.}(7ft)}\right)$$
 rounded to nearest even number.
of Segments = $\left(\frac{115*12}{84}\right)$ = 16.428, so 14 was selected.
B. End Segment Length = $\frac{((\text{Length}*12) - (\# \text{ of Segments}*84))}{2}$.
End Segment Length = $\frac{((115*12) - (14*84))}{2}$ = 102 in.

C. Theta = $\frac{Length}{Radius}$.

$$Theta = \frac{115}{1909.86} = 0.0602 \ rad \ or \ 3.450 \ degrees$$

D. Determine the radial offsets using the outside edge, the outside flange, the inner flange, and CL of the bridge.

Offsets (in.)				
Edge	158.5			
Outside Flange	122.5			
Inner Flange	36.5			
CL of Bridge	0			

Rad	lial Spacin	ng (in.)
А	23077.3	CL + Edge
В	23041.3	CL + OF
С	22955.3	CL + IF
Center Line	22918.8	or 1909.86 (ft)
D	22882.3	CL – IF
Е	22796.3	CL – OF
F	22760.3	CL – Edge

- E. Determine the longitudinal spacing along theta by converting the longitudinal segment lengths into degrees.
 - The first and last segments are 102 in., and the intermediate segments are 84 in. The total length is 115 ft, or 1380 in. • Radial Spacing (rad) = $\frac{Long.Spacing}{Radius}$ • Radial Spacing (degree) = Radius Spacing (rad) * $\frac{180}{\pi}$

Long. Spacing (in.)	Radial Spacing (rad.)	Radial Spacing (degrees)
0	0.0000	0.000
102	0.0045	0.255
186	0.0081	0.465
270	0.0118	0.675
354	0.0154	0.885
438	0.0191	1.095
522	0.0228	1.305
606	0.0264	1.515
690	0.0301	1.725
774	0.0338	1.935
858	0.0374	2.145
942	0.0411	2.355
1026	0.0448	2.565
1110	0.0484	2.775
1194	0.0521	2.985
1278	0.0558	3.195
1380	0.0602	3.450

- *Int.Transverse Element width* = 84 *in.*
- End Transverse Element = $102 \left(\frac{84}{2}\right) = 60$ in.
- 5. Input the coordinate system into SAP2000.
 - A. Select File > New Model > Blank model (making sure units are in kips and in.).
 - B. Right click on the blank workspace and select Edit Grid Data > Modify/Show System > Quick Start > Cylindrical.
 - In the Number of Gridlines panel, set "Along Z = 1."
 - In the Grid Spacing panel, set "Along Z = 1."
 - Select OK.
 - Delete all R and T Coordinates that were generated.
 - C. Add correct coordinates for R.

All radial coordinates (A, B, C, D, E, and F).

- D. Add correct coordinates for T.
 - All theta coordinates for T (0 to 1380 in.).
 - Click OK.
- E. The grid system is now formed.

System Name	•	GLO	BAL				Quick Start
R Grid							
Grid ID	Ordinate (In)	Line Type	Visible	Bubble Loc	Grid Color		
A	23077.3	Primary	Yes	End		Add	
В	23041.3	Primary	Yes	End			
с	22955.3	Primary	Yes	End		Delete	
D	22882.3	Primary	Yes	End			
F	22796.3	Primary	Yes	End			
F	22760.3	Primary	Yes	End			
							Display Grids as
Grid							Ordinates Spacing
Grid ID	Angle (deg)	Line Type	Visible	Bubble Loc	Grid Color		G ordinates C spacing
1	0	Primary	Yes	Stat		Add	
2	0.255	Primary	Yes	Stat			Hide All Grid Lines
3	0.465	Primary	Үсэ	Stat		Delete	Glue to Grid Lines
4	0.675	Primary	Yes	Stat			
5	0.885	Primary	Yes	Stat			Bubble Size 36.
6	1.095	Primary	Yes	Stat			
7	1.305	Primary	Yes	Stat	~		
Z Grid Data							Reset to Default Color
Crid ID	Ordinate (i	in) Lir	пс Турс	Visible	Bubblc Loc		Reorder Ordinates
Z1	0	P	nmary	Yes	End	Add	noordor ordinated
						Delete	
							OK Cancel



- 6. Define material in SAP2000.
 - A. Click Define > Materials > Add New Material > Material Type (Steel, Concrete, or Rebar) > Standard (User) > OK.
 - B. At the bottom of the window, select the box that states, "Switch to Advanced Properties."
 - C. In the open window, name the material "Concrete," "Steel," or "Rebar" depending on which material is being defined. Then click "Modify/Show Material Properties."
 - D. In the Material Property Data window, click the "Nonlinear Material Data" icon.
 - E. In the Nonlinear Material Data window, select the "Convert to User Defined" icon.
 - F. Input the number of data points for the stress strain behavior (10 for concrete, 13 for rebar, 17 for steel).
 - G. Input the data points for the stress strain behavior.
 - H. Select "OK."
 - I. Repeat this process again for the remaining materials.

atena	IName		Material Type			
Concr	ete		Concrete			
ystere	esis Type	Drucker-Prager Pa	arameters	Units		
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tress-	Strain Curve Defir	ition Options				
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- 7. Define frame cross sections in SAP2000.
 - A. Click Define > Section Properties > Frame Sections > Add New Properties.
 - B. In the Frame Section Properties drop-down box, select Other and click Section Designer. In the SD Section Designer window, name the section B2Long and click the Section Designer icon.
 - C. Using the Polygon feature, draw the features of the half width of the bridge from Step 3. Features include one rail, two concrete deck pieces, two concrete haunches, two top flanges, two webs, and two pieces of the bottom flange.
 - To change material types for the polygons, right-click on the polygon and select the desired material type from the material drop-down menu.
 - To change the coordinates of the polygon's nodes, use the Reshaper tool.
 - D. Add in the longitudinal rebar to both concrete deck elements by using the Line Bar from the Draw Reinforcing Shape tool. From the design drawings, it can be determined that there are 11 #5 top bars and nine #5 bottom bars in the outer concrete deck element and nine #5 top bars and seven #5 bottom bars in the inner concrete element, with a 2 in. top cover and transverse reinforcement. The top bars are located at 5.0625 in. with a 1.25-in. bottom cover, and transverse reinforcements are located at 2.1875.
 - E. Click Done.



- F. Repeat this process for the transverse elements.
 - At the SD Section Designer window select Modifiers and set Mass and Weight to 0 in order to not double-count the dead weight.
 - The interior transverse members are 84 in. wide (end members are 60 in. wide), with #5 rebar at 5 in. spacing at 5.6875 in. and 1.5625 in.



G. To generate an exterior longitudinal member and an interior longitudinal member, make two copies of the B2Long section. Label one LongOut and the other LongInt.



• For the LongOut, delete every element right of the CL.

• For the LongInt, delete every element left of the CL.



H. To generate a simulated fracture section, make copies of LongOut and LongInt.Name the LongOut copy FracOut.



Delete the bottom flange, web, and top flange of the steel tub.

• Name the LongInt copy FracInt.

Delete the bottom flange, web, and top flange of the steel tub.



- 8. Generate plastic hinges for frame elements in SAP2000.
 - A. Define > Section Properties > Frame Sections.
 - B. Select the desired cross-section. Hinges will need to be made for the LongOut, LongInt, FracOut, FracInt, Trans, and TransEnd.
 - C. Once selected, click Modify/Show Property > Section Designer.
 - D. Once in the section designer, select the Moment Curvature Curve tool.
 - E. In the Moment Curvature Curve window, select Details.
 - Copy the Moment Curvature data to an Excel file.
 - Select OK.
 - F. In the Moment Curvature Curve window, change the angle (deg) to 180, then select Details.

- Copy the Moment Curvature data to the same Excel file as was done previously.
- Select OK.
- G. Generate a Normalized Moment Curvature diagram.
 - Normalize the moments by dividing each of the positive moments by the maximum positive moment and the negative moment by the maximum negative moment. Next, divide the curvatures by the curvatures corresponding to the maximum and negative moments.
 - Plot the normalized positive and negative moment curvatures on a chart.
 - Create a hinge moment curvature plot on the same chart with four positive moment points and four negative moment points without generating a negative slope.

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6	-1.84E-03	1.106	2.43E-03	0	-2473	-213.797	2685.704	0	-0.7189	3.31E-05	228216		0.909959	0.56							1				
7	-3.06E-03	5.3861	4.63E-03	0	-2986	-205.926	3191.444	0	-4.14E-02	5.96E-05	250798		1	1.00											
8	-4.63E-03	6.6997	7.32E-03	0	-3161	-234.918	3396.077	0	-0.1158	9.26E-05	245937		0.980618	1.56			÷				1.5				
9	-6.64E-03	6.5511	0.0104	0	-3129	-349.403	3477.927	0	-0.0659	0.000132	234212		0.933867	2.22			ner								
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12	-0.0256	-31.1492	0.012	0	0	-1883	1882.917	0	-0.4252	0.000291	153264		0.611105	4.89			aliz				0.5				
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17	-0.0603	-30.9435	0.0285	0	0	-2048	2048.255	0	0.0298	0.000688	167050		0.666074	11.55						Normaliza	d Cuprati				
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19	-0.0785	-31.1408	0.0368	0	0	-2208	2208.25	0	0.0249	0.000893	180046		0.717892	15.00											
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21	-0.0988	-31.1509	0.0463	0	0	-2327	2326.757	0	0.0915	0.001125	189616		0.756051	18.89		-1.35	5 -35								
22	-0.1099	-31.156	0.0515	0	0	-2371	2370.534	0	0.0316	0.001251	193172		0.770229	21.00		-1.35	5 -25								
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H. Define Hinge Length.

The hinge length is one half of the section depth.

- Hinge_{long} = 0.5 * (Deck thickness + haunch height + top flange thickness + web height + bottom flange thickness)
 a. Hinge_{long} = 45.5 in.
- $Hinge_{Frac} = 0.5 * (Deck thickness + haunch height)$
 - a. $Hinge_{Frac} = 6$ in.
- $\circ \quad Hinge_{Trans} = 0.5 * (Deck \ thickness)$

a.
$$Hinge_{Trans} = 4$$
 in.

- I. Make the plastic hinge in SAP2000.
 - Select Define > Section Properties > Hinge Properties > Add New Properties.
 - In the Type window, select Moment Curvature and input the corresponding correct hinge length.

- In the Moment Curvature table, insert the four positive and four negative normalized moment curvatures and the zero point.
- Uncheck the symmetric box and select the Is Extrapolated option in the Load Carrying Capacity beyond Point E window.
- In the Scaling for Moment and Curvature window, insert the maximum positive moment and corresponding curvature as well as the maximum negative curvature and corresponding curvature.
- In the Acceptance Criteria, use the values 1, 2, and 3 for Immediate Occupancy, Life Safety, and Collapse Prevention in the positive column and -1, -2, and -3 for the negative column.
- Repeat for all remaining frame sections (LongOut, LongInt, FracOut, FracInt, Trans, and TransOut).

Frame Hin Edit Displacemen	ge Property Data t Control Paramete Moment/SF	a for LongOUT - ers Curvature/SF	Moment M	3	Type	n	>
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Use	Yield Moment Yield Curvature eel Objects Only) ice Criteria (Plastic inmediate Occupar	Moment SF Curvature SF c Curvature/SF)	250798. 5.960E-05 Positive	144085. 5.960E-05 Negative -1.			
Li C Shov	ife Safety ollapse Preventior v Acceptance Crit	n teria on Plot	2.	-2.	ОК	Cancel	

- 9. Assign frame members to grid.
 - A. Select the Draw tab > Quick Draw Frame/Cable/Tendon.
 - B. In the Section drop-down menu, select LongOut.
 - C. Click on every grid segment on second-to-last longitudinal grids (B and E).



- D. Change the section to LongInt and repeat Step C but for the two interior longitudinal grids (C and D).
- E. Change the Section to TransEnd and repeat Step C but for the end transverse grids (1 and 17).
- F. Change the Section to Trans and repeat Step C but for all other transverse grids (2 to 16).



- 10. Assign hinges to frame elements.
 - A. The longitudinal hinges are placed at the ends of the longitudinal frame elements or at a relative distance of 0 and 1.
 - B. The transverse hinges are placed at a distance of half a top flange width away from the node. 40

• Hinge
$$Loc_{F to E} = \frac{half flange width}{Element Width} = 1 - \frac{\frac{18}{2}}{36} = 0.75.$$

• Hinge $Loc_{E to D}$ and $C to B = \frac{half flange width}{Element Width} = \frac{18/2}{86} = 0.1046$ and $(1 - 0.1046)$ or 0.8954 .

- Hinge Loc._{D to C} = $\frac{half flange width}{Element Width} = \frac{18/2}{73} = 0.1233$ and $(1 1)^{-1}$ 0.1046) or 0.8767.
- Hinge Loc._{B to A} = $\frac{half flange width}{Element Width} = \frac{18/2}{36} = 0.25.$
- C. In SAP2000, assign the hinges to corresponding frame elements.
 - Select the desired frame elements you wish to assign hinges to, such as LongOut. (The elements will turn from blue to yellow).
 - In SAP2000, select the Assign tab > Frame > Hinges.
 - From the drop-down menu, select LongOut and set relative distance to 0 and click Add.
 - From the drop-down menu, select LongOut and set relative distance to 1 and click Add.
 - Click OK.

🐹 Assign Frame Hinges

Hinge Property	Relative Distance	
LongOUT	v 1	
LongOUT	0	
LongOUT	1	Add Hinge
		Modify/Show Auto Hinge
		Delete Hinge
DOP: Moment MS		
ptions		
ptions Add Specified Hinge A	ssigns to Existing Hinge	Assigns
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• Repeat the previous step of assigning the hinges for all other frame elements.



Х

- 11. Assign loads to the frame elements.
 - A. Wheel Axle Loads
 - Axle loads will be placed at distances of 36 in., 108 in., 180 in., and 252 in. from the outside of the curved edge.
 - One line of the 16 kip axles will be placed at the halfway point of the bridge, or Transverse Grid 9, with the second line 14 ft away at Transverse Grid 7. One line of 4 kip axles will be placed 14 ft away from the first line of axles, at Gridline 11.
 - In SAP2000, the loads have to be placed at a relative distance, so this value needs to be calculated.

$$\begin{array}{l} \circ \quad HS20_{Axel\ 1\ Loc\ (B-A)} = \frac{L_1 - 36}{L_1} = \frac{36 - 36}{36} = 0.\\ \circ \quad HS20_{Axel\ 2\ Loc\ (C-B)} = \frac{L_1 + L_2 - 108}{L_2} = \frac{36 + 86 - 108}{86} = 0.1628.\\ \circ \quad HS20_{Axel\ 3\ Loc\ (D-C)} = \frac{L_1 + L_2 + L_3 - 180}{L_3} = \frac{36 + 86 + 73 - 180}{73} = 0.2055.\\ \circ \quad HS20_{Axel\ 4\ Loc\ (E-D)} = \frac{L_1 + L_2 + L_3 + L_4 - 252}{L_4} = \frac{36 + 86 + 73 + 86 - 252}{86} = 0.3372. \end{array}$$

- B. Lane Loads
 - Lane Loads are lane loads of 0.640 kip/ft (0.05333 kip/in.) centered at a distance of 96 in. and 240 in. from outside of the curved edge.
 - These lane loads will be placed on the longitudinal frame elements. They will be assigned to elements along the B, C, D, and E longitudinal elements according to the appropriate tributary distance.
 - $LaneLoad_B = \left(\frac{L_1 + L_2 96}{L_2}\right) * laneload = \left(\frac{36 + 86 96}{86}\right) * 0.05333 = 0.016124 \frac{kip}{in}.$
 - $LaneLoad_{C} = laneload laneload_{B} = 0.05333 0.016124 = 0.037209 \frac{kip}{in}$.
 - $\circ \quad LaneLoad_{D} = \left(\frac{L_{1} + L_{2} + L_{3} + L_{4} 240}{L_{4}}\right) * laneload = \left(\frac{36 + 86 + 73 + 86 240}{86}\right) * 0.05333 = 0.025426\frac{kip}{in}.$
 - $\circ LaneLoad_E = laneload laneload_D = 0.05333 0.025426 = 0.027907 \frac{kip}{in}.$
- C. In SAP2000, first define the load patterns.
 - Select the Define tab > Load Patterns.
 - Under the Load Pattern Name, enter HS-20 and change the type in the drop-down menu to Live. The self-weight multiplier should be set to 0. Then click Add New Load Pattern.
 - Under the Load Pattern Name, enter LaneLoad and change the type in the drop-down menu to Live. The self-weight multiplier should be set to 0. Then click Add New Load Pattern.
 - Click OK.

oad Patterns				Click To:
Load Pattern Name	Туре	Self Weight Multiplier	Auto Lateral Load Pattern	Add New Load Pattern
DEAD	Dead	~ <u>1</u>	\sim	Modify Load Pattern
DEAD	Dead	1		Modify Lateral Load Pattern
LaneLoad	Live	0		Delete Load Pattern
				Show Load Pattern Notes
				ОК

- Assign the wheel loads in SAP2000.
 - Select the exterior transverse element of Gridline 9 and 7.
 - Click Assign > Frame Loads > Point.
 - From the Load Pattern drop-down menu, select HS-20 and verify that the coordinate system is set to Global, the Load Direction is Gravity, and the Load Type is Force.
 - In Column 1, enter a relative distance of 0 (Axle 1, Loc. B-A) and load of 16 kips.
 - o Click OK.
 - Repeat for Gridline 11 to assign the 4 kip load.
 - Repeat Steps 1–6 for Axle 2,3, 4, Loc. C-B, D-C, and E-D.



- Assign the Lane Load in SAP2000.
 - Select all exterior longitudinal frame elements along Gridline B.
 - Click Assign > Frame Loads > Distributed.

- From the Load Pattern drop-down menu, select Lane Load and verify that the coordinate system is set to Global, the Load Direction is Gravity, and the Load Type is Force.
- In the Uniform Load box, enter 0.016124 (Lane Load B).
- o Click OK.
- Repeat Steps 1–5 for all of the longitudinal elements along gridlines C, D, and E.



12. Define load cases.

- A. The load case being used to determine redundancy is 1.25DL + 1.75(LL + IM). Where DL = Dead Load, LL = Live Load, and IM = Impact Load. When substituting in the truck load and the lane load, the preceding equation reduces to 1.25DL + 1.75LaneLoad + 2.33HS20.
- B. Generate load cases in SAP2000.
 - Click Define > Load Cases > Add New Load Case.
 - In the Load Case Name Panel, name the load case "LC1."
 - In the Analysis Type, select "Nonlinear."
 - For the LC1 Load Case in the Stiffness to Use panel, select "Zero Initial Conditions."
 - In the Loads Applied panel, leave the Load Type "Load Pattern" in the drop-down Load Name menu, select DEAD and change the Scale Factor to 1.25. Click Add. Change the Load Name menu, select HS-20, and change the Scale Factor to 2.33. Click ADD. Change the Load Name menu, select Lane Load, and change the Scale Factor to 1.75. Click ADD.
 - In the Other Parameters panel, in the Results Saved section, click Modify/Show.

- In the Results Saved for Nonlinear Static Load Cases window, change the Results Saved to Multiple States, and in the For Each Stage panel, change the Minimum Number of Saved Steps and the Maximum Number of Saved Steps to 20. Click OK.
- Click the OK on the Load Case Data window.
- Repeat the previous 8 steps mentioned in B to create an LC2, LC3, and LC4. However, in the Initial Conditions window, select Continue from State at End of Nonlinear Case, and from the drop-down menu, select the preceding load case. (For LC2, the Nonlinear Case LC1 would be selected).

.oad Case Name		1	Votes	Load Case Type
LC1	Set Def N	lame	Modify/Show	Static ~ Design
nitial Conditions Zero Initial Condi Continue from Sta Important Note: Modal Load Case	ions - Start from Unstressed Stat te at End of Nonlinear Case Loads from this previous case a	e are included in	the current case	Analysis Type Linear Nonlinear Nonlinear Staged Construction Geometric Nonlinearity Parameters
All Modal Loads App	olied Use Modes from Case	М	ODAL ~	 None P-Deita
Load Type	Load Name	Scale Factor	r	P-Delta plus Large Displacements
Load Pattern	DEAD	/ 1.25]	Mass Source
Load Pattern Load Pattern Load Pattern	DEAD HS20 LaneLoad	1.25 2.33 1.75	Add Modify Delete	Previous
Other Parameters				ОК
Other Parameters	Full Load		Modify/Show	
Other Parameters Load Application Results Saved	Full Load Multiple States	_ [Modify/Show	Cancel

- 13. Define end supports.
 - A. The elastomeric bearing pads for each girder have a lateral stiffness of 12 kip/in. and a vertical stiffness of 6100 kip/in. Since the tub girders are divided in half, the lateral stiffness will be 6 kip/in., and the vertical stiffness will be 3050 kip/in.
 - B. Assign spring supports in SAP2000.
 - Select the eight nodes at the very end of the longitudinal members.
 - Click Assign > Joint- >Springs.
 - In the Assign Joint Springs window in the Simple Springs Stiffness panel, enter 6 for Translation 1 and 2 and 3050 for Translation 3.
 - Click OK.



- 14. Define CL data acquisition points mid-span of girder.
 - A. Select the transverse frame elements between B and C and between D and E at mid-span (Gridline 9).
 - B. Click Edit > Edit Lines > Divide Frames.
 - C. In the Divide into Specified Number of Frames window, enter 2 for Number of Frames.
 - D. Click OK.



- 15. Analyze the nonfracture structure for dead load only.
 - A. In SAP2000, click Analyze > Run Analysis.
 - B. In the Set Load Cases to Run window, click the Run/Do Not Run All button until every Action is Do Not Run.
 - C. Select DEAD then click Run/Do Not Run Case until the action is run.
 - D. Click Run Now. Let SAP2000 run the load cases until the screen says the analysis is complete.
 - E. Once the analysis is complete, select Spring Reactions.
 - F. Click Display > Show Tables
 - G. In the Choose Table for Display window, click the + symbol beside Joint Output and select the square box beside Reactions.
 - H. In the Output Options window in the Nonlinear Static Results panel, select Last Step.
 - I. Select and copy the information from the F3 column.

- J. The sum of the F3 values is the dead load.
- K. Click Done.
- L. Unlock the structure.
- 16. Analyze the nonfractured structure.
 - A. In SAP2000, click Analyze > Run Analysis
 - B. In the Set Load Cases to Run window, click the Run/Do Not Run All button until every action is Do Not Run.
 - C. Select LC1, LC2, LC3, and LC4 and then click Run/Do Not Run Case until the action for all four is run.
 - D. Then click Run Now.
 - E. Let SAP2000 run the load cases until the screen says the analysis is complete.
 - F. Once the analysis is complete, select the data collection point on the transverse member on the outside girder (C-B).
 - G. Click Display > Show Tables.
 - H. In the Choose Table for Display window, click the + symbol beside Joint Output and select the square box beside Displacements.

💢 Choose Tables for Display	×
Edit	
MODEL DEFINITION (0 of 70 tables selected) System Data Property Definitions Load Pattern Definitions Load Case Definitions Load Case Definitions Connectivity Data Joint Assignments Frame Assignments Options/Preferences Data Miscellaneous Data Miscellaneous Data Miscellaneous Data Solution Upput Solution Masses Element Output Structure Output	Load Patterns (Model Def.) Select Load Patterns 3 of 3 Selected Load Cases (Results) Select Load Cases 4 of 4 Selected Modify/Show Options Set Output Selections Options Selection Only Show Unformatted
	Named Sets
	Save Named Set
	Show Named Set
	Delete Named Set
	OK Cancel
Table Formats File Current Table Formats File: Program Default	

- I. Click the Modify/Show Options button.
- J. In the Output Options window in the Nonlinear Static Results panel, select Stepby-Step.

IX 0 IY 0 IZ 0	in in in	Mode to			
IZ 0	in	All Modes			
ar Static Results					
		Steady State Results			
velopes	Envelopes				
ep-by-Step	O At Frequencies				
st Step		In and Out of Phase			
ep Static Results					
velopes		RMS			
ep-by-Step		O sqrt(PSD)			
st Step					
Design Results					
ntrolling Combo					
Combos					
	ep-by-step ist Step ep Static Results welopes ep-by-Step ist Step Design Results introlling Combo I Combos	ep-by-step st Step ep Static Results velopes ep-by-Step st Step Design Results introlling Combo I Combos			

- K. Click OK.
- L. Select and copy the information from the Output Case, StepNum Unitless, and the U3 in. column and paste them into an Excel worksheet. These columns represent Load Case, Step Number, and Deflection for the Outside Girder, respectively.

ol 💢	int Displa	icemen	ts								-		\times
File	View	Edit	Format-Filter	-Sort Select	Options								
Units: As Noted Joint Displacements													~
	Join Tex	t t	Outputcase	Case Type Text	StepType Text	StepNum Unitless	U1 in	U2 in	in	R1 Radians	R2 Radians	R3 Radians	^
•	10)3	LC1	NonStatic	Step	0	C	0	0	0	0	0)
	10)3	LC1	NonStatic	Step	1	0.002545	7.7E-05	-0.087926	-2.681E-06	0.000169	5.324E-08	3
	10)3	LC1	NonStatic	Step	2	0.00509	0.000154	-0.175851	-5.363E-06	0.000338	1.065E-07	7

M. Click Done.

- N. Select the data collection point on the transverse member on the inside girder (E-D) and repeat Steps G–L. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- O. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element B) and repeat Steps G–L. This information goes into the Delta 4 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- P. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element C) and repeat Steps G–L. This information goes into the Delta 3 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.

- Q. Select the joint on the transverse member on the inside girder at Transverse Element 9 (at Longitudinal Element D) and repeat Steps G–L. This information goes into the Delta 2 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- R. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element E) and repeat Steps G–L. This information goes into the Delta 1 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.

4	A	8	С	D	E	F	G	н	1.1	J	K	L	м	V	х	Y	Z	AA	AB	AC	AD
1	Intact				OG-CL	IG	CL	Point 4	Point 3	Point 2	Point 1			Applied		0	G	IC	5		
	Lond Core	Load Stop	•	Delta (In)	Chard (red)	Dalta (In)	Chard (rad)	Dolta (In)	Dalta (In)	Delta (In)	Delta (In)	a 22 (rad)	a 22 (and)	Load	O (ml)	Dolta (In)	Chord	Dolta (In)	Chord	α 23	α 32
2	Load Case	Load Step	u	Delta (in)	Chord (rad)	Deita (in)	Chord (rad)	Deita (in)	Deita (in)	Delta (in)	Delta (in)	α 23 (rad)	α 32 (rad)	(kip)	LI (cal)	Deita (in)	(deg)	Deita (in)	(deg)	(deg)	(deg)
3	LC1	0	0	0	0.00000	0	0.00000	0	0	0	0	0	0	0	0.00	0.00	0.00000	0.00	0.00000	0	0
-4	LC1	1	0.05	-0.1037	0.00015	-0.087926	0.00013	-0.10541	-0.10129	-0.0942	-0.07905	7.89E-05	4.94E-05	66	0.05	0.10	0.00861	0.09	0.00730	0.004522	0.002831
5	LC1	2	0.1	-0.2074	0.00030	-0.175851	0.00025	-0.21081	-0.20259	-0.18839	-0.15809	0.000158	9.88E-05	132	0.10	0.21	0.01722	0.18	0.01460	0.009043	0.005662

- S. Once all the data are collected, unlock the model by selecting the Lock tool on the left hand side of the SAP2000 screen.
- 17. Analyze the fractured structure.
 - A. At mid-span along Gridline 9, replace the hinges of the outside longitudinal element (Gridline B) with FracOUT hinges according to Step 10.
 - B. At mid-span along Gridline 9, replace the hinges of the first interior longitudinal element (Gridline C) with FracInt hinges according to Step 10.



C. Repeat Step 15 for the Fractured Case and collect the data accordingly. 18. Post-process the data.

A. In the Excel Sheet, the following values need to be calculated for each step.

• Omega (Ω).

$$\Omega_{i} = \Omega_{i-1} + \left(\frac{1}{\# \ of \ Steps \ in \ Load \ case}\right)$$

• Longitudinal Chord Rotation of Interior and Exterior Girder.

- Chord Rot._{Single Span} = $-1 * \left(\frac{\delta_{CL}}{0.5 * L}\right)$ (rad).
- Transverse Deck Rotation.

$$\circ \quad \alpha_{2-3} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_2 - \delta_1}{w}\right) \text{ (rad).}$$

$$\circ \quad \alpha_{2-3} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_2 - \delta_1}{w}\right) \text{ (rad).}$$

- Where s = spacing between the interior top flanges of the inside and outside girders and w = spacing between the top flanges of the same girder.
- Applied Load.
 - o Calculate unit applied load or applied load at 1 Ω .

 - $\circ \quad Applied \ Load = Unit \ Applied \ Load * \Omega$
- B. Repeat Step A for the fractured case.
- C. Calculate the initial stiffness for the intact bridge and instantaneous stiffness for fractured bridge.
 - For the nonfractured condition (intact bridge), find the absolute displacement for the outside girder at an Ω value of 0.4:

Initial Stiffness =
$$\frac{0.4}{Absolute Displacement OG (at \Omega = 0.4)}$$
.

• For the Fractured case, add an additional column labeled stiffness:

Instantaneous Stiffness_{OG-Frac.i} =
$$\frac{\Omega_i - \Omega_{i-1}}{\delta_i - \delta_{i-1}}$$
.

- D. Failure of the structure occurs at the Ω of the fractured bridge at the first of the following criteria:
 - The instantaneous stiffness for the fractured outside girder is less than 5 percent of the initial stiffness of the intact outside girder.
 - The chord angle of the outside girder for a simple span or interior spans is greater than 2 degrees. The chord angle for exterior spans of multi-span bridges is greater than 3 degrees.
 - The transverse deck rotation is greater than 5 degrees.
- E. On a chart, plot the nonfractured outside and inside girder as well as the fractured outside and inside girder with displacement on the primary x axis and the total force on the primary y axis and Ω on the secondary y axis.



5.3 OVERALL RESULTS FOR BRIDGE 2

Table 5.1 and Figure 5.2 presents the overall results for Bridge 2. In addition to plotting the foregoing upper- and lower-bound yield line solutions and the grillage overstrength (load) versus deflection results, for the sake of completeness, FEM results have also been included in Table 5.1 and in Figure 5.2. Clearly, all methods indicate that Bridge 2 could be reclassified as nonfracture critical. It should be noted that while yield line results provide the collapse load capacity, the analysis is silent on the deformation limitations. Such results, however, are provided by both grillage and FEM solutions.

Method of Analysis	Overstrength Factors	Single Span (115 ft)				
Yield Line Theories	$\Omega^{Upper \; Bound}_{Yield \; Line}$	1.17				
Tield Line Theories	$\Omega^{Lower \; Bound}_{Yield \; Line}$	1.02				
Grillage Analysis	$\Omega_{Grillage}$	1.11				
FEM	Ω_{FEM}	1.65				

 Table 5.1. Results Summary of Bridge 2.



Figure 5.2. Comparison of the Results for Bridge 2, $L_x = 115$ ft.
6. BRIDGE 5

6.1 YIELD LINE ANALYSIS EXAMPLE OF BRIDGE 5

This section presents the steps to calculate the reserve capacity (overstrength factor) of Bridge 5 using the yield line analysis (grillage analysis is explained in the next sub-section). Figure 6.1 presents the geometric details of the two exterior spans of Bridge 5, which are further illustrated in detail in Appendix B.



(*a*) Bridge 5, Span 1 ($L_x = 140$ ft)



(*b*) Bridge 5, Span 2 ($L_x = 139.60$ ft)

Figure 6.1. Schematic Diagrams of Bridge 5 ($R_{\rm L}$ = 450 ft).

This section presents the stepwise procedure of the yield line analysis conducted to establish the upper-bound and lower-bound solution range for the overstrength factor as per the procedure described in Section 3.

The moment capacities of the deck slab are computed in the same way as described in Section 5. The overstrength factors are also calculated using the same procedure as mentioned in Section 5. Bridge 5 is a two-span bridge which is why the spans have one continuous support. The moment capacities of the intact section at the continuous support are computed as the positive and negative moment of the intact section using the similar procedure as that followed for calculating the moment capacities of the deck slab alone (explained in Section 5). The difference being the additional elements of the girder cross-section that contribute to the flexural strength. The intact cross-section of the bridge, including the girders, at the continuous supports is considered.

The geometric parameters namely B = the total width (breadth) of the deck slab, t = thickness and h = thickness of haunch are noted from the structural plans. The various material properties of concrete and steel are obtained from the bridge plans and the reports associated with the respective bridges. These properties include $f_c' =$ the specified compressive strength of concrete, $\varepsilon_{cu} =$ the maximum strain at the extreme concrete compression fiber (computed as per Section 22.2.2 of ACI-318 (2017) which states the "assumptions for concrete."), $f_y =$ the yield strength of steel of the reinforcement bars, $F'_y =$ the yield strength of the steel of STTG, E = the Young's modulus of steel of and ε_y and $\varepsilon_s =$ the yield strain that is the ratio of yield strength and Young's modulus of steel, of the reinforcement and the SSTG, respectively.

There are additional number of bars provided for extra strength at the supports that are denoted by "bent." The details of the reinforcement such as $\#_{top \ bars} =$ the number of top bars, $\#_{top \ bars \ bent} =$ the number of top bars at bent, $\#_{bottom \ bars} =$ the number of bottom bars, $d_{b \ top \ tra} =$ the diameters of the transverse top bars, $d_{b \ bot \ tra} =$ the diameters of the transverse bottom bars, $d_{b \ top \ long} =$ the diameters of the longitudinal top bars, $d_{b \ top \ long} =$ the diameters of the longitudinal top bars at bent, $d_{b \ bot \ long} =$ the diameters of the longitudinal bottom bars and the clear cover for each type of bar are recorded with suitable subscript to *cc*.

The STTG dimensions such as D_g = depth of girder (overall), b_{tf} = top flange width, t_{tf} = top flange thickness, b_w = web width, t_w = web thickness, b_{bf} = bottom flange width and t_{bf} = bottom flange thickness are also tabulated.

The composite area neutral axis is found by using n which denotes the ratio of yield strength of steel of reinforcement and that of the STTG to express the area computations in terms of an effective area. The positive moment capacity is found by the similar procedure for computing the compressive and tensile forces as explained in Section 5 whereas the negative moment capacity is found by using the equal area method of plastic analysis where the areas of compression and tension zones are calculated. The neutral axis for the positive moment capacity is obtained by equating the compressive and the tensile forces while the plastic neutral axis for the negative moment capacity is calculated by equating the areas in compression and tension. The depth of compression zones from the top is denoted by c and is indicated by blue highlighting. y = the portion of the width of web in compression and y' = the portion of the width of web in tension. The depths of the compressive and tensile forces from the neutral axis are computed for the positive moment capacity while the depths of the center of gravity of compressive and the tensile areas from the plastic neutral axis are calculated for the negative moment capacity. The neutral axis and the plastic neutral axis may lie either in the web or in the top flanges and those sections are denoted by the subscripts 1 and 2 for the top-half and the bottom-half of the element of girder in which the neutral axes lie, respectively.

For the positive moment capacity, the net tensile strain in the extreme steel components of the section respectively, are determined from a linear distribution (ACI-318 2017). The compressive and tensile forces are computed in the similar method explained in Section 5. The compressive forces and areas are denoted by a negative sign while the tensile forces and areas are denoted by a negative sign while the tensile forces and areas are denoted by a negative sign while the tensile forces and areas are denoted by a negative sign while the tensile forces and areas are denoted by positive sign. The concrete is assumed to be completely cracked for the negative moment capacity and is not considered for the tensile area. The positive moment capacity is obtained by taking the moments about the neutral axis and negative moment capacity is calculated by taking moments about the plastic neutral axis.

The yield line mechanism engages only the outer half of the bridge cross-section. Therefore, only half of the intact moment capacities are used. The positive intact moment capacity is used for obtaining the exact location of fracture in the end spans that is needed to develop the value of the fraction of the span length from the exterior support at which the girder is fractured, λ . The λ is set to 0.4 as a result of a detailed analysis explained in the TxDOT (0-6937) report on Fracture Critical Steel Twin Tub Girder Bridges. The negative intact moment capacity is used for the internal work done at the interior supports.

63

		MOMENT CAPACITY OF DECK SLAB					1
			(BRIDGE 5)				<u> </u>
		Descríptíon Parameter Data					
V V	1	Total Width	B in	360			
CTI	2	Thickness	t in	8			
д Ш У	3	Section width	<i>b</i> in	12			
RETE	4	Characterístíc Compressíve Strength	f_c' ksi	4			
NC1	5		β_1	0.85			
У С У	6	Straín	ε _{cu}	0.003			
R	F	Yield Strength	<i>fy</i> ksi	60			
ER	8	Young's Modulus	E ksi	29000			
RE	9	Strain ε_y 0.002070					
		Moment Computations		Longítu	dinal	Tra	insverse
	10			m'_x	m_x	m'_y	m_y
	11			Тор	Bottom	Тор	Bottom
	12	Bar No.		4	5	5	5
Ŀ	13	Díameter of Bar	d_b in	0.5	0.625	0.625	0.625
TA ATA	14	Area of Bar	$A_{\phi} \operatorname{in}^2$	0.2	0.31	0.31	0.31
S DE	15	Spacing	s in	_	_	5	5
BAI	16	No. of Bars	# _{bar}	41	33	_	_
NIN	17	Area of Steel	$A_s \operatorname{in}^2$	0.273	0.341	0.744	0.744
ORC	18	Clear Cover	cc in	2	1.25	2	1.25
	19	Effective depth (tension)	<i>d</i> in	5.125	5.8125	5.6875	6.4375
Ŕ	20	Effective depth (comp.)	<i>d'</i> in	2.1875	2.875	1.5625	2.3125
21 Depth of NA c in		1.06	1.06	1.44	1.81		

		MOMENT CAPACITY OF DECK SLAB					
		Monarent Compute	(BRID	(GES)	udínal	Trains	yeve e
	~~~	Moment Comput	1CLOVIS	Congu		170705	
	22			$m_x$	$m_x$	$m_y$	m _y
-	23			Тор	Bottom	Тор	Bottom
RAIN	24		$\varepsilon_{top}$	0.011466	0.005115	0.008829	0.000837
1 TS	25		$\varepsilon_{bot}$	0.003174	0.013406	0.000250	0.007682
	26	βıc	<i>a</i> in	0.90	0.90	1.23	1.54
10	27	Compression-Concrete	C _c kip	-36.86	-36.86	-50.03	-62.70
CES	28	Tension/	T _{top} kip	16.40	16.40	44.64	18.06
OR	29	Compression-Steel	T _{bot} kip	20.46	20.46	5.39	44.64
Ŧ	30	Equilibrium Check	$\Sigma T + C$ kip	0.00	0.00	0.00	0.00
RESULTS	31	Moment	<i>M_n</i> k-in/in	9.35	12.45	19.30	23.41
						I	
		<u>Remarks:</u>		Top Yíelded	Top Yíelded	Top Yielded	Top Not Yíelded
				Bottom Yielded	Bottom Yíelded	Bottom Not Yielded	Bottom Yíelded
Both steel ín Tensíon Ín Tensíon Tensíon ín T							Both steel ín Tensíon

		POSITIVE AND NEGATIVE MOMENT OF INTACT SECTION 3					
		(BRIDGE 5, SPANS 1 AND 2)					
	1		Deci	M	10ment	(a	
		Description	POSU	tive	Negativ	<i>1</i> 00	
л Л Л	1	Total width	B in	180	<i>B</i> 1n	180	
SEC T	2	Thíckness	t in	8	tin	8	
T S	3	Haunch thickness	<i>h</i> in	4	<i>h</i> in	4	
RIAL	4	Characterístic Compressive Strength	$f_c'$ ksi	4			
AT CI	5		$\beta_1$	0.85			
3 X	6	Strain	Е _{си}	0.003			
	F	Yield Strength	<i>f_y</i> ksi	60	$f_y$ ksi	60	
	8	Straín	Rebar $\varepsilon_y$	0.002070			
	9	No of Bars at Top	#top bars	20.5	#top bars	20.5	
	10	No of Bars at Top Bent			#top bars bent	20	
S	11	No of Bars at Bottom	#bot bars	16.5	#bot bars	16.5	
ALL	12	Clear Cover Top	<i>CCtop</i> in	2	<i>CCtop</i> in	2	
ET.	13	Clear Cover Bottom	<i>CC_{bot}</i> in	1.25	<i>CC</i> bot in	1.25	
4R I	14	Transverse Top Díameter	<i>d_{b top tra}</i> in	0.625	$d_{b top tra}$ in	0.625	
A U	15	Transverse Bottom Díameter	<i>d</i> b bot tra in	0.625	<i>d</i> b bot tra in	0.625	
CIN C	16	Longítudínal Top Díameter	<i>d</i> b top long in	0.5	<i>d</i> b top long in	0.5	
-OR	17	Longitudinal Top Diameter at Bents			$d_{b \ top \ long}$ 'in	0.625	
E N	18	Longítudínal Bottom Díameter	<i>d</i> b bot long in	0.625	<i>d</i> b bot long in	0.625	
Ŕ	19	Effective Depth Top	<i>d</i> _{top} in	2.88	$d_{top}$ in	2.88	
	20				<i>d</i> _{top bent} in	2.94	
	21	Effective Depth Bottom	<i>d</i> _{bot} in	5.81	<i>d</i> _{bot} in	5.81	
	22	Area of Steel for #4 Bars	<i>A</i> #4 in ²	0.196	A#4 in ²	0.196	
	23	Area of Steel for #5 Bars	<i>A</i> #5 in ²	0.307	<i>A</i> #5 in ²	0.307	
EL	24	Yield Strength	$F'_{\mathcal{Y}}$ ksi	50	$F'_{y}$ ksi	50	
<u><u></u></u>	25	Straín	$\mathcal{E}_{S}$	0.001720			
4L 6	26	Young's Modulus	Eksi	29000			
LLR.	27	$f_y/F'_y$	n	1.2	п	1.2	
enct	28	Effective Area of Steel for #4 Bars	A#4. eff in ²	0.236	$A$ #4. eff $\operatorname{in}^2$	0.236	
STR 8	29	Effective Area of Steel for #5 Bars	A#5. eff in ²	0.368	$A$ #5. eff $in^2$	0.368	

		POSITIVE AND NEGATIVE MOMENT OF INTACT SECTION 4						
		(BRIDG	IDGE 5, SPANS 1 AND 2)					
				Мо	ment			
		Description	Posítín	/e	Negatír	/e		
	30	Gírder Depth	$D_g$ in	55.75	$D_g$ in	58		
Ŋ	31	Top Flange Width	<i>b</i> tf in	18	<i>b</i> _{tf} in	18		
TAIL	32	Top Flange Thickness	<i>t</i> _{tf} in	1	<i>t</i> tf in	2.5		
S DE	33	web width	$b_w$ in	54	$b_w$ in	54		
DER	34	Web Thickness	$t_w$ in	0.5	$t_w$ in	0.6		
ЦR	35	Bottom Flange Width	<i>b</i> _{bf} in	56	<i>b</i> _{bf} in	56		
	36	Bottom Thickness	t _{bf} in	0.75	<i>t_{bf}</i> in	1.5		
Ś	37	Compression Zone	<i>c</i> in	12.89	<i>c</i> in	39.29		
ЕРТН	38	b _w in Comp.	<i>y</i> in	0.89	<i>y</i> in	37.79		
Id	39	$b_w$ in Tension	y'in	0.11	y'in	16.21		
	40		d Cconc in	7.41	<i>d Arebar top</i> in	27.84		
	41	Dístances of Compressíve	d Crebar top in	10.02	<i>d Arebar top bent</i> in	27.78		
CES	42	Forces and Tensile Forces	d Crebar bottom in	7.08	d Arebar bot in	24.90		
TAN	43	Trom Neutral Axis/ Dístances of Areas ín	<i>d C</i> _{tf1} in	0.45	$dA_{tf}$ in	17.46		
DISID	44	Tension and Compression	<i>d T_{tf 2}</i> in	0.05	$dA_{w1}$ in	8.11		
	45	Trom PNA	$d T_w$ in	54.05	$dA_{w2}$ in	18.89		
	46		$d T_{bf}$ in	54.48	<i>d</i> A _{bf} in	38.54		
	4F		E Crebar top	0.002331				
	48		E Crebar bottom	0.001648				
ΑIΛ	49		E Ctf	0.000104				
L R	50		$\mathcal{E} \mathcal{C}_{W1}$	0.000012				
Ŋ	51		$\varepsilon T_{w2}$	0.012578				
	52		E Tbf	0.012677				

POSITIVE AND NEGATIVE MOMENT OF INTACT SECTION					5	
		(BRIDG	C 3, SPANS I A	Mov	nent	/ 9
		Description	Positive		Negatí	ve
	53		<i>0.85f_c' βBT</i> kip	-4162	As top kip	5
	54				As top bent kip	7
¥	55	Compression <i>Fc</i> / Tension <i>Ar</i>	As topfy kip	-273	As bottom kip	6
ARE	56		$A_{sbottom}f_y$ kip	-270	A _{tf} kip	90
DRCE	57		<i>2b_{tf}yF'_y</i> kip	-97	<i>A_{w1}</i> kip	18
F(	58		$2b_{tf}y'F'_y$ kip	1.39		
	59	Tension <i>F</i> 1/ Compression <i>Ac</i>	<i>2b_wt_wF'_y</i> kip	2700	$A_{W2}$ kip	-43
	60		<i>bыtыF'y</i> kip	2100	Abf kip	-84
N	61		Fc kip	-4800	<i>A</i> _T kip	127
CHECH	62	Equilibrium of Forces/Areas	$F_T$ kip	4800	$A_C$ kip	-127
)	63		$F_C + F_T$ kip	0	$A_T + A_C$ kip	0
ENT	64	(Half Section)	$0.5 M_p^+$ k-ft	24658	$0.5 M_p^-$ k-ft	26450
MOM	65	(Full Section)	$M_p^+$ k-ft	49317	$M_p^-$ k-ft	52901

COMPUTATION OF OVERSTRE		COMPUTA	TION OF OVERSTRENGTH FACTOR EX	TERIOR	: 2-	6
			SPAN CASE (BRIDGE 5, SPAN 1)			9
		Parameter	Formula/Definition/Equation		Dat	а
	1	$L_x$	Span Length	Span Length 140.00 ft		
	$2 R_{CL}$ Radíus of center líne		450	ft		
<u>8</u> 3		В	Width	30	ft	
OME	4	$L_x^*$	Outer region length $L_x^* = \left(1 + \frac{B}{4R_{C_L}}\right)L_x$	142.33	ft	
dfi Ufi	5	S	Inter-Girder Spacing	9.7	ft	
	6	b	Width of Girder + Edge	10	ft	
	F	t	Deck Thickness	8	in	
	8	m _x	Longitudinal Positive Moment per ft	12	k-in/i	n
	9	<i>m′</i> _{<i>x</i>}	Longítudínal Negatíve Moment per ft	9	k-in/i	n
	10	$m_y$	Transverse Positive Moment per A	23	k-in/i	n
	11	<i>m′y</i>	Transverse Negatíve Moment per ft	19	k-in/i	n
	12	$0.5M_p^-$	- Negative Moment at Support		k-ft	
₫ <u>₩</u>	13	λ	Fraction of Length from the exterior support at which girder is fractured	0.40		
K DONE,	14	tan θ	$\tan \theta = \sqrt{\frac{m'_y + m_y}{m'_x + m_x}}$	1.40	(Θ =	54.4°)
- WORI	15	tan α	$\tan \alpha = \frac{2s}{L_x}$	0.14	(Θ =	7.9°)
ERNAU	16	$k_{bound}^{upper}$	$\left[1+2\frac{\tan\alpha}{\tan\theta}\right] = 1 + \frac{4s}{L_x}\sqrt{\frac{m'_x + m_x}{m'_y + m_y}}$	1.20		
LN.	17	k ^{lower} k ^{bound}	$\left[1 + 2\frac{\tan^{2}\alpha}{\tan^{2}\theta}\right] = 1 + \frac{8s^{2}}{L_{x}^{2}}\sqrt{\frac{m'_{x} + m_{x}}{m'_{y} + m_{y}}}$	1.02		
	18	IWD _{upper}	$\left(m_{y}'+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{upper}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}+\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	689	k-ft*	
	19	IWD _{lower}	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{lower}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	634	k-ft *	

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of internal work is in k-ft

		COMP	COMPUTATION OF OVERSTRENGTH FACTOR 7					
		TTX3	ZIOR: 2-SPAN CASE (BRIDGE 5, SPAN	1)	9			
		Parameter	Formula/Definition/Equation	Þ	vata			
	20	DL	Dead Load Factor	Dead Load Factor 1.25				
	21	LL	Live Load Factor 1.75					
	22	SAF	Stiffener Allowance Factor	Stiffener Allowance Factor 1				
	23	γc	Unit weight of reinforced concrete 0.15 k		kcf			
	24	$W_u$ Area load due to reinforced concrete + lane load $DL\gamma_c \frac{t}{12} + LL \cdot \frac{0.64}{12}$ 0.218		ksf				
	25	$\gamma_s$	Unit weight of steel	0.49	kcf			
EWI	26	$V_g$	Volume of Girder	145	ft ³			
N Ш	27	Ar	Area of Rail Cross-Section (T4(S))	1.25	ft ²			
DO	28	Vr	Volume of Rail = $L_x A_r$	Volume of Rail = $L_x A_r$ 178 f				
VORK	29	W _x	$1.25 \left(1.15 V_g \gamma_s + V_r \gamma_c\right) / L_x \qquad \qquad 0.$		k/ft			
FRNAL V	30	<i>y</i> (Lane 2)	(b + s - 15) for $(b + s) < 21(b + s - 18)$ for $(b + s) > 21$	4.84	ft			
EXTE	31	Klane	$1 + 0.5 \frac{y}{s}$ for $(b+s) < 21; 1 + \frac{y}{s}$ for $(b+s) > 21$	1.25				
	32	EWD _{HS-20}	$\left(168 - \frac{261.33}{\lambda L_{\rm x}} - \frac{1045.33}{(1-\lambda)L_{\rm x}}\right) K_{Lane}$	189	k-ft			
	33	WET	$w_u L_x(b+0.5s) + W_x L_x + 2EWD_{HS20}$	979	k-ft			
	34	EWD	0.5 Wet	490	k-ft *			
MLTS	35	$arOmega_{upper}$	IWD _{upper} /EWD	1.41				
RESI	36	$arOmega_{lower}$	IWD _{lower} /EWD	1.29				

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of external work is in k-ft

COMPUTATION OF OVERSTRENGTH FACTOR		ર	8		
		EXTE	RIDR: 2-SPAN CASE (BRIDGE 5, SPAN :	2)	9
		Parameter	Formula/Definition/Equation		Data
	1	L _x	Span Length	139.60	ft
	2	R _{CL}	Radíus of center líne	450	ft
ж К		В	Width	30	ft
SOMET	4	$L_x^*$	Outer region length $L_{\chi}^{*} = \left(1 + \frac{B}{4R_{C_{L}}}\right)L_{\chi}$	141.93	ft
G.	5	S	Inter Gírder Spacing	10	ft
	6	b	Width of Girder + Edge	10	ft
	F	t	Deck Thickness	8	in
	8	$m_x$	Longitudinal Positive Moment per A	12	k-in/in
	9	m'x	Longítudinal Negative Moment per ft	9	k-in/in
	10	$m_y$	Transverse Positive Moment per ft	23	k-in/in
	11	<i>m′y</i>	Transverse Negative Moment per ft	19	k-in/in
	12	0.5Mp ⁻	Negative Moment at Support	26450	k-ft
dw)	13	λ	Fraction of Length from the exterior support at which girder is fractured	0.40	
K DONE	14	tan θ	$\tan \theta = \sqrt{\frac{m'_y + m_y}{m'_x + m_x}}$	1.40	(Θ = 54.4°)
IL WOR	15	tan α	$\tan \alpha = \frac{2s}{L_x}$	0.14	( $\Theta = 7.9^{\circ}$ )
TERNA	16	k ^{upper} k _{bound}	$\left[1+2\frac{\tan\alpha}{\tan\theta}\right] = 1 + \frac{4s}{L_x}\sqrt{\frac{m'_x + m_x}{m'_y + m_y}}$	1.20	
Ž	17	k ^{lower} k ^{bound}	$\left[1+2\frac{\tan^2\alpha}{\tan^2\theta}\right] = 1 + \frac{8s^2}{L_x^2}\sqrt{\frac{m'_x + m_x}{m'_y + m_y}}$	1.02	
18 <i>IWDupper</i> $\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{upper}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}+\frac{m_{y}b}{2s}$		$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{upper}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}+\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	683	k-ft *	
	19	IWDlower	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{lower}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	628	k-ft *

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of internal work is in k-ft

		COMP	COMPUTATION OF OVERSTRENGTH FACTOR 9					
		EXTER	CIOR: 2-SPAN CASE (BRIDGE 5, SPAN :	2)	9			
		Parameter	Parameter Formula/Definition/Equation Dat					
	20	DL	Dead Load Factor	1.25				
	21	LL	Live Load Factor	1.75				
	22	SAF	Stíffener Allowance Factor	1				
	23	γc	Unit weight of reinforced concrete	kcf				
	24	Wu	Area load due to reinforced concrete + lane load $DL\gamma_{c}\frac{t}{12} + LL \cdot \frac{0.64}{12}$	0.218	ksf			
Ą	25	γs	Unit weight of steel	0.49	kcf			
EV.	26	$V_g$	Volume of Girder	152	ft ³			
NE,	27	Ar	Area of Rail Cross-Section (T4(S))	1.25	ft ²			
р Д Х	28 $V_r$ Volume of Rail = $L_x A_r$		177	ft ³				
WORI	29	W _x	$1.25 \left(1.15 V_g \gamma_s + V_r \gamma_c\right) / L_x$	0.99	k/ft			
NAL	30	<i>y</i> (Lane 2)	(b + s - 15) for $(b+s) < 21(b + s - 18)$ for $(b+s) > 21$	4.84	ft			
EXTER	31	Klane	$1 + 0.5 \frac{y}{s}$ for $(b+s) < 21; 1 + \frac{y}{s}$ for $(b+s) > 21$	1.25				
	32	EWD _{HS-20}	$\left(168 - \frac{261.33}{\lambda L_{x}} - \frac{1045.33}{(1-\lambda)L_{x}}\right) K_{Lane}$	189	k-ft			
	33	Wet	$w_u L_x(b+0.5s) + W_x L_x + 2EWD_{HS20}$	983	k-ft			
	34 EWD 0.5 W _{ET}		0.5 W _{ET}	491	k-ft *			
MLTS	35	$\Omega_{upper}$	IWDupper/EWD	1.39				
RESI	36	$\Omega_{lower}$	IWD _{lower} /EWD	1.28				

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of external work is in k-ft

# 6.2 GRILLAGE ANALYSIS EXAMPLE OF BRIDGE 5

The following steps are explained to conduct the computational implementation of grillage analysis of Bridge 5.

Location	<b>Top Flange</b>		Web		<b>Bottom Flange</b>	
Location ft	Width	Thickness	Width	Thickness	Width	Thickness
11	in.	in.	in.	in.	in.	in.
0–105	18	1.00	54	0.5	56	0.75
105-122	18	1.00	54	0.5625	56	1.250
122-140	18	1.75	54	0.5625	56	1.250
140–157	18	1.75	54	0.5625	56	1.250
157–174	18	1.57	54	0.5625	56	1.250
174–192	18	1.00	54	0.5625	56	0.75
192–280	18	1.00	54	0.5	56	0.75

1. Gather bridge geometry and material information.

Location	Parameter	<b>Description/Value</b>
	Location	Travis County, I-35
	Year Designed/Year Built	1998/2002
Dridaa	Design Load	
Бпаде	Length, ft	279.58
	Spans, ft	140, 139.58
	Radius of Curvature, ft	450
	Width, ft	30
Dealr	Thickness, in.	8
Deck	Haunch, in.	4
	Rail Type	T4(S)
	# of Bar Longitudinal Top Row (#4)	40
	# of Bar Longitudinal Bottom Row (#5)	36
	# of Bar Longitudinal Top Row (#4)	41
	@support	71
Rebar	# of Bar Longitudinal Top Row (#5)	40
	@support	
	# of Bar Longitudinal Bottom Row (#5)	36
	@support	
	Transverse Spacing Top Row (#5), in.	5
	Transverse Spacing Bottom Row (#5), in.	5
Girder	CL of Bridge to CL of Girder (in.)	56.5
Girder	CL of Top Flange to CL of Top Flange (in.)	83

Concrete (4 ksi)				
Stress	Strain			
(ksi)	(1/in)			
-4	-3.79E-03			
-4	-3.56E-03			
-4	-2.69E-03			
-4	-1.78E-03			
-3.8205	-1.40E-03			
-2.8718	-8.69E-04			
-0.6403	-1.78E-04			
0	0			
0.378	1.06E-04			
0.378	1.16E-03			

2.	Determine	material	constitutive	behavior.

Rebar	(60 ksi)
Stress	Strain
(ksi)	(1/in)
-87.9	-0.095
-87.9	-0.0944
-86.6	-0.0761
-78	-0.0386
-60.7	-9.80E-03
-60.3	-2.08E-03
0	0
60.3	2.08E-03
60.7	9.80E-03
78	0.0386
86.6	0.0761
87.9	0.0944
87.9	0.095

Steel	(50 ksi)
Stress	Strain
(ksi)	(1/in)
-71.6	-0.1
-71.6	-0.097
-71.6	-0.095
-71.6	-0.0946
-70.3	-0.0764
-62.5	-0.039
-50	-0.0196
-50	-1.72E-03
0	0
50	1.72E-03
50	0.0196
62.5	0.039
70.3	0.0764
71.6	0.0946
71.6	0.095
71.6	0.097
71.6	0.1

3. Create a coordinate system for the half width of the span for each cross-section in 1. An example of the first cross-section is shown below.



4. Create a cylindrical coordinate system for the curved bridge and ensure that the middle transverse divisions are 7 ft because this will aid is applying the truck load whose axles are separated by 14 ft.

A. # of Segments_{per span} = 
$$\left(\frac{Length (ft)*12}{84 in (7ft)}\right)$$
 rounded to nearest even number:  
# of Segments_{span 1&2} =  $\left(\frac{140*12}{84}\right)$  = 20.  
B. End Segment Length =  $\frac{((Length*12)-(\# of Segments*84))}{2}$ :  
End Segment Length =  $\frac{((140*12) - (14*84))}{2}$  = 0 in.

Therefore, the end segments are also equal to 84 in.

C. Theta =  $\frac{Total Length}{Radius}$ .

$$Theta = \frac{280}{450} = 0.6222 \ rad \ or \ 36.65 \ degrees.$$

D. Determine the radial offsets using the outside edge, the outside flange, the inner flange, and CL of the bridge.

Offsets (in.)							
Edge	180						
Outside Flange	139.5						
Inner Flange	56.5						
CL of Bridge	0						

Radial Spacing (in.)								
А	5580	CL + Edge						
В	5539.5	CL + OF						
С	5456.5	CL + IF						
Center Line	5400	or 450 (ft)						
D	5343.5	CL – IF						
Е	5260.5	CL – OF						
F	5220	CL – Edge						

- E. The longitudinal spacing along theta is determined by converting the longitudinal segment lengths into degrees.
  - All segments are 84 in. The total length is 280 ft, or 3360 in. Radial Spacing (rad) =  $\frac{Long.Spacing}{Radius}$

  - Radial Spacing (degree) = Radius Spacing (rad)  $*\frac{180}{\pi}$

Long. Spacing (in.)	Radial Spacing (rad.)	Radial Spacing (deg.)	Cross- Section		
0	0.0000	0.000	Long1		
84	0.0156	0.891	Long1		
168	0.0311	1.783	Long1		
252	0.0467	2.674	Long1		
336	0.0622	3.565	Long1		
420	0.0778	4.456	Long1		
504	0.0933	5.348	Long1		
588	0.1089	6.239	Long1		
672	0.1244	7.130	Long1		
756	0.1400	8.021	Long1		
840	0.1556	8.913	Long1		
924	0.1711	9.804	Long1		
1008	0.1867	10.695	Long1		
1092	0.2022	11.586	Long1		
1176	0.2178	12.478	Long2		
1260	0.2333	13.369	Long2		
1344	0.2489	14.260	Long2		
1428	0.2644	15.152	Long3		
1512	0.2800	16.043	Long3		
1596	0.2956	16.934	Long4		
1680	0.3111	17.825	Long4		
1764	0.3267	18.717	Long4		
1848	0.3422	19.608	Long3		
1932	0.3578	20.499	Long3		
2016	0.3733	21.390	Long2		
2100	0.3889	22.282	Long2		
2184	0.4044	23.173	Long2		
2268	0.4200	24.064	Long1		
2352	0.4356	24.955	Long1		
2436	0.4511	25.847	Long1		

2520	0.4667	26.738	Long1
2604	0.4822	27.629	Long1
2688	0.4978	28.521	Long1
2772	0.5133	29.412	Long1
2856	0.5289	30.303	Long1
2940	0.5444	31.194	Long1
3024	0.5600	32.086	Long1
3108	0.5756	32.977	Long1
3192	0.5911	33.868	Long1
3276	0.6067	34.759	Long1
3360	0.6222	35.651	Long1

- Int. Transverse Element width = 84 in.
- End Transverse Element =  $84 \left(\frac{84}{2}\right) = 42$  in.
- *Pier Tansverse Element* = 84 + 84 84 = 84 *in.*
- 5. Input the coordinate system into SAP2000.
  - A. Select File > New Model > Blank model (making sure units are in kips and inches).
  - B. Right click on the blank workspace and select Edit Grid Data > Modify/Show System > Quick Start >Cylindrical.
    - In the Number of Gridlines panel, set "Along Z = 1."
    - In the Grid Spacing panel, set "Along Z = 1."
    - Select OK.
    - Delete all R and T Coordinates that were generated.
  - C. Add correct coordinates for R.

All radial coordinates are A, B, C, D, E, and F where A = outside edge; B = centerline of outer top flange of outside girder; C = Centerline of inner top flange of outside girder; D = Centerline of inner top flange of inside girder and F = inside edge.

- D. Add correct coordinates for T
  - All theta coordinates for T (0 to 3360 in.).
  - Click OK.
- E. The grid system is now formed.





- 6. Define material in SAP200.
  - A. Click Define > Materials- > Add New Material > Material Type (Steel, Concrete, or Rebar) > Standard (User) > OK.
  - B. At the bottom of the window, select the box that states, "Switch to Advanced Properties."
  - C. In the open window, name the material "Concrete," "Steel," or "Rebar" depending on which material is being defined. Then click "Modify/Show Material Properties."
  - D. In the Material Property Data window, click the Nonlinear Material Data icon.
  - E. In the Nonlinear Material Data window select the Convert to User Defined icon.
  - F. Input the number of number of data points for the stress strain behavior (10 for concrete, 13 for rebar, 17 for steel).
  - G. Input the data points for the stress strain behavior.
  - H. Select "OK."

	Name		Material Type				
Concr	ete		Concrete				
Hystere	sis Type	Drucker-Prager Pa	arameters	Units			
Taked	a v	Friction Angle	0.	0. Kip, in, F			
		Dilatational Angle					
Stress-	Strain Curve Defir	ition Options					
	rametric			Convert To User Defined			
-							
● Us User St Numbe	er Defined ress-Strain Curve er of Points in Stre	Data ss-Strain Curve		10			
Us User St Numbe	er Defined ress-Strain Curve er of Points in Stre Strain	Data ss-Strain Curve Stress	Point ID	10			
User St Number 1	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03	Data ss-Strain Curve Stress -4.	Point ID -E				
User St Number 1 2	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03	Data ss-Strain Curve Stress -4. -4.	Point ID -E				
User St Number 1 2 3	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03	Data ss-Strain Curve Stress -4. -4. -4. -4.	Point ID -E				
User St Number 1 2 3 4	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03	Data ss-Strain Curve Stress -4. -4. -4. -4. -4.	Point ID -E -C				
User St Number	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 8.600E.04	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -3.8205 2.9718	Point ID E				
<ul> <li>Us</li> <li>User St</li> <li>Number</li> <li>1</li> <li>2</li> <li>3</li> <li>4</li> <li>5</li> <li>6</li> <li>7</li> </ul>	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -4. -3.8205 -2.8718 0.6403	Point ID -E -C				
<ul> <li>Us</li> <li>User St</li> <li>Number</li> <li>1</li> <li>2</li> <li>3</li> <li>4</li> <li>5</li> <li>6</li> <li>7</li> <li>8</li> </ul>	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04 0	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -4. -4. -4.	Point ID -E -C	10			
<ul> <li>Us</li> <li>User St</li> <li>Number</li> <li>1</li> <li>2</li> <li>3</li> <li>4</li> <li>5</li> <li>6</li> <li>7</li> <li>8</li> <li>9</li> </ul>	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04 0. 1.060E-04	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -3.8205 -2.8718 -0.6403 0. 0. 0.378	Point ID -E -C A B	10			
<ul> <li>Us</li> <li>User St</li> <li>Number</li> <li>1</li> <li>2</li> <li>3</li> <li>4</li> <li>5</li> <li>6</li> <li>7</li> <li>8</li> <li>9</li> <li>10</li> </ul>	er Defined ress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04 0. 1.060E-04 1.160E-03	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -3.8205 -2.8718 -0.6403 0. 0. 0.378 0.378 0.378	Point ID -E -C A B F	10			

I. Repeat this process again for the remaining materials.

- 7. Define frame cross-sections in SAP2000.
  - A. Click Define > Section Properties > Frame Sections > Add New Properties.
  - B. In the Frame Section Properties drop-down box, select "Other" and click Section Designer. In the SD Section Designer window, name the section B5Long1 and click the Section Designer icon.
  - C. Using the Polygon feature, draw the features of the half width of the bridge from Step 3. These features include one rail, two concrete deck pieces, two concrete haunches, two top flanges, two webs, and two pieces of the bottom flange.
    - To change material types for the polygons, right-click on the polygon and select the desired material type from the material drop-down menu.
    - To change the coordinates of the polygon's nodes, use the Reshaper tool.
  - D. Add in the longitudinal rebar to both concrete deck elements by using the Line Bar from the Draw Reinforcing Shape tool. From the design drawings, it can be

determined that there are seven #4 top bars and nine #5 bottom bars in the outer concrete deck element and 11 #4 top bars and 11 #5 bottom bars in the inner concrete element. At the pier support, nine additional #5 top bars are added to the top of the outer concrete deck and 10 additional #5 top bars are added to the inner concrete deck with a 2 in. top cover and transverse reinforcement. The top bars are located at 5.0625 in., with a 1.25 in. bottom cover and transverse reinforcement. The top bars are located at 2.1875.



E. Click Done (Below is an example of the 1st cross-section).

F. Repeat this process for the remaining longitudinal elements.

G. Repeat the process for the transverse elements.

- In the SD Section Designer window, select Modifiers and set Mass and Weight to 0 to not double-count the dead weight.
- The interior transverse members are 84 in. wide (end members are 42 in. wide), with #5 rebar at 5 in. spacing at 5.6875 in. and 1.5625 in.



H. To generate an exterior longitudinal member and an interior longitudinal member, make two copies of the B5Long1 section. Label one Long1Out and one Long1Int.



• For the Long1Out, delete every element right of the CL.

• For the Long1Int, delete every element left of the CL.



- Repeat this process for all cross-sections.
- I. To generate a simulated fracture section, make copies of Long1Out and Long1Int.

- The reason Long1Out and Long1Int are chosen for the fractured section is because the fracture occurs at 0.4*L, or 672 in. On the radial spacing table, Long1 is the section used at 672 in.
- Name the Long1Out copy Frac1Out.

Delete the bottom flange, web, and top flange of the steel tub.



• Name the LongInt copy FracInt.

Delete the bottom flange, web, and top flange of the steel tub.



- 8. Generate plastic hinges for frame elements in SAP2000.
  - A. Define > Section Properties > Frame Sections.
  - B. Select the desired cross-section. Hinges will need to be made for the Long(1–4)Out, Long(1–4)Int, FracOut, FracInt, Trans, and TransEnd.
  - C. Once selected, click Modify/Show Property > Section Designer.
  - D. Once in the section designer, select the Moment Curvature Curve tool.
  - E. In the Moment Curvature Curve window, select Details.
    - Copy the moment curvature data to an Excel file.

- Select OK.
- F. In the Moment Curvature Curve window, change the angle (deg) to 180, then select Details.
  - Copy the moment curvature data to the same Excel file as the one used previously.
  - Select OK.
- G. Generate a Normalized Moment Curvature Diagram.
  - Normalize the moments by dividing each of the positive moments by the maximum positive moment and the negative moment by the maximum negative moment, and divide the curvatures by the curvatures corresponding to the maximum and negative moments.
  - Plot the normalized positive and negative moment curvatures on a chart.
  - Create a hinge moment curvature plot on the same chart with four positive moment points and four negative moment points without generating a negative slope.

A	A	8	с	D	E	F	G	н	- 1 I	<u></u>	К	L.	M	N	0	P	Q	R	s	T	U	V	W
1 2	Positive																						
3	Concrete !	Neutral A:	Steel Strail	Tendon St	Concrete	Steel Con	Steel Ten:	Prestress	Net Force	Curvature	Moment		M										
4	.0	0	0	0	0	0	0	0	0	0	0		0	0.00									
5	-6.42E-04	-0.0653	9.56E-04	0	-849.525	-71.5223	921.2336	0	0.1867	1.86E-05	56261		0.447094	0.14							15		
6	-1.53E-03	1.4924	2.46E-03	0	-1758	-130.476	1888.462	0	-0.0221	4.66E-05	111681		0.887505	0.36									
7	-2.49E-03	4.6917	4.70E-03	0	-2119	-90.6898	2208.734	0	-5.01E-01	8.39E-05	121852		0.968332	0.64							1 24		
8	-3.67E-03	6.2978	7.52E-03	0	-2351	-110.929	2462.295	0	0.0725	0.000131	125837		1	1.00							1	*******	
9	-5.21E-03	6.4308	0.0108	0	-2439	-154.905	2593.535	0	-0.0456	0.000186	121679		0.966957	1.43							0.5		
10	-7.04E-03	6.4221	0.0145	0	-2495	-209.435	2703.974	0	-0.4069	0.000252	119891		0.952748	1.93									
11	-9.17E-03	6.2948	0.0188	0	-2537	-256.469	2793.547	0	-0.0143	0.000326	119168		0.947003	2.50							0		
12	-0.0225	-20.3633	0.0127	0	0	-1461	1460.759	0	-0.0104	0.00041	82471		0.65538	3.14		-4	5,00 -	30.00	-20.00	-10.00	0.00	10.00	20.00
13	-0.0275	-20.1838	0.0157	0	0	-1465	1465.611	0	0.1893	0.000503	82810		0.658074	3.86							0.5		
14	-0.0329	-19.9925	0.019	0	0	-1470	1470.333	0	0.2082	0.000606	83150		0.660775	4.64							4		
15	-0.0393	-20.4119	0.0222	0	0	-1494	1493.947	0	0.0502	0.000718	84541		0.671829	5.50							- ch		
16	-0.0464	-20.868	0.0256	0	0	-1533	1533.258	0	0.0173	0.000839	86803		0.689805	6.43					-	- A-B-B-B-B-B-B-B-B-B-B-B-B-B-B-B-B-B-B-	erow.		
17	-0.0537	-20.9742	0.0294	0	0	-1590	1590.45	0	0.3643	0.000969	90046		0.715576	7.43			-	- 0					
18	-0.0615	-21.0802	0.0336	0	0	-1653	1653.258	0	0.0119	0.001109	93606		0.743867	8.50							-1.3		
19	-0.07	-21.2377	0.0379	0	0	-1718	1718.782	0	0.3503	0.001258	97263		0.772928	9.64									
20	-0.0786	-21.1113	0.0428	0	0	-1771	1771.636	0	0.1748	0.001416	100205		0.796308	10.85									
21	-0.0875	-20.8117	0.0484	0	0	-1816	1816.259	0	0.1105	0.001584	102695		0.816095	12.14		-1.35	-35						
22	-0.0973	-20.8379	0.0537	0	0	-1849	1849.515	0	0.1005	0.001761	104568		0.83098	13.49		-1.35	-25						
23	-0.1077	-20.877	0.0593	0	0	-1883	1882.613	0	0.0149	0.001948	106436		0.845824	14.93		-1	-1						
24																+0.7	-0.56						
25																0	0						
26	Negative															0.97	0.64						
27	Concrete !	Neutral A:	Steel Strail	Tendon St	Concrete	Steel Com	Steel Ten:	Prestress	Net Force	Curvature	Moment					1	1						
28	0	0	0	0	0	0	0	0	0	0	0		0	0.00		1	3						
29	+7.09E-04	13.306	4.99E-04	0	-341.349	-577.707	235.8451	0	-683.211	1.86E-05	35333		-0.44896	-0.22		1	13						

H. Define Hinge Length.

The hinge length is one half of the section depth.

 Hinge_{long} = 0.5 * (Deck thickness + haunch height + top flange thickness + web height + bottom flange thickness).

 $Hinge_{long} = 30$  in.

•  $Hinge_{Frac} = 0.5 * (Deck thickness + haunch height).$ 

 $Hinge_{Frac} = 6$  in..

•  $Hinge_{Trans} = 0.5 * (Deck thickness).$ 

 $Hinge_{Trans} = 4$  in.

- I. Make the plastic hinge in SAP2000.
  - Select Define > Section Properties > Hinge Properties > Add New Properties.
  - In the Type window, select Moment Curvature and input the corresponding correct hinge length.
  - In the Moment Curvature table, insert the four positive and four negative normalized moment curvatures and the zero point.
  - Uncheck the symmetric box and select the Is Extrapolated option in the Load Carrying Capacity beyond Point E window.
  - In the Scaling for Moment and Curvature window, insert the maximum positive moment and corresponding curvature as well as the maximum negative curvature and corresponding curvature.
  - In the Acceptance Criteria, use the values 1, 2, and 3 for Immediate Occupancy, Life Safety, and Collapse Prevention in the positive column and -1, -2, and -3 for the negative column.
  - Repeat for all remaining frame sections (Long(1–4)Out, Long(1–4)Int, FracOut, FracInt, Trans, and TransEnd).

					туре				
Point	Moment/SF	Curvature/SF	^		O Moment - Rotati	DN			
E-	-1.35	-35.			Moment - Curva	ture			
D-	-1.35	-25.			Hinge Length	35.			
C-	-1.	-1.			Pelative Length				
В-	-0.7	-0.56			Relative	Length			
А	0.	0.			Hysteresis Type And	Parameters			
В	0.97	0.64							
С	1.	1.		Cummatria	Hysteresis Type	Isotropic N			
D	4								
oad Car O Dro Is E caling fo	rying Capacity Bey ps To Zero xtrapolated	3. 13 yond Point E	~		No Parameter Hysteresis Ty	s Are Required For This pe			
E oad Car O Dro O Is E caling fi	rying Capacity Bey ps To Zero xtrapolated or Moment and Cur	3. 42 yond Point E	Posit 125837.	ive Negative	No Parameter Hysteresis Ty	s Are Required For This pe			
oad Car O Dro Is E caling fi Use	r, rying Capacity Bey ps To Zero xtrapolated or Moment and Cur e Yield Moment	yond Point E	Posit 125837.	ive Negative 78699.	No Parameter Hysteresis Ty	s Are Required For This pe			
caling fr Caling fr Use Caling fr	r, rying Capacity Bey ps To Zero xtrapolated or Moment and Cur e Yield Moment e Yield Curvature eel Objects Only)	3. 42 yond Point E rvature Moment SF Curvature SF	Posit 125837. 1.305E-0	ive Negative 78699. 04 8.390E-05	No Parameter Hysteresis Ty	s Are Required For This pe			
oad Car O Dro O Is E caling fi Use (St .cceptar	rying Capacity Bey ps To Zero xtrapolated or Moment and Cur e Yield Moment e Yield Curvature eel Objects Only) nce Criteria (Plastic	3. 13 vond Point E rvature Moment SF Curvature SF c Curvature/SF)	Posit 125837. 1.305E-0 Posit	ive Negative 78699. 14 8.390E-05	No Parameter Hysteresis Ty	s Are Required For This			
oad Car Dro Is E caling fo Use (St .cceptar	rying Capacity Bey ps To Zero xtrapolated or Moment and Cur e Yield Moment e Yield Curvature eel Objects Only) nce Criteria (Plastic mmediate Occupar	3. 42 yond Point E rvature Moment SF Curvature SF c Curvature/SF) icy	Posit 125837. 1.305E-0 Posit 1.	tive Negative 78699. 04 8.390E-05 tive Negative -1.	No Parameter Hysteresis Ty	s Are Required For This			
oad Car O Dro Is E caling fr Use (St .cceptar Ir L	rying Capacity Bey ps To Zero xtrapolated or Moment and Cur e Yield Moment e Yield Curvature eel Objects Only) nce Criteria (Plastic nmediate Occupar	3. 13 vond Point E rvature Moment SF Curvature SF Curvature/SF) icy	Posit 125837. 1.305E-0 Posit 1. 2.	ive Negative 78699. 14 8.390E-05 ive Negative -1. -2.	No Parameter Hysteresis Ty	s Are Required For This pe Cancel			

- 9. Assign frame members to grid.
  - A. Select the Draw tab > Quick Draw Frame/Cable/Tendon.

- B. In the Section drop-down menu, select Long1Out.
- C. Click on every grid segment on the second-to-last longitudinal grids (B and E) from Transverse Grids 1–15 and 28–41.
- D. Change the section to Long1Int and repeat Step C but for the two interior longitudinal grids (C and D).
- E. In the Section drop-down menu, select Long2Out.
- F. Click on every grid segment on the second-to-last longitudinal grids (B and E) from Transverse Grids 15–18 and 25–28.
- G. Change the section to Long2Int and repeat Step E but for the two interior longitudinal grids (C and D).
- H. In the Section drop-down menu, select Long3Out.
- I. Click on every grid segment on the second-to-last longitudinal grids (B and E) from Transverse Grids 18–20 and 23–25.
- J. Change the section to Long3Int and repeat Step I but for the two interior longitudinal grids (C and D).
- K. In the Section drop-down menu, select Long4Out.
- L. Click on every grid segment on the second-to-last longitudinal grids (B and E) from Transverse Grids 20–23.
- M. Change the Section to LongInt and repeat Step L but for the two interior longitudinal grids (C and D).
- N. Change the Section to TransEnd and repeat Step C but for the end transverse grids (1 and 41).
- O. Change the Section to Trans and repeat Step C but for all other transverse grids (2–40).



- 10. Assign hinges to frame elements.
  - A. The longitudinal hinges are placed at the ends of the longitudinal frame elements or at a relative distance of 0 and 1.
  - B. The transverse hinges are placed at a distance of half a top flange width away from the node.

• Hinge Loc._{F to E} = 
$$\frac{half flange width}{Element Width} = 1 - \frac{\frac{18}{2}}{40.5} = 0.7778.$$

- Hinge Loc._{E to D} and C to B =  $\frac{half flange width}{Element Width} = \frac{18/2}{83} = 0.1084$  and (1 0.1084) or 0.8916.
- 0.1084) or 0.8916. • Hinge  $Loc_{.D \ to \ C} = \frac{half \ flange \ width}{Element \ Width} = \frac{18/2}{113} = 0.0796 \ and \ (1 - 0.0796) \ or \ 0.9204.$
- Hinge Loc._{B to A} =  $\frac{half flange width}{Element Width} = \frac{18/2}{40.5} = 0.2222.$
- C. In SAP2000, assign the hinges to corresponding frame elements.
  - Select the desired frame elements you wish to assign hinges to, such as Long1Out. (The elements will turn from blue to yellow).
  - In SAP2000, select the Assign tab > Frame > Hinges

- From the drop-down menu, select Long1Out and set relative distance to 0 and click ADD.
- From the drop-down menu, select Long1Out and set relative distance to 1 and click ADD.
- Click OK.

💢 Assign Frame Hinges	×
Frame Hinge Assignment Data	
Relative Hinge Property Distance	
LongOUT 1 Add Hinge	
Modify/Show Auto Hinge	
Delete Hinge	
<u>Current Hinge Information</u> Type: User Defined DOF: Moment M3	
Options O Add Specified Hinge Assigns to Existing Hinge Assigns	
Replace Existing Hinge Assigns with Specified Hinge Assigns	
Existing Hinge Assignments on Currently Selected Frame Objects Number of Selected Frame Objects: 32 Total Number of Hinges on All Selected Frame Objects: 0	
Fill Form with Hinges on Selected Frame Object	
OK Close Apply	

• Repeat previous step of assigning the hinges for all other frame elements.



- 11. Assign loads to the frame elements.
  - A. Wheel Axle Loads.
    - Axle loads will be placed at distances of 36 in., 108 in., 180 in., and 252 in. from the outside of the curved edge.
    - One line of the 16 kip axles will be placed at the 0.4L point of the bridge, or Transverse Grid 9, with the second line 14 ft away at Transverse Grid 7. One line of 4 kip axles will be placed 14 ft away from the first line of axles, at Gridline 11.
    - In SAP2000, the loads have to be placed at a relative distance, so this value needs to be calculated.

$$\begin{array}{l} \circ \quad HS20_{Axel\ 1\ Loc\ (B-A)} = \frac{L_1 - 36}{L_1} = \frac{40.5 - 36}{40.5} = 0.1111. \\ \circ \quad HS20_{Axel\ 2\ Loc\ (C-B)} = \frac{L_1 + L_2 - 108}{L_2} = \frac{40.5 + 83 - 108}{83} = 0.1867. \\ \circ \quad HS20_{Axel\ 3\ Loc\ (D-C)} = \frac{L_1 + L_2 + L_3 - 180}{L_3} = \frac{40.5 + 83 + 113 - 180}{113} = 0.5. \\ \circ \quad HS20_{Axel\ 4\ Loc\ (E-D)} = \frac{L_1 + L_2 + L_3 + L_4 - 252}{L_4} = \frac{40.5 + 83 + 113 + 83 - 252}{83} = 0.8133. \end{array}$$

- B. Lane Loads
  - Lane Loads are lane loads of 0.640 kip/ft (0.05333 kip/in.) centered at a distance of 96 in. and 240 in. from the outside of the curved edge.
  - These lane loads will be placed on the longitudinal frame elements. They will be assigned to elements along the B, C, D, and E longitudinal elements according to the appropriate tributary distance.
    - $\circ \quad LaneLoad_B = \left(\frac{L_1 + L_2 96}{L_2}\right) * laneload = \left(\frac{40.5 + 83 96}{83}\right) * \\ 0.05333 = 0.017671 \frac{kip}{in}.$
    - $LaneLoad_{c} = laneload laneload_{B} = 0.05333 0.017671 = 0.035663 \frac{kip}{in}$ .
    - $LaneLoad_D = \left(\frac{L_1 + L_2 + L_3 + L_4 240}{L_4}\right) * laneload = \left(\frac{40.5 + 83 + 113 + 83 240}{83}\right) * 0.05333 = 0.051084 \frac{kip}{in}.$
    - $\circ \quad LaneLoad_E = laneload laneload_D = 0.05333 0.051084 = 0.002249 \frac{kip}{in}.$
- C. In SAP2000, the load patterns must first be defined.
  - Select the Define tab > Load Patterns.
    - Under the Load Pattern Name, enter HS20_1 and change the type in the drop-down menu to Live. The self-weight multiplier should be set to 0. Then click Add New Load Pattern.
    - Under the Load Pattern Name, enter LaneLoad1 and change the type in the drop-down menu to Live. The self-weight multiplier should be set to 0. Then click Add New Load Pattern.
    - o Click OK.

oad Patterns				Click To:
Load Pattern Name	Туре	Self Weight Multiplier	Auto Lateral Load Pattern	Add New Load Pattern
DEAD	Dead	~ 1	~	Modify Load Pattern
DEAD LaneLoad1	Dead Live	1 0		Modify Lateral Load Pattern
HS20_1	Live	0		Delete Load Pattern
				Show Load Pattern Notes
				OK

- Assign the HS-20 wheel loads in SAP2000.
  - Select the exterior transverse element of Gridlines 9 and 7.
  - Click Assign > Frame Loads > Point.
  - From the Load Pattern drop-down menu, select HS20 and verify that the coordinate system is set to Global, the Load Direction is Gravity, and the Load Type is Force.
  - In Column 1, enter a relative distance of 0 (Axle 1, Loc. B-A) and load of 16 kips.
  - o Click OK.
  - Repeat for Gridline 11 to assign the 4 kip load.
  - Repeat Steps 1–6 for Axle 2, 3, and 4, Loc. C-B, D-C, and E-D.



- Assign the Lane Load in SAP2000.
  - Select all exterior longitudinal frame elements along Gridline B.
  - Click Assign > Frame Loads > Distributed.
  - From the Load Pattern drop-down menu, select Lane Load and verify that the coordinate system is set to Global, the Load Direction is Gravity, and the Load Type is Force.
  - In the Uniform Load box, enter 0.016124 (Lane Load B).
  - o Click OK.

• Repeat Steps 1–5 for all of the longitudinal elements along gridlines C, D, and E.



## 12. Define load cases.

- A. The load case being used to determine redundancy is 1.25DL + 1.75(LL + IM), where DL = Dead Load, LL = Live Load, and IM = Impact Load. When substituting in the truck load and the lane load, the preceding equation reduces to 1.25DL + 1.75LaneLoad + 2.33HS20.
- B. Generate Load Cases in SAP2000.
  - Click Define > Load Cases > Add New Load Case.
  - In the Load Case Name Panel, name the load case "LC1_1."
  - In the Analysis Type, select "Nonlinear."
  - For the LC1_1 Load Case in the Stiffness to Use panel, select "Zero Initial Conditions."
  - In the Loads Applied panel, leave the Load Type "Load Pattern" in the dropdown Load Name menu, select DEAD and change the Scale Factor to 1.25. Click Add. Change the Load Name menu to select HS20 and change the Scale Factor to 2.33. Click ADD. Change the Load Name menu to select Lane Load and change the Scale Factor to 1.75. Click ADD.
  - In the Other Parameters panel, in the Results Saved section, click Modify/Show.
  - In the Results Saved for Nonlinear Static Load Cases window, change the Results Saved to Multiple States, and in the For Each Stage panel, change the Minimum Number of Saved Steps and the Maximum Number of Saved Steps to 20. Click OK.
  - Click OK on the Load Case Data window.
  - Repeat the previous 8 steps listed under B to create an LC2_1, LC3_1, and LC4_1. However, in the Initial Conditions window, select Continue from State at End of Nonlinear Case, and from the drop-down menu select the preceding load case. (For LC2_1, the Nonlinear Case LC1_1 would be selected).

oad Case Name			Notes	Load Case Type	
LC1_1	5	Set Def Name	Modify/Show	Static V Design	n
nitial Conditions				Analysis Type	
Zero Initial Condition	ons - Start from Unstres	sed State		O Linear	
O Continue from State	e at End of Nonlinear Ca	se	~	Nonlinear	
Important Note:	Loads from this previou	s case are included	in the current case	O Nonlinear Staged Construction	
Iodal Load Case				Geometric Nonlinearity Parameters	
All Modal Loads Appli	ed Use Modes from Cas	e	MODAL $\checkmark$	None	
oade Applied				O P-Delta	
Load Type	Load Name	Scale Fac	tor	P-Delta plus Large Displacements	
Load Pattern V	DEAD	~ 1.25		Mass Source	
Load Pattern	DEAD	1.25	Add	Previous	$\sim$
Load Pattern	HS20_1	2.33	Aug		_
Load Pattern	LaneLoad1	1.75	Modify		
			mouny		
			Delete		
Other Parameters					
Load Application	Full Los	ad	Modify/Show	ок	
Results Saved	Multiple St	ates	Modify/Show	Cancel	
	Defeu				

## 13. Define end supports.

- A. The elastomeric bearing pads for each girder have a lateral stiffness of 12 kip/in. and a vertical stiffness of 6100 kip/in. Since the tub girders are divided in half, the lateral stiffness will be 6 kip/in., and the vertical stiffness will be 3050 kip/in.
- B. Assign spring supports in SAP2000.
  - Select the 12 nodes, eight at the very end of the longitudinal members, four at the location of the pier.
  - Click Assign > Joint > Springs.
  - In the Assign Joint Springs window in the Simple Springs Stiffness panel, enter 6 for Translation 1 and 2 and 3050 for Translation 3.
  - Click OK.



14. Define CL data acquisition points mid-span of girder.

- A. Select the transverse frame elements between B and C and between D and E at 0.4L (Gridline 9).
- B. Click Edit > Edit Lines > Divide Frames.
- C. In the Divide into Specified Number of Frames window, enter 2 for Number of Frames.
- D. Click OK.



- 15. Analyze the nonfracture structure for dead load only.
  - A. In SAP2000, click Analyze > Run Analysis
  - B. In the Set Load Cases to Run window, click the Run/Do Not Run All button until every action is Do Not Run.
  - C. Select DEAD then click Run/Do Not Run Case until the action is run.
  - D. Click Run Now. Let SAP2000 Run the Load Cases until the screen says the analysis is complete.
  - E. Once the analysis is complete, select Spring Reactions.
  - F. Click Display > Show Tables.
  - G. In the Choose Table for Display window, click the + symbol beside Joint Output and select the square box beside Reactions.
  - H. In the Output Options window in the Nonlinear Static Results panel, select Last Step.
  - I. Select and copy the information from the F3 column.
  - J. The sum of the F3 values is the dead load.
  - K. Click Done.
  - L. Unlock the structure.
- 16. Analyze the nonfractured structure.
  - A. In SAP2000, click Analyze > Run Analysis.
  - B. In the Set Load Cases to Run window, click the run/Do Not Run All button until every action is Do Not Run.
  - C. Select LC1_1, LC2_1, LC3_1, and LC4_1 then click Run/Do Not Run Case until the action for all four is run.
  - D. Click Run Now.

- E. Let SAP2000 Run the Load Cases until the screen says the analysis is complete.
- F. Once the analysis is complete, select the data collection point on the transverse member on the outside girder (C-B).
- G. Click Display > Show Tables.
- H. In the Choose Table for Display window, click the + symbol beside Joint Output and select the square box beside Displacements.

choose rubles for proping	
dit	
MODEL DEFINITION (0 of 70 tables selected)	Load Patterns (Model Def.)
⊕. □ System Data	Select Load Patterns
	3 of 3 Selected
⊡ Load Pattern Definitions	Lond Cores (Denutto)
Other Definitions	Load Cases (Results)
er Connectivity Data	Select Load Cases
	4 of 4 Selected
Options/Preferences Data	Modify/Show Options
	Set Output Selections
NALYSIS RESULTS (1 of 10 tables selected)	Outiens
⊡ M Joint Output	Options
	Selection Only
	Show Unformatted
⊡ Element Output	
Structure Output	
	Named Sets
	Save Named Set
	Show Named Set
	Delete Named Set
	OK Cancel

- I. Click the Modify/Show Options button.
- J. In the Output Options window in the Nonlinear Static Results panel, select Stepby-Step.

lode Shapes	Base Reactio	ns Location	Buckling Modes			
Mode to	Global X	0	Mode to			
All Modes	Global Y Global Z	0	in in	All Modes		
Iodal History Results	Nonlinear Sta	tic Results		Steady State Results		
Envelopes	O Envelop	es		Envelopes		
O Step-by-Step	Step-by-Step			<ul> <li>At Frequencies</li> </ul>		
<ul> <li>Last Step</li> </ul>	O Last Step			In and Out of Phase $\sim$		
Direct History Results	Multi-step Sta	itic Results		Power Spectral Density Results		
Envelopes	Envelopes			RMS		
O Step-by-Step	O Step-by-Step			🔘 sqrt(PSD)		
🔘 Last Step	🔘 Last Ste	эр				
.oad Combos	Bridge Design	n Results				
Envelopes	O Controlli	ing Combo				
O Correspondence	O All Com	bos				
Multiple values if possible				OK Cancel		

- K. Click OK.
- L. Select and copy the information from the Output Case, StepNum Unitless, and the U3 in. column and paste them into an Excel worksheet. These columns represent Load Case, Step Number, and Deflection for the Outside Girder, respectively.

🔀 Joint Displacements									-		Х		
File	View	Edit	Format-Filter	-Sort Select	Options								
Units: Filter:	Units: As Noted Filter: Joint Displacements											~	
	Joint Text		OutputCase	Cube Type Text	StepType Text	StopNum Unitless	U1 in	U2 in	os in	R1 Radians	R2 Radians	R3 Radians	>
•	103	3	LC1	NonStatic	Step	0	0	0	0	0	0		0
	103	3	LC1	NonStatic	Step	1	0.002545	7.7E-05	-0.087926	-2.681E-06	0.000169	5.324E-0	8
	103	3	LC1	NonStatic	Step	2	0.00509	0.000154	-0.175851	-5.363E-06	0.000338	1.065E-0	7

M. Click Done.

- N. Select the data collection point on the transverse member on the inside girder (E-D) and repeat Steps G–L. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- O. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element B) and repeat Steps G–L. This information goes into the Delta 4 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- P. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element C) and repeat Steps G–L. This information goes into the Delta 3 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- Q. Select the joint on the transverse member on the inside girder at Transverse Element 9 (at Longitudinal Element D) and repeat Steps G–L. This information goes into the Delta 2 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- R. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element E) and repeat Steps G–L. This information goes into the Delta 1 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.

4	A	8	С	D	E	F	G	н	1.1	J	K	L	м	V	х	Y	Z	AA	AB	AC	AD
1	Intact			0	OG-CL	IG	CL	Point 4	Point 3	Point 2	Point 1			Applied		0	G	IC	5		
	Lond Core	Load Stop	•	Dalta (In)	Chard (red)	Dalta (In)	Chard (rad)	Dolta (In)	Dalta (In)	Delta (In)	Delta (In)	a 22 (rad)	a 22 (md)	Load	O (ml)	Delta (In)	Chord	Dolta (In)	Chord	α 23	α 32
2	Load Case	Load Step	u	Deita (in)	Chord (rad)	Deita (in)	Chord (rad)	Deita (in)	Deita (in)	Delta (in)	Delta (in)	α 23 (rad)	α 32 (rad)	(kip)	LI (cal)	Deita (in)	(deg)	Deita (in)	(deg)	(deg)	(deg)
3	LC1	0	0	0	0.00000	0	0.00000	0	0	0	0	0	0	0	0.00	0.00	0.00000	0.00	0.00000	0	0
-4	LC1	1	0.05	-0.1037	0.00015	-0.087926	0.00013	-0.10541	-0.10129	-0.0942	-0.07905	7.89E-05	4.94E-05	66	0.05	0.10	0.00861	0.09	0.00730	0.004522	0.002831
5	LC1	2	0.1	-0.2074	0.00030	-0.175851	0.00025	-0.21081	-0.20259	-0.18839	-0.15809	0.000158	9.88E-05	132	0.10	0.21	0.01722	0.18	0.01460	0.009043	0.005662

- S. Once all the data are collected, unlock the model by selecting the Lock tool on the left hand side of the SAP2000 screen.
- 17. Analyze the fractured structure.
  - A. At mid-span along Gridline 9, replace the hinges of the outside longitudinal element (Gridline B) with FracOUT hinges according to Step 10.
  - B. At mid-span along Gridline 9, replace the hinges of the first interior longitudinal element (Gridline C) with FracInt hinges according to Step 10.



C. Repeat Step 15 for the fractured case and collect the data accordingly. 18. Post-process the data.

A. In the Excel Sheet, the following values need to be calculated for each step.

• Omega ( $\Omega$ ):

$$\varOmega_{i} = \varOmega_{i-1} + \left(\frac{1}{\# \ of \ Steps \ in \ Load \ case}\right)$$

• Longitudinal Chord Rotation of Interior and Exterior Girder:

Chord Rot._{Single Span} = 
$$-1 * \left(\frac{\delta_{CL}}{0.5 * L}\right)$$
 (rad).

• Transverse Deck Rotation:

$$\alpha_{2-3} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_2 - \delta_1}{w}\right) \text{ (rad).}$$
  
 
$$\alpha_{2-3} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_2 - \delta_1}{w}\right) \text{ (rad).}$$

- Where s = spacing between the interior top flanges of the inside and outside girders and w = spacing between the top flanges of the same girder.
- Applied Load:
  - ο Calculate unit applied load or applied load at 1 Ω.
  - Unit Applied Load_{single span} = 1.25 * Total Reactions from Dead Load Case + 2 * (2.33 * HS20 truck + 1.75 * Lane Load).
  - $\circ \quad Applied \ Load = Unit \ Applied \ Load * \Omega.$
- B. Repeat Step A for the fractured case.
- C. Calculate the initial stiffness for the intact bridge and instantaneous stiffness for the fractured bridge.
  - For the nonfractured condition (intact bridge), find the absolute displacement for the outside girder at an  $\Omega$  value of 0.4:

Initial Stiffness = 
$$\frac{0.4}{Absolute \ Displacement \ OG \ (at \ \Omega = 0.4)}$$
.

• For the Fractured case, add an additional column labeled stiffness:

Instantaneous Stiffness_{OG-Frac.i} = 
$$\frac{\Omega_i - \Omega_{i-1}}{\delta_i - \delta_{i-1}}$$
.

- D. Failure of the structure occurs at the  $\Omega$  of the fractured bridge at the first of the following criteria:
  - The instantaneous stiffness for the fractured outside girder is less than 5 percent of the initial stiffness of the intact outside girder.
  - The chord angle of the outside girder for a simple spans or interior spans is greater than 2 degrees. The chord angle for exterior spans of multi-span bridges is greater than 3 degrees.
  - The transverse deck rotation is greater than 5 degrees.
- E. On a chart, plot the nonfractured outside and inside girder as well as the fractured outside and inside girder with displacement on the primary x axis and the total force on the primary y axis and  $\Omega$  on the secondary y axis.
- F. Repeat Steps 11–18 for Span 2, if a different length. For Bridge 5, both Span 1 and Span 2 are 140 ft long, so no repletion is needed.



### 6.3 OVERALL RESULTS FOR BRIDGE 5

Table 6.1 and Figure 6.2 depicts the overall results for Bridge 5, comparing all the three methods using the upper-bound and lower-bound solution range of the yield line theory and the grillage overstrength (load) versus deflection plots. FEM data are also shown in Table 6.1 and in Figure 6.2 to facilitate a thorough comparison. All three methods suggest that Bridge 5 could be reclassified as nonfracture critical.

Method of Analysis	<b>Overstrength Factors</b>	Span 1 (140 ft)	Span 2 (139.60 ft)
Yield Line Theories	$arOmega_{Yield\ Line}^{Upper\ Bound}$	1.40	1.39
Tield Enic Theories	$\Omega^{Lower \; Bound}_{Yield \; Line}$	1.28	1.28
Grillage Analysis	$arOmega_{Grillage}$	1.10	1.10
FEM	$arOmega_{FEM}$	1.20	1.20

 Table 6.1. Results Summary of Bridge 5.



Figure 6.2. Comparison of the Results for Bridge 5, Span 1 and 2,  $L_x = 140$  ft.

#### 7. **BRIDGE 10**

### 7.1 YIELD LINE ANALYSIS EXAMPLE OF BRIDGE 10

This section documents the steps to compute the overstrength factor of Bridge 10 using the yield line analysis (grillage analysis is explained in the next sub-section). Figure 7.1 presents the dimensional details of the three spans of Bridge 10. Further details may be found in Appendix C.



(*a*) Bridge 10, Span 1 ( $L_x = 148$  ft)



(*b*) Bridge 10, Span 2 ( $L_x = 265$  ft)



(c) Bridge 10, Span 3 ( $L_x$  = 189.58 ft)

#### Figure 7.1. Schematic Diagrams of Bridge 10 ( $R_{\rm fL}$ = 716 ft).

This section presents the stepwise procedure of the yield line analysis conducted to establish the upper-bound and lower-bound solution range for the overstrength factor using the procedure of Section 3. The moment capacities are computed using the methods explained in Section 5 and 6. It should be noted that this bridge consists of two typical exterior spans with one support fixity and one interior span with fixity at both the supports.

		MOMENT CAPACITY OF DECK SLAB						
		(	BRIDGE 10)				14	
		Description	Parameter	Data				
, Z	1	Total Width	<i>B</i> in	360				
ECK CTIC	2	Thíckness	tin	8				
A D	3	Section width	<i>b</i> in	12				
LETE	4	Characterístíc Compressíve Strength	$f_c'$ ksi	4				
NCR	5		$oldsymbol{eta}_{{\scriptscriptstyle \mathcal{I}}}$	0.85				
9 X	6	Strain	E _{cu}	0.003				
AL	F	Yield Strength	f _y ksi	60				
LERA	8	Young's Modulus	<i>E</i> ksi	29000				
μ Ž	9	Straín	$\varepsilon_y$	0.002070				
		Moment Computat	íons	Longíti	ıdinal	Trai	nsverse	
	10			$m'_x$	$m_x$	$m'_y$	$m_y$	
	11			тор	Bottom	Тор	Bottom	
	12	Bar No.		4	5	5	5	
	13	Díameter of Bar	$d_b$ in	0.5	0.625	0.625	0.625	
FTAIL	14	Area of Bar	$A\phi \operatorname{in}^2$	0.2	0.31	0.31	0.31	
R DI	15	Spacing	<i>s</i> in	-	-	6	6	
L BA	16	No. of Bars	#bar	41	33	-	-	
ÇIN	17	Area of Steel	$A_s in^2$	0.273	0.341	0.620	0.620	
IFOR	18	Clear Cover	<i>cc</i> in	2	1.25	2	1.25	
SEIN	19	Effective depth (tension)	<i>d</i> in	5.125	5.8125	5.6875	6.4375	
	20	Effective depth (comp.)	<i>d</i> ′in	2.1875	2.875	1.5625	2.3125	
	21	Depth of NA	<i>c</i> in	1.06	1.06	1.34	1.67	

MOMENT CAPACITY OF DECK SLAB							2
			(BRIDGI	E 10)			14
		Moment Computatí	ons	Longi	tudínal	Tran	sverse
	22			$m'_x$	$m_x$	$m'_y$	$m_y$
	23			тор	Bottom	Тор	Bottom
Ϋ́Ν	24		$\varepsilon_{top}$	0.011466	0.005115	0.009770	0.001153
STR	25		E _{bot}	0.003174	0.013406	0.000508	0.008561
	26	$\beta_{1}c$	<i>a</i> in 0.90 0.90		1.14	1.42	
	27	Compression-Concrete	$C_c$ kip	-36.86	-36.86	-46.34	-57.93
CES	28	Tension/	<i>T</i> top kip	16.40	16.40	37.20	20.73
FOR(	29	Compression-Steel	T bot kip	20.46	20.46	9.14	37.20
	30	Equílíbríum Check	Σ <i>T</i> + <i>C</i> kip	0.00	0.00	0.00	0.00
RESMLTS	31	Moment	<i>M_n</i> k-ft	9.35	12.45	16.63	20.52
		<u>Remarks</u>	<u>.</u>	⊤op Yíelded	⊤op Yíelded	Top Yielded Bottom	Top Not Yielded
				Bottom Yíelded	Bottom Yíelded	Not Yíelded	Bottom Yíelded
			Both steel ín Tensíon	Both steel ín Tensíon	Both steel ín Tensíon	Both steel ín Tensíon	

POSITIVE AND NEGATIVE MOMENT OF INTACT SECTION						3
		(BRIDGE 1	0, SPAN 1)			14
	T			M	oment	
		Description	Posít	íve	Negatív	e
N N N N	1	Total width	<i>B</i> in	180	Bin	180
ECT I	2	Thíckness	<i>t</i> in	8	<i>t</i> in	8
ч В С	3	Haunch thickness	<i>h</i> in	5	<i>h</i> in	5
eete Rial	4	Characterístíc Compressíve Strength	<i>f</i> _c ′ksi	4		
NCH LEI	5		$\beta_1$	0.85		
Ne CO	6	Strain	E _{cu}	0.003		
	F	Yield Strength	<i>fy</i> ksi	60	<i>fy</i> ksi	60
	8	Straín	Rebar $\varepsilon_y$	0.002070		
	9	No of Bars at Top	#top bars	20.5	#top bars	20.5
	10	No of Bars at Top Bent			#top bars bent	20
	11	No of Bars at Bottom	#bot bars	16.5	#bot bars	16.5
പി	12	Clear Cover Top	<i>CC_{top}</i> in	2	<i>CCtop</i> in	2
ETA	13	Clear Cover Bottom	<i>CC_{bot}</i> in	1.25	<i>CC_{bot}</i> in	1.25
LR D	14	Transverse Top Díameter	<i>d_{b top tra}</i> in	0.625	$d_{b top tra}$ in	0.625
5 B	15	Transverse Bottom Díameter	<i>d_{b bot tra}</i> in	0.625	<i>d_{b bot tra}</i> in	0.625
CIN	16	Longitudinal Top Diameter	<i>d_{b top long}</i> in	0.5	<i>d_{b top long}</i> in	0.5
FOR	17	Longitudinal Top Diameter at Bents			<i>d</i> b top long 'in	0.625
N IN	18	Longitudinal Bottom Diameter	<i>d</i> b bot long in	0.625	<i>d</i> b bot long in	0.625
Ŕ	19	Effective Depth Top	<i>d</i> _{top} in	2.88	<i>d</i> _{top} in	2.88
	20				<i>d</i> top bent in	2.94
	21	Effective Depth Bottom	<i>d</i> _{bot} in	5.81	<i>d</i> _{bot} in	5.81
	22	Area of Steel for #4 Bars	<i>A</i> #4 in ²	0.196	<i>A</i> #4 in ²	0.196
	23	Area of Steel for #5 Bars	$A_{\#5}  { m in}^2$	0.307	<i>A</i> #5 in ²	0.307
3EL	24	Yield Strength	<i>F</i> ' _y ksi	50	<i>F</i> ′ _y ksi	50
STE	25	Strain	E _s	0.00172		
AL.	26	Young's Modulus	<i>E</i> ksi	29000		
L R	27	$f_y/F'_y$	п	1.2	п	1.2
NCT	28	Effective Area of Steel for #4 Bars	A#4. eff in ²	0.236	$A$ #4. eff $in^2$	0.236
STR	29	Effective Area of Steel for #5 Bars	A#5. eff in ²	0.368	A#5. eff in ²	0.368

POSITIVE AND NEGATIVE MOMENT OF INTACT SECTION								
		(BRIDO	GE 10, SPAN 1	_)		14		
	1			Ma	oment			
		Descríptíon	Posítív	e	Negatíve			
	30	Gírder Depth	D _g in	80	$D_g$ in	83		
S	31	Top Flange Width	<i>b</i> _{tf} in	24	<i>b</i> tf in	24		
TAII	32	Top Flange Thickness	<i>t</i> _{tf} in	1	<i>t</i> _{tf} in	3		
2 DE	33	web width	$b_w$ in	78	$b_w$ in	78		
IRDEF	34	Web Thickness	$t_w$ in	0.625	$t_w$ in	0.9		
С Щ	35	Bottom Flange Width	<i>b</i> _{bf} in	59	<i>b bf</i> in	59		
	36	Bottom Thickness	<i>t</i> _{bf} in	1.25	<i>t</i> _{bf} in	2		
Şţ	37	Compressíon Z.one	<i>c</i> in	30.14	<i>c</i> in	53.65		
114	38	b _w ín Comp.	yin	16.14	<i>y</i> in	51.65		
DÐ	39	b _w in Tension	y'in	61.86	y'in	26.35		
	40		d Cconc in	17.33	<i>d Arebar top</i> in	39.48		
	41		d C _{rebar top} in	27.26	<i>d A_{rebar top bent}</i> in	39.41		
ξS	42	Distances of Compressive Forces	<i>d Crebar bottom</i> in	24.32	d Arebar bot in	36.54		
ANC	43	Neutral Axís/	d C _{tf1} in	16.64	$dA_{tf}$ in	27.85		
NST.	44	Distances of Areas in Tension	$d T_{W 1}$ in	8.07	$dA_{w1}$ in	13.18		
4	45	una compression from PNA	$d T_{W2}$ in	30.93	$dA_{W2}$ in	25.82		
	46		$d T_{bf}$ in	62.49	d Abf in	52.65		
	47		E Crebar top	0.00271				
	48		E Crebar bottom	0.00242				
AIN	49		E Ctf	0.00166				
STR	50		<i>E Cw</i> 1	0.00080				
	51		$\mathcal{E} T_{w2}$	0.00308				
	52		E Tbf	0.00622				

		POSITIVE AND NEGATIVE	MOMENT OF I	NTACT	SECTION	5		
			Moment					
		Description	Posítíve	2	Negatí	ve		
	53		<i>0.85f_c' βbt</i> kip	-4162	<i>As top</i> kip	5		
	54				As top bent kip	7		
Υŝ	55	Compression Fc/	As topfy kip	-273	As bottom kip	6		
/AR	56	Tension Ar	As bottom fy kip	-344	Att kip	144		
DRCE	57		$2b_{tf}t_{tf}F'_y$ kip	-2305	<i>A_{w1}</i> kip	46		
F(	58		<i>2t_wyF</i> ' _y kip	-469.82				
	59	Tension $F_{T/}$	$2t_w y' F'_y$ kip	3866	<i>Aw2</i> kip	-90		
	60	Compression $A_C$	<i>bbftbfF</i> 'y kip	3688	Abf kip	-118		
X	61		Fc kip	-7554	<i>A</i> _T kip	208		
HEC	62	Equilibrium of Forces/Areas	F _T kip	7554	<i>Ac</i> kip	-208		
S	63		$F_C + F_T$ kip	0	<i>At</i> + <i>Ac</i> kip	0		
ENT	64	(Half Section)	$0.5 M_p^+$ k-ft	40008	$0.5 M_p^-$ k-ft	58368		
MOM	65	(Full Section)	$M_p^+$ k-ft	80016	$M_p^-$ k-ft	116736		

	SECTION	6				
		(BRIDGE 10	D, SPAN 3)	)		14
				N	10ment	
		Description	Posít	<i>sive</i>	Negatín	le
N N N N	1	Total width	<i>B</i> in	180	<i>B</i> in	180
SEC T	2	Thickness	<i>t</i> in	8	<i>t</i> in	8
U D	3	Haunch thickness	<i>h</i> in	5	<i>h</i> in	5
2ETE RIAL	4	Characterístíc Compressíve Strength	<i>f</i> [′] ksi	4		
NCH LIEI	5		βι	0.85		
MA	6	Straín	Е _{си}	0.003		
	F	Yield Strength	<i>f_y</i> ksi	60	$f_y$ ksi	60
	8	Strain	Rebar $\varepsilon_y$	0.002070		
	9	No of Bars at Top	#top bars	20.5	#top bars	20.5
	10	No of Bars at Top Bent			#top bars bent	20
Ŋ	11	No of Bars at Bottom	#bot bars	16.5	#bot bars	16.5
LAIL	12	Clear Cover Top	cctop in	2	cctop in	2
DE	13	Clear Cover Bottom	ccbot in	1.25	ccbot in	1.25
AR	14	Transverse Top Díameter	$d_{b\ top\ tra}$ in	0.625	$d_{b\ top\ tra}$ in	0.625
40 13	15	Transverse Bottom Díameter	$d_{b\ bot\ tra}$ in	0.625	$d_{b\ bot\ tra}$ in	0.625
SCIP	16	Longitudinal Top Diameter	db top long in	0.5	$d_{b\ top\ long}  { m in}$	0.5
FOR	17	Longitudinal Top Diameter at Bents			$d_{b\ top\ long}$ ' in	0.625
SEIN	18	Longitudinal Bottom Diameter	$d_b$ bot long ${ m in}$	0.625	$d_{b\ bot\ long}{ m in}$	0.625
-	19	Effective Depth Top	$d_{top}$ in	2.88	$d_{top}$ in	2.88
	20				dtop bent in	2.94
	21	Effective Depth Bottom	$d_{bot}$ in	5.81	$d_{bot}$ in	5.81
	22	Area of Steel for #4 Bars	A#4 in ²	0.196	A#4 in ²	0.196
	23	Area of Steel for #5 Bars	A#5 in ²	0.307	A#5 in ²	0.307
EL	24	Yield Strength	F'y ksi	50	$F'_y$ ksi	50
STE	25	Straín	E _s	0.00172		
AL.	26	Young's Modulus	E ksi	29000		
TUR	27	$f_y/F'_y$	n	1.2	Ν	1.2
RUC	28	Effective Area of Steel for #4 Bars	$A$ #4. eff $\ln^2$	0.236	A#4. eff $in^2$	0.236
S T	29	Effective Area of Steel for #5 Bars	A#5. eff in ²	0.368	<i>A</i> #5. <i>eff</i> in ²	0.368

		POSITIVE AND NEGATIVE MOMENT OF INTACT SECTION						
		(BRID	GE 10, SPAN :	3)		14		
				M	oment			
		Description	Positivo	2	Negatív	e		
	30	Gírder Depth	$D_g$ in	79.75	$D_g$ in	83.5		
SJ1	31	Top Flange Width	<i>b</i> _{tf} in	24	<i>b</i> _{tf} in	24		
ETA	32	Top Flange Thickness	<i>t</i> _{tf} in	1	<i>t</i> _{tf} in	3		
R D	33	web width	$b_w$ in	78	$b_W$ in	78		
E E	34	Web Thickness	$t_w$ in	0.625	$t_w$ in	0.9		
d la	35	Bottom Flange Width	<i>b</i> _{bf} in	59	<i>b</i> _{bf} in	59		
	36	Bottom Thickness	t _{bf} in	0.75	<i>t</i> _{bf} in	2.5		
Sł	37	Compressíon Zone	<i>c</i> in	22.18	<i>c</i> in	45.72		
ертн	38	b _w in Comp.	<i>y</i> in	8.18	<i>y</i> in	43.22		
Ð	39	$b_w$ in Tension	y'in	69.82	y'in	34.78		
	40		d C _{conc} in	12.75	<i>d A_{rebar top}</i> in	47.91		
	41		<i>d Crebar top</i> in	19.30	d Arebar top bent in	47.84		
SE(S)	42	Distances of Compressive Earces and Tensile Earces	d Crebar bottom in	16.36	<i>d A_{rebar bot}</i> in	44.97		
<b></b> ₹NC	43	from Neutral Axís/	$d C_{tf}$ in	8.68	$dA_{tf}$ in	36.28		
IST,	44	Distances of Areas in Tension	$d T_{W 1}$ in	4.09	$dA_{w1}$ in	17.39		
Д	45	and Compression from PNA	$d T_{w 2}$ in	34.91	$dA_{w2}$ in	21.61		
	46		$d T_{bf}$ in	70.20	$dA_{bf}$ in	44.47		
	4 <del>7</del>		E Crebar top	0.00261				
	48		E Crebar bottom	0.00221				
AIN	49		E Ctf	0.00117				
STR	50		<i>E C</i> _{<i>W</i> 1}	0.00055				
vj	51		$\mathcal{E} T_{W2}$	0.00472				
	52		E Tbf	0.00950				

		POSITIVE AND NEGATIV	E MOMENT O	FINTA(	SECTION	8
				., М	oment	<u> </u>
		Descríptíon	Posítívo	2	Negi	atíve
	53		<i>0.85f_c' βbt</i> kip	-4162	As top kip	5
	54				As top bent kip	7
4	55	COMPRESSION Fol	<i>As topfy</i> kip	-273	As bottom kip	6
ARE	56	Tension $A_T$	As bottom fy kip	-344	Att kip	144
SCE/1	57	2	$2b_{tf}t_{tf}F'_y$ kip	-1634	<i>A_{w1}</i> kip	61
FOF	58		$2t_w y F'_y$ kip	-163.91		
	59	Tension Fr/	$2t_w y' F'_y$ kip	4364	<i>A_{w2}</i> kip	-76
	60	Compression $A_C$	<i>bbftbfF</i> 'y kip	2213	Abf kip	-148
2	61		Fc kip	-6577	<i>A</i> _T kip	223
HECH	62	Equílíbríum of Forces/Areas	F _T kip	6577	<i>Ac</i> kip	-223
S	63		$F_C + F_T$ kip	0	$A_T + A_C$ kip	0
ENT	64	(Half Section)	$0.5 M_p^+$ k-ft	32207	$0.5 M_p^-$ k-ft	64603
IMOM	65	(Full Section)	$M_p^+$ k-ft	64414	$M_p^-$ k-ft	129206

		C	OMPUTATION OF OVERSTRENGTH FACTOR	R	9
		EX	TERIOR: 3-SPAN CASE (BRIDGE 10, SPAN	1)	14
		Parameter	Formula/Definition/Equation		Data
	1	$L_x$	Span Length	148.00	ft
	2	$R_{CL}$	Radíus of center líne	716	ft
R	3	В	Width	30	ft
DMET	4	$L_x^*$	Outer region length $L_x^* = \left(1 + \frac{B}{4R_{C_L}}\right)L_x$	149.55	ft
d El	5	S	Inter Gírder Spacing	7.7	ft
-	6	b	Width of Girder + Edge	11.2	ft
	F	t	Deck Thickness	8.00	in
	8	m _x	Longítudinal Posítive Moment per A	12	k-in/in
	9	<i>m′</i> _x	Longítudínal Negatíve Moment per A	9	k-in/in
	10	$m_y$	Transverse Positive Moment per A	21	k-in/in
	11	<i>m′y</i>	Transverse Negative Moment per ft	17	k-in/in
	12	0.5Mp ⁻	Negative Moment at Support	58368	k-ft
₫M]	13	λ	Fraction of Length from the exterior support at which girder is fractured	0.40	
K DONE,	14	tan θ	$\tan \theta = \sqrt{\frac{m_y' + m_y}{m_x' + m_x}}$	1.31	(Θ = 52.5°)
WOR	15	tan α	$\tan \alpha = \frac{2s}{L_x}$	0.10	(Θ = 5.9°)
ERNAI	16	$k_{bound}^{upper}$	$\left[1+2\frac{\tan\alpha}{\tan\theta}\right] = 1 + \frac{4s}{L_x}\sqrt{\frac{m'_x + m_x}{m'_y + m_y}}$	1.16	
IN I	17	$k_{bound}^{lower}$	$\left[1 + 2\frac{\tan^{2}\alpha}{\tan^{2}\theta}\right] = 1 + \frac{8s^{2}}{L_{x}^{2}}\sqrt{\frac{m'_{x} + m_{x}}{m'_{y} + m_{y}}}$	1.01	
	18	IWDupper	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{upper}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}+\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	1070	k-ft *
	19	IWDiower	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{lower}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}+\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	1018	k-ft *

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of internal work is in k-ft

		COMP	COMPUTATION OF OVERSTRENGTH FACTOR							
		EXTER	LIOR: 3-SPAN CASE (BRIDGE 10, SPAN	1)	14					
		Parameter	Formula/Definition/Equation	Þ	pata					
	20	DL	Dead Load Factor	1.25						
	21	LL	Live Load Factor	1.75						
	22	SAF	Stíffener Allowance Factor	1						
	23	γc	Unit weight of reinforced concrete	0.15	kcf					
	24	Wu	Area load due to reinforced concrete + lane load $DL\gamma_c \frac{t}{12} + LL \cdot \frac{0.64}{12}$	0.218	ksf					
	25	γs	Unit weight of steel	0.49	kcf					
EWI	26	$V_g$	Volume of Girder	252	ft ³					
NE	27	Ar	Area of Rail Cross-Section (T4(S))	1.25	ft ²					
DQ	28	Vr	Volume of $Rail = L_x A_r$	187	ft ³					
NORK	29	W _x	$1.25\left(1.15V_g\gamma_s+V_r\gamma_c\right)/L_x$	1.42	k/ft					
ERNAL (	30	<i>y</i> (Lane 2)	(b + s - 15) for $(b + s) < 21(b + s - 18)$ for $(b + s) > 21$	3.83	ft					
EXT	31	Klane	$1 + 0.5 \frac{y}{s}$ for $(b+s) < 21; 1 + \frac{y}{s}$ for $(b+s) > 21$	1.25						
	32	EWD _{HS-20}	$\left(168 - \frac{261.33}{\lambda L_{x}} - \frac{1045.33}{(1-\lambda)L_{x}}\right) K_{Lane}$	190	k-ft					
	33	WET	$w_u L_x(b+0.5s) + W_x L_x + 2EWD_{HS20}$	1082	k-ft					
	34	EWD	0.5 Wet	541	k-ft *					
NLTS	35	$arOmega_{upper}$	IWD _{upper} /EWD	1.98						
RESI	36	$\Omega_{lower}$	IWD _{lower} /EWD	1.88						

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of external work is in k-ft

		COMPUTATION OF OVERSTRENGTH FACTOR				
		INTERIOR: 3-SPAN CASE (BRIDGE 10, SPAN 2)			14	
	T	Parameters	Formula/Definition/Equation	Data		
	1	L _x	Span Length	265.00	ft	
~	2	<i>R</i> _{CL}	Radíus of center líne	716	ft	
Ŕ	3	В	Width	30	ft	
OME	4	$L_x^*$	Outer region length $L_{\chi}^{*}=\left(1+rac{B}{4R_{C_{L}}} ight)L_{\chi}$	267.78	ft	
d E	5	S	Inter Gírder Spacing	7.7	ft	
	6	b	Width of Girder + Edge	11	ft	
	F	t	deck thíckness	8	in	
	8	$m_x$	Longitudinal Positive Moment per A	12	k-in/in	
	9	m' _x	Longítudínal Negatíve Moment per ft	9	k-in/in	
	10	my	Transverse Positive Moment per ft	21	k-in/in	
	11	$m'_y$	Transverse Negative Moment per ft	17	k-in/in	
	12	$0.5M_{p1}$	Negative Moment at Support 1	58368	k-ft	
1	13	$0.5 M_{p2}$	Negative Moment at Support 2	64603		
C DONE, I	14	tan θ	$\tan \theta = \sqrt{\frac{m'_y + m_y}{m'_x + m_x}}$	1.31	(Θ = 53°)	
WORK	15	tan α	$\tan \alpha = \frac{2s}{L_x}$	0.06	$(\Theta = 3.3^{\circ})$	
FRNAL	16	$k_{bound}^{upper}$	$\left[1+2\frac{\tan\alpha}{\tan\theta}\right] = 1 + \frac{4s}{L_x}\sqrt{\frac{m'_x + m_x}{m'_y + m_y}}$	1.09		
PLN	17	$k_{bound}^{lower}$	$\left[1 + 2\frac{\tan^{2}\alpha}{\tan^{2}\theta}\right] = 1 + \frac{8s^{2}}{L_{x}^{2}}\sqrt{\frac{m'_{x} + m_{x}}{m'_{y} + m_{y}}}$	1.00		
	18	IWDupper	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{upper}+\frac{4m_{x}b}{L_{x}}+\left(0.5M_{p1}^{-}+0.5M_{p2}^{-}\right)\frac{2}{L_{x}}$	1620	k-ft *	
	19	IWD _{lower}	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{lower}+\frac{4m_{x}b}{L_{x}}+\left(0.5M_{p1}^{-}+0.5M_{p2}^{-}\right)\frac{2}{L_{x}}$	1566	k-ft *	

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of internal work is in k-ft

		COMPUTATION OF OVERSTRENGTH FACTOR			12
		INTER	IOR: 3-SPAN CASE (BRIDGE 10, SPAN	2)	14
		Parameter	Formula/Definition/Equation	Þ	ata
	20	DL	Dead Load Factor	1.25	
	21	LL	Líve Load Factor	1.75	
	22	SAF	Stíffener Allowance Factor	1.15	
	23	γc	Unit weight of reinforced concrete	0.15	kcf
	24	Wu	Area load due to reinforced concrete + lane load $DL\gamma_c \frac{t}{12} + LL \cdot \frac{0.64}{12}$	0.22	ksf
Ą	25	γs	Unit weight of steel	0.49	kcf
Ц Ц Д	26	$V_g$	Volume of Girder	519	ft ³
DNE	27	Ar	Area of Rail Cross-Section (T4(S))	1.25	ft ²
Z Z Z Z	28	Vr	Volume of $Rail = L_x A_r$	335	ft ³
VOR	29	W _x	$1.25\left(1.15V_g\gamma_s+V_r\gamma_c\right)/L_x$	1.60	k/ft
RNAL V	30	<i>y</i> (Lane 2)	(b + s - 15) for $(b + s) < 21(b + s - 18)$ for $(b + s) > 21$	3.83	ft
EXTEI	31	Klane	$1 + 0.5 \frac{y}{s}$ for $(b+s) < 21$ ; $1 + \frac{y}{s}$ for $(b+s) > 21$	1.25	
	32	EWDHS-20	$\left(168 - \frac{2613}{L_{\chi}}\right) K_{Lane}$	198	k-ft
	33	W _{ET}	$w_u L_x(b+0.5s) + W_x L_x + 2EWD_{HS20}$	1701	k-ft
	34	EWD	0.5 Wet	851	k-ft *
NLTS	35	$\Omega_{upper}$	IWDupper/EWD	1.90	
RES	36	$arOmega_{lower}$	IWD _{lower} /EWD	1.84	

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of external work is in k-ft

		COMPUTATION OF OVERSTRENGTH FACTOR				
		EXTER	.)	14		
		Parameters	Formula/Definition/Equation	I	>ata	
	1	L _x	Span Length	189.58	ft	
~	2	R _{CL}	Radíus of center líne	716	ft	
R	3	В	Width	30	ft	
OMET	4	$L_x^*$	Outer region length $L_{\chi}^{*}=\left(1+rac{B}{4R_{C_{L}}} ight)L_{\chi}$	191.57	ft	
E E	5	S	Inter Girder Spacing	7.7	ft	
	6	b	Width of Girder + Edge	11.2	ft	
	F	t	Deck Thickness	8.00	in	
	8	$m_x$	Longitudinal Positive Moment per A	12	k-in/in	
	9	m'x	Longitudinal Negative Moment per ft	9	k-in/in	
	10	$m_y$	Transverse Positive Moment per ft	21	k-in/in	
	11	$m'_y$	Transverse Negative Moment per ft	17	k-in/in	
	12	$M_p^-$	Negative Moment at Support	64603	k-ft	
CTWD (	13	λ	Fraction of Length from the exterior support at which girder is fractured	0.40		
K DONE	14	tan θ	$\tan \theta = \sqrt{\frac{m'_y + m_y}{m'_x + m_x}}$	1.31	(=55.5)	
L WOR	15	tan α	$\tan \alpha = \frac{2s}{L_x}$	0.08	(⊖=4.6°)	
ERNA	16	$k_{bound}^{upper}$	$\left[1+2\frac{\tan\alpha}{\tan\theta}\right] = 1 + \frac{4s}{L_x}\sqrt{\frac{m'_x + m_x}{m'_y + m_y}}$	1.12		
FNI	17	k ^{lower} k ^{bound}	$\left[1 + 2\frac{\tan^{2}\alpha}{\tan^{2}\theta}\right] = 1 + \frac{8s^{2}}{L_{x}^{2}}\sqrt{\frac{m'_{x} + m_{x}}{m'_{y} + m_{y}}}$	1.01		
	18	IWDupper	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{upper}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}+\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	189.58	k-ft*	
	19	IWD _{lower}	$\left(m_{y}^{'}+m_{y}\right)\left(\frac{L_{x}}{2s}\right)k_{bound}^{lower}+\frac{m_{x}b}{L_{x}(\lambda-\lambda^{2})}+\frac{0.5M_{p}^{-}}{(1-\lambda)L_{x}}$	716	k-ft *	

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of internal work is in k-ft

		COMP	COMPUTATION OF OVERSTRENGTH FACTOR		
		EXTER	LIOR: 3-SPAN CASE (BRIDGE 10, SPAN	з)	14
		Parameter	Formula/Definition/Equation	Þ	,ata
	20	DL	Dead Load Factor	1.25	
	21	LL	Líve Load Factor	1.75	
	22	SAF	Stiffener Allowance Factor	1	
	23	γ _c	Unit weight of reinforced concrete	0.15	kcf
	24	Wu	Area load due to reinforced concrete + lane load $DL\gamma_c \frac{t}{12} + LL \cdot \frac{0.64}{12}$	0.218	ksf
Δ	25	γs	Unit weight of steel	0.49	kcf
ΕŃ	26	$V_g$	Volume of Gírder	335	ft ³
NE,	27	Ar	Area of Rail Cross-Section (T4(S))	1.25	ft ²
D D D D D D D D D D	28	Vr	Volume of $Rail = L_x A_r$	239	ft ³
WORK	29	W _x	$1.25\left(1.15V_g\gamma_s+V_r\gamma_c\right)/L_x$	1.46	k/ft
RNALI	30	<i>y</i> (Lane 2)	(b + s - 15) for $(b+s) < 21(b + s - 18)$ for $(b+s) > 21$	3.83	ft
EXTE	31	Klane	$1 + 0.5 \frac{y}{s}$ for $(b+s) < 21$ ; $1 + \frac{y}{s}$ for $(b+s) > 21$	1.25	
	32	EWD _{HS-20}	$\left(168 - \frac{261.33}{\lambda L_{x}} - \frac{1045.33}{(1-\lambda)L_{x}}\right) K_{Lane}$	194	k-ft
	33	W _{ET}	$w_{u}L_{x}(b+0.5s) + W_{x}L_{x} + 2EWD_{HS20}$	1297	k-ft
	34	EWD	0.5 Wet	648	k-ft *
NLTS	35	$\Omega_{upper}$	IWDupper/EWD	1.67	
RESI	36	$arOmega_{lower}$	IWD _{lower} /EWD	1.59	

Note: *: A unit deflection ( $\delta = 1$ ) is considered; therefore, the unit of external work is in k-ft

# 7.2 GRILLAGE ANALYSIS EXAMPLE OF BRIDGE 10

The following steps are explained to conduct the computational implementation of grillage analysis of Bridge 10.

Teettee	Top Flange		Web		Bottom Flange		Sec. Type	
ft	Width in.	Thickness in.	Width in.	Thickness in.	Width in.	Thickness in.	Section	Rebar
0–50	24	1.00	78	0.625	59	0.750	1	Reg.
50–98	24	1.00	78	0.625	59	1.250	2	Pier
98–131	24	2.00	78	0.75	59	2.000	3	Pier
131–181	24	3.00	78	0.875	59	2.000	4	Pier
181-230	24	1.00	78	0.875	59	1.250	5	Pier
230-247	24	1.00	78	0.75	59	1.000	6	Reg.
247-297	24	1.00	78	0.75	59	1.250	7	Reg.
297-330	24	1.00	78	0.75	59	1.000	8	Reg.
330-380	24	1.00	78	0.875	59	1.250	5	Pier
380–396	24	2.00	78	0.875	59	1.250	5	Pier
396–430	24	3.00	78	0.875	59	2.000	9	Pier
430-447	24	3.00	78	0.875	59	2.000	9	Pier
447-464	24	2.00	78	0.75	59	1.250	10	Pier
464-499	24	1.00	78	0.75	59	1.250	10	Pier
499-602	24	1.00	78	0.625	59	0.750	1	Reg.

## 1. Gather bridge geometry and material information.

Location	Parameter	Description/Value
	Location	Harris County,
	Vear Designed/Vear Built	IH 10 1998/2002
Bridge	Design Load	1770/2002
Diluge	Length. ft	602.58
	Spans, ft	148, 265, 189.58
	Radius of Curvature, ft	716.2
	Width, ft	30
Deek	Thickness, in.	8
DECK	Haunch, in.	5
	Rail Type	T4(s)
	# of Bar Longitudinal Top Row (#4)	42
	# of Bar Longitudinal Bottom Row (#5)	32
	# of Bar Longitudinal Top Row (#4) @sup- port	42
Rebar	# of Bar Longitudinal Top Row (#5) @support	40
	# of Bar Longitudinal Bottom Row (#5)	
	@support	32
	Transverse Spacing Top Row (#5), in.	6
	Transverse Spacing Bottom Row (#5), in.	6
Girder	CL of Bridge to CL of Girder (in.)	45
Under	CL of Top Flange to CL of Top Flange (in.)	96

Concre	te (4 ksi)
Stress	Strain
(ksi)	(1/in)
-4	-3.79E-03
-4	-3.56E-03
-4	-2.69E-03
-4	-1.78E-03
-3.8205	-1.40E-03
-2.8718	-8.69E-04
-0.6403	-1.78E-04
0	0
0.378	1.06E-04
0.378	1.16E-03

Rebar	(60 ksi)
Stress	Strain
(ksi)	(1/in)
-87.9	-0.095
-87.9	-0.0944
-86.6	-0.0761
-78	-0.0386
-60.7	-9.80E-03
-60.3	-2.08E-03
0	0
60.3	2.08E-03
60.7	9.80E-03
78	0.0386
86.6	0.0761
87.9	0.0944
87.9	0.095

Steel	( <b>50 ksi</b> )
Stress	Strain
(ksi)	(1/in)
-71.6	-0.1
-71.6	-0.097
-71.6	-0.095
-71.6	-0.0946
-70.3	-0.0764
-62.5	-0.039
-50	-0.0196
-50	-1.72E-03
0	0
50	1.72E-03
50	0.0196
62.5	0.039
70.3	0.0764
71.6	0.0946
71.6	0.095
71.6	0.097
71.6	0.1

3. Create a coordinate system for the half width of the span for each cross-section. An example of the first cross-section is shown below.



4. Create a cylindrical coordinate system for the curved bridge and ensure that the middle transverse divisions are 7 ft because this will aid is applying the truck load, whose axles are separated by 14 ft.

A. # of Segments _{per span} = $\left(\frac{\text{Length } (ft)*12}{84 \text{ in } (7ft)}\right)$ rounded to nearest even number:
• # of Segments _{Span 1} = $\left(\frac{148*12}{84}\right)$ = 21.428, or 18.
• # of Segments _{Span 2} = $\left(\frac{265*12}{84}\right)$ = 37.857, or 36.
• # of Segments _{Span 3} = $\left(\frac{190*12}{84}\right)$ = 27.143, or 24.
B. End Segment Length = $\frac{((Length*12) - (\# of Segments*84))}{2}$ :
• End Segment Length _{Span 1} = $\frac{((148*12)-(18*84))}{2} = 132$ in.
• End Segment Length _{Span 2} = $\frac{((265*12)-(36*84))}{2}$ = 78 in.
• End Segment Length _{Span 3} = $\frac{((190*12)-(24*84))}{2} = 132$ in.
C. Theta = $\frac{Total Length}{Radius}$ :

Theta = ⁶⁰³/_{716.2} = 0.84194 rad., or 48.24 degrees.
 D. Determine the radial offsets using the outside edge, the outside flange, the inner flange, and CL of the bridge.

Offsets (in.)		
Edge	180	
Outside Flange	141	
Inner Flange	45	
CL of Bridge	0	

Rad	ial Spaci	ng (in.)
А	8774.4	CL + Edge
В	8735.4	CL + OF
С	8639.4	CL + IF
Center Line	8594.4	or 450 (ft)
D	8549.4	CL – IF
Е	8453.4	CL – OF
F	8414.4	CL – Edge

- E. The longitudinal spacing along theta is determined by converting the longitudinal segment lengths into degrees.
  - The segments vary in length. The total length is 603 ft, or 7236 in.
  - Radial Spacing  $(rad) = \frac{Long.Spacing}{Radius}$ .
  - Radial Spacing (degree) = Radius Spacing (rad)  $*\frac{180}{\pi}$ .

Long. Spacing (in.)	Radial Spacing (rad.)	Radial Spacing (deg.)
0	0.0000	0.000
132	0.0000	0.000
216	0.0154	1 440
300	0.0349	2 000
384	0.0317	2.560
468	0.0545	3.120
552	0.0642	3.680
636	0.0740	4.240
720	0.0838	4.800
804	0.0935	5.360
888	0.1033	5.920
972	0.1131	6.480
1056	0.1229	7.040
1140	0.1326	7.600
1224	0.1424	8.160
1308	0.1522	8.720
1392	0.1620	9.280
1476	0.1717	9.840
1560	0.1815	10.400
1644	0.1913	10.960
1776	0.2066	11.840
1854	0.2157	12.360
1938	0.2255	12.920
2022	0.2353	13.480
2106	0.2450	14.040
2190	0.2548	14.600
2274	0.2646	15.160
2358	0.2744	15.720
2442	0.2841	16.280
2526	0.2939	16.840
2610	0.3037	17.400
2694	0.3135	17.960
2778	0.3232	18.520
2862	0.3330	19.080
2946	0.3428	19.640
3030	0.3526	20.200
3114	0.3623	20.760
3198	0.3721	21.320

3282	0.3819	21.880
3366	0.3917	22.440
3450	0.4014	23.000
3534	0.4112	23.560
3618	0.4210	24.120
3702	0.4307	24.680
3786	0.4405	25.240
3870	0.4503	25.800
3954	0.4601	26.360
4038	0.4698	26.920
4122	0.4796	27.480
4206	0.4894	28.040
4290	0.4992	28.600
4374	0.5089	29.160
4458	0.5187	29.720
4542	0.5285	30.280
4626	0.5383	30.840
4710	0.5480	31.400
4794	0.5578	31.960
4878	0.5676	32.520
4956	0.5767	33.040
5088	0.5920	33.920
5172	0.6018	34.480
5256	0.6116	35.040
5340	0.6213	35.600
5424	0.6311	36.160
5508	0.6409	36.720
5592	0.6507	37.280
5676	0.6604	37.840
5760	0.6702	38.400
5844	0.6800	38.960
5928	0.6898	39.520
6012	0.6995	40.080
6096	0.7093	40.640
6180	0.7191	41.200
6264	0.7288	41.760
6348	0.7386	42.320
6432	0.7484	42.880
6516	0.7582	43.440
6600	0.7679	44.000
6684	0.7777	44.560

6768	0.7875	45.120
6852	0.7973	45.680
6936	0.8070	46.240
7020	0.8168	46.800
7104	0.8266	47.360
7236	0.8419	48.240

- Int. Transverse Element width = 84 in.
- End Transverse Element =  $132 \left(\frac{84}{2}\right) = 90$  in.
- *Pier Tansverse Element* = 132 + 78 84 = 126 *in.*
- 5. Input the coordinate system into SAP2000.
  - A. Select File > New Model > Blank model (making sure units are in kips and in.).
  - B. Right click on the blank workspace and select Edit Grid Data > Modify/Show System > Quick Start > Cylindrical.
    - In the Number of Gridlines panel, set "Along Z = 1."
    - In the Grid Spacing panel, set "Along Z = 1."
    - Select OK.
    - Delete all R and T Coordinates that were generated.
  - C. Add correct coordinates for R.

All radial coordinates (A, B, C, D, E, and F).

- D. Add correct coordinates for T.
  - All theta coordinates for T (0 to 7236 in.)
  - Click OK.
- E. The grid system is now formed.

#### 💢 Define Grid System Data

ystem Nam	e	GLU	JDAL				QUICK Start
Grid							
Grid ID	Ordinate (in)	Line Type	Visible	Bubble Loc	Grid Colo	r	65
F	8414.4	Primary	Yes	End		Add	
E	8453.4	Primary	Yes	End			
D	8549.4	Primary	Yes	End		Delete	
С	8639.4	Primary	Yes	End			
В	8735.4	Primary	Yes	End			
Α	8774.4	Primary	Yes	End			
							Disalau Osida es
rid							Display Grids as
Grid ID	Angle (deg)	Line Type	Visible	Bubble Loc	Grid Color	^	Ordinates O Spacing
1	0	Primary	Yes	End		Add	
2	0.88	Primary	Yes	End			Hide All Grid Lines
3	1 44	Primary	Yes	End		Delete	Glue to Grid Lines
4	2	Primary	Yes	End			
5	2.56	Primary	Yes	End			Bubble Size 50.
6	3.12	Primary	Yes	End			
-	2.00	n.	N.			<b>~</b>	
rid Data							Reset to Default Colo
Grid ID	Ordinate (i	n) Lin	e Type	Visible	Bubble Loc		
310 10		n) En	стурс	VISIDIC	Euro		Reorder Ordinates
21	U	FI	imary	Tes	End	Add	
						Delete	
						-	

 $\times$ 



- 6. Define material in SAP2000.
  - A. Click Define > Materials -> Add New Material > Material Type (Steel, Concrete, or Rebar) > Standard (User) > OK.
  - B. At the bottom of the window, select the box that states, "Switch to Advanced Properties."
  - C. In the open window, name the material "Concrete," "Steel," or "Rebar" depending on which material is being defined. Then click "Modify/Show Material Properties."
  - D. In the Material Property Data window, click the "Nonlinear Material Data" icon.
  - E. In the Nonlinear Material Data window, select the "Convert to User Defined" icon.
  - F. Input the number of data points for the stress strain behavior (10 for concrete, 13 for rebar, 17 for steel).
  - G. Input the data points for the stress strain behavior.
  - H. Select "OK."
  - I. Repeat this process again for the remaining materials.

	IName		Material Type	;
Concr	rete		Concrete	
lystere	esis Type	Drucker-Prager Pa	arameters	Units
Taked	a v	Friction Angle	0.	Kip, in, F 🗸
		Dilatational Angle	0.	
Stress-	Strain Curve Defir	ition Options		
O Pa	rametric			Convert To User Defined
	an Defined	1		
-		Dete		
Jser St	tress-Strain Curve	Data		
Jser St	tress-Strain Curve	Data ss-Strain Curve		10
Jser St	tress-Strain Curve er of Points in Stre	Data ss-Strain Curve		10
Jser St	tress-Strain Curve er of Points in Stre Strain	Data ss-Strain Curve Stress	Point ID	
Jser St Numbe	tress-Strain Curve er of Points in Stre Strain -3.790E-03	Data ss-Strain Curve Stress -4.	Point ID -E	
Jser St Numbo	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03	Data ss-Strain Curve Stress -4. -4.	Point ID -E	
Jser Si Number 1 2 3	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03	Data ss-Strain Curve Stress -4. -4. -4. -4.	Point ID -E	
Jser Sf Numbo 1 2 3 4	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03	Data ss-Strain Curve Stress -4. -4. -4. -4. -4.	Point ID E C	
Jser Sf Numb 1 2 3 4 5	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03	Data ss-Strain Curve Stress -4. -4. -4. -4. -3.8205	Point ID E C	
Jser St Numbo 1 2 3 4 5 6	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04	Data ss-Strain Curve Stress -4. -4. -4. -4. -3.8205 -2.8718	Point ID E C	
Jser St Numbe 1 2 3 4 5 6 7	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -3.8205 -2.8718 -0.6403	Point ID -E -C	
Jser St Number 1 2 3 4 5 6 7 8	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04 0.	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -3.8205 -2.8718 -0.6403 0.	Point ID E C A	10
Jser St Number 1 2 3 4 5 6 7 8 9	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04 0. 1.060E-04	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -3.8205 -2.8718 -0.6403 0. 0. 0.378	Point ID E C C A B	10
Jser SI Numbo 1 2 3 4 5 6 7 7 8 9 9 10	tress-Strain Curve er of Points in Stre Strain -3.790E-03 -3.560E-03 -2.690E-03 -1.780E-03 -1.400E-03 -8.690E-04 -1.780E-04 0. 1.060E-04 1.160E-03	Data ss-Strain Curve Stress -4. -4. -4. -4. -4. -3.8205 -2.8718 -0.6403 0. 0. 0.378 0.378 0.378	Point ID E C C C C C 	10

- 7. Define frame cross-sections in SAP2000.
  - A. Click Define > Section Properties > Frame Sections > Add New Properties.
  - B. In the Frame Section Properties drop-down box, select Other and click Section Designer. In the SD Section Designer window, name the section B5Long1 and click the Section Designer icon.
  - C. Using the Polygon feature, draw the features of the half width of the bridge from Step 3. These features include one rail, two concrete deck pieces, two concrete haunches, two top flanges, two webs, and two pieces of the bottom flange.
    - To change material types for the polygons, right-click on the polygon and select the desired material type from the material drop-down menu.
    - To change the coordinates of the polygon's nodes, use the Reshaper tool.
  - D. Add in the longitudinal rebar to both concrete deck elements by using the Line Bar from the Draw Reinforcing Shape tool. From the design drawings, it can be determined that there are 10 #4 top bars and seven #5 bottom bars in the outer concrete deck element and 11 #4 top bars and nine #5 bottom bars in the inner concrete element. At the pier support, 10 additional #5 top bars are added to the top of the outer concrete deck and 10 additional #5 top bars are added to the inner concrete deck, with a 2 in. top cover and transverse reinforcement. The top bars

are located at 5.0625 in., with a 1.25-in. bottom cover and transverse reinforcement. The top bars are located at 2.1875.

E. Click Done (below is an example of the 1st cross-section).



- F. Repeat this process for the remaining longitudinal elements.
- G. Repeat the process for the transverse elements.
  - In the SD Section Designer window, select Modifiers and set Mass and Weight to 0 to not double-count the dead weight.
  - The interior transverse members are 84 in. wide (end members are 90 in. wide and pier members are 126 in.), with #5 rebar at 5 in. spacing at 5.6875 in. and 1.5625 in.



- H. To generate an exterior longitudinal member and an interior longitudinal member, make two copies of the B10Long1.1 section. Label one Long1Out and one Long1Int.
  - For the Long1Out, delete every element right of the CL.



• For the Long1Int, delete every element left of the CL.



- Repeat this process for all cross-sections.
- I. To generate a simulated fracture section, make copies of Long1Out and Long1Int.
  - The reason Long1Out and Long1Int are chosen for the fractured section is because the fracture occurs at 0.4*L, or 710 in. On the radial spacing table, Long1 is the section used at 710 in.
  - Name the Long1Out copy FracInt1.1.
    - Delete the bottom flange, web, and top flange of the steel tub.



• Name the Long1Int copy FracInt1.1.

• Delete the bottom flange, web, and top flange of the steel tub.



- 8. Generate plastic hinges for frame elements in SAP2000.
  - A. Define > Section Properties > Frame Sections.
  - B. Select the desired cross-section. Hinges will need to be made for all of the necessary cross-sections.
  - C. Once selected, click Modify/Show Property > Section Designer.
  - D. Once in the section designer, select the Moment Curvature Curve tool.
  - E. In the Moment Curvature Curve window, select Details.
    - Copy the moment curvature data to an Excel file.
    - Select OK.
  - F. In the Moment Curvature Curve window, change the angle (deg) to 180, then select Details.
    - Copy the moment curvature data to the same Excel file as the one used previously.
    - Select OK.
  - G. Generate a Normalized Moment Curvature Diagram.

- Normalize the moments by dividing each of the positive moments by the maximum positive moment and the negative moment by the maximum negative moment, and divide the curvatures by the curvatures corresponding to the maximum and negative moments.
- Plot the normalized positive and negative moment curvatures on a chart.
- Create a hinge moment curvature plot on the same chart with four positive moment points and four negative moment points without generating a negative slope.

1	A	8	C	D	E	F.	G	н	- E	1	K	L	M	N	0	P	Q	R	S	T	U	V.	W
2 Pc	sitive																						
3 Co	oncrete :	Neutral A:	Steel Strail	Tendon St	Concrete	Steel Com	Steel Ten:	Prestress	Net Force	Curvature I	Moment		M	0									
4	0	0	0	0	0	0	0	0	0	0	0		0	0.00							10		
5 -5	.83E-04	-0.202	9.54E-04	0	-1023	-134.148	1157.457	0	0.6999	1.39E-05	92327		0.424291	0.14							1.9		
6 -1	.39E-03	1.8021	2.46E-03	0	-2182	-278.572	2460.218	0	-0.5396	3.47E-05	188632		0.866863	0.36									
7 -2	.20E-03	6.5568	4.72E-03	0	-2709	-258.87	2968.071	0	-4.31E-02	6.24E-05	210140		0.965704	0.64							1 84		•
8 -3	.16E-03	9.2404	7.60E-03	0	-3022	-189.735	3211.643	0	-0.0135	9.71E-05	217603		1	1.00							14	100000	
9 -4	.38E-03	10.2529	0.011	0	-3170	-162.774	3332.986	0	0.022	0.000139	216988		0.997174	1.43							0.5		
10 -5	.91E-03	10.2344	1.48E-02	0	-3162	-221.373	3383.742	0	0.4763	0.000187	213039		0.979026	1.93									
11 -7	71E-03	10.0496	1.92E-02	0	-3122	-281.487	3403,438	0	-0.0672	0.000243	210855		0.968989	2.50							-0		
12	-0.0198	-22.9408	0.014	0	0	-1854	1854.169	0	-0.0301	0.000305	148132		0.680744	3.14		-44	1.00	30.00	-20.00	-10.00	0.00	10,00	20.00
13	-0.0242	-22.8196	0.0173	0	0	-1858	1858.773	0	0.3035	0.000375	148558		0.682702	3.86							-0.5		
14	-0.0292	-22.9173	0.0208	0	0	-1871	1871.122	0	0.052	0.000451	149608		0.687527	4.64									
15	-0.0349	-23.6182	0.0242	0	0	-1906	1906.531	0	0.1244	0.000534	152481		0.70073	5.50							1		
16	-0.0411	-24.0453	0.028	0	0	-1961	1960.493	0	-9.66E-02	0.000624	156827		0.720702	6.43					1	- ARADA	4446		
17	-0.0476	-24.1875	0.0323	0	0	-2034	2033.797	0	0.0141	0.000722	162696		0.747674	7.43			-						
18	-0.0546	-24.3215	0.0369	0	0	-2115	2115.214	0	0.015	0.000826	169211		0.777613	8.50							-15		
19	-0.0618	-24.2485	0.0419	0	0	-2189	2189.226	0	0.1694	0.000937	175107		0.804709	9.64									
20	-0.0693	-23.9243	0.0475	0	0	-2249	2249.521	0	0.0233	0.001055	179894		0.826707	10.86									
21	-0.0772	-23.6615	0.0534	0	0	+2307	2306.81	0	0.2443	0.001179	184442		0.847608	12.14		-1.36	-35						
22	-0.0859	-23.727	0.0593	0	0	-2351	2350.667	0	0.0432	0.001311	187939		0.863678	13.50		-1.36	-24						
23	-0.0951	-23.7901	0.0655	0	0	-2395	2394.726	0	0.0178	0.00145	191454		0.879832	14.93		-1	-1						
24																-0.7	-0.56						
25																0	0						
26 N	egative															0.966	0.64						
27 Co	oncrete !!	Neutral A:	Steel Strai	Tendon St	Concrete	Steel Com	Steel Tens	Prestress	Net Force	Curvature	Moment					1	1						
28	0	0	0	0	0	0	0	0	0	0	0		0	0.00		1	3						
29 -7	.27E-04	16.5861	5.20E-04	0	-374.912	-698.732	323.5235	0	-750.12	1.39E-05	59719		-0.42617	-0.22		1	13						

- H. Define Hinge Length.
  - The hinge length is one half of the section depth.
    - Hinge_{long} = 0.5 * (Deck thickness + haunch height + top flange thickness + web height + bottom flange thickness):
      - a.  $Hinge_{long} = 45$  in.
    - $Hinge_{Frac} = 0.5 * (Deck thickness + haunch height):$ a.  $Hinge_{Frac} = 7 in.$
    - $Hinge_{Trans} = 0.5 * (Deck thickness):$

a. 
$$Hinge_{Trans} = 4$$
 in.

- I. Make the plastic hinge in SAP2000.
  - Select Define > Section Properties > Hinge Properties > Add New Properties.
  - In the Type window, select Moment Curvature and input the corresponding correct hinge length.
  - In the Moment Curvature table, insert the four positive and four negative normalized moment curvatures and the zero point.
  - Uncheck the symmetric box and select the Is Extrapolated option in the Load Carrying Capacity beyond Point E window.
  - In the Scaling for Moment and Curvature window, insert the maximum positive moment and corresponding curvature as well as the maximum negative curvature and corresponding curvature.

- In the Acceptance Criteria, use the values 1, 2, and 3 for Immediate Occupancy, Life Safety, and Collapse Prevention in the positive column and -1, -2, and -3 for the negative column.
- Repeat for all remaining frame sections.

F Curvature/SF -35. -24.		O Moment - Rotation
-35. -24.		
-24.		Moment - Curvature
		Hinge Length 45.
-1.		
-0.56		Relative Length
0.		Hysteresis Type And Parameters
0.64		
1.		Hysteresis Type Isotropic 🗸
3.	Symmetric	No Parameters Are Required For This
13		Hysteresis Type
d Curvature		
d Curvature	Positive         Negative           217603.         140128.	
d Curvature nt Moment SF ture Curvature SF nly)	Positive         Negative           217603.         140128.           9.710E-05         6.240E-05	
d Curvature It Moment SF ture Curvature SF inly) Vastic Curvature/SF)	Positive         Negative           217603.         140128.           9.710E-05         6.240E-05           Positive         Negative	
d Curvature It Moment SF ture Curvature SF Inly) Plastic Curvature/SF) supancy	Positive         Negative           217603.         140128.           9.710E-05         6.240E-05           Positive         Negative           1.         -1.	
d Curvature ht Moment SF ture Curvature SF inly) Plastic Curvature/SF) supancy	Positive         Negative           217603.         140128.           9.710E-05         6.240E-05           Positive         Negative           1.         -1.           2         2	
d Curvature ht Moment SF ture Curvature SF inly) Plastic Curvature/SF) supancy	Positive         Negative           217603.         140128.           9.710E-05         6.240E-05           Positive         Negative           1.         -1.           2.         -2.	OK Cancel
	0. 0. 0.64 1. 3. 43 y Beyond Point E	0.64 1. 3. y Beyond Point E

- 9. Assign frame members to grid.
  - A. Select the Draw tab > Quick Draw Frame/Cable/Tendon.
  - B. In the Section drop-down menu, select the appropriate cross-section and click on the grillage grid member.



- 10. Assign hinges to frame elements.
  - A. The longitudinal hinges are placed at the ends of the longitudinal frame elements or at a relative distance of 0 and 1.
  - B. The transverse hinges are placed at a distance of half a top flange width away from the node.

    - Hinge Loc._{F to E} =  $\frac{half flange width}{Element Width} = 1 \frac{\frac{24}{2}}{39} = 0.6923.$  Hinge Loc._{E to D and C to B} =  $\frac{half flange width}{Element Width} = \frac{24/2}{96} = 0.125 and (1 10.125)$
    - 0.125) or 0.875. Hinge  $Loc_{.D \ to \ C} = \frac{half \ flange \ width}{Element \ Width} = \frac{24/2}{90} = 0.1333 \ and \ (1 1)^{-1}$ 0.1333) or 0.8667.

• Hinge Loc._{B to A} = 
$$\frac{half flange width}{Element Width} = \frac{24/2}{39} = 0.3077$$

- C. In SAP2000, assign the hinges to corresponding frame elements.
  - Select the desired frame elements you wish to assign hinges to, such as Long1Out. (The elements will turn from blue to yellow).
  - In SAP2000, select the Assign tab > Frame > Hinges.
- From the drop-down menu, select Long1Out and set relative distance to 0 and click ADD.
- From the drop-down menu, select Long1Out and set relative distance to 1 and click ADD.
- Click OK.

🂐 Assign Frame Hinges			×							
Frame Hinge Assignment Data										
Hinge Property	Relative Distance									
Auto v	0									
L1.1OUT	0									
L1.1OUT	1	Add Hinge								
		Modify/Show Auto Hinge								
		Delete Hinge								
Current Hinge Information No hinge is currently selected Options O Add Specified Hinge Assi	Current Hinge Information No hinge is currently selected									
Replace Existing Hinge As	signs with Specific	ed Hinge Assigns								
Existing Hinge Assignments of Number of Selected Frame C Total Number of Hinges on A All 2 existing hinge assignme	on Currently Select bjects: 1 Il Selected Frame nts will be remove	<u>ed Frame Objects</u> Objects: 2 d when the above hinge assignment is applied	ł							
Fill Form with Hinges on Selected Frame Object										
	OK Clo	se Apply								

• Repeat the previous step of assigning the hinges for all other frame elements.



- 11. Assign loads to the frame elements.
  - A. Wheel Axle Loads.
    - Axle loads will be placed at distances of 36 in., 108 in., 180 in., and 252 in. from the outside of the curved edge.
    - One line of the 16 kip axles will be placed at the 0.4L point of the bridge, or Transverse Grid 9, with the second line 14 ft away at Transverse Grid 7. One line of 4 kip axles will be placed 14 ft away from the first line of axles, at Gridline 11.
    - In SAP2000, the loads have to be placed at a relative distance, so this value needs to be calculated.

$$\begin{array}{l} \circ \quad HS20_{Axel\ 1\ Loc\ (B-A)} = \frac{L_1 - 36}{L_1} = \frac{39 - 36}{39} = 0.0792. \\ \circ \quad HS20_{Axel\ 2\ Loc\ (C-B)} = \frac{L_1 + L_2 - 108}{L_2} = \frac{39 + 96 - 108}{96} = 0.2813. \\ \circ \quad HS20_{Axel\ 3\ Loc\ (D-C)} = \frac{L_1 + L_2 + L_3 - 180}{L_3} = \frac{39 + 96 + 90 - 180}{90} = 0.5. \\ \circ \quad HS20_{Axel\ 4\ Loc\ (E-D)} = \frac{L_1 + L_2 + L_3 + L_4 - 252}{L_4} = \frac{39 + 96 + 90 + 96 - 252}{90} = 0.7188. \end{array}$$

- B. Lane Loads
  - Lane Loads are lane loads of 0.640 kip/ft (0.05333 kip/in.) centered at a distance of 96 in. and 240 in. from outside of the curved edge.
  - These lane loads will be placed on the longitudinal frame elements. They will be assigned to elements along the B, C, D, and E longitudinal elements according to the appropriate tributary distance.
    - $LaneLoad_B = \left(\frac{L_1 + L_2 96}{L_2}\right) * laneload = \left(\frac{39 + 96 96}{96}\right) * 0.05333 = 0.021667 \frac{kip}{in}.$
    - $LaneLoad_{C} = laneload laneload_{B} = 0.05333 0.021667 = 0.031667 \frac{kip}{in}$ .

- $\begin{array}{l} \circ \quad LaneLoad_{D} = \left(\frac{L_{1}+L_{2}+L_{3}+L_{4}-240}{L_{4}}\right)*laneload = \\ \left(\frac{39+96+90+96-240}{96}\right)*0.05333 = 0.045\frac{kip}{in.} \\ \circ \quad LaneLoad_{E} = laneload laneload_{D} = 0.05333 0.045 = \\ 0.00833\frac{kip}{in.} \end{array}$
- C. In SAP2000, the load patterns must first be defined.
  - Select the Define tab > Load Patterns.
    - Under the Load Pattern Name, enter HS20_1 and change the type in the drop-down menu to Live. The self-weight multiplier should be set to 0. Then click Add New Load Pattern.
    - Under the Load Pattern Name, enter LaneLoad1 and change the type in the drop-down menu to Live. The self-weight multiplier should be set to 0. Then click Add New Load Pattern.
    - o Click OK.



- Assign the wheel loads in SAP2000.
  - Select the exterior transverse element of Gridlines 9 and 7.
  - Click Assign > Frame Loads > Point.
  - From the Load Pattern drop-down menu, select HS20 and verify that the coordinate system is set to Global, the Load Direction is Gravity, and the Load Type is Force.
  - In Column 1, enter a relative distance of 0 (Axle 1, Loc. B-A) and load of 16 kips.
  - Click OK.
  - Repeat for Gridline 11 to assign the 4 kip load.
  - Repeat Steps 1–6 for Axle 2, 3, and 4, Loc. C-B, D-C, and E-D.



- Assign the Lane Load in SAP2000.
  - Select all exterior longitudinal frame elements along Gridline B.
  - Click Assign > Frame Loads > Distributed
  - From the Load Pattern drop-down menu, select Lane Load and verify that the coordinate system is set to Global, the Load Direction is Gravity, and the Load Type is Force.
  - In the Uniform Load box, enter 0.021667 (Lane Load B).
  - o Click OK.
  - Repeat Steps 1–5 for all of the longitudinal elements along gridlines C, D, and E.



### 12. Define load cases.

- A. The load case being used to determine redundancy is 1.25DL + 1.75(LL + IM). Where DL = Dead Load, LL = Live Load, and IM = Impact Load. When substituting in the truck load and the lane load, the preceding equation reduces to 1.25DL + 1.75LaneLoad + 2.33HS20.
- B. Generate Load Cases in SAP2000.
  - Click Define > Load Cases > Add New Load Case.
  - In the Load Case Name Panel, name the load case "LC1_1."
  - In the Analysis Type, select "Nonlinear."

- For the LC1_1 Load Case in the Stiffness to Use panel, select "Zero Initial Conditions."
- In the Loads Applied panel, leave the Load Type "Load Pattern" in the drop-down Load Name menu, select DEAD and change the Scale Factor to 1.25. Click Add. Change the Load Name menu, select HS20, and change the Scale Factor to 2.33. Click ADD. Change the Load Name menu, select Lane Load, and change the Scale Factor to 1.75. Click ADD.
- In the Other Parameters panel, in the Results Saved section, click Modify/Show.
- In the Results Saved for Nonlinear Static Load Cases window, change the Results Saved to Multiple States, and in the For Each Stage panel, change the Minimum Number of Saved Steps and the Maximum Number of Saved Steps to 20. Click OK.
- Click the OK on the Load Case Data window.
- Repeat the previous 8 steps listed out in B to create an LC2_1, LC3_1, and LC4_1. However, in the Initial Conditions window, select Continue from State at End of Nonlinear Case, and from the drop-down menu select the preceding load case. (For LC2_1, the Nonlinear Case LC1_1 would be selected).
- Repeat the immediately previous step to create LC(1–4)_2 for Span 2 and LC(1–4)_3 for Span 3.

oad Case Name		Notes	Load Case Type
LC1_1	Set Def Nan	Modify/Show	Static V Design
nitial Conditions			Analysis Type
Zero Initial Condition	ns - Start from Unstressed State		O Linear
O Continue from State	e at End of Nonlinear Case	$\sim$	Nonlinear
Important Note:	Loads from this previous case are	included in the current case	O Nonlinear Staged Construction
Iodal Load Case			Geometric Nonlinearity Parameters
All Modal Loads Appli	ed Use Modes from Case	MODAL $\sim$	None
anda Applied			O P-Delta
Load Type	Load Name	Scale Factor	O P-Delta plus Large Displacements
Load Pattern V	DEAD v	1.25	Mass Source
Load Pattern	DEAD	1.25 Add	Previous 🗸
Load Pattern	HS20_1	2.33	
Load Pattern	LaneLoadi	Delete	
)ther Parameters			
Load Application	Full Load	Modify/Show	UK
Results Saved	Multiple States	Modify/Show	Cancel
Nankara Davantara	Default	Madifield	

- 13. Define end supports.
  - A. The elastomeric bearing pads for each girder have a lateral stiffness of 12 kip/in. and a vertical stiffness of 6100 kip/in. Since the tub girders are divided in half, the lateral stiffness will be 6 kip/in., and the vertical stiffness will be 3050 kip/in.
  - B. Assign spring supports in SAP2000.
    - Select the 16 nodes—8 at the very end of the longitudinal members, 8 at the location of the piers.
    - Click Assign > Joint > Springs.
    - In the Assign Joint Springs window in the Simple Springs Stiffness panel, enter 6 for Translation 1 and 2 and 3050 for Translation 3.
    - Click OK.



- 14. Define CL data acquisition points mid-span of girder.
  - A. Select the transverse frame elements between B and C and between D and E at 0.4L (Gridline 9).
  - B. Click Edit > Edit Lines > Divide Frames.
  - C. In the Divide into Specified Number of Frames window, enter 2 for Number of Frames.
  - D. Click OK.



- 15. Analyze the nonfracture structure for dead load only.
  - A. In SAP2000, click Analyze > Run Analysis.
  - B. In the Set Load Cases to Run window, click the Run/Do Not Run All button until every action is Do Not Run.
  - C. Select DEAD then click Run/Do Not Run Case until the action is run.
  - D. Then click Run Now. Let SAP2000 Run the Load Cases until the screen says the analysis is complete.
  - E. Once the analysis is complete, select Spring Reactions.
  - F. Click Display > Show Tables.
  - G. In the Choose Table for Display window, click the + symbol beside Joint Output and select the square box beside Reactions.
  - H. In the Output Options window in the Nonlinear Static Results panel, select Last Step.
  - I. Select and copy the information from the F3 column.
  - J. The sum of the F3 values is the dead load.
  - K. Click Done.
  - L. Unlock the structure.
- 16. Analyze the nonfractured structure.
  - A. In SAP2000, click Analyze > Run Analysis.
  - B. In the Set Load Cases to Run window, click the Run/Do Not Run All button until every Action is Do Not Run.
  - C. Select LC1_1, LC2_1, LC3_1, and LC4_1, then click Run/Do Not Run Case until the action for all four is run.
  - D. Click Run Now.
  - E. Let SAP2000 Run the Load Cases until the screen says the analysis is complete.
  - F. Once the analysis is complete, select the data collection point on the transverse member on the outside girder (C-B).
  - G. Click Display > Show Tables.
  - H. In the Choose Table for Display window, click the + symbol beside Joint Output and select the square box beside Displacements.

💢 Choose Tables for Display	×
Edit	
Image: Constraint of the selected         Image: Constraint of the selected	Load Patterns (Model Def.) Select Load Patterns 3 of 3 Selected Load Cases (Results) Select Load Cases 4 of 4 Selected Modify/Show Options Set Output Selections Options Selection Only Show Unformatted
	Named Sets
	Save Named Set
	Show Named Set
	Delete Named Set
	OK Cancel
Table Formats File Current Table Formats File: Program Default	

- I. Click the Modify/Show Options button.
- J. In the Output Options window in the Nonlinear Static Results panel, select Stepby-Step.

Vode Shapes	Base Reactions Location		Buckling Modes				
Mode to	Global X 0	in	Mode to				
	Global Y 0	in					
All Modes	Global Z 0	in	All Modes				
Iodal History Results	Nonlinear Static Results		Steady State Results				
Envelopes	O Envelopes		Envelopes				
O Step-by-Step	Step-by-Step		<ul> <li>At Frequencies</li> </ul>				
O Last Step	O Last Step	In and Out of Phase $\qquad \lor$					
Direct History Results	Multi-step Static Results	Power Spectral Density Results					
Envelopes	Envelopes		RMS				
O Step-by-Step	O Step-by-Step		O sqrt(PSD)				
🔘 Last Step	🔘 Last Step						
.oad Combos	Bridge Design Results						
Envelopes	O Controlling Combo						
O Correspondence	O All Combos						
O Multiple values, if possible			OK Cancel				

- K. Click OK.
- L. Select and copy the information from the Output Case, StepNum Unitless, and the U3 in. column and paste them into an Excel worksheet. These columns represent Load Case, Step Number, and Deflection for the Outside Girder, respectively.

)	🗙 Joir	nt Displa	cemen	ts								-		$\times$
	File	View	Edit	Format-Filter	-Sort Select	Options								
Units: As Noted Filter:											~			
		Joint Text	t	OutputCase	CaseType Text	StepType Text	StepNum Unitless	U1 in	U2 in	U3 in	R1 Radians	R2 Radians	R3 Radians	^
	•	10	3	LC1	NonStatic	Step	0	(	0 0	0	0	0	(	D
		10	3	LC1	NonStatic	Step	1	0.002545	5 7.7E-05	-0.087926	-2.681E-06	0.000169	5.324E-08	3
		10	3	LC1	NonStatic	Step	2	0.00509	0.000154	-0.175851	-5.363E-06	0.000338	1.065E-07	7

- M. Click Done.
- N. Select the data collection point on the transverse member on the inside girder (E-D) and repeat Steps G–L. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- O. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element B) and repeat Steps G–L. This information goes into the Delta 4 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- P. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element C) and repeat Steps G–L. This information goes into the Delta 3 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- Q. Select the joint on the transverse member on the inside girder at Transverse Element 9 (at Longitudinal Element D) and repeat Steps G–Ll. This information goes into the Delta 2 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.
- R. Select the joint on the transverse member on the outside girder at Transverse Element 9 (at Longitudinal Element E) and repeat Steps G–L. This information goes into the Delta 1 column. However, for Step L, there is no need to copy Output Case, Step Num Unitless again.

	Α	8	с	D	E	F	G	н	1	J	К	L	м	v	х	Y	Z	AA	AB	AC	AD
1	Intact				OG-CL	IG	CL	Point 4	Point 3	Point 2	Point 1			Applied		0	G	IC	3		
2	Load Case	Load Step	Ω	Delta (in)	Chord (rad)	Delta (in)	Chord (rad)	Delta (in)	Delta (in)	Delta (in)	Delta (in)	α 23 (rad)	α 32 (rad)	Load (kip)	Ω (cal)	Delta (in)	Chord (deg)	Delta (in)	Chord (deg)	α 23 (deg)	α 32 (deg)
3	LC1	0	0	0	0.00000	0	0.00000	0	0	0	0	0	0	0	0.00	0.00	0.00000	0.00	0.00000	0	0
4	LC1	1	0.05	-0.1037	0.00015	-0.087926	0.00013	-0.10541	-0.10129	-0.0942	-0.07905	7.89E-05	4.94E-05	66	0.05	0.10	0.00861	0.09	0.00730	0.004522	0.002831
5	LC1	2	0.1	-0.2074	0.00030	-0.175851	0.00025	-0.21081	-0.20259	-0.18839	-0.15809	0.000158	9.88E-05	132	0.10	0.21	0.01722	0.18	0.01460	0.009043	0.005662

- S. Once all of the data are collected, unlock the model by selecting the Lock tool on the left hand side of SAP2000 screen.
- 17. Analyze the fractured structure.
  - A. At mid-span along Gridline 9, replace the hinges of the outside longitudinal element (Gridline B) with FracOUT hinges according to Step 10.
  - B. At mid-span along Gridline 9, replace the hinges of the first interior longitudinal element (Gridline C) with FracInt hinges according to Step 10.



C. Repeat Step 15 for the fractured case and collect the data accordingly. 18. Post-process the data.

- A. In the Excel sheet, the following values need to be calculated for each step.
  - Omega ( $\Omega$ ):

$$\circ \quad \Omega_i = \Omega_{i-1} + \left(\frac{1}{\# \, of \, Steps \, in \, Load \, case}\right).$$

- Longitudinal Chord Rotation of Interior and Exterior Girder:
  - Chord Rot._{Single Span} =  $-1 * \left(\frac{\delta_{CL}}{0.5 * L}\right)$  (rad).
- Transverse Deck Rotation:

$$\alpha_{2-3} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_2 - \delta_1}{w}\right) \text{ (rad).}$$
  
 
$$\alpha_{2-3} = \left(\frac{\delta_3 - \delta_2}{s}\right) - \left(\frac{\delta_2 - \delta_1}{w}\right) \text{ (rad).}$$

- Where s = spacing between the interior top flanges of the inside and outside girders and w = spacing between the top flanges of the same girder.
- Applied Load:
  - ο Calculate unit applied load or applied load at 1 Ω.
  - Unit Applied Load_{single span} = 1.25 * Total Reactions from Dead Load Case + 2 * (2.33 * HS20 truck + 1.75 * Lane Load).
  - Applied Load = Unit Applied Load  $* \Omega$ .
- B. Repeat Step A for the fractured case.
- C. Calculate the initial stiffness for the intact bridge and the instantaneous stiffness for the fractured bridge.
  - For the nonfractured condition (intact bridge) find the absolute displacement for the outside girder at an Ω value of 0.4:

• Initial Stiffness =  $\frac{0.4}{Absolute Displacement OG (at \Omega=0.4)}$ .

- For the Fractured case, add an additional column labeled stiffness:
  - Instantaneous Stiffness_{OG-Frac.i} =  $\frac{\Omega_i \Omega_{i-1}}{\delta_i \delta_{i-1}}$ .
- D. Failure of the structure occurs at the  $\Omega$  of the fractured bridge at the first of the following criteria:
  - The instantaneous stiffness for the fractured outside girder is less than 5 percent of the initial stiffness of the intact outside girder.
  - The chord angle of the outside girder for a simple span or interior spans is greater than 2 degrees. The chord angle for exterior spans of multi-span bridges is greater than 3 degrees.
  - The transverse deck rotation is greater than 5 degrees.
- E. On a chart, plot the nonfractured outside and inside girder as well as the fractured outside and inside girder with displacement on the primary x axis and the total force on the primary y axis and  $\Omega$  on the secondary y axis.



19. Repeat Steps 11–18 for Span 2 and 3 (Span 1 is pictured below).

#### 7.3 OVERALL RESULTS FOR BRIDGE 10

Table 7.1 and Figure 7.2 presents the overall results for Bridge 10. Similar to the previous sections, a comparison between the three methods is drawn using the results of the yield line band, which is flanked by the upper-bound and the lower-bound solutions, and the overstrength (load) versus deflection plots of the results from grillage and FEM analyses. It is evident, as shown in Figure 7.2, that all three methods indicate that Bridge 10 could be reclassified as nonfracture critical.

Method of Analysis	Overstrength Factors	Span 1 (140 ft)	Span 2 (139.60 ft)	Span 3 (139.60 ft)
Yield Line	$arOmega_{Yield\ Line}^{Upper\ Bound}$	1.98	1.90	1.67
Theories	$arOmega_{Yield\ Line}^{Lower\ Bound}$	1.88	1.84	1.59
Grillage Analysis	$arOmega_{Grillage}$	1.71	1.25	1.25
FEM	$arOmega_{FEM}$	1.70	1.45	1.45

 Table 7.1. Results Summary of Bridge 10.



(a) Span 1,  $L_x = 148 ft$ 



(*b*) Span 2,  $L_x = 265 ft$ 



Figure 7.2. Comparison of the Results for Bridge 10.

## 8. CLOSURE AND CAVEATS

In comparison to the more accurate FEM analysis, the yield line and grillage push-down methods are reputable simplified analysis models for predicting the overstrength capacity of TTGBs with one fractured girder.

When compared to the FEM and yield line methods, the grillage analysis tends to provide a lower-bound solution for overstrength factor and thereby underestimates the reserve capacity. The yield line method is unable to determine displacement limitations. Therefore, either the grillage or FEM approaches are needed to identify the overall deflection limits based on slab hinge (yield line) rotational restrictions.

The key features of the three methods are discussed as follows:

- Yield line method:
  - The least intensive modeling method (by hand) with limited parameters. The yield line method includes the plates of the steel tubs, reinforcing bars, and the deck concrete.
  - Upper-bound and lower-bound yield line solutions are indicative of strength limits.
  - The yield line approaches do not provide information on practical displacement limitations.
- Grillage method:
  - The grillage method is a computational solution that is somewhat similar to the lower-bound strip methods of plastic analysis of slabs.
  - As an incremental analysis, the grillage approach is able to overcome the shortcoming of identifying displacement limits that are absent in the yield line approach.
  - The initial stiffness of the grillage is not captured well and thereby tends to be a lower-bound solution.
- FEM:
  - The nonlinear FEM is the most computationally intensive and accurate modeling approach.

- Because the method is a full 3D approach, other mechanisms that can only be explained by large-deflection theory may be identified.
- At large deflections, it is often observed that the FEM strength may exceed the upper-bound yield line solution. In the FEM, this is attributed to the development of tensile membrane action. Strength from such membrane action cannot be captured by the grillage and yield line methods because they are both predominately based on flexural collapse.
- It should be noted that while membrane action may theoretically be mobilized, the deflection generally exceeds the practical rotation limits that can be achieved in the slab that exists between the twin tubs. Therefore, the reserve strength capacity must be restricted to be not more than the yield line solution.

## REFERENCES

- ACI-318 (2017). "Building code requirements for structural concrete (ACI 318-17) and commentary." ACI Committee, American Concrete Institute, and International Organization for StandardizationDassault Systèmes. (2017). Abaqus, version 2017. Author, Providence, RI.
- FHWA. (2012). Clarification of requirements for fracture critical members. Memorandum from Office of Bridge Technology, Federal Highway Administration, Washington, DC.
- Park, R., and Gamble, W. L. (2000). Reinforced concrete slabs. New York, John Wiley & Sons.
- Wilson, E., and Habibullah, A. (1997). SAP2000: integrated finite element analysis and design of structures. Computers and Structures Inc., Berkeley, CA.

APPENDIX A. STRUCTURAL DRAWINGS—BRIDGE 2























A-8



A-9

# APPENDIX B. STRUCTURAL DRAWINGS—BRIDGE 5



















Installed in shop except at locations spanning a field splice, which shall be installed in the field.
APPENDIX C. STRUCTURAL DRAWINGS—BRIDGE 10





C-3









C-5











Installed in shop except at locations spanning a field splice, which shall be installed in the fleid.