

A DISCRETE-ELEMENT METHOD OF ANALYSIS FOR COMBINED BENDING
AND SHEAR DEFORMATIONS OF A BEAM

by

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Development of Methods for Computer Simulation
of Beam-Columns and Grid-Beam and Slab Systems

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PREFACE

This report presents a method for determining the contribution of shear deformations to the total deflections of beam-columns. The theory for the required discrete-element analytical procedure is developed and a computer program is documented. The method permits the solution of complex problems with any reasonable variation in member properties, loads or supports.

This is the twelfth in a series of reports that describe the work in Research Project No. 3-5-63-56, entitled "Development of Methods for Computer Simulation of Beam-Columns and Grid-Beam and Slab Systems." A review of earlier reports (see list of reports) will provide useful background information for the procedures presented herein.

The computer program described in this report is written in American Standard FORTRAN language for the CDC 6600 computer. Only minimal changes are required to make the program operational on other systems. Duplicate copies of the program deck and test data cards for the example problems may be obtained from the Center for Highway Research, The University of Texas at Austin.

This report is a product of the efforts of many people. Particular acknowledgement is made of the advice and assistance of Messrs. Hudson Matlock, Art Frakes and Don Fenner. The support of the U. S. Bureau of Public Roads is gratefully acknowledged. The excellent facilities of the Computation Center of The University of Texas and the cooperation of its staff have contributed significantly.

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ABSTRACT

A method of analysis for the combined effects of bending and shear deformations is presented. Although the method is primarily designed for the analysis of beams, it can be applied to other structural problems. The method is applicable to beams that have abrupt, or point-by-point, variations in their structural properties. Also, the beam may be subjected to any configuration of transverse or longitudinal loads, and it may be supported in any reasonable manner.

There are three important features of the method presented. First, a discrete-element model is substituted for the real structure. Second, a system of algebraic equations which describe the load-deflection behavior of the model is written. Finally, these equations are solved for the unknown deflections by a modified form of Gaussian elimination.

A computer program, SHRBM 1, which utilizes the method of analysis presented herein provides an efficient means of solving practical problems. The correct usage of the program and the generality of the method are demonstrated by a series of example problems.

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NOMENCLATURE

<u>Symbol</u>	<u>Typical Units</u>	<u>Definition</u>
a	in.	Distance from the joint at which the R force is applied
$a_{i,1} - a_{i,8}$	-	Terms in coefficient and load matrices
A	in^2	Cross-section area
A_s	in^2	Effective shear area
b	in.	Width of beam section
$b_{i,1} - b_{i,7}$	-	Terms in coefficient and load matrices
d	in.	Depth of beam section
E	lb/in^2	Modulus of elasticity
F	lb-in^2	Flexural stiffness = EI
G	lb/in^2	Shear modulus
h	in.	Increment length
i	-	Station and bar number
I	in^4	Moment of inertia
k	-	Constant = $\frac{A_s}{I^2} \iint \left(\frac{Q_y^2}{b^2} \right) dydz$
K	lb/in	Shear stiffness
L	in.	Beam length
M	in-lb	Bending moment

<u>Symbol</u>	<u>Typical Units</u>	<u>Definition</u>
P	lb	Longitudinal load
Q	lb	Transverse force
Q_y	in^3	First moment, about the neutral axis, of the cross-section area outside a distance y from the neutral axis of the beam
Q_R	lb	Support reaction
R	in-lb/rad	Rotational restraint
S	lb/in	Support spring stiffness
t	in.	Beam dimension
T	in-lb	Applied couple
V	lb	Shear
w	in.	Total vertical deflection
W_e	in-lb	External work
W_i	in-lb	Internal work
x	in.	Distance along the beam
Δx	in.	Incremental length of the beam
y	in.	Distance from the neutral axis of the beam
α	rad	Incremental angular deformation due to shear
γ	rad	Total angular deformation due to shear
δ_i	in.	Shear deflection in Bar i
θ_i	rad	Bar slope
$(\theta_J)_i$	rad	Average slope of two neighboring bars
τ_y	lb/in^2	Shear stress at a distance y from the neutral axis of the beam

<u>Symbol</u>	<u>Typical Units</u>	<u>Definition</u>
ϕ_i	rad	Angle change between two neighboring bars

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LIST OF REPORTS

Report No. 56-1, "A Finite-Element Method of Solution for Linearly Elastic Beam-Columns" by Hudson Matlock and T. Allan Haliburton, presents a finite-element solution for beam-columns that is a basic tool in subsequent reports.

Report No. 56-2, "A Computer Program to Analyze Bending of Bent Caps" by Hudson Matlock and Wayne E. Ingram, describes the application of the beam-column solution to the particular problem of bent caps.

Report No. 56-3, "A Finite-Element Method of Solution for Structural Frames" by Hudson Matlock and Berry Ray Grubbs, describes a solution for frames with no sway.

Report No. 56-4, "A Computer Program to Analyze Beam-Columns under Movable Loads" by Hudson Matlock and Thomas P. Taylor, describes the application of the beam-column solution to problems with any configuration of movable non-dynamic loads.

Report No. 56-5, "A Finite-Element Method for Bending Analysis of Layered Structural Systems" by Wayne B. Ingram and Hudson Matlock, describes an alternating-direction iteration method for solving two-dimensional systems of layered grids-over-beams and plates-over-beams.

Report No. 56-6, "Discontinuous Orthotropic Plates and Pavement Slabs" by W. Ronald Hudson and Hudson Matlock, describes an alternating-direction iteration method for solving complex two-dimensional plate and slab problems with emphasis on pavement slabs.

Report No. 56-7, "A Finite-Element Analysis of Structural Frames" by T. Allan Haliburton and Hudson Matlock, describes a method of analysis for rectangular plane frames with three degrees of freedom at each joint.

Report No. 56-8, "A Finite-Element Method for Transverse Vibrations of Beams and Plates" by Harold Salani and Hudson Matlock, describes an implicit procedure for determining the transient and steady-state vibrations of beams and plates, including pavement slabs.

Report No. 56-9, "A Direct Computer Solution for Plates and Pavement Slabs" by C. Fred Stelzer, Jr., and W. Ronald Hudson, describes a direct method for solving complex two-dimensional plate and slab problems.

Report No. 56-10, "A Finite-Element Method of Analysis for Composite Beams" by Thomas P. Taylor and Hudson Matlock, describes a method of analysis for composite beams with any degree of horizontal shear interaction.

Report No. 56-11, "A Discrete-Element Solution of Plates and Pavement Slabs Using a Variable-Increment-Length Model" by Charles M. Pearre, III, and W. Ronald Hudson, presents a method of solving for the deflected shape of freely discontinuous plates and pavement slabs subjected to a variety of loads.

Report No. 56-12, "A Discrete-Element Method of Analysis for Combined Bending and Shear Deformations of a Beam" by David F. Tankersley and William P. Dawkins, presents a method of analysis for the combined effects of bending and shear deformations.

Report No. 56-13, "A Discrete-Element Method of Multiple-Loading Analysis for Two-Way Bridge Floor Slabs" by John J. Panak and Hudson Matlock, includes a procedure for analysis of two-way bridge floor slabs continuous over many supports.

CHAPTER 1. INTRODUCTION

Purpose

The purpose of this study is to develop an efficient method for the analysis of beams in which shear deformations are significant. In the analysis of many structural problems, the dimensions of the members are such that it is sufficient to consider only the effects of bending deformations. However, for large-flange, thin-web beams and short, deep beams the effects of shear deformations are significant. By considering these effects in the analysis of the structure, a more flexible member is represented. Consequently, for some members a significantly different moment and shear variation is predicted than would be expected by neglecting shear deformations. Typical structural members which should be analyzed for shear deformations are girders for supporting bridge floor beams, traveling-crane girders, bridge diaphragms, and flexural components of aircraft structures.

Brief Description of the Method of Analysis

The method of analysis presented herein allows for a very general description of the physical problem. A wide variation in loads, supports, and beam section and material properties is possible. A basic requirement of the method is the necessity to solve a large number of simultaneous equations. This is not practical without the aid of a computer.

The proposed method of analysis consists of three fundamental steps. First, the actual structure is replaced by an idealized model composed of a system of discrete bars and springs. Second, two load-deflection equations are derived from the conditions of equilibrium and geometric compatibility by means of the force-deformation relations for the model. The load-deflection equations are written in terms of two unknown deflections: the shear deflection and the total deflection. When the two governing equations are written for each bar and joint in the model, a diagonally-banded set of equations results. Solution of this system of equations is the final step of the analysis.

A computer program has been written to perform the three steps outlined above. The physical problem is described according to a data input form that is provided and the analysis of a structure is automatically completed by the program.

Order of the Discussion

The assumptions of elementary beam theory and some methods of analysis are presented in Chapter 2. In Chapter 3 a discrete-element model representation of the actual system is introduced, and from this model the governing equations are derived. The method of solution is discussed in Chapter 4. A detailed explanation of the computer program used in the solution is given in Chapter 5. Several example problems are solved in Chapter 6 to illustrate the usage of the program. In Chapter 7 a summary of the method and some recommendations for further work are given.

CHAPTER 2. ELEMENTARY BEAM THEORY

Contents of the Chapter

The analysis of structural problems based on elementary beam theory incorporates several simplifying assumptions which yield sufficiently accurate results for most engineering problems. The assumptions related to the study of combined bending and shear deformations will be presented in this chapter. The behavior of a beam element due to the effects of bending and shear is examined. Also, some methods of analysis are discussed.

Bending Deformations

Bending deformations are generally the most significant part of the total deformation in a beam. For the case of pure bending, the beam curvature is related to the bending moment M and the flexural stiffness EI by

$$\frac{d^2w}{dx^2} = \frac{M}{EI} \quad (2.1)$$

where the second derivative of the deflection w with respect to the distance x along the beam is an approximation of the beam curvature. This relationship neglects both the axial and shear deformations of the element and is further limited by the following assumptions:

- (1) The beam is assumed to be initially straight.
- (2) Plane cross sections are assumed to remain plane.
- (3) Deflections are assumed to be small compared to the original cross-section dimensions.
- (4) The beam is assumed to be of symmetrical cross section about the plane of loading.
- (5) The material of the beam is assumed to be isotropic and linearly elastic.

Equation 2.1 is a force-deformation relation for a beam subjected to pure bending. However, it represents a good approximation of the beam behavior

when the deformations due to shear and axial loads are insignificant. A more general relation can be derived which includes the effect of shear deformations.

Shear Deformations

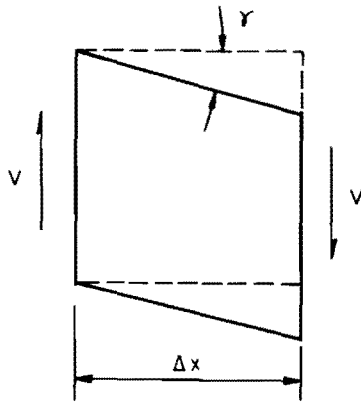
A force-deformation relation which accounts for the shear deflection is developed in this section by application of the method of virtual work to an incremental length of an idealized beam. The deformed shape of the beam element due to the shear V is illustrated in Fig 2.1(a). The indicated shear strain γ is the total shear deformation which occurs over a finite distance Δx of the beam, and it represents a change in slope of the element. An expression for the total shear deformation is derived by equating the external work to the internal work done by the unit shear on the element shown in Fig 2.1(b).

The shear stress τ_y on any fiber at a distance y from the neutral axis of a prismatic¹ beam is determined from elementary beam theory by satisfying the condition of longitudinal equilibrium of a beam element. Crandall (Ref 3) gives the expression for the shear stress distribution as

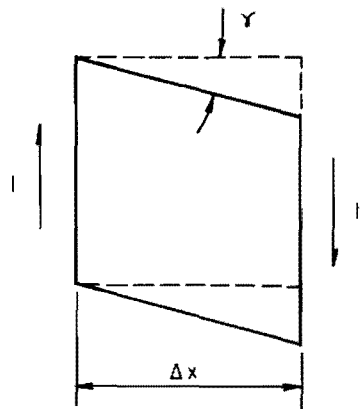
$$\tau_y = \frac{VQ_y}{Ib} \quad (2.2)$$

where Q_y is the first moment, with respect to the neutral axis, of the cross-section area which is outside a distance y from the neutral axis. The moment of inertia of the cross section, referred to the neutral axis, is represented by I , and the width of the cross section at the distance y is given by b . The shear distortion of a fiber of the beam, shown in cross section in Fig 2.1(c), is illustrated in Fig 2.1(d). The angle of this shear distortion at the distance y from the neutral axis is designated by α .

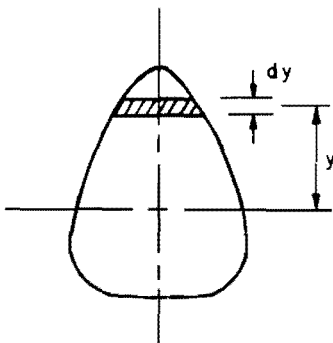
¹A more general expression for the shear stress distribution in symmetrically tapered beams is derived by Timoshenko (Ref 18).



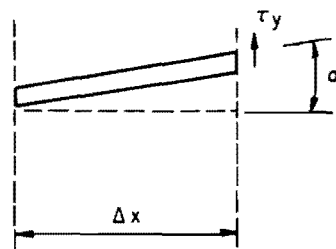
(a) Total shear on a beam element.



(b) Unit shear on a beam element.



(c) Beam cross section.



(d) Shear stress on a beam fiber.

Fig 2.1. Deformation of a beam element due to shear.

The internal work W_i done by the unit shear in distorting the beam element is

$$W_i = \int \frac{1 \cdot Q_y}{Ib} dA \cdot \alpha \cdot \Delta x \quad (2.3)$$

Since $dA = dydz$ and $\alpha = \frac{\tau_y}{G} = \frac{VQ_y}{GIb}$, where G is the shear modulus, this equation reduces to

$$W_i = \frac{V}{I^2 G} \iiint \frac{Q_y^2}{b} dydz \Delta x \quad (2.4)$$

The corresponding external work W_e done by the unit shear on the beam element is

$$W_e = 1 \cdot \gamma \cdot \Delta x \quad (2.5)$$

Combination of Eqs 2.4 and 2.5 gives

$$\gamma = \frac{V}{I^2 G} \iiint \frac{Q_y^2}{b} dydz \quad (2.6)$$

which relates the total shear strain on a finite beam element to the shear on the element. Evaluation of the double integral in Eq 2.6 for a particular beam cross section yields the equivalent relation

$$\gamma = k \frac{V}{A_s G} \quad , \quad k = \frac{A_s}{I^2} \iiint \frac{Q_y^2}{b} dydz \quad (2.7)$$

where A_s is the effective shear area. The value of the factor k depends on the shape of the cross section. For example, k is 1.2 for a rectangular cross section. Other values of k are given by Fife (Ref 5). Since k is very nearly equal to 1.0 for most cross sections, Eq 2.7 can be approximated by

$$\gamma = \frac{V}{A_s G} \quad (2.8)$$

Combined Bending and Shear Deformations

The total deformation due to bending and shear is determined by superposition. The justification for this can be obtained from an examination of the effect of shear deformations on a beam. For a beam of rectangular cross section, Eq 2.2 indicates a parabolic shear stress distribution as shown in Fig 2.2(a). The corresponding strain is also parabolically distributed. This implies that the originally plane cross sections distort in the manner shown in Fig 2.2(b). Thus, due to the effect of shear, plane sections do not remain plane as assumed. However, for a constant shear force, it is observed that any longitudinal line parallel to the neutral surface does not change its length as the beam element deforms. A constant shear force, therefore, has little effect on the bending strain and, hence, the bending strain distribution. Crandall (Ref 3) also indicates that the error due to a varying shear force is small, particularly for long slender beams. The combined effect of bending and shear may, therefore, be determined by superposition.

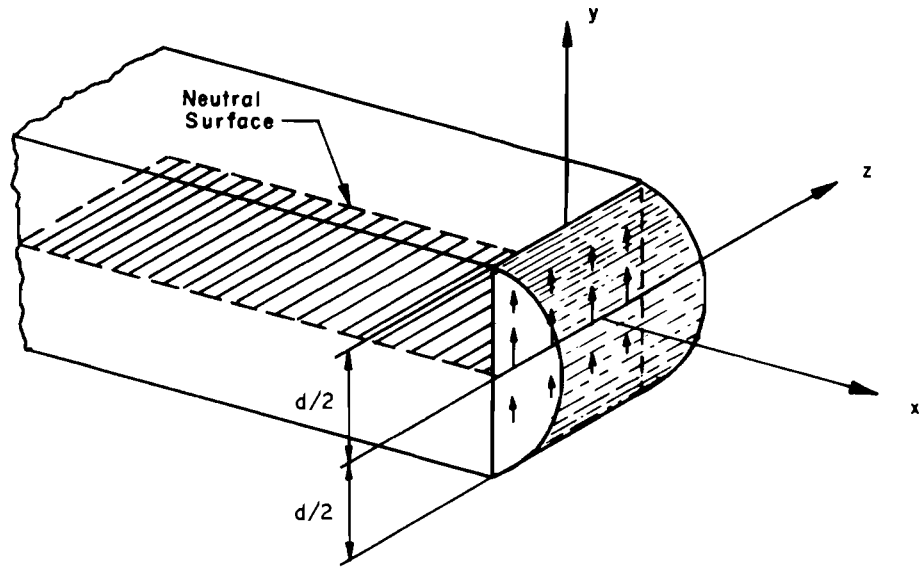
For the case of pure bending, the total slope $\frac{dw}{dx}$ of a beam element is due only to bending. Since shear deformations cause a change in slope equal to γ , the slope due to combined bending and shear is $\frac{dw}{dx} + \gamma$ by superposition. Substitution of this modified slope in Eq 2.1 yields the moment-curvature relation

$$\frac{d}{dx} \left(\frac{dw}{dx} + \gamma \right) = \frac{M}{EI} \quad (2.9)$$

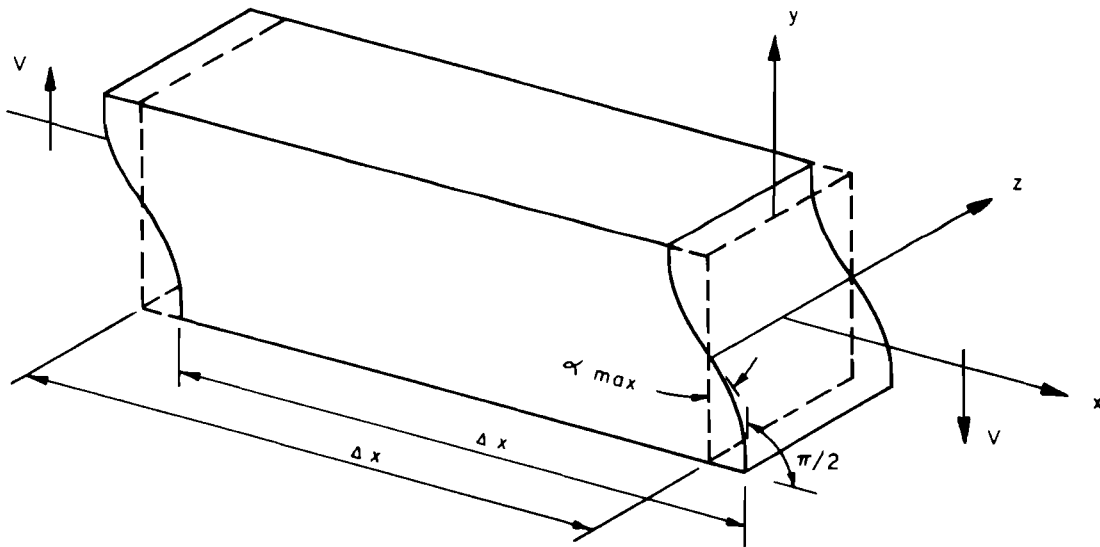
or

$$\frac{d^2 w}{dx^2} + \frac{d\gamma}{dx} = \frac{M}{EI} \quad (2.10)$$

which is the required relation.



(a) Parabolic shear stress distribution.



(b) Warped cross section due to parabolic shear stress distribution.

Fig 2.2. Effect of a constant shear force on a rectangular beam.

Methods of Analysis

Some methods of analysis which have contributed to this study of combined bending and shear deformations are mentioned in this section. No attempt is made to discuss each method in detail.

Two conventional methods of analysis are the virtual work method (Ref 10) and the conjugate beam method (Ref 17). With the virtual work method, deflections are determined by relating the internal energy of the structure to the external work done by a unit load. The conjugate beam method is a numerical technique for solution of the differential equation of the beam. Both of these methods utilize the principle of superposition to determine the combined effect due to shear and bending.

Numerical techniques have also been applied to the solution of beam problems by replacing the governing differential equation with finite-difference approximations of the derivatives.

Another approach to the problem is to model the structure physically by a system of discrete elements whose behavior can be described with algebraic equations. Newmark (Ref 7) was one of the first to use this method of analysis. For many problems, the finite-difference equations developed by direct substitution for the differential equation and the discrete-element model equations developed from a free-body analysis of the model are equivalent.

The method of analysis presented in this text has been greatly influenced by the numerical solutions for beam-columns on elastic foundations by Matlock (Ref 12). With this method the actual structure is replaced with a discrete-element model composed of rigid, weightless bars with hinged ends. The beam stiffness of each finite beam element is concentrated in the springs at the hinges. The development of the bar-and-spring model from a section of a beam element subjected to pure bending is shown in Fig 2.3.

Figure 2.3(a) shows an element of a beam subjected to bending moments and Fig 2.3(b) illustrates the stresses acting on the element. These distributed forces may be replaced by concentrated forces as indicated in Fig 2.3(c). The beam element may also be represented by the plates and springs as shown in Fig 2.3(d), and Fig 2.3(e) is an idealization of the structure by use of a series of these beam element models. A cruder model can be made by using rigid bars and springs as shown in Fig 2.3(f). Based on the latter model, a system of algebraic equations which express the load-deflection characteristics

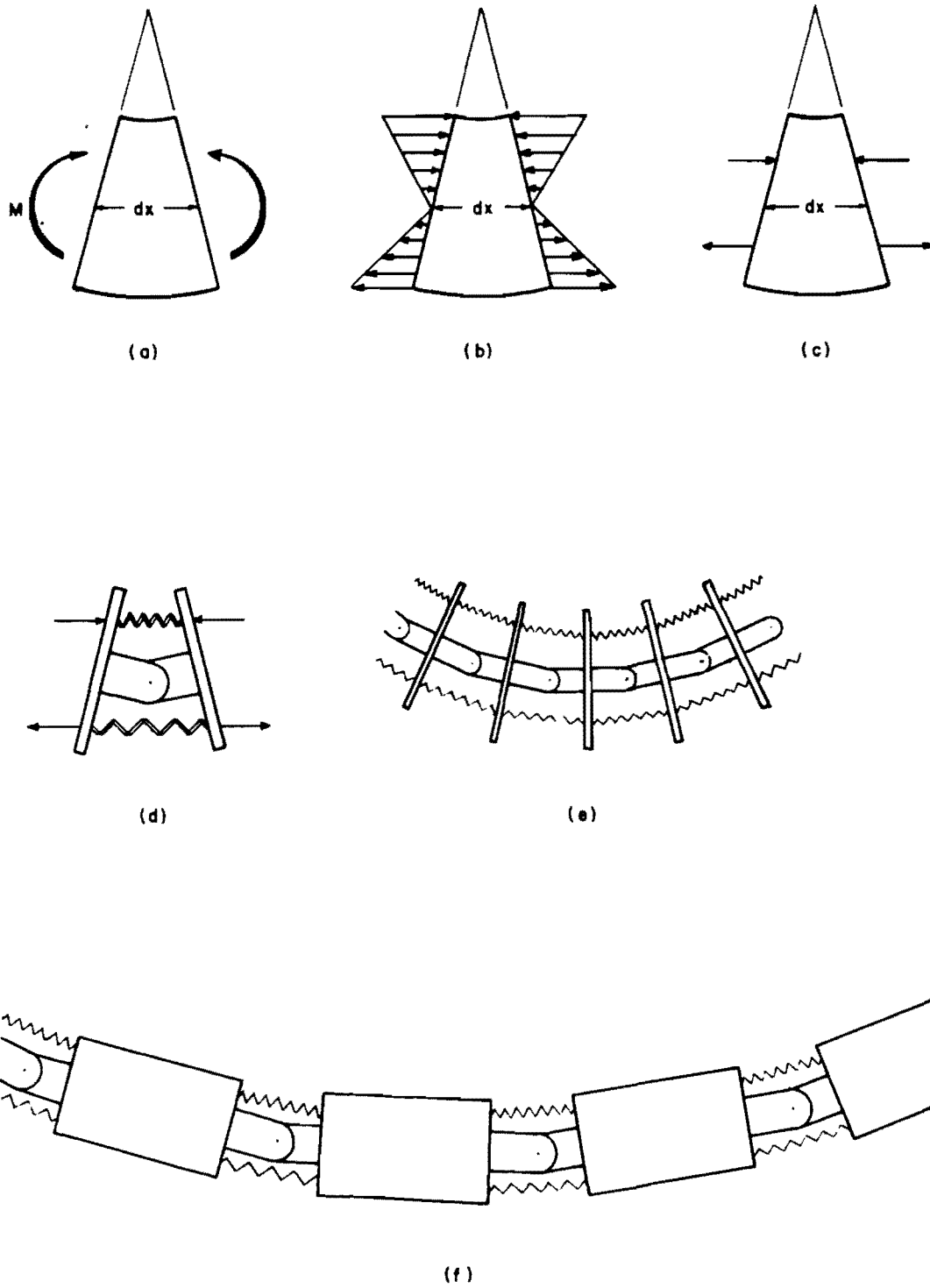


Fig 2.3. Discrete mechanical representation of a conventional beam.

of the system is derived. The accuracy of the solution for this model has been well documented previously (Ref 12).

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CHAPTER 3. DISCRETE-ELEMENT MODEL

Contents of the Chapter

In this chapter a discrete-element model of the actual beam is presented. From this model the equilibrium equations and the geometric compatibility equations are derived. From these equations and the beam force-deformation relationships, the load-deflection equations for the beam are developed.

The Bar-and-Spring Model

The model that is used to represent the actual system is illustrated in Fig 3.1. The beam is represented by a series of bars and springs. The bending stiffness F of the actual system is concentrated in the springs shown at the hinged joints between each bar. Likewise, the shear stiffness K is concentrated at the center of each bar. In this manner the shear deformation δ is confined to the bar, whereas the bending deformation ϕ occurs only at the joint.

An important feature of the method is that a very general description of the actual system is possible. A point-by-point variation of all loads and stiffnesses may be modeled. All properties of the real system are represented as "lumped" quantities at discrete points in the model. As indicated above, a distinction can be made between the quantities associated with the bar and those associated with the joint. Other terms which may be defined as properties of the bar are the slope θ , the shear V , the axial tension P , and the shear stiffness K . Typical joint properties are the total deflection w , the bending stiffness F , the bending moment M , the transverse load Q , and the support spring stiffness S .

The rotational restraint R and the applied couple T may be considered as affecting the joint. Both the R and the T terms are felt by the model as two equal and opposite forces as indicated in Fig 3.2. Typical units of the rotational restraint are in-lb/rad. Shear deformations are

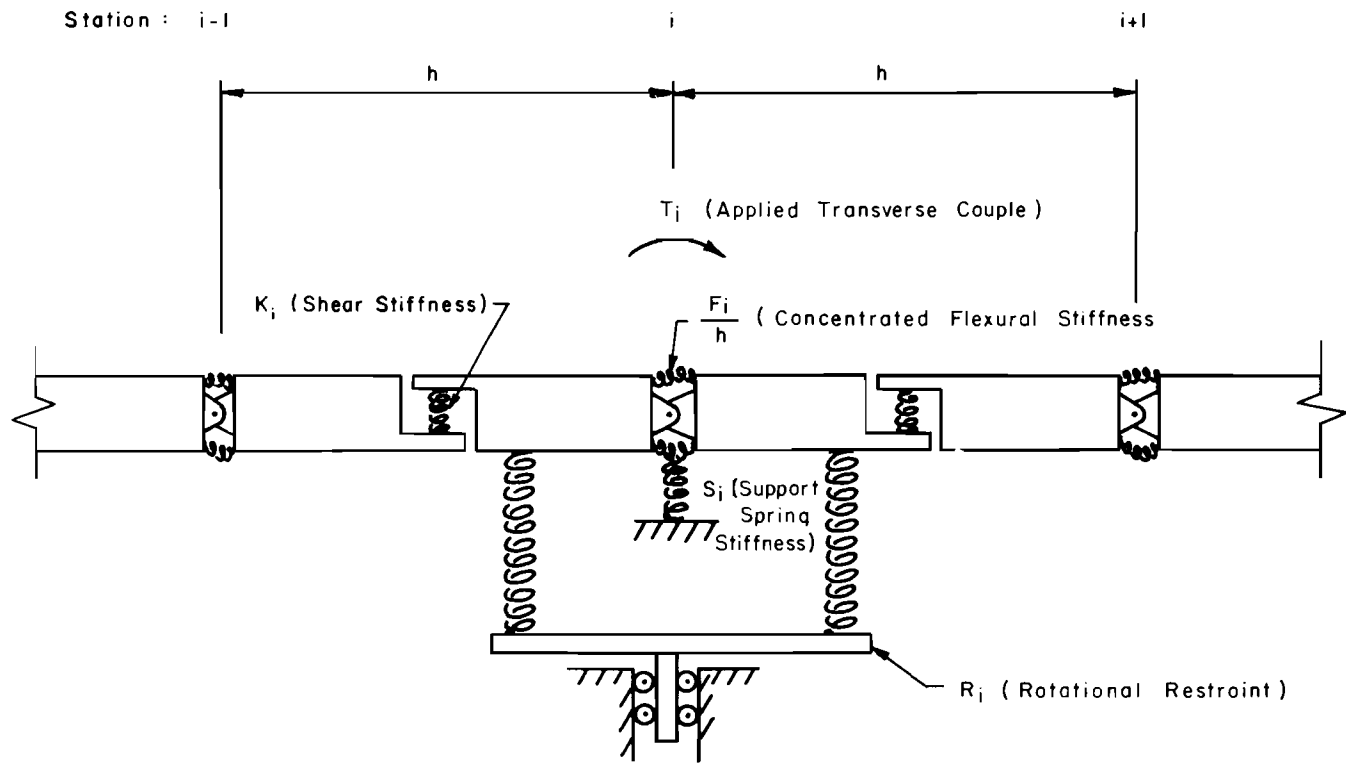


Fig 3.1. Bar-and-spring model.

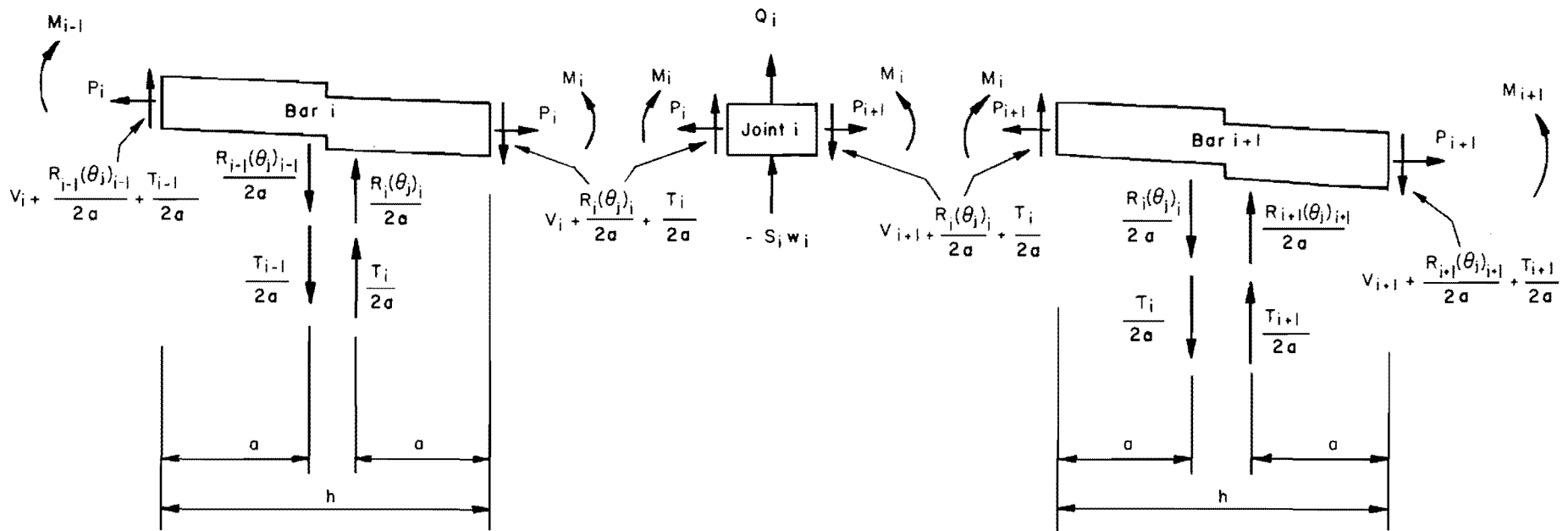


Fig 3.2. Free-body diagram.

unaltered when the rotational restraint forces and the applied couple forces are applied at the points indicated.

The sign convention which has been adopted is indicated in Fig 3.3. All quantities are shown in the positive sense. A subscript is used to identify the bar or joint with which each of the quantities is associated.

The entire model of the beam is composed of a number of these bar-and-spring elements. A typical bar-and-spring model representation of a beam is shown in Fig 3.4. Each joint, which is referred to as a "station," is numbered from left to right beginning with the left extremity of the beam as Station zero. In addition, two imaginary stations are defined at each end of the model. Likewise, the bars are numbered from left to right beginning with the first real bar as bar number one. It should be noted that all support points and end points of the actual beam correspond to joint locations in the model.

Equilibrium Equations

Two equations of equilibrium must be satisfied by the free-bodies shown in Fig 3.2. From the summation of vertical forces acting on Joint i

$$V_i + \frac{R_i(\theta_J)_i}{2a} + \frac{T_i}{2a} + Q_i - S_i w_i - \left(V_{i+1} + \frac{R_i(\theta_J)_i}{2a} + \frac{T_i}{2a} \right) = 0 \quad (3.1)$$

or, upon collecting like terms,

$$V_i - S_i w_i - V_{i+1} + Q_i = 0 \quad (3.2)$$

The second equilibrium condition to be satisfied, which results from the summation of moments about Station $i-1$ for Bar i , is

$$M_i - M_{i-1} - P_i \left(w_i - w_{i-1} \right) - \left(V_i + \frac{R_i(\theta_J)_i}{2a} + \frac{T_i}{2a} \right) h$$

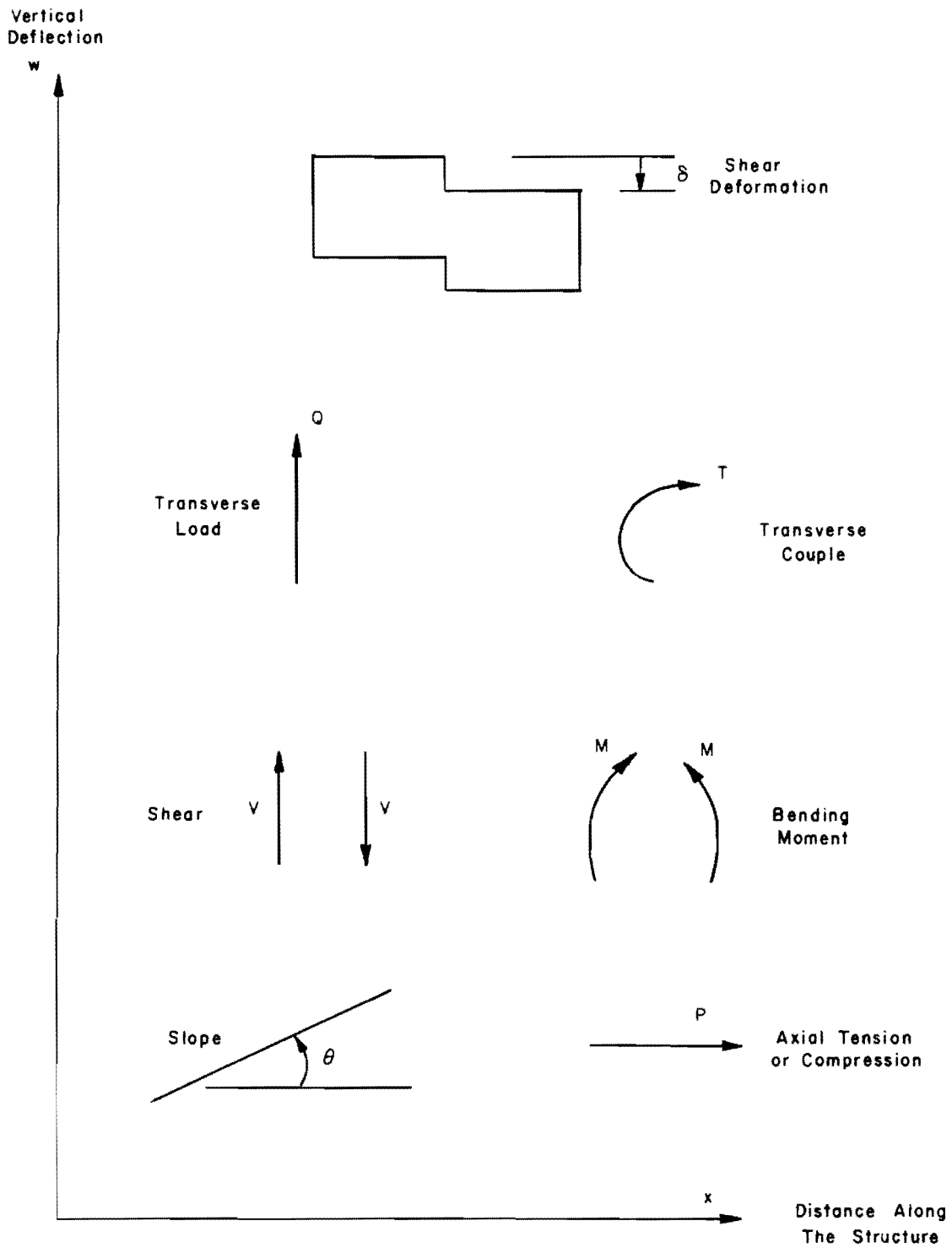
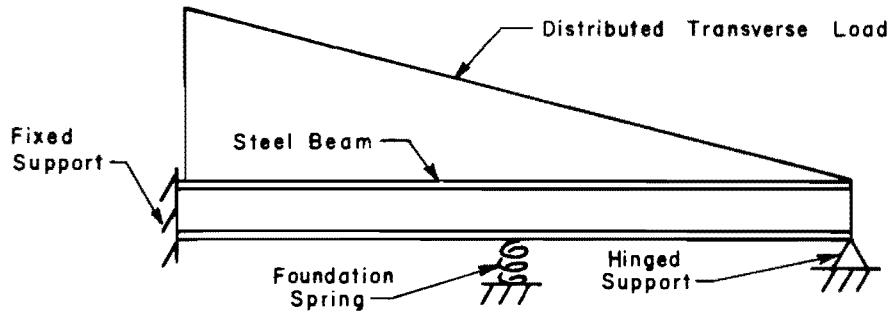
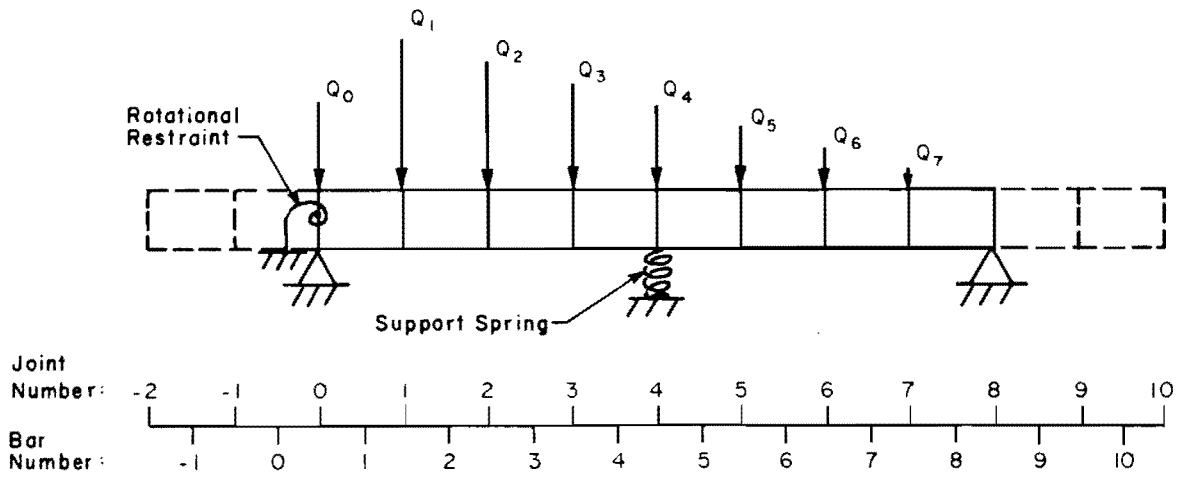


Fig 3.3. Sign convention. All quantities are shown in the positive sense with respect to distance along the beam, x .



(a) The actual beam to be analyzed.



(b) Bar-and-spring model of the beam.

Fig 3.4. Typical bar-and-spring model representation of a beam.

$$\begin{aligned}
& + \left(\frac{R_i(\theta_J)_i}{2a} + \frac{T_i}{2a} \right) (h - a) - \left(\frac{R_{i-1}(\theta_J)_{i-1}}{2a} + \frac{T_{i-1}}{2a} \right) a \\
& = 0
\end{aligned} \tag{3.3}$$

or after collecting like terms,

$$\begin{aligned}
M_i - M_{i-1} - P_i(w_i - w_{i-1}) - V_i h - \frac{R_i(\theta_J)_i}{2} - \frac{T_i}{2} - \frac{R_{i-1}(\theta_J)_{i-1}}{2} \\
- \frac{T_{i-1}}{2} = 0
\end{aligned} \tag{3.4}$$

Note that this relation is independent of the distance "a" at which the R and T terms are applied to the bar.

Geometric Compatibility

The deformed shape of the centerline of the model is illustrated in Fig 3.5. This configuration corresponds to positive deformations. As indicated, two deflections are defined for the model. The total deflection w is measured at each joint with the undeflected centerline of the model as a reference. The shear deflection δ is measured at the center of each bar and represents only the deflection due to shear in a single bar. Expressions for the bar slope, the joint rotation, and the relative angle change between adjacent bars can be formulated based on the deformed shape of the model.

The slope θ for the bar-and-spring model is defined as the slope of each bar. Thus, from the model geometry, the slope of Bar i is related to the deflections by

$$\theta_i = \frac{w_i - (w_{i-1} - \delta_i)}{h} \tag{3.5}$$

where h is the length of the bar. The slope of the model is not defined at the joint, but the slope of the actual beam represented by the model can be approximated as the average of the slopes of the adjacent bars:

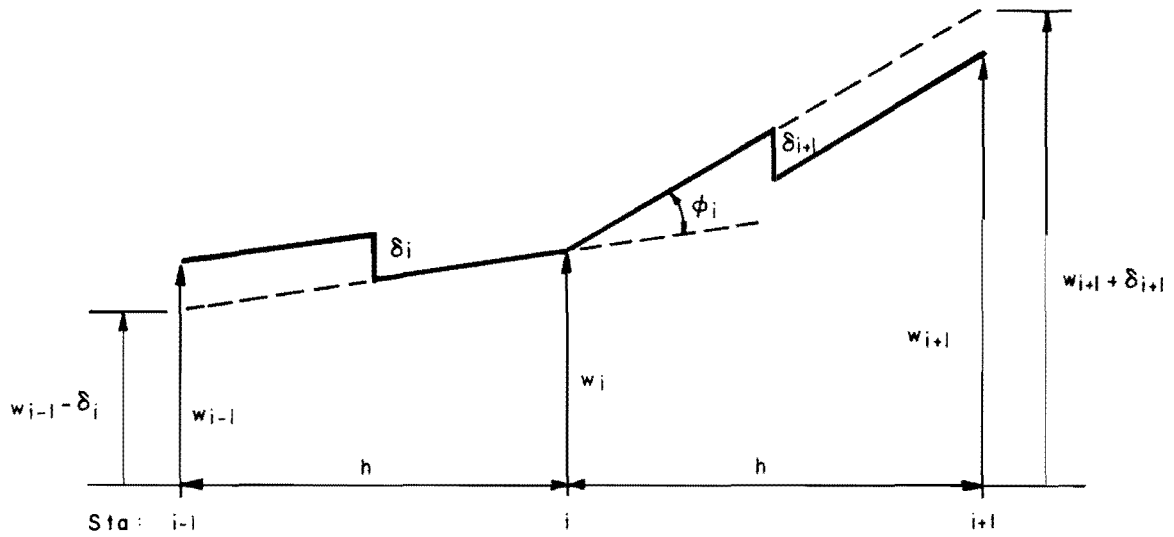


Fig 3.5. Deformed shape of the bar-and-spring model.

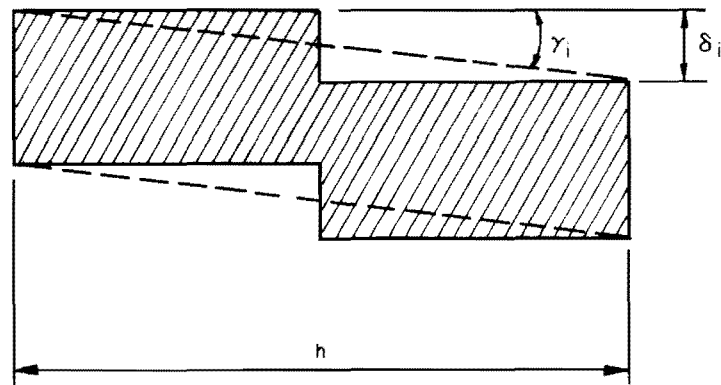


Fig 3.6. Geometric relation between a bar and an idealized beam element subjected to pure shear.

$$(\theta_J)_i = \frac{\theta_i + \theta_{i+1}}{2} = \frac{w_{i+1} + \delta_{i+1} - w_{i-1} + \delta_i}{2h} \quad (3.6)$$

This is the angle of rotation which occurs at Joint i.

The relative angle change between adjacent bars of the model represents the deformation due to bending. At Station i the angle change is determined from the difference in slopes of Bar i+1 and Bar i:

$$\phi_i = \theta_{i+1} - \theta_i = \frac{w_{i-1} - 2w_i + w_{i+1} - \delta_i + \delta_{i+1}}{h} \quad (3.7)$$

Each of the above expressions reduces to the relations for pure bending if the shear deflections are eliminated.

Force-Deformation Relations

In conventional beams, the bending moment is equal to the product of the flexural stiffness and the curvature. In the discrete-element model the flexibility of the beam and the curvature are lumped at the station points. The corresponding relation for bending moment M in the model is

$$M_i = \frac{F_i}{h} \phi_i \quad (3.8)$$

where the flexural stiffness F_i is equal to the product $E_i I_i$ and the relative angle change ϕ_i is given by Eq 3.7 (Ref 12).

The shear deflection has been lumped at the center of each bar in the model. The geometric relation between the beam element discussed in Chapter 2 and a bar of the model is illustrated in Fig 3.6 with the incremental length Δx replaced by the bar length h . The total shear strain γ is, therefore, related to the shear deflection δ by

$$\gamma = \frac{\delta}{h} \quad (3.9)$$

From Eqs 2.8 and 3.9 the shear in Bar i is

$$V_i = K_i \delta_i \quad (3.10)$$

where

$$K_i = \frac{G_i (A_s)_i}{h} \quad (3.11)$$

which is the desired shear spring constant for a bar. This shear spring constant implies a value of unity for the k factor discussed in Chapter 2. Although no explicit allowance has been made for other values of k , this effect can be considered by using $\frac{(A_s)_i}{k}$ in place of $(A_s)_i$.

By using Eq 3.10, it is possible to apply beam theory to the analysis of fabricated structures. From detailed considerations of the particular structure, an appropriate spring constant K_i representative of the shear stiffness may be determined. For example, a latticed strut or an airplane wing may be regarded as a beam (Ref 19). Even though Eq 3.11 does not apply for these structures, the simple relation between V and δ expressed in Eq 3.10 will still be admissible.

Load-Deflection Equations

The equilibrium equations can now be expressed in terms of the unknown deflections. Substitution of Eq 3.10 into Eq 3.2 gives

$$K_i \delta_i - S_i w_i - K_{i+1} \delta_{i+1} = -Q_i \quad (3.12)$$

Similarly, using Eqs 3.8 and 3.10 in Eq 3.4,

$$\begin{aligned} \frac{F_i \phi_i}{h} - \frac{F_{i-1} \phi_{i-1}}{h} - P_i (w_i - w_{i-1}) - K_i h \delta_i - \frac{R_i (\theta_J)_i}{2} \\ - \frac{T_i}{2} - \frac{R_{i-1} (\theta_J)_{i-1}}{2} - \frac{T_{i-1}}{2} = 0 \end{aligned} \quad (3.13)$$

Substitution of Eqs 3.6 and 3.7 into Eq 3.13,

$$\begin{aligned}
& \frac{F_i}{h^2} \left(w_{i-1} - 2w_i + w_{i+1} - \delta_i + \delta_{i+1} \right) \\
& - \frac{F_{i-1}}{h^2} \left(w_{i-2} - 2w_{i-1} + w_i - \delta_{i-1} + \delta_i \right) \\
& - P_i \left(w_i - w_{i-1} \right) - K_i h \delta_i - \frac{R_i}{4h} \left(w_{i+1} + \delta_{i+1} - w_{i-1} + \delta_i \right) \\
& - \frac{T_i}{2} - \frac{R_{i-1}}{4h} \left(w_i + \delta_i - w_{i-2} + \delta_{i-1} \right) - \frac{T_{i-1}}{2} = 0 \quad (3.14)
\end{aligned}$$

Or

$$\begin{aligned}
& \left(-\frac{F_{i-1}}{h^2} + \frac{R_{i-1}}{4h} \right) w_{i-2} + \left(\frac{F_{i-1}}{h^2} - \frac{R_{i-1}}{4h} \right) \delta_{i-1} + \left(\frac{F_i}{h^2} + \frac{2F_{i-1}}{h^2} \right. \\
& \left. + P_i + \frac{R_i}{4h} \right) w_{i-1} + \left(-\frac{F_i}{h^2} - \frac{F_{i-1}}{h^2} - K_i h - \frac{R_{i-1}}{4h} - \frac{R_i}{4h} \right) \delta_i \\
& + \left(-\frac{2F_i}{h^2} - \frac{F_{i-1}}{h^2} - P_i - \frac{R_{i-1}}{4h} \right) w_i + \left(\frac{F_i}{h^2} - \frac{R_i}{4h} \right) \delta_{i+1} \\
& + \left(\frac{F_i}{h^2} - \frac{R_i}{4h} \right) w_{i+1} = \frac{1}{2} \left(T_i + T_{i-1} \right) \quad (3.15)
\end{aligned}$$

Equations 3.12 and 3.15 are the two load-deflection equations which must be satisfied for each bar and joint of the idealized system. By also requiring that these equations be satisfied at the first imaginary bar and joint at each end of the model, the beam end conditions are accounted for automatically. Thus, for a beam which is represented by m discrete bars, a system of $2m + 5$ algebraic equations in $2m + 5$ unknown deflections results.

General Remarks

The bar-and-spring model which has been discussed is limited by the assumptions of elementary beam theory. The primary purpose of the model has been to study shear deformations. For this reason effects of torsion, axial deformation, biaxial bending, and other conditions have not been considered. These effects could be included in the analysis by means of the discrete-element model approach. In each case the properties of the real system are felt at discrete points.

The basic difference between the model used for pure bending, as discussed in Chapter 2, and the model for combined bending and shear is that the latter has additional degrees of freedom representing the shear deflections. This results in a more flexible system which is a better approximation for some structural elements.

CHAPTER 4. METHOD OF SOLUTION

Contents of the Chapter

In this chapter the load-deflection equations derived in Chapter 3 are rewritten in a more convenient form for computer solution. The form of the resulting equations written about each bar and joint of the system is discussed. A matrix formulation of these equations is introduced and the elimination procedure used to solve the equations is described. Boundary conditions are discussed and the accuracy of the solution is evaluated. Finally, the formulas used to solve for the slope, moment, shear, and support reaction are given.

Conversion of Equations to Standard Form

The two load-deflection equations can be rewritten in a more convenient form. Equation 3.12 may be expressed as

$$\begin{aligned} b_{i,1}\delta_i + b_{i,2}w_i + b_{i,3}\delta_{i+1} + b_{i,4}w_{i+1} + b_{i,5}\delta_{i+2} + b_{i,6}w_{i+2} \\ = b_{i,7} \end{aligned} \quad (4.1)$$

where the coefficients $b_{i,n}$ are given by

$$b_{i,1} = K_i$$

$$b_{i,2} = -S_i$$

$$b_{i,3} = -K_{i+1}$$

$$b_{i,4} = 0.0$$

$$b_{i,5} = 0.0$$

$$b_{i,6} = 0.0$$

$$b_{i,7} = -Q_i$$

In a similar manner, Eq 3.15 may be rewritten in the form

$$\begin{aligned} a_{i,1}w_{i-2} + a_{i,2}\delta_{i-1} + a_{i,3}w_{i-1} + a_{i,4}\delta_i + a_{i,5}w_i + a_{i,6}\delta_{i+1} \\ + a_{i,7}w_{i+1} = a_{i,8} \end{aligned} \quad (4.2)$$

where the coefficients $a_{i,n}$ are

$$a_{i,1} = -\frac{F_{i-1}}{h^2} + \frac{R_{i-1}}{4h}$$

$$a_{i,2} = \frac{F_{i-1}}{h^2} - \frac{R_{i-1}}{4h}$$

$$a_{i,3} = \frac{F_i}{h^2} + 2\frac{F_{i-1}}{h^2} + P_i + \frac{R_i}{4h}$$

$$a_{i,4} = -\frac{F_i}{h^2} - \frac{F_{i-1}}{h^2} - K_i h - \frac{R_{i-1}}{4h} - \frac{R_i}{4h}$$

$$a_{i,5} = -2\frac{F_i}{h^2} - \frac{F_{i-1}}{h^2} - P_i - \frac{R_{i-1}}{4h}$$

$$a_{i,6} = \frac{F_i}{h^2} - \frac{R_i}{4h}$$

$$a_{i,7} = \frac{F_i}{h} - \frac{R_i}{4h}$$

$$a_{i,8} = \frac{1}{2} (T_i + T_{i-1})$$

The first subscript of the $a_{i,n}$ and $b_{i,n}$ coefficients represents the beam station or bar at which the equation is to be satisfied, while the second subscript indicates the position of the coefficient in the equation. The $a_{i,n}$ and $b_{i,n}$ are referred to as continuity coefficients.

Pattern of the Equations

Equations 4.1 and 4.2, respectively, must be satisfied at each bar and joint of the idealized model. The matrix form of these equations is shown in Fig 4.1. The system of $2m + 5$ equations has been written in sequential order of the station numbers from Station -1 to Station $m+1$ in obtaining the matrix representation. Thus, the coefficients of the two load-deflection equations are alternated throughout the coefficient and load matrices. The coefficient matrix represents the stiffness of the system. All non-zero terms of this matrix are centered about the diagonal, indicated in Fig 4.2, in a seven-wide band.

The set of simultaneous equations is solved by use of a modified Gaussian elimination. In using this elimination procedure, the non-zero coefficients below the main diagonal of the coefficient matrix are eliminated, allowing the unknown deflections to be solved for by back substitution. An efficient and rapid solution is obtained by taking advantage of the large number of zeros in the square matrix. Most of the terms which are zero before the elimination are not even considered. The $b_{i,4}$ term must be considered, however, since the initial zero value of this term may be altered by the elimination procedure. Two other terms, $b_{i,5}$ and $b_{i,6}$, which are initially zero must also be included since it is sometimes necessary to interchange rows of the matrix. If a zero is encountered on the diagonal of the coefficient matrix, terms below the diagonal cannot be eliminated unless the zero can be replaced by a non-zero term. This is effected by a row

$$\begin{array}{c}
 \text{Coefficient} \\
 \text{Matrix} \\
 \left[\begin{array}{cccccccc}
 & b_{i-1,1} & b_{i-1,2} & b_{i-1,3} & b_{i-1,4} & b_{i-1,5} & b_{i-1,6} & \\
 a_{i,1} & a_{i,2} & a_{i,3} & a_{i,4} & a_{i,5} & a_{i,6} & a_{i,7} & \\
 & & b_{i,1} & b_{i,2} & b_{i,3} & b_{i,4} & b_{i,5} & b_{i,6} \\
 & a_{i+1,1} & a_{i+1,2} & a_{i+1,3} & a_{i+1,4} & a_{i+1,5} & a_{i+1,6} & a_{i+1,7} \\
 & & & b_{i+1,1} & b_{i+1,2} & b_{i+1,3} & b_{i+1,4} & b_{i+1,5} & b_{i+1,6} \\
 & a_{i+2,1} & a_{i+2,2} & a_{i+2,3} & a_{i+2,4} & a_{i+2,5} & a_{i+2,6} & a_{i+2,7} \\
 & & & & \cdot & \cdot & \cdot & \cdot & \cdot \\
 & & & & & \cdot & \cdot & \cdot & \cdot \\
 & & & & & & \cdot & \cdot & \cdot \\
 & & & & & & & \cdot & \cdot \\
 & & & & & & & & \cdot \\
 & & & & & & & & & b_{m-1,1} & b_{m-1,2} & b_{m-1,3} & b_{m-1,4} & b_{m-1,5} & b_{m-1,6} \\
 a_{m,1} & a_{m,2} & a_{m,3} & a_{m,4} & a_{m,5} & a_{m,6} & a_{m,7} & & & & & & & & & & \\
 & & & & & b_{m,1} & b_{m,2} & b_{m,3} & b_{m,4} & b_{m,5} & b_{m,6} & & & & & & \\
 & & & & & & a_{m+1,1} & a_{m+1,2} & a_{m+1,3} & a_{m+1,4} & a_{m+1,5} & a_{m+1,6} & a_{m+1,7} & & & & \\
 & & & & & & & & & b_{m+1,1} & b_{m+1,2} & b_{m+1,3} & b_{m+1,4} & b_{m+1,5} & b_{m+1,6} & & \\
 \end{array} \right]
 \end{array}
 \begin{array}{c}
 \text{Deflection} \\
 \text{Matrix} \\
 \left[\begin{array}{c}
 W_{i-2} \\
 \delta_{i-1} \\
 W_{i-1} \\
 \delta_i \\
 W_i \\
 \delta_{i+1} \\
 W_{i+2} \\
 \delta_{i+2} \\
 \cdot \\
 \cdot \\
 \cdot \\
 W_{m-1} \\
 \delta_m \\
 W_m \\
 \delta_{m+1} \\
 W_{m+1} \\
 \delta_{m+2} \\
 W_{m+2}
 \end{array} \right]
 \end{array}
 =
 \begin{array}{c}
 \text{Load} \\
 \text{Matrix} \\
 \left[\begin{array}{c}
 b_{i-1,7} \\
 a_{i,8} \\
 b_{i,7} \\
 a_{i+1,8} \\
 b_{i+1,7} \\
 a_{i+2,8} \\
 \cdot \\
 \cdot \\
 \cdot \\
 b_{m-1,7} \\
 a_{m,8} \\
 b_{m,7} \\
 a_{m+1,8} \\
 b_{m+1,7}
 \end{array} \right]
 \end{array}$$

Fig 4.1. Matrix representation of the equations of the model.

Coefficient Matrix											Deflection Matrix	Load Matrix				
0	0	0	$b_{-1,3}$	0	0	0						W_{-2}				
0	0	$a_{0,3}$	$a_{0,4}$	$a_{0,5}$	$a_{0,6}$	$a_{0,7}$						δ_{-1}	$b_{-1,7}$			
			$b_{0,1}$	$b_{0,2}$	$b_{0,3}$	0	0	0				W_0	$a_{0,8}$			
		$a_{1,1}$	$a_{1,2}$	$a_{1,3}$	$a_{1,4}$	$a_{1,5}$	$a_{1,6}$	$a_{1,7}$				δ_1	$a_{1,8}$			
					$b_{1,1}$	$b_{1,2}$	$b_{1,3}$	0	0	0		W_1	$b_{1,7}$			
			$a_{2,1}$	$a_{2,2}$	$a_{2,3}$	$a_{2,4}$	$a_{2,5}$	$a_{2,6}$	$a_{2,7}$			δ_2	$a_{2,8}$			
												.	.			
												.	.			
												.	.			
												W_{m-1}	$b_{m-1,7}$			
						$b_{m-1,1}$	$b_{m-1,2}$	$b_{m-1,3}$	0	0	0	δ_m	$a_{m,8}$			
			$a_{m,1}$	$a_{m,2}$	$a_{m,3}$	$a_{m,4}$	$a_{m,5}$	$a_{m,6}$	$a_{m,7}$			W_m	$b_{m,7}$			
								$b_{m,1}$	$b_{m,2}$	$b_{m,3}$	$b_{m,4}$	0	0			
										$a_{m+1,1}$	$a_{m+1,2}$	$a_{m+1,3}$	$a_{m+1,4}$	$a_{m+1,5}$	0	0
													W_{m+1}	$a_{m+1,8}$		
													δ_{m+1}	$b_{m+1,7}$		
													W_{m+2}			
													δ_{m+2}			
													W_{m+2}			

Fig 4.2. Illustration of the initial zero terms in the equations of the model.

interchange. The first two rows of the matrix must always be interchanged since the $b_{-1,2}$ term is zero.

Two back substitution formulas are required since there are two sets of coefficients used in the matrix. The total deflection and shear deflection for each station are determined by substitution into the equations

$$w_i = \frac{b_{i,7} - b_{i,6}w_{i+2} - b_{i,5}\delta_{i+2} - b_{i,4}w_{i+1} - b_{i,3}\delta_{i+1}}{b_{i,2}} \quad (4.3)$$

$$\delta_i = \frac{a_{i,8} - a_{i,7}w_{i+1} - a_{i,6}\delta_{i+1} - a_{i,5}w_i}{a_{i,4}} \quad (4.4)$$

beginning at the lower end of the matrix.

Boundary Equations and Specified Conditions

A very general description of support points is possible by use of several conditions which can be freely enforced at any desired station along the structure. In addition some boundary conditions are automatically created at the ends of the structure.

Two boundary conditions are implied by requiring that the load-deflection equations be satisfied at the imaginary stations of the beam. By writing Eq 4.1 at Station -1 and Eq 4.2 at Bar 0, it is observed that some of the continuity coefficients are always zero. These zero terms, some of which are outside the coefficient matrix, are automatically set by the physical properties of the system beyond the ends of the structure. The first of these equations indicates that the shear deformation δ_0 in Bar 0 is always zero. The second equation physically signifies that there is zero curvature at the ends of the beam. Thus, the condition of zero moment is automatically created by this equation.

The deflection of any station may be specified by proper adjustment of coefficients of Eq 4.1. Thus, by setting $b_{i,2}$ equal to 1.0, $b_{i,7}$ equal to the desired deflection, and all other $b_{i,n}$ coefficients equal to zero, a specified deflection is enforced at Station i . A zero deflection can also be prescribed at a station by placing a very stiff support spring at the joint.

Other boundary conditions can be enforced by proper adjustment of the values of the physical properties of the system. For example, the fixity or resistance to rotation of a member may be controlled at any station by specifying a rotational restraint. A rotational restraint spring acts as a couple which is equal to the product of the spring constant and the average slope at the joint. Zero slope can be enforced by a sufficiently large spring constant. The zero curvature condition at the ends of the beam is automatically overridden by the specification of a rotational restraint at these points.

A hinge can be created at any station by a zero flexural stiffness at this point. This condition is not applicable to the ends of the beam in creating a pin-ended condition, however. The automatically created conditions of zero curvature and the specification of a deflection or spring support are sufficient to insure the required end conditions for a pinned end.

Shear deformations can be prevented by specifying a large shear spring stiffness. This provides a means of obtaining the pure bending solution for a problem.

It is also possible to obtain several independent solutions from a single problem should this ever be desirable. A series of beams can be solved simultaneously by requiring that the beams be separated by two or more stations with zero flexural stiffness. If more than two consecutive stations have zero flexural stiffness, however, the resulting system of equations will not be independent. By placing a support spring at stations located two or more increments from all boundaries, no dependency will be introduced and in addition the solution will not be affected.

Results

Other results can be computed from the known deflections. Substitution of the shear deflections and the total deflections into the equations

$$\theta_i = \frac{w_i - w_{i-1} + \delta_i}{h} \quad (3.5)$$

$$V_i = K_i \delta_i \quad (3.10)$$

and

$$M_i = \frac{F_i}{2} (w_{i-1} - 2w_i + w_{i+1} - \delta_i + \delta_{i+1}) \quad (3.7) \text{ and } (3.8)$$

determines the value of slope, shear, and bending moment, respectively, for each station. Also the support reaction can be computed from

$$(Q_R)_i = -V_i + V_{i+1} - Q_i \quad (4.5)$$

at support points where the deflection is specified. The above equation results from the summation of vertical forces at a joint which is not free to deflect vertically. For all other joints, the support reaction is determined by

$$(Q_R)_i = S_i w_i \quad (4.6)$$

Accuracy of the Solution

Approximation errors are introduced when the discrete-element model is substituted for the real structure. Within the limits of the assumptions of the model, this type of error can be reduced by dividing the model into more increments. The number of increments required to obtain a specified degree of accuracy depends upon the complexity of the particular problem. An excessive number of increments should be avoided because computation time increases in simple proportion to the number of increments.

Because of the large number of arithmetic operations involved in the solutions, round-off errors may occur. A CDC 6600 computer using approximately 14 decimal digits has been used to verify the method of solution and no significant round-off errors have been observed in the practical problems that have been solved. Errors have been found to occur when an unreasonably large value of the rotational restraint is specified. In general the

magnitude of this quantity should not exceed 10^4 times the bending stiffness of the member at that station.

To verify that this value of rotational restraint is sufficient to prevent joint rotation for a particular problem, the solution should be checked to assure that the bar slopes on either side of the joint are equal in magnitude and opposite in direction.

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CHAPTER 5. THE COMPUTER PROGRAM

General

Program SHRBM 1 (SHeaR BeaM - 1st version) has been developed for an efficient solution of the equations derived in Chapter 3. The computer program is written in FORTRAN 63 language for the Control Data Corporation 6600 Computer. With minor changes, the program would be compatible with other computers. Compile time for this program on the 6600 Computer is approximately 8 seconds. The program storage requirement is 22,954 decimal words. Two subroutines, INTERP4 and TIC TOC (J), are included in the program.

A summary flow diagram for SHRBM 1 is given in Appendix 2. A more detailed flow diagram of the main program and subroutine INTERP4 are also included in Appendix 2. Appendix 3 contains a glossary of the notation used in program SHRBM 1, and is followed by a complete listing of the program in Appendix 4. Several comment cards have been inserted in the program to facilitate the understanding of the program logic.

To describe a physical problem for a computer solution using SHRBM 1, it is first necessary to divide the member into a number of equal increments and to designate points between the increments by station numbers. The program is currently dimensioned for a maximum of 200 increments. The left end of the structure must correspond to Station 0. The bar numbering convention adopted in Chapter 3 and used in program SHRBM 1 identifies each bar in the model with the same number as the station to its right.

All loads and stiffness properties of an actual system are defined at discrete points of the beam model. As discussed in Chapter 3, some properties of the physical system are associated with the joints while others are associated with the bars. Distributed quantities which are "lumped" at the joints are representative of one-half an increment of the physical system on each side of the joint. Joint quantities should, therefore, receive half values at the end stations of a distribution sequence. Terms which are concentrated in the bars, however, should be distributed as full values throughout the

model. Subroutine INTERP4 was written by Taylor (Ref 16) and is used in program SHRBM 1 for the distribution of input data. For given values at the initial and final stations in a distribution sequence, INTERP4 performs a linear interpolation between these extreme values and stores the appropriate value at each station, including half values at the end station. Concentrated loads that occur between stations should be divided between the two adjacent stations such that the amount of load placed at each station is inversely proportional to the distance between the station and the point of application of the load.

Any system of units may be used to describe the problem (for example, pounds and inches), but the system must be used consistently.

A plot routine, FORTRAN 63 Plot Routine, is used in SHRBM 1 to plot all results computed by the main program. This routine was written for The University of Texas Computation Center and instructions for its use are given in the Plot Routine Operation Manual (Ref 6). The required plot parameters are automatically determined by program SHRBM 1.

The time required to run problems depends on the desired results and size of the particular system. Problems with a greater number of increments and which exercise the plot option require more time. Subroutine TIC TOC (J) provides a check on the time requirement by automatically printing the run time for each problem.

Procedure for Data Input

A guide for data input is provided in Appendix 1. The guide is a self-contained instruction manual which is designed such that additional copies may be furnished as separately bound extracts for routine use.

Included in the guide is an illustration of the proper deck sequence. Any number of problems may be stacked and run together. The problem sequence is preceded by two cards which describe the run. The first card of each problem is provided for the problem number and a brief description of the problem. A blank card at the end of each run terminates the program.

Tables of Data Input

All data input is arranged in tabular form for convenience and clarity. Table 1 contains the control data. It consists of a single card which must

be input in each problem. The data-hold options, the number of cards in the remaining tables, and the plot options are specified in this table. The data-hold options allow the user to retain data from any of the tables of the preceding problem. If the data from Table 2 or 3 are held, no cards may be added to these tables. Data in Tables 4A, 4B, and 5 may be modified by the addition of new cards, but the total number of cards accumulated in each of these tables must not exceed 100.

One of the following three plot options may be specified in Table 1:

- (1) Plot on 8-1/2 × 11-inch paper with 1-inch and 5-inch axes.
- (2) Plot on 8-1/2 × 11-inch paper with 2-inch and 10-inch axes.
- (3) Plot on 11 × 17-inch paper with 1-inch and 15-inch axes.

If one of these options is specified, scales for six sets of axes are automatically determined for plotting the total deflection, shear deflection, slope, moment, shear, and support reaction vs distance along the beam. A blank or zero plot option indicates no plots are desired.

In Table 2 one card is used to specify the number of increments and the increment length. The number of increments includes only the number of bars which are used to represent the actual structure. The imaginary bars are created internally by SHRBM 1. A maximum of 200 increments is allowed.

Table 3 provides for vertical deflections to be specified at any station along the beam. Not more than 20 deflections may be specified. Each specification requires a separate card. The cards in this table may be stacked in any order.

The beam section and material properties are described in Table 4A. Any variation of the modulus of elasticity, the shear modulus, the moment of inertia, and the effective shear area may be specified in this table. These properties are used to compute the shear and flexural stiffnesses of the beam. The method used for description of distributed data is illustrated in Appendix 1. Half-values of the modulus of elasticity are automatically created at the end of each distribution sequence. Half-values are not created for the moment of inertia since the beam flexural stiffness (EI) would then only be a quarter-value. Likewise, the modulus of shear and the effective shear area are bar properties and must, therefore, receive full-values. All of the values in Table 4A are accumulated algebraically in storage. There are no restrictions on the order of the cards, except that within a distribution sequence the stations must be in ascending order.

Additional stiffness data may be specified in Table 4B. The data from this table are automatically added to the stiffnesses computed in Table 4A. Table 4B provides a means for specifying more general stiffness values. The data-input rules of Table 4A apply to Table 4B.

Table 5 is used to describe transverse forces, spring supports, applied couples, rotational restraints, and axial tensions or compressions. Applied couples and rotational restraints are concentrated effects. The transverse forces and spring supports are distributed with half-values. Axial tensions or compressions receive full values. The rules of Table 4A for data input apply to Table 5.

Special Programming Features

Certain programming features concerning data input should be mentioned at this point. For example, zero data are automatically created for points where no data are specified in Tables 4A, 4B, and 5. Also, an arbitrarily large value of shear stiffness is assigned to points where zero shear stiffnesses are computed. Furthermore, provision has been made to automatically insert the necessary spring supports in order to solve a discontinuous series of beams as discussed in Chapter 4.

Output

All data input in Tables 1 through 5 is reflected in tabular form in the output. In addition, the computed values of total deflection, moment, and support reaction are printed with the corresponding beam station in Table 6. The values of shear deflection, slope, and shear are printed between the quantities computed for the stations. If one of the plot options is specified, the values used at the ends of the axes are printed in Table 7.

Error Messages

Several checks on data errors are provided in SHRBM 1. An error message is printed if any of the following conditions occurs:

- (1) More than one deflection is specified at a station.
- (2) No data are specified for a table (excluding Tables 3 and 4A).
- (3) The hold option is exercised without having input data for the previous problem.

- (4) Data are erroneously held in Table 2 or 3.
- (5) The allowable number of cards for an input table is exceeded.
- (6) The station numbers in a distribution sequence are out of order.

In addition to these specific error messages, a general purpose error message - "UNSPECIFIED ERROR STOP" - is provided. Should this error message occur, the program must be investigated to determine its source since this message may be caused by one of several different errors. Any error detected by the program will cause the run to be terminated. No check is provided for the specification of data beyond the ends of the beam. These data will interfere with the automatically created boundary conditions and cause the solution to be in error.

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CHAPTER 6. SAMPLE PROBLEMS

General

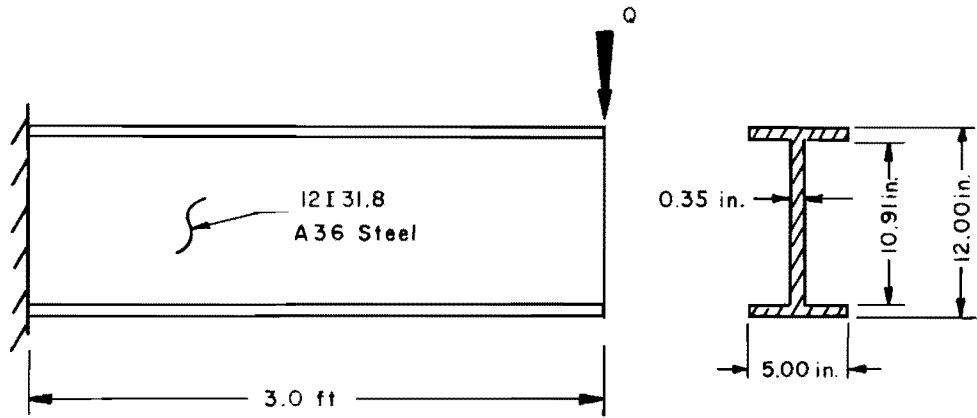
This chapter contains several example problems which have been solved to verify the method of analysis and to demonstrate the use of program SHRBM 1. It is primarily intended that these sample problems be helpful to the reader in modeling the actual system and in coding problems for the computer solution.

Several different types of problems have been selected. The first problem, Problem Series 100, is a determinate structural problem which has been solved to compare the computer results with closed-form solutions. In Problem Series 200, the solution of a highway bridge diaphragm subjected to three different load conditions is examined. Problem Series 300 illustrates the effect of shear deformations on a three-span overhanging beam. Each of these problems has been solved in two parts. In Part A the actual shear stiffness of the member was modeled while in Part B a very large shear stiffness was used to determine the effect of neglecting shear deformations. A listing of the data input and sample codings for each problem are given in Appendix 5. Appendix 6 contains the computer output including plots for these problems.

Problem Series 100 - Comparison of the Method with Closed-Form Solutions

Two sample problems have been solved in this section using the discrete-element method of analysis to compare the results with closed-form solutions. The member analyzed in the first problem of this series, Problem 101, is illustrated in Fig 6.1(a). It is a 3-foot cantilevered 12I31.8 steel beam. The material and section properties of this member are tabulated in Fig 6.1. The beam is subjected to a 22,000-pound concentrated load and a uniform beam dead weight of 2.65 lbs/in.

The idealized model of the structure is illustrated in Fig 6.1(b). The beam is represented by 36 1-inch increments. The fixed end support is



(a) Cantilevered prismatic beam.

Section Properties

$I = 215.8 \text{ in}^4$

$A_s = 3.82 \text{ in}^2$

Material Properties

$E = 30 \times 10^6 \text{ lb/in}^2$

$G = 11 \times 10^6 \text{ lb/in}^2$

Load

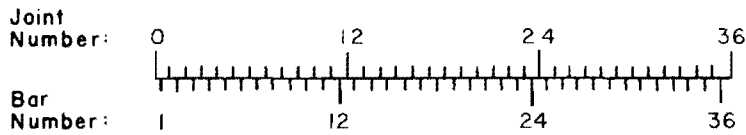
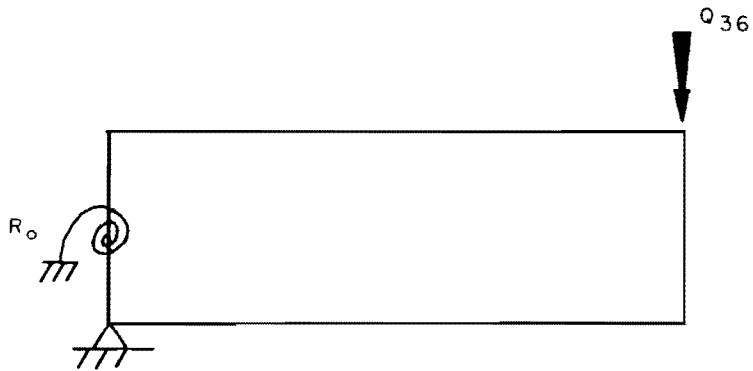
$q = -2.65 \text{ lb/in}$

$Q = 22,000 \text{ lb}$

Other Data

$R_o = 1 \times 10^{15} \text{ in-lb/rad}$

$h = 1.0 \text{ in}$



(b) Bar-and-spring model.

Fig 6.1. Beam solved in Problem 101.

modeled with a zero specified deflection and a rotational restraint of 1×10^{15} in-lb/rad to enforce a zero slope. It should be noted that the limits of the data represent bar numbers for bar quantities and joint numbers for joint quantities.

A comparison of the results computed by SHRBM 1 with the closed-form results for this problem are shown in Table 1(a). The beam deflection at Station 36 and the bending moment at Station 0 are selected as a basis for comparison. Shear deformations account for approximately 26 percent of the total deflection for this problem. The bending moment at Station 0 given by the computer program is approximately one-half the theoretical value since this station has a half-value of stiffness.

In the second problem of this series, Problem 102, the 3-foot tapered cantilevered beam shown in Fig 6.2(a) is analyzed. The member section properties, expressed as a function of distance along the beam, are tabulated in Fig 6.2. The properties at selected points along the structure were used to model the actual system. The idealized representation of the beam is illustrated in Fig 6.2(b). The only load considered in this problem is the 7,000-pound concentrated load at the end of the beam.

The closed-form solutions of the beam deflections at Station 36 and the bending moment at Station 0 are compared with the computer solutions for this problem. These results are shown in Table 1(b). A value of 1.2 was used for the k factor in both solutions. Also the beam taper was included in the closed-form solution. The computer solutions indicate close agreement with the closed-form solutions.

Problem Series 200 - A Bridge Diaphragm

The interior diaphragm of a highway bridge illustrated in Fig 6.3(a) is analyzed in this problem series. A 15I33.9 section has been selected for the diaphragms and 30 WF 99 sections are used for the supporting girders. Section and material properties for these members are given in Fig 6.3. To analyze the diaphragm using Program SHRBM 1, it is necessary to model the system as shown in Fig 6.3(b). The bridge deck slab has not been considered with this model. The effect of each girder is represented by two springs. The stiffness of the rotational restraint spring is determined from the

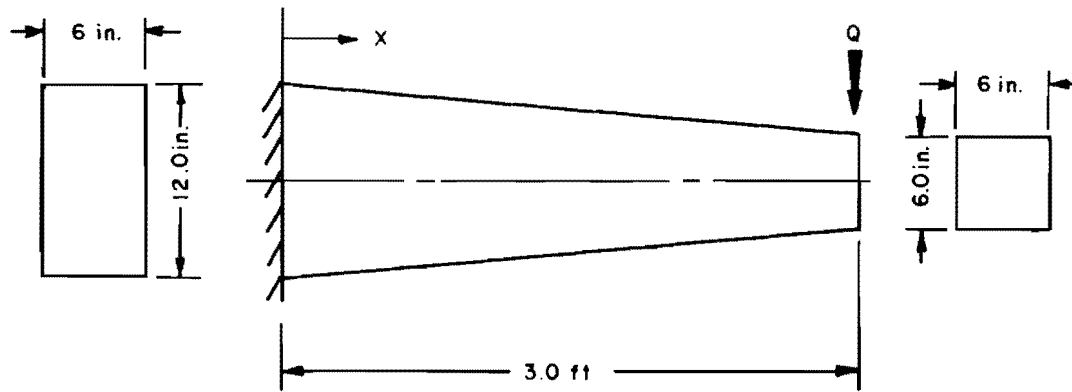
TABLE 1. COMPARISON OF THE DISCRETE-ELEMENT METHOD WITH THEORY

	<u>SHRBM 1</u>	<u>CLOSED-FORM SOLUTION</u>
Deflection at Station 36 due to bending	-5.296×10^{-2} in	-5.294×10^{-2} in
Deflection at Station 36 due to shear	-1.888×10^{-2} in	-1.889×10^{-2} in
Total deflection at Station 36	-7.184×10^{-2} in	-7.183×10^{-2} in
Bending moment at Station 0	-3.969×10^5 in-lb	-7.937×10^5 in-lb

(a) Problem 101 - Cantilevered I-beam

	<u>SHRBM 1</u>	<u>CLOSED-FORM SOLUTION</u>
Deflection at Station 36 due to bending	-6.834×10^{-2} in	-6.870×10^{-2} in
Deflection at Station 36 due to shear	-0.459×10^{-2} in	-0.492×10^{-2} in
Total deflection at Station 36	-7.293×10^{-2} in	-7.362×10^{-2} in
Bending moment at Station 0	-1.260×10^5 in-lb	-2.520×10^5 in-lb

(b) Problem 102 - Cantilevered tapered beam



(a) Symmetrically tapered cantilever beam.

Section Properties

$$A_s = (72 - X) \text{ in}^2$$

$$I = 1/2 (12 - X/6)^3 \text{ in}^4$$

$$k = 1.2$$

Material Properties

$$E = 3 \times 10^6 \text{ lb/in}^2$$

$$G = 1.27 \times 10^6 \text{ lb/in}^2$$

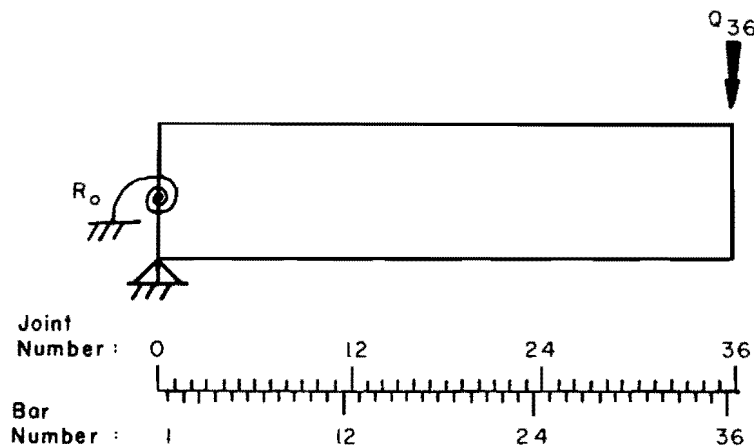
Loads

$$Q = 7,000 \text{ lb}$$

Other Data

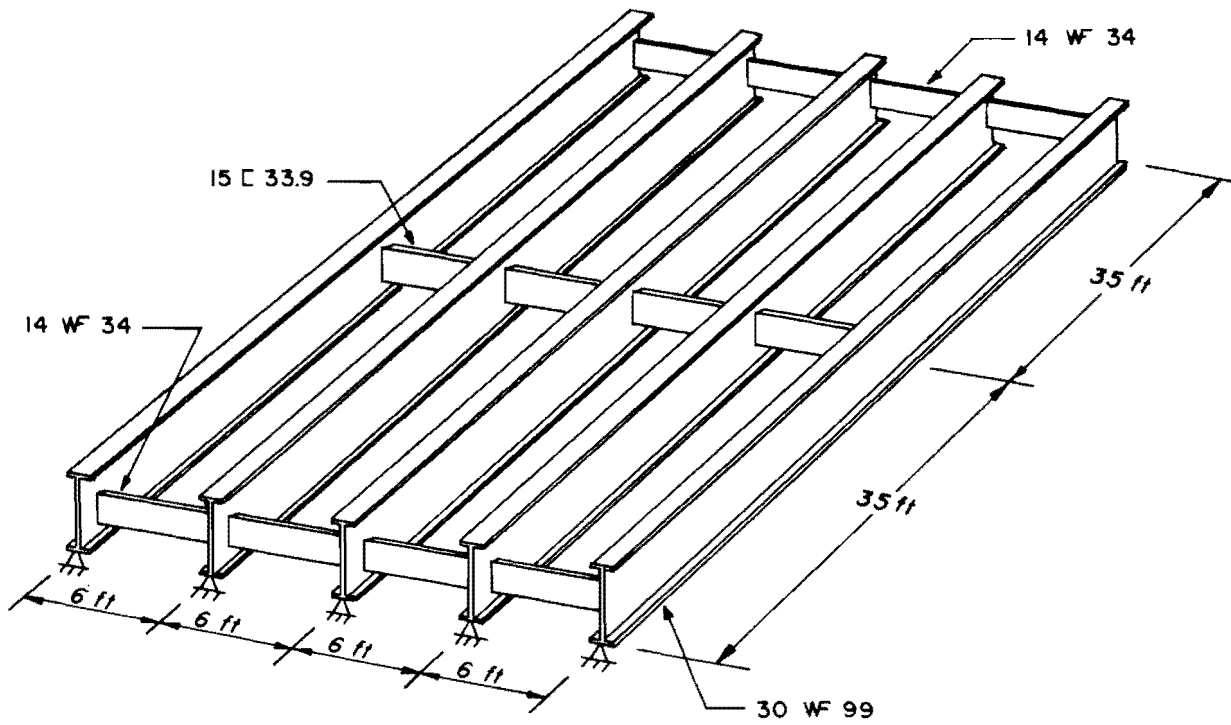
$$R_o = 1 \times 10^{13} \text{ in-lb/rad}$$

$$h = 1.0 \text{ in.}$$

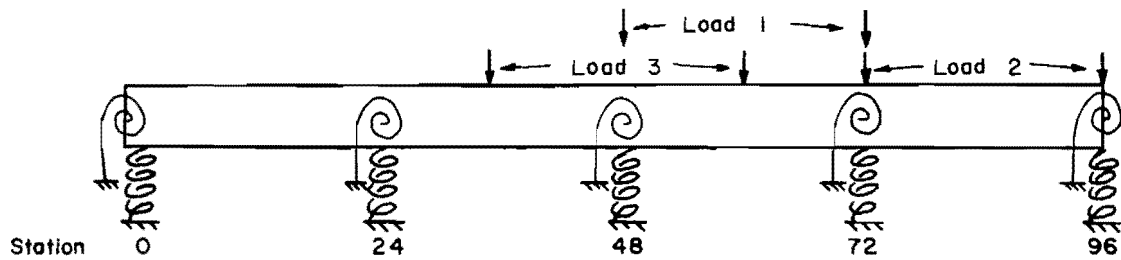


(b) Bar-and-spring model.

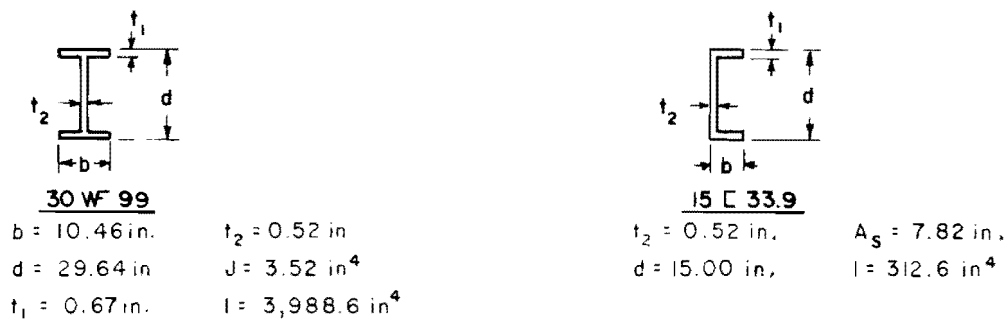
Fig 6.2. Beam solved in Problem 102.



(a) Highway bridge beam grid system.



(b) Model of interior bridge diaphragm.



(c) Section properties.

Fig 6.3. Structure analyzed in Problem Series 200.

torsional resistance of the girder assuming fixity at each end. Using the torsional constant of the girder computed by

$$J = \frac{2bt_1^3 + dt_2^3}{3} = \frac{(2)(10.46 \text{ in})(0.67 \text{ in})^3 + (29.64 \text{ in})(0.52 \text{ in})^3}{3}$$

$$= 3.52 \text{ in}^4$$

the rotational restraint is determined to be

$$R = \frac{4GJ}{L} = \frac{(4)(11.0 \times 10^6 \text{ lb/in}^2)(3.52 \text{ in}^4)}{(70 \text{ ft})(12 \text{ in/ft})}$$

$$= 1.84 \times 10^5 \text{ in-lb/rad}$$

Determination of the support spring stiffness which replaces each of the girders is based on the deflection of a simply supported beam with a concentrated load at the center. Thus,

$$S = \frac{48EI}{L^3} = \frac{(48)(30 \times 10^6 \text{ lb/in}^2)(3988.6 \text{ in}^4)}{(70 \text{ ft})^3(12 \text{ in/ft})^3}$$

$$= 9.7 \times 10^5 \text{ lb/in}$$

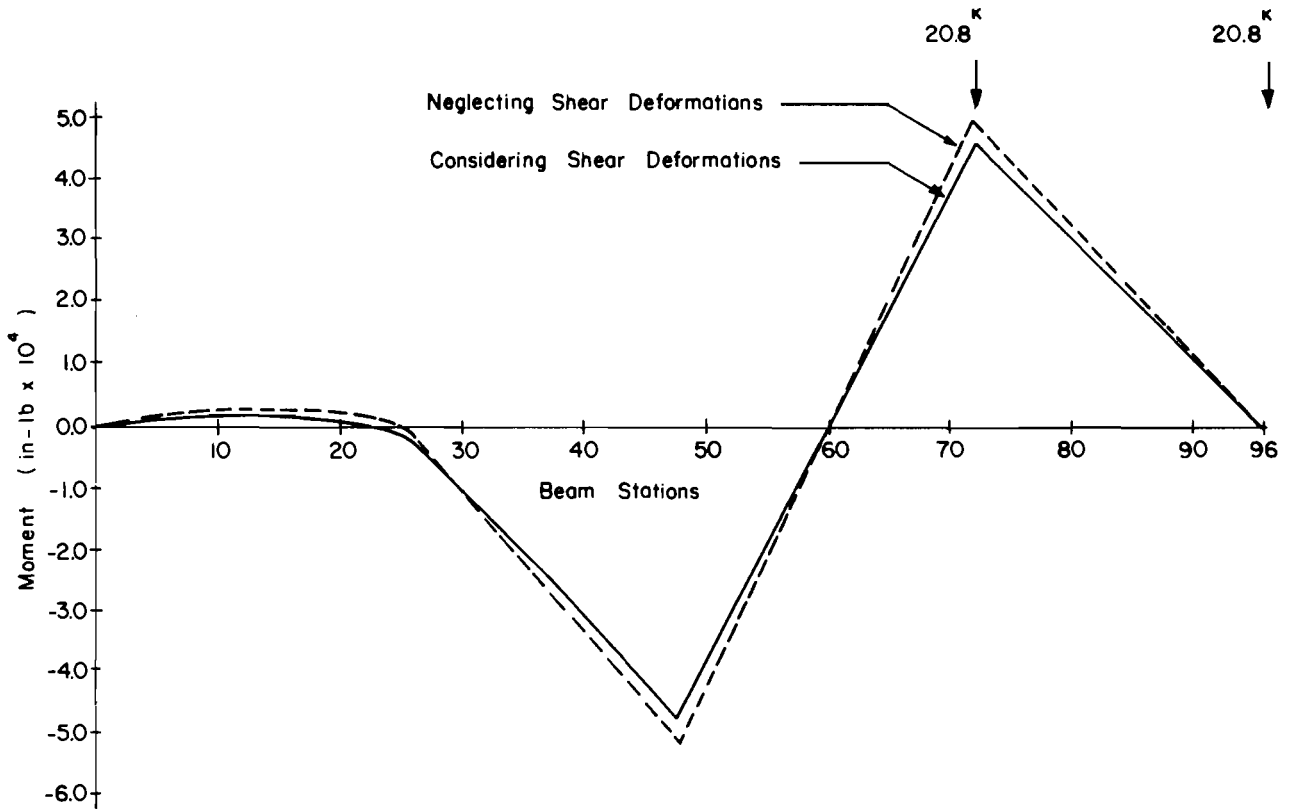
The uniform dead weight of the diaphragm of

$$q = (33.9 \text{ lb/ft})(1 \text{ ft}/12 \text{ in})(3 \text{ in/sta}) = 8.48 \text{ lb/sta}$$

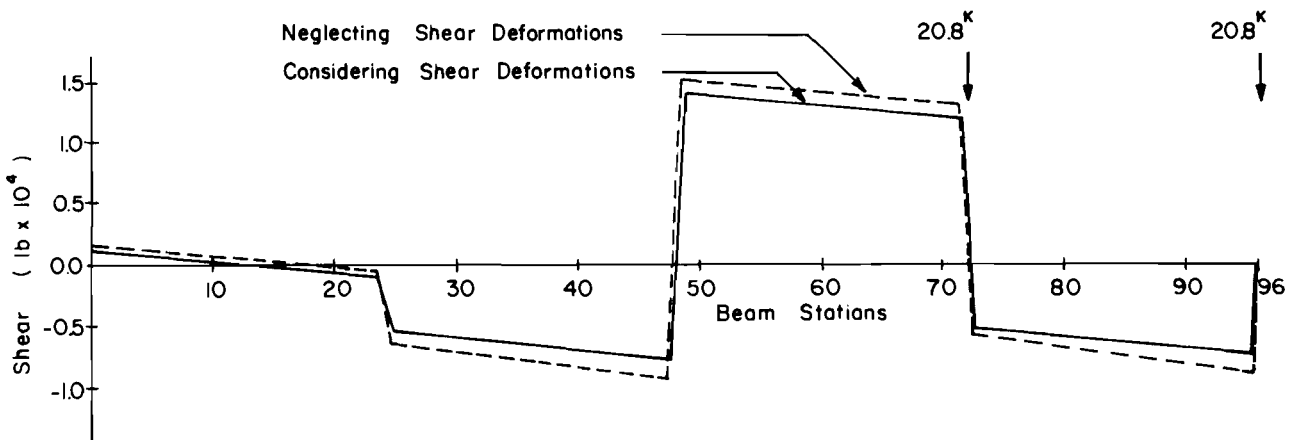
is included in the analysis.

The member is also subjected to the AASHO standard HS 20 truck wheel design load of 20.8 kips, including a 30 percent impact factor. A solution is given for each of the three load positions indicated in Fig 6.3(b).

The computer results using SHRBM 1 indicate that the shear deformations for this problem are relatively small compared with the deflections due to



(a) Moment diagram.



(b) Shear diagram.

Fig 6.4. Effect of shear deformations in a bridge diaphragm.

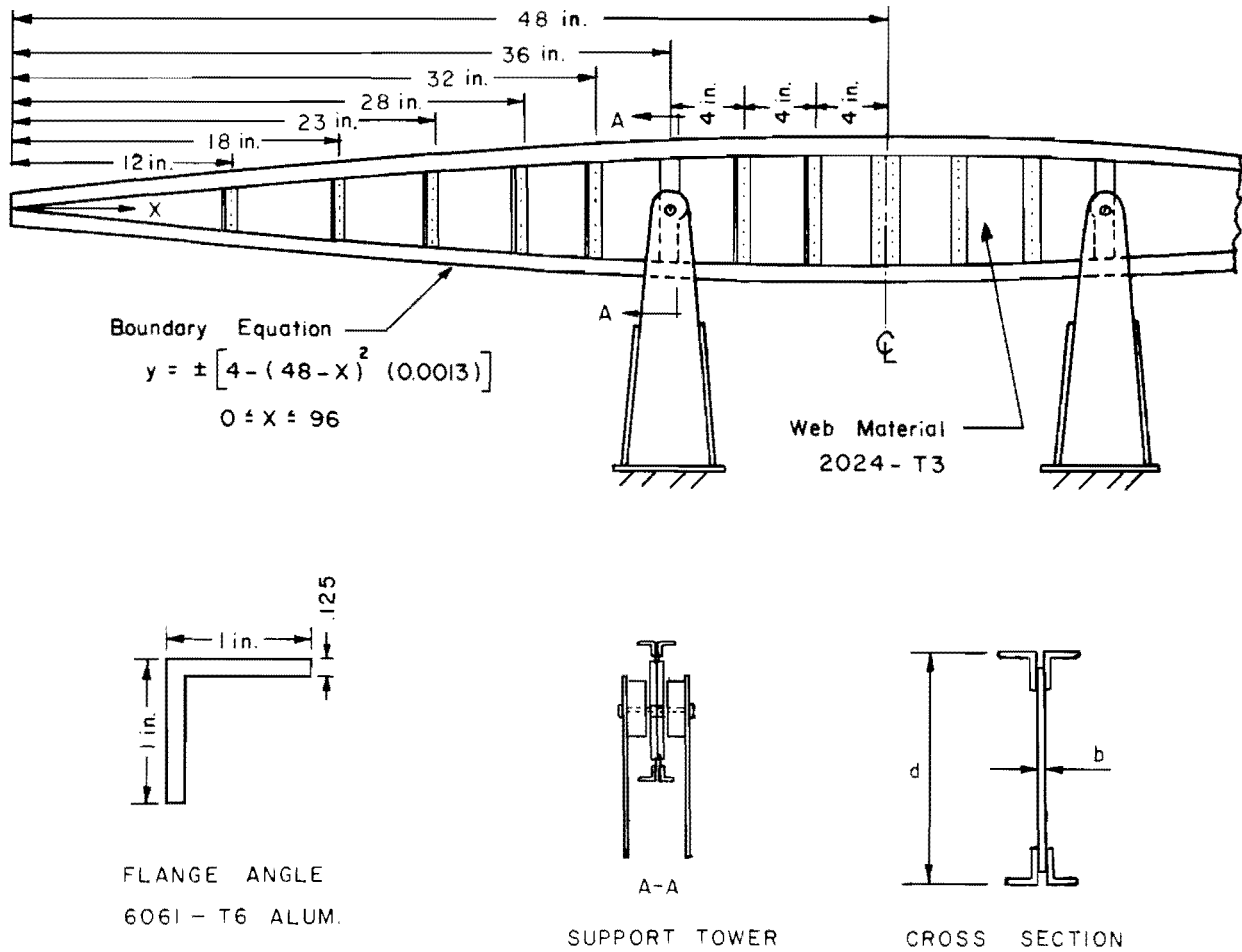
the combined effect of bending and shear. Significant effects in the moments and shears in the structure are apparent, however. Moment and shear diagrams for the diaphragm with the wheel loads at Stations 72 and 96 are shown in Figs 6.4(a) and (b). The solid-line curves indicate the solutions using the actual shear stiffness of the structure and the dashed-line curves indicate the results considering only bending effects. Both the moment and the shear in the diaphragm are reduced when shear deformations are considered. For this problem, the maximum negative and positive moments are decreased by 7.7 percent, and the maximum negative and positive shears are decreased 9.0 percent and 7.2 percent, respectively.

Problem Series 300 - A Three-Span Overhanging Beam

The purpose of this problem series is to verify the method of solution by comparison with the results of an experimental investigation conducted by Duncan (Ref 4). The beam used in this study was a built-up aluminum section as shown in Fig 6.5(a). The flange angle material used was 6061-T6 aluminum and the web was constructed from 2024-T3 aluminum. The beam is symmetrical about the vertical and horizontal centerlines, and has a second degree taper. Stiffeners were placed only on the left and center spans of the beam. In the tests performed by Duncan, an incremental loading was applied through a screw jack. Dial indicators were placed at selected stations along the beam and initial readings were recorded before the concentrated load was applied. The measured deflections do not include the effect of the dead weight of the structure.

The member is modeled by 96 1-inch increments with zero deflection specified at the supports. Section properties were computed with the parabolic boundary equation for selected points along the beam. The stiffeners were not modeled for the computer solution. A modulus of elasticity of 10.0×10^6 lb/in² was used for the 6061-T6 aluminum. A shear modulus of 4.0×10^6 lb/in² was used for the web material. Required section and material properties are tabulated in Fig 6.6(b).

The deflections measured by Duncan for 300, 500, and 700-pound loads at Station 0 are shown in Table 2. Computer solutions using SHREB 1 were obtained for these same load conditions. A comparison of these results is shown in Fig 6.6. The experimental values are indicated by the "+" symbols. The



(a) Three-span overhanging beam with a concentrated load at the left end.

$$b = 0.032 \text{ in.}$$

$$d = 2 [4 - (48 - x)^2 (0.0013)] \text{ in } 0 \leq x \leq 96$$

$$I = \frac{bd^3}{12} + 2 \left[\frac{(2)(0.125)^3}{12} + \left(\frac{d - 0.125}{2} \right)^2 (2)(0.125) \right. \\ \left. + \frac{(0.250)(0.875)^3}{12} + (0.250)(0.875) \left(\frac{d}{2} - 0.125 - \frac{0.875}{2} \right)^2 \right] \text{ in}^4$$

$$A_s = bd \text{ in}^2$$

$$E = 1 \times 10^7 \text{ lb/in}^2$$

$$G = 4 \times 10^6 \text{ lb/in}^2$$

(b) Section and material properties.

Fig 6.5. Beam solved in Problem Series 300.

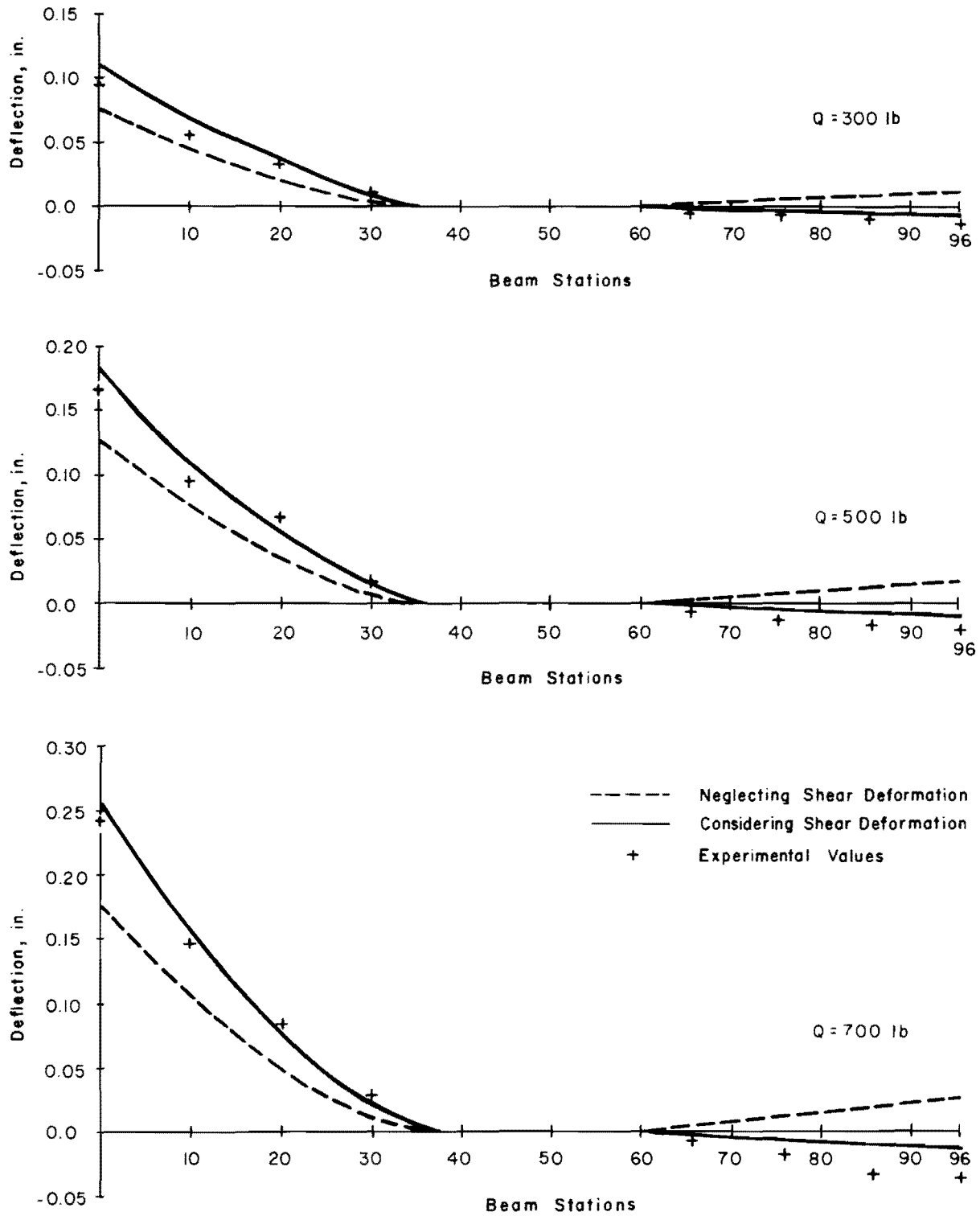


Fig 6.6. Deflection curves for a concentrated load Q at the left end of the beam.

solid-line curves represent the computed values of deflection considering shear deformations, and the dashed-line curves represent the deflection neglecting shear deformations. In most cases the computed values of deflection due to combined bending and shear are very nearly equal to the measured deflections. The maximum difference in calculated and measured deflections for the left span is 0.0152 inch or approximately 8 percent at Station 0. Some error in the analysis is due to the fact that the effect of moment on the shear deformations in a tapered beam has not been considered (Ref 18). Significant differences in the computed and the measured values of deflection occur in the right span.

It is important to note the effect of shear deformations on the beam deflections. The negative deflections in the right span of the structure were not predicted by the bending solution.

TABLE 2. MEASURED VALUES OF DEFLECTION OF A
THREE-SPAN OVERHANGING BEAM

<u>Station</u>	<u>LOAD AT STATION 0</u>		
	<u>300 lb</u>	<u>500 lb</u>	<u>700 lb</u>
0	0.0982 in.	0.1685 in.	0.2431 in.
10	0.0547	0.0966	0.1460
20	0.0326	0.0679	0.0821
30	0.0114	0.0171	0.0288
66	-0.0032	-0.0056	-0.0089
76	-0.0060	-0.0112	-0.0174
86	-0.0099	-0.0191	-0.0312
96	-0.0104	-0.0200	-0.0348

CHAPTER 7. SUMMARY AND CONCLUSIONS

A method of analyzing a structure for the combined effects of bending and shear deformations has been presented. The method provides for a more accurate solution of problems in which shear deflections are significant. The principal features of the method are

- (1) A finite-element model is used to simulate the real structure.
- (2) The load-deflection behavior of the system is described by two equations which are written for each station in the structure.
- (3) A special version of Gaussian elimination is used for an efficient solution of the system of equations.

A computer program, SHRBM 1, has been written which utilizes the method of analysis. The accuracy of the method and of the program have been demonstrated by comparison of the solutions for a range of problems with experimental data or results obtained by other accepted analytical procedures.

The analysis of highway bridge grid floor systems has been investigated by Ingram (Ref 9). The effects of shear deformations were not included in that analysis. By a combination of that work and the results of the study included in this report, a method of analysis could be developed for structural elements in which the effects of shear deformations are important. In particular a more accurate determination of the benefits of diaphragms in bridge floor systems may be made using such a combined analytical procedure.

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APPENDIX 1

GUIDE FOR DATA INPUT FOR SHRBM 1

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GUIDE FOR DATA INPUT FOR SHRBM 1
With Supplementary Notes

extract from

A DISCRETE-ELEMENT METHOD OF ANALYSIS FOR COMBINED BENDING
AND SHEAR DEFORMATIONS OF A BEAM

Report No. 56-12

by

David F. Tankersley and William P. Dawkins

December 1969

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SHRBM 1 GUIDE FOR DATA INPUT - Card Forms

IDENTIFICATION OF PROGRAM AND RUN (2 alphanumeric cards per run)

1	Account number, project, date, etc.	60
1	Description of run	60

IDENTIFICATION OF PROBLEM (one card each problem; program stops if PROB NUM is blank)

1	5	11	Description of problem (alphanumeric)	60
---	---	----	-----------------------------------------	----

TABLE 1. PROGRAM CONTROL DATA (one card each problem)

ENTER "1" TO HOLD PRIOR TABLE					NUM CARDS ADDED FOR TABLE					ENTER 1,2, OR 3 TO PLOT	
2	3	4A	4B	5	2	3	4A	4B	5		
5	10	15	20	25	35	38 40	43 45	48 50	53 55	65	

TABLE 2. CONSTANTS (one card, or none if Table 2 of preceding problem is held)

NUM INCRS		INCR LENGTH		
6	10	21	30	

TABLE 3. SPECIFIED DEFLECTIONS (number of cards according to Table 1; none if preceding Table 3 is held)

STATION		DEFLECTION		
6	10	21	30	

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TABLE 4A. SECTION AND MATERIAL PROPERTIES

FROM STA	TO STA	ENTER 1 IF CONT'D ON NEXT CARD	E Modulus Of Elasticity	G Modulus Of Shear	I Moment Of Inertia	A _s Effective Shear Area	
6	10	15	20	30	40	50	60

TABLE 4B. STIFFNESS DATA

FROM STA	TO STA	ENTER 1 IF CONT'D ON NEXT CARD	K Shear Stiffness	F Flexural Stiffness	
6	10	15	20	30	40

TABLE 5. LOAD DATA

FROM STA	TO STA	ENTER 1 IF CONT'D ON NEXT CARD	Q Transverse Force	S Spring Support	T Applied Couple	R Rotational Restraint	P Axial Tension or Compression	
6	10	15	20	30	40	50	60	70

STOP CARD (one blank card at end of run stops program)

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GENERAL PROGRAM NOTES

The data cards must be stacked in proper order for the program to run.

A consistent system of units must be used for all input data, for example, pounds and inches.

All 5-space words are understood to be right-justified integers or whole decimal numbers -43

All 10-space words are right-justified floating-point decimal numbers -4.321E + 03

TABLE 1. PROGRAM-CONTROL DATA

In Tables 2 and 3, either the data from the preceding problem must be held or entirely new data must be entered. If the hold option for either of these tables is set equal to 1, the number of cards input for that table must be zero.

In Tables 4A, 4B, and 5, the data are accumulated by adding to previously stored data. The number of cards input is independent of the hold option, except the cumulative total of cards in each of the tables cannot exceed 100.

Card counts in Table 1 should be rechecked after the coding of each problem is completed.

The plot option has four possible values. If the plot is blank or zero, no plots are drawn. If the plot option is specified as 1, 2, or 3 then each of w , δ , dw , M , V , and Q_R is plotted versus station number, with the size of the axes depending on the value of the plot option. If the plot option is 1, the plots are drawn on 1 × 5 in. axes such that the group of six plots will fit on an 8-1/2 × 11 in. page. If the plot option is 2, the plots are drawn on 2 × 10 in. axes such that the group of six plots will fit on 2 8-1/2 × 11 in. pages. If the plot option is 3, the plots are drawn on 1 × 15 in. axes such that the group of six plots will fit on an 11 × 17 in. page.

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TABLE 2. CONSTANTS

Typical units for the increment length are inches.

The maximum number of increments into which the beam may be divided is 200.

TABLE 3. SPECIFIED DEFLECTIONS

The maximum number of stations for which deflections may be specified is 20.

TABLE 4A. SECTION AND MATERIAL PROPERTIES

Typical units				
variables:	E	G	I	A _s
values per station:	lb/in ²	lb/in ²	in ⁴	in ²

Data should not be entered (nor held from the preceding problem) which would express effects beyond the ends of the beam.

The left end of the beam must be located at Station 0.

The variations in the interpolation and distribution process are explained and illustrated on page 71.

There are no restrictions on the order of cards in Table 4A, except that within a distribution sequence the stations must be in ascending order.

At end stations of each distribution sequence, half-values are automatically created for the modulus of elasticity.

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TABLE 4B. STIFFNESS DATA

Typical units
variables: K F
values per station: 1b/in 1b-in²

Data in Table 4B are governed by the same rules as Table 4A.

The distributed values of stiffness input in Table 4B are automatically added to the stiffnesses computed from the section and material properties of Table 4A.

At end stations of each distribution sequence, half-values are automatically created for the flexural stiffness.

TABLE 5. LOAD DATA

Typical units
variables: Q S T R P
values per station: 1b 1b/in in-lb in-lb/rad 1b

Data in Table 5 are governed by the same rules as Table 4A.

At end stations of each distribution sequence, half-values are automatically created for the transverse force, spring support, applied couple, and rotational restraint.

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Individual Card Input

Data concentrated at one station.....
 Data uniformly distributed.....

FROM STA	TO STA	CONT'D TO NEXT CARD?	TYPE 1 DATA	TYPE 2 DATA	TYPE 3 DATA
2	2	0=NO	3.0		3.0
0	10	0=NO		2.0	
11	15	0=NO	1.0	4.0	2.0
5	15	0=NO	2.0		
1	15	0=NO			1.0

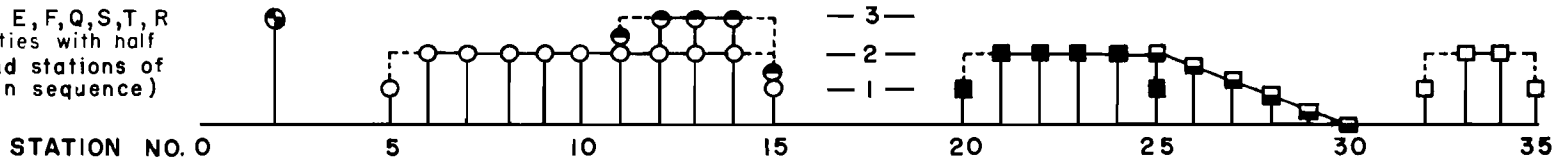
Multiple Card Sequence

First of sequence.....
 Interior of sequence.....
 End of sequence.....

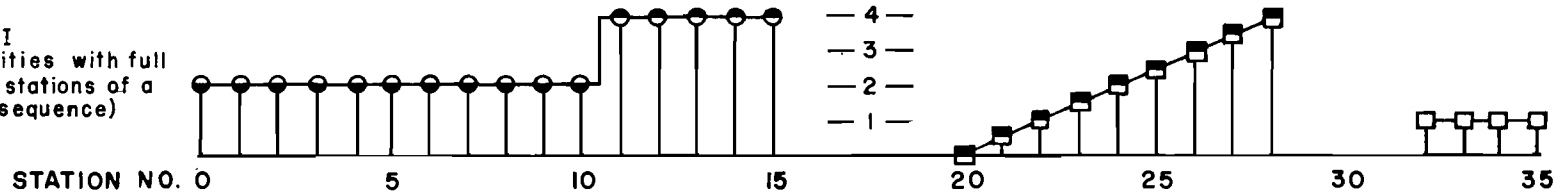
20		1=YES	2.0		
	25	1=YES	2.0		
	30	0=NO	0.0		
20		1=YES		0.0	2.0
	28	0=NO		4.0	2.0
32		1=YES	2.0	1.0	0.0
	35	0=NO	2.0	1.0	3.0

Resulting Distribution of Data

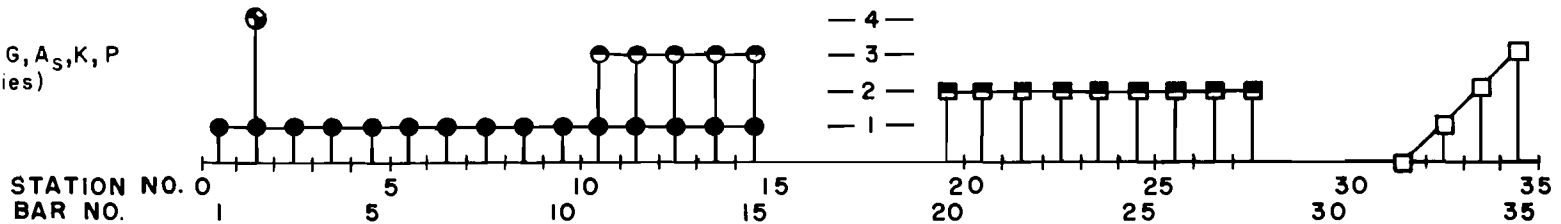
TYPE 1 DATA: E, F, Q, S, T, R
 (Joint quantities with half values at end stations of a distribution sequence)



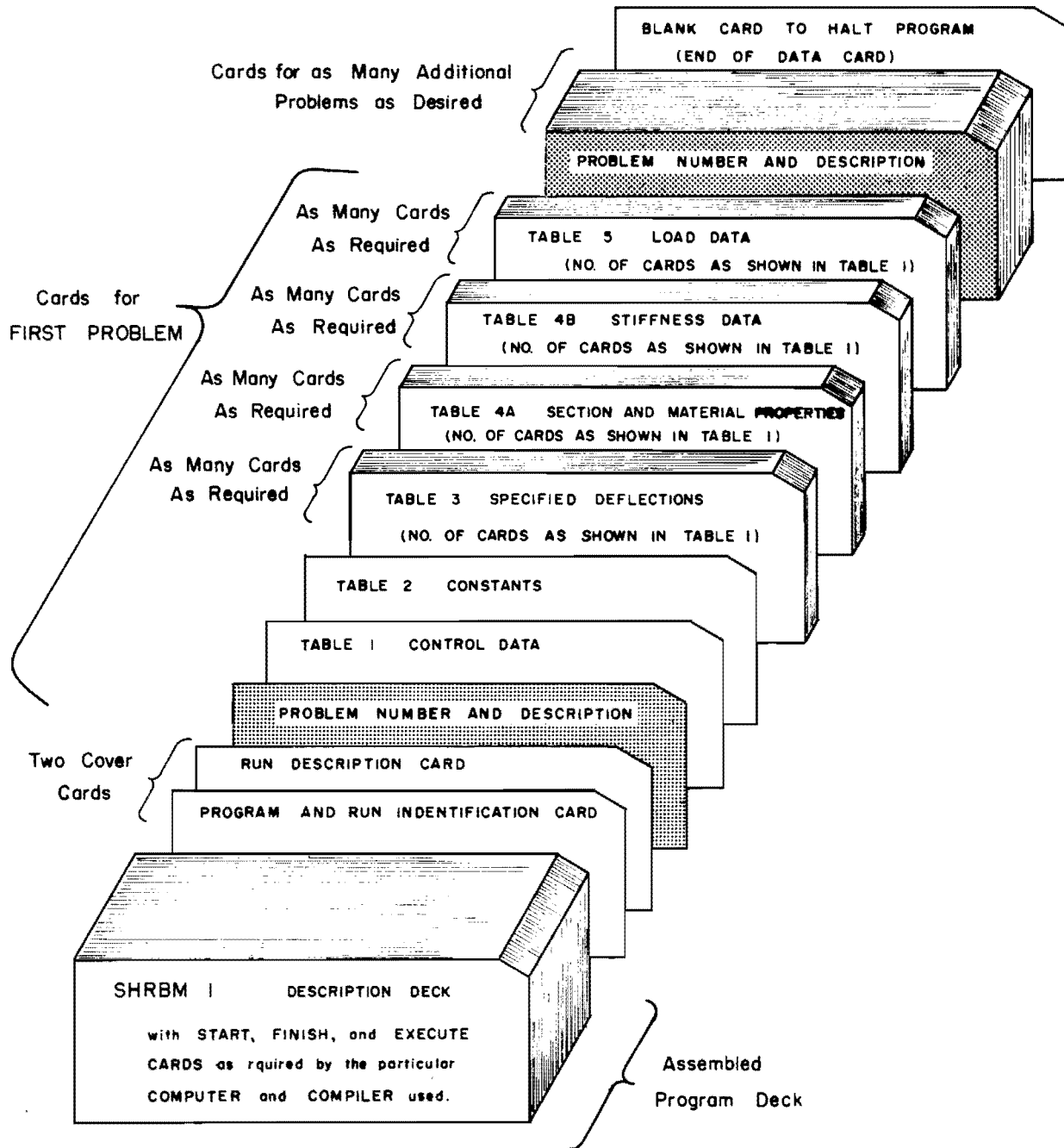
TYPE 2 DATA: I
 (Joint quantities with full values at end stations of a distribution sequence)



TYPE 3 DATA: G, A_S, K, P
 (Bar quantities)



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Organization of program deck.

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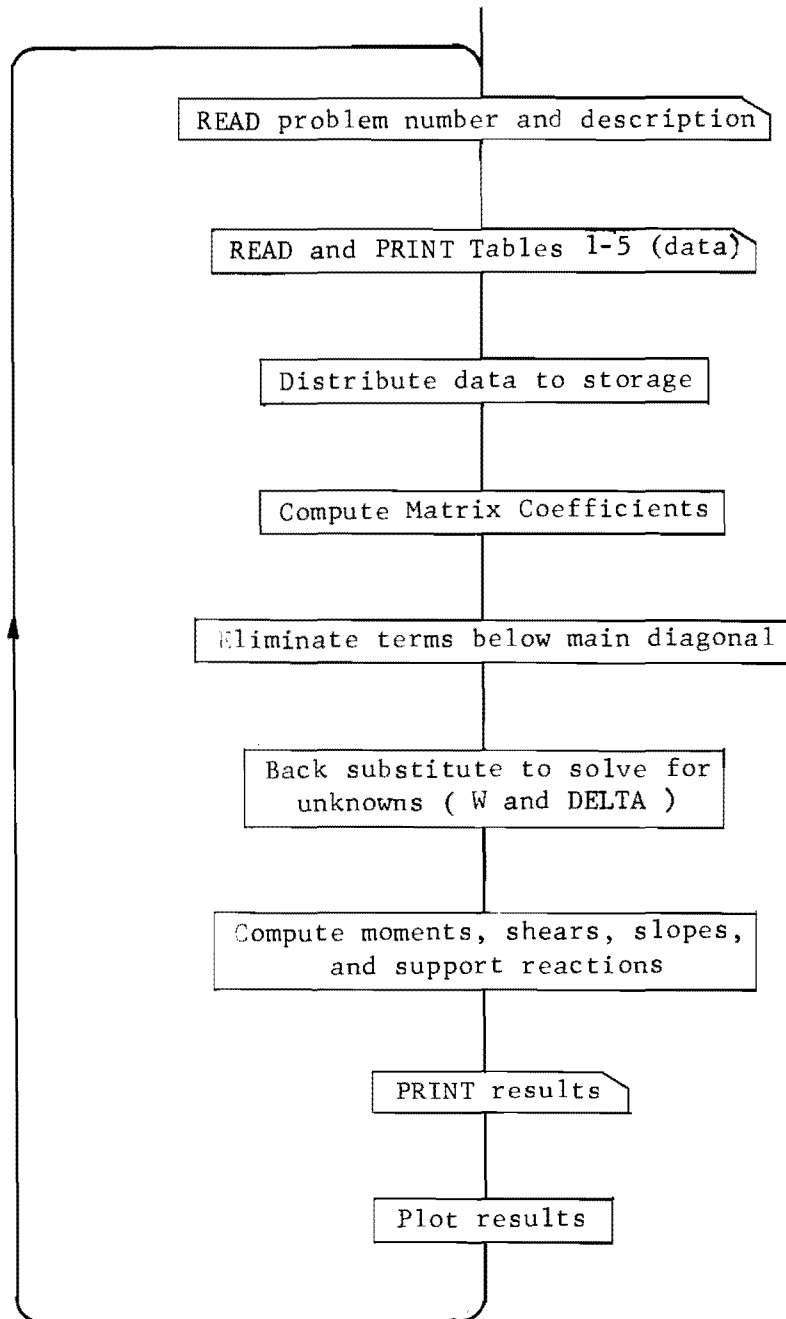
APPENDIX 2

FLOW DIAGRAMS FOR PROGRAM SHRBM 1

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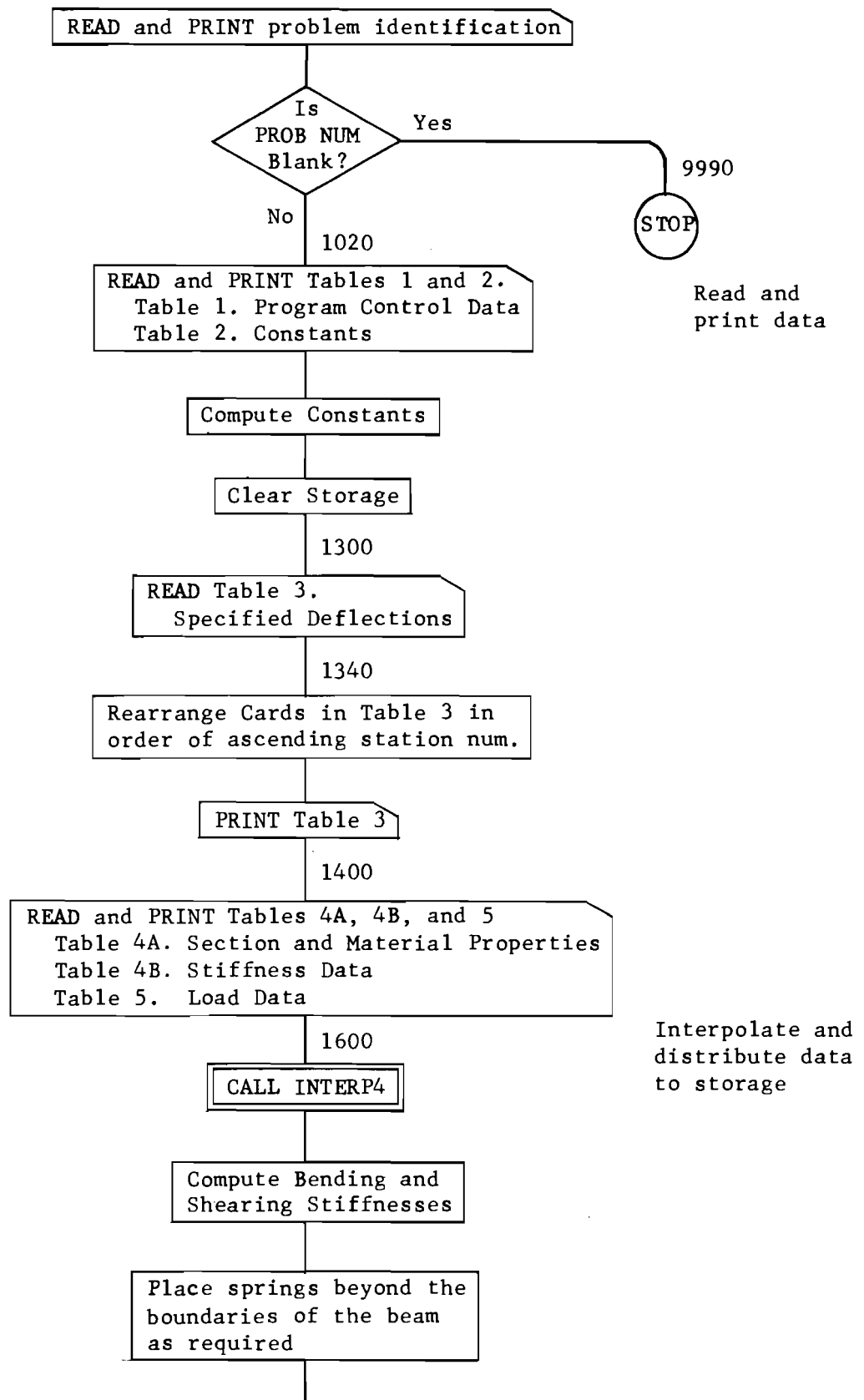
SUMMARY FLOW DIAGRAM FOR PROGRAM SHRBM 1

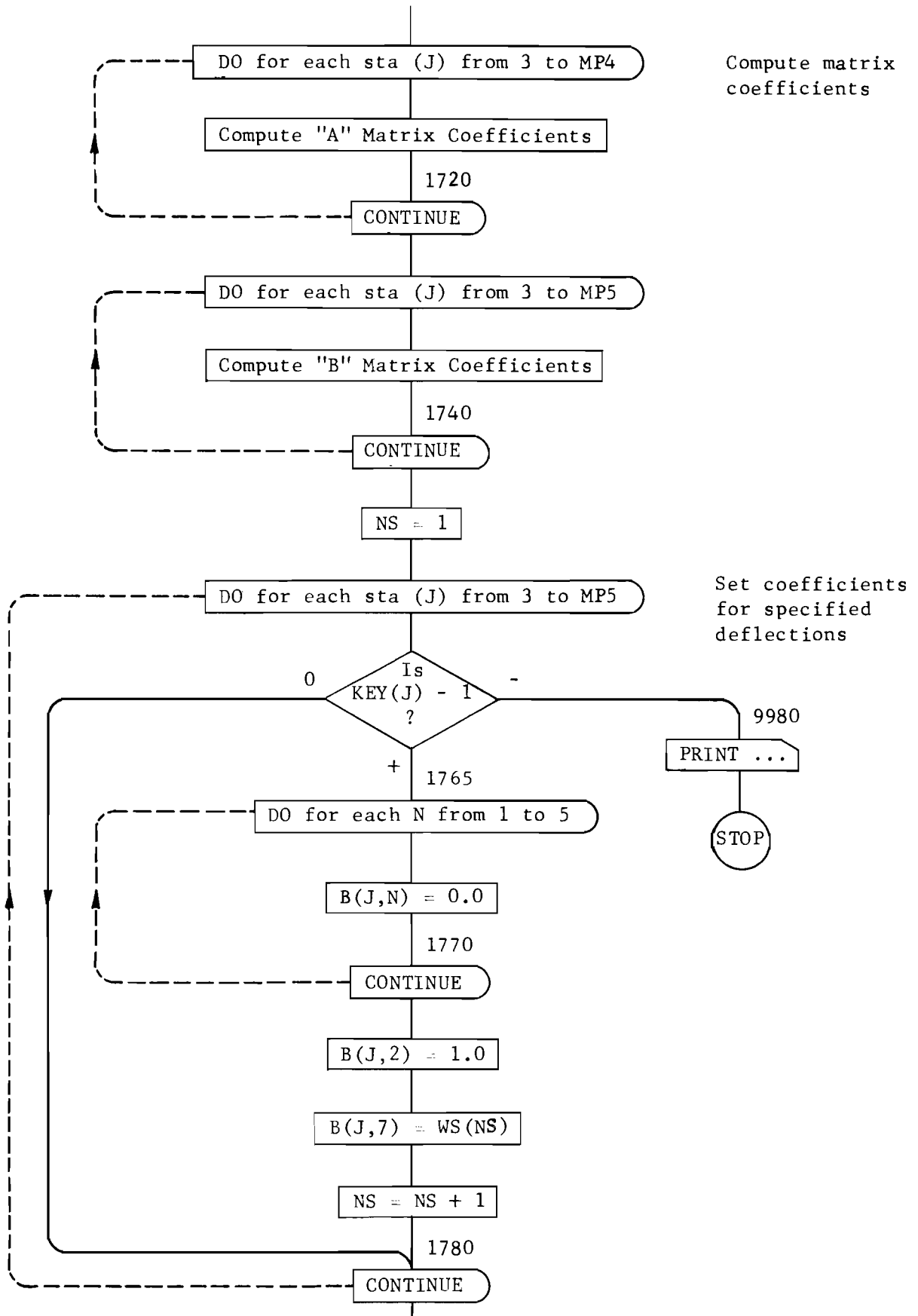


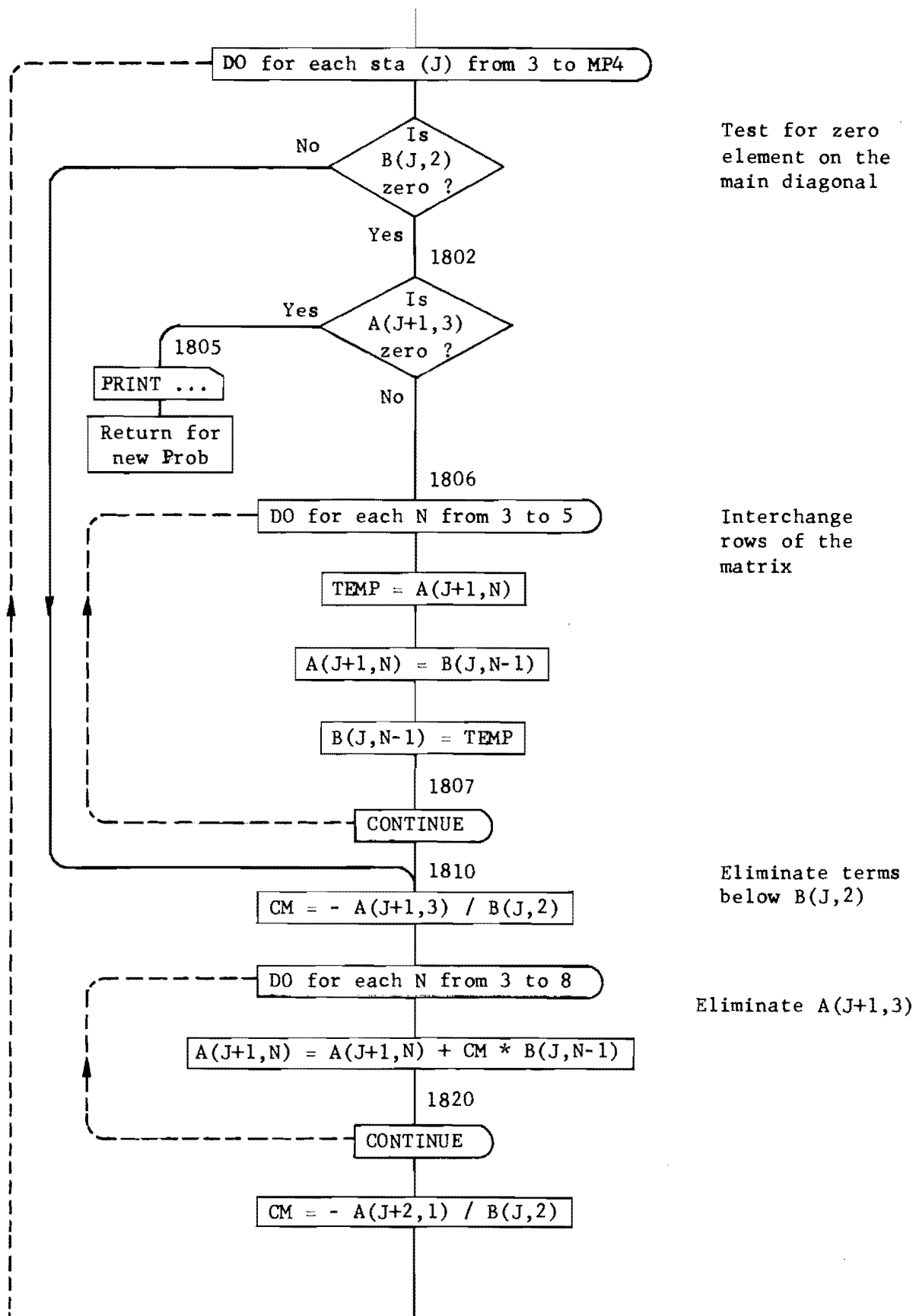
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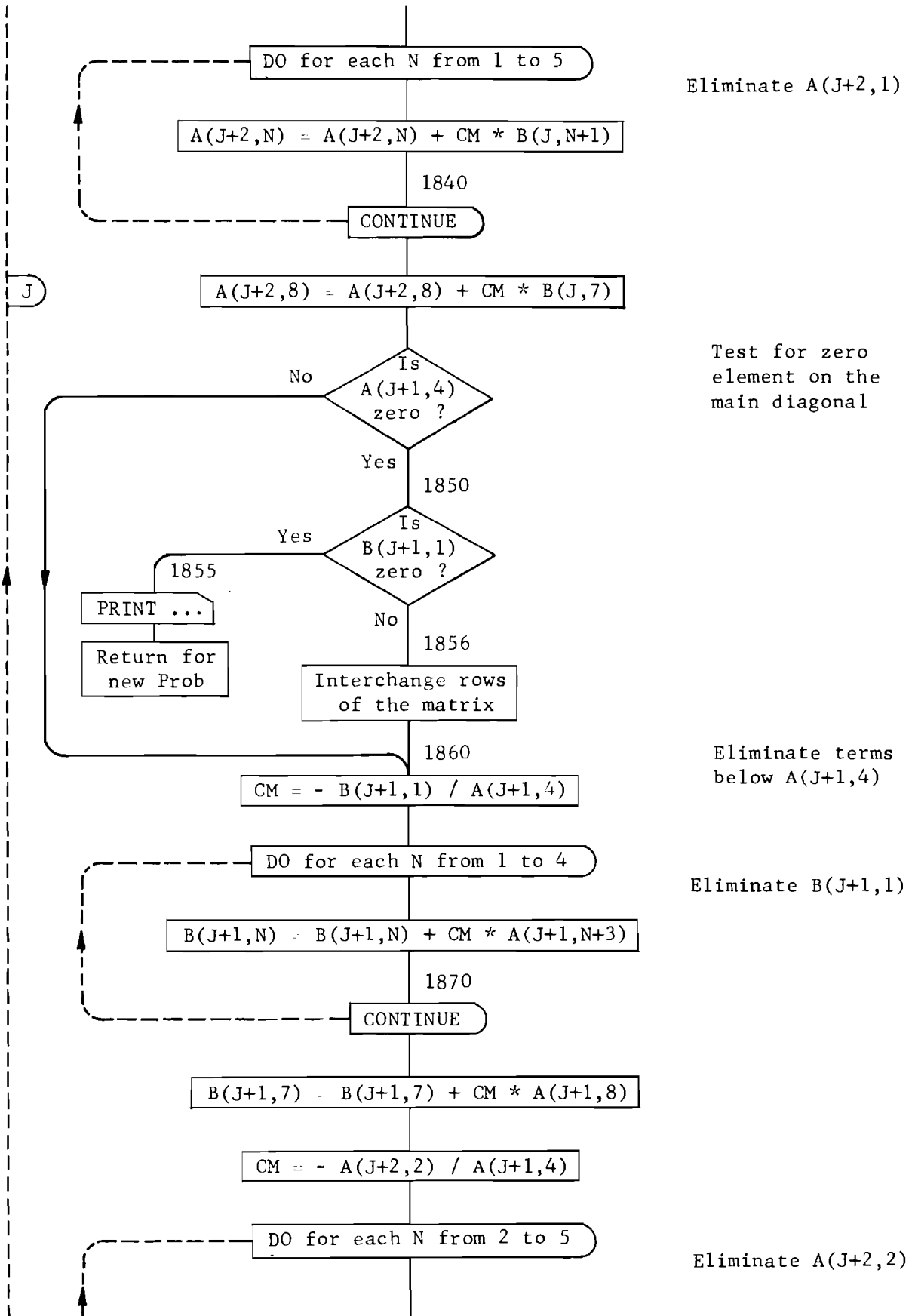
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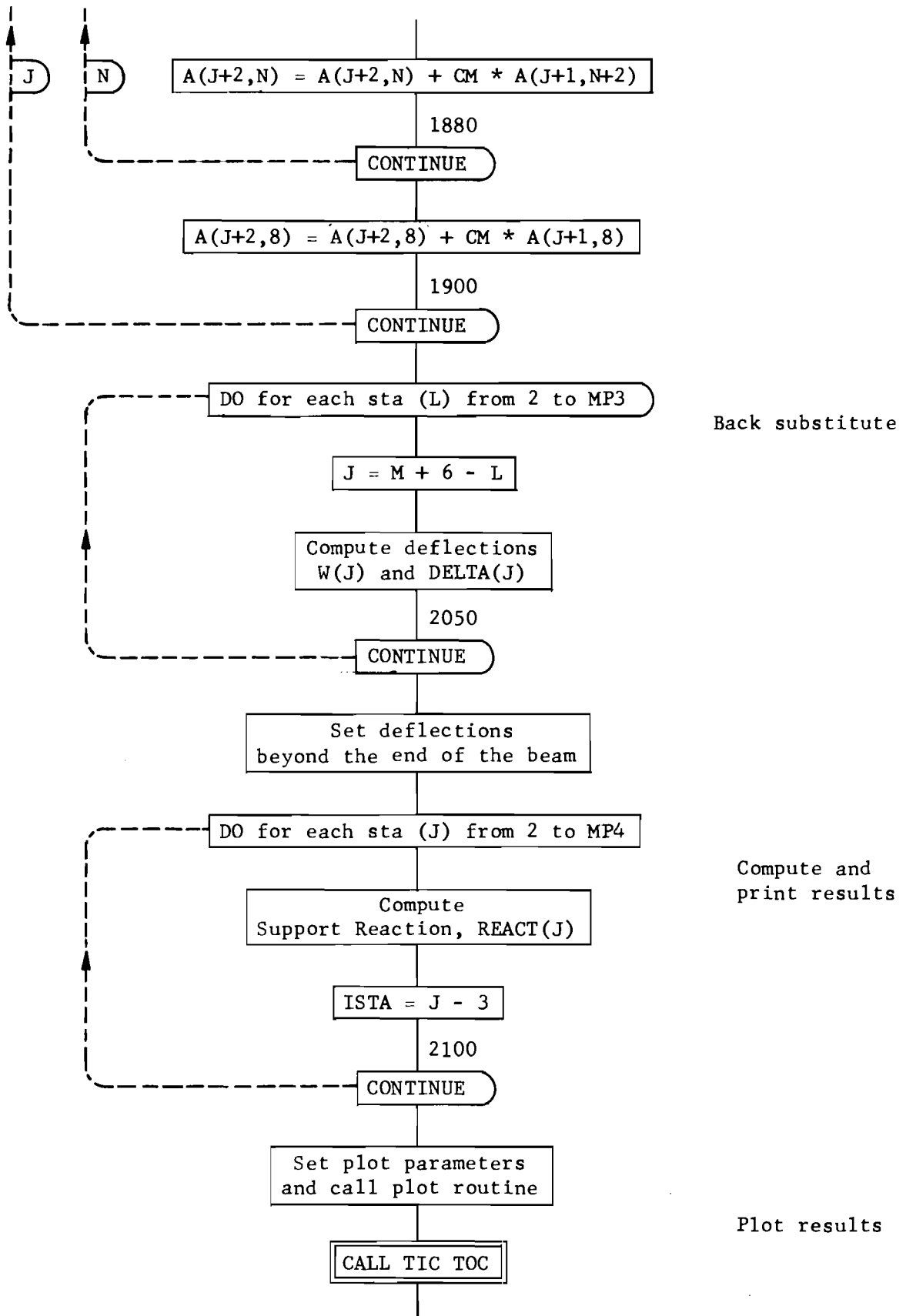
GENERAL FLOW DIAGRAM FOR PROGRAM SHRBM 1

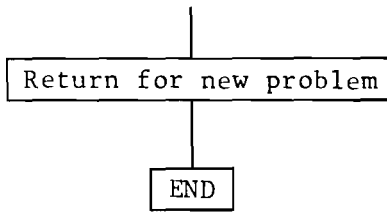




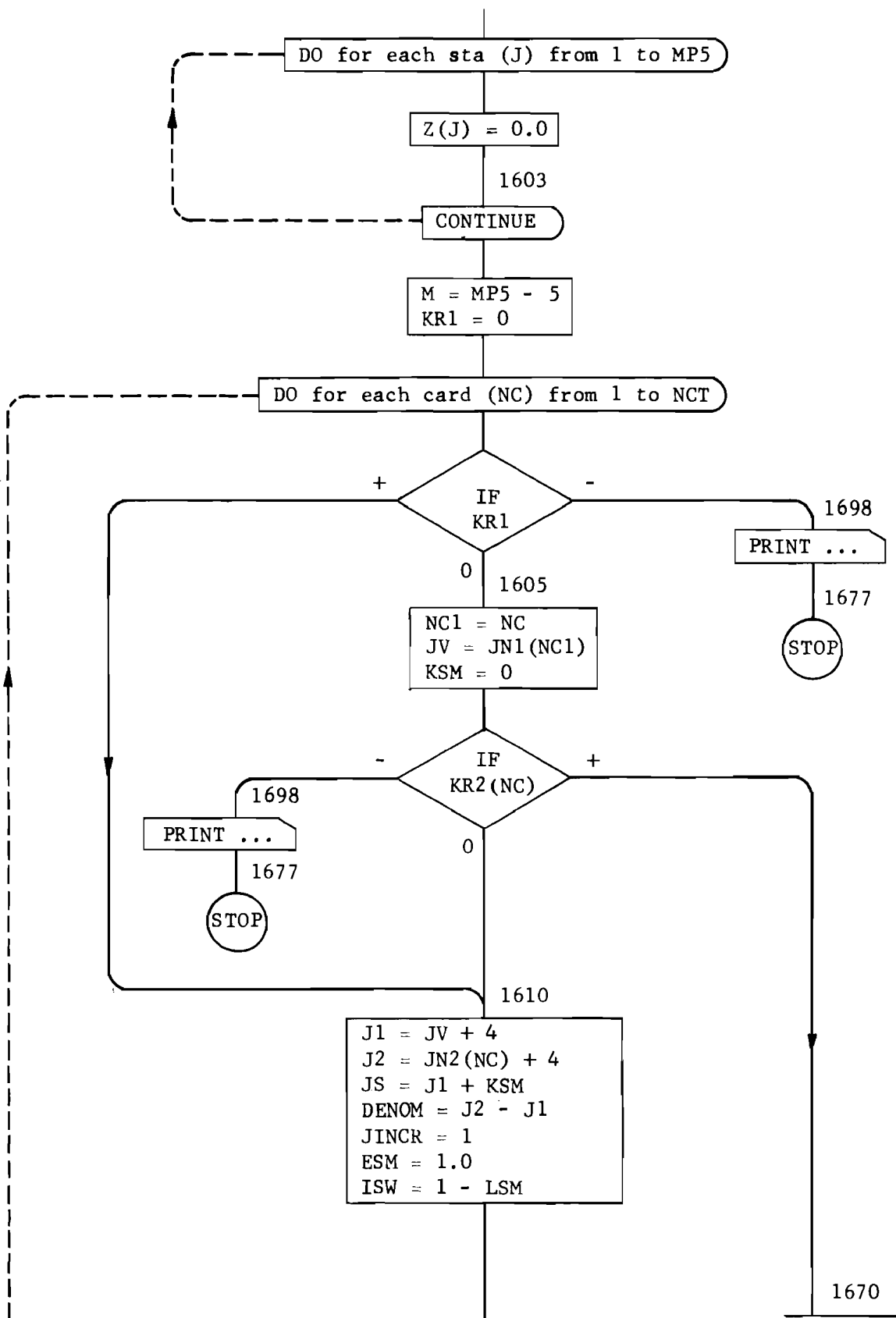


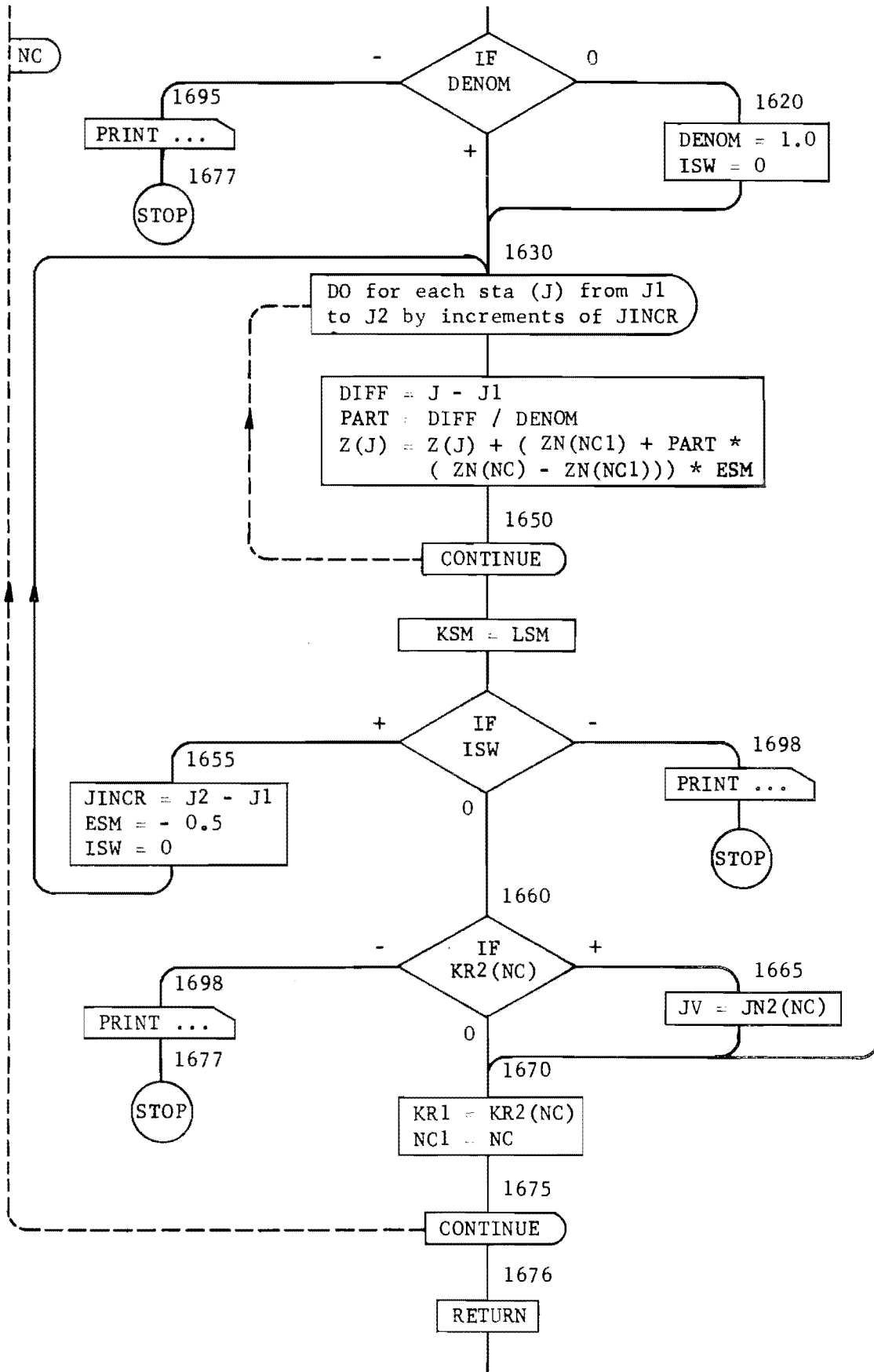






SUBROUTINE INTERP4





APPENDIX 3

GLOSSARY OF NOTATION FOR SHREM 1

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C-----NOTATION FOR SHRBM 1		15AP8
C	A(,)	28FE8
C	AN1(), AN2()	28FE8
C	B(,)	28FE8
C	BM(J)	28FE8
C	CM	28FE8
C		ELIMINATION
C	DELTA(J)	28FE8
C	DENOM	28FE8
C	DIFF	28FE8
C	DW(J)	13MR8
C	E(), E1()	28FE8
C	ESM	28FE8
C	F(), F1()	07MR8
C		INPUT)
C	G(), G1()	28FE8
C	H	28FE8
C	HE2	28FE8
C	HE3	28FE8
C	HT2	28FE8
C	IN13()	28FE8
C	IN14A() THRU	07MR8
C	IN15()	07MR8
C	IN24A() THRU	07MR8
C	IN25()	07MR8
C	IP1	28FE8
C	ITEST	28FE8
C		PROGRAM
C	ISTA	28FE8
C	ISW	28FE8
C	J	28FE8
C	J1	28FE8
C	J2	28FE8
C	JINCR	28FE8
C	JN1(), JN2()	28FE8
C		OF THE DISTRIBUTION SEQUENCE
C	JS	28FE8
C	JV	28FE8
C	KEEP2 THRU KEEP5	28FE8
C	KERR	15AP8
C	KEY(J), KEYJ	28FE8
C	KPLOT	28FE8
C		IF = 0 DO NOT PLOT
C		IF = 1, PLOT ON 8.5 X 11 PAGE WITH
C		1 INCH AND 5 INCH AXES
C		IF = 2, PLOT ON 2 - 8.5 X 11 PAGES WITH
C		2 INCH AND 10 INCH AXES
C		IF = 3, PLOT ON 8.5 X 11 PAGE WITH
C		1 INCH AND 15 INCH AXES
C	KR1	15AP8
C		PRIOR VALUE OF KR24A(), KR24B(), OR
C		KR25()
C	KR24A() THRU	07MR8
C	KR25()	07MR8
C	KSM	28FE8
C		SWITCH USED FOR DISTRIBUTING VALUES TO
C		HALF STATIONS
C		28FE8

C	KSTOP	ROUTING SWITCH FOR PLOT TAPE TERMINATION	28FE8
C	KSW4A() THRU	CONTINUE SWITCH USED IN TABLE 4A THRU	07MR8
C	KSW5()	TABLE 5	07MR8
C	L	DO LOOP INDEX	28FE8
C	LSM	SWITCH USED FOR DISTRIBUTING VALUES TO	28FE8
C		HALF STATIONS	28FE8
C	M	NUMBER OF INCREMENTS	28FE8
C	MP2 THRU MP7	M PLUS 2 THRU M PLUS 7	28FE8
C	N	DO LOOP INDEX	28FE8
C	NCD2 THRU NCD5	NUM CARDS IN TABLES 2 THRU 5, THIS PROB	28FE8
C	NCT2 THRU NCT5	TOTAL NUM CARDS IN TABLES 2 THRU 5	28FE8
C	NC13 THRU NC15	INITIAL INDEX FOR DATA INPUT THIS PROB	28FE8
C		FOR TABLES 3 THRU 5	28FE8
C	NCT3M1	NCT3 MINUS 1	28FE8
C	NS	INDEX NUMBER FOR SPECIFIED CONDITIONS	28FE8
C	NPROB	PROBLEM NUMBER (PROG STOPS IF ZERO)	28FE8
C	ODH	ONE DIVIDED BY H	15AP8
C	ODHE2	ONE DIVIDED BY H SQUARED	15AP8
C	ODHT4	ONE DIVIDED BY H TIMES FOUR	15AP8
C	P(), P1()	AXIAL LOAD (DISTRIBUTED, INPUT)	28FE8
C	PART	INTERPOLATION FRACTION	28FE8
C	Q(), Q1()	TRANSVERSE LOAD (DISTRIBUTED, INPUT)	28FE8
C	REACT()	SUPPORT REACTION	28FE8
C	R(), R1()	ROTATIONAL RESTRAINT (DISTRIBUTED, INPUT)	28FE8
C	S(), S1()	SPRING SUPPORT STIFFNESS (DISTRIBUTED,	13MR8
C		INPUT)	13MR8
C	SA(), SA1()	SHEAR AREA (DISTRIBUTED, INPUT)	07AP8
C	SI(), SI1()	MOMENT OF INERTIA (DISTRIBUTED, INPUT)	28FE8
C	SK(), SK1()	SHEAR SPRING STIFFNESS (DISTRIBUTED,	07MR8
C		INPUT)	07MR8
C	T(), T1()	TRANSVERSE COUPLE (DISTRIBUTED, INPUT)	07MR8
C	TEMP	TEMPORARY STORAGE LOCATION	28FE8
C	V(J)	SHEAR IN BAR J	28FE8
C	W(J)	TOTAL DEFLECTION AT STA J	28FE8
C	WS()	SPECIFIED VALUE OF DEFLECTION	28FE8
C	Z()	INTERPOLATED VALUE	28FE8
C	ZN()	INPUT VALUE FOR INTERPOLATION	28FE8

APPENDIX 4

PROGRAM LISTING OF SHRBM 1

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DTANK,17,160,60000,200.CE244010,MATLOCK.
 RUN(S,,,,,20000)
 LGO.

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PROGRAM SHRBM 1 ( INPUT, OUTPUT )
DIMENSION A(205,8), AN1(32), AN2(14), B(203,7), BM(205),      06MR8
1 DELTA(205), DW(205), E(205),                                08AP8
2 E1(100), F1(100), F(205), G(205), G1(100), HAXIS(7),06MR8
3 HPLOT(205), IN13(20), IN14A(100), IN24A(100),              06MR8
4 IN14B(100), IN24B(100), IN15(100), IN25(100),            06MR8
5 KEY(205), KR24A(100), KR24B(100), KR25(100),            06MR8
6 KSW4A(100), KSW4B(100), KSW5(100), P(205), P1(100),    06MR8
7 Q(205), Q1(100), R(205), R1(100), REACT(205), S(205)08AP8
8 , S1(100), SA(205), SA1(100), SI(205), SI1(100),        08AP8
9 SK1(100), SK(205), T(205), T1(100), V(205),            06MR8
1 VPLOT(205), W(205), WS(20), ZI(205)                      06MR8
1 FORMAT ( 43H PROGRAM SHRBM 1 - DAWKINS - TANKERSLEY )    26AP7
10 FORMAT ( 5H ,80X, 10HI-----TRIM )                    26AP7
11 FORMAT ( 5H1 ,80X, 10HI-----TRIM )                   04MY7
12 FORMAT ( 16A5 )                                         26AP7
13 FORMAT ( 5X, 16A5 )                                     26AP7
14 FORMAT ( A5, 5X, 14A5 )                                 26AP7
15 FORMAT ( ///10H PROB , / , 4X, A5, 5X, 14A5 )          09JE7
16 FORMAT ( ///17H PROB (CONTD), / , 4X, A5, 5X, 14A5 )  28FE8
19 FORMAT ( ///48H RETURN THIS PAGE TO TIME RECORD FILE -- HM ) 04MY7
20 FORMAT ( 2(5I5,5X), I5 )                               05MR8
21 FORMAT ( 5X, I5, 10X, E10.3 )                         26AP7
31 FORMAT ( 5X, I5, 10X, E10.3 )                         26AP7
41 FORMAT ( 5X, 3I5, 4E10.3 )                            08AP8
45 FORMAT ( 5X, 3I5, 2E10.3 )                            10AP8
51 FORMAT ( 5X, 3I5, 5E10.3 )                            26AP7
100 FORMAT ( ///35H TABLE 1 - PROGRAM-CONTROL DATA      26AP7
1 / 58X, 30H TABLES NUMBER                                08MR8
2 / 57X, 30H 2 3 4A 4B 5                                  05MR8
3 // 39H PRIOR-DATA OPTIONS (1 = HOLD), 19X, 5I5,        05MR8
4 / 38H NUM CARDS INPUT THIS PROBLEM, 20X, 5I5,          05MR8
5 // 37H OPTION (IF=1, 2, 3) TO PLOT, 21X, I5 )          05MR8
200 FORMAT ( ///24H TABLE 2 - CONSTANTS / )             26AP7
202 FORMAT ( 25H NUM INCREMENTS , 53X, I5,              09JE7
1 / 27H INCREMENT LENGTH , 46X, E10.3 )                 09JE7
300 FORMAT ( ///36H TABLE 3 - SPECIFIED DEFLECTIONS     26AP7
1 // 5X, 25H STA DEFLECTION , / )                       26AP7
302 FORMAT ( 46H MORE THAN ONE DEFLECTION SPECIFIED AT STA, 26AP7
1 5X, I4 )                                               26AP7
311 FORMAT ( 8X, I4, 9X, E10.3 )                          28FE8
400 FORMAT ( ///47H TABLE 4A - SECTION AND MATERIAL PROPERTIES , 05MR8
1 // 48H FROM TO CONID E G I                            28FE8
2 11H SA , / )                                          08AP8
411 FORMAT ( 4X, 3I4, 1X, 4E11.3 )                       08AP8
412 FORMAT ( 4X, I4, 4X, I4, 1X, 4E11.3 )               08AP8
413 FORMAT ( 8X, 2I4, 1X, 4E11.3 )                       08AP8
450 FORMAT ( ///30H TABLE 4B - STIFFNESS DATA ,        05MR8
1 // 35H FROM TO CONTD SK F , / )                     05MR8
461 FORMAT ( 4X, 3I4, 1X, 2E11.3 )                       05MR8
462 FORMAT ( 4X, I4, 4X, I4, 1X, 2E11.3 )               05MR8
463 FORMAT ( 8X, 2I4, 1X, 2E11.3 )                       05MR8

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500 FORMAT ( // 24H      TABLE 5 - LOAD DATA          26AP7
1 //      48H      FROM TO CONTD      Q          S          T          26AP7
2          20H      R          P , / )          26AP7
511 FORMAT ( 4X, 3I4, 1X, 5E11.3 )          28FE8
512 FORMAT ( 4X, 14, 4X, 14, 1X, 5E11.3 )          28FE8
513 FORMAT ( 8X, 2I4, 1X, 5E11.3 )          28FE8
801 FORMAT ( 29H      ZERO ELEMENT ON DIAGONAL )          26AP7
805 FORMAT ( //51H      TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT 20NO4
1          13H ENDS OF AXES          20NO4
2 // 51H      STA      X          W          DELTA          DW          28FE8
3          35H      M          V          SUP REACT )          28FE8
810 FORMAT ( 4X, 14, 7E11.3 )          28FE8
901 FORMAT ( 38H      NO DATA SPECIFIED FOR THIS TABLE )          15AP8
902 FORMAT ( 50H      NO DATA IN THIS TABLE FROM THE PREVIOUS PROB ) 15AP8
903 FORMAT ( / 25H      NONE )          26AP7
904 FORMAT ( // 40H      TOO MUCH DATA FOR AVAILABLE STORAGE // )          26AP7
905 FORMAT ( 45H      DATA HELD FROM THE PREVIOUS PROBLEM )          26AP7
906 FORMAT ( 50H      DATA HELD ERRONEOUSLY FROM THE PREVIOUS PROB ) 15AP8
910 FORMAT ( 48H      ADDITIONAL DATA INPUT FOR THIS PROBLEM )          26AP7
980 FORMAT ( / 51H      UNSPECIFIED ERROR STOP -- PROGRAM TERMINATED ) 04MY7
991 FORMAT ( //22H      TABLE 6 - RESULTS          28FE8
1 //      41H      STA      W          DELTA      SLOPE          26AP7
2          35H      MOM      SHEAR      SUP REACT , / )          26AP7
992 FORMAT ( 21X, 2E11.3, 11X, E11.3 )          26AP7
993 FORMAT ( 5X, 14, 1X, E11.3, 22X, E11.3, 11X, E11.3 )          26AP7
C-----START EXECUTION OF PROGRAM          26AP7
      KSTOP = 0          26AP7
      ITEST = 5H          26AP7
1000 PRINT 10          26AP7
      CALL TIC TOC(1)          26AP7
      READ 12, ( AN1(N), N = 1, 32 )          26AP7
1010 READ 14, NPROB, ( AN2(N), N = 1, 14 )          28AP7
      IF( NPROB - ITEST ) 1020, 9990, 1020          26AP7
1020 PRINT 11          26AP7
      PRINT 13, ( AN1(N), N = 1, 32 )          26AP7
      PRINT 15, NPROB, ( AN2(N), N = 1, 14 )          26AP7
C-----INPUT TABLE 1. PROGRAM CONTROL DATA          28FE8
1100 READ 20, KEEP2, KEEP3, KEEP4A, KEEP4B, KEEP5, NCD2, NCD3, NCD4A, 05MR8
1          NCD4B, NCD5, KPLOTT          05MR8
      PRINT 100, KEEP2, KEEP3, KEEP4A, KEEP4B, KEEP5, NCD2, NCD3, 05MR8
1          NCD4A, NCD4B, NCD5, KPLOTT          05MR8
C-----INPUT TABLE 2. CONSTANTS          28FE8
1200 PRINT 200          26AP7
      IF( KEEP2 ) 9980, 1205, 1203          11JL7
1203 PRINT 905          11JL7
      IF ( NCD2.LT.1 ) GO TO 1230          15AP8
      PRINT 906          15AP8
      GO TO 9990          15AP8
1205      IF( NCD2 ) 9980, 1207, 1210          26AP7
1207 PRINT 901          15AP8
      GO TO 9990          26AP7
1210 READ 21, M, H          26AP7
1230 PRINT 202, M, H          11JL7
C-----COMPUTE CONSTANTS AND INDICES          26AP7
      HT2 = H + H          26AP7

```

HT4 = HT2 + HT2	08AP8
HE2 = H * H	26AP7
HE3 = H * HE2	26AP7
ODH = 1.0 / H	08AP8
ODHE2 = 1.0 / HE2	08AP8
ODHT4 = 1.0 / HT4	08AP8
MP1 = M + 1	28FE8
MP2 = M + 2	26AP7
MP3 = M + 3	26AP7
MP4 = M + 4	26AP7
MP5 = M + 5	26AP7
C-----CLEAR STORAGE	26AP7
DO 1260 J = 1, MP5	06MR8
W(J) = 0.0	26AP7
V(J) = 0.0	26AP7
DW(J) = 0.0	26AP7
BM(J) = 0.0	26AP7
DELTA(J) = 0.0	26AP7
REACT(J) = 0.0	26AP7
DO 1240 N = 1, 8	26AP7
A(J,N) = 0.0	26AP7
1240 CONTINUE	26AP7
DO 1250 N = 1, 7	09AP8
B(J,N) = 0.0	26AP7
1250 CONTINUE	26AP7
1260 CONTINUE	26AP7
C-----INPUT TABLE 3. SPECIFIED DEFLECTIONS	28FE8
1300 PRINT 300	26AP7
IF (KEEP3) 9980, 1310, 1305	28FE8
1305 PRINT 905	26AP7
IF (NCD3.LT.1) GO TO 1307	15AP8
PRINT 906	15AP8
GO TO 9990	15AP8
1307 NCD3 = NCT3	15AP8
IF (NCD3) 9980, 1308, 1360	15AP8
1308 PRINT 902	15AP8
GO TO 9990	15AP8
1310 DO 1315 J = 2, MP4	06MR8
KEY(J) = 1	26AP7
1315 CONTINUE	26AP7
IF (NCD3) 9980, 1320, 1325	15AP8
1320 PRINT 903	15AP8
GO TO 1399	15AP8
1325 IF(NCD3 - 20) 1327, 1327, 1326	26AP7
1326 PRINT 904	26AP7
GO TO 9990	28FE8
1327 DO 1330 N = 1, NCD3	28FE8
READ 31, IN13(N), WS(N)	09JE7
1330 CONTINUE	16MY7
C-----ARRANGE CARDS IN ASCENDING ORDER OF STA NUMBER	26AP7
IF(NCD3 - 2) 1360, 1340, 1340	26AP7
1340 NCD3M1 = NCD3 - 1	26AP7
DO 1357 I = 1, NCD3M1	12MY7
IP1 = I + 1	26AP7
DO 1355 N = IP1, NCD3	04MY7

IF(IN13(I) - IN13(N)) 1355, 1348, 1350	12MY7
1348 PRINT 302, IN13(N)	26AP7
GO TO 9990	26AP7
1350 TEMP = IN13(I)	26AP7
IN13(I) = IN13(N)	26AP7
IN13(N) = TEMP	26AP7
TEMP = WS(I)	26AP7
WS(I) = WS(N)	26AP7
WS(N) = TEMP	26AP7
1355 CONTINUE	26AP7
1357 CONTINUE	12MY7
1360 CONTINUE	26AP7
C-----SET INDICES FOR FUTURE CONTROL OF SPECIFIED CONDITION ROUTINES	26AP7
DO 1398 N = 1, NCD3	12JL7
JS = IN13(N) + 3	06MR8
KEY(JS) = 2	26AP7
PRINT 311, IN13(N), WS(N)	26AP7
1398 CONTINUE	12MY7
1399 CONTINUE	26AP7
NCT3 = NCD3	12JL7
C-----INPUT TABLE 4A. SECTION AND MATERIAL PROPERTIES	05MR8
1400 PRINT 400	26AP7
KERR = 0	15AP8
IF(KEEP4A) 9980, 1401, 1409	05MR8
1401 NCT4A = 0	05MR8
IF(NCD4A) 9980, 1402, 1428	05MR8
1402 PRINT 903	05MR8
KERR = 1	15AP8
GO TO 1450	05MR8
1409 PRINT 905	28FE8
IF(NCT4A) 9980, 1410, 1411	05MR8
1410 PRINT 902	28FE8
GO TO 9990	28FE8
1411 DO 1427 N = 1, NCT4A	05MR8
KSW4AN = KSW4A(N)	05MR8
GO TO (1413, 1417, 1421, 1421), KSW4AN	05MR8
1413 PRINT 411, IN14A(N), IN24A(N), KR24A(N), E1(N), G1(N), SI1(N),	05MR8
1 SA1(N)	08AP8
GO TO 1425	05MR8
1417 PRINT 412, IN14A(N), KR24A(N), E1(N), G1(N), SI1(N), SA1(N)	08AP8
GO TO 1425	05MR8
1421 PRINT 413, IN24A(N), KR24A(N), E1(N), G1(N), SI1(N), SA1(N)	08AP8
1425 CONTINUE	05MR8
1427 CONTINUE	05MR8
IF(NCD4A) 9980, 1450, 1428	05MR8
1428 NCT4A = NCT4A + 1	05MR8
NCT4A = NCT4A + NCD4A	05MR8
IF(NCT4A - 100) 1435, 1435, 1433	05MR8
1433 PRINT 904	26AP7
GO TO 9990	26AP7
1435 PRINT 910	28FE8
1440 KR1 = 0	26AP7
DO 1449 N = NCT4A, NCT4A	05MR8
READ 41, IN14A(N), IN24A(N), KR24A(N), E1(N), G1(N), SI1(N),	05MR8
1 SA1(N)	08AP8

```

        KSW4A(N) = 1 + KR24A(N) + 2.0 * KR1      05MR8
        KR1 = KR24A(N)                          05MR8
        KSW4AN = KSW4A(N)                       05MR8
        GO TO ( 1443, 1445, 1447, 1447 ), KSW4AN 05MR8
1443 PRINT 411, IN14A(N), IN24A(N), KR24A(N), E1(N), G1(N), S11(N), 05MR8
    1 SA1(N)                                     08AP8
        GO TO 1448                               05MR8
1445 PRINT 412, IN14A(N), KR24A(N), E1(N), G1(N), S11(N), SA1(N) 08AP8
        GO TO 1448                               05MR8
1447 PRINT 413, IN24A(N), KR24A(N), E1(N), G1(N), S11(N), SA1(N) 08AP8
1448 CONTINUE                                   05MR8
1449 CONTINUE                                   05MR8
C-----INPUT TABLE 4B. STIFFNESS DATA      05MR8
1450 PRINT 450                                  05MR8
        IF( KEEP4B ) 9980, 1451, 1459          05MR8
1451        NCT4B = 0                            05MR8
        IF( NCD4B ) 9980, 1452, 1478          05MR8
1452 PRINT 903                                  05MR8
        IF ( KERR.NE.1 ) GO TO 1500           15AP8
        PRINT 901                               15AP8
        GO TO 9990                              15AP8
1459 PRINT 905                                  05MR8
        IF( NCT4B ) 9980, 1460, 1461          05MR8
1460 PRINT 902                                  05MR8
        GO TO 9990                              05MR8
1461 DO 1477 N = 1, NCT4B                       05MR8
        KSW4BN = KSW4B(N)                       05MR8
        GO TO ( 1463, 1467, 1471, 1471 ), KSW4BN 05MR8
1463 PRINT 461, IN14B(N), IN24B(N), KR24B(N), SK1(N), F1(N)      05MR8
        GO TO 1475                              05MR8
1467 PRINT 462, IN14B(N), KR24B(N), SK1(N), F1(N)      05MR8
        GO TO 1475                              05MR8
1471 PRINT 463, IN24B(N), KR24B(N), SK1(N), F1(N)      05MR8
1475 CONTINUE                                   05MR8
1477 CONTINUE                                   05MR8
        IF( NCD4B ) 9980, 1500, 1478          05MR8
1478        NC14B = NCT4B + 1                   05MR8
        NCT4B = NCT4B + NCD4B                 05MR8
        IF( NCT4B - 100 ) 1485, 1485, 1483      05MR8
1483 PRINT 904                                  05MR8
        GO TO 9990                              05MR8
1485 PRINT 910                                  05MR8
1490        KR1 = 0                             05MR8
        DO 1499 N = NC14B, NCT4B               05MR8
        READ 45, IN14B(N), IN24B(N), KR24B(N), SK1(N), F1(N)      05MR8
        KSW4B(N) = 1 + KR24B(N) + 2.0 * KR1      05MR8
        KR1 = KR24B(N)                           05MR8
        KSW4BN = KSW4B(N)                       05MR8
        GO TO ( 1493, 1495, 1497, 1497 ), KSW4BN 05MR8
1493 PRINT 461, IN14B(N), IN24B(N), KR24B(N), SK1(N), F1(N)      05MR8
        GO TO 1498                              05MR8
1495 PRINT 462, IN14B(N), KR24B(N), SK1(N), F1(N)      05MR8
        GO TO 1498                              05MR8
1497 PRINT 463, IN24B(N), KR24B(N), SK1(N), F1(N)      05MR8
1498 CONTINUE                                   05MR8

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1499	CONTINUE	05MR8
C-----	INPUT TABLE 5. LOAD DATA	28FE8
1500	PRINT 500	26AP7
	IF(KEEP5) 9980, 1501, 1509	26AP7
1501	NCT5 = 0	28FE8
	IF(NCD5) 9980, 1502, 1528	05SE7
1502	PRINT 901	15AP8
	GO TO 9990	26AP7
1509	PRINT 905	26AP7
	IF(NCT5) 9980, 1510, 1511	28FE8
1510	PRINT 902	28FE8
	GO TO 9990	28FE8
1511	DO 1527 N = 1, NCT5	12MY7
	KSW5N = KSW5(N)	26AP7
	GO TO (1513, 1517, 1521, 1521), KSW5N	26AP7
1513	PRINT 511, IN15(N), IN25(N), KR25(N), Q1(N), S1(N), T1(N), R1(N),	26AP7
1	P1(N)	26AP7
	GO TO 1525	26AP7
1517	PRINT 512, IN15(N), KR25(N), Q1(N), S1(N), T1(N), R1(N), P1(N)	26AP7
	GO TO 1525	26AP7
1521	PRINT 513, IN25(N), KR25(N), Q1(N), S1(N), T1(N), R1(N), P1(N)	26AP7
1525	CONTINUE	26AP7
1527	CONTINUE	26AP7
	IF(NCD5) 9980, 1600, 1528	05SE7
1528	NC15 = NCT5 + 1	26AP7
	NCT5 = NCT5 + NCD5	26AP7
	IF(NCT5 - 100) 1535, 1535, 1533	26AP7
1533	PRINT 904	26AP7
	GO TO 9990	26AP7
1535	PRINT 910	28FE8
1540	KR1 = 0	26AP7
	DO 1571 N = NC15, NCT5	12MY7
	READ 51, IN15(N), IN25(N), KR25(N), Q1(N), S1(N), T1(N), R1(N),	28AP7
1	P1(N)	26AP7
	KSW5(N) = 1 + KR25(N) + 2 * KR1	05MY7
	KR1 = KR25(N)	26AP7
	KSW5N = KSW5(N)	28AP7
	GO TO (1550, 1555, 1560, 1560), KSW5N	28AP7
1550	PRINT 511, IN15(N), IN25(N), KR25(N), Q1(N), S1(N), T1(N), R1(N),	26AP7
1	P1(N)	26AP7
	GO TO 1570	26AP7
1555	PRINT 512, IN15(N), KR25(N), Q1(N), S1(N), T1(N), R1(N), P1(N)	26AP7
	GO TO 1570	26AP7
1560	PRINT 513, IN25(N), KR25(N), Q1(N), S1(N), T1(N), R1(N), P1(N)	26AP7
1570	CONTINUE	26AP7
1571	CONTINUE	12MY7
C-----	INTERPOLATE AND DISTRIBUTE DATA FROM TABLE 4A	05MR8
1600	LSM = 1	26AP7
	CALL INTERP4 (MP5, NCT4A, IN14A, IN24A, KR24A, G1, G, LSM)	06MR8
	CALL INTERP4 (MP5, NCT4A, IN14A, IN24A, KR24A, S11, S1, LSM)	06MR8
	CALL INTERP4 (MP5, NCT4A, IN14A, IN24A, KR24A, SA1, SA, LSM)	08AP8
	LSM = 0	26AP7
	CALL INTERP4 (MP5, NCT4A, IN14A, IN24A, KR24A, E1, E, LSM)	06MR8
C-----	INTERPOLATE AND DISTRIBUTE DATA FROM TABLE 4B	05MR8
1675	LSM = 1	05MR8


```

1775     CONTINUE
1780     CONTINUE
C-----BEGIN GAUSSIAN ELIMINATION
      DO 1900 J = 2, MP3
C-----TEST FOR ZERO ELEMENT ON THE DIAGONAL
      IF( ABSF( B(J,2) ) ) 9980, 1802, 1810
1802     IF ( ABSF( A(J+1,3) ) ) 9980, 1805, 1806
1805     PRINT 801
      GO TO 9970
C-----INTERCHANGE ROWS OF THE MATRIX
1806     DO 1807 N = 3, 8
          TEMP = A(J+1,N)
          A(J+1,N) = B(J,N-1)
          B(J,N-1) = TEMP
1807     CONTINUE
C-----ELIMINATE ALL TERMS BELOW B(J,2)
1810     CM = - A(J+1,3) / B(J,2)
      DO 1820 N = 3, 8
          A(J+1,N) = A(J+1,N) + CM * B(J,N-1)
1820     CONTINUE
          CM = - A(J+2,1) / B(J,2)
      DO 1840 N = 1, 5
          A(J+2,N) = A(J+2,N) + CM * B(J,N+1)
1840     CONTINUE
          A(J+2,8) = A(J+2,8) + CM * B(J,7)
C-----TEST FOR ZERO ELEMENT ON THE DIAGONAL
      IF ( ABSF ( A(J+1,4) ) ) 9980, 1850, 1860
1850     IF( ABSF( B(J+1,1) ) ) 9980, 1855, 1856
1855     PRINT 801
      GO TO 9970
C-----INTERCHANGE ROWS OF THE MATRIX
1856     DO 1857 N = 1, 4
          TEMP = B(J+1,N)
          B(J+1,N) = A(J+1,N+3)
          A(J+1,N+3) = TEMP
1857     CONTINUE
          TEMP = B(J+1,7)
          B(J+1,7) = A(J+1,8)
          A(J+1,8) = TEMP
C-----ELIMINATE ALL TERMS BELOW A(J+1,4)
1860     CM = - B(J+1,1) / A(J+1,4)
      DO 1870 N = 1, 4
          B(J+1,N) = B(J+1,N) + CM * A(J+1,N+3)
1870     CONTINUE
          B(J+1,7) = B(J+1,7) + CM * A(J+1,8)
          CM = - A(J+2,2) / A(J+1,4)
      DO 1880 N = 2, 5
          A(J+2,N) = A(J+2,N) + CM * A(J+1,N+2)
1880     CONTINUE
          A(J+2,8) = A(J+2,8) + CM * A(J+1,8)
1900     CONTINUE
C-----BEGIN BACK SUBSTITUTION
      DO 2050 L = 2, MP3
          J = M + 6 - L
          W(J) = ( B(J,7) - B(J,6) * W(J+2) - B(J,5) * DELTA(J+2)

```

```

28FE8
26AP7
26AP7
06MR8
28FE8
05MR8
19AP8
26AP7
28FE8
28FE8
28FE8
28FE8
05MR8
05MR8
28FE8
05MR8
08AP8
26AP7
08AP8
26AP7
08AP8
28AP7
08AP8
26AP7
05MR8
28FE8
05MR8
07MR8
05MR8
05MR8
05MR8
07MR8
07MR8
07MR8
07MR8
05MR8
05MR8
05MR8
05MR8
26AP7
05MR8
26AP7
26AP7
19AP8
26AP7
19AP8
26AP7
26AP7
06MR8
06MR8
05MR8

```

```

1          - B(J,4) * W(J+1) - B(J,3) * DELTA(J+1) ) / B(J,2) 05MR8
          DELTA(J) = ( A(J,8) - A(J,7) * W(J+1) - A(J,6) *      08AP8
1          DELTA(J+1) - A(J,5) * W(J) ) / A(J,4)                08AP8
2050      CONTINUE                                             26AP7
          W(2) = ( B(2,7) - B(2,6) * W(4) - B(2,5) * DELTA(4)  06MR8
1          - B(2,4) * W(3) - B(2,3) * DELTA(3) ) / B(2,2)      06MR8
          W(1) = 2.0 * W(2) - W(3) - DELTA(3)                  06MR8
          W(MP5) = 2.0 * W(MP4) - W(MP3) + DELTA(MP4)          06MR8
C-----COMPUTE AND PRINT RESULTS                               26AP7
          DO 2070 J = 2, MP4                                    06MR8
1          BM(J) = F(J) * ( W(J-1) - 2.0 * W(J) + W(J+1) - DELTA(J) 26AP7
          + DELTA(J+1) ) * ODHE2                                09AP8
          V(J) = SK(J) * DELTA(J)                               26AP7
          DW(J) = ( W(J) - W(J-1) + DELTA(J) ) * ODH           09AP8
2070      CONTINUE                                             26AP7
          PRINT 11                                             26AP7
          PRINT 13, ( AN1(N), N = 1, 32 )                      28AP7
          PRINT 16, NPROB, ( AN2(N), N = 1, 14 )               26AP7
          PRINT 991                                             26AP7
          DO 2100 J = 2, MP4                                    06MR8
          IF ( KEY(J) - 1 ) 9980, 2080, 2075                   28FE8
2075      REACT(J) = V(J+1) - V(J) - Q(J)                       26DE7
          GO TO 2082                                           28FE8
2080      REACT(J) = S(J) * W(J)                               28FE8
2082      ISTA = J - 3                                         10MR8
          ZI(J) = ISTA                                         28FE8
          IF ( J - 3 ) 2085, 2084, 2084                        10MR8
2084      PRINT 992, DELTA(J), DW(J), V(J)                     26AP7
2085      PRINT 993, ISTA, W(J), BM(J), REACT(J)               26AP7
2100      CONTINUE                                             26AP7
C-----BEGIN AUTOMATIC PLOT ROUTINE                           16MR4
          PRINT 805                                             28FE8
          IF ( KPLOT ) 9980, 8300, 8310                         19JE4
8300      PRINT 903                                           23MR4
          GO TO 8700                                           23MR4
C-----DETERMINE STATION NUMBER AT END OF STATION AXES      19JE4
8310      KSTOP = 1                                           19JE4
          IF ( M - 200 ) 8320, 8350, 8340                     19JE4
8320      IF ( M - 100 ) 8330, 8360, 8350                     25FE4
8330      IF ( M - 50 ) 8370, 8370, 8360                      25FE4
8340      VEND = -500.0                                        01JE4
          GO TO 8400                                           25FE4
8350      VEND = -200.0                                        01JE4
          GO TO 8400                                           25FE4
8360      VEND = -100.0                                       01JE4
          GO TO 8400                                           25FE4
8370      VEND = -50.0                                         01JE4
C-----SET PLOT ARRAY EQUAL TO W, DELTA, DW, M, V, SUP REACT 28FE8
8400      DO 8650 N = 1, 6                                     14MR8
          GO TO ( 8410, 8420, 8430, 8440, 8450, 8456 ), N     28FE8
8410      DO 8415 J = 3, MP3                                    06MR8
          HPLOT(J-2) = W(J)                                     09MR8
          VPLOT(J-2) = -ZI(J)                                  10MR8
8415      CONTINUE                                             25FE4
          NT = 1                                               14MR8

```

	SPACE = 2.0	12JA8
	GO TO 8465	16MR4
8420	DO 8425 J = 3, MP3	06MR8
	HPlot(J-2) = DELTA(J)	09MR8
8425	CONTINUE	25FE4
	NT = 1	14MR8
	GO TO 8460	25FE4
8430	DO 8435 J = 3, MP3	06MR8
	HPlot(J-2) = DW(J)	09MR8
8435	CONTINUE	25FE4
	NT = 2	14MR8
	GO TO 8460	25FE4
8440	DO 8445 J = 3, MP3	06MR8
	HPlot(J-2) = BM(J)	09MR8
8445	CONTINUE	25FE4
	NT = 1	14MR8
	GO TO 8460	25FE4
8450	DO 8455 J = 3, MP3	06MR8
	HPlot(J-2) = V(J)	09MR8
8455	CONTINUE	25FE4
	NT = 1	14MR8
	GO TO 8460	28FE8
8456	DO 8458 J = 3, MP3	06MR8
	HPlot(J-2) = REACT(J)	09MR8
8458	CONTINUE	28FE8
8460	SPACE = SPACE + 1.5	23MR4
8465	HMAX = 0.0	01JE4
	C-----DETERMINE LARGEST VALUE IN NEXT ARRAY TO BE PLOTTED	16MR4
	DO 8470 J = 3, MP3	06MR8
	ZTMP = ABSF(HPlot(J-2))	09MR8
	HMAX = MAX1F(ZTMP, HMAX)	01JE4
8470	CONTINUE	25FE4
	C-----DETERMINE EXPONENT OF MAX VALUE OF FUNCTION AND SET H-AXIS SCALE	23JE4
	KEXP = 0	01JE4
	IF (HMAX - 1.0E-10) 8570, 8570, 8500	06MR8
8500	DO 8530 L = 1, 100	25JE4ARB
	IF (1.0 - HMAX) 8505, 8570, 8510	01JE4
8505	HMAX = HMAX / 10.0	01JE4
	KEXP = KEXP + 1	01JE4
	GO TO 8530	18MR4
8510	IF (0.1 - HMAX) 8540, 8560, 8520	01JE4
8520	HMAX = HMAX * 10.0	01JE4
	KEXP = KEXP - 1	01JE4
8530	CONTINUE	18MR4
	GO TO 9980	19JE4
8540	IF (0.2 - HMAX) 8550, 8580, 8580	01JE4
8550	IF (0.5 - HMAX) 8570, 8590, 8590	01JE4
8560	HEND = 0.1	01JE4
	GO TO 8595	18MR4
8570	HEND = 1.0	01JE4
	GO TO 8595	18MR4
8580	HEND = 0.2	01JE4
	GO TO 8595	18MR4
8590	HEND = 0.5	01JE4
	C-----SET PLOT VARIABLE VALUE AT END OF VARIABLE AXES	01JE4

```

8595          HEND = HEND * 10.0**KEXP
              HAXIS(N) = HEND
C-----SET REMAINING AXES AND PLOT ARGUMENTS
              VPOS = 0.0
              VNEG = 5.0 * KPLOT
              HTCKS = HEND
              VTCKS = -VEND
              NUMPTS = MPI
              KPS = 1
              IF ( KPLOT - 2 ) 8596, 8597, 8596
C-----DRAW AXES AND PLOT FOR KPLOT = 1 AND 3
8596          HPOS = 0.5
              HNEG = 0.5
              KAXES = 0
              CALL AXES ( HEND, HPOS, HNEG, SPACE, VEND, VPOS, VNEG, HTCKS,
1              VTCKS, KAXES )
              CALL PLOT ( HPLLOT, VPLOT, NUMPTS, KPS )
C-----SET PEN FOR NEXT PLOT
              HPLTTMP = 0.0
              VPLTTMP = VEND
              NUMPTS = 1
              KPS = 8
              CALL PLOT ( HPLTTMP, VPLTTMP, NUMPTS, KPS )
              GO TO 8650
C-----DRAW AXES AND PLOT FOR KPLOT = 2
8597          HPOS = 1.0
              HNEG = 1.0
              SPACE = 0.75
              KAXES = 1
              IF ( NT - 2 ) 8599, 8598, 8599
8598          VINCH = 4.0
              GO TO 8600
8599          VINCH = 0.5
8600 CALL AXES ( HEND, HPOS, HNEG, SPACE, VEND, VPOS, VNEG, HTCKS,
1              VTCKS, KAXES )
              CALL PLOT ( HPLLOT, VPLOT, NUMPTS, KPS )
C-----SET PEN FOR NEXT PLOT
              CALL AXTERM ( 1 )
              INCH = 0
              INCV = ( -4.0 + VINCH ) * 100.0
              KPS = 8
              CALL STRPLOT ( INCH, INCV, KPS )
8650          CONTINUE
C-----PRINT PLOT SCALES AND ROLL PAPER FOR NEXT PROBLEM
              MAXSTA = -VEND
              VAXIS = -VEND * H
              PRINT 810, MAXSTA, VAXIS, ( HAXIS(N), N = 1, 6 )
              CALL AXTERM ( 1 )
8700          CONTINUE
9970 CALL TIC TOC(4)
C-----RETURN FOR NEW PROBLEM
              GO TO 1010
9980 PRINT 980
9990          CONTINUE
              IF ( KSTOP ) 9995, 9995, 9992

```

```

01JE4
19JE4
23JE4
19JE4
19JE4
19JE4
19JE4
30JE4
19JE4
19JE4
19JE4
19JE4
23JE4
19JE4
19JE4
19JE4
19JE4
19JE4
19JE4
19JE4
23JE4
19JE4
19JE4
24JL4
24JL4
19JE4
19JE4
24JL4
02AP8
23JE4
19JE4
19JE4
19JE4
19JE4
14MR8
15MR8
14MR8
15MR8
19JE4
01JE4
01JE4
23JE4
21JL7
06JL4
06JL4
06JL4
06JL4
14MR8
01JE4
01JE4
19JE4
28FE8
21JL7
23MR4
26AP7
26AP7
26AP7
26AP7
28AP7
19JE4

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9992	CALL AXTERM (0)	21JL7
9995	CONTINUE	26AP7
9999	CONTINUE	26AP7
	PRINT 11	26AP7
	PRINT 1	26AP7
	PRINT 13, (AN1(N), N = 1, 32)	26AP7
	PRINT 19	26AP7
	END	26AP7

	SUBROUTINE INTERP4 (MP5, NCT, JN1, JN2, KR2, ZN, Z, LSM)	06MR8
	DIMENSION JN1(100), JN2(100), KR2(100), ZN(100), Z(205)	06MR8
905	FORMAT (//40H ERROR STOP -- STATIONS NOT IN ORDER)	14MY5
908	FORMAT (//43H UNDESIGNATED ERROR STOP IN SUBROUTINE)	14MY5
	DO 1603 J = 1, MP5	06MR8
	Z(J) = 0.0	28FE8
1603	CONTINUE	28FE8
	KR1 = 0	12JA5
	IF(1 .GT. NCT) GO TO 1676	01SE66
	DO 1675 NC = 1, NCT	26MR5
1605	IF (KR1) 1698, 1605, 1610	02JL6
	NC1 = NC	12JA5
	JV = JN1(NC1)	28JE6
	KSM = 0	28JE6
1610	IF (KR2(NC)) 1698, 1610, 1670	02JL6
	J1 = JV + 3	06MR8
	J2 = JN2(NC) + 3	06MR8
	JS = J1 + KSM	02JL6
	DENOM = J2 - J1	28JE6
	JINCR = 1	28JE6
	ESM = 1.0	28JE6
	ISW = 1 - LSM	28JE6
1620	IF (DENOM) 1695, 1620, 1630	12JA5
	DENOM = 1.0	12JA5
	ISW = 0	12JA5
1630	DO 1650 J = JS, J2, JINCR	28JE6
	DIFF = J - J1	28JE6
	PART = DIFF / DENOM	02JL6
	Z(J) = Z(J) + (ZN(NC1) + PART * (ZN(NC) - ZN(NC1)))	28JE6
	* ESM	28JE6
1650	CONTINUE	12JA5
	KSM = LSM	28JE6
1655	IF (ISW) 1698, 1660, 1655	28JE6
	JINCR = J2 - J1	12JA5
	ESM = -0.5	12JA5
	ISW = 0	12JA5
	GO TO 1630	12JA5
1660	IF (KR2(NC)) 1698, 1670, 1665	12JA5
1665	JV = JN2(NC)	28JE6
1670	KR1 = KR2(NC)	12JA5
	NC1 = NC	12JA5
1675	CONTINUE	12JA5
1676	RETURN	14MY5
1677	STOP 1677	26OC4
1695	PRINT 905	14MY5
	GO TO 1677	12JA5
1698	PRINT 908	14MY5
	GO TO 1677	12JA5
	END	04MA3 ID

```

SUBROUTINE TIC TOC (J)
10 FORMAT(///30X19HELAPSED CPU TIME = I5,8H MINUTESF9.3,8H SECONDS ) 240C66
11 FORMAT(///30X15HCOMPILE TIME = ,I5,8H MINUTES,F9.3,8H SECONDS ) 25SE66
12 FORMAT(///30X24HTIME FOR THIS PROBLEM = ,I5,8H MINUTES,F9.3, 25SE66
1   8H SECONDS ) 25SE66
      I = J - 2 240C66
      IF( I-1 ) 40,30,30 25SE66
30     FI4 = F 25SE66
40 CALL SECOND (F) 25SE66
      I11 = F 25SE66
      I1 = I11 / 60 25SE66
      FI2 = F - I1*60 25SE66
      IF( I ) 50,70,60 25SE66
50 PRINT 11, I1,FI2 25SE66
      GO TO 990 25SE66
60     FI3 = F - FI4 25SE66
      I2 = FI3 / 60 25SE66
      FI3 = FI3 - I2*60 25SE66
      PRINT 12, I2, FI3 25SE66
      IF( I-1 ) 990,990,70 25SE66
70 PRINT 10, I1,FI2 25SE66
990 CONTINUE 25SE66
      RETURN 25SE66
      END 25SE66
28FE8

```


APPENDIX 5

DATA LISTING AND CODING FORMS FOR EXAMPLE PROBLEMS

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SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

101A CANTILEVERED PRISMATIC BEAM WITH CONCENTRATED LOAD
1 1 2 0 3 1

36 1.000E 00
0 0.000E 00
0 36 3.000E 07 2.158E 02
1 36 1.100E 07 3.820E 00
0 36 -2.650E 00
0 0 1.000E 15
36 36 -2.200E 04

101B SAME AS PROBLEM 101A - ADD INFINITE SHEAR STIFFNESS
1 1 1 0 1 0 0 0 1 0 1
1 36 1.000E 99

102A CANTILEVERED SYMMETRICALLY TAPERED BEAM WITH CONCENTRATED LOAD
1 1 10 0 2 1

36 1.000E 00
0 0.000E 00
0 36 3.000E 06
1 36 1 1.270E 06 7.200E 01
0 36 1.270E 06 3.600E 01
0 1 8.640E 02
6 1 6.660E 02
12 1 5.000E 02
18 1 3.640E 02
24 1 2.560E 02
30 1 1.720E 02
36 1.080E 02
36 36 -7.000E 03
0 0 1.000E 15

102B SAME AS PROBLEM 102A - ADD INFINITE SHEAR STIFFNESS
1 1 1 0 1 0 0 0 1 0 1
1 36 1.000E 99

102C SAME AS PROBLEM 102A WITH MODIFIED SHEAR AREA FOR K = 1.2
1 1 1 0 1 0 0 2 0 0 1
1 1 1 -1.200E 01
36 0 -0.600E 01

```

201A    STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD ON OUTSIDE BEAM
        1    0    2    0    8    1
96      3.000E 00
0 96    3.000E 07      3.126E 02
1 96    1.100E 07      7.820E 00
0 0     9.700E 05      1.840E 05
24 24   9.700E 05      1.840E 05
48 48   9.700E 05      1.840E 05
72 72   9.700E 05      1.840E 05
96 96   9.700E 05      1.840E 05
0 96    -8.480E 00
72 72   -2.080E 04
96 96   -2.080E 04
201B    SAME AS PROBLEM 201A - ADD INFINITE SHEAR STIFFNESS
1 0 1 0 1 0 0 0 0 1 0 1
1 96 1.000E 99
202A    STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD ON INTERIOR BEAMS
1 0 1 0 1 0 0 0 0 2 1
96 96 2.080E 04
48 48 -2.080E 04
202B    SAME AS PROBLEM 202A - ADD INFINITE SHEAR STIFFNESS
1 0 1 0 1 0 0 0 0 1 0 1
1 96 1.000E 99
203A    STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD AT CENTER OF DIAPHRAGM
1 0 1 0 1 0 0 0 0 0 4 1
48 48 2.080E 04
72 72 2.080E 04
36 36 -2.080E 04
60 60 -2.080E 04
203B    SAME AS PROBLEM 203A - ADD INFINITE SHEAR STIFFNESS
1 0 1 0 1 0 0 0 0 1 0 1
1 96 1.000E 99

```

301A TAPERED THREE-SPAN OVERHANGING BEAM WITH 300 LB LOAD AT STA 0
 1 2 22 0 1 1

96			1.000E 00						
60			0.000E 00						
36			0.000E 00						
0	96	0	1.000E 07						
1	96	0		4.000E 06					
0	0	0			5.546E-01				
1		1			7.421E-01	7.200E-02			
	3	1			1.193E 00	8.740E-02			
	6	1			2.034E 00	1.090E-01			
	10	1			3.395E 00	1.360E-01			
	14	1			4.937E 00	1.600E-01			
	18	1			6.566E 00	1.810E-01			
	23	1			8.592E 00	2.040E-01			
	28	1			1.047E 01	2.220E-01			
	35	1			1.260E 01	2.420E-01			
	48	1			1.429E 01	2.560E-01			
	61	1			1.260E 01	2.420E-01			
	68	1			1.047E 01	2.220E-01			
	73	1			8.592E 00	2.040E-01			
	78	1			6.566E 00	1.810E-01			
	82	1			4.937E 00	1.600E-01			
	86	1			3.395E 00	1.360E-01			
	90	1			2.034E 00	1.090E-01			
	93	1			1.193E 00	8.740E-02			
	96				5.546E-01	6.440E-02			

0 0 3.000E 02

301B SAME AS PROBLEM 301A - ADD INFINITE SHEAR STIFFNESS

1 1 1 0 1 0 0 0 1 0 1

1 96 1.000E 99

302A TAPERED THREE-SPAN OVERHANGING BEAM WITH 500 LB LOAD AT STA 0

1 1 1 0 0 0 0 0 1 1

0 0 5.000E 02

302B SAME AS PROBLEM 302A - ADD INFINITE SHEAR STIFFNESS

1 1 1 0 1 0 0 0 1 0 1

1 96 1.000E 99

303A TAPERED THREE-SPAN OVERHANGING BEAM WITH 700 LB LOAD AT STA 0

1 1 1 0 0 0 0 0 1 1

0 0 7.000E 02

303B SAME AS PROBLEM 303A - ADD INFINITE SHEAR STIFFNESS

1 1 1 0 1 0 0 0 1 0 1

1 96 1.000E 99

IDENTIFICATION Example problems CODED BY D.F.T. DATE 18 April 68 PAGE 1 of 7

1	5	10	15	20	25	30	35	40	45	50	55	60	65	70	75	80
GE2440	10	MASTERS THESIS D.E. TAHKERSLEY - 15APR														
SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SUBRM 1 - APPENDIX 6																
101A	CANTILEVERED PRISMATIC BEAM WITH CONCENTRATED LOAD															
							1	1	2	0	3				1	
	36									1.000E 00						
	0									0.000E 00						
	0	36								3.000E 07				2.158E 02		
	1	36								1.100E 07				3.820E 00		
	0	36								-2.650E 00						
	0	0												1.000E 15		
	36	36								-2.200E 04						
101B	SAME AS PROBLEM 101A - ADD INFINITE SHEAR STIFFNESS															
	1	1	1	0	1				0	0	0		1	0		1
	1	36								1.000E 99						
102A	CANTILEVERED SYMMETRICALLY TAPERED BEAM WITH CONCENTRATED LOAD															
								1	1	10	0	2			1	

IDENTIFICATION Example problems CODED BY D.F.T. DATE 18 April 68 PAGE 2 of 7

1	5	10	15	20	25	30	35	40	45	50	55	60	65	70	75	80
		36								1.000E 00						
		0								0.000E 00						
		0	36							3.000E 06						
										1.270E 06					7.200E 01	
			36							1.270E 06					3.600E 01	
			0												8.640E 02	
					6										6.660E 02	
					12										5.000E 02	
					18										3.640E 02	
					24										2.560E 02	
					30										1.720E 02	
					36										1.080E 02	
			36	36											-7.000E 03	
			0	0											1.000E 15	
102B	SAME AS PROBLEM 102A - ADD INFINITE SHEAR STIFFNESS															
	1	1	1	0	1					0	0	0		1	0	1

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APPENDIX 6

COMPUTER RESULTS FOR EXAMPLE PROBLEMS

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SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
101A CANTILEVERED PRISMATIC BEAM WITH CONCENTRATED LOAD

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	-0	-0	-0	-0	-0
NUM CARDS INPUT THIS PROBLEM	1	1	2	0	3
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

NUM INCREMENTS	36
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
0	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
ADDITIONAL DATA INPUT FOR THIS PROBLEM						
0	36	-0	3.000E+07	-0.	2.158E+02	-0.
1	36	-0	-0.	1.100E+07	-0.	3.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
NONE				

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
ADDITIONAL DATA INPUT FOR THIS PROBLEM							
0	36	-0	-2.650E+00	-0.	-0.	-0.	-0.
0	0	-0	-0.	-0.	-0.	1.000E+15	-0.
36	36	-0	-2.200E+04	-0.	-0.	-0.	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)

101A CANTILEVERED PRISMATIC BEAM WITH CONCENTRATED LOAD

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	-6.130E-05	0.	6.130E-05	0.	0.	0.
0	0.	5.258E-04	-6.130E-05	-3.969E+05	2.209E+04	2.210E+04
1	-5.871E-04	5.257E-04	-1.805E-04	-7.716E+05	2.209E+04	0.
2	-1.293E-03	5.257E-04	-2.963E-04	-7.495E+05	2.209E+04	0.
3	-2.115E-03	5.256E-04	-4.086E-04	-7.274E+05	2.209E+04	0.
4	-3.050E-03	5.255E-04	-5.176E-04	-7.054E+05	2.208E+04	0.
5	-4.093E-03	5.255E-04	-6.231E-04	-6.833E+05	2.208E+04	0.
6	-5.241E-03	5.254E-04	-7.253E-04	-6.612E+05	2.208E+04	0.
7	-6.492E-03	5.254E-04	-8.240E-04	-6.391E+05	2.208E+04	0.
8	-7.841E-03	5.253E-04	-9.193E-04	-6.170E+05	2.207E+04	0.
9	-9.286E-03	5.252E-04	-1.011E-03	-5.950E+05	2.207E+04	0.
10	-1.082E-02	5.252E-04	-1.100E-03	-5.729E+05	2.207E+04	0.
11	-1.245E-02	5.251E-04	-1.185E-03	-5.508E+05	2.206E+04	0.
12	-1.416E-02	5.250E-04	-1.266E-03	-5.288E+05	2.206E+04	0.
13	-1.595E-02	5.250E-04	-1.345E-03	-5.067E+05	2.206E+04	0.
14	-1.782E-02	5.249E-04	-1.420E-03	-4.846E+05	2.206E+04	0.
15	-1.976E-02	5.249E-04	-1.491E-03	-4.626E+05	2.205E+04	0.
16	-2.178E-02	5.248E-04	-1.559E-03	-4.405E+05	2.205E+04	0.
17	-2.386E-02	5.247E-04	-1.624E-03	-4.185E+05	2.205E+04	0.
18	-2.601E-02	5.247E-04	-1.685E-03	-3.964E+05	2.205E+04	0.
19	-2.822E-02	5.246E-04	-1.743E-03	-3.744E+05	2.204E+04	0.
20	-3.049E-02	5.245E-04	-1.797E-03	-3.523E+05	2.204E+04	0.
21	-3.281E-02	5.245E-04	-1.848E-03	-3.303E+05	2.204E+04	0.

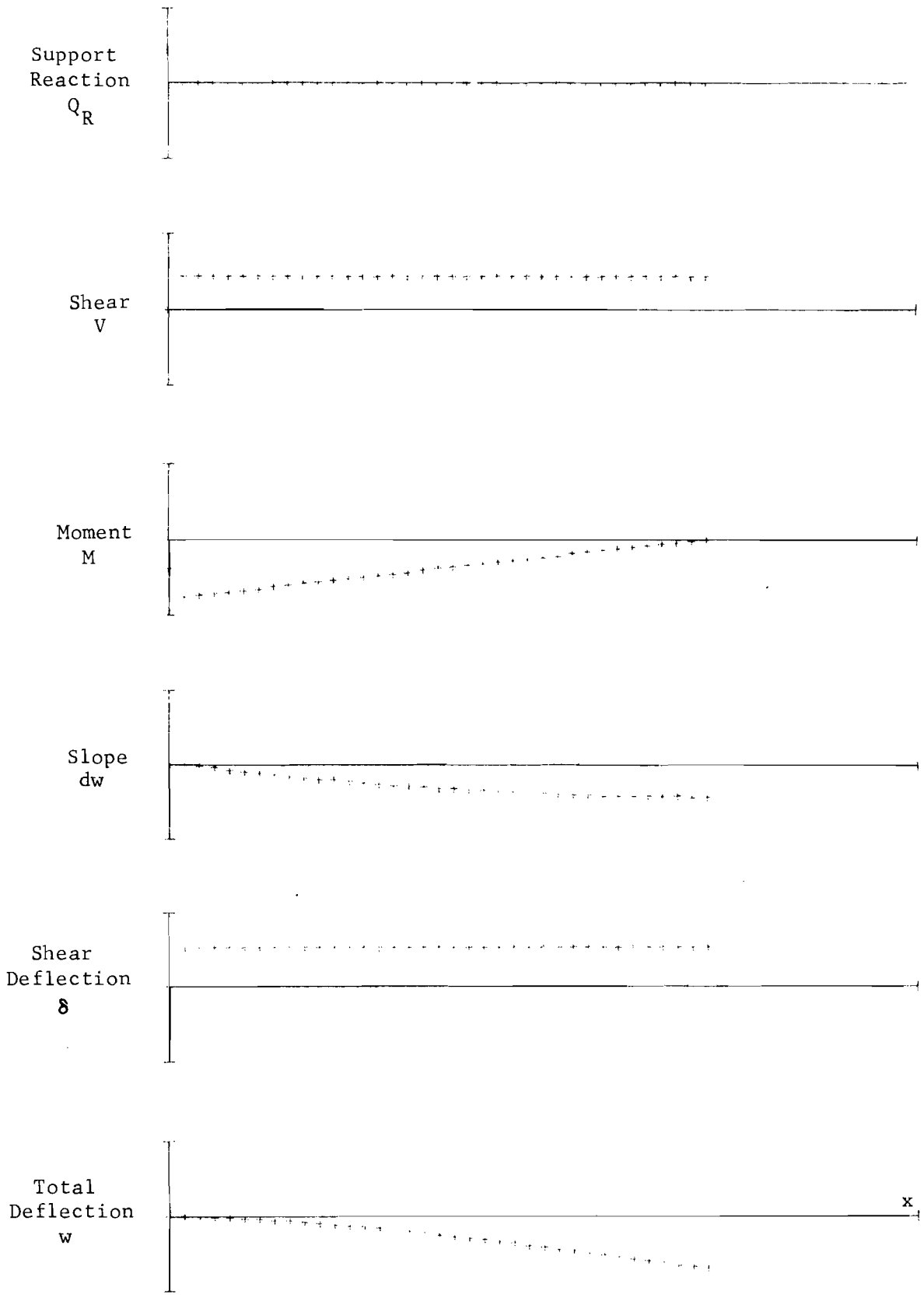
22	-3.518E-02			-3.083E+05		0.
		5.244E-04	-1.896E-03		2.204E+04	0.
23	-3.760E-02			-2.862E+05		0.
		5.243E-04	-1.940E-03		2.203E+04	0.
24	-4.007E-02			-2.642E+05		0.
		5.243E-04	-1.981E-03		2.203E+04	0.
25	-4.257E-02			-2.422E+05		0.
		5.242E-04	-2.018E-03		2.203E+04	0.
26	-4.511E-02			-2.201E+05		0.
		5.242E-04	-2.052E-03		2.203E+04	0.
27	-4.769E-02			-1.981E+05		0.
		5.241E-04	-2.083E-03		2.202E+04	0.
28	-5.030E-02			-1.761E+05		0.
		5.240E-04	-2.110E-03		2.202E+04	0.
29	-5.293E-02			-1.541E+05		0.
		5.240E-04	-2.134E-03		2.202E+04	0.
30	-5.539E-02			-1.320E+05		0.
		5.239E-04	-2.154E-03		2.201E+04	0.
31	-5.827E-02			-1.100E+05		0.
		5.238E-04	-2.171E-03		2.201E+04	0.
32	-6.096E-02			-8.802E+04		0.
		5.238E-04	-2.185E-03		2.201E+04	0.
33	-6.367E-02			-6.601E+04		0.
		5.237E-04	-2.195E-03		2.201E+04	0.
34	-6.639E-02			-4.401E+04		0.
		5.237E-04	-2.202E-03		2.200E+04	0.
35	-6.912E-02			-2.200E+04		0.
		5.236E-04	-2.205E-03		2.200E+04	0.
36	-7.184E-02			-6.264E-07		0.
		-2.690E-18	-2.205E-03		-1.130E-10	0.
37	-7.405E-02			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
50	5.000E+01	1.000E-01	1.000E-03	5.000E-03	1.000E+06	5.000E+04	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 2.696 SECONDS

ELAPSED CPU TIME = 0 MINUTES 10.586 SECONDS



Prob 101A. Cantilevered prismatic beam with concentrated load.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
101B SAME AS PROBLEM 101A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	36
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
0	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	36	-0	3.000E+07	-0.	2.158E+02	-0.
1	36	-0	-0.	1.100E+07	-0.	3.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
ADDITIONAL DATA INPUT FOR THIS PROBLEM				
1	36	-0	1.000E+99	-0.

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
DATA HELD FROM THE PREVIOUS PROBLEM							
0	36	-0	-2.650E+00	-0.	-0.	-0.	-0.
0	0	-0	-0.	-0.	-0.	1.000E+15	-0.
36	36	-0	-2.200E+04	-0.	-0.	-0.	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
 1018 SAME AS PROBLEM 101A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	-6.130E-05			0.		0.
0	0.	0.	6.130E-05		0.	
		2.209E-95	-6.130E-05	-3.969E+05	2.209E+04	2.210E+04
1	-6.130E-05	2.209E-95	-1.805E-04	-7.716E+05	2.209E+04	0.
2	-2.418E-04	2.209E-95	-2.963E-04	-7.495E+05	2.209E+04	0.
3	-5.381E-04	2.209E-95	-4.086E-04	-7.274E+05	2.209E+04	0.
4	-9.467E-04	2.208E-95	-5.176E-04	-7.054E+05	2.208E+04	0.
5	-1.464E-03	2.208E-95	-6.231E-04	-6.833E+05	2.208E+04	0.
6	-2.087E-03	2.208E-95	-7.253E-04	-6.612E+05	2.208E+04	0.
7	-2.813E-03	2.208E-95	-8.240E-04	-6.391E+05	2.208E+04	0.
8	-3.637E-03	2.207E-95	-9.193E-04	-6.170E+05	2.207E+04	0.
9	-4.556E-03	2.207E-95	-1.011E-03	-5.950E+05	2.207E+04	0.
10	-5.567E-03	2.207E-95	-1.100E-03	-5.729E+05	2.207E+04	0.
11	-6.667E-03	2.206E-95	-1.185E-03	-5.508E+05	2.206E+04	0.
12	-7.852E-03	2.206E-95	-1.266E-03	-5.288E+05	2.206E+04	0.
13	-9.118E-03	2.206E-95	-1.345E-03	-5.067E+05	2.206E+04	0.
14	-1.046E-02	2.206E-95	-1.420E-03	-4.846E+05	2.206E+04	0.
15	-1.188E-02	2.205E-95	-1.491E-03	-4.626E+05	2.205E+04	0.
16	-1.337E-02	2.205E-95	-1.559E-03	-4.405E+05	2.205E+04	0.
17	-1.493E-02	2.205E-95	-1.624E-03	-4.185E+05	2.205E+04	0.
18	-1.656E-02	2.205E-95	-1.685E-03	-3.964E+05	2.205E+04	0.
19	-1.824E-02	2.204E-95	-1.743E-03	-3.744E+05	2.204E+04	0.
20	-1.998E-02	2.204E-95	-1.797E-03	-3.523E+05	2.204E+04	0.
21	-2.178E-02	2.204E-95	-1.848E-03	-3.303E+05	2.204E+04	0.

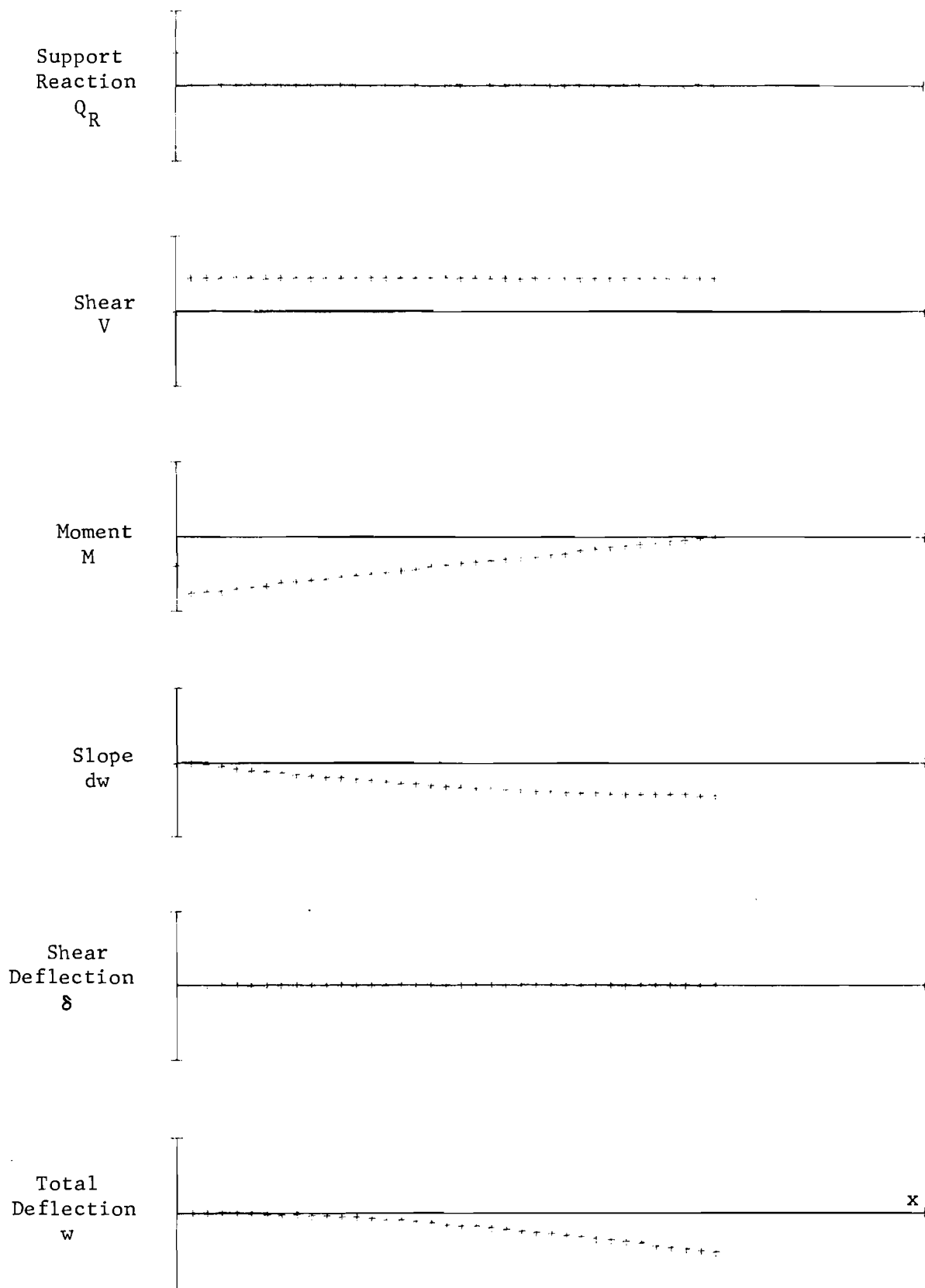
22	-2.363E-02			-3.083E+05		0.
		2.204E-95	-1.896E-03		2.204E+04	
23	-2.552E-02			-2.862E+05		0.
		2.203E-95	-1.940E-03		2.203E+04	
24	-2.746E-02			-2.642E+05		0.
		2.203E-95	-1.981E-03		2.203E+04	
25	-2.945E-02			-2.422E+05		0.
		2.203E-95	-2.018E-03		2.203E+04	
26	-3.146E-02			-2.201E+05		0.
		2.203E-95	-2.052E-03		2.203E+04	
27	-3.352E-02			-1.981E+05		0.
		2.202E-95	-2.083E-03		2.202E+04	
28	-3.560E-02			-1.761E+05		0.
		2.202E-95	-2.110E-03		2.202E+04	
29	-3.771E-02			-1.541E+05		0.
		2.202E-95	-2.134E-03		2.202E+04	
30	-3.984E-02			-1.320E+05		0.
		2.201E-95	-2.154E-03		2.201E+04	
31	-4.200E-02			-1.100E+05		0.
		2.201E-95	-2.171E-03		2.201E+04	
32	-4.417E-02			-8.802E+04		0.
		2.201E-95	-2.185E-03		2.201E+04	
33	-4.635E-02			-6.601E+04		0.
		2.201E-95	-2.195E-03		2.201E+04	
34	-4.855E-02			-4.401E+04		0.
		2.200E-95	-2.202E-03		2.200E+04	
35	-5.075E-02			-2.200E+04		0.
		2.200E-95	-2.205E-03		2.200E+04	
36	-5.296E-02			-7.122E-86		0.
		-7.559E-110	-2.205E-03		-7.559E-11	
37	-5.516E-02			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
50	5.000E+01	1.000E-01	1.000E+00	5.000E-03	1.000E+06	5.000E+04	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 2.282 SECONDS

ELAPSED CPU TIME = 0 MINUTES 12.868 SECONDS



Prob 101B. Same as Prob 101A, neglecting shear deformations.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
102A CANTILEVERED SYMMETRICALLY TAPERED BEAM WITH CONCENTRATED LOAD

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	-0	-0	-0	-0	-0
NUM CARDS INPUT THIS PROBLEM	1	1	10	0	2
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

NUM INCREMENTS	36
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
0	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
ADDITIONAL DATA INPUT FOR THIS PROBLEM						
0	36	-0	3.000E+06	-0.	-0.	-0.
1		1	-0.	1.270E+06	-0.	7.200E+01
	36	-0	-0.	1.270E+06	-0.	3.600E+01
0		1	-0.	-0.	8.640E+02	-0.
	6	1	-0.	-0.	6.660E+02	-0.
	12	1	-0.	-0.	5.000E+02	-0.
	18	1	-0.	-0.	3.640E+02	-0.
	24	1	-0.	-0.	2.560E+02	-0.
	30	1	-0.	-0.	1.720E+02	-0.
	36	-0	-0.	-0.	1.080E+02	-0.

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
NONE				

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
ADDITIONAL DATA INPUT FOR THIS PROBLEM							
36	36	-0	-7.000E+03	-0.	-0.	-0.	-0.
0	0	-0	-0.	-0.	-0.	1.000E+15	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
102A CANTILEVERED SYMMETRICALLY TAPERED BEAM WITH CONCENTRATED LOAD

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	-4.861E-05			0.		0.
0	0.	0.	4.861E-05	-1.260E+05	0.	7.000E+03
1	-1.252E-04	7.655E-05	-4.861E-05	-2.450E+05	7.000E+03	0.
2	-3.497E-04	7.766E-05	-1.469E-04	-2.380E+05	7.000E+03	0.
3	-6.748E-04	7.880E-05	-2.463E-04	-2.310E+05	7.000E+03	0.
4	-1.102E-03	7.998E-05	-3.470E-04	-2.240E+05	7.000E+03	0.
5	-1.632E-03	8.119E-05	-4.490E-04	-2.170E+05	7.000E+03	0.
6	-2.267E-03	8.244E-05	-5.524E-04	-2.100E+05	7.000E+03	0.
7	-3.008E-03	8.373E-05	-6.575E-04	-2.030E+05	7.000E+03	0.
8	-3.857E-03	8.506E-05	-7.636E-04	-1.960E+05	7.000E+03	0.
9	-4.814E-03	8.643E-05	-8.705E-04	-1.890E+05	7.000E+03	0.
10	-5.880E-03	8.785E-05	-9.786E-04	-1.820E+05	7.000E+03	0.
11	-7.057E-03	8.931E-05	-1.088E-03	-1.750E+05	7.000E+03	0.
12	-8.346E-03	9.083E-05	-1.198E-03	-1.680E+05	7.000E+03	0.
13	-9.749E-03	9.239E-05	-1.310E-03	-1.610E+05	7.000E+03	0.
14	-1.127E-02	9.401E-05	-1.423E-03	-1.540E+05	7.000E+03	0.
15	-1.290E-02	9.569E-05	-1.536E-03	-1.470E+05	7.000E+03	0.
16	-1.464E-02	9.743E-05	-1.649E-03	-1.400E+05	7.000E+03	0.
17	-1.651E-02	9.924E-05	-1.763E-03	-1.330E+05	7.000E+03	0.
18	-1.849E-02	1.011E-04	-1.878E-03	-1.260E+05	7.000E+03	0.
19	-2.058E-02	1.031E-04	-1.993E-03	-1.190E+05	7.000E+03	0.
20	-2.279E-02	1.051E-04	-2.108E-03	-1.120E+05	7.000E+03	0.
21	-2.512E-02	1.072E-04	-2.222E-03	-1.050E+05	7.000E+03	0.
		1.094E-04	-2.335E-03		7.000E+03	

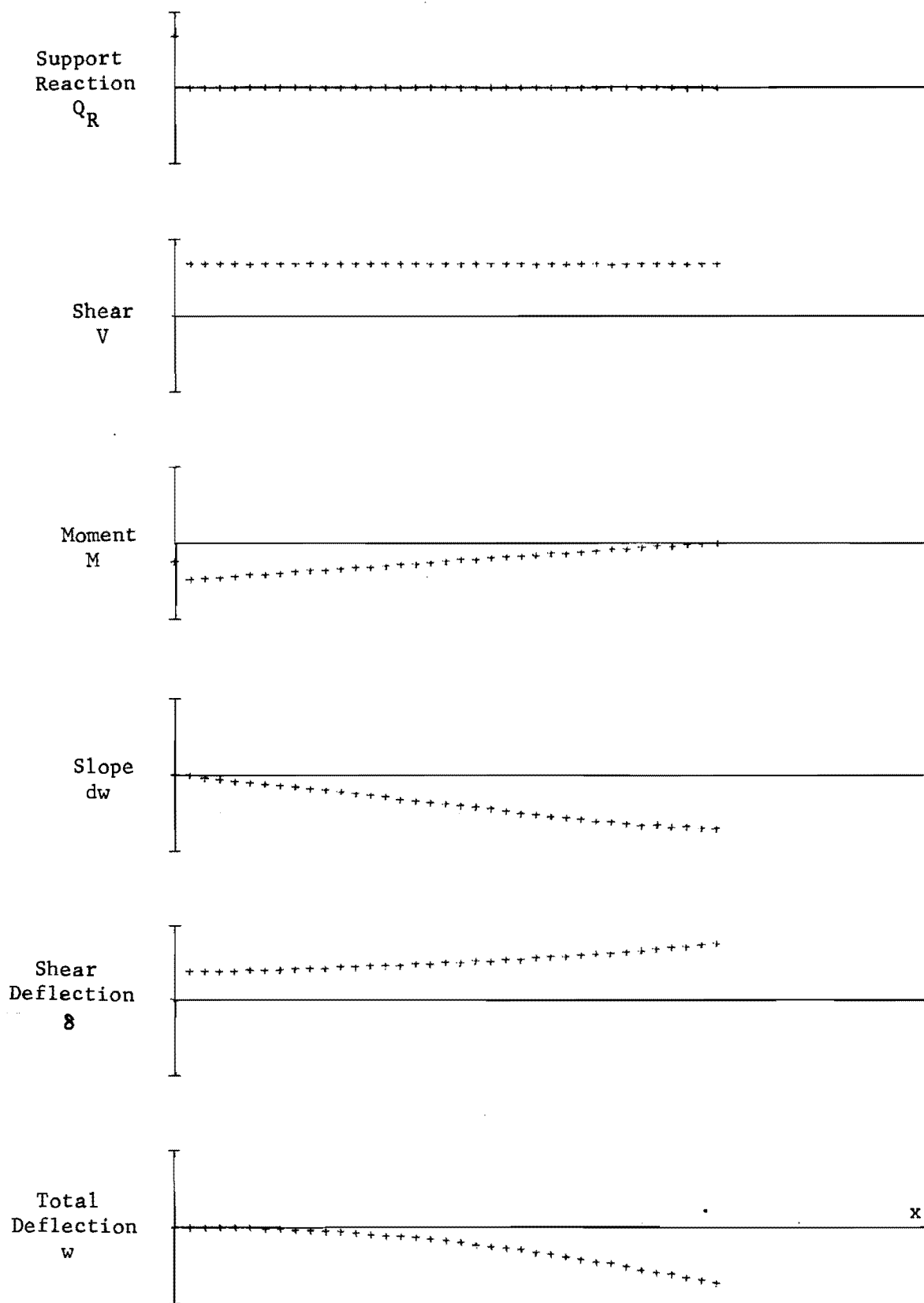
22	-2.757E-02			-9.800E+04		0.
23	-3.013E-02	1.116E-04	-2.446E-03	-9.100E+04	7.000E+03	0.
24	-3.280E-02	1.140E-04	-2.557E-03	-8.400E+04	7.000E+03	0.
25	-3.558E-02	1.165E-04	-2.667E-03	-7.700E+04	7.000E+03	0.
26	-3.847E-02	1.191E-04	-2.773E-03	-7.000E+04	7.000E+03	0.
27	-4.147E-02	1.218E-04	-2.875E-03	-6.300E+04	7.000E+03	0.
28	-4.457E-02	1.246E-04	-2.973E-03	-5.600E+04	7.000E+03	0.
29	-4.776E-02	1.276E-04	-3.066E-03	-4.900E+04	7.000E+03	0.
30	-5.104E-02	1.307E-04	-3.154E-03	-4.200E+04	7.000E+03	0.
31	-5.441E-02	1.340E-04	-3.236E-03	-3.500E+04	7.000E+03	0.
32	-5.786E-02	1.374E-04	-3.308E-03	-2.800E+04	7.000E+03	0.
33	-6.137E-02	1.410E-04	-3.370E-03	-2.100E+04	7.000E+03	0.
34	-6.493E-02	1.448E-04	-3.420E-03	-1.400E+04	7.000E+03	0.
35	-6.854E-02	1.489E-04	-3.456E-03	-7.000E+03	7.000E+03	0.
36	-7.217E-02	1.531E-04	-3.476E-03	-1.179E-07	7.000E+03	0.
37	-7.564E-02	-9.718E-19	-3.476E-03	0.	-4.443E-11	0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
50	5.000E+01	1.000E-01	2.000E-04	5.000E-03	5.000E+05	1.000E+04	1.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 2.337 SECONDS

ELAPSED CPU TIME = 0 MINUTES 15.205 SECONDS



Prob 102A. Cantilevered symmetrically tapered beam with concentrated load.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
102B SAME AS PROBLEM 102A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	36
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
0	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	36	-0	3.000E+06	-0.	-0.	-0.
1		1	-0.	1.270E+06	-0.	7.200E+01
	36	-0	-0.	1.270E+06	-0.	3.600E+01
0		1	-0.	-0.	8.640E+02	-0.
	6	1	-0.	-0.	6.660E+02	-0.
	12	1	-0.	-0.	5.000E+02	-0.
	18	1	-0.	-0.	3.640E+02	-0.
	24	1	-0.	-0.	2.560E+02	-0.
	30	1	-0.	-0.	1.720E+02	-0.
	36	-0	-0.	-0.	1.080E+02	-0.

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
------	----	-------	----	---

ADDITIONAL DATA INPUT FOR THIS PROBLEM
1 36 -0 1.000E+99 -0.

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
DATA HELD FROM THE PREVIOUS PROBLEM							
36	36	-0	-7.000E+03	-0.	-0.	-0.	-0.
0	0	-0	-0.	-0.	-0.	1.000E+15	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)

102B SAME AS PROBLEM 102A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	-4.861E-05			0.		0.
0	0.	0.	4.861E-05	-1.260E+05	0.	7.000E+03
1	-4.861E-05	7.000E-96	-4.861E-05	-2.450E+05	7.000E+03	0.
2	-1.955E-04	7.000E-96	-1.469E-04	-2.380E+05	7.000E+03	0.
3	-4.418E-04	7.000E-96	-2.463E-04	-2.310E+05	7.000E+03	0.
4	-7.888E-04	7.000E-96	-3.470E-04	-2.240E+05	7.000E+03	0.
5	-1.238E-03	7.000E-96	-4.490E-04	-2.170E+05	7.000E+03	0.
6	-1.790E-03	7.000E-96	-5.524E-04	-2.100E+05	7.000E+03	0.
7	-2.448E-03	7.000E-96	-6.575E-04	-2.030E+05	7.000E+03	0.
8	-3.211E-03	7.000E-96	-7.636E-04	-1.960E+05	7.000E+03	0.
9	-4.082E-03	7.000E-96	-8.705E-04	-1.890E+05	7.000E+03	0.
10	-5.060E-03	7.000E-96	-9.786E-04	-1.820E+05	7.000E+03	0.
11	-6.148E-03	7.000E-96	-1.088E-03	-1.750E+05	7.000E+03	0.
12	-7.347E-03	7.000E-96	-1.198E-03	-1.680E+05	7.000E+03	0.
13	-8.657E-03	7.000E-96	-1.310E-03	-1.610E+05	7.000E+03	0.
14	-1.008E-02	7.000E-96	-1.423E-03	-1.540E+05	7.000E+03	0.
15	-1.162E-02	7.000E-96	-1.536E-03	-1.470E+05	7.000E+03	0.
16	-1.326E-02	7.000E-96	-1.649E-03	-1.400E+05	7.000E+03	0.
17	-1.503E-02	7.000E-96	-1.763E-03	-1.330E+05	7.000E+03	0.
18	-1.691E-02	7.000E-96	-1.878E-03	-1.260E+05	7.000E+03	0.
19	-1.890E-02	7.000E-96	-1.993E-03	-1.190E+05	7.000E+03	0.
20	-2.101E-02	7.000E-96	-2.108E-03	-1.120E+05	7.000E+03	0.
21	-2.323E-02	7.000E-96	-2.222E-03	-1.050E+05	7.000E+03	0.
		7.000E-96	-2.335E-03		7.000E+03	

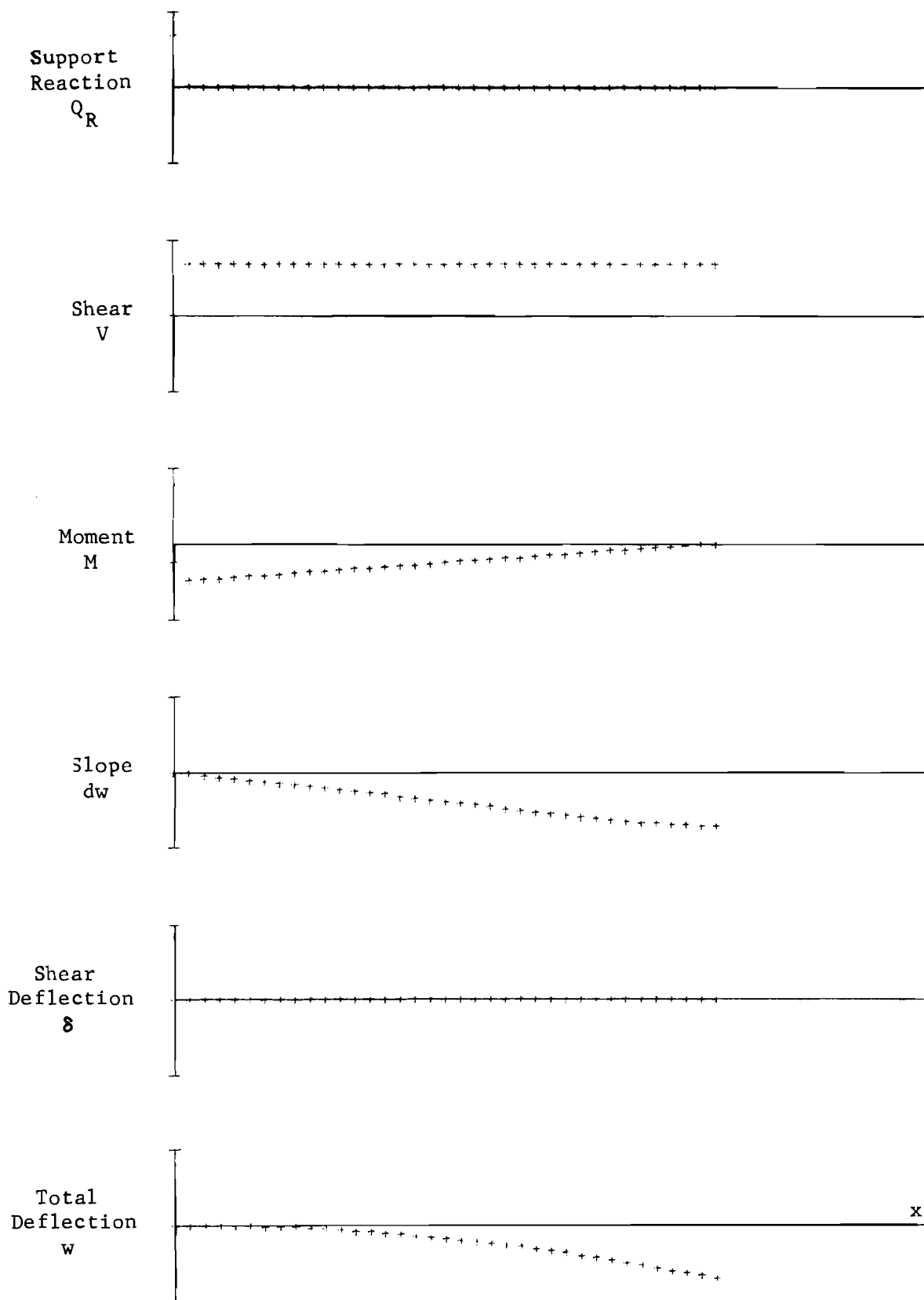
22	-2.556E-02			-9.800E+04		0.
		7.000E-96	-2.446E-03		7.000E+03	0.
23	-2.801E-02			-9.100E+04		0.
		7.000E-96	-2.557E-03		7.000E+03	0.
24	-3.057E-02			-8.400E+04		0.
		7.000E-96	-2.667E-03		7.000E+03	0.
25	-3.323E-02			-7.700E+04		0.
		7.000E-96	-2.773E-03		7.000E+03	0.
26	-3.601E-02			-7.000E+04		0.
		7.000E-96	-2.875E-03		7.000E+03	0.
27	-3.888E-02			-6.300E+04		0.
		7.000E-96	-2.973E-03		7.000E+03	0.
28	-4.185E-02			-5.600E+04		0.
		7.000E-96	-3.066E-03		7.000E+03	0.
29	-4.492E-02			-4.900E+04		0.
		7.000E-96	-3.154E-03		7.000E+03	0.
30	-4.807E-02			-4.200E+04		0.
		7.000E-96	-3.236E-03		7.000E+03	0.
31	-5.131E-02			-3.500E+04		0.
		7.000E-96	-3.308E-03		7.000E+03	0.
32	-5.462E-02			-2.800E+04		0.
		7.000E-96	-3.370E-03		7.000E+03	0.
33	-5.799E-02			-2.100E+04		0.
		7.000E-96	-3.420E-03		7.000E+03	0.
34	-6.141E-02			-1.400E+04		0.
		7.000E-96	-3.456E-03		7.000E+03	0.
35	-6.486E-02			-7.000E+03		0.
		7.000E-96	-3.476E-03		7.000E+03	0.
36	-6.834E-02			7.194E-08		0.
		-4.677E-110	-3.476E-03		-4.677E-11	0.
37	-7.181E-02			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
50	5.000E+01	1.000E-01	1.000E+00	5.000E-03	5.000E+05	1.000E+04	1.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 2.303 SECONDS

ELAPSED CPU TIME = 0 MINUTES 17.508 SECONDS



Prob 102B. Same as Prob 102A, neglecting shear deformations.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
102C SAME AS PROBLEM 102A WITH MODIFIED SHEAR AREA FOR K = 1.2

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	2	0	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	36
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
0	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	36	-0	3.000E+06	-0.	-0.	-0.
1	36	1	-0.	1.270E+06	-0.	7.200E+01
0	0	1	-0.	1.270E+06	-0.	3.600E+01
	6	1	-0.	-0.	8.640E+02	-0.
	12	1	-0.	-0.	6.660E+02	-0.
	18	1	-0.	-0.	5.000E+02	-0.
	24	1	-0.	-0.	3.640E+02	-0.
	30	1	-0.	-0.	2.560E+02	-0.
	36	-0	-0.	-0.	1.720E+02	-0.
					1.080E+02	-0.
ADDITIONAL DATA INPUT FOR THIS PROBLEM						
1	1	1	-0.	-0.	-0.	-1.200E+01
	36	0	-0.	-0.	-0.	-6.000E+00

TABLE 4B - STIFFNESS DATA

FROM TO CONTD SK F

NONE

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
DATA HELD FROM THE PREVIOUS PROBLEM							
36	36	-0	-7.000E+03	-0.	-0.	-0.	-0.
0	0	-0	-0.	-0.	-0.	1.000E+15	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)

102C SAME AS PROBLEM 102A WITH MODIFIED SHEAR AREA FOR K = 1.2

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	-4.861E-05			0.		0.
0	0.	0.	4.861E-05	-1.260E+05	0.	7.000E+03
1	-1.405E-04	9.186E-05	-4.861E-05	-2.450E+05	7.000E+03	0.
2	-3.806E-04	9.319E-05	-1.469E-04	-2.380E+05	7.000E+03	0.
3	-7.214E-04	9.457E-05	-2.463E-04	-2.310E+05	7.000E+03	0.
4	-1.164E-03	9.598E-05	-3.470E-04	-2.240E+05	7.000E+03	0.
5	-1.711E-03	9.743E-05	-4.490E-04	-2.170E+05	7.000E+03	0.
6	-2.362E-03	9.893E-05	-5.524E-04	-2.100E+05	7.000E+03	0.
7	-3.120E-03	1.005E-04	-6.575E-04	-2.030E+05	7.000E+03	0.
8	-3.986E-03	1.021E-04	-7.636E-04	-1.960E+05	7.000E+03	0.
9	-4.960E-03	1.037E-04	-8.705E-04	-1.960E+05	7.000E+03	0.
10	-6.044E-03	1.054E-04	-9.786E-04	-1.820E+05	7.000E+03	0.
11	-7.239E-03	1.072E-04	-1.088E-03	-1.750E+05	7.000E+03	0.
12	-8.546E-03	1.090E-04	-1.198E-03	-1.680E+05	7.000E+03	0.
13	-9.968E-03	1.109E-04	-1.310E-03	-1.610E+05	7.000E+03	0.
14	-1.150E-02	1.128E-04	-1.423E-03	-1.540E+05	7.000E+03	0.
15	-1.315E-02	1.148E-04	-1.536E-03	-1.470E+05	7.000E+03	0.
16	-1.492E-02	1.169E-04	-1.649E-03	-1.400E+05	7.000E+03	0.
17	-1.680E-02	1.191E-04	-1.763E-03	-1.330E+05	7.000E+03	0.
18	-1.880E-02	1.213E-04	-1.878E-03	-1.260E+05	7.000E+03	0.
19	-2.092E-02	1.237E-04	-1.993E-03	-1.260E+05	7.000E+03	0.
20	-2.315E-02	1.261E-04	-2.108E-03	-1.190E+05	7.000E+03	0.
21	-2.550E-02	1.286E-04	-2.222E-03	-1.120E+05	7.000E+03	0.
		1.312E-04	-2.335E-03	-1.050E+05	7.000E+03	0.

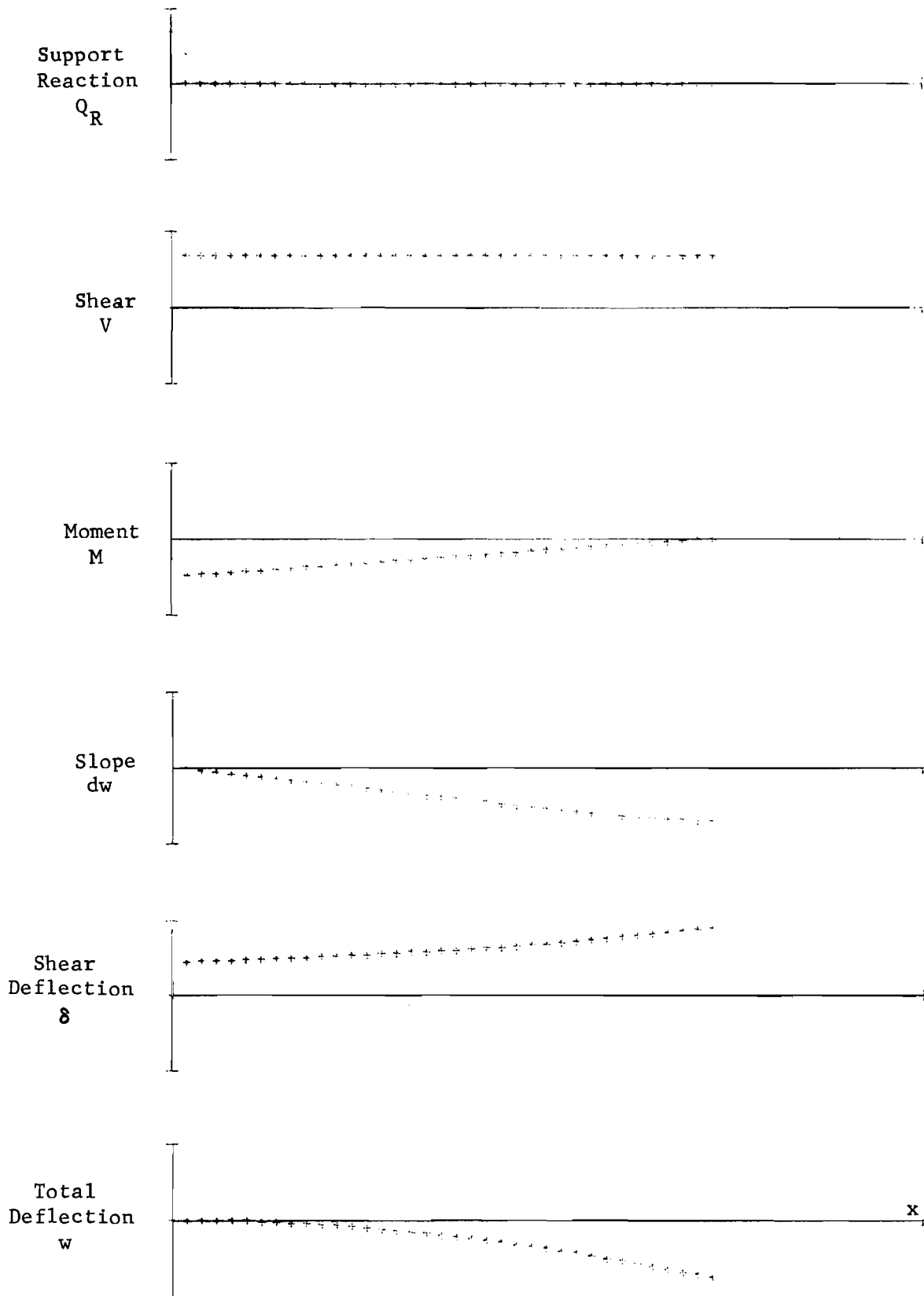
22	-2.797E-02			-9.800E+04		0.
		1.340E-04	-2.446E-03		7.000E+03	
23	-3.055E-02			-9.100E+04		0.
		1.368E-04	-2.557E-03		7.000E+03	
24	-3.324E-02			-8.400E+04		0.
		1.398E-04	-2.667E-03		7.000E+03	
25	-3.605E-02			-7.700E+04		0.
		1.429E-04	-2.773E-03		7.000E+03	
26	-3.896E-02			-7.000E+04		0.
		1.461E-04	-2.875E-03		7.000E+03	
27	-4.199E-02			-6.300E+04		0.
		1.495E-04	-2.973E-03		7.000E+03	
28	-4.511E-02			-5.600E+04		0.
		1.531E-04	-3.066E-03		7.000E+03	
29	-4.833E-02			-4.900E+04		0.
		1.568E-04	-3.154E-03		7.000E+03	
30	-5.164E-02			-4.200E+04		0.
		1.608E-04	-3.236E-03		7.000E+03	
31	-5.503E-02			-3.500E+04		0.
		1.649E-04	-3.308E-03		7.000E+03	
32	-5.851E-02			-2.800E+04		0.
		1.692E-04	-3.370E-03		7.000E+03	
33	-6.205E-02			-2.100E+04		0.
		1.738E-04	-3.420E-03		7.000E+03	
34	-6.564E-02			-1.400E+04		0.
		1.786E-04	-3.456E-03		7.000E+03	
35	-6.927E-02			-7.000E+03		0.
		1.837E-04	-3.476E-03		7.000E+03	
36	-7.293E-02			7.752E-08		0.
		-1.155E-18	-3.476E-03		-4.399E-11	
37	-7.641E-02			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
50	5.000E+01	1.000E-01	2.000E-04	5.000E-03	5.000E+05	1.000E+04	1.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 2.306 SECONDS

ELAPSED CPU TIME = 0 MINUTES 19.814 SECONDS



Prob 102C. Same as Prob 102A, considering the factor $K = 1.2$.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
201A STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD ON OUTSIDE BEAM

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	-0	-0	-0	-0	-0
NUM CARDS INPUT THIS PROBLEM	1	0	2	0	8
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

NUM INCREMENTS	96
INCREMENT LENGTH	3.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
	NONE

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
ADDITIONAL DATA INPUT FOR THIS PROBLEM						
0	96	-0	3.000E+07	-0.	3.126E+02	-0.
1	96	-0	-0.	1.100E+07	-0.	7.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
				NONE

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
ADDITIONAL DATA INPUT FOR THIS PROBLEM							

0	0	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
24	24	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
48	48	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
72	72	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
96	96	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
0	96	-0	-8.480E+00	-0.	-0.	-0.	-0.
72	72	-0	-2.080E+04	-0.	-0.	-0.	-0.
96	96	-0	-2.080E+04	-0.	-0.	-0.	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
201A STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD ON OUTSIDE BEAM

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	-1.124E-04			0.		0.
0	-1.016E-04	0.	3.621E-06	3.331E-01	0.	-9.853E+01
1	-9.400E-05	3.288E-06	3.621E-06	2.835E+02	9.429E+01	0.
2	-8.586E-05	2.993E-06	3.711E-06	5.410E+02	8.581E+01	0.
3	-7.690E-05	2.697E-06	3.885E-06	7.730E+02	7.733E+01	0.
4	-6.691E-05	2.401E-06	4.132E-06	9.795E+02	6.885E+01	0.
5	-5.568E-05	2.105E-06	4.445E-06	1.161E+03	6.037E+01	0.
6	-4.304E-05	1.810E-06	4.816E-06	1.316E+03	5.189E+01	0.
7	-2.884E-05	1.514E-06	5.237E-06	1.447E+03	4.341E+01	0.
8	-1.296E-05	1.218E-06	5.700E-06	1.551E+03	3.493E+01	0.
9	4.708E-06	9.224E-07	6.196E-06	1.551E+03	2.645E+01	0.
10	2.424E-05	6.267E-07	6.718E-06	1.631E+03	1.797E+01	0.
11	4.568E-05	3.309E-07	7.257E-06	1.685E+03	9.489E+00	0.
12	6.906E-05	3.520E-08	7.805E-06	1.713E+03	1.009E+00	0.
13	9.438E-05	-2.605E-07	8.354E-06	1.716E+03	-7.471E+00	0.
14	1.216E-04	-5.563E-07	8.896E-06	1.694E+03	-1.595E+01	0.
15	1.507E-04	-8.520E-07	9.422E-06	1.646E+03	-2.443E+01	0.
16	1.817E-04	-1.148E-06	9.925E-06	1.572E+03	-3.291E+01	0.
17	2.143E-04	-1.444E-06	1.040E-05	1.474E+03	-4.139E+01	0.
18	2.485E-04	-1.739E-06	1.083E-05	1.350E+03	-4.987E+01	0.
19	2.842E-04	-2.035E-06	1.121E-05	1.200E+03	-5.835E+01	0.
20	3.211E-04	-2.331E-06	1.154E-05	1.025E+03	-6.683E+01	0.
21	3.592E-04	-2.627E-06	1.180E-05	8.244E+02	-7.531E+01	0.
		-2.922E-06	1.200E-05	5.985E+02	-8.379E+01	0.

22	3.981E-04			3.471E+02		0.
23	4.376E-04	-3.218E-06	1.211E-05	7.031E+01	-9.227E+01	0.
24	4.775E-04	-3.514E-06	1.213E-05	-2.308E+02	-1.008E+02	4.632E+02
25	5.337E-04	-1.996E-05	1.206E-05	-1.947E+03	-5.724E+02	0.
26	5.882E-04	-2.026E-05	1.143E-05	-3.690E+03	-5.809E+02	0.
27	6.395E-04	-2.056E-05	1.025E-05	-5.458E+03	-5.894E+02	0.
28	6.859E-04	-2.085E-05	8.506E-06	-7.252E+03	-5.979E+02	0.
29	7.256E-04	-2.115E-05	6.186E-06	-9.071E+03	-6.064E+02	0.
30	7.569E-04	-2.144E-05	3.284E-06	-1.092E+04	-6.148E+02	0.
31	7.780E-04	-2.174E-05	-2.073E-07	-1.279E+04	-6.233E+02	0.
32	7.872E-04	-2.203E-05	-4.297E-06	-1.468E+04	-6.318E+02	0.
33	7.825E-04	-2.233E-05	-8.993E-06	-1.660E+04	-6.403E+02	0.
34	7.622E-04	-2.263E-05	-1.430E-05	-1.855E+04	-6.488E+02	0.
35	7.244E-04	-2.292E-05	-2.024E-05	-2.052E+04	-6.572E+02	0.
36	6.672E-04	-2.322E-05	-2.680E-05	-2.252E+04	-6.657E+02	0.
37	5.887E-04	-2.351E-05	-3.400E-05	-2.454E+04	-6.742E+02	0.
38	4.870E-04	-2.381E-05	-4.185E-05	-2.659E+04	-6.827E+02	0.
39	3.600E-04	-2.410E-05	-5.036E-05	-2.866E+04	-6.912E+02	0.
40	2.058E-04	-2.440E-05	-5.953E-05	-3.076E+04	-6.996E+02	0.
41	2.242E-05	-2.470E-05	-6.937E-05	-3.288E+04	-7.081E+02	0.
42	-1.922E-04	-2.499E-05	-7.989E-05	-3.503E+04	-7.166E+02	0.
43	-4.402E-04	-2.529E-05	-9.109E-05	-3.721E+04	-7.251E+02	0.
44	-7.237E-04	-2.558E-05	-1.030E-04	-3.941E+04	-7.336E+02	0.
45	-1.045E-03	-2.588E-05	-1.156E-04	-4.164E+04	-7.420E+02	0.
46	-1.405E-03	-2.617E-05	-1.289E-04	-4.389E+04	-7.505E+02	0.
47	-1.808E-03	-2.647E-05	-1.430E-04	-4.616E+04	-7.590E+02	0.
48	-2.254E-03	-2.677E-05	-1.577E-04	-4.848E+04	-7.675E+02	0.
49	-2.823E-03	4.919E-05	-1.732E-04	-4.427E+04	1.410E+03	-2.186E+03
50	-3.434E-03	4.889E-05	-1.874E-04	-4.006E+04	1.402E+03	0.
51	-4.083E-03	4.860E-05	-2.002E-04	-3.588E+04	1.393E+03	0.
52	-4.767E-03	4.830E-05	-2.117E-04	-3.172E+04	1.385E+03	0.

53	-5.480E-03	4.801E-05	-2.218E-04	-2.759E+04	1.377E+03	0.
54	-6.220E-03	4.771E-05	-2.307E-04	-2.349E+04	1.368E+03	0.
55	-6.982E-03	4.742E-05	-2.382E-04	-1.941E+04	1.360E+03	0.
56	-7.762E-03	4.712E-05	-2.444E-04	-1.536E+04	1.351E+03	0.
57	-8.557E-03	4.682E-05	-2.493E-04	-1.133E+04	1.343E+03	0.
58	-9.362E-03	4.653E-05	-2.529E-04	-7.328E+03	1.334E+03	0.
59	-1.017E-02	4.623E-05	-2.553E-04	-3.351E+03	1.326E+03	0.
60	-1.099E-02	4.594E-05	-2.563E-04	6.003E+02	1.317E+03	0.
61	-1.180E-02	4.564E-05	-2.562E-04	4.526E+03	1.309E+03	0.
62	-1.261E-02	4.535E-05	-2.547E-04	8.427E+03	1.300E+03	0.
63	-1.341E-02	4.505E-05	-2.520E-04	1.230E+04	1.292E+03	0.
64	-1.420E-02	4.475E-05	-2.481E-04	1.615E+04	1.283E+03	0.
65	-1.498E-02	4.446E-05	-2.429E-04	1.998E+04	1.275E+03	0.
66	-1.573E-02	4.416E-05	-2.365E-04	2.378E+04	1.266E+03	0.
67	-1.646E-02	4.387E-05	-2.289E-04	2.755E+04	1.258E+03	0.
68	-1.716E-02	4.357E-05	-2.201E-04	3.130E+04	1.249E+03	0.
69	-1.784E-02	4.328E-05	-2.101E-04	3.502E+04	1.241E+03	0.
70	-1.848E-02	4.298E-05	-1.989E-04	3.872E+04	1.232E+03	0.
71	-1.908E-02	4.268E-05	-1.865E-04	4.239E+04	1.224E+03	0.
72	-1.964E-02	4.239E-05	-1.729E-04	4.602E+04	1.215E+03	0.
73	-2.010E-02	-1.888E-05	-1.582E-04	4.960E+04	-5.415E+02	-1.905E+04
74	-2.051E-02	-1.918E-05	-1.440E-04	4.438E+04	-5.500E+02	0.
75	-2.088E-02	-1.948E-05	-1.304E-04	4.273E+04	-5.584E+02	0.
76	-2.121E-02	-1.977E-05	-1.172E-04	4.105E+04	-5.669E+02	0.
77	-2.151E-02	-2.007E-05	-1.046E-04	3.935E+04	-5.754E+02	0.
78	-2.176E-02	-2.036E-05	-9.259E-05	3.763E+04	-5.839E+02	0.
79	-2.199E-02	-2.066E-05	-8.112E-05	3.588E+04	-5.924E+02	0.
80	-2.218E-02	-2.095E-05	-7.021E-05	3.410E+04	-6.008E+02	0.
81	-2.233E-02	-2.125E-05	-5.988E-05	3.230E+04	-6.093E+02	0.
82	-2.246E-02	-2.155E-05	-5.013E-05	3.047E+04	-6.178E+02	0.
		-2.184E-05	-4.098E-05	2.861E+04	-6.263E+02	0.

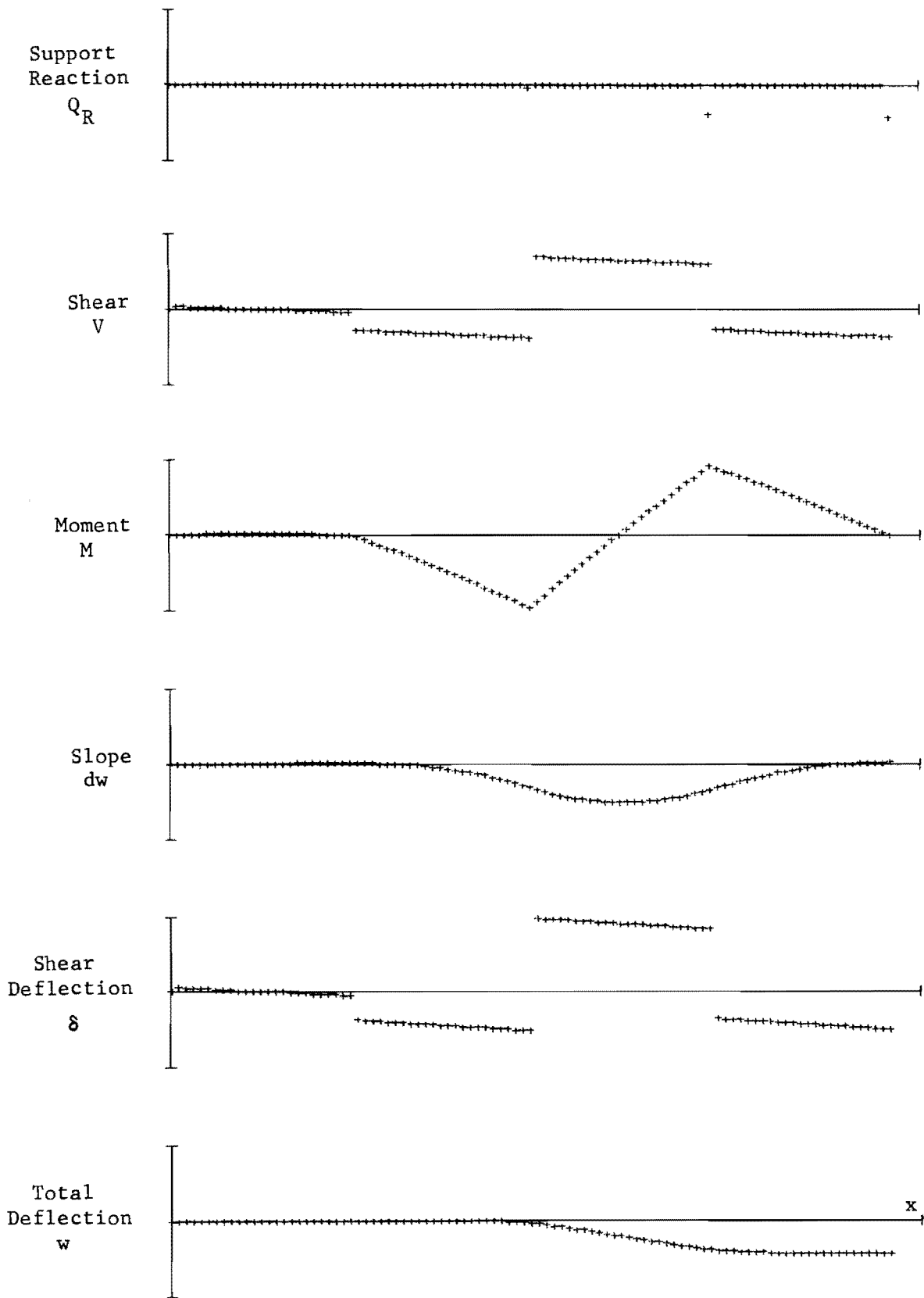
83	-2.256E-02			2.674E+04		0.
84	-2.264E-02	-2.214E-05	-3.243E-05	2.483E+04	-6.348E+02	0.
85	-2.269E-02	-2.243E-05	-2.448E-05	2.290E+04	-6.432E+02	0.
86	-2.272E-02	-2.273E-05	-1.716E-05	2.095E+04	-6.517E+02	0.
87	-2.273E-02	-2.302E-05	-1.045E-05	1.897E+04	-6.602E+02	0.
88	-2.272E-02	-2.332E-05	-4.388E-06	1.696E+04	-6.687E+02	0.
89	-2.269E-02	-2.362E-05	1.038E-06	1.493E+04	-6.772E+02	0.
90	-2.265E-02	-2.391E-05	5.814E-06	1.287E+04	-6.856E+02	0.
91	-2.260E-02	-2.421E-05	9.931E-06	1.079E+04	-6.941E+02	0.
92	-2.253E-02	-2.450E-05	1.338E-05	8.682E+03	-7.026E+02	0.
93	-2.246E-02	-2.480E-05	1.616E-05	6.549E+03	-7.111E+02	0.
94	-2.238E-02	-2.509E-05	1.825E-05	4.390E+03	-7.196E+02	0.
95	-2.229E-02	-2.539E-05	1.966E-05	2.206E+03	-7.280E+02	0.
96	-2.221E-02	-2.569E-05	2.036E-05	-1.874E+00	-7.365E+02	-2.154E+04
97	-2.215E-02	-2.922E-18	2.036E-05	0.	-8.379E-11	0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	3.000E+02	5.000E-02	5.000E-05	5.000E-04	5.000E+04	2.000E+03	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 3.555 SECONDS

ELAPSED CPU TIME = 0 MINUTES 21.167 SECONDS



Prob 201A. Steel bridge diaphragms with wheel load on outside beam.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
201B SAME AS PROBLEM 201A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	0	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	3.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
	NONE

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
			DATA HELD FROM THE PREVIOUS PROBLEM			
0	96	-0	3.000E+07	-0.	3.126E+02	-0.
1	96	-0	-0.	1.100E+07	-0.	7.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
			ADDITIONAL DATA INPUT FOR THIS PROBLEM	
1	96	-0	1.000E+99	-0.

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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DATA HELD FROM THE PREVIOUS PROBLEM

0	0	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
24	24	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
48	48	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
72	72	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
96	96	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
0	96	-0	-8.480E+00	-0.	-0.	-0.	-0.
72	72	-0	-2.080E+04	-0.	-0.	-0.	-0.
96	96	-0	-2.080E+04	-0.	-0.	-0.	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
 2018 SAME AS PROBLEM 201A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	-1.323E-04	0.	3.501E-06	0.	0.	0.
0	-1.218E-04	1.139E-97	3.502E-06	3.221E-01	1.139E+02	-1.182E+02
1	-1.113E-04	1.055E-97	3.611E-06	3.424E+02	1.055E+02	0.
2	-1.005E-04	9.697E-98	3.822E-06	6.588E+02	9.697E+01	0.
3	-8.902E-05	8.849E-98	4.126E-06	9.497E+02	8.849E+01	0.
4	-7.665E-05	8.001E-98	4.514E-06	1.215E+03	8.001E+01	0.
5	-6.310E-05	7.153E-98	4.980E-06	1.455E+03	7.153E+01	0.
6	-4.816E-05	6.305E-98	5.514E-06	1.670E+03	6.305E+01	0.
7	-3.162E-05	5.457E-98	6.109E-06	1.859E+03	5.457E+01	0.
8	-1.330E-05	4.609E-98	6.756E-06	2.023E+03	4.609E+01	0.
9	6.971E-06	3.761E-98	7.447E-06	2.161E+03	3.761E+01	0.
10	2.931E-05	2.913E-98	8.174E-06	2.274E+03	2.913E+01	0.
11	5.384E-05	2.065E-98	8.930E-06	2.361E+03	2.065E+01	0.
12	8.063E-05	1.217E-98	9.705E-06	2.423E+03	1.217E+01	0.
13	1.097E-04	3.693E-99	1.049E-05	2.460E+03	3.693E+00	0.
14	1.412E-04	-4.787E-99	1.128E-05	2.471E+03	-4.787E+00	0.
15	1.751E-04	-1.327E-98	1.207E-05	2.456E+03	-1.327E+01	0.
16	2.113E-04	-2.175E-98	1.284E-05	2.417E+03	-2.175E+01	0.
17	2.498E-04	-3.023E-98	1.359E-05	2.351E+03	-3.023E+01	0.
18	2.906E-04	-3.871E-98	1.432E-05	2.261E+03	-3.871E+01	0.
19	3.335E-04	-4.719E-98	1.500E-05	2.145E+03	-4.719E+01	0.
20	3.785E-04	-5.567E-98	1.564E-05	2.003E+03	-5.567E+01	0.
21	4.255E-04	-6.415E-98	1.623E-05	1.836E+03	-6.415E+01	0.

22	4.741E-04			1.644E+03		0.
		-7.263E-98	1.676E-05		-7.263E+01	
23	5.244E-04			1.426E+03		0.
		-8.111E-98	1.721E-05		-8.111E+01	
24	5.761E-04			1.184E+03		5.588E+02
		-6.484E-97	1.759E-05		-6.484E+02	
25	6.288E-04			-7.595E+02		0.
		-6.568E-97	1.735E-05		-6.568E+02	
26	6.809E-04			-2.730E+03		0.
		-6.653E-97	1.647E-05		-6.653E+02	
27	7.303E-04			-4.726E+03		0.
		-6.738E-97	1.496E-05		-6.738E+02	
28	7.752E-04			-6.747E+03		0.
		-6.823E-97	1.280E-05		-6.823E+02	
29	8.136E-04			-8.794E+03		0.
		-6.908E-97	9.991E-06		-6.908E+02	
30	8.436E-04			-1.087E+04		0.
		-6.992E-97	6.515E-06		-6.992E+02	
31	8.631E-04			-1.296E+04		0.
		-7.077E-97	2.368E-06		-7.077E+02	
32	8.702E-04			-1.509E+04		0.
		-7.162E-97	-2.458E-06		-7.162E+02	
33	8.629E-04			-1.724E+04		0.
		-7.247E-97	-7.972E-06		-7.247E+02	
34	8.389E-04			-1.941E+04		0.
		-7.332E-97	-1.418E-05		-7.332E+02	
35	7.964E-04			-2.161E+04		0.
		-7.416E-97	-2.109E-05		-7.416E+02	
36	7.331E-04			-2.383E+04		0.
		-7.501E-97	-2.872E-05		-7.501E+02	
37	6.470E-04			-2.608E+04		0.
		-7.586E-97	-3.706E-05		-7.586E+02	
38	5.358E-04			-2.836E+04		0.
		-7.671E-97	-4.614E-05		-7.671E+02	
39	3.974E-04			-3.066E+04		0.
		-7.756E-97	-5.594E-05		-7.756E+02	
40	2.295E-04			-3.299E+04		0.
		-7.840E-97	-6.650E-05		-7.840E+02	
41	3.003E-05			-3.534E+04		0.
		-7.925E-97	-7.780E-05		-7.925E+02	
42	-2.034E-04			-3.772E+04		0.
		-8.010E-97	-8.987E-05		-8.010E+02	
43	-4.730E-04			-4.012E+04		0.
		-8.095E-97	-1.027E-04		-8.095E+02	
44	-7.811E-04			-4.255E+04		0.
		-8.180E-97	-1.163E-04		-8.180E+02	
45	-1.130E-03			-4.500E+04		0.
		-8.264E-97	-1.307E-04		-8.264E+02	
46	-1.522E-03			-4.748E+04		0.
		-8.349E-97	-1.459E-04		-8.349E+02	
47	-1.960E-03			-4.999E+04		0.
		-8.434E-97	-1.619E-04		-8.434E+02	
48	-2.446E-03			-5.253E+04		-2.372E+03
		1.520E-96	-1.787E-04		1.520E+03	
49	-2.982E-03			-4.799E+04		0.
		1.512E-96	-1.940E-04		1.512E+03	
50	-3.564E-03			-4.345E+04		0.
		1.503E-96	-2.079E-04		1.503E+03	
51	-4.188E-03			-3.894E+04		0.
		1.495E-96	-2.204E-04		1.495E+03	
52	-4.849E-03			-3.446E+04		0.

53	-5.543E-03	1.486E-96	-2.314E-04	-3.000E+04	1.486E+03	0.
		1.478E-96	-2.410E-04		1.478E+03	0.
54	-6.266E-03	1.469E-96	-2.492E-04	-2.557E+04	1.469E+03	0.
		1.461E-96	-2.560E-04		1.461E+03	0.
55	-7.014E-03	1.461E-96	-2.560E-04	-2.116E+04	1.461E+03	0.
		1.452E-96	-2.613E-04		1.452E+03	0.
56	-7.782E-03	1.452E-96	-2.613E-04	-1.677E+04	1.452E+03	0.
		1.444E-96	-2.653E-04		1.444E+03	0.
57	-8.566E-03	1.444E-96	-2.653E-04	-1.242E+04	1.444E+03	0.
		1.436E-96	-2.679E-04		1.436E+03	0.
58	-9.362E-03	1.436E-96	-2.679E-04	-8.085E+03	1.436E+03	0.
		1.427E-96	-2.691E-04		1.427E+03	0.
59	-1.017E-02	1.427E-96	-2.691E-04	-3.778E+03	1.427E+03	0.
		1.419E-96	-2.689E-04		1.419E+03	0.
60	-1.097E-02	1.419E-96	-2.689E-04	5.031E+02	1.419E+03	0.
		1.410E-96	-2.674E-04		1.410E+03	0.
61	-1.178E-02	1.410E-96	-2.674E-04	4.759E+03	1.410E+03	0.
		1.402E-96	-2.645E-04		1.402E+03	0.
62	-1.258E-02	1.402E-96	-2.645E-04	8.989E+03	1.402E+03	0.
		1.393E-96	-2.603E-04		1.393E+03	0.
63	-1.338E-02	1.393E-96	-2.603E-04	1.319E+04	1.393E+03	0.
		1.385E-96	-2.548E-04		1.385E+03	0.
64	-1.416E-02	1.385E-96	-2.548E-04	1.737E+04	1.385E+03	0.
		1.376E-96	-2.479E-04		1.376E+03	0.
65	-1.492E-02	1.376E-96	-2.479E-04	2.153E+04	1.376E+03	0.
		1.368E-96	-2.397E-04		1.368E+03	0.
66	-1.566E-02	1.368E-96	-2.397E-04	2.566E+04	1.368E+03	0.
		1.359E-96	-2.302E-04		1.359E+03	0.
67	-1.638E-02	1.359E-96	-2.302E-04	2.976E+04	1.359E+03	0.
		1.351E-96	-2.193E-04		1.351E+03	0.
68	-1.707E-02	1.351E-96	-2.193E-04	3.384E+04	1.351E+03	0.
		1.342E-96	-2.072E-04		1.342E+03	0.
69	-1.773E-02	1.342E-96	-2.072E-04	3.789E+04	1.342E+03	0.
		1.334E-96	-1.938E-04		1.334E+03	0.
70	-1.835E-02	1.334E-96	-1.938E-04	4.191E+04	1.334E+03	0.
		1.325E-96	-1.791E-04		1.325E+03	0.
71	-1.893E-02	1.325E-96	-1.791E-04	4.592E+04	1.325E+03	0.
		1.325E-96	-1.791E-04	4.988E+04	1.325E+03	-1.889E+04
72	-1.947E-02	1.325E-96	-1.791E-04	4.988E+04	1.325E+03	-1.889E+04
		-5.951E-97	-1.632E-04		-5.951E+02	0.
73	-1.996E-02	-5.951E-97	-1.632E-04	4.808E+04	-5.951E+02	0.
		-6.035E-97	-1.478E-04		-6.035E+02	0.
74	-2.041E-02	-6.035E-97	-1.478E-04	4.626E+04	-6.035E+02	0.
		-6.120E-97	-1.330E-04		-6.120E+02	0.
75	-2.080E-02	-6.120E-97	-1.330E-04	4.443E+04	-6.120E+02	0.
		-6.205E-97	-1.188E-04		-6.205E+02	0.
76	-2.116E-02	-6.205E-97	-1.188E-04	4.257E+04	-6.205E+02	0.
		-6.290E-97	-1.052E-04		-6.290E+02	0.
77	-2.148E-02	-6.290E-97	-1.052E-04	4.068E+04	-6.290E+02	0.
		-6.375E-97	-9.214E-05		-6.375E+02	0.
78	-2.175E-02	-6.375E-97	-9.214E-05	3.877E+04	-6.375E+02	0.
		-6.459E-97	-7.973E-05		-6.459E+02	0.
79	-2.199E-02	-6.459E-97	-7.973E-05	3.683E+04	-6.459E+02	0.
		-6.544E-97	-6.795E-05		-6.544E+02	0.
80	-2.220E-02	-6.544E-97	-6.795E-05	3.487E+04	-6.544E+02	0.
		-6.629E-97	-5.680E-05		-6.629E+02	0.
81	-2.237E-02	-6.629E-97	-5.680E-05	3.288E+04	-6.629E+02	0.
		-6.714E-97	-4.628E-05		-6.714E+02	0.
82	-2.250E-02	-6.714E-97	-4.628E-05	3.086E+04	-6.714E+02	0.
		-6.799E-97	-3.641E-05		-6.799E+02	0.

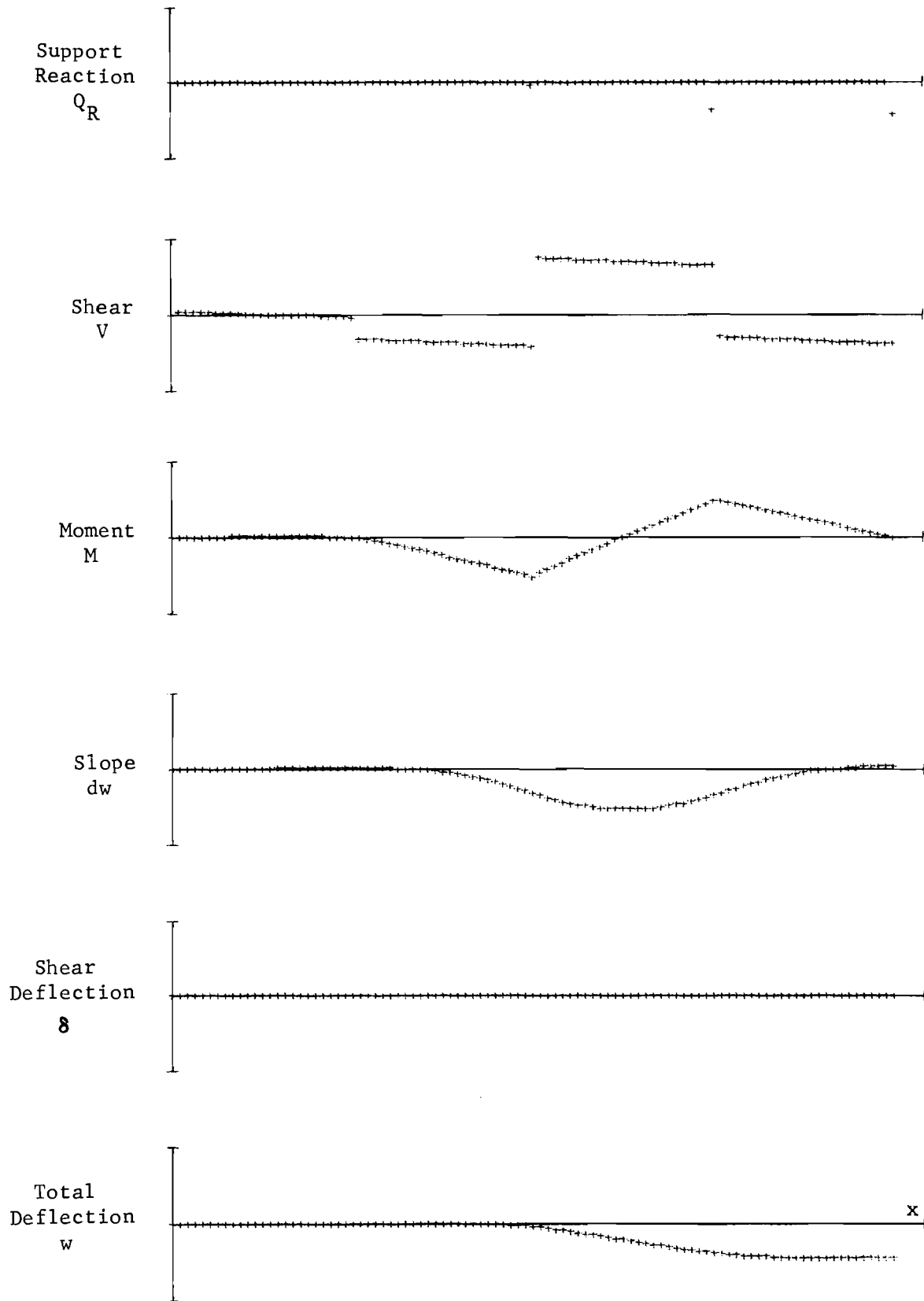
83	-2.261E-02			2.882E+04		0.
		-6.883E-97	-2.719E-05		-6.883E+02	0.
84	-2.270E-02			2.676E+04		0.
		-6.968E-97	-1.863E-05		-6.968E+02	0.
85	-2.275E-02			2.467E+04		0.
		-7.053E-97	-1.074E-05		-7.053E+02	0.
86	-2.278E-02			2.255E+04		0.
		-7.138E-97	-3.521E-06		-7.138E+02	0.
87	-2.279E-02			2.041E+04		0.
		-7.223E-97	3.008E-06		-7.223E+02	0.
88	-2.278E-02			1.824E+04		0.
		-7.307E-97	8.845E-06		-7.307E+02	0.
89	-2.276E-02			1.605E+04		0.
		-7.392E-97	1.398E-05		-7.392E+02	0.
90	-2.272E-02			1.383E+04		0.
		-7.477E-97	1.841E-05		-7.477E+02	0.
91	-2.266E-02			1.159E+04		0.
		-7.562E-97	2.211E-05		-7.562E+02	0.
92	-2.259E-02			9.323E+03		0.
		-7.647E-97	2.510E-05		-7.647E+02	0.
93	-2.252E-02			7.029E+03		0.
		-7.731E-97	2.734E-05		-7.731E+02	0.
94	-2.244E-02			4.710E+03		0.
		-7.816E-97	2.885E-05		-7.816E+02	0.
95	-2.235E-02			2.365E+03		0.
		-7.901E-97	2.961E-05		-7.901E+02	0.
96	-2.226E-02			-2.724E+00		-2.159E+04
		-1.718E-109	2.961E-05		-1.718E-10	0.
97	-2.217E-02			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	3.000E+02	5.000E-02	1.000E+00	5.000E-04	1.000E+05	2.000E+03	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 3.515 SECONDS

ELAPSED CPU TIME = 0 MINUTES 24.682 SECONDS



Prob 201B. Same as Prob 201A, neglecting shear deformations.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
202A STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD ON INTERIOR BEAMS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	0	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	0	2
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	3.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
	NONE

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	-0	3.000E+07	-0.	3.126E+02	-0.
1	96	-0	-0.	1.100E+07	-0.	7.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
				NONE

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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DATA HELD FROM THE PREVIOUS PROBLEM
0  0  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
24 24  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
48 48  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
72 72  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
96 96  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
0  96  -0  -8.480E+00 -0.      -0.      -0.      -0.
72 72  -0  -2.080E+04 -0.      -0.      -0.      -0.
96 96  -0  -2.080E+04 -0.      -0.      -0.      -0.
ADDITIONAL DATA INPUT FOR THIS PROBLEM
96 96  -0  2.080E+04 -0.      -0.      -0.      -0.
48 48  -0  -2.080E+04 -0.      -0.      -0.      -0.
```

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
202A STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD ON INTERIOR BEAMS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	5.705E-04			0.		0.
0	6.030E-04	0.	1.082E-05	9.952E-01	0.	5.849E+02
1	6.560E-04	-2.055E-05	1.082E-05	-1.765E+03	-5.891E+02	0.
2	7.076E-04	-2.084E-05	1.025E-05	-3.558E+03	-5.976E+02	0.
3	7.561E-04	-2.114E-05	9.115E-06	-5.377E+03	-6.061E+02	0.
4	7.997E-04	-2.143E-05	7.395E-06	-7.220E+03	-6.146E+02	0.
5	8.367E-04	-2.173E-05	5.085E-06	-9.090E+03	-6.231E+02	0.
6	8.653E-04	-2.203E-05	2.178E-06	-1.098E+04	-6.315E+02	0.
7	8.836E-04	-2.232E-05	-1.336E-06	-1.290E+04	-6.400E+02	0.
8	8.898E-04	-2.262E-05	-5.464E-06	-1.485E+04	-6.485E+02	0.
9	8.821E-04	-2.291E-05	-1.021E-05	-1.682E+04	-6.570E+02	0.
10	8.585E-04	-2.321E-05	-1.560E-05	-1.882E+04	-6.655E+02	0.
11	8.171E-04	-2.350E-05	-2.162E-05	-2.084E+04	-6.739E+02	0.
12	7.561E-04	-2.380E-05	-2.828E-05	-2.289E+04	-6.824E+02	0.
13	6.734E-04	-2.410E-05	-3.560E-05	-2.496E+04	-6.909E+02	0.
14	5.670E-04	-2.439E-05	-4.359E-05	-2.706E+04	-6.994E+02	0.
15	4.350E-04	-2.469E-05	-5.224E-05	-2.918E+04	-7.079E+02	0.
16	2.752E-04	-2.498E-05	-6.158E-05	-3.133E+04	-7.163E+02	0.
17	8.569E-05	-2.528E-05	-7.160E-05	-3.350E+04	-7.248E+02	0.
18	-1.357E-04	-2.557E-05	-8.232E-05	-3.570E+04	-7.333E+02	0.
19	-3.910E-04	-2.587E-05	-9.374E-05	-3.793E+04	-7.418E+02	0.
20	-6.825E-04	-2.617E-05	-1.059E-04	-4.018E+04	-7.503E+02	0.
21	-1.012E-03	-2.646E-05	-1.187E-04	-4.246E+04	-7.587E+02	0.
		-2.676E-05	-1.323E-04		-7.672E+02	

22	-1.382E-03			-4.476E+04		0.
		-2.705E-05	-1.466E-04		-7.757E+02	
23	-1.795E-03			-4.709E+04		0.
		-2.735E-05	-1.617E-04		-7.842E+02	
24	-2.253E-03			-4.945E+04		-2.185E+03
		4.857E-05	-1.775E-04		1.393E+03	
25	-2.834E-03			-4.529E+04		0.
		4.827E-05	-1.920E-04		1.384E+03	
26	-3.458E-03			-4.114E+04		0.
		4.798E-05	-2.052E-04		1.376E+03	
27	-4.122E-03			-3.701E+04		0.
		4.768E-05	-2.170E-04		1.367E+03	
28	-4.820E-03			-3.291E+04		0.
		4.739E-05	-2.275E-04		1.359E+03	
29	-5.550E-03			-2.883E+04		0.
		4.709E-05	-2.368E-04		1.350E+03	
30	-6.308E-03			-2.478E+04		0.
		4.680E-05	-2.447E-04		1.342E+03	
31	-7.089E-03			-2.076E+04		0.
		4.650E-05	-2.513E-04		1.333E+03	
32	-7.889E-03			-1.676E+04		0.
		4.620E-05	-2.567E-04		1.325E+03	
33	-8.705E-03			-1.278E+04		0.
		4.591E-05	-2.608E-04		1.316E+03	
34	-9.533E-03			-8.834E+03		0.
		4.561E-05	-2.636E-04		1.308E+03	
35	-1.037E-02			-4.910E+03		0.
		4.532E-05	-2.652E-04		1.299E+03	
36	-1.121E-02			-1.012E+03		0.
		4.502E-05	-2.655E-04		1.291E+03	
37	-1.205E-02			2.861E+03		0.
		4.473E-05	-2.646E-04		1.282E+03	
38	-1.289E-02			6.708E+03		0.
		4.443E-05	-2.624E-04		1.274E+03	
39	-1.372E-02			1.053E+04		0.
		4.413E-05	-2.591E-04		1.265E+03	
40	-1.454E-02			1.433E+04		0.
		4.384E-05	-2.545E-04		1.257E+03	
41	-1.535E-02			1.810E+04		0.
		4.354E-05	-2.487E-04		1.249E+03	
42	-1.614E-02			2.184E+04		0.
		4.325E-05	-2.417E-04		1.240E+03	
43	-1.691E-02			2.556E+04		0.
		4.295E-05	-2.335E-04		1.232E+03	
44	-1.765E-02			2.926E+04		0.
		4.266E-05	-2.242E-04		1.223E+03	
45	-1.837E-02			3.293E+04		0.
		4.236E-05	-2.136E-04		1.215E+03	
46	-1.905E-02			3.657E+04		0.
		4.206E-05	-2.019E-04		1.206E+03	
47	-1.970E-02			4.019E+04		0.
		4.177E-05	-1.891E-04		1.198E+03	
48	-2.031E-02			4.376E+04		-1.970E+04
		3.054E-06	-1.751E-04		8.756E+01	
49	-2.084E-02			4.401E+04		0.
		2.758E-06	-1.610E-04		7.908E+01	
50	-2.132E-02			4.425E+04		0.
		2.462E-06	-1.468E-04		7.060E+01	
51	-2.176E-02			4.446E+04		0.
		2.167E-06	-1.326E-04		6.212E+01	
52	-2.216E-02			4.465E+04		0.

53	-2.252E-02	1.871E-06	-1.183E-04	4.481E+04	5.364E+01	0.
54	-2.284E-02	1.575E-06	-1.040E-04	4.494E+04	4.516E+01	0.
55	-2.311E-02	1.279E-06	-8.963E-05	4.505E+04	3.668E+01	0.
56	-2.333E-02	9.836E-07	-7.521E-05	4.514E+04	2.820E+01	0.
57	-2.351E-02	6.879E-07	-6.077E-05	4.520E+04	1.972E+01	0.
58	-2.365E-02	3.922E-07	-4.632E-05	4.523E+04	1.124E+01	0.
59	-2.375E-02	9.641E-08	-3.185E-05	4.524E+04	2.765E+00	0.
60	-2.380E-02	-1.993E-07	-1.738E-05	4.522E+04	-5.715E+00	0.
61	-2.381E-02	-4.951E-07	-2.909E-06	4.518E+04	-1.420E+01	0.
62	-2.377E-02	-7.908E-07	1.154E-05	4.511E+04	-2.268E+01	0.
63	-2.370E-02	-1.087E-06	2.597E-05	4.502E+04	-3.116E+01	0.
64	-2.357E-02	-1.382E-06	4.038E-05	4.490E+04	-3.964E+01	0.
65	-2.341E-02	-1.678E-06	5.474E-05	4.475E+04	-4.812E+01	0.
66	-2.320E-02	-1.974E-06	6.905E-05	4.475E+04	-5.660E+01	0.
67	-2.295E-02	-2.270E-06	8.332E-05	4.458E+04	-6.508E+01	0.
68	-2.265E-02	-2.565E-06	9.752E-05	4.439E+04	-7.356E+01	0.
69	-2.231E-02	-2.861E-06	1.116E-04	4.417E+04	-8.204E+01	0.
70	-2.193E-02	-3.157E-06	1.257E-04	4.392E+04	-9.052E+01	0.
71	-2.151E-02	-3.453E-06	1.397E-04	4.365E+04	-9.900E+01	0.
72	-2.105E-02	-3.748E-06	1.535E-04	4.335E+04	-1.075E+02	0.
73	-2.053E-02	-3.748E-06	1.535E-04	4.305E+04	-1.075E+02	-2.041E+04
74	-1.997E-02	-1.749E-05	1.673E-04	4.156E+04	-5.014E+02	0.
75	-1.937E-02	-1.778E-05	1.806E-04	4.003E+04	-5.099E+02	0.
76	-1.873E-02	-1.808E-05	1.934E-04	3.847E+04	-5.184E+02	0.
77	-1.806E-02	-1.837E-05	2.057E-04	3.689E+04	-5.268E+02	0.
78	-1.736E-02	-1.867E-05	2.175E-04	3.689E+04	-5.353E+02	0.
79	-1.662E-02	-1.897E-05	2.288E-04	3.529E+04	-5.438E+02	0.
80	-1.585E-02	-1.897E-05	2.288E-04	3.365E+04	-5.438E+02	0.
81	-1.505E-02	-1.926E-05	2.396E-04	3.200E+04	-5.523E+02	0.
82	-1.423E-02	-1.956E-05	2.498E-04	3.031E+04	-5.608E+02	0.
		-1.985E-05	2.595E-04	2.861E+04	-5.692E+02	0.
		-2.015E-05	2.686E-04	2.687E+04	-5.777E+02	0.
		-2.044E-05	2.772E-04		-5.862E+02	0.

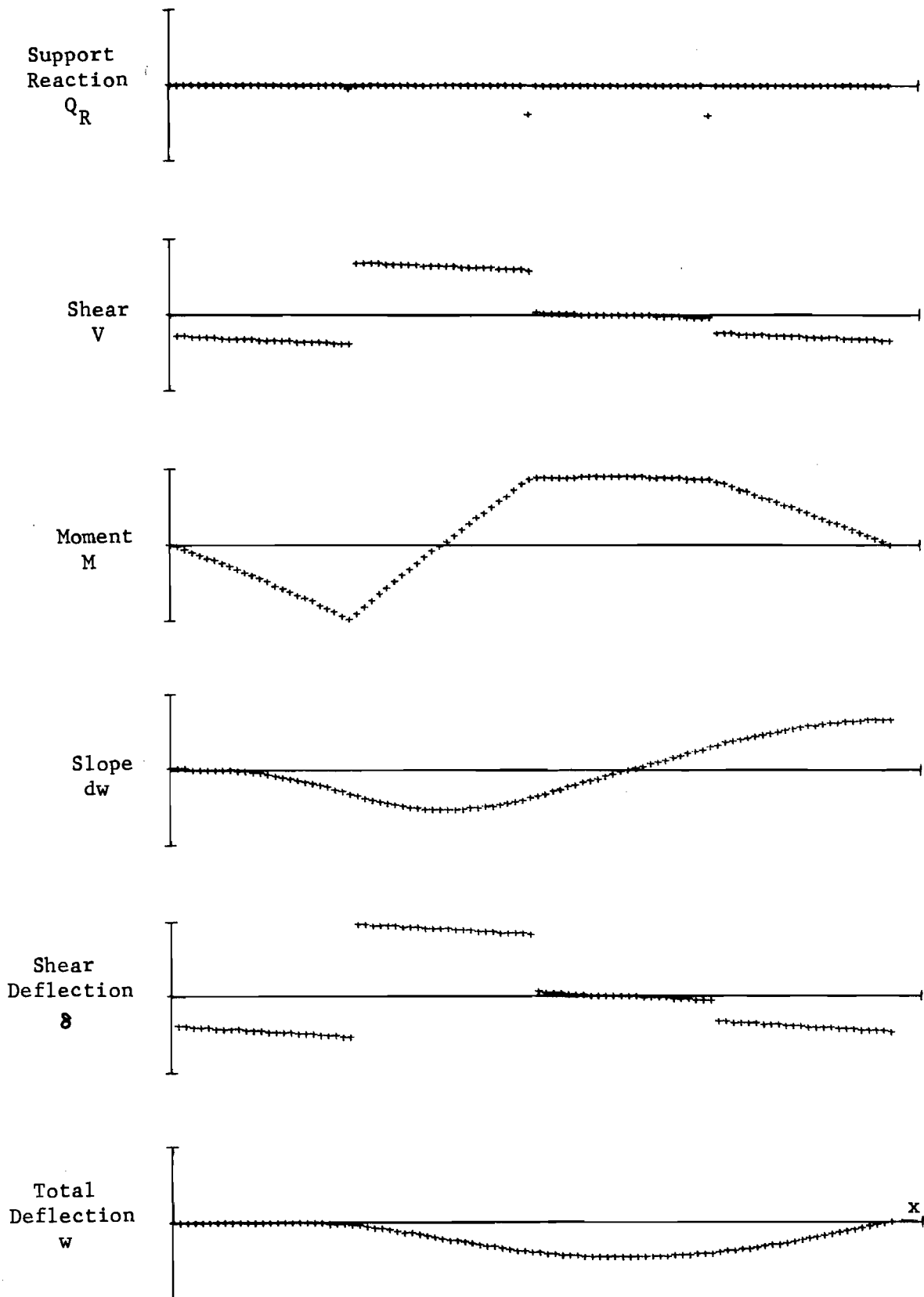
83	-1.337E-02			2.512E+04		0.
84	-1.250E-02	-2.074E-05	2.853E-04	2.333E+04	-5.947E+02	0.
85	-1.160E-02	-2.104E-05	2.927E-04	2.152E+04	-6.032E+02	0.
86	-1.068E-02	-2.133E-05	2.996E-04	1.969E+04	-6.116E+02	0.
87	-9.738E-03	-2.163E-05	3.059E-04	1.783E+04	-6.201E+02	0.
88	-8.781E-03	-2.192E-05	3.116E-04	1.594E+04	-6.286E+02	0.
89	-7.809E-03	-2.222E-05	3.167E-04	1.403E+04	-6.371E+02	0.
90	-6.822E-03	-2.251E-05	3.212E-04	1.209E+04	-6.456E+02	0.
91	-5.824E-03	-2.281E-05	3.251E-04	1.013E+04	-6.540E+02	0.
92	-4.816E-03	-2.311E-05	3.283E-04	8.143E+03	-6.625E+02	0.
93	-3.800E-03	-2.340E-05	3.309E-04	6.130E+03	-6.710E+02	0.
94	-2.778E-03	-2.370E-05	3.329E-04	4.092E+03	-6.795E+02	0.
95	-1.751E-03	-2.399E-05	3.342E-04	2.028E+03	-6.880E+02	0.
96	-7.224E-04	-2.429E-05	3.348E-04	-3.080E+01	-6.964E+02	-7.007E+02
97	2.821E-04	4.566E-20	3.348E-04	0.	1.309E-12	0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	3.000E+02	5.000E-02	5.000E-05	5.000E-04	5.000E+04	2.000E+03	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 3.554 SECONDS

ELAPSED CPU TIME = 0 MINUTES 28.236 SECONDS



Prob 202A. Steel bridge diaphragms with wheel loads on interior beams.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
202B SAME AS PROBLEM 202A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	0	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	3.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
	NONE

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	-0	3.000E+07	-0.	3.126E+02	-0.
1	96	-0	-0.	1.100E+07	-0.	7.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
ADDITIONAL DATA INPUT FOR THIS PROBLEM				
1	96	-0	1.000E+99	-0.

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
------	----	-------	---	---	---	---	---

DATA HELD FROM THE PREVIOUS PROBLEM

0	0	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
24	24	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
48	48	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
72	72	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
96	96	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
0	96	-0	-8.480E+00	-0.	-0.	-0.	-0.
72	72	-0	-2.080E+04	-0.	-0.	-0.	-0.
96	96	-0	-2.080E+04	-0.	-0.	-0.	-0.
96	96	-0	2.080E+04	-0.	-0.	-0.	-0.
48	48	-0	-2.080E+04	-0.	-0.	-0.	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)

202B SAME AS PROBLEM 202A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	5.958E-04			0.		0.
		0.	2.091E-05		0.	
0	6.586E-04			1.924E+00		6.388E+02
		-6.431E-97	2.091E-05		-6.431E+02	
1	7.213E-04			-1.925E+03		0.
		-6.515E-97	2.030E-05		-6.515E+02	
2	7.822E-04			-3.880E+03		0.
		-6.600E-97	1.906E-05		-6.600E+02	
3	8.394E-04			-5.860E+03		0.
		-6.685E-97	1.718E-05		-6.685E+02	
4	8.909E-04			-7.865E+03		0.
		-6.770E-97	1.467E-05		-6.770E+02	
5	9.349E-04			-9.896E+03		0.
		-6.855E-97	1.150E-05		-6.855E+02	
6	9.694E-04			-1.195E+04		0.
		-6.939E-97	7.678E-06		-6.939E+02	
7	9.925E-04			-1.403E+04		0.
		-7.024E-97	3.188E-06		-7.024E+02	
8	1.002E-03			-1.614E+04		0.
		-7.109E-97	-1.976E-06		-7.109E+02	
9	9.961E-04			-1.827E+04		0.
		-7.194E-97	-7.822E-06		-7.194E+02	
10	9.726E-04			-2.043E+04		0.
		-7.279E-97	-1.436E-05		-7.279E+02	
11	9.296E-04			-2.262E+04		0.
		-7.363E-97	-2.159E-05		-7.363E+02	
12	8.648E-04			-2.483E+04		0.
		-7.448E-97	-2.953E-05		-7.448E+02	
13	7.762E-04			-2.706E+04		0.
		-7.533E-97	-3.819E-05		-7.533E+02	
14	6.616E-04			-2.932E+04		0.
		-7.618E-97	-4.757E-05		-7.618E+02	
15	5.189E-04			-3.160E+04		0.
		-7.703E-97	-5.768E-05		-7.703E+02	
16	3.459E-04			-3.392E+04		0.
		-7.787E-97	-6.853E-05		-7.787E+02	
17	1.403E-04			-3.625E+04		0.
		-7.872E-97	-8.013E-05		-7.872E+02	
18	-1.001E-04			-3.861E+04		0.
		-7.957E-97	-9.248E-05		-7.957E+02	
19	-3.775E-04			-4.100E+04		0.
		-8.042E-97	-1.056E-04		-8.042E+02	
20	-6.943E-04			-4.341E+04		0.
		-8.127E-97	-1.195E-04		-8.127E+02	
21	-1.053E-03			-4.585E+04		0.
		-8.211E-97	-1.342E-04		-8.211E+02	

22	-1.455E-03			-4.831E+04		0.
23	-1.904E-03	-8.296E-97	-1.496E-04	-5.080E+04	-8.296E+02	0.
24	-2.402E-03	-8.381E-97	-1.659E-04	-5.333E+04	-8.381E+02	-2.330E+03
25	-2.950E-03	1.483E-96	-1.829E-04	-4.890E+04	1.483E+03	0.
26	-3.546E-03	1.475E-96	-1.986E-04	-4.448E+04	1.475E+03	0.
27	-4.184E-03	1.466E-96	-2.128E-04	-4.008E+04	1.466E+03	0.
28	-4.861E-03	1.458E-96	-2.256E-04	-3.571E+04	1.458E+03	0.
29	-5.572E-03	1.449E-96	-2.370E-04	-3.136E+04	1.449E+03	0.
30	-6.314E-03	1.441E-96	-2.471E-04	-2.704E+04	1.441E+03	0.
31	-7.081E-03	1.432E-96	-2.557E-04	-2.274E+04	1.432E+03	0.
32	-7.870E-03	1.424E-96	-2.630E-04	-1.847E+04	1.424E+03	0.
33	-8.676E-03	1.415E-96	-2.689E-04	-1.422E+04	1.415E+03	0.
34	-9.497E-03	1.407E-96	-2.734E-04	-1.000E+04	1.407E+03	0.
35	-1.033E-02	1.398E-96	-2.766E-04	-5.810E+03	1.398E+03	0.
36	-1.116E-02	1.390E-96	-2.785E-04	-1.641E+03	1.390E+03	0.
37	-1.200E-02	1.381E-96	-2.790E-04	2.503E+03	1.381E+03	0.
38	-1.283E-02	1.373E-96	-2.782E-04	6.621E+03	1.373E+03	0.
39	-1.366E-02	1.364E-96	-2.761E-04	1.071E+04	1.364E+03	0.
40	-1.448E-02	1.356E-96	-2.727E-04	1.478E+04	1.356E+03	0.
41	-1.528E-02	1.347E-96	-2.680E-04	1.882E+04	1.347E+03	0.
42	-1.607E-02	1.339E-96	-2.619E-04	2.284E+04	1.339E+03	0.
43	-1.683E-02	1.330E-96	-2.546E-04	2.683E+04	1.330E+03	0.
44	-1.757E-02	1.322E-96	-2.460E-04	3.080E+04	1.322E+03	0.
45	-1.828E-02	1.313E-96	-2.362E-04	3.474E+04	1.313E+03	0.
46	-1.896E-02	1.305E-96	-2.251E-04	3.865E+04	1.305E+03	0.
47	-1.959E-02	1.296E-96	-2.127E-04	4.254E+04	1.296E+03	0.
48	-2.019E-02	1.288E-96	-1.991E-04	4.639E+04	1.288E+03	-1.959E+04
49	-2.074E-02	6.514E-98	-1.843E-04	4.656E+04	6.514E+01	0.
50	-2.125E-02	5.666E-98	-1.694E-04	4.673E+04	5.666E+01	0.
51	-2.172E-02	4.818E-98	-1.544E-04	4.688E+04	4.818E+01	0.
52	-2.213E-02	3.970E-98	-1.394E-04	4.700E+04	3.970E+01	0.

53	-2.251E-02	3.122E-98	-1.244E-04	4.709E+04	3.122E+01	0.
54	-2.284E-02	2.274E-98	-1.093E-04	4.716E+04	2.274E+01	0.
55	-2.312E-02	1.426E-98	-9.425E-05	4.720E+04	1.426E+01	0.
56	-2.336E-02	5.781E-99	-7.915E-05	4.722E+04	5.781E+00	0.
57	-2.355E-02	-2.699E-99	-6.404E-05	4.721E+04	-2.699E+00	0.
58	-2.369E-02	-1.118E-98	-4.894E-05	4.718E+04	-1.118E+01	0.
59	-2.380E-02	-1.966E-98	-3.384E-05	4.712E+04	-1.966E+01	0.
60	-2.385E-02	-2.814E-98	-1.877E-05	4.703E+04	-2.814E+01	0.
61	-2.386E-02	-3.662E-98	-3.725E-06	4.692E+04	-3.662E+01	0.
62	-2.383E-02	-4.510E-98	1.129E-05	4.679E+04	-4.510E+01	0.
63	-2.375E-02	-5.358E-98	2.625E-05	4.663E+04	-5.358E+01	0.
64	-2.363E-02	-6.206E-98	4.117E-05	4.644E+04	-6.206E+01	0.
65	-2.346E-02	-7.054E-98	5.603E-05	4.623E+04	-7.054E+01	0.
66	-2.325E-02	-7.902E-98	7.082E-05	4.599E+04	-7.902E+01	0.
67	-2.299E-02	-8.750E-98	8.553E-05	4.573E+04	-8.750E+01	0.
68	-2.269E-02	-9.598E-98	1.002E-04	4.544E+04	-9.598E+01	0.
69	-2.235E-02	-1.045E-97	1.147E-04	4.513E+04	-1.045E+02	0.
70	-2.196E-02	-1.129E-97	1.291E-04	4.479E+04	-1.129E+02	0.
71	-2.153E-02	-1.214E-97	1.435E-04	4.443E+04	-1.214E+02	0.
72	-2.105E-02	-1.299E-97	1.577E-04	4.405E+04	-1.299E+02	-2.042E+04
73	-2.054E-02	-5.154E-97	1.718E-04	4.252E+04	-5.154E+02	0.
74	-1.998E-02	-5.239E-97	1.854E-04	4.095E+04	-5.239E+02	0.
75	-1.939E-02	-5.324E-97	1.985E-04	3.935E+04	-5.324E+02	0.
76	-1.875E-02	-5.408E-97	2.111E-04	3.773E+04	-5.408E+02	0.
77	-1.809E-02	-5.493E-97	2.231E-04	3.608E+04	-5.493E+02	0.
78	-1.738E-02	-5.578E-97	2.347E-04	3.441E+04	-5.578E+02	0.
79	-1.664E-02	-5.663E-97	2.457E-04	3.271E+04	-5.663E+02	0.
80	-1.588E-02	-5.748E-97	2.561E-04	3.099E+04	-5.748E+02	0.
81	-1.508E-02	-5.832E-97	2.661E-04	2.924E+04	-5.832E+02	0.
82	-1.425E-02	-5.917E-97	2.754E-04	2.746E+04	-5.917E+02	0.
		-6.002E-97	2.842E-04		-6.002E+02	

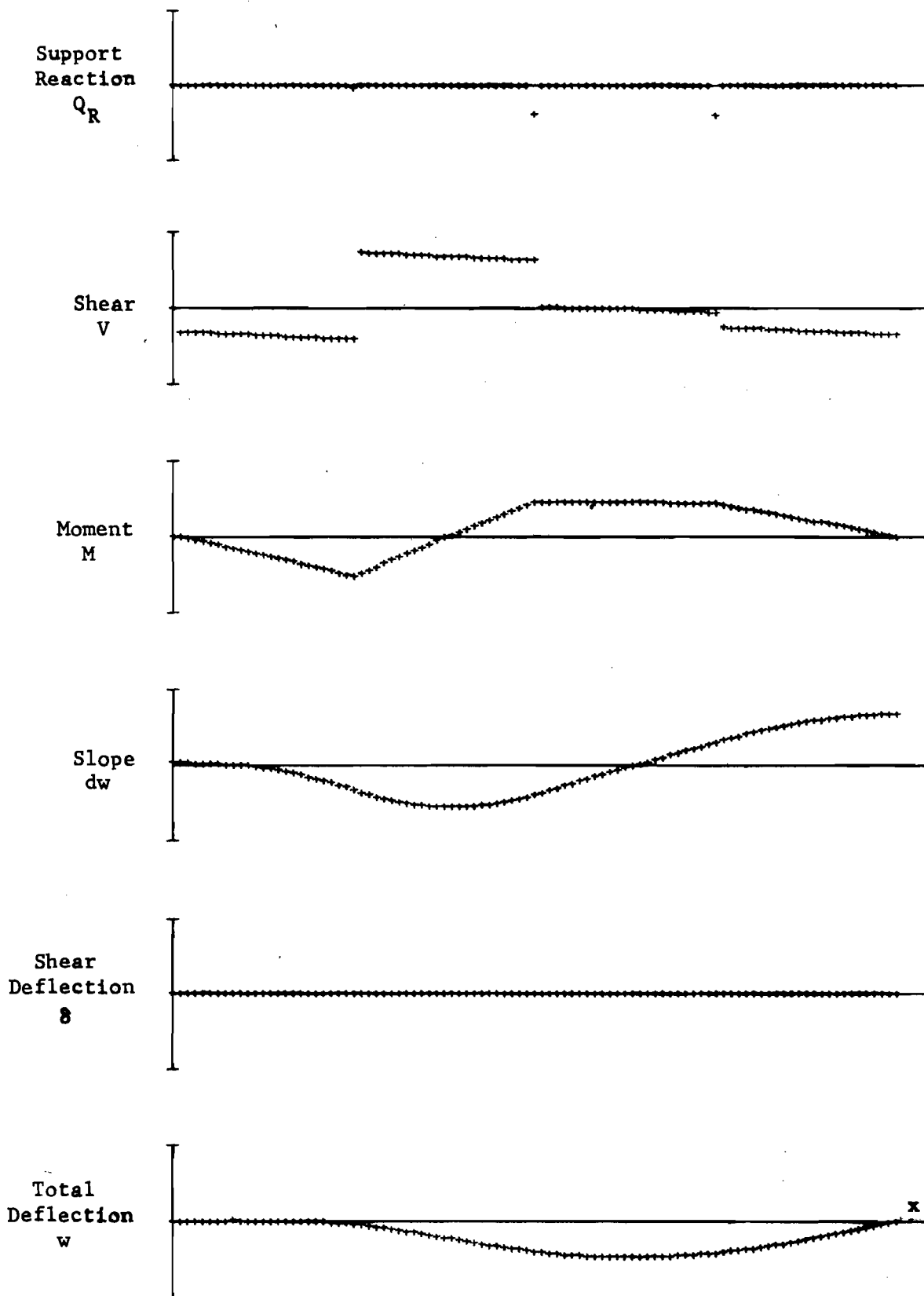
83	-1.340E-02			2.566E+04		0.
		-6.087E-97	2.924E-04		-6.087E+02	
84	-1.252E-02			2.383E+04		0.
		-6.172E-97	3.000E-04		-6.172E+02	
85	-1.162E-02			2.198E+04		0.
		-6.256E-97	3.071E-04		-6.256E+02	
86	-1.070E-02			2.011E+04		0.
		-6.341E-97	3.135E-04		-6.341E+02	
87	-9.760E-03			1.820E+04		0.
		-6.426E-97	3.193E-04		-6.426E+02	
88	-8.802E-03			1.628E+04		0.
		-6.511E-97	3.245E-04		-6.511E+02	
89	-7.828E-03			1.432E+04		0.
		-6.596E-97	3.291E-04		-6.596E+02	
90	-6.841E-03			1.234E+04		0.
		-6.680E-97	3.330E-04		-6.680E+02	
91	-5.842E-03			1.034E+04		0.
		-6.765E-97	3.364E-04		-6.765E+02	
92	-4.833E-03			8.310E+03		0.
		-6.850E-97	3.390E-04		-6.850E+02	
93	-3.816E-03			6.255E+03		0.
		-6.935E-97	3.410E-04		-6.935E+02	
94	-2.793E-03			4.174E+03		0.
		-7.020E-97	3.423E-04		-7.020E+02	
95	-1.766E-03			2.068E+03		0.
		-7.104E-97	3.430E-04		-7.104E+02	
96	-7.368E-04			-3.156E+01		-7.147E+02
		1.343-111	3.430E-04		1.343E-12	
97	2.922E-04			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	3.000E+02	5.000E-02	1.000E+00	5.000E-04	1.000E+05	2.000E+03	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 3.523 SECONDS

ELAPSED CPU TIME = 0 MINUTES 31.759 SECONDS



Prob 202B. Same as Prob 202A, neglecting shear deformations.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
203A STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD AT CENTER OF DIAPHRAGM

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	0	1	0	1
NUM CARDS INPJT THIS PROBLEM	0	0	0	0	4
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	3.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
	NONE

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	-0	3.000E+07	-0.	3.126E+02	-0.
1	96	-0	-0.	1.100E+07	-0.	7.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
				NONE

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
------	----	-------	---	---	---	---	---


```
DATA HELD FROM THE PREVIOUS PROBLEM
 0  0  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
24 24  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
48 48  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
72 72  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
96 96  -0  -0.      9.700E+05 -0.      1.840E+05 -0.
 0 96  -0 -8.480E+00 -0.      -0.      -0.
72 72  -0 -2.080E+04 -0.      -0.      -0.
96 96  -0 -2.080E+04 -0.      -0.      -0.
96 96  -0  2.080E+04 -0.      -0.      -0.
48 48  -0 -2.080E+04 -0.      -0.      -0.
ADDITIONAL DATA INPUT FOR THIS PROBLEM
48 48  -0  2.080E+04 -0.      -0.      -0.
72 72  -0  2.080E+04 -0.      -0.      -0.
36 36  -0 -2.080E+04 -0.      -0.      -0.
60 60  -0 -2.080E+04 -0.      -0.      -0.
```

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
203A STEEL BRIDGE DIAPHRAGMS WITH WHEEL LOAD AT CENTER OF DIAPHRAGM

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	1.520E-03	0.	-7.133E-05	0.	0.	0.
0	1.306E-03	-4.433E-05	-7.133E-05	-6.563E+00	-1.271E+03	1.267E+03
1	1.136E-03	-4.462E-05	-7.256E-05	-3.826E+03	-1.279E+03	0.
2	9.632E-04	-4.492E-05	-7.501E-05	-7.664E+03	-1.288E+03	0.
3	7.831E-04	-4.521E-05	-7.870E-05	-1.153E+04	-1.296E+03	0.
4	5.922E-04	-4.551E-05	-8.363E-05	-1.542E+04	-1.305E+03	0.
5	3.868E-04	-4.581E-05	-8.981E-05	-1.933E+04	-1.313E+03	0.
6	1.632E-04	-4.610E-05	-9.726E-05	-2.327E+04	-1.322E+03	0.
7	-8.248E-05	-4.640E-05	-1.060E-04	-2.724E+04	-1.330E+03	0.
8	-3.540E-04	-4.669E-05	-1.160E-04	-3.123E+04	-1.339E+03	0.
9	-6.552E-04	-4.699E-05	-1.272E-04	-3.525E+04	-1.347E+03	0.
10	-9.899E-04	-4.728E-05	-1.398E-04	-3.929E+04	-1.356E+03	0.
11	-1.362E-03	-4.758E-05	-1.537E-04	-4.335E+04	-1.364E+03	0.
12	-1.775E-03	-4.788E-05	-1.689E-04	-4.745E+04	-1.373E+03	0.
13	-2.234E-03	-4.817E-05	-1.853E-04	-5.157E+04	-1.381E+03	0.
14	-2.742E-03	-4.847E-05	-2.032E-04	-5.571E+04	-1.390E+03	0.
15	-3.303E-03	-4.876E-05	-2.223E-04	-5.988E+04	-1.398E+03	0.
16	-3.921E-03	-4.906E-05	-2.428E-04	-6.407E+04	-1.407E+03	0.
17	-4.601E-03	-4.935E-05	-2.647E-04	-6.829E+04	-1.415E+03	0.
18	-5.345E-03	-4.965E-05	-2.879E-04	-7.254E+04	-1.424E+03	0.
19	-6.159E-03	-4.995E-05	-3.124E-04	-7.681E+04	-1.432E+03	0.
20	-7.047E-03	-5.024E-05	-3.384E-04	-8.111E+04	-1.441E+03	0.
21	-8.012E-03	-5.054E-05	-3.657E-04	-8.543E+04	-1.449E+03	0.

22	-9.058E-03			-8.977E+04		0.
		-5.083E-05	-3.944E-04		-1.458E+03	
23	-1.019E-02			-9.415E+04		0.
		-5.113E-05	-4.246E-04		-1.466E+03	
24	-1.141E-02			-9.859E+04		-1.107E+04
		3.347E-04	-4.561E-04		9.596E+03	
25	-1.312E-02			-6.984E+04		0.
		3.344E-04	-4.784E-04		9.588E+03	
26	-1.489E-02			-4.107E+04		0.
		3.341E-04	-4.916E-04		9.579E+03	
27	-1.669E-02			-1.234E+04		0.
		3.338E-04	-4.955E-04		9.571E+03	
28	-1.852E-02			1.638E+04		0.
		3.335E-04	-4.903E-04		9.562E+03	
29	-2.032E-02			4.506E+04		0.
		3.332E-04	-4.759E-04		9.554E+03	
30	-2.208E-02			7.373E+04		0.
		3.329E-04	-4.523E-04		9.545E+03	
31	-2.377E-02			1.024E+05		0.
		3.326E-04	-4.195E-04		9.537E+03	
32	-2.536E-02			1.310E+05		0.
		3.323E-04	-3.776E-04		9.529E+03	
33	-2.683E-02			1.596E+05		0.
		3.320E-04	-3.266E-04		9.520E+03	
34	-2.814E-02			1.881E+05		0.
		3.317E-04	-2.664E-04		9.512E+03	
35	-2.927E-02			2.167E+05		0.
		3.314E-04	-1.971E-04		9.503E+03	
36	-3.019E-02			2.452E+05		0.
		-3.943E-04	-1.187E-04		-1.131E+04	
37	-3.015E-02			2.112E+05		0.
		-3.946E-04	-5.110E-05		-1.131E+04	
38	-2.991E-02			1.773E+05		0.
		-3.949E-04	5.621E-06		-1.132E+04	
39	-2.950E-02			1.433E+05		0.
		-3.952E-04	5.147E-05		-1.133E+04	
40	-2.895E-02			1.093E+05		0.
		-3.955E-04	8.645E-05		-1.134E+04	
41	-2.830E-02			7.533E+04		0.
		-3.958E-04	1.106E-04		-1.135E+04	
42	-2.757E-02			4.129E+04		0.
		-3.961E-04	1.238E-04		-1.136E+04	
43	-2.680E-02			7.216E+03		0.
		-3.964E-04	1.261E-04		-1.136E+04	
44	-2.603E-02			-2.688E+04		0.
		-3.966E-04	1.175E-04		-1.137E+04	
45	-2.528E-02			-6.100E+04		0.
		-3.969E-04	9.796E-05		-1.138E+04	
46	-2.459E-02			-9.514E+04		0.
		-3.972E-04	6.752E-05		-1.139E+04	
47	-2.399E-02			-1.293E+05		0.
		-3.975E-04	2.615E-05		-1.140E+04	
48	-2.351E-02			-1.635E+05		-2.281E+04
		3.975E-04	-2.615E-05		1.140E+04	
49	-2.399E-02			-1.293E+05		0.
		3.972E-04	-6.752E-05		1.139E+04	
50	-2.459E-02			-9.514E+04		0.
		3.969E-04	-9.796E-05		1.138E+04	
51	-2.528E-02			-6.100E+04		0.
		3.966E-04	-1.175E-04		1.137E+04	
52	-2.603E-02			-2.688E+04		0.

53	-2.680E-02	3.964E-04	-1.261E-04	7.216E+03	1.136E+04	0.
54	-2.757E-02	3.961E-04	-1.238E-04	4.129E+04	1.136E+04	0.
55	-2.830E-02	3.958E-04	-1.106E-04	7.533E+04	1.135E+04	0.
56	-2.895E-02	3.955E-04	-8.645E-05	1.093E+05	1.134E+04	0.
57	-2.950E-02	3.952E-04	-5.147E-05	1.433E+05	1.133E+04	0.
58	-2.991E-02	3.949E-04	-5.621E-06	1.773E+05	1.132E+04	0.
59	-3.015E-02	3.946E-04	5.110E-05	2.112E+05	1.131E+04	0.
60	-3.019E-02	3.943E-04	1.187E-04	2.452E+05	1.131E+04	0.
61	-2.927E-02	-3.314E-04	1.971E-04	2.167E+05	-9.503E+03	0.
62	-2.814E-02	-3.317E-04	2.664E-04	1.881E+05	-9.512E+03	0.
63	-2.683E-02	-3.320E-04	3.266E-04	1.596E+05	-9.520E+03	0.
64	-2.536E-02	-3.323E-04	3.776E-04	1.310E+05	-9.529E+03	0.
65	-2.377E-02	-3.326E-04	4.195E-04	1.024E+05	-9.537E+03	0.
66	-2.208E-02	-3.329E-04	4.523E-04	7.373E+04	-9.545E+03	0.
67	-2.032E-02	-3.332E-04	4.759E-04	4.506E+04	-9.554E+03	0.
68	-1.852E-02	-3.335E-04	4.903E-04	1.638E+04	-9.562E+03	0.
69	-1.669E-02	-3.338E-04	4.955E-04	-1.234E+04	-9.571E+03	0.
70	-1.489E-02	-3.341E-04	4.916E-04	-4.107E+04	-9.579E+03	0.
71	-1.312E-02	-3.344E-04	4.784E-04	-6.984E+04	-9.588E+03	0.
72	-1.141E-02	-3.347E-04	4.561E-04	-9.859E+04	-9.596E+03	0.
73	-1.019E-02	5.113E-05	4.246E-04	-9.415E+04	1.466E+03	-1.107E+04
74	-9.058E-03	5.083E-05	3.944E-04	-8.977E+04	1.458E+03	0.
75	-8.012E-03	5.054E-05	3.657E-04	-8.543E+04	1.449E+03	0.
76	-7.047E-03	5.024E-05	3.384E-04	-8.111E+04	1.441E+03	0.
77	-6.159E-03	4.995E-05	3.124E-04	-7.681E+04	1.432E+03	0.
78	-5.345E-03	4.965E-05	2.879E-04	-7.254E+04	1.424E+03	0.
79	-4.601E-03	4.935E-05	2.647E-04	-6.829E+04	1.415E+03	0.
80	-3.921E-03	4.906E-05	2.428E-04	-6.407E+04	1.407E+03	0.
81	-3.303E-03	4.876E-05	2.223E-04	-5.988E+04	1.398E+03	0.
82	-2.742E-03	4.847E-05	2.032E-04	-5.571E+04	1.390E+03	0.
		4.817E-05	1.853E-04		1.381E+03	

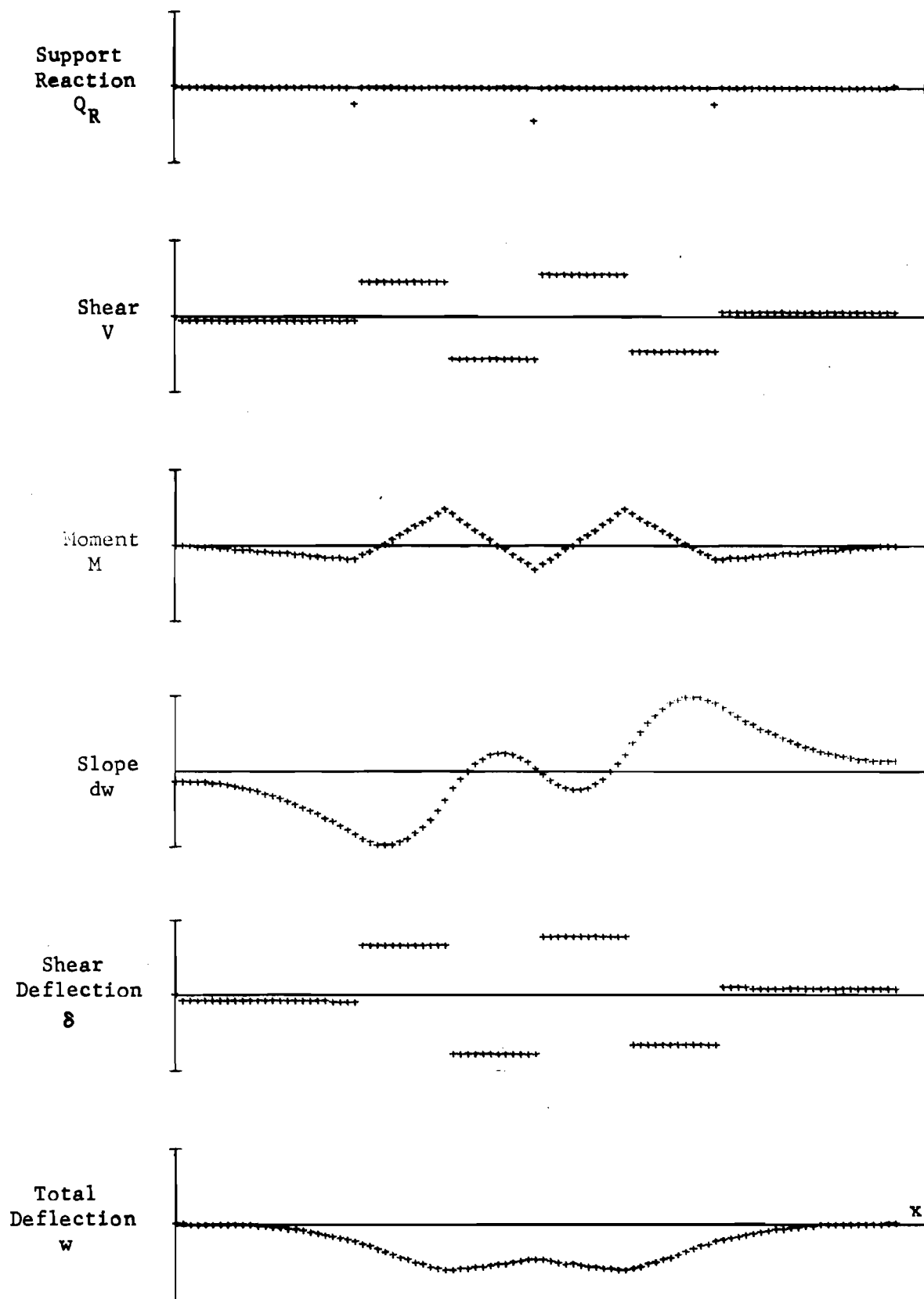
83	-2.234E-03	4.788E-05	1.689E-04	-5.157E+04	1.373E+03	0.
84	-1.775E-03	4.758E-05	1.537E-04	-4.745E+04	1.364E+03	0.
85	-1.362E-03	4.728E-05	1.398E-04	-4.335E+04	1.356E+03	0.
86	-9.899E-04	4.699E-05	1.272E-04	-3.929E+04	1.347E+03	0.
87	-6.552E-04	4.669E-05	1.160E-04	-3.525E+04	1.339E+03	0.
88	-3.540E-04	4.640E-05	1.060E-04	-3.123E+04	1.330E+03	0.
89	-8.248E-05	4.610E-05	9.726E-05	-2.724E+04	1.322E+03	0.
90	1.632E-04	4.581E-05	8.981E-05	-2.327E+04	1.313E+03	0.
91	3.868E-04	4.551E-05	8.363E-05	-1.933E+04	1.305E+03	0.
92	5.922E-04	4.521E-05	7.870E-05	-1.542E+04	1.296E+03	0.
93	7.831E-04	4.492E-05	7.501E-05	-1.153E+04	1.288E+03	0.
94	9.632E-04	4.462E-05	7.256E-05	-7.664E+03	1.279E+03	0.
95	1.136E-03	4.433E-05	7.133E-05	-3.826E+03	1.271E+03	0.
96	1.306E-03	3.653E-19	7.133E-05	-6.563E+00	1.047E-11	1.267E+03
97	1.520E-03			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	3.000E+02	5.000E-02	5.000E-04	5.000E-04	5.000E+05	2.000E+04	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 3.573 SECONDS

ELAPSED CPU TIME = 0 MINUTES 35.332 SECONDS



Prob 203A. Steel bridge diaphragms with wheel load at center of diaphragm.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
203B SAME AS PROBLEM 203A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	0	1	0	1
NUM CARDS INPJT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	3.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
	NONE

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	-0	3.000E+07	-0.	3.126E+02	-0.
1	96	-0	-0.	1.100E+07	-0.	7.820E+00

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
ADDITIONAL DATA INPUT FOR THIS PROBLEM				
1	96	-0	1.000E+99	-0.

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
------	----	-------	---	---	---	---	---

DATA HELD FROM THE PREVIOUS PROBLEM

0	0	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
24	24	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
48	48	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
72	72	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
96	96	-0	-0.	9.700E+05	-0.	1.840E+05	-0.
0	96	-0	-8.480E+00	-0.	-0.	-0.	-0.
72	72	-0	-2.080E+04	-0.	-0.	-0.	-0.
96	96	-0	-2.080E+04	-0.	-0.	-0.	-0.
96	96	-0	2.080E+04	-0.	-0.	-0.	-0.
48	48	-0	-2.080E+04	-0.	-0.	-0.	-0.
48	48	-0	2.080E+04	-0.	-0.	-0.	-0.
72	72	-0	2.080E+04	-0.	-0.	-0.	-0.
36	36	-0	-2.080E+04	-0.	-0.	-0.	-0.
60	60	-0	-2.080E+04	-0.	-0.	-0.	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
 203B SAME AS PROBLEM 203A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	1.478E-03	0.	-5.438E-05	0.	0.	0.
0	1.315E-03	-1.279E-96	-5.438E-05	-5.003E+00	-1.279E+03	1.275E+03
1	1.151E-03	-1.288E-96	-5.562E-05	-3.848E+03	-1.288E+03	0.
2	9.845E-04	-1.296E-96	-5.808E-05	-7.711E+03	-1.296E+03	0.
3	8.103E-04	-1.305E-96	-6.179E-05	-1.160E+04	-1.305E+03	0.
4	6.249E-04	-1.313E-96	-6.676E-05	-1.551E+04	-1.313E+03	0.
5	4.246E-04	-1.322E-96	-7.298E-05	-1.945E+04	-1.322E+03	0.
6	2.057E-04	-1.330E-96	-8.047E-05	-2.342E+04	-1.330E+03	0.
7	-3.574E-05	-1.339E-96	-8.924E-05	-2.741E+04	-1.339E+03	0.
8	-3.035E-04	-1.347E-96	-9.929E-05	-3.143E+04	-1.347E+03	0.
9	-6.013E-04	-1.356E-96	-1.106E-04	-3.547E+04	-1.356E+03	0.
10	-9.333E-04	-1.364E-96	-1.233E-04	-3.953E+04	-1.364E+03	0.
11	-1.303E-03	-1.373E-96	-1.372E-04	-4.363E+04	-1.373E+03	0.
12	-1.715E-03	-1.381E-96	-1.525E-04	-4.774E+04	-1.381E+03	0.
13	-2.172E-03	-1.390E-96	-1.691E-04	-5.189E+04	-1.390E+03	0.
14	-2.680E-03	-1.398E-96	-1.870E-04	-5.606E+04	-1.398E+03	0.
15	-3.241E-03	-1.407E-96	-2.063E-04	-6.025E+04	-1.407E+03	0.
16	-3.860E-03	-1.415E-96	-2.269E-04	-6.447E+04	-1.415E+03	0.
17	-4.541E-03	-1.423E-96	-2.489E-04	-6.872E+04	-1.423E+03	0.
18	-5.287E-03	-1.432E-96	-2.723E-04	-7.299E+04	-1.432E+03	0.
19	-6.104E-03	-1.440E-96	-2.970E-04	-7.728E+04	-1.440E+03	0.
20	-6.995E-03	-1.449E-96	-3.231E-04	-8.160E+04	-1.449E+03	0.
21	-7.965E-03	-1.457E-96	-3.506E-04	-8.595E+04	-1.457E+03	0.

22	-9.016E-03			-9.032E+04		0.
23	-1.015E-02	-1.466E-96	-3.795E-04	-9.472E+04	-1.466E+03	0.
24	-1.138E-02	-1.474E-96	-4.098E-04	-9.918E+04	-1.474E+03	-1.104E+04
25	-1.271E-02	9.560E-96	-4.415E-04	-7.054E+04	9.560E+03	0.
26	-1.410E-02	9.551E-96	-4.641E-04	-4.189E+04	9.551E+03	0.
27	-1.553E-02	9.543E-96	-4.775E-04	-1.326E+04	9.543E+03	0.
28	-1.698E-02	9.534E-96	-4.817E-04	1.534E+04	9.534E+03	0.
29	-1.841E-02	9.526E-96	-4.768E-04	4.392E+04	9.526E+03	0.
30	-1.980E-02	9.517E-96	-4.628E-04	7.247E+04	9.517E+03	0.
31	-2.112E-02	9.509E-96	-4.396E-04	1.010E+05	9.509E+03	0.
32	-2.234E-02	9.500E-96	-4.073E-04	1.295E+05	9.500E+03	0.
33	-2.344E-02	9.492E-96	-3.658E-04	1.580E+05	9.492E+03	0.
34	-2.438E-02	9.483E-96	-3.153E-04	1.864E+05	9.483E+03	0.
35	-2.515E-02	9.475E-96	-2.557E-04	2.149E+05	9.475E+03	0.
36	-2.571E-02	9.467E-96	-1.869E-04	2.433E+05	9.467E+03	0.
37	-2.604E-02	-1.134E-95	-1.091E-04	2.092E+05	-1.134E+04	0.
38	-2.616E-02	-1.135E-95	-4.219E-05	1.752E+05	-1.135E+04	0.
39	-2.617E-02	-1.136E-95	1.385E-05	1.411E+05	-1.136E+04	0.
40	-2.594E-02	-1.137E-95	5.899E-05	1.070E+05	-1.137E+04	0.
41	-2.566E-02	-1.138E-95	9.321E-05	7.287E+04	-1.138E+04	0.
42	-2.532E-02	-1.138E-95	1.165E-04	3.872E+04	-1.138E+04	0.
43	-2.493E-02	-1.139E-95	1.289E-04	4.538E+03	-1.139E+04	0.
44	-2.454E-02	-1.140E-95	1.304E-04	-2.967E+04	-1.140E+04	0.
45	-2.417E-02	-1.141E-95	1.209E-04	-6.390E+04	-1.141E+04	0.
46	-2.387E-02	-1.142E-95	1.004E-04	-9.815E+04	-1.142E+04	0.
47	-2.367E-02	-1.143E-95	6.903E-05	-1.324E+05	-1.143E+04	0.
48	-2.359E-02	-1.144E-95	2.667E-05	-1.667E+05	-1.144E+04	-2.288E+04
49	-2.367E-02	1.144E-95	-2.667E-05	-1.324E+05	1.144E+04	0.
50	-2.387E-02	1.143E-95	-6.903E-05	-9.815E+04	1.143E+04	0.
51	-2.417E-02	1.142E-95	-1.004E-04	-6.390E+04	1.142E+04	0.
52	-2.454E-02	1.141E-95	-1.209E-04	-2.967E+04	1.141E+04	0.

53	-2.493E-02	1.140E-95	-1.304E-04	4.538E+03	1.140E+04	0.
		1.139E-95	-1.289E-04		1.139E+04	
54	-2.532E-02			3.872E+04		0.
		1.138E-95	-1.165E-04		1.138E+04	
55	-2.566E-02			7.287E+04		0.
		1.138E-95	-9.321E-05		1.138E+04	
56	-2.594E-02			1.070E+05		0.
		1.137E-95	-5.899E-05		1.137E+04	
57	-2.612E-02			1.411E+05		0.
		1.136E-95	-1.385E-05		1.136E+04	
58	-2.616E-02			1.752E+05		0.
		1.135E-95	4.219E-05		1.135E+04	
59	-2.604E-02			2.092E+05		0.
		1.134E-95	1.091E-04		1.134E+04	
60	-2.571E-02			2.433E+05		0.
		-9.467E-96	1.869E-04		-9.467E+03	
61	-2.515E-02			2.149E+05		0.
		-9.475E-96	2.557E-04		-9.475E+03	
62	-2.438E-02			1.864E+05		0.
		-9.483E-96	3.153E-04		-9.483E+03	
63	-2.344E-02			1.580E+05		0.
		-9.492E-96	3.658E-04		-9.492E+03	
64	-2.234E-02			1.295E+05		0.
		-9.500E-96	4.073E-04		-9.500E+03	
65	-2.112E-02			1.010E+05		0.
		-9.509E-96	4.396E-04		-9.509E+03	
66	-1.980E-02			7.247E+04		0.
		-9.517E-96	4.628E-04		-9.517E+03	
67	-1.841E-02			4.392E+04		0.
		-9.526E-96	4.768E-04		-9.526E+03	
68	-1.698E-02			1.534E+04		0.
		-9.534E-96	4.817E-04		-9.534E+03	
69	-1.553E-02			-1.326E+04		0.
		-9.543E-96	4.775E-04		-9.543E+03	
70	-1.410E-02			-4.189E+04		0.
		-9.551E-96	4.641E-04		-9.551E+03	
71	-1.271E-02			-7.054E+04		0.
		-9.560E-96	4.415E-04		-9.560E+03	
72	-1.138E-02			-9.918E+04		-1.104E+04
		1.474E-96	4.098E-04		1.474E+03	
73	-1.015E-02			-9.472E+04		0.
		1.466E-96	3.795E-04		1.466E+03	
74	-9.016E-03			-9.032E+04		0.
		1.457E-96	3.506E-04		1.457E+03	
75	-7.965E-03			-8.595E+04		0.
		1.449E-96	3.231E-04		1.449E+03	
76	-6.995E-03			-8.160E+04		0.
		1.440E-96	2.970E-04		1.440E+03	
77	-6.104E-03			-7.728E+04		0.
		1.432E-96	2.723E-04		1.432E+03	
78	-5.297E-03			-7.294E+04		0.
		1.423E-96	2.489E-04		1.423E+03	
79	-4.541E-03			-6.872E+04		0.
		1.415E-96	2.269E-04		1.415E+03	
80	-3.860E-03			-6.447E+04		0.
		1.407E-96	2.063E-04		1.407E+03	
81	-3.241E-03			-6.025E+04		0.
		1.398E-96	1.870E-04		1.398E+03	
82	-2.640E-03			-5.606E+04		0.
		1.390E-96	1.691E-04		1.390E+03	

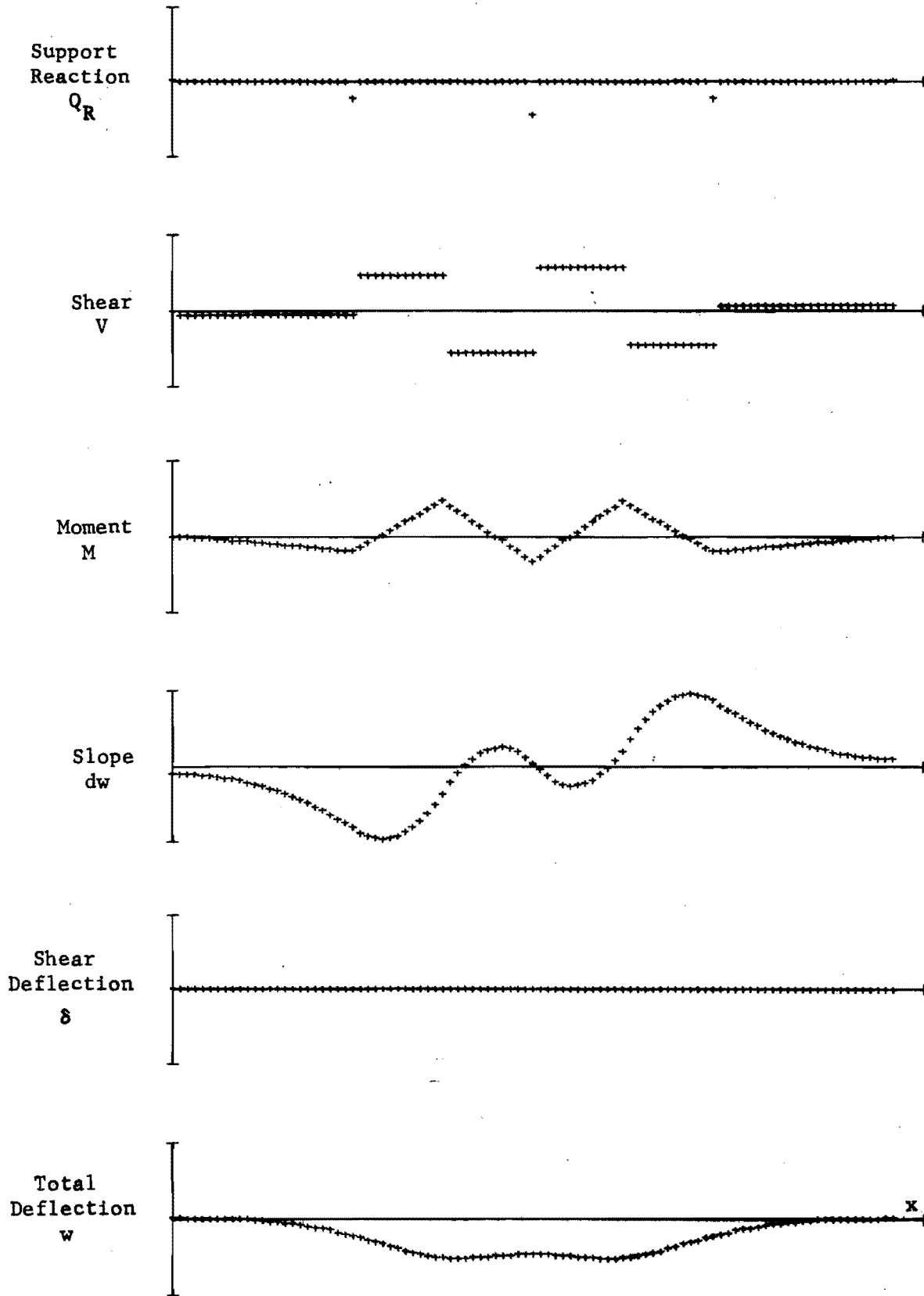
83	-2.172E-03			-5.189E+04		0.
		1.381E-96	1.525E-04		1.381E+03	0.
84	-1.715E-03			-4.774E+04		0.
		1.373E-96	1.372E-04		1.373E+03	0.
85	-1.303E-03			-4.363E+04		0.
		1.364E-96	1.233E-04		1.364E+03	0.
86	-9.333E-04			-3.953E+04		0.
		1.356E-96	1.106E-04		1.356E+03	0.
87	-6.013E-04			-3.547E+04		0.
		1.347E-96	9.929E-05		1.347E+03	0.
88	-3.035E-04			-3.143E+04		0.
		1.339E-96	8.924E-05		1.339E+03	0.
89	-3.574E-05			-2.741E+04		0.
		1.330E-96	8.047E-05		1.330E+03	0.
90	2.057E-04			-2.342E+04		0.
		1.322E-96	7.298E-05		1.322E+03	0.
91	4.246E-04			-1.945E+04		0.
		1.313E-96	6.676E-05		1.313E+03	0.
92	6.249E-04			-1.551E+04		0.
		1.305E-96	6.179E-05		1.305E+03	0.
93	8.103E-04			-1.160E+04		0.
		1.296E-96	5.808E-05		1.296E+03	0.
94	9.845E-04			-7.711E+03		0.
		1.288E-96	5.562E-05		1.288E+03	0.
95	1.151E-03			-3.848E+03		0.
		1.279E-96	5.438E-05		1.279E+03	0.
96	1.315E-03			-5.003E+00		1.275E+03
		1.074-110	5.438E-05		1.074E-11	0.
97	1.478E-03			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	3.000E+02	5.000E-02	1.000E+00	5.000E-04	5.000E+05	2.000E+04	5.000E+04

TIME FOR THIS PROBLEM = 0 MINUTES 3.533 SECONDS

ELAPSED CPU TIME = 0 MINUTES 38.865 SECONDS



Prob 203B. Same as Prob 203A, neglecting shear deformations.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
301A TAPERED THREE-SPAN OVERHANGING BEAM WITH 300 LB LOAD AT STA 0

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	-0	-0	-0	-0	-0
NUM CARDS INPUT THIS PROBLEM	1	2	22	0	1
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

NUM INCREMENTS	96
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
36	0.
60	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
ADDITIONAL DATA INPUT FOR THIS PROBLEM						
0	96	0	1.000E+07	-0.	-0.	-0.
1	96	0	-0.	4.000E+06	-0.	-0.
0	0	0	-0.	-0.	5.546E-01	-0.
1	1	1	-0.	-0.	7.421E-01	7.200E-02
	3	1	-0.	-0.	1.193E+00	8.740E-02
	6	1	-0.	-0.	2.034E+00	1.090E-01
	10	1	-0.	-0.	3.395E+00	1.360E-01
	14	1	-0.	-0.	4.937E+00	1.600E-01
	18	1	-0.	-0.	6.566E+00	1.810E-01
	23	1	-0.	-0.	8.592E+00	2.040E-01
	28	1	-0.	-0.	1.047E+01	2.220E-01
	35	1	-0.	-0.	1.260E+01	2.420E-01
	48	1	-0.	-0.	1.429E+01	2.560E-01
	61	1	-0.	-0.	1.260E+01	2.420E-01
	68	1	-0.	-0.	1.047E+01	2.220E-01
	73	1	-0.	-0.	8.592E+00	2.040E-01
	78	1	-0.	-0.	6.566E+00	1.810E-01
	82	1	-0.	-0.	4.937E+00	1.600E-01

86	1	-0.	-0.	3.395E+00	1.360E-01
90	1	-0.	-0.	2.034E+00	1.090E-01
93	1	-0.	-0.	1.193E+00	8.740E-02
96	-0	-0.	-0.	5.546E-01	6.440E-02

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
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NONE

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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ADDITIONAL DATA INPUT FOR THIS PROBLEM

0	0	-0	3.000E+02	-0.	-0.	-0.	-0.
---	---	----	-----------	-----	-----	-----	-----

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)

301A TAPERED THREE-SPAN OVERHANGING BEAM WITH 300 LB LOAD AT STA 0

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	1.142E-01			0.		0.
0	1.102E-01	0.	-3.984E-03	-1.078E-09	0.	0.
1	1.052E-01	1.042E-03	-3.984E-03	3.000E+02	3.000E+02	0.
2	1.003E-01	9.410E-04	-3.944E-03	6.000E+02	3.000E+02	0.
3	9.556E-02	8.581E-04	-3.882E-03	9.000E+02	3.000E+02	0.
4	9.097E-02	7.928E-04	-3.806E-03	1.200E+03	3.000E+02	0.
5	8.650E-02	7.367E-04	-3.725E-03	1.500E+03	3.000E+02	0.
6	8.219E-02	6.881E-04	-3.639E-03	1.800E+03	3.000E+02	0.
7	7.798E-02	6.479E-04	-3.551E-03	2.100E+03	3.000E+02	0.
8	7.390E-02	6.122E-04	-3.462E-03	2.400E+03	3.000E+02	0.
9	6.995E-02	5.803E-04	-3.374E-03	2.700E+03	3.000E+02	0.
10	6.611E-02	5.515E-04	-3.285E-03	3.000E+03	3.000E+02	0.
11	6.239E-02	5.282E-04	-3.197E-03	3.300E+03	3.000E+02	0.
12	5.877E-02	5.068E-04	-3.110E-03	3.600E+03	3.000E+02	0.
13	5.526E-02	4.870E-04	-3.023E-03	3.900E+03	3.000E+02	0.
14	5.185E-02	4.687E-04	-2.938E-03	4.200E+03	3.000E+02	0.
15	4.855E-02	4.539E-04	-2.853E-03	4.500E+03	3.000E+02	0.
16	4.534E-02	4.399E-04	-2.768E-03	4.800E+03	3.000E+02	0.
17	4.223E-02	4.267E-04	-2.685E-03	5.100E+03	3.000E+02	0.
18	3.921E-02	4.144E-04	-2.602E-03	5.400E+03	3.000E+02	0.
19	3.629E-02	4.041E-04	-2.520E-03	5.700E+03	3.000E+02	0.
20	3.346E-02	3.943E-04	-2.438E-03	6.000E+03	3.000E+02	0.
21	3.071E-02	3.850E-04	-2.357E-03	6.300E+03	3.000E+02	0.
		3.761E-04	-2.276E-03		3.000E+02	

22	2.806E-02			6.600E+03		0.
23	2.550E-02	3.676E-04	-2.195E-03	6.900E+03	3.000E+02	0.
24	2.302E-02	3.613E-04	-2.115E-03	7.200E+03	3.000E+02	0.
25	2.063E-02	3.551E-04	-2.035E-03	7.500E+03	3.000E+02	0.
26	1.833E-02	3.492E-04	-1.954E-03	7.800E+03	3.000E+02	0.
27	1.611E-02	3.434E-04	-1.874E-03	8.100E+03	3.000E+02	0.
28	1.398E-02	3.378E-04	-1.794E-03	8.400E+03	3.000E+02	0.
29	1.193E-02	3.335E-04	-1.714E-03	8.700E+03	3.000E+02	0.
30	9.972E-03	3.294E-04	-1.633E-03	9.000E+03	3.000E+02	0.
31	8.095E-03	3.253E-04	-1.552E-03	9.300E+03	3.000E+02	0.
32	6.304E-03	3.213E-04	-1.470E-03	9.600E+03	3.000E+02	0.
33	4.598E-03	3.174E-04	-1.388E-03	9.900E+03	3.000E+02	0.
34	2.980E-03	3.136E-04	-1.305E-03	1.020E+04	3.000E+02	0.
35	1.447E-03	3.099E-04	-1.222E-03	1.050E+04	3.000E+02	0.
36	0.	3.085E-04	-1.139E-03	1.080E+04	3.000E+02	-7.500E+02
37	-5.933E-04	-4.608E-04	-1.054E-03	1.035E+04	-4.500E+02	0.
38	-1.108E-03	-4.588E-04	-9.736E-04	9.900E+03	-4.500E+02	0.
39	-1.549E-03	-4.567E-04	-8.974E-04	9.450E+03	-4.500E+02	0.
40	-1.919E-03	-4.548E-04	-8.253E-04	9.000E+03	-4.500E+02	0.
41	-2.224E-03	-4.528E-04	-7.574E-04	8.550E+03	-4.500E+02	0.
42	-2.467E-03	-4.508E-04	-6.935E-04	8.100E+03	-4.500E+02	0.
43	-2.651E-03	-4.489E-04	-6.335E-04	7.650E+03	-4.500E+02	0.
44	-2.782E-03	-4.470E-04	-5.775E-04	7.200E+03	-4.500E+02	0.
45	-2.862E-03	-4.451E-04	-5.252E-04	6.750E+03	-4.500E+02	0.
46	-2.895E-03	-4.432E-04	-4.766E-04	6.300E+03	-4.500E+02	0.
47	-2.886E-03	-4.413E-04	-4.317E-04	5.850E+03	-4.500E+02	0.
48	-2.837E-03	-4.395E-04	-3.904E-04	5.400E+03	-4.500E+02	0.
49	-2.748E-03	-4.413E-04	-3.526E-04	4.950E+03	-4.500E+02	0.
50	-2.622E-03	-4.432E-04	-3.176E-04	4.500E+03	-4.500E+02	0.
51	-2.463E-03	-4.451E-04	-2.856E-04	4.050E+03	-4.500E+02	0.
52	-2.272E-03	-4.470E-04	-2.564E-04	3.600E+03	-4.500E+02	0.

		-4.489E-04	-2.303E-04		-4.500E+02	
53	-2.054E-03			3.150E+03		0.
		-4.508E-04	-2.072E-04		-4.500E+02	
54	-1.410E-03			2.700E+03		0.
		-4.528E-04	-1.872E-04		-4.500E+02	
55	-1.544E-03			2.250E+03		0.
		-4.548E-04	-1.704E-04		-4.500E+02	
56	-1.260E-03			1.800E+03		0.
		-4.567E-04	-1.568E-04		-4.500E+02	
57	-9.601E-04			1.350E+03		0.
		-4.588E-04	-1.465E-04		-4.500E+02	
58	-6.479E-04			9.000E+02		0.
		-4.608E-04	-1.396E-04		-4.500E+02	
59	-3.267E-04			4.500E+02		0.
		-4.628E-04	-1.361E-04		-4.500E+02	
60	0.			-4.124E-09		4.500E+02
		-1.382E-16	-1.361E-04		-1.338E-10	
61	-1.361E-04			-3.286E-09		0.
		-1.383E-16	-1.361E-04		-1.323E-10	
62	-2.722E-04			-3.201E-09		0.
		-1.366E-16	-1.361E-04		-1.291E-10	
63	-4.083E-04			-3.095E-09		0.
		-1.381E-16	-1.361E-04		-1.289E-10	
64	-5.444E-04			-4.247E-09		0.
		-1.362E-16	-1.361E-04		-1.257E-10	
65	-6.805E-04			-3.742E-09		0.
		-1.344E-16	-1.361E-04		-1.224E-10	
66	-8.166E-04			-4.378E-09		0.
		-1.358E-16	-1.361E-04		-1.221E-10	
67	-9.527E-04			-4.703E-09		0.
		-1.309E-16	-1.361E-04		-1.162E-10	
68	-1.089E-03			-5.838E-09		0.
		-1.242E-16	-1.361E-04		-1.085E-10	
69	-1.225E-03			-6.336E-09		0.
		-1.176E-16	-1.361E-04		-1.010E-10	
70	-1.361E-03			-7.458E-09		0.
		-1.110E-16	-1.361E-04		-9.380E-11	
71	-1.497E-03			-4.587E-09		0.
		-1.046E-16	-1.361E-04		-8.687E-11	
72	-1.633E-03			-4.965E-09		0.
		-1.045E-16	-1.361E-04		-8.524E-11	
73	-1.769E-03			-6.691E-09		0.
		-9.907E-17	-1.361E-04		-7.902E-11	
74	-1.905E-03			-5.754E-09		0.
		-9.996E-17	-1.361E-04		-7.789E-11	
75	-2.041E-03			-9.802E-09		0.
		-1.010E-16	-1.361E-04		-7.686E-11	
76	-2.177E-03			-6.789E-09		0.
		-9.592E-17	-1.361E-04		-7.121E-11	
77	-2.314E-03			-6.935E-09		0.
		-8.438E-17	-1.361E-04		-6.109E-11	
78	-2.450E-03			-5.628E-09		0.
		-7.294E-17	-1.361E-04		-5.128E-11	
79	-2.586E-03			-3.531E-09		0.
		-7.476E-17	-1.361E-04		-5.099E-11	
80	-2.722E-03			-4.913E-09		0.
		-7.691E-17	-1.361E-04		-5.084E-11	
81	-2.858E-03			-4.972E-09		0.
		-8.668E-17	-1.361E-04		-5.547E-11	
82	-2.994E-03			-4.962E-09		0.
		-9.004E-17	-1.361E-04		-5.547E-11	

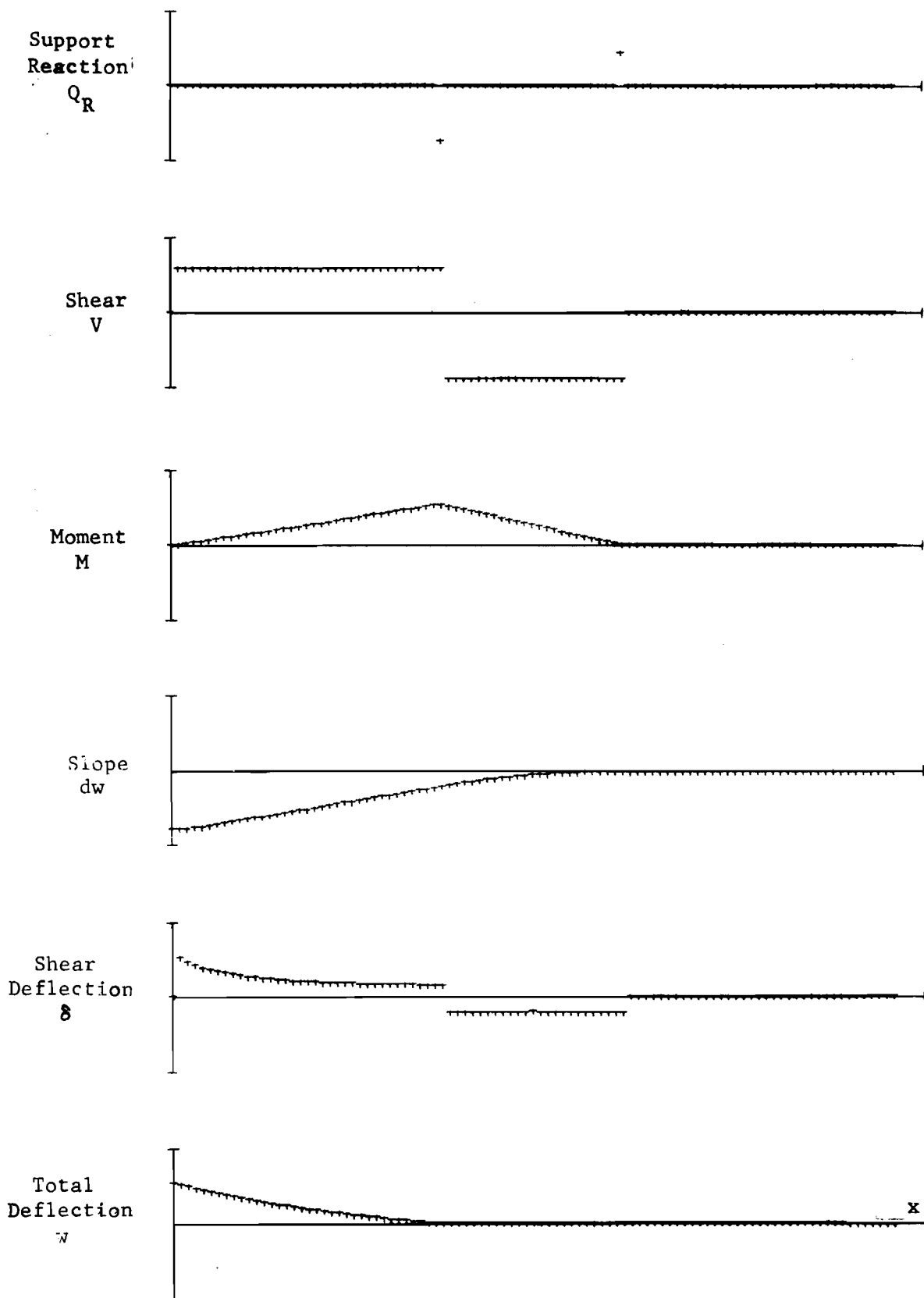
83	-3.130E-03			-1.090E-09		0.
		-8.624E-17	-1.361E-04		-5.105E-11	
84	-3.266E-03			-3.312E-09		0.
		-8.247E-17	-1.361E-04		-4.684E-11	
85	-3.402E-03			-2.481E-09		0.
		-7.870E-17	-1.361E-04		-4.281E-11	
86	-3.538E-03			-1.266E-09		0.
		-7.435E-17	-1.361E-04		-3.844E-11	
87	-3.675E-03			-1.712E-09		0.
		-7.488E-17	-1.361E-04		-3.669E-11	
88	-3.811E-03			-1.244E-09		0.
		-6.519E-17	-1.361E-04		-3.018E-11	
89	-3.947E-03			2.641E-10		0.
		-5.407E-17	-1.361E-04		-2.357E-11	
90	-4.083E-03			1.576E-10		0.
		-4.632E-17	-1.361E-04		-1.886E-11	
91	-4.219E-03			-1.301E-09		0.
		-3.725E-17	-1.361E-04		-1.410E-11	
92	-4.355E-03			-1.187E-09		0.
		-3.454E-17	-1.361E-04		-1.207E-11	
93	-4.491E-03			-8.176E-10		0.
		-1.980E-17	-1.361E-04		-6.316E-12	
94	-4.627E-03			-5.820E-10		0.
		-2.366E-17	-1.361E-04		-6.822E-12	
95	-4.763E-03			-1.016E-10		0.
		-9.147E-18	-1.361E-04		-2.356E-12	
96	-4.899E-03			1.786E-10		0.
		-2.583E-19	-1.361E-04		-6.653E-14	
97	-5.035E-03			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	1.000E+02	2.000E-01	2.000E-03	5.000E-03	2.000E+04	5.000E+02	1.000E+03

TIME FOR THIS PROBLEM = 0 MINUTES 3.599 SECONDS

ELAPSED CPU TIME = 0 MINUTES 23.413 SECONDS



Prob 301A. Tapered three-span overhanging beam with 300-lb load at Sta 0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
301B SAME AS PROBLEM 301A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
36	0.
60	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	0	1.000E+07	-0.	-0.	-0.
1	96	0	-0.	4.000E+06	-0.	-0.
0	0	0	-0.	-0.	5.546E-01	-0.
1		1	-0.	-0.	7.421E-01	7.200E-02
	3	1	-0.	-0.	1.193E+00	8.740E-02
	6	1	-0.	-0.	2.034E+00	1.090E-01
	10	1	-0.	-0.	3.395E+00	1.360E-01
	14	1	-0.	-0.	4.937E+00	1.600E-01
	18	1	-0.	-0.	6.566E+00	1.810E-01
	23	1	-0.	-0.	8.592E+00	2.040E-01
	28	1	-0.	-0.	1.047E+01	2.220E-01
	35	1	-0.	-0.	1.260E+01	2.420E-01
	48	1	-0.	-0.	1.429E+01	2.560E-01
	61	1	-0.	-0.	1.260E+01	2.420E-01
	68	1	-0.	-0.	1.047E+01	2.220E-01
	73	1	-0.	-0.	8.592E+00	2.040E-01

78	1	-0.	-0.	6.566E+00	1.810E-01
82	1	-0.	-0.	4.937E+00	1.600E-01
86	1	-0.	-0.	3.395E+00	1.360E-01
90	1	-0.	-0.	2.034E+00	1.090E-01
93	1	-0.	-0.	1.193E+00	8.740E-02
96	-0	-0.	-0.	5.546E-01	6.440E-02

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
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ADDITIONAL DATA INPUT FOR THIS PROBLEM

1	96	-0	1.000E+99	-0.
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TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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DATA HELD FROM THE PREVIOUS PROBLEM

0	0	-0	3.000E+02	-0.	-0.	-0.	-0.
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SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
 301B SAME AS PROBLEM 301A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	8.010E-02	0.	-3.533E-03	0.	0.	0.
0	7.657E-02	3.000E-97	-3.533E-03	-2.463E-09	3.000E+02	0.
1	7.304E-02	3.000E-97	-3.493E-03	3.000E+02	3.000E+02	0.
2	6.955E-02	3.000E-97	-3.431E-03	6.000E+02	3.000E+02	0.
3	6.612E-02	3.000E-97	-3.355E-03	9.000E+02	3.000E+02	0.
4	6.276E-02	3.000E-97	-3.274E-03	1.200E+03	3.000E+02	0.
5	5.949E-02	3.000E-97	-3.188E-03	1.500E+03	3.000E+02	0.
6	5.630E-02	3.000E-97	-3.100E-03	1.800E+03	3.000E+02	0.
7	5.320E-02	3.000E-97	-3.011E-03	2.100E+03	3.000E+02	0.
8	5.019E-02	3.000E-97	-2.923E-03	2.400E+03	3.000E+02	0.
9	4.726E-02	3.000E-97	-2.834E-03	2.700E+03	3.000E+02	0.
10	4.443E-02	3.000E-97	-2.746E-03	3.000E+03	3.000E+02	0.
11	4.168E-02	3.000E-97	-2.659E-03	3.300E+03	3.000E+02	0.
12	3.903E-02	3.000E-97	-2.572E-03	3.600E+03	3.000E+02	0.
13	3.645E-02	3.000E-97	-2.487E-03	3.900E+03	3.000E+02	0.
14	3.397E-02	3.000E-97	-2.402E-03	4.200E+03	3.000E+02	0.
15	3.156E-02	3.000E-97	-2.317E-03	4.500E+03	3.000E+02	0.
16	2.925E-02	3.000E-97	-2.234E-03	4.800E+03	3.000E+02	0.
17	2.701E-02	3.000E-97	-2.151E-03	5.100E+03	3.000E+02	0.
18	2.486E-02	3.000E-97	-2.069E-03	5.400E+03	3.000E+02	0.
19	2.279E-02	3.000E-97	-1.987E-03	5.700E+03	3.000E+02	0.
20	2.081E-02	3.000E-97	-1.906E-03	6.000E+03	3.000E+02	0.
21	1.890E-02	3.000E-97	-1.825E-03	6.300E+03	3.000E+02	0.

22	1.708E-02			6.600E+03		0.
23	1.533E-02	3.000E-97	-1.744E-03	6.900E+03	3.000E+02	0.
24	1.367E-02	3.000E-97	-1.664E-03	7.200E+03	3.000E+02	0.
25	1.208E-02	3.000E-97	-1.584E-03	7.500E+03	3.000E+02	0.
26	1.058E-02	3.000E-97	-1.503E-03	7.800E+03	3.000E+02	0.
27	9.157E-03	3.000E-97	-1.423E-03	8.100E+03	3.000E+02	0.
28	7.814E-03	3.000E-97	-1.343E-03	8.400E+03	3.000E+02	0.
29	6.552E-03	3.000E-97	-1.263E-03	8.700E+03	3.000E+02	0.
30	5.370E-03	3.000E-97	-1.182E-03	9.000E+03	3.000E+02	0.
31	4.269E-03	3.000E-97	-1.101E-03	9.300E+03	3.000E+02	0.
32	3.250E-03	3.000E-97	-1.019E-03	9.600E+03	3.000E+02	0.
33	2.313E-03	3.000E-97	-9.368E-04	9.900E+03	3.000E+02	0.
34	1.459E-03	3.000E-97	-8.542E-04	1.020E+04	3.000E+02	0.
35	6.880E-04	3.000E-97	-7.713E-04	1.050E+04	3.000E+02	0.
36	0.	3.000E-97	-6.880E-04	1.080E+04	3.000E+02	0.
37	-6.031E-04	-4.500E-97	-6.031E-04	1.035E+04	-4.500E+02	-7.500E+02
38	-1.126E-03	-4.500E-97	-5.226E-04	9.900E+03	-4.500E+02	0.
39	-1.572E-03	-4.500E-97	-4.464E-04	9.450E+03	-4.500E+02	0.
40	-1.947E-03	-4.500E-97	-3.744E-04	9.000E+03	-4.500E+02	0.
41	-2.253E-03	-4.500E-97	-3.065E-04	8.550E+03	-4.500E+02	0.
42	-2.496E-03	-4.500E-97	-2.426E-04	8.100E+03	-4.500E+02	0.
43	-2.678E-03	-4.500E-97	-1.826E-04	7.650E+03	-4.500E+02	0.
44	-2.805E-03	-4.500E-97	-1.265E-04	7.200E+03	-4.500E+02	0.
45	-2.879E-03	-4.500E-97	-7.424E-05	6.750E+03	-4.500E+02	0.
46	-2.905E-03	-4.500E-97	-2.567E-05	6.300E+03	-4.500E+02	0.
47	-2.885E-03	-4.500E-97	1.923E-05	5.850E+03	-4.500E+02	0.
48	-2.825E-03	-4.500E-97	6.054E-05	5.400E+03	-4.500E+02	0.
49	-2.727E-03	-4.500E-97	9.833E-05	4.950E+03	-4.500E+02	0.
50	-2.593E-03	-4.500E-97	1.333E-04	4.500E+03	-4.500E+02	0.
51	-2.428E-03	-4.500E-97	1.654E-04	4.050E+03	-4.500E+02	0.
52	-2.233E-03	-4.500E-97	1.945E-04	3.600E+03	-4.500E+02	0.

53	-2.013E-03	-4.500E-97	2.206E-04	3.150E+03	-4.500E+02	0.
54	-1.769E-03	-4.500E-97	2.437E-04	2.700E+03	-4.500E+02	0.
55	-1.505E-03	-4.500E-97	2.637E-04	2.250E+03	-4.500E+02	0.
56	-1.225E-03	-4.500E-97	2.805E-04	1.800E+03	-4.500E+02	0.
57	-9.306E-04	-4.500E-97	2.941E-04	1.350E+03	-4.500E+02	0.
58	-6.262E-04	-4.500E-97	3.044E-04	9.000E+02	-4.500E+02	0.
59	-3.148E-04	-4.500E-97	3.113E-04	4.500E+02	-4.500E+02	0.
60	0.	-4.500E-97	3.148E-04	9.054E-09	-4.500E+02	4.500E+02
61	3.148E-04	3.260-108	3.148E-04	1.180E-08	3.260E-09	0.
62	6.297E-04	2.522-108	3.148E-04	1.450E-08	2.522E-09	0.
63	9.445E-04	2.154-108	3.148E-04	1.581E-08	2.154E-09	0.
64	1.259E-03	1.663-108	3.148E-04	1.541E-08	1.663E-09	0.
65	1.574E-03	1.365-108	3.148E-04	1.817E-08	1.365E-09	0.
66	1.889E-03	1.166-108	3.148E-04	1.845E-08	1.166E-09	0.
67	2.204E-03	1.024-108	3.148E-04	1.944E-08	1.024E-09	0.
68	2.519E-03	8.262-109	3.148E-04	1.944E-08	8.262E-10	0.
69	2.834E-03	8.398-109	3.148E-04	2.034E-08	8.398E-10	0.
70	3.148E-03	7.770-109	3.148E-04	1.821E-08	7.770E-10	0.
71	3.463E-03	7.259-109	3.148E-04	2.023E-08	7.259E-10	0.
72	3.778E-03	6.837-109	3.148E-04	2.334E-08	6.837E-10	0.
73	4.093E-03	6.053-109	3.148E-04	1.493E-08	6.053E-10	0.
74	4.408E-03	5.431-109	3.148E-04	1.669E-08	5.431E-10	0.
75	4.723E-03	4.918-109	3.148E-04	1.591E-08	4.918E-10	0.
76	5.037E-03	4.812-109	3.148E-04	1.512E-08	4.812E-10	0.
77	5.352E-03	4.429-109	3.148E-04	1.228E-08	4.429E-10	0.
78	5.667E-03	4.429-109	3.148E-04	1.548E-08	4.106E-10	0.
79	5.982E-03	4.106-109	3.148E-04	1.276E-08	4.106E-10	0.
80	6.297E-03	3.834-109	3.148E-04	1.026E-08	3.834E-10	0.
81	6.612E-03	3.364-109	3.148E-04	9.578E-09	3.364E-10	0.
82	6.927E-03	3.184-109	3.148E-04	1.038E-08	3.184E-10	0.
		3.033-109	3.148E-04	-2.741E-09	3.033E-10	0.
		2.484-109	3.148E-04		2.484E-10	0.

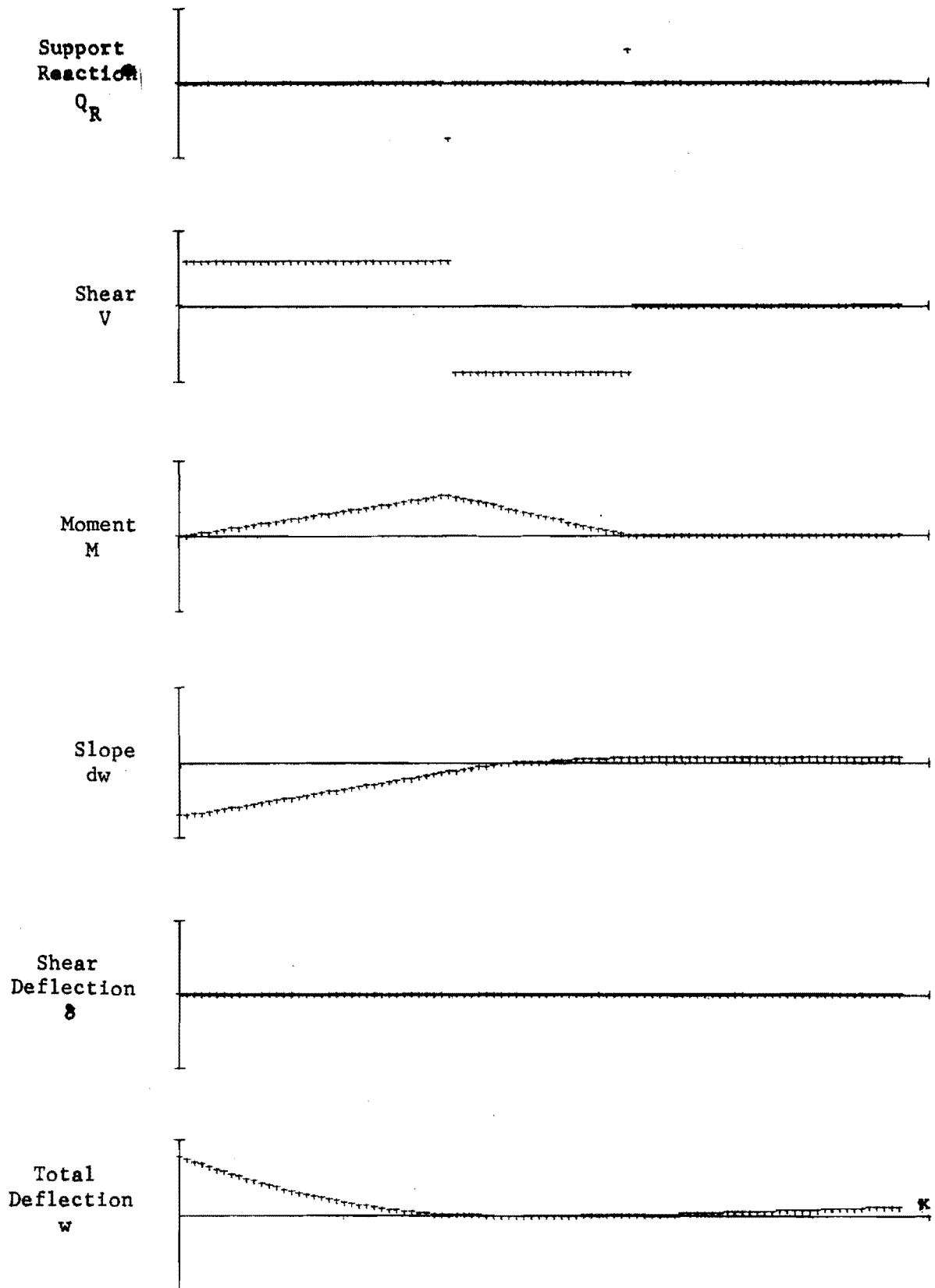
83	7.241E-03			-5.053E-09		0.
84	7.556E-03	2.391-109	3.148E-04	-2.313E-09	2.391E-10	0.
85	7.871E-03	2.125-109	3.148E-04	-6.296E-09	2.125E-10	0.
86	8.186E-03	1.889-109	3.148E-04	-9.423E-09	1.889E-10	0.
87	8.501E-03	1.658-109	3.148E-04	-6.783E-09	1.658E-10	0.
88	8.816E-03	1.451-109	3.148E-04	-6.027E-09	1.451E-10	0.
89	9.130E-03	1.085-109	3.148E-04	-1.054E-08	1.085E-10	0.
90	9.445E-03	9.129-110	3.148E-04	-6.775E-09	9.129E-11	0.
91	9.760E-03	7.302-110	3.148E-04	-3.894E-09	7.302E-11	0.
92	1.007E-02	5.577-110	3.148E-04	-2.454E-09	5.577E-11	0.
93	1.039E-02	3.877-110	3.148E-04	-2.649E-09	3.877E-11	0.
94	1.070E-02	2.969-110	3.148E-04	-8.626-104	2.969E-11	0.
95	1.102E-02	2.089-110	3.148E-04	-1.704E-09	2.089E-11	0.
96	1.133E-02	9.116-111	3.148E-04	-1.539E-10	9.116E-12	0.
97	1.165E-02	1.709-112	3.148E-04	0.	1.709E-13	0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	1.000E+02	1.000E-01	1.000E+00	5.000E-03	2.000E+04	5.000E+02	1.000E+03

TIME FOR THIS PROBLEM = 0 MINUTES 3.532 SECONDS

ELAPSED CPU TIME = 0 MINUTES 26.945 SECONDS



Prob 301B. Same as Prob 301A, neglecting shear deformations.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
302A TAPERED THREE-SPAN OVERHANGING BEAM WITH 500 LB LOAD AT STA 0

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	0
NUM CARDS INPUT THIS PROBLEM	0	0	0	0	1
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
36	0.
60	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	0	1.000E+07	-0.	-0.	-0.
1	96	0	-0.	4.000E+06	-0.	-0.
0	0	0	-0.	-0.	5.546E-01	-0.
1		1	-0.	-0.	7.421E-01	7.200E-02
	3	1	-0.	-0.	1.193E+00	8.740E-02
	6	1	-0.	-0.	2.034E+00	1.090E-01
	10	1	-0.	-0.	3.395E+00	1.360E-01
	14	1	-0.	-0.	4.937E+00	1.600E-01
	18	1	-0.	-0.	6.566E+00	1.810E-01
	23	1	-0.	-0.	8.592E+00	2.040E-01
	29	1	-0.	-0.	1.047E+01	2.220E-01
	35	1	-0.	-0.	1.260E+01	2.420E-01
	48	1	-0.	-0.	1.429E+01	2.560E-01
	61	1	-0.	-0.	1.260E+01	2.420E-01
	68	1	-0.	-0.	1.047E+01	2.220E-01
	73	1	-0.	-0.	8.592E+00	2.040E-01

78	1	-0.	-0.	6.566E+00	1.810E-01
82	1	-0.	-0.	4.937E+00	1.600E-01
86	1	-0.	-0.	3.395E+00	1.360E-01
90	1	-0.	-0.	2.034E+00	1.090E-01
93	1	-0.	-0.	1.193E+00	8.740E-02
96	-0	-0.	-0.	5.546E-01	6.440E-02

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
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NONE

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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ADDITIONAL DATA INPUT FOR THIS PROBLEM

0	0	-0	5.000E+02	-0.	-0.	-0.	-0.
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SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)

302A TAPERED THREE-SPAN OVERHANGING BEAM WITH 500 LB LOAD AT STA 0

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	1.903E-01			0.		0.
0	1.837E-01	0.	-6.640E-03	3.540E-09	0.	0.
1	1.753E-01	1.736E-03	-6.640E-03	5.000E+02	5.000E+02	0.
2	1.672E-01	1.568E-03	-6.573E-03	1.000E+03	5.000E+02	0.
3	1.593E-01	1.430E-03	-6.469E-03	1.500E+03	5.000E+02	0.
4	1.516E-01	1.321E-03	-6.344E-03	2.000E+03	5.000E+02	0.
5	1.442E-01	1.228E-03	-6.208E-03	2.500E+03	5.000E+02	0.
6	1.370E-01	1.147E-03	-6.065E-03	3.000E+03	5.000E+02	0.
7	1.300E-01	1.080E-03	-5.918E-03	3.500E+03	5.000E+02	0.
8	1.232E-01	1.020E-03	-5.770E-03	4.000E+03	5.000E+02	0.
9	1.166E-01	9.671E-04	-5.623E-03	4.500E+03	5.000E+02	0.
10	1.102E-01	9.191E-04	-5.476E-03	5.000E+03	5.000E+02	0.
11	1.040E-01	8.803E-04	-5.328E-03	5.500E+03	5.000E+02	0.
12	9.795E-02	8.446E-04	-5.183E-03	6.000E+03	5.000E+02	0.
13	9.210E-02	8.117E-04	-5.039E-03	6.500E+03	5.000E+02	0.
14	8.642E-02	7.812E-04	-4.896E-03	7.000E+03	5.000E+02	0.
15	8.091E-02	7.564E-04	-4.754E-03	7.500E+03	5.000E+02	0.
16	7.557E-02	7.331E-04	-4.614E-03	8.000E+03	5.000E+02	0.
17	7.038E-02	7.112E-04	-4.475E-03	8.500E+03	5.000E+02	0.
18	6.535E-02	6.906E-04	-4.337E-03	9.000E+03	5.000E+02	0.
19	6.048E-02	6.735E-04	-4.200E-03	9.500E+03	5.000E+02	0.
20	5.576E-02	6.572E-04	-4.063E-03	1.000E+04	5.000E+02	0.
21	5.119E-02	6.417E-04	-3.928E-03	1.050E+04	5.000E+02	0.
		6.269E-04	-3.793E-03		5.000E+02	

22	4.677E-02			1.100E+04		0.
		6.127E-04	-3.659E-03		5.000E+02	
23	4.250E-02			1.150E+04		0.
		6.021E-04	-3.525E-03		5.000E+02	
24	3.837E-02			1.200E+04		0.
		5.919E-04	-3.391E-03		5.000E+02	
25	3.439E-02			1.250E+04		0.
		5.819E-04	-3.257E-03		5.000E+02	
26	3.055E-02			1.300E+04		0.
		5.723E-04	-3.123E-03		5.000E+02	
27	2.685E-02			1.350E+04		0.
		5.631E-04	-2.990E-03		5.000E+02	
28	2.330E-02			1.400E+04		0.
		5.559E-04	-2.856E-03		5.000E+02	
29	1.989E-02			1.450E+04		0.
		5.489E-04	-2.721E-03		5.000E+02	
30	1.662E-02			1.500E+04		0.
		5.421E-04	-2.586E-03		5.000E+02	
31	1.349E-02			1.550E+04		0.
		5.355E-04	-2.450E-03		5.000E+02	
32	1.051E-02			1.600E+04		0.
		5.290E-04	-2.313E-03		5.000E+02	
33	7.664E-03			1.650E+04		0.
		5.227E-04	-2.175E-03		5.000E+02	
34	4.966E-03			1.700E+04		0.
		5.165E-04	-2.037E-03		5.000E+02	
35	2.412E-03			1.750E+04		0.
		5.142E-04	-1.898E-03		5.000E+02	
36	0.			1.800E+04		-1.250E+03
		-7.580E-04	-1.757E-03		-7.500E+02	
37	-9.888E-04			1.725E+04		0.
		-7.646E-04	-1.623E-03		-7.500E+02	
38	-1.847E-03			1.650E+04		0.
		-7.612E-04	-1.496E-03		-7.500E+02	
39	-2.581E-03			1.575E+04		0.
		-7.579E-04	-1.376E-03		-7.500E+02	
40	-3.199E-03			1.500E+04		0.
		-7.546E-04	-1.262E-03		-7.500E+02	
41	-3.706E-03			1.425E+04		0.
		-7.514E-04	-1.156E-03		-7.500E+02	
42	-4.111E-03			1.350E+04		0.
		-7.482E-04	-1.056E-03		-7.500E+02	
43	-4.419E-03			1.275E+04		0.
		-7.450E-04	-9.624E-04		-7.500E+02	
44	-4.636E-03			1.200E+04		0.
		-7.418E-04	-8.753E-04		-7.500E+02	
45	-4.770E-03			1.125E+04		0.
		-7.386E-04	-7.943E-04		-7.500E+02	
46	-4.825E-03			1.050E+04		0.
		-7.355E-04	-7.195E-04		-7.500E+02	
47	-4.809E-03			9.750E+03		0.
		-7.324E-04	-6.507E-04		-7.500E+02	
48	-4.728E-03			9.000E+03		0.
		-7.355E-04	-5.877E-04		-7.500E+02	
49	-4.580E-03			8.250E+03		0.
		-7.386E-04	-5.294E-04		-7.500E+02	
50	-4.370E-03			7.500E+03		0.
		-7.418E-04	-4.760E-04		-7.500E+02	
51	-4.105E-03			6.750E+03		0.
		-7.450E-04	-4.274E-04		-7.500E+02	
52	-3.787E-03			6.000E+03		0.

53	-3.423E-03	-7.482E-04	-3.838E-04	5.250E+03	-7.500E+02	0.
54	-3.017E-03	-7.514E-04	-3.453E-04	4.500E+03	-7.500E+02	0.
55	-2.574E-03	-7.546E-04	-3.120E-04	3.750E+03	-7.500E+02	0.
56	-2.100E-03	-7.579E-04	-2.840E-04	3.000E+03	-7.500E+02	0.
57	-1.600E-03	-7.612E-04	-2.614E-04	2.250E+03	-7.500E+02	0.
58	-1.080E-03	-7.646E-04	-2.442E-04	1.500E+03	-7.500E+02	0.
59	-5.445E-04	-7.680E-04	-2.327E-04	7.500E+02	-7.500E+02	0.
60	0.	-7.714E-04	-2.268E-04	-4.620E-09	-7.500E+02	7.500E+02
61	-2.268E-04	-2.167E-16	-2.268E-04	-4.918E-09	-2.098E-10	0.
62	-4.536E-04	-2.158E-16	-2.268E-04	-5.642E-09	-2.065E-10	0.
63	-6.805E-04	-2.132E-16	-2.268E-04	-5.451E-09	-2.015E-10	0.
64	-9.073E-04	-2.170E-16	-2.268E-04	-6.910E-09	-2.026E-10	0.
65	-1.134E-03	-2.206E-16	-2.268E-04	-7.501E-09	-2.034E-10	0.
66	-1.361E-03	-2.240E-16	-2.268E-04	-6.586E-09	-2.041E-10	0.
67	-1.588E-03	-2.210E-16	-2.268E-04	-7.984E-09	-1.988E-10	0.
68	-1.815E-03	-2.119E-16	-2.268E-04	-8.790E-09	-1.881E-10	0.
69	-2.041E-03	-1.987E-16	-2.268E-04	-1.114E-08	-1.736E-10	0.
70	-2.268E-03	-1.980E-16	-2.268E-04	-1.343E-08	-1.701E-10	0.
71	-2.495E-03	-1.974E-16	-2.268E-04	-7.883E-09	-1.667E-10	0.
72	-2.722E-03	-1.846E-16	-2.268E-04	-8.688E-09	-1.533E-10	0.
73	-2.949E-03	-1.843E-16	-2.268E-04	-8.688E-09	-1.504E-10	0.
74	-3.176E-03	-1.858E-16	-2.268E-04	-1.324E-08	-1.482E-10	0.
75	-3.402E-03	-1.874E-16	-2.268E-04	-1.036E-08	-1.460E-10	0.
76	-3.629E-03	-1.768E-16	-2.268E-04	-1.537E-08	-1.345E-10	0.
77	-3.856E-03	-1.663E-16	-2.268E-04	-1.151E-08	-1.234E-10	0.
78	-4.083E-03	-1.688E-16	-2.268E-04	-1.178E-08	-1.222E-10	0.
79	-4.310E-03	-1.591E-16	-2.268E-04	-1.213E-08	-1.119E-10	0.
80	-4.536E-03	-1.591E-16	-2.268E-04	-8.791E-09	-1.119E-10	0.
81	-4.763E-03	-1.631E-16	-2.268E-04	-6.655E-09	-1.112E-10	0.
82	-4.990E-03	-1.678E-16	-2.268E-04	-9.197E-09	-1.109E-10	0.
		-1.734E-16	-2.268E-04	-7.442E-09	-1.109E-10	0.
		-1.576E-16	-2.268E-04	-9.707E-11		

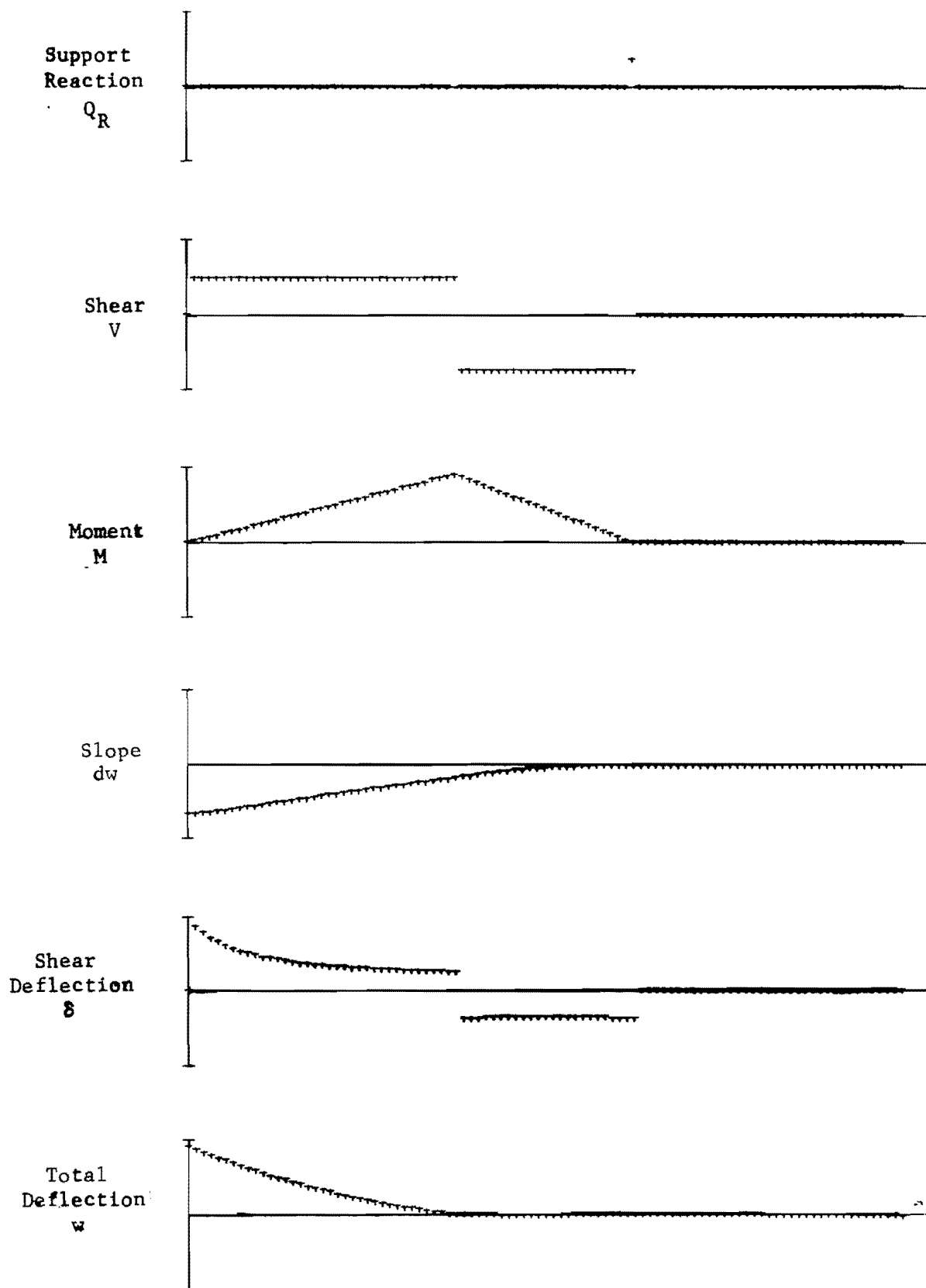
83	-5.217E-03			-1.778E-09		0.
		-1.411E-16	-2.268E-04		-8.354E-11	
84	-5.444E-03			-1.931E-09		0.
		-1.319E-16	-2.268E-04		-7.495E-11	
85	-5.671E-03			-4.225E-09		0.
		-1.049E-16	-2.268E-04		-5.708E-11	
86	-5.897E-03			7.227E-10		0.
		-8.365E-17	-2.268E-04		-4.324E-11	
87	-6.124E-03			-1.276E-09		0.
		-6.989E-17	-2.268E-04		-3.425E-11	
88	-6.351E-03			-6.259E-10		0.
		-6.519E-17	-2.268E-04		-3.018E-11	
89	-6.578E-03			-8.228E-10		0.
		-7.209E-17	-2.268E-04		-3.143E-11	
90	-6.805E-03			-1.573E-09		0.
		-6.617E-17	-2.268E-04		-2.695E-11	
91	-7.031E-03			-2.805E-09		0.
		-5.960E-17	-2.268E-04		-2.255E-11	
92	-7.258E-03			-2.212E-09		0.
		-4.317E-17	-2.268E-04		-1.509E-11	
93	-7.485E-03			-8.327E-10		0.
		-2.970E-17	-2.268E-04		-9.474E-12	
94	-7.712E-03			-4.849E-10		0.
		-2.366E-17	-2.268E-04		-6.822E-12	
95	-7.939E-03			5.140E-10		0.
		-1.220E-17	-2.268E-04		-3.142E-12	
96	-8.166E-03			3.239E-11		0.
		-5.166E-19	-2.268E-04		-1.331E-13	
97	-8.392E-03			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	1.000E+02	2.000E-01	2.000E-03	1.000E-02	2.000E+04	1.000E+03	2.000E+03

TIME FOR THIS PROBLEM = 0 MINUTES 3.555 SECONDS

ELAPSED CPU TIME = 0 MINUTES 30.500 SECONDS



Prob 302A. Tapered three-span overhanging beam
with 500-lb load at Sta 0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
302B SAME AS PROBLEM 302A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
36	0.
60	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	0	1.000E+07	-0.	-0.	-0.
1	96	0	-0.	4.000E+06	-0.	-0.
0	0	0	-0.	-0.	5.546E-01	-0.
1	1	1	-0.	-0.	7.421E-01	7.200E-02
	3	1	-0.	-0.	1.193E+00	8.740E-02
	6	1	-0.	-0.	2.034E+00	1.090E-01
	10	1	-0.	-0.	3.395E+00	1.360E-01
	14	1	-0.	-0.	4.937E+00	1.600E-01
	18	1	-0.	-0.	6.566E+00	1.810E-01
	23	1	-0.	-0.	8.592E+00	2.040E-01
	28	1	-0.	-0.	1.047E+01	2.220E-01
	35	1	-0.	-0.	1.260E+01	2.420E-01
	48	1	-0.	-0.	1.429E+01	2.560E-01
	61	1	-0.	-0.	1.260E+01	2.420E-01
	68	1	-0.	-0.	1.047E+01	2.220E-01
	73	1	-0.	-0.	8.592E+00	2.040E-01

78	1	-0.	-0.	6.566E+00	1.810E-01
82	1	-0.	-0.	4.937E+00	1.600E-01
86	1	-0.	-0.	3.395E+00	1.360E-01
90	1	-0.	-0.	2.034E+00	1.090E-01
93	1	-0.	-0.	1.193E+00	8.740E-02
96	-0	-0.	-0.	5.546E-01	6.440E-02

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
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ADDITIONAL DATA INPUT FOR THIS PROBLEM

1	96	-0	1.000E+99	-0.
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TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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DATA HELD FROM THE PREVIOUS PROBLEM

0	0	-0	5.000E+02	-0.	-0.	-0.	-0.
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SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
 302B SAME AS PROBLEM 302A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	1.335E-01			0.		0.
0	1.276E-01	0.	-5.888E-03	-3.694E-09	0.	0.
1	1.217E-01	5.000E-97	-5.888E-03	5.000E+02	5.000E+02	0.
2	1.159E-01	5.000E-97	-5.821E-03	1.000E+03	5.000E+02	0.
3	1.102E-01	5.000E-97	-5.718E-03	1.500E+03	5.000E+02	0.
4	1.046E-01	5.000E-97	-5.592E-03	2.000E+03	5.000E+02	0.
5	9.914E-02	5.000E-97	-5.456E-03	2.500E+03	5.000E+02	0.
6	9.383E-02	5.000E-97	-5.314E-03	3.000E+03	5.000E+02	0.
7	8.866E-02	5.000E-97	-5.166E-03	3.500E+03	5.000E+02	0.
8	8.365E-02	5.000E-97	-5.019E-03	4.000E+03	5.000E+02	0.
9	7.877E-02	5.000E-97	-4.871E-03	4.500E+03	5.000E+02	0.
10	7.405E-02	5.000E-97	-4.724E-03	5.000E+03	5.000E+02	0.
11	6.947E-02	5.000E-97	-4.577E-03	5.500E+03	5.000E+02	0.
12	6.504E-02	5.000E-97	-4.431E-03	6.000E+03	5.000E+02	0.
13	6.075E-02	5.000E-97	-4.287E-03	6.500E+03	5.000E+02	0.
14	5.661E-02	5.000E-97	-4.144E-03	7.000E+03	5.000E+02	0.
15	5.261E-02	5.000E-97	-4.003E-03	7.500E+03	5.000E+02	0.
16	4.875E-02	5.000E-97	-3.862E-03	8.000E+03	5.000E+02	0.
17	4.502E-02	5.000E-97	-3.723E-03	8.500E+03	5.000E+02	0.
18	4.144E-02	5.000E-97	-3.585E-03	9.000E+03	5.000E+02	0.
19	3.799E-02	5.000E-97	-3.448E-03	9.500E+03	5.000E+02	0.
20	3.468E-02	5.000E-97	-3.312E-03	1.000E+04	5.000E+02	0.
21	3.150E-02	5.000E-97	-3.176E-03	1.050E+04	5.000E+02	0.
		5.000E-97	-3.041E-03		5.000E+02	

22	2.846E-02			1.100E+04		0.
23	2.555E-02	5.000E-97	-2.907E-03	1.150E+04	5.000E+02	0.
24	2.278E-02	5.000E-97	-2.773E-03	1.200E+04	5.000E+02	0.
25	2.014E-02	5.000E-97	-2.639E-03	1.250E+04	5.000E+02	0.
26	1.763E-02	5.000E-97	-2.506E-03	1.300E+04	5.000E+02	0.
27	1.526E-02	5.000E-97	-2.372E-03	1.350E+04	5.000E+02	0.
28	1.302E-02	5.000E-97	-2.238E-03	1.400E+04	5.000E+02	0.
29	1.092E-02	5.000E-97	-2.104E-03	1.450E+04	5.000E+02	0.
30	8.950E-03	5.000E-97	-1.970E-03	1.500E+04	5.000E+02	0.
31	7.115E-03	5.000E-97	-1.834E-03	1.550E+04	5.000E+02	0.
32	5.417E-03	5.000E-97	-1.698E-03	1.600E+04	5.000E+02	0.
33	3.856E-03	5.000E-97	-1.561E-03	1.650E+04	5.000E+02	0.
34	2.432E-03	5.000E-97	-1.424E-03	1.700E+04	5.000E+02	0.
35	1.147E-03	5.000E-97	-1.285E-03	1.750E+04	5.000E+02	0.
36	0.	5.000E-97	-1.147E-03	1.800E+04	5.000E+02	-1.250E+03
37	-1.005E-03	-7.500E-97	-1.005E-03	1.725E+04	-7.500E+02	0.
38	-1.876E-03	-7.500E-97	-8.710E-04	1.650E+04	-7.500E+02	0.
39	-2.620E-03	-7.500E-97	-7.440E-04	1.575E+04	-7.500E+02	0.
40	-3.244E-03	-7.500E-97	-6.240E-04	1.500E+04	-7.500E+02	0.
41	-3.755E-03	-7.500E-97	-5.108E-04	1.425E+04	-7.500E+02	0.
42	-4.159E-03	-7.500E-97	-4.043E-04	1.350E+04	-7.500E+02	0.
43	-4.464E-03	-7.500E-97	-3.043E-04	1.275E+04	-7.500E+02	0.
44	-4.675E-03	-7.500E-97	-2.109E-04	1.200E+04	-7.500E+02	0.
45	-4.798E-03	-7.500E-97	-1.237E-04	1.125E+04	-7.500E+02	0.
46	-4.841E-03	-7.500E-97	-4.279E-05	1.050E+04	-7.500E+02	0.
47	-4.809E-03	-7.500E-97	3.205E-05	9.750E+03	-7.500E+02	0.
48	-4.708E-03	-7.500E-97	1.009E-04	9.000E+03	-7.500E+02	0.
49	-4.544E-03	-7.500E-97	1.639E-04	8.250E+03	-7.500E+02	0.
50	-4.322E-03	-7.500E-97	2.221E-04	7.500E+03	-7.500E+02	0.
51	-4.046E-03	-7.500E-97	2.756E-04	6.750E+03	-7.500E+02	0.
52	-3.722E-03	-7.500E-97	3.242E-04	6.000E+03	-7.500E+02	0.

53	-3.355E-03	-7.500E-97	3.677E-04	5.250E+03	-7.500E+02	0.
54	-2.948E-03	-7.500E-97	4.062E-04	4.500E+03	-7.500E+02	0.
55	-2.509E-03	-7.500E-97	4.395E-04	3.750E+03	-7.500E+02	0.
56	-2.041E-03	-7.500E-97	4.676E-04	3.000E+03	-7.500E+02	0.
57	-1.551E-03	-7.500E-97	4.902E-04	2.250E+03	-7.500E+02	0.
58	-1.044E-03	-7.500E-97	5.074E-04	1.500E+03	-7.500E+02	0.
59	-5.247E-04	-7.500E-97	5.189E-04	7.500E+02	-7.500E+02	0.
60	0.	-7.500E-97	5.247E-04	1.634E-08	-7.500E+02	7.500E+02
61	5.247E-04	4.657-108	5.247E-04	2.098E-08	4.657E-09	0.
62	1.049E-03	3.491-108	5.247E-04	2.474E-08	3.491E-09	0.
63	1.574E-03	3.016-108	5.247E-04	2.663E-08	3.016E-09	0.
64	2.099E-03	3.048-108	5.247E-04	2.919E-08	3.048E-09	0.
65	2.624E-03	2.535-108	5.247E-04	2.919E-08	2.535E-09	0.
66	3.148E-03	2.187-108	5.247E-04	3.317E-08	2.187E-09	0.
67	3.673E-03	1.821-108	5.247E-04	3.075E-08	1.821E-09	0.
68	4.198E-03	1.744-108	5.247E-04	3.589E-08	1.744E-09	0.
69	4.723E-03	1.527-108	5.247E-04	4.068E-08	1.527E-09	0.
70	5.247E-03	1.295-108	5.247E-04	2.802E-08	1.295E-09	0.
71	5.772E-03	1.117-108	5.247E-04	3.237E-08	1.117E-09	0.
72	6.297E-03	9.767-109	5.247E-04	3.371E-08	9.767E-10	0.
73	6.822E-03	9.767-109	5.247E-04	2.240E-08	8.647E-10	0.
74	7.346E-03	8.647-109	5.247E-04	3.100E-08	8.647E-10	0.
75	7.871E-03	8.535-109	5.247E-04	3.181E-08	8.535E-10	0.
76	8.396E-03	8.431-109	5.247E-04	3.181E-08	8.431E-10	0.
77	8.920E-03	7.700-109	5.247E-04	3.024E-08	7.700E-10	0.
78	9.445E-03	7.087-109	5.247E-04	2.047E-08	7.087E-10	0.
79	9.970E-03	6.022-109	5.247E-04	2.322E-08	6.022E-10	0.
80	1.049E-02	6.022-109	5.247E-04	1.822E-08	6.022E-10	0.
81	1.102E-02	5.113-109	5.247E-04	1.822E-08	5.113E-10	0.
82	1.154E-02	4.326-109	5.247E-04	1.709E-08	4.326E-10	0.
		3.184-109	5.247E-04	1.277E-08	3.184E-10	0.
		3.466-109	5.247E-04	1.137E-08	3.466E-10	0.
		3.312-109	5.247E-04	-7.617-103	3.312E-10	0.

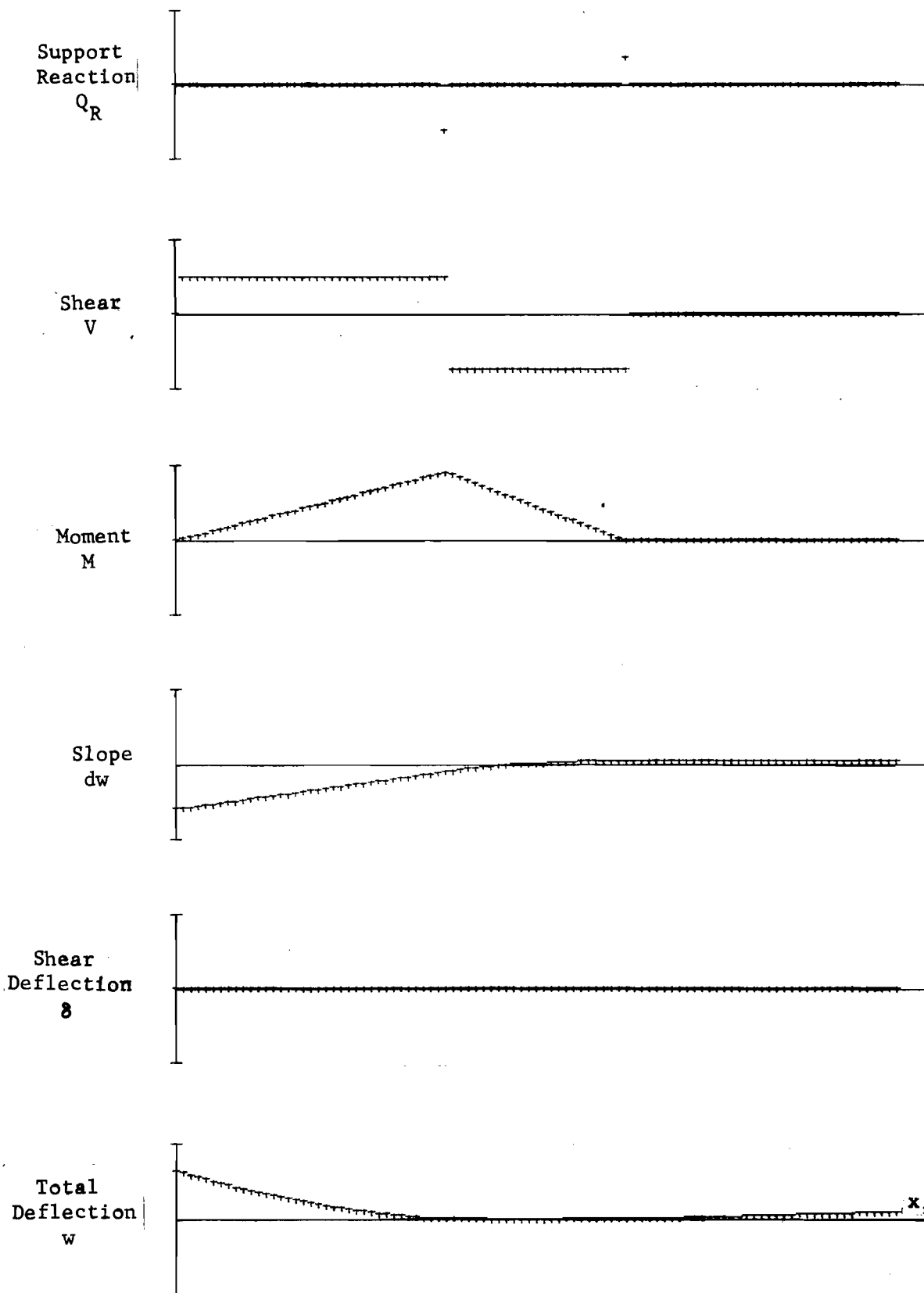
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85	1.312E-02	2.705-109	5.247E-04	-1.049E-08	2.705E-10	0.
86	1.364E-02	2.267-109	5.247E-04	-1.508E-08	2.267E-10	0.
87	1.417E-02	2.026-109	5.247E-04	-1.187E-08	2.026E-10	0.
88	1.469E-02	1.451-109	5.247E-04	-1.356E-08	1.451E-10	0.
89	1.522E-02	9.038-110	5.247E-04	-1.845E-08	9.038E-11	0.
90	1.574E-02	7.303-110	5.247E-04	-9.033E-09	7.303E-11	0.
91	1.627E-02	5.477-110	5.247E-04	-3.894E-09	5.477E-11	0.
92	1.679E-02	5.577-110	5.247E-04	-3.271E-09	5.577E-11	0.
93	1.732E-02	5.815-110	5.247E-04	-1.324E-09	5.815E-11	0.
94	1.784E-02	4.949-110	5.247E-04	-2.176E-09	4.949E-11	0.
95	1.837E-02	4.178-110	5.247E-04	-1.704E-09	4.178E-11	0.
96	1.889E-02	1.823-110	5.247E-04	-4.866-104	1.823E-11	0.
97	1.942E-02	6.836-112	5.247E-04	0.	6.836E-13	0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	1.000E+02	2.000E-01	1.000E+00	1.000E-02	2.000E+04	1.000E+03	2.000E+03

TIME FOR THIS PROBLEM = 0 MINUTES 3.564 SECONDS

ELAPSED CPU TIME = 0 MINUTES 34.064 SECONDS



Prob 302B. Same as Prob 302A, neglecting shear deformations.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
303A TAPERED THREE-SPAN OVERHANGING BEAM WITH 700 LB LOAD AT STA 0

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	0
NUM CARDS INPUT THIS PROBLEM	0	0	0	0	1
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
36	0.
60	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	0	1.000E+07	-0.	-0.	-0.
1	96	0	-0.	4.000E+06	-0.	-0.
0	0	0	-0.	-0.	5.546E-01	-0.
1	1	1	-0.	-0.	7.421E-01	7.200E-02
3	1	1	-0.	-0.	1.193E+00	8.740E-02
6	1	1	-0.	-0.	2.034E+00	1.090E-01
10	1	1	-0.	-0.	3.395E+00	1.360E-01
14	1	1	-0.	-0.	4.937E+00	1.600E-01
18	1	1	-0.	-0.	6.566E+00	1.810E-01
23	1	1	-0.	-0.	8.592E+00	2.040E-01
28	1	1	-0.	-0.	1.047E+01	2.220E-01
35	1	1	-0.	-0.	1.260E+01	2.420E-01
43	1	1	-0.	-0.	1.429E+01	2.560E-01
51	1	1	-0.	-0.	1.260E+01	2.420E-01
58	1	1	-0.	-0.	1.047E+01	2.220E-01
73	1	1	-0.	-0.	8.592E+00	2.040E-01

78	1	-0.	-0.	6.566E+00	1.810E-01
82	1	-0.	-0.	4.937E+00	1.600E-01
86	1	-0.	-0.	3.395E+00	1.360E-01
90	1	-0.	-0.	2.034E+00	1.090E-01
93	1	-0.	-0.	1.193E+00	8.740E-02
96	-0	-0.	-0.	5.546E-01	6.440E-02

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
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NONE

TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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ADDITIONAL DATA INPUT FOR THIS PROBLEM							
0	0	-0	7.000E+02	-0.	-0.	-0.	-0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
303A TAPERED THREE-SPAN OVERHANGING BEAM WITH 700 LB LOAD AT STA 0

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	2.665E-01	0.	-9.296E-03	0.	0.	0.
0	2.572E-01	2.431E-03	-9.296E-03	-7.851E-09	7.000E+02	0.
1	2.454E-01	2.196E-03	-9.202E-03	7.000E+02	7.000E+02	0.
2	2.340E-01	2.002E-03	-9.057E-03	1.400E+03	7.000E+02	0.
3	2.230E-01	1.850E-03	-8.881E-03	2.100E+03	7.000E+02	0.
4	2.123E-01	1.719E-03	-8.691E-03	2.800E+03	7.000E+02	0.
5	2.018E-01	1.606E-03	-8.491E-03	3.500E+03	7.000E+02	0.
6	1.917E-01	1.512E-03	-8.285E-03	4.200E+03	7.000E+02	0.
7	1.819E-01	1.429E-03	-8.078E-03	4.900E+03	7.000E+02	0.
8	1.724E-01	1.354E-03	-7.872E-03	5.600E+03	7.000E+02	0.
9	1.632E-01	1.287E-03	-7.666E-03	6.300E+03	7.000E+02	0.
10	1.543E-01	1.232E-03	-7.460E-03	7.000E+03	7.000E+02	0.
11	1.456E-01	1.182E-03	-7.256E-03	7.700E+03	7.000E+02	0.
12	1.371E-01	1.136E-03	-7.054E-03	8.400E+03	7.000E+02	0.
13	1.289E-01	1.094E-03	-6.854E-03	9.100E+03	7.000E+02	0.
14	1.210E-01	1.059E-03	-6.656E-03	9.800E+03	7.000E+02	0.
15	1.133E-01	1.026E-03	-6.459E-03	1.050E+04	7.000E+02	0.
16	1.058E-01	9.957E-04	-6.265E-03	1.120E+04	7.000E+02	0.
17	9.853E-02	9.669E-04	-6.072E-03	1.190E+04	7.000E+02	0.
18	9.149E-02	9.429E-04	-5.880E-03	1.260E+04	7.000E+02	0.
19	8.467E-02	9.201E-04	-5.689E-03	1.330E+04	7.000E+02	0.
20	7.806E-02	8.984E-04	-5.499E-03	1.400E+04	7.000E+02	0.
21	7.167E-02	8.776E-04	-5.310E-03	1.470E+04	7.000E+02	0.

22	6.548E-02			1.540E+04		0.
		8.578E-04	-5.122E-03		7.000E+02	
23	5.950E-02			1.610E+04		0.
		8.430E-04	-4.935E-03		7.000E+02	
24	5.372E-02			1.680E+04		0.
		8.286E-04	-4.747E-03		7.000E+02	
25	4.814E-02			1.750E+04		0.
		8.147E-04	-4.560E-03		7.000E+02	
26	4.277E-02			1.820E+04		0.
		8.013E-04	-4.373E-03		7.000E+02	
27	3.760E-02			1.890E+04		0.
		7.883E-04	-4.186E-03		7.000E+02	
28	3.262E-02			1.960E+04		0.
		7.783E-04	-3.998E-03		7.000E+02	
29	2.785E-02			2.030E+04		0.
		7.685E-04	-3.810E-03		7.000E+02	
30	2.327E-02			2.100E+04		0.
		7.590E-04	-3.620E-03		7.000E+02	
31	1.889E-02			2.170E+04		0.
		7.497E-04	-3.430E-03		7.000E+02	
32	1.471E-02			2.240E+04		0.
		7.406E-04	-3.238E-03		7.000E+02	
33	1.073E-02			2.310E+04		0.
		7.318E-04	-3.045E-03		7.000E+02	
34	6.952E-03			2.380E+04		0.
		7.231E-04	-2.852E-03		7.000E+02	
35	3.377E-03			2.450E+04		0.
		7.199E-04	-2.657E-03		7.000E+02	
36	0.			2.520E+04		-1.750E+03
		-1.075E-03	-2.459E-03		-1.050E+03	
37	-1.384E-03			2.415E+04		0.
		-1.070E-03	-2.272E-03		-1.050E+03	
38	-2.586E-03			2.310E+04		0.
		-1.066E-03	-2.094E-03		-1.050E+03	
39	-3.614E-03			2.205E+04		0.
		-1.061E-03	-1.926E-03		-1.050E+03	
40	-4.478E-03			2.100E+04		0.
		-1.057E-03	-1.767E-03		-1.050E+03	
41	-5.189E-03			1.995E+04		0.
		-1.052E-03	-1.618E-03		-1.050E+03	
42	-5.755E-03			1.890E+04		0.
		-1.047E-03	-1.478E-03		-1.050E+03	
43	-6.186E-03			1.785E+04		0.
		-1.043E-03	-1.347E-03		-1.050E+03	
44	-6.491E-03			1.680E+04		0.
		-1.038E-03	-1.225E-03		-1.050E+03	
45	-6.677E-03			1.575E+04		0.
		-1.034E-03	-1.112E-03		-1.050E+03	
46	-6.755E-03			1.470E+04		0.
		-1.030E-03	-1.007E-03		-1.050E+03	
47	-6.733E-03			1.365E+04		0.
		-1.025E-03	-9.109E-04		-1.050E+03	
48	-6.619E-03			1.260E+04		0.
		-1.030E-03	-8.227E-04		-1.050E+03	
49	-6.412E-03			1.155E+04		0.
		-1.034E-03	-7.412E-04		-1.050E+03	
50	-6.119E-03			1.050E+04		0.
		-1.038E-03	-6.663E-04		-1.050E+03	
51	-5.747E-03			9.450E+03		0.
		-1.043E-03	-5.983E-04		-1.050E+03	
52	-5.302E-03			8.400E+03		0.

53	-4.792E-03	-1.047E-03	-5.373E-04	7.350E+03	-1.050E+03	0.
54	-4.223E-03	-1.052E-03	-4.835E-04	6.300E+03	-1.050E+03	0.
55	-3.604E-03	-1.057E-03	-4.368E-04	5.250E+03	-1.050E+03	0.
56	-2.940E-03	-1.061E-03	-3.976E-04	4.200E+03	-1.050E+03	0.
57	-2.240E-03	-1.066E-03	-3.659E-04	3.150E+03	-1.050E+03	0.
58	-1.512E-03	-1.070E-03	-3.419E-04	2.100E+03	-1.050E+03	0.
59	-7.624E-04	-1.075E-03	-3.257E-04	1.050E+03	-1.050E+03	0.
60	0.	-1.080E-03	-3.176E-04	-7.560E-09	-1.050E+03	1.050E+03
61	-3.176E-04	-1.843E-16	-3.176E-04	-7.170E-09	-1.784E-10	0.
62	-6.351E-04	-1.787E-16	-3.176E-04	-7.815E-09	-1.710E-10	0.
63	-9.527E-04	-1.798E-16	-3.176E-04	-8.042E-09	-1.700E-10	0.
64	-1.270E-03	-1.775E-16	-3.176E-04	-8.593E-09	-1.658E-10	0.
65	-1.588E-03	-1.817E-16	-3.176E-04	-8.593E-09	-1.675E-10	0.
66	-1.905E-03	-1.856E-16	-3.176E-04	-7.560E-09	-1.691E-10	0.
67	-2.223E-03	-1.768E-16	-3.176E-04	-9.787E-09	-1.590E-10	0.
68	-2.540E-03	-1.869E-16	-3.176E-04	-1.156E-08	-1.660E-10	0.
69	-2.858E-03	-1.863E-16	-3.176E-04	-1.446E-08	-1.627E-10	0.
70	-3.176E-03	-1.980E-16	-3.176E-04	-1.799E-08	-1.701E-10	0.
71	-3.493E-03	-1.850E-16	-3.176E-04	-1.763E-08	-1.563E-10	0.
72	-3.811E-03	-1.846E-16	-3.176E-04	-1.163E-08	-1.533E-10	0.
73	-4.128E-03	-1.491E-08	-3.176E-04	-1.491E-08	-1.504E-10	0.
74	-4.446E-03	-1.343E-16	-3.176E-04	-1.707E-08	-1.284E-10	0.
75	-4.763E-03	-1.610E-16	-3.176E-04	-1.375E-08	-1.266E-10	0.
76	-5.081E-03	-1.624E-16	-3.176E-04	-1.859E-08	-1.153E-10	0.
77	-5.398E-03	-1.515E-16	-3.176E-04	-1.259E-08	-9.494E-11	0.
78	-5.716E-03	-1.279E-16	-3.176E-04	-1.664E-08	-8.458E-11	0.
79	-6.033E-03	-1.168E-16	-3.176E-04	-1.483E-08	-6.526E-11	0.
80	-6.351E-03	-9.284E-17	-3.176E-04	-9.527E-09	-7.416E-11	0.
81	-6.669E-03	-1.087E-16	-3.176E-04	-9.758E-09	-7.394E-11	0.
82	-6.986E-03	-1.119E-16	-3.176E-04	-1.284E-08	-8.321E-11	0.
		-1.300E-16	-3.176E-04	-1.058E-08	-9.244E-11	0.
		-1.501E-16	-3.176E-04			

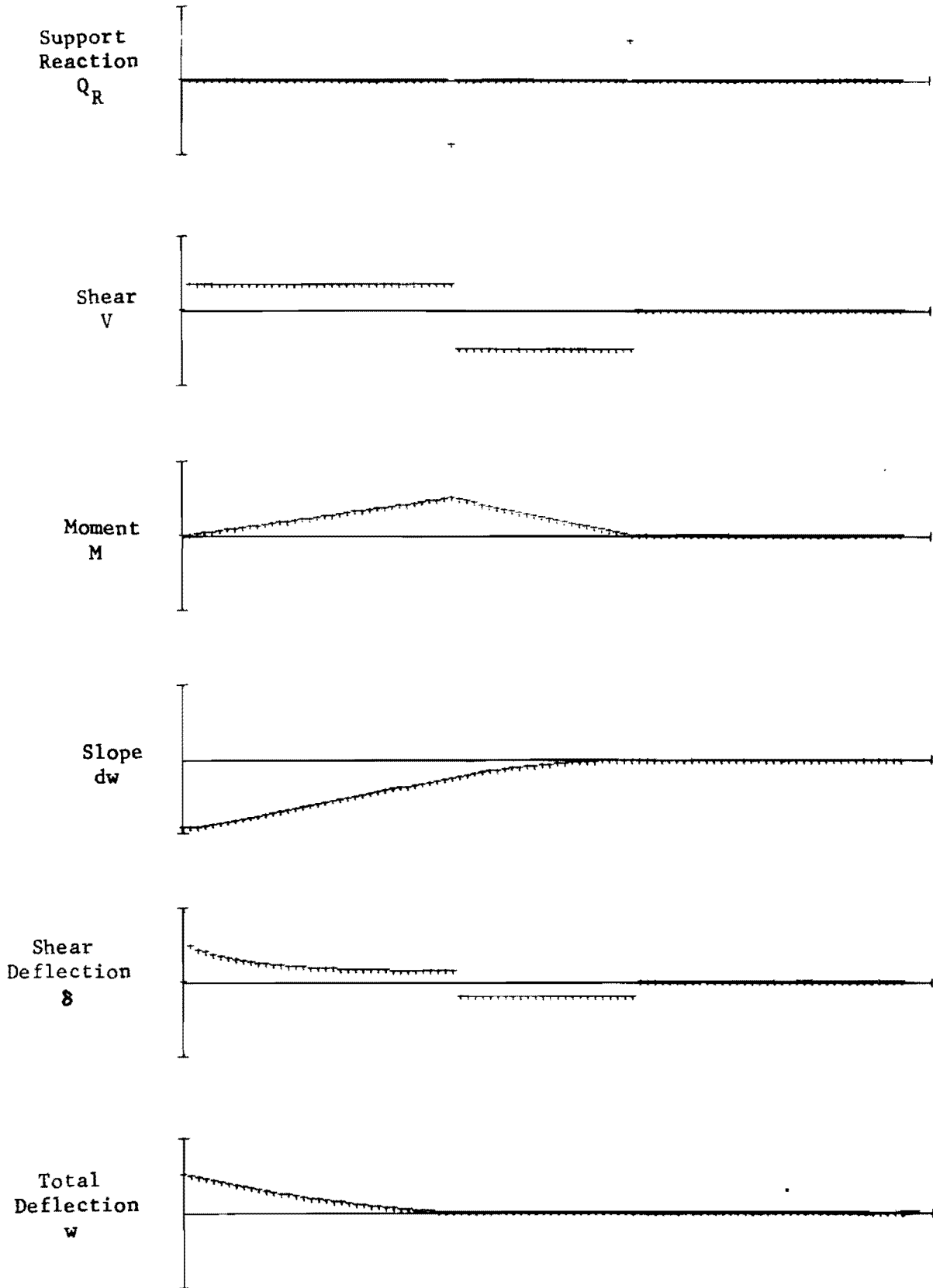
83	-7.304E-03			-4.646E-09		0.
		-1.411E-16	-3.176E-04		-8.354E-11	
84	-7.621E-03			-6.556E-09		0.
		-1.319E-16	-3.176E-04		-7.495E-11	
85	-7.939E-03			-8.034E-09		0.
		-1.224E-16	-3.176E-04		-6.660E-11	
86	-8.256E-03			-2.768E-09		0.
		-9.294E-17	-3.176E-04		-4.805E-11	
87	-8.574E-03			-4.212E-09		0.
		-1.198E-16	-3.176E-04		-5.871E-11	
88	-8.891E-03			-1.204E-09		0.
		-1.087E-16	-3.176E-04		-5.031E-11	
89	-9.209E-03			5.828E-10		0.
		-8.411E-17	-3.176E-04		-3.667E-11	
90	-9.527E-03			-1.893E-09		0.
		-6.617E-17	-3.176E-04		-2.695E-11	
91	-9.844E-03			-3.779E-09		0.
		-5.960E-17	-3.176E-04		-2.255E-11	
92	-1.016E-02			-1.521E-09		0.
		-5.181E-17	-3.176E-04		-1.811E-11	
93	-1.048E-02			-1.959E-09		0.
		-4.951E-17	-3.176E-04		-1.579E-11	
94	-1.080E-02			-9.509E-10		0.
		-3.550E-17	-3.176E-04		-1.023E-11	
95	-1.111E-02			6.048E-10		0.
		-1.220E-17	-3.176E-04		-3.142E-12	
96	-1.143E-02			1.863E-10		0.
		-5.166E-19	-3.176E-04		-1.331E-13	
97	-1.175E-02			0.		0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	1.000E+02	5.000E-01	5.000E-03	1.000E-02	5.000E+04	2.000E+03	2.000E+03

TIME FOR THIS PROBLEM = 0 MINUTES 3.527 SECONDS

ELAPSED CPU TIME = 0 MINUTES 37.591 SECONDS



Prob 303A. Tapered three-span overhanging beam with 700-lb load at Sta 0.

SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB
303B SAME AS PROBLEM 303A - ADD INFINITE SHEAR STIFFNESS

TABLE 1 - PROGRAM-CONTROL DATA

	TABLES NUMBER				
	2	3	4A	4B	5
PRIOR-DATA OPTIONS (1 = HOLD)	1	1	1	0	1
NUM CARDS INPUT THIS PROBLEM	0	0	0	1	0
OPTION (IF=1, 2, 3) TO PLOT	1				

TABLE 2 - CONSTANTS

DATA HELD FROM THE PREVIOUS PROBLEM	
NUM INCREMENTS	96
INCREMENT LENGTH	1.000E+00

TABLE 3 - SPECIFIED DEFLECTIONS

STA	DEFLECTION
DATA HELD FROM THE PREVIOUS PROBLEM	
36	0.
60	0.

TABLE 4A - SECTION AND MATERIAL PROPERTIES

FROM	TO	CONTD	E	G	I	SA
DATA HELD FROM THE PREVIOUS PROBLEM						
0	96	0	1.000E+07	-0.	-0.	-0.
1	96	0	-0.	4.000E+06	-0.	-0.
0	0	0	-0.	-0.	5.546E-01	-0.
1	1	1	-0.	-0.	7.421E-01	7.200E-02
	3	1	-0.	-0.	1.193E+00	8.740E-02
	6	1	-0.	-0.	2.034E+00	1.090E-01
	10	1	-0.	-0.	3.395E+00	1.360E-01
	14	1	-0.	-0.	4.937E+00	1.600E-01
	18	1	-0.	-0.	6.566E+00	1.810E-01
	23	1	-0.	-0.	8.592E+00	2.040E-01
	28	1	-0.	-0.	1.047E+01	2.220E-01
	35	1	-0.	-0.	1.260E+01	2.420E-01
	48	1	-0.	-0.	1.429E+01	2.560E-01
	61	1	-0.	-0.	1.260E+01	2.420E-01
	68	1	-0.	-0.	1.047E+01	2.220E-01
	73	1	-0.	-0.	8.592E+00	2.040E-01

78	1	-0.	-0.	6.566E+00	1.810E-01
82	1	-0.	-0.	4.937E+00	1.600E-01
86	1	-0.	-0.	3.395E+00	1.360E-01
90	1	-0.	-0.	2.034E+00	1.090E-01
93	1	-0.	-0.	1.193E+00	8.740E-02
96	-0	-0.	-0.	5.546E-01	6.440E-02

TABLE 4B - STIFFNESS DATA

FROM	TO	CONTD	SK	F
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ADDITIONAL DATA INPUT FOR THIS PROBLEM

1	96	-0	1.000E+99	-0.
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TABLE 5 - LOAD DATA

FROM	TO	CONTD	Q	S	T	R	P
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DATA HELD FROM THE PREVIOUS PROBLEM

0	0	-0	7.000E+02	-0.	-0.	-0.	-0.
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SAMPLE INPUT FOR EXAMPLE PROBLEMS FOR SHRBM 1 - APPENDIX 6

PROB (CONTD)
 303B SAME AS PROBLEM 303A - ADD INFINITE SHEAR STIFFNESS

TABLE 6 - RESULTS

STA	W	DELTA	SLOPE	MOM	SHEAR	SUP REACT
-1	1.869E-01	0.	-8.244E-03	0.	0.	0.
0	1.787E-01	7.000E-97	-8.244E-03	1.941E-90	7.000E+02	0.
1	1.704E-01	7.000E-97	-8.149E-03	7.000E+02	7.000E+02	0.
2	1.623E-01	7.000E-97	-8.005E-03	1.400E+03	7.000E+02	0.
3	1.543E-01	7.000E-97	-7.829E-03	2.100E+03	7.000E+02	0.
4	1.464E-01	7.000E-97	-7.639E-03	2.800E+03	7.000E+02	0.
5	1.388E-01	7.000E-97	-7.439E-03	3.500E+03	7.000E+02	0.
6	1.314E-01	7.000E-97	-7.233E-03	4.200E+03	7.000E+02	0.
7	1.241E-01	7.000E-97	-7.026E-03	4.900E+03	7.000E+02	0.
8	1.171E-01	7.000E-97	-6.820E-03	5.600E+03	7.000E+02	0.
9	1.103E-01	7.000E-97	-6.614E-03	6.300E+03	7.000E+02	0.
10	1.037E-01	7.000E-97	-6.408E-03	7.000E+03	7.000E+02	0.
11	9.726E-02	7.000E-97	-6.204E-03	7.700E+03	7.000E+02	0.
12	9.106E-02	7.000E-97	-6.002E-03	8.400E+03	7.000E+02	0.
13	8.506E-02	7.000E-97	-5.802E-03	9.100E+03	7.000E+02	0.
14	7.925E-02	7.000E-97	-5.604E-03	9.800E+03	7.000E+02	0.
15	7.365E-02	7.000E-97	-5.407E-03	1.050E+04	7.000E+02	0.
16	6.824E-02	7.000E-97	-5.213E-03	1.120E+04	7.000E+02	0.
17	6.303E-02	7.000E-97	-5.019E-03	1.190E+04	7.000E+02	0.
18	5.801E-02	7.000E-97	-4.827E-03	1.260E+04	7.000E+02	0.
19	5.318E-02	7.000E-97	-4.637E-03	1.330E+04	7.000E+02	0.
20	4.855E-02	7.000E-97	-4.447E-03	1.400E+04	7.000E+02	0.
21	4.410E-02	7.000E-97	-4.258E-03	1.470E+04	7.000E+02	0.

22	3.984E-02			1.540E+04		0.
		7.000E-97	-4.070E-03		7.000E+02	
23	3.577E-02			1.610E+04		0.
		7.000E-97	-3.882E-03		7.000E+02	
24	3.189E-02			1.680E+04		0.
		7.000E-97	-3.695E-03		7.000E+02	
25	2.820E-02			1.750E+04		0.
		7.000E-97	-3.508E-03		7.000E+02	
26	2.469E-02			1.820E+04		0.
		7.000E-97	-3.321E-03		7.000E+02	
27	2.137E-02			1.890E+04		0.
		7.000E-97	-3.133E-03		7.000E+02	
28	1.823E-02			1.960E+04		0.
		7.000E-97	-2.946E-03		7.000E+02	
29	1.529E-02			2.030E+04		0.
		7.000E-97	-2.758E-03		7.000E+02	
30	1.253E-02			2.100E+04		0.
		7.000E-97	-2.568E-03		7.000E+02	
31	9.961E-03			2.170E+04		0.
		7.000E-97	-2.378E-03		7.000E+02	
32	7.584E-03			2.240E+04		0.
		7.000E-97	-2.186E-03		7.000E+02	
33	5.398E-03			2.310E+04		0.
		7.000E-97	-1.993E-03		7.000E+02	
34	3.405E-03			2.380E+04		0.
		7.000E-97	-1.800E-03		7.000E+02	
35	1.605E-03			2.450E+04		0.
		7.000E-97	-1.605E-03		7.000E+02	
36	0.			2.520E+04		-1.750E+03
		-1.050E-96	-1.407E-03		-1.050E+03	
37	-1.407E-03			2.415E+04		0.
		-1.050E-96	-1.219E-03		-1.050E+03	
38	-2.627E-03			2.310E+04		0.
		-1.050E-96	-1.042E-03		-1.050E+03	
39	-3.668E-03			2.205E+04		0.
		-1.050E-96	-8.736E-04		-1.050E+03	
40	-4.542E-03			2.100E+04		0.
		-1.050E-96	-7.151E-04		-1.050E+03	
41	-5.257E-03			1.995E+04		0.
		-1.050E-96	-5.660E-04		-1.050E+03	
42	-5.823E-03			1.890E+04		0.
		-1.050E-96	-4.261E-04		-1.050E+03	
43	-6.249E-03			1.785E+04		0.
		-1.050E-96	-2.952E-04		-1.050E+03	
44	-6.544E-03			1.680E+04		0.
		-1.050E-96	-1.732E-04		-1.050E+03	
45	-6.718E-03			1.575E+04		0.
		-1.050E-96	-5.991E-05		-1.050E+03	
46	-6.777E-03			1.470E+04		0.
		-1.050E-96	4.487E-05		-1.050E+03	
47	-6.733E-03			1.365E+04		0.
		-1.050E-96	1.413E-04		-1.050E+03	
48	-6.591E-03			1.260E+04		0.
		-1.050E-96	2.294E-04		-1.050E+03	
49	-6.362E-03			1.155E+04		0.
		-1.050E-96	3.110E-04		-1.050E+03	
50	-6.051E-03			1.050E+04		0.
		-1.050E-96	3.858E-04		-1.050E+03	
51	-5.665E-03			9.450E+03		0.
		-1.050E-96	4.538E-04		-1.050E+03	
52	-5.211E-03			8.400E+03		0.

53	-4.696E-03	-1.050E-96	5.148E-04	7.350E+03	-1.050E+03	0.
54	-4.128E-03	-1.050E-96	5.687E-04	6.300E+03	-1.050E+03	0.
55	-3.512E-03	-1.050E-96	6.154E-04	5.250E+03	-1.050E+03	0.
56	-2.858E-03	-1.050E-96	6.546E-04	4.200E+03	-1.050E+03	0.
57	-2.171E-03	-1.050E-96	6.863E-04	3.150E+03	-1.050E+03	0.
58	-1.461E-03	-1.050E-96	7.103E-04	2.100E+03	-1.050E+03	0.
59	-7.346E-04	-1.050E-96	7.265E-04	1.050E+03	-1.050E+03	0.
60	0.	-1.050E-96	7.346E-04	1.678E-08	-1.050E+03	1.050E+03
61	7.346E-04	8.382-108	7.346E-04	2.623E-08	8.382E-09	0.
62	1.469E-03	6.983-108	7.346E-04	3.413E-08	6.983E-09	0.
63	2.204E-03	5.600-108	7.346E-04	3.828E-08	5.600E-09	0.
64	2.939E-03	5.543-108	7.346E-04	4.055E-08	5.543E-09	0.
65	3.673E-03	4.290-108	7.346E-04	4.423E-08	4.290E-09	0.
66	4.408E-03	4.373-108	7.346E-04	5.227E-08	4.373E-09	0.
67	5.142E-03	3.869-108	7.346E-04	5.084E-08	3.869E-09	0.
68	5.877E-03	3.489-108	7.346E-04	5.812E-08	3.489E-09	0.
69	6.612E-03	3.206-108	7.346E-04	4.483E-08	3.206E-09	0.
70	7.346E-03	2.849-108	7.346E-04	4.856E-08	2.849E-09	0.
71	8.081E-03	2.457-108	7.346E-04	5.705E-08	2.457E-09	0.
72	8.816E-03	2.051-108	7.346E-04	3.485E-08	2.051E-09	0.
73	9.550E-03	1.902-108	7.346E-04	4.770E-08	1.902E-09	0.
74	1.028E-02	1.862-108	7.346E-04	4.999E-08	1.862E-09	0.
75	1.102E-02	1.546-108	7.346E-04	3.024E-08	1.546E-09	0.
76	1.175E-02	1.347-108	7.346E-04	2.457E-08	1.347E-09	0.
77	1.249E-02	1.181-108	7.346E-04	3.096E-08	1.181E-09	0.
78	1.322E-02	1.040-108	7.346E-04	2.916E-08	1.040E-09	0.
79	1.396E-02	1.023-108	7.346E-04	2.735E-08	1.023E-09	0.
80	1.469E-02	8.171-109	7.346E-04	1.277E-08	8.171E-10	0.
81	1.543E-02	7.278-109	7.346E-04	1.780E-08	7.278E-10	0.
82	1.616E-02	6.066-109	7.346E-04	-5.481E-09	6.066E-10	0.
		4.968-109	7.346E-04		4.968E-10	

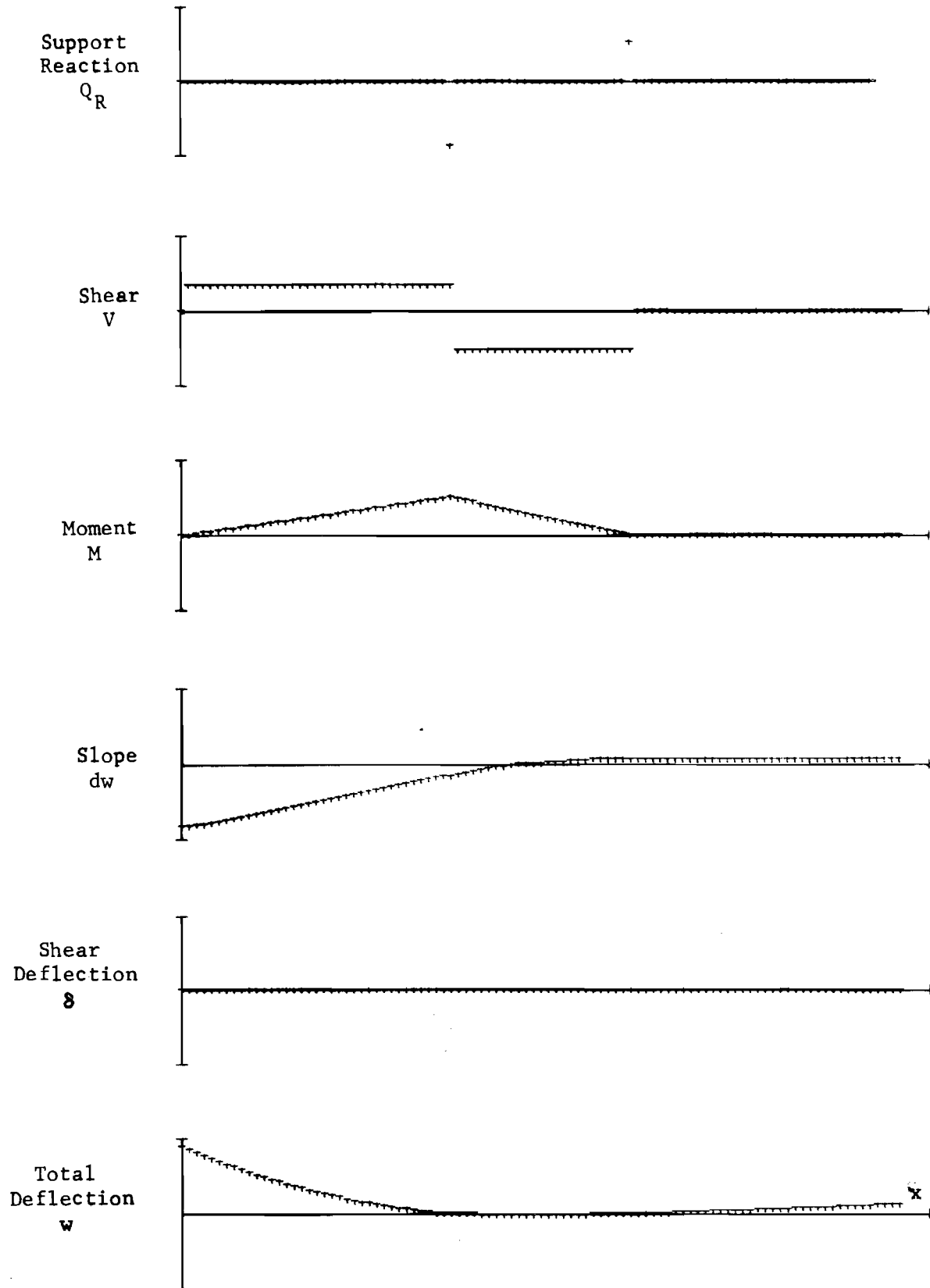
83	1.690E-02			-2.021E-08		0.
84	1.763E-02	3.985-109	7.346E-04	-9.250E-09	3.985E-10	0.
85	1.837E-02	3.478-109	7.346E-04	-1.259E-08	3.478E-10	0.
86	1.910E-02	3.022-109	7.346E-04	-2.262E-08	3.022E-10	0.
87	1.983E-02	2.947-109	7.346E-04	-1.357E-08	2.947E-10	0.
88	2.057E-02	2.540-109	7.346E-04	-1.808E-08	2.540E-10	0.
89	2.130E-02	2.169-109	7.346E-04	-2.109E-08	2.169E-10	0.
90	2.204E-02	2.191-109	7.346E-04	-1.355E-08	2.191E-10	0.
91	2.277E-02	1.460-109	7.346E-04	-1.363E-08	1.460E-10	0.
92	2.351E-02	9.296-110	7.346E-04	-8.179E-09	9.296E-11	0.
93	2.424E-02	7.754-110	7.346E-04	-1.324E-09	7.754E-11	0.
94	2.498E-02	5.938-110	7.346E-04	-7.014-104	5.938E-11	0.
95	2.571E-02	5.223-110	7.346E-04	-3.408E-09	5.223E-11	0.
96	2.645E-02	9.116-111	7.346E-04	-6.157E-10	9.116E-12	0.
97	2.718E-02	0.	7.346E-04	0.	0.	0.

TABLE 7 -- SCALES FOR PLOT OUTPUT -- VALUES AT ENDS OF AXES

STA	X	W	DELTA	DW	M	V	SUP REACT
100	1.000E+02	2.000E-01	1.000E+00	1.000E-02	5.000E+04	2.000E+03	2.000E+03

TIME FOR THIS PROBLEM = 0 MINUTES 3.569 SECONDS

ELAPSED CPU TIME = 0 MINUTES 41.160 SECONDS



Prob 303B. Same as Prob 303A, neglecting shear deformations.